11.1 11.6 GIVEN: X=4t4-6t3+2t-1 X-m. t-s GIVEN: X=3t3-6t2-12t+5 X-m, t-5 FIND: X, N, AND Q AT t= 25 FIND: (a) t WHEN 5=0 (b) X, Q, TOTAL DISTANCES TRAVELED x = 4t4 - 6t3 + 2t -1 HAVE ... WHEN L = 45 N = 9x = 16t3 - 18t2 + 2 a = 9x = 48t2 - 36t THEN X=3t3-6t2-12t+5 AND HAVE -ル=器=9だ-124-12 THEN AT t= 25: X= 4(2) -6(2) +2(2)-1 X= 19 m OR O=能=184-15 MA N=58 m N=16(2)3-18(2)2+2 OR 0= 150 35 a= 4B(2)2-36(2) (a) WHEN 15=0: 9t2-12t-12 = 3(3t2-4t-4)=0 OR OR (31+2)(1-2)=0 11.2 OR t=25 AND t=- 35 (REJECT) :: t=25 GIVEN: X=3t4+4t3-7t2-5t+8 X~ MM. t~S (b) AT L=45: X=3(4)3- L(4)2-12(4)+5 OR X=53m FIND: X, N, AND Q AT 1:35 a=18(4)-12 or a=60 52 X=314+413-722-51+B HAVE -FIRST OBSERVE THAT .. OSECS: NOCO N= dx = 12t3+12t2-14t-5 f>52: 12>0 THEN a=#= 36t2+24t-14 AND Now. AT t=0: X=5 m t=25: X=3(2)3-6(2)2-12(2)+5=-19m AT t=35: X=3(3)+4(5)-7(3)2-5(3)+B OR X=2B1 mm OR N= 385 5 N=12(3)3+12(3)2-14(3)-5 a=36(3) +24(3)-14 OR 0=382 77 11.3 THEN 1x2-x01=1-19-51=24 m GIVEN: X=6t2-8+40 cos 17t X~IN., t-5 X4 - X2 = 53 - (-19) = 72 m FIND: X, N, AND a AT t= 65 .. TOTAL DISTANCE TRAVELEN = (24+72) m = 96 m X= 622-8+40 cosmt HAVE ... 11.7 N= # = 12+ - 4017 SINTE. GIVEN: X=t3-9t2+24t-8 X~IN. t-5 THEN a = # = 12 - 4012 cos 11t FIND: (a) t. WHEN N=0 AND (b) X AND TOTAL DISTANCE TRAVELED WHEN Q=0 AT t=65: X=6(6) -8+40 cos 67 OR X = Z48 IN. OR N= 72 3 17-12 (6) - 407 SINGT OR 0= -383 1 1 X=13-922+24t-B Q= 12 - 40 112 COS 617 HAVE ... N= 2 = 3t2-18+24 THEN 11.4 GIVEN: X=3t3-2t2-30+8 X-ft, t-5 a=#= 6t-18 AND FIND: t, X, AND Q WHEN 15=0 (a) WHEN N=0: 3t2-18t+24=3(t2-6t+8)=0 X= 3t3- 2t2-30+B OR (t-2)(t-4)=0 HAVE _ N=#=5t2-St-30 THEN OR 1=25 AND 1=45 a = dy = 10t-5 (b) WHEN Q=0: 6t-18=0 OR t=35 dus AT 1=35: X=(3)=9(3)=24(3)-8 OR X=10 IN. WHEN N=0: St2-St-30=S(t2-t-6)=0 FIRST OBJERVE THAT .. DELCES! NOO OR t=35 AND t=-25 (REJECT) : 1:35 256 £ = 35: 560 AT t=35: X= 3(3)= 2(3)-30(3)+8 OR X=-59.5 Ft NOW .. AT t=0: X=-BIN. AT t=25: X,=(2)3-9(2)2+24(2)-B=12 IN. 0:10(3)-5 OR 0 = 25 12 11.5 GIVEN: X= 6t - 2t3-12t2+3t+3 x~m. t~s 101N. 12 IN. X FIND: t, x, AND NO WHEN Q = 0 (0) (35) (25) HAVE .. X = 6t4 - 2t3 - 12t2+ 3t+3 X2- X0 = 12- (-B) = 20 IN. THEN W= SX = 24t3-6t2-24t+3 THEN 1x3-x21=110-121=2 IN. a = dt = 72t2-12t-24 ". TOTAL DISTANCE TRAVELED = (20+2)IN. = 22 IN. AND WHEN Q=0: 722-12t-24=12(62-t-2)=0 OR (31-2)(2++1)=0 OR t= 35 AND t=-25 (REJECT) : t=0.6675 AT t= 35: X=6(3)4-2(3)3-12(3)2+3(3)+3 OR X=0.259m N= 24(3) -6(3)-24(3)+3 OR N=-8.55

11.8 11.10 GIVEN: X=t3-6t2-36t-40 X-11, t-5 GIVEN: Q & t; AT t=0, N=16" b; AT t=15, N=15" N, x=20 IN. FIND: (a) t WHEN N=0 (b) N, Q, AND TOTAL DISTANCE FIND: N. X. AND TOTAL DISTANCE TRAVELED TRAVELED WHEN X=0 AT tals HAVE __ X = +3- 6+2-36+-40 HAVE .. a = kt k - CONSTANT N=dx - 3t2-12t-36 THEN Now # = a= kt a= dr = 6t-12 AT too, N=16 ": Indus lo ktat dNA OR N-16 = 2kt OR 15 = 16+2/2/2 (1) (a) WHEN N=0: 3t2-12t-36=3(t2-4t-12)=0 AT t=15, 15:15": 15" = 16" + 2 k(15)2 OR (++2)(+-6)=0 ALSO \$\frac{4x}{4x} = 16 - t^2 OR \$ = -2 \frac{10}{50} \text{ and } 15 \cdot 16 - t^2 OR t=-25 (REJECT) AND t=65 : t=65 (b) WHEN X=0: t3-6t2-36t-40=0 AT t=15, x=20 m : 120 dx = 1 (16-t2)dt FACTORING .. (t-10)(t+2)(t+2)=0 OR t=105 OR X-20 = [16t-3t3], NOW OBSERVE THAT .. OST +65: NO OR X = -3+3+16++13 (IN.) 654 t 5105: NOO AND AT t=0: X=-40 ft THEN .. AT t=75: N=16-1712 OR N=-33 5 X7 = -3(7)3+16(7)+13 OR X7=2.00 IN. t=65: X=(6)36(6)2-36(6)-40=-256 ft WHEN N=0: 16-t2=0 OR t=45 AT t=0: X0=3 1=105: N=3(10)2-12(10)-36 OR N=144 \$ 0=6(10)-12 OR 0=48 3 t=45: X4 =- 3(4) +16(4) + 13 = 47 IN. NOW OBSERVE THAT .. US + LAS: 15 >0 -40ft 0 X -256ft 4562575: 1540 (0) (105) THEN | X6-X0 = 1-256-(-40) = 216 ft X10-X6 = D-(-256) = 256 St (75) 301. .. TOTAL DISTANCE TRAVELED = (216+256) At = 472 ft X4-X3 = 47- 13 = 42.67 IN. 11.9 GIVEN: a = 6 5tz; AT t=0, X=-32 st; AT t=25, N=-6 ts 1x7- x41 = 1 2-471 = 45 IN . : TOTAL DISTANCE TRAVELED = (4247+45)IN. = 87.7 IN. FIND: N, X, AND TOTAL DISTANCE TRAVELED 11.11 AT 1=55 GIVEN: Q = A- 62; AT t=0, X=8 m, N=0; AT 1=15, N=30 5 HAVE .. JE = a = 6 52 FIND: (a) t WHEN NO AT t=25, N=-6 \$: [dN= 12 6dt (b) TOTAL DISTANCE TRAVELED WHEN OR N-(-6) = 6(t-2) t=55 OR N= 6t-18 (3) ALSO .. dx = N = 61-18 HAVE _ a= A- 6t2 A ~ CONSTANT AT t=0, X=-32 ft: 132 dx = 6(6t-18)dt SE = a = 4-642 AT t=0, 5=0: 1, ds = 1 (A-6+2)dt OR X- (-32) = 3t2-18t OR X=322-18t-32 (ft) OR N = At-2t3 (2) AT t=15, N=30 = 30 = A(1)-2(1)3 AT t=55: 15=6(5)-18 OR 15=12 \$ X=3(5)2-18(5)-32 OR X=-47 ft OR A = 32 5 AND N= 32t-2t3 ALSO dx = N = 321-213 WHEN N=0: 6t-18=0 OR t=35 AT t=0, x=8m: 18dx = 10(32t-2t3)dt AT t=35: X3=3(3)2-18(3)-32 = -59 ft NOW OBSERVE THAT DS E < 35: N < 0 OR X= 8+162-2t4 (m) (a) WHEN 15=0: 32t-223 = 2t(16-t2)=0 354 £ 555: N5>0 OR t=0 NN t=45 (b) AT t=45: X4=8+16(4)2- 2(4)4 = 136 m -59 ft -47 ft -32 ft t=53: Xs=B+16(5)2-2(5)4=95.5 m (35) (55) (0) NOW OBSERVE THAT OCELAS: NO >0 THEN | X3 - X0 = 1-59 - (-32) = 27 ft 4562555: NGO 1 x5- x31 = 1-47- (-59)1 = 12 ft .. TOTAL DISTANCE TRAVELED = (27+12)ft = 39 ft X4-X0= 136-8 = 128 m 1x5-x41=195,5-1361 + 40.5 m .. TOTAL DISTANCE TRAVELED = (128+40.5)m = 168.5 m

11.12 GIVEN: axt ; AT t=0, X=24 m; AT t=65, X=96m, N=18 m/s FIND: X(t) AND W(t) HAVE .. a= kt2 k ~ CONSTANT gr = a = ktz Now. AT t= 65, N=18 =: 18 do = 1 kt dt

OR N-18 = 3k(t3-216) OR N=18+3k(t3 216) (5) #= N= 18+3k(t3-216) ALSO AT t=0, X=24 m: | x dx = | t [18+3k(t3-216)]dt OR X-24 = 18t+3k(4t4-216t) Now. AT t= 65, X=96 m: 96-24 = 18(6)+3 R[4(6) -216(6)] OR R = 9 54 THEN. X-24=18+3(+)(+++-216+) OR X(t) = roet +10t + 24 AND N=18+3(4)(23-216) OR 15(t) = = + + 10 11.13 GIVEN: FOR 255t = 105, Qa to; AT t=25, 5 = -15 m/s; AT t=10 5, 5=0.36 m/s; 1X21=21 X101 FIND: (a) X AT 1=25 AND AT 1=105 (b) TOTAL DISTANCE TRAVELED FROM t=25 TO t= 105 HAVE .. a : £3 Now # = a : £3 K ~ CONSTANT AT t= 25, 15=-15 ": [15 d15 = 12 to dt OR NS-(-15) = $-\frac{1}{2}\left(\frac{1}{4}z^{2}-\frac{1}{(2)^{2}}\right)$ OR NS = $\frac{1}{2}\left(\frac{1}{4}-\frac{1}{10}z^{2}\right)$ - 15 OR NS = $\frac{1}{2}\left(\frac{1}{4}-\frac{1}{10}z^{2}\right)$ - 15 OR NS = $\frac{1}{2}\left(\frac{1}{4}-\frac{1}{10}z^{2}\right)$ - 15 OR NS = 128 M.S (a) HAVE $\frac{1}{4}z^{2}=1-\frac{1}{4}z^{2}=\frac{1}{4$ THEN $|dx = \int (1 - \frac{1}{2}) dt + C$ (-constant)or $x = t + \frac{1}{2} + C$ (m)Now $x_2 = 2x_{10}$: $z + \frac{1}{2} + C = 2(10 + \frac{10}{10} + C)$ AND X= ++ ++ 1.2 (m) 1. AT 1=25: X2=2+ 2+1.2 OR X2=35.2 m t=105: X10=10+10+1.2 OR X10=17.6 m NOTE: A SECOND SOLUTION EXISTS FOR THE CASE X2 >0, X10 CO. FOR THIS CASE C=- 22 is m AND X2=1115m, X10=-515m (b) WHEN N=0: 1- \$2=0 OR to BS AT 1=85: XB = 8+ 8+1.2 = 17.2 m NOW OBSERVE THAT 255 £ < 85! N < 0 BSETEIDS: 1500 35.2m X 17.2m 17.6m (BS) (105) THEN 1X8-X21=117.2-35.21=18 m X10-XB = 17.6-17.2 = 0.4m

.. TOTAL DISTANCE TRAVELED: (18+0.4)m: 18.4 m.
NOTE: THE TOTAL DISTANCE TRAVELED IS THE SAME

FOR BOTH CASES.

11.14 GIVEN: Q =- 8 TYS2; AT t= 45, X=20 m; WHEN 15=16 M/s, X= 4 m FIND: (a) t WHEN NO (b) NT AND TOTAL DISTANCE TRAVELED AT t= 11 5 E= Q= - B 52 HAVE THEN IdW = I-Bdt + C C- CONSTANT OR N=-8++ (") #= N=-Bt+C ALSO AT t= 45, X=20m: 120 dx + 141-8t+c)dt OR X-20 = [-4t2+Ct]2 OR X=-4t2+C(t-4)+84 WHEN 5=16 5, X=4m: 16=-8++ C = C=16+8+ 4=-42+0(1-4)+84 COMBINING .. 0 = - 42 + (16+8+)(+-4) + 80 SIMPLIFYING .. t2-4++=0 OR 1=25 AND C = 32 5 N=-81+32 (3) x=-42+32t-44 (m) (a) WHEN N=0: -8t+32=0 OR t=45 (b) AT t=0: X= - 44 m t=45: X4 = 20 m t=115: x11=-4(11)2+32(11)-44=-176 m 0= £445: 150 NOW OBSERVE THAT 456 £ \$115: 05 60 O SOM X (115) THEN X4-X0 = 20-(-44) = 64 m 1x11- x41= 1-176-201= 196m ". TOTAL DISTANCE TRAVELED = (64+196) M = 260 M 11.15 GIVEN: Q= K(100-X), R - CONSTANT; N=0 AT X = 40 mm, X = 160 mm; WHEN X=100 mm, N= 18 mm/s FIND: (a) R (P) N WHEN X= 120 MM (a) HAVE N = a = k(100-x) WHEN X=40 mm, N=0: 10 Ndor = 10 R(100-X) dx OR 25 = K(100x-2x2-3200) WHEN X=100 MM, N=18 mm; \$(18)2 = k[100(100) - \$(100) - 3200] OR K=0.0932 (b) WHEN X=120 mm: \$102 = 0.09[100[120]-\$(120) -3200] = 144 OR 15= = 16,97 mm

11.16 11.18 CONTINUED GWEN: Q = K/(X+4)2, R ~ CONSTANT; WHEN Now .. No = 2: \frac{1}{2}Not = (2) = \frac{k(51.5-ALNB)}{k(\frac{2}{2}-ALNB)} X=0, N=0; WHEN X=BM. 5=4 m/s FIND: (a) k OR 6-4ALNZ=31.5-ALNB (b) X WHEN N= 4.5 M/S OR 25.5 = A(LNB-4LNZ) = A(LNB-LNZ) = ALN(2) (C) SMAX OR A = - 36.8 ft2 WHEN X=16 \$2, 15=29 \$: \$(29)2=k[\$(16)2-25.5 LN(16)-\$] (a) HAVE NOX = a = (x+4)2 /x /k /x /4)2 dx (SIM-=(3)N) dua SN1+=(11)N1 TAHT DUITON HAVE .. 841 = K[256 - 2x255 + 4 LN(2) - 1] OR \$ N2 = - k (x+4 - 4) OR K=1.832 52 WHEN X=8m, N=4 5: \$(4)2=-k(8+4-4) 11.19 (b) WHEN N= 4.5 \(\frac{m}{2} : \frac{1}{2} (4.5)^2 = -48 \(\frac{1}{244} - \frac{1}{4} \) GIVEN: Q= k(1-k"), k = constant; WHEN X = - 2 M, IT & 6 5; WHEN X = 21.6 m X=0. 5=0 (C) NOTE THAT WHEN IS I DMAX, Q=0. NOW .. FIND: (a) k Q -0 AS X -00 SO THAT

\[\frac{1}{2} N \tag{7.00} = 48 \tag{1.00} \left(\frac{1}{2} - \frac{1}{244}\right) = 48 (\frac{1}{4}\right)
\] (b) IT WHEN X=-1 M (a) HAVE No 3x = a = k(1-e-x) NMAX = 4.90 5 WHEN X = - 2 m, N= 6 3: | " wdw = | R (1-ex) dx OR 2 (12-36) = 12 (x+ex) = 2)+18 11.17 GIVEN: Q = 6x-14, Q = \$4/52, X = \$2; WHEN X=0, N=492/s FIND: (a) XMAX WHEN X=0, 15=0: 0= k(1+2-e)+18 (b) IT WHEN TOTAL DISTANCE OR R = 4.1011 52 (b) WHEN X=-1 m: 2N2 + 4.1011(-1+e'+2-e')+18 TRAVELED = 1 St OR N= 2,43 % N = a = 6x-14 HAVE WHEN X=0, 15=4\$: 14 15 du = 1 (6x-14) dx 11.20 GIVEN: Q =- (0.1+ SIN 6), Q- 15, X-m; OR [252] = [3x2-14x] b= 0.8 m; WHEN X=0, N=1 m/s OR 2502 = 3x2-14x+8 FIND: (a) NJ WHEN X=-1 M (a) FIRST DETERMINE WHERE N=0 .. (b) X WHERE IT = Nomax 3x2-14x+8=(3x-2)(x-4)=0 (C) June X= 3ft AND X=4ft NOW OBSERVE THAT AS THE PARTICLE PASSES 2 3x = 0 = - (01+ 214 28) HAVE WHEN X=0, N= 1 3: [" wdw = [- (0.1+ sm & b) dx THROUGH X=0, 500 AND Q CO AND THAT AT X= \$ft, 5=0 AND Q<0. THUS, THE PARTICLE WILL OR \$ (152-1) = - [0.1x - 0.8 cos 0.8] NEVER REACH X= 4 ft AND, THEREFORE, XMAX = 0.667 ft OR 2N2 = -0.1x + 0.8 cos 5.8 - 0.3 (b) THE PARTICLE WILL HAVE TRAVELED A TOTAL (a) WHEN X=-1 m: 2152=-0.1(-1)+0.8cos 0.8 -0.3 DISTANCE OF I ST WHEN IT PASSES THROUGH OR 15= = 0.323 5 X = 3 ft FOR THE SECOND TIME AND IS MOVING (P) WHEN D= DMAX, Q=0: -(0.1+5100B)=0 TO THE LEFT. THEN -- AT $X = \frac{1}{3}H$: $\frac{1}{2}U^2 = 3(\frac{1}{3})^2 - 14(\frac{1}{3}) + \theta = \frac{11}{3}$ OR X=-0.080 134 m (C) WHEN X = -0.080 134 m: 2 5 mm = -0.1 (-0.080 134) +0.8 cos -0.080 134 - 0.3 N=271 \$ -OR Nome 1.004 5 11.18 GIVEN: Q= k(X-X), k AND A ARE CONSTANTS; AT t=0, X=1ft, N=0; WHEN X=16 ft, 15 = 29 \$45; 12 (X=Bff) = 5(2(X= 5 ff)) FIND: A AND K HAVE JAX = a = k(x-X) WHEN X=1ft, 5=0: [55065 = [x (x-X)]X OR 202 = k[2x2-ALNX]X

= k(2x2- A LNX - 1)

(CONTINUED)

AT X= 2 ft: 2 52 = k(2(2)2-ALN2-2]=k(3-ALN2) X=BA & NB = R[2(B)2-ALNB-2]= K(31.5-ALNB) R=4.10 52

X=-0.0801m €

11.21 11.23 CONTINUED GIVEN: Q = 0.8 [5+49 , Q- "/5", 5- "/5 (b) HAVE du = 0 = -12 12 25 AT t=0, 15 = 16 5: 16 5-25 du = 5 - 12 dt (b) HAVE WHEN X=0, 5=0 FIND: (a) X WHEN IS= 24 M/s (b) IT WHEN X = 40 m OR - 3[15 3/2] " = - 12 t HAVE NOTE = 0 = 0.8 (52+49) WHEN X=0, 15=0: 5 (52+49) = 50.8dx OR \$ (15 12 - 14) = 12 WHEN 15=9 13: 3 (9th -64)= 12 OR 1=0.17135 OR [15+49] = 0.8x 11.24 GIVEN: Q =- 5/(205-15) a-1/3, 15-1/3; (a) WHEN 1=243: 124+49 -7 = 0.8x AT t=0, X=0, N=No; AT t=25, OR X = 22.5 m 5=0.550 (b) WHEN X=40 m: 15+49-7=0.8(40) FIND: (a) No N=38.4 m (P) + WHEN D=0 (C) X WHEN WEI ST 11.22 GIVEN: Q = - KIN, K ~ CONSTANT; AT 1=0, (a) HAVE dr. a = - 215-15 X=0, 15=81 M/5; WHEN X=18 m, AT t=0, 15=15: 10 (215-15) ds = 10-5 dt 15= 36 m/s OR (215-11)2 No - 5t FIND: (a) IJ WHEN X=20 M (b) t WHEN N=0 AT 1=25, 5=0505: (255-0.555)-55=10(2) (a) HAVE N 3x = a =- k Tu DR \$ 50 = 20 WHEN X=0, 15=81 =: | "Todu = | - kdx WHEN 5=0: (8-5)2-16=10t WHEN 5=0: (8)2-16=10t (b) HAVE OR t=4.85 OR 3 (NT) 1 - - KX (C) HAVE NOW = a = - 5 OR - - 8-1 OR \$ (1312-729) = -kx

WHEN X=18 m, 5=36 \$ (362-729) = -k(18)

OR R=19 (356-729) = -19(20) WHEN X=0, 0-15=4#: | 1508-17/00 = 15-5dx OR $[4u^2 - \frac{1}{3}u^3]_4^N = -5x$ OR $(4u^2 - \frac{1}{3}u^3) - [4(4)^2 - \frac{1}{3}(4)^3] = -5x$ OR $(4u^2 - \frac{1}{3}u^3) - \frac{128}{5} = -5x$ OR 15 159

(b) Have at = 0 = -19 (5)

At t=0, 15 = 81 %: | 15 dy = 5 -19 dt J= 29.3 5 WHEN N=1 \$ [4(1) = 3(1)] - 128 =- 5X OR X=7.80 St 11.25 or 2[[] = -19t GIVEN: Q= 0.4(1-RW) R-CONSTANT; 2(15-9)=-19t OR AT 1=0, X=4 m, N=0; AT 1=155, WHEN 15=0: 2(-9)=-19t 15 = 4 m/s ce t=0.9475 € FIND: Lai k (b) X WHEN N= 6 M/S 11.23 GIVEN: Q = - RU . R - CONSTANT; AT t=0, (C) Jmax X=0, 15=16 14/5; WHEN X=614., (a) HAVE # a = 0.4 (1- RN) N= 4 14/5 AT t=0, 15=0: 10 du = 10.4dt FIND: (a) IS WHEN X=5 IN. (b) t WHEN 15=9 IN/S OR - = [[LN (1- KN)] = 0.4 t (a) HAVE IT OX = a = - KIT S.S OR LN(1- ku) = -0.4 kt WHEN X=0, 15=16 5: \(10 - 1.5 dis = \int \) - kdx AT t= 155, J=43: LN(1-4k) = -0.4k(15) SOLVING YIELDS R = 0.145703 m OR - 2 [N =] = - XX OR R=0.1457 # -(b) HAVE 13 dx = a = 0.4(1-ks) OR 2(7-4)= kx WHEN X=4 m, $\sqrt{5} = 0$: $\int_{1-RD}^{D} \frac{D^{2}dD}{1-RD} = \int_{4}^{X} 0.4 dx$ NOW - $\frac{D^{2}}{1-RD} = -\frac{1}{R} + \frac{\sqrt{R}}{1-RD}$ WHEN X=6 IN, 15=4 INS: 2(14-4)= K(6) CR R = 12 12 FINALLY -- WHEN X=5 IN .: 2(- 4)= 15(5) THEN SU[- + + + (1-EN) du = 10.4 dx 四 帝= 24 5=476学 OR [- \ - \ \ 2 LN (1- \ 20) \ 0 = 0.4 (X) \ 4

(CONTINUES)

(CONTINUED)

11.25 CONTINUED

OR - [+ to LN(1-KN)] = 0.4(x-4) WHEN W= 6 =:

 $-\left(\frac{6}{0.145703} + \frac{1}{(0.145703)^2} \ln(1-0.145703 \times 6)\right) = 0.4(4-4)$

OR 0.4(x-4)= 56.4778

OR X= 145.2 m

(C) THE MAXIMUM VELOCITY OCCURS WHEN Q = O.

:. a=0: 0.4(1-kuma)=0 OR 15 MAX = 0.145703

OR Nomax = 6.86 5

AN ALTERNATIVE SOLUTION IS TO BEGIN WITH EQ.(1). W(1-ks) = -0.4kt

N= = (1- 0.4kt)

THUS, DANK IS ATTAINED AS \$ - 00 ... NMAX = K .. AS ABOVE

11.26 GIVEN: a=-0.65 to a- 7/5: 15- 1/5; AT t=0, x=0, 15=9 m/s

FIND: (a) X WHEN No 4 m/s (b) + WHEN N= 1 m/s

(C) t WHEN X=6 m

(a) HAVE 12 dx = a = -0.6 x 12 dx = 1 -0.6 dx OR 2[15 2] = -0.6 X

WHEN $J=4\frac{W}{5}$: $X=\frac{1}{0.3}(3-J^{1k})$ (1)

(b) HAVE dy = a = -0.65% OR x = 3.33m WHEN t=0, 5 = 9%: $\int_{0}^{5} d^{3}x dx = \int_{0}^{1} -0.6dt$

OR - 2 [15 1/2] = -0.6t

WHEN 15=1 3: 1 - 3 = 0.3t

OR t= 2.72 5

(C) HAVE $\frac{1}{10} - \frac{1}{3} = 0.3t$ OR $S = \left(\frac{3}{1+0.9t}\right)^2 = \frac{9}{(1+0.9t)^2}$

Now .. dx = N = (1+09+)2

AT t=0, x=0: | dx = | t 9 (1+09t) 2 dt

OR X = 9[- 09 1109E]

= 10 (1 - 1+0.9t)

WHEN X=6m: 6= 9t

OR t=1.6675

AN ALTERNATIVE SOLUTION IS TO BEGIN WITH EQ. (1)

X = 0.3 (3-1512)

THEN $\frac{dx}{dt} = 15 = (3 - 0.3 \times)^2$ Now. At t=0, x=0: $\int_0^x \frac{dx}{(3 - 0.3 \times)^2} = \int_0^t dt$

or $t = \frac{1}{0.3} \left(\frac{1}{3 - 0.3x} \right)_0^x = \frac{x}{9 - 0.9x}$

WHICH LEADS TO THE SAME EQUATION AS ABOVE.

11.27



GIVEN: N=7.5(1-0.04x) 0.3 5- mi/h x- mi : AT t=0, x=0 FIND: (a) X AT tolh

(b) a (4/52) AT too (C) t WHEN X=6 mi

(a) HAVE dx = 1.5(1-0.04x) 03
AT t=0, x=0: \(\frac{x}{(1-0.04x)^{0.3}} = \frac{t}{1} \cdot 2.5 dt

or $\frac{1}{\sqrt{3}}(-\frac{1}{\sqrt{3}})[(1-\alpha)4x)^{0,7}]_{0}^{x}$. 7.5t or $1-(1-\alpha)4x)^{0,7}=0.21t$ or $1-(1-\alpha)4x)^{0,7}=0.21t$

AT L=1 h: X=0.04 [1-[1-0.21(1)] YO.7

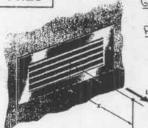
OR X=7.15 mi (b) HAVE Q= N ST (b) HAVE $\alpha = x \cdot \frac{1}{2} \frac{1}$

OR a = - 275 MO 1

(C) FROM EQ (1) .. t = 0.21 [1-(1-0.04x)0.7] WHEN X=6 mi: t= 121 [1-[1-0.04(6)] 0.7] = 0.832 29 h

OR t = 49.9 MIN

11.28



GIVEN: N= Q.IBNO J- 5, X-M No= 3.6 m/s FIND: (a) a WHEN X=2 m (b) TIME FOR AIR TO FLOW FROM XEIM

TO X=3 m

= 0.18 dx (0.18 dx) (a) HAVE

WHEN X=2 m: $Q = -\frac{\chi^3}{(2)^5}$

(P) HAVE # " " 0.18 "

OR Q=-0.0525 # FROM X=1 m TO X=3m: 1, Xdx = 1,0.1856dt

OR [12x2] = 0.18 No (ta-ti) OR (t3-t1) = \frac{\frac{1}{2}(9-1)}{0.18(3.6)}

OR ta-t, = 6.175



GIVEN: Q = -52.2/[1+(4/20,9 <10")]2

Q = 4/52, 4 = 18

FIND: 4 MAX WHEN

(Q) 15 = 1800 4/5

(b) 15 = 36, 700 4/5

HAVE $N \frac{dN}{dq} = \alpha = -\frac{32.2}{(1 + \frac{4}{20.9 \times 10^6})^2}$ WHEN q = 0, $N = N_0$ THEN... $\int_{0}^{0} J dN = \int_{0}^{0} \frac{1 + \frac{4}{4}}{(1 + \frac{4}{20.9 \times 10^6})^2} dq$ OR $-\frac{1}{2}N_0^2 = -32.2 \left[\frac{20.9 \times 10^6}{1 + \frac{4}{20.9 \times 10^6}} \right]_{0}^{0}$ OR $J_0 = 1345.96 \times 10^6 \left(1 - \frac{1}{1 + \frac{4000}{20.9 \times 10^6}} \right)$ OR $J_{0} = 1345.96 \times 10^6 \left(1 - \frac{1}{1 + \frac{4000}{20.9 \times 10^6}} \right)$ OR $J_{0} = 1345.96 \times 10^6 \left(1 - \frac{1}{1 + \frac{4000}{20.9 \times 10^6}} \right)$ $J_{0} = 1800 \frac{51}{5}$ $J_{0} = 1800 \frac{51}{5}$ $J_{0} = 1800 \frac{51}{5}$

(c) $N_0 = 36,700 \frac{\text{ft}}{5}$: $V_{\text{MAX}} = \frac{(36,700)^2}{64.4 - \frac{(36,700)^2}{20,9x10^6}}$

THE VELOCITY 36,700 \$ 15 APPROXIMATELY THE ESCAPE VELOCITY No FROM THE EARTH. FOR NO



HAVE

SIVEN: Q = - 2R2, R=3960 mi; WHEN T=0, 15=0

5 do = a = - 9 R2

WHEN T=R, 15=150 T=0, 15=0 THEN Sudds = SR - 9R2 dr

(CONTINUED)

11.30 CONTINUED

OR $-\frac{1}{2}IS_{2}^{2} = 9R^{2}[\frac{1}{r}]_{R}^{\infty}$ OR $IS_{2} = \sqrt{2qR}$ = $(2\times32.2\frac{f_{2}}{5}\times3960mi \times \frac{5280ft}{1mi})^{\frac{1}{2}}$

OR NE = 36,7MD \$

11.31 GIVEN: N= 5. [1- SIN(+)]; AT t=0, X=0,

FIND: (a) X AND a AT t=3T

(b) NAVE DURING t=0 TO t=T

(a) Have $\frac{dx}{dt} = 15 = 15 \left[1 - \sin(\frac{\pi t}{T})\right]$ At t = 0, x = 0: $\int_{0}^{x} dx = \int_{0}^{x} u_{0} \left[1 - \sin(\frac{\pi t}{T})\right] dt$ or $x = 15 \left[t + \frac{\pi}{\pi} \cos(\frac{\pi t}{T}) - \frac{\pi}{\pi}\right]$ (1)

At t = 3T: $X_{3T} = N_0 \left[\frac{3T + \frac{T}{\pi} \cos(\frac{\pi \cdot sT}{T}) - \frac{T}{\pi} \right]$ = $N_0 \left(\frac{3T}{T} - \frac{2T}{\pi} \right)$

ALSO.. a= dr = dr { vo[1-51N(7)]} xst = 2.36vo.T

AT t=3T: $a_{37} = -u_3 \frac{\pi}{T} \cos \frac{\pi t}{T}$ or $a_{37} = \frac{\pi u_3}{T}$

(b) Using Eq. (1)..

At t=0: $X_0 = \kappa_0 [0 + \frac{\pi}{\pi} \cos(0) - \frac{\pi}{\pi}] = 0$ At t=T: $X_T = \kappa_0 [T + \frac{\pi}{\pi} \cos(\frac{\pi T}{T}) - \frac{\pi}{\pi}]$ $= \kappa_0 (T - \frac{\pi T}{\pi})$

Now.. $N_{AVE} = \frac{0.363 N_0 T}{N_0 T}$ $\frac{0.363 N_0 T}{T-0}$

OR ISME = 0,363No

11.32 GIVEN: $J=\sigma'\sin(\omega_n t+\phi)$; AT t=0, $X=X_0$, $J=\sigma_0$; $X_m x_m = 2X_0$ Show: (a) $J'=(N_0^2+X_0^2\omega_n^2)/2X_0\omega_n$

 $X = X_0 [3 - (J_0^2 + X_0^2 \omega_h^2)/2X_0 \omega_h]$ $(b) J_{MAX} OCCURS WHEN$ $<math>X = X_0 [3 - (J_0 | X_0 \omega_h)^2]/2$

(a) At t=0, 15=15: 15 = 15' sin (0+4) = 15' sin 4

THEN COS\$ = \(\sigma \sigma^{12} - 15' \sigma^{1} \sigma^{

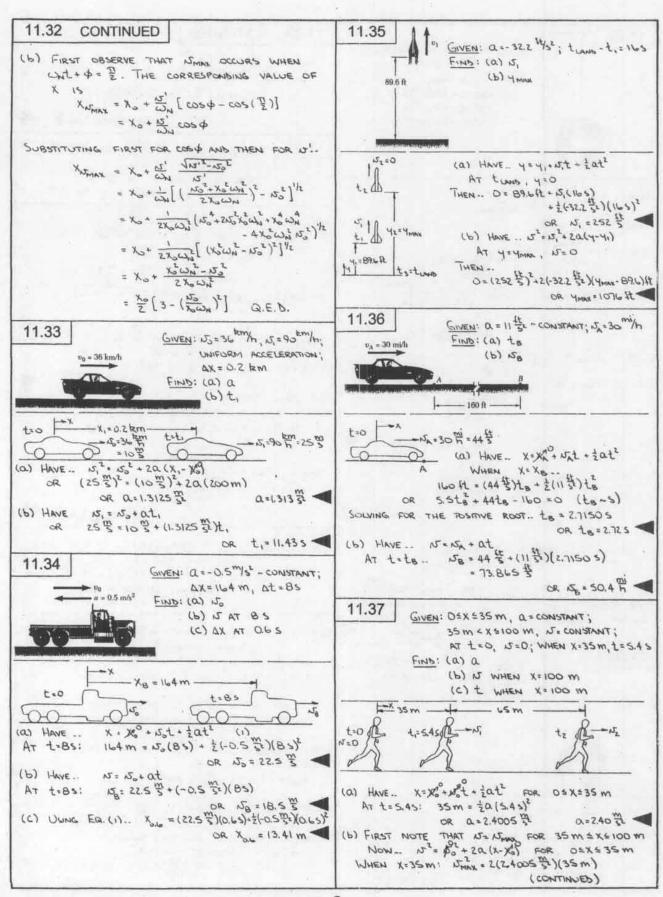
Now $\frac{dx}{dt} = \omega = \omega' \sin(\omega_N t + \phi)$ At t = 0, $x = x_0 : \int_{x_0}^{x} dx = \int_{x_0}^{t} \omega' \sin(\omega_N t + \phi) dt$

> OR $X-X_0 = nT' \left[-\frac{1}{2n} \cos(\omega_n t + \phi) \right]_0^T$ OR $X=X_0 + \frac{nT'}{2n} \left[\cos \phi - \cos(\omega_n t + \phi) \right]$

NOW OBSERVE THAT XMAX OCCURS WHEN $\cos(\omega_n t + \phi) = -1$. THEN... $\chi_{\text{MMX}} = 2\chi_0 = \chi_0 + \frac{\omega^2}{2} \left[\cos \phi - (-1) \right]$

Substituting for $\cos \phi - x_0 = \frac{\omega^1}{\omega_N} \left(\frac{\sqrt{\omega^2 - \omega_0^2}}{\omega^1} + 1 \right)$ or $x_0 \omega_N - \omega^1 = \sqrt{\omega^{12} - \omega_0^2}$

SQUARUNG BOTH SIDES OF THIS EQUATION. $\chi_{0}^{2} \omega_{N}^{1} - 2\lambda_{0}\omega_{N} + \omega_{1}^{12} = \omega_{1}^{12} - \omega_{0}^{2}$ $OR \qquad \omega_{1}^{1} = \frac{\kappa_{0}^{2} + \chi_{0}^{2} \omega_{N}^{2}}{2\lambda_{0}\omega_{N}} \qquad Q. E. D.$ (CONTINUED)

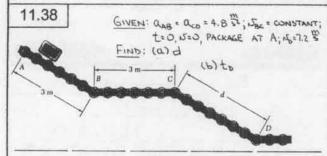


11.37 CONTINUED

OR UMAX = 12.9628 \$ UMX = 12.96 \$ (C) HAVE.. X = X, + V5 (t-t,) FOR 35 m < X = 100 m

WHEN X=100 m: 100 m = 35 m + (12.9628 \$)(t_2.4)5

OR t_2 = 10.41 5



(a) FOR $A \rightarrow B$ AND $(\rightarrow D)$ HAVE $D^2 = D_0^2 + 2\alpha(X - X_0)$ THEN... AT B.. $D_{RC}^2 = D_0^2 + 2(4.8 \frac{m^2}{5})(3-0)m$ $= 28.8 \frac{m^2}{5}$ ($D_{RC} = 5.3 + 6.6 \frac{m^2}{5}$) AND AT D.. $D_0^2 = D_0^2 + 2\alpha_{CD}(X_0 - X_0)$ $D_0^2 = 2X_0 - X_0$ OR $(7.2 \frac{m^2}{5})^2 = (28.8 \frac{m^2}{5^2}) + 2(4.8 \frac{m^2}{5}) d$ OR d = 2.40 m

(b) FOR A = B AND (=0) HAVE... N=NS+at

THEN.. A = B... 5.3666 = 0+ (4.8 = 1)tag

OR LAG = 1.118 045

AND (=0)... 7.2 = 5.3666 = + (4.8 = 1)tag

OR toc = 0.559015 FINALLY, to=tas+toc+tco=(1.11804+0.55901+0.38196)5 OR to=2.065◀

11.39

GIVEN: AT t=0, XM = Xp =0; AT t= 425, XM=Xp;

NM = CONSTANT; FOR 05 t = 185, Np=0;

FOR 185 < t > 265, Ap = CONSTANT;

AT t = 265, Np = 90 m;

FOR 265 < t < 425, Np = 90 m

FIND: (a) Xp AT t = 425

(b) NM

- (XP)42 5 P (Np)42 P (Up) 210 500 1:185 1=265 1=425 (126)50=00 H=52 m (126)45=00 H=52 m (Np)10=0 (a) PATROL CAR: FOR 185<15265: 15p=(15p/18+ap(t-18) AT t= 265: 25 \$ = ap(26-18)5 OR ap = 3.125 52 ALSO, Xp = (XP/18+(15/18(t-18)+2ap(t-18)2 AT t=265: (XP) = = 2 (3.125 m)(26-18)2 = 100 m FOR 2654 £ 5 425: Xp = (Xp)26 + (Np)26 (t-26) AT t=425: (Xp)42 = 100m + (25 3)(42-26)5 (Xp)42=0.5km (b) FOR THE MOTORIST'S CAR. Xm = (XM) + 15mt AT t=425, Xm= Xp: SOOM = Nm (425)

OR Nom = 11.9048 3

OR 15m = 42.9 5

11.40

($v_A |_{0} = 12.9 \text{ m/s}$ ($v_B |_{0} = 1$ ($v_B |_{0}$

FIND: (a) an and as knowing
THAT BOTH ARE UNIFORM
(b) ts when runner 8
STARTS TO RUN

(a) FOR RUNNER A: $x_n = (120)^2 + (101)_0 + \frac{1}{2}a_n t^2$ At t = 1.825: $20 m = (12.9 \%)(1.825)^2 + \frac{1}{2}a_n (1.825)^2$ OR $a_n = -2.10 \%$

ALSO.. IS = (ISA) + CAT AT t=1.825: (ISA), BZ = (12.9 =)+(-2.10 =)(1.825) = 9,078 =

FOR RUNNER B: NB2 = (NB2) + 200 (x0-(N6)) WHEN X8=20 M, N8=NA: (9.078=)2=200 (20 M) OR 08=2.0603=2

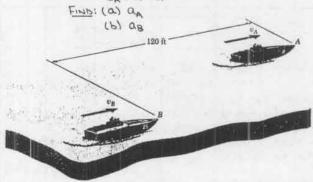
(b) FOR RUNNER B: $N_B = (\hat{A}_B^0)_0 + \Omega_B(t-t_B)$ WHERE t_B is the time at which he begins to run.

AT t=1.825: 9,078 = (2.0603 =)(1.82-to)5

OR to=-2.595

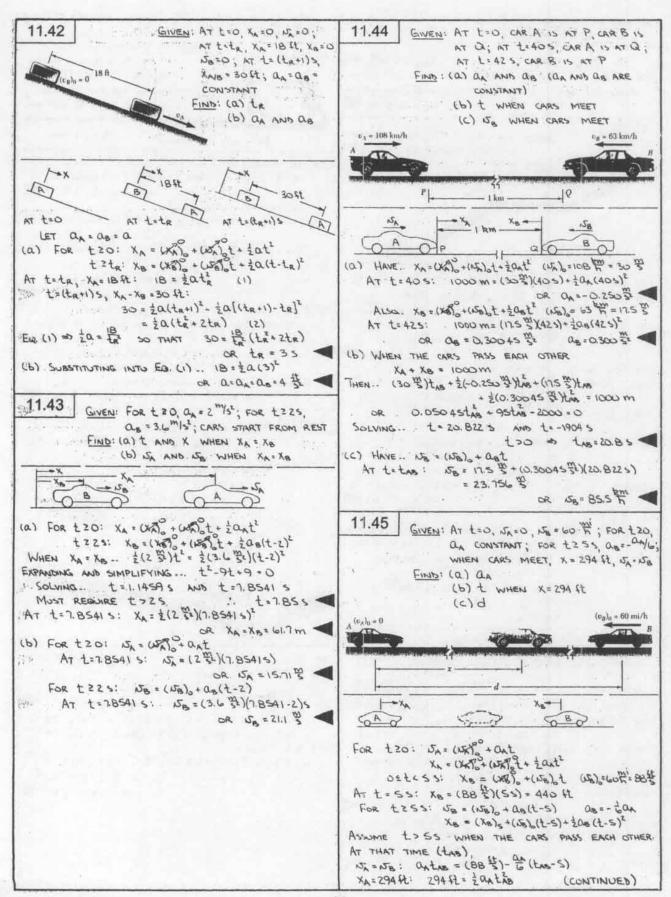
.. RUNNER B SHOULD START TO RUN 2.59 S BEFORE A REACHES THE EXCHANGE BONE.

11.41 GIVEN: AT t=0, $X_{NB}=120$ ft, $N_{A}=N_{B}=105$ mi α_{A} , $\alpha_{B}=constants$; AT t=85, $X_{A}=X_{B}$, $N_{A}=135$ m



(a) HAVE.. UT = (UT) + OAT (UT) = 105 M=154 \$ AT L=85: UT = 135 M=198 \$ THEN 198 \$ = 154 \$ + OA(85)

(b) HAVE.. $X_A = (X_A)_0 + (N_A)_0 + \frac{1}{2} \alpha_A t^2$ $(X_A)_0 = 120 \text{ ft}$ AND $X_B = (X_B)_0 + (N_B)_0 + \frac{1}{2} \alpha_B t^2$ $(X_B)_0 = 154 \frac{15}{5}$ At t = 8 s: $X_A = X_B$.: $120 \text{ ft} + (154 \frac{15}{5})(8 \text{ s}) + \frac{1}{2}(5.50 \frac{15}{5})(8 \text{ s})^2 = (154 \frac{15}{5})(8 \text{ s})^2$ $+ \frac{1}{2} \alpha_B (8 \text{ s})^2$ OR $\alpha_B = 9.25 \frac{15}{12}$



11.45 CONTINUED

THEN Qx (2tA8-2) = 88 = 20xtx8 = 294 OR 44t2-343tA8 +245=0

SOLVING .. + + = 0.795 5 AND + = 7.005
(a) WITH + + > 55, 294 ft = 20,(7.005)

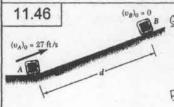
OR ON 1200 32

(b) From above tas=7.00 s ...

Note: An acceptable solution cannot be found

IF IT IS Assumed that tas \$55.

(C) HAVE.. d = X + (XB) + 18 + (88 = X + 100 - 5) 5 + 2 (-6 × 12 × 20 × 12 × 30 × 5) 5 OR d = 906 ft



GIVEN: (M) AND (M); AT t=15,

BLOCKS PASS EACH OTHER,

AT t=3.45, X8=d;

(XA)MAX = 21 St; QA, QB

ARE CONSTANT AND

FIND: (Q) QA AND QB

(b) d (c) No AT t=15

(a) Have.. $J_A^2 = (J_A)_0^2 + 2Q_1(X_1 - Q_1^2)_0^2$ WHEN $J_A = (X_A)_{AA}$ THEN.. $O = (27 \frac{1}{5})^2 + 2Q_A(21 \frac{1}{5})^2$ OR $Q_A = -17.3571 \frac{1}{5}$ OR $Q_A = 17.36 \frac{1}{5}$

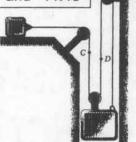
NOW.. $X_A = (X_A^A)_0 + (X_A)_0 + \frac{1}{2}\alpha_A t^2$ AND $X_B = (X_B^A)_0 + (X_B^A)_0 + \frac{1}{2}\alpha_B t^2$ AT t=1 5, THE BLOCKS PAST EACH OTHER ... $(X_A)_1 + (X_B)_1 = d$ AT t=3.4 5, $X_B = d$ THUS... $(X_A)_1 + (X_B)_2 = (X_B)_2 a$

OR $(27\frac{4}{5})(15)+\frac{1}{2}(-17.3571\frac{1}{5}(15)^2]+[\frac{1}{2}a_8(15)^4]=\frac{1}{2}a_8(3.45)^4$ OR $(27\frac{4}{5})(15)+\frac{1}{2}(-17.3571\frac{1}{5}(15)^2]+[\frac{1}{2}a_8(15)^4]=\frac{1}{2}a_8(3.45)^4$

(b) At t=3.45, X8=d: d=2(3.4700 \$\$\) OR d=20.1 ft

(C) HAVE.. No = (No) + OAt AT t=15: No = 2) \$+ (-17.3571 \$)(15) OR No = 9.64 \$

11.47 and 11.48



GIVEN: BLOCKS A AND B AND THE DULLEY/ CABLE SYSTEM SHOWN

FROM THE DIAGRAM (NEXT COLUMN) HAVE ..

(2)

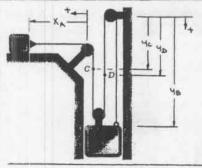
 $X_A + 3 Y_B = CONSTANT$ THEN $J_A + 3 J_B = 0$ (1)

ax + 30 g = 0

AND

(CONTINUED)

11.47 and 11.48 CONTINUED



11.47 GIVEN: UZ = 6 3 + (b) UB

(C) ½C/b
(a) SUBSTITUTING INTO EQ. (1)... 6 3 + 35 = 0

OR 15=2 3 1 ◀

(b) From the DIAGRAM.. 48+40 = CONSTANT
THEN UB+ UD = 0

(C) FROM THE BIAGRAM. X+4c=CONSTANT
THEN IS+IS=0 :. IS=-6 =

NOM - 7518 = 15-12P (5 2) = -8 2

: 248=821

11.48 GIVEN: AT t=0, 15 =0; QB = CONSTANT 1; WHEN | DXA = 0.4 m, |SA = 4 %

FIND: (a) QA AND QB

CB) IS AND [40-40) AT t=25

(a) Eq.(2): $Q_A + 3Q_B = 0$ and $Q_B = 0$ constant and positive $\Rightarrow Q_A = 0$ constant and negative

ALSO, EQ.(1) AND $(U_B)_{a=0} \Rightarrow (U_A)_{a=0}$ THEN $U_A^2 = (V_A)_a^2 + 2Q_A(X_A - (X_A)_a)^2$ WHEN $|Q_{X_A}| = 0.4 m$: $(4 \%)_a^2 = 2Q_A(0.4 m)$

THEN .. SUBSTITUTING INTO EQ (2) .. - 20 52+3 ag = 0

OR ag = 30 52

OR ag = 20 54

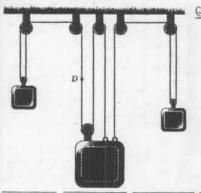
OR ag = 20 54

(b) HAVE.. US = (NE) + ast AT t=25: NE = (3 %)(25)

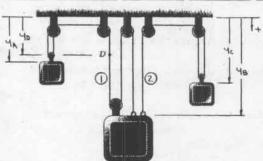
ALSO -- 48 = (48) + (188) 2+ 20et2 AT t=25: 48-(48) = 2(23 5)(25)2

OR 48- (48)=13.33m1

11.49 and 11.50



GIVEN: BLOCKS A, B, AND C AND THE PULLEY / CABLE SYSTEM SHOWN



FROM THE BIAGRAM ..

CABLE 1: 244+348 = CONSTANT THEN .. SULT + BUB = O (1)

20A+308=0 (2)

CABLE 2: 48+24c = CONSTANT THEN - NE+2NE = 0 (3) AND 08+20 =0 (4)

GIVEN: NB = 24 5 11.49 FIND: (a) UTA

(P) RE

(C) No

(9) PAB

(a) Substituting into Ea. (1). 205 +3(24 5)=0 OR NT : 36 31

(b) SUBSTITUTING INTO EQ. (3) .. (24 5)+24 = 0 DR 5 = 12 1 1

(C) FROM THE DIAGRAM _. 24x + 46 = CONSTANT THEN .. SUS + NO = O

SUBSTITUTING FOR UT .. 2(-36 5)+15=0 OR U5 = 72 13 1

(d) HAVE - Noja = No-No = 72 8 - 24 8

OR WHB = 48 51 .

11.50 GIVEN: (US)=0; Q = CONSTANT 1; AT 1=125, 5 = 18 IN/S

FIND: (a) QA, QB, AND QC

(b) NB AND [48-(48)] AT t=85

(a) Eas. (3) AND (1) AND (UC) =0 => (UA) = (UB) =0 ALSO, EQS. (4) AND (2) AND QC IS CONSTANT AND POSITIVE \$ Q IS CONSTANT AND NEGATIVE QA IS CONSTANT AND POSITIVE

THEN .. IS = (ISK) + ant

AT t=125: 18 1 = ax (125) OR ax=1.531 (CONTINUED)

11.50 CONTINUED

SUBSTITUTING INTO EQ. (2) .. 2(1.5 52)+300 =0 OR Q8=1.031

SUBSTITUTING INTO EQ. (4) .. (-1.0 52) + 20c = 0 OR QL = 0.5 511

(b) HAVE .. No = (NO)+ ast

AT t= BS: NB= (-10 54) (BS)

OR 58 = 8.0 5

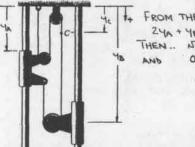
ALSO .. 48 = (48) + 158 of + 2 ast AT t=85: 48-(48) = 2(-10 12)(85)2

OR 40-(48) = 32.0 IN.

11.51 and 11.52



GIVEN: COLLARS A AND B AND THE PULLEY CABLE SYSTEM SHOWN



+ FROM THE DIAGRAM .. ZYA + YB + (YB-YA)= CONSTANT THEN .. NA+ 2NE =0 (1) 04+200=0

11.51 GIVEN: (NA)0=0; QA = CONSTANT ; AT L=85, |NB/A| = 24 111/5

FIND: (a) QA AND QB (b) we AND 48-(48)0 AT 1:65

(a) Eq. (1) AND (NA)=0 => (NB)0 ALSO, EQ. (2) AND QA IS CONSTANT AND NEGATIVE

A QB IS CONSTANT AND POSITIVE

No = (45%) + ast THEN .. NA = (NATO + OAT NOW - UBIA = UB - UA = (as-aa)t

FROM Ea.(2).. a==-2aa so THAT NEWA =- 2aat AT t= 85: 24 1 = - 204 (85)

OR Q4 = 2 32

AND THEN -- OB =- 2 (-2 52) OR QB = 1 52 (b) AT t= 65: NB = (1 15)(65)

OR 158= 6 5 1 Now .. 40 = (48) + (15) t + 200t2 AT t= 65: 48 - (48) = = (1 1/2)(65)2 OR 48-(40) = 18 IN.

(CONTINUED)

11.52 CONTINUED

GIVEN: NB = 12 5 11.52 FIND: (a) JA (P) VE

(c) NG/B

(a) Substituting into Ea. (1). Un+2(12 11)=0 OR 154 - 24 31

(b) FROM THE DIAGRAM .. 24x+ 4c = CONSTANT THEN .. ZNA + NC = 0 SUBSTITUTING .. 2 (-24 5) + 150 = 0

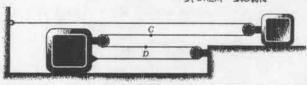
OR NE = 48 51

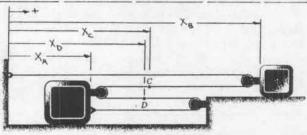
(C) HAVE .. NEIB = NC-NB = (48 3)-(12 3)

OR 5/8-36 51

11.53 and 11.54

GIVEN: BLOCKS A AND B AND THE PULLEY/CABLE SYSTEM SHOWN





FROM THE DIAGRAM .. XB+ (XB-XA)- ZXA = CONSTANT THEN .. 258-35A=0 (1)

208-30A=0 (2) AND

11.53 GIVEN: NE = 300 FIND: (a) Un (b) Jc

(C) 50 (d) 50/A

(a) SUBSTITUTING INTO EQ. (1). 2(300 5)-354=0 OR 5 = 200 5

(b) FROM THE DIAGRAM. XB+(XB-Xc) = CONSTANT THEN -- 258 -50 = 0 SUBSTITUTING -. 2(300 5)-NE = 0 OR 150=600 5

(C) FROM THE DIAGRAM -- (XC-XA)+ (Xb-XA) = CONSTANT THEN - No - 2NA + NO = 0 SUBSTITUTING -- 600 5 - 2(200 5) + 45 = 0

OR NS = 200 mm +

(d) HAVE - NGA - NG - NA = 600 mm - 200 mm

OR 154 = 400 TO -* ALSO HAVE .. - X6 - X4 = CONSTANT

THEN - UD + UT = 0 (3) (CONTINUED) 11.54 CONTINUED

GIVEN: (Ug) = 150 5; QB = CONSTANT; WHEN 11.54 1 - (XA) = 240 mm - , 15x = 60 5 FIND: (a) QA AND QB

(b) an

(C) 58 AND X8-(X8) AT 1=45

(a) FIRST OBSERVE THAT IF BLOCK A MOVES TO THE RIGHT, NA - AND EQ. (1) => NE -. THEN USING EQ. (1) AT t=0.. 2(150 5)-3(NA)=0 OR (N) = 100 5

ALSO, EQ. (2) AND QB = CONSTANT => QA = CONSTANT

THEN -. No = (No) = + 20a [Xa - (Xa)] = (100 mm) = 20a (240 mm) OR 04 = - 40 MM

OR Qx=13.33 52 -THEN, SUBSTITUTING INTO EQ. (2)__

zag-3(-40 mm)=0

OR an = - 20 50 ag=20.0 52

(b) FROM THE SOLUTION TO PROBLEM 11.53 ... 15p+15A=0

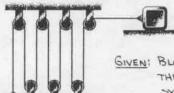
THEN .. as + an = 0 SUBSTITUTING ... ab + (- 40 mm) = 0

OR ab= 13.33 52 -

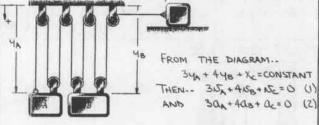
(C) HAVE -- NB = (NB) + ast AT t=45: NB = 150 "5" + (-200 "5")(45)

OR 158 = 70.0 mm ALSO - 48 = (48) + (48) ot + 2 ast2 AT t= 45: 48-(48) = (150 5)(45)+ 2(-200 52)(45)2 OR 48- (48)= 440 mm -

11.55 and 11.56



GIVEN: BLOCKS A, B, AND C AND THE PULLEY/CABLE SYSTEM SHOWN



(CONTINUED)

11.55 and 11.56 CONTINUED

GIVEN: NE = 20 51; (NE) = 30 51; 11.55 an = constant; AT t-35, Xc-(Xc) = ST mm -FIND: (a) (NE)

(b) QA AND QC

(C) 44- (HA) AT t=55

(a) SUBSTITUTING INTO EQ. (1) AT t=0 ... 3(-30 mm) + 4(20 mm) + (NE) = 0

OR (NE) = 10 mm (b) HAVE .. Xc = (Xdo + (x)ot + Eact

AT t=35: 57 mm = (10 5)(35)+ 20c(35)2 OR ac & ST -

NOW .. IS = CONSTANT => QB = 0 THEN .. SUBSTITUTING INTO EQ. (2) .. 30x+4(0)+(6 mm)=0

OR QL= 2 SE (C) HAVE - 4A = (4A) + (UT)) + 2 ant AT t= 55: 4-(4A)0 = (-30 5)(55)+2(-2 5)2(55)2 OR 44-4410=175 mm 1

11.56 GIVEN: (NB) = 0, Qx = CONSTANT, (Qc) = 75 mm; AT t=25, NB = 480 mm 1, NE = 280 mm

FIND: (a) QA AND QB (b) (15A) AND (15c). (C) Xc-(Xc) AT t=35

(a) Eq.(2) AND Q = CONSTANT AND $a_c = constant \Rightarrow a_B = constant$ Then.. $a_B = (a_B) + a_B t$ At t = 2 s: $480 \% = a_B(2 s)$

OR QB = 240 55

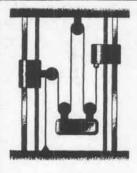
SUBSTITUTING INTO EQ. (2) .. 30x + 4(240 52)+ (75 52)=0 OR Q4 = 345 52

(b) HAVE .. JE = (NE) + act AT t=25: 280 mm = (15) + (75 mm)(25)

OR (UE) = 130 5 -> THEN, SUBSTITUTING INTO EQ. (1) AT t=0 .. 3(5) +4(0)+(130 3)=0

OR (N) = 43.3 mm (C) HAVE -- Xe = (Xe) + (NE) ot + 2 act2 AT t= 3 5: X - (Xc) = (130) (3 5)+ 2(75 52) (35)

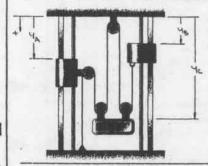
11.57 and 11.58



GIVEN: COLLARS A AND B. BLOCK C. AND THE PULLEY / CABLE SYSTEM SHOWN

OR X - (Xc) = 728 mm -

11.57 and 11.58 CONTINUED



FROM THE DIAGRAM .. -4A + (4c-4A)+24c + (4c-48)= CONSTANT -215-15a+415-0 (1)

-204-08+40c=0 (2)

AND

GIVEN: (NT) =0, (Qx)=7501; (NE)=8151 11.57 ag = constant ; AT t= 25, 48-486= 20 IN.

FIND: (a) as AND QL (b) t WHEN NE =0

(C) 4c-(4c)0 WHEN NE=0

(a) HAVE .. Ye = (40) + (No) t + 2 ast AT t=25: -20 IN. = (-8 5)(25)+ 2 ab(25)2 OR ag = 2 52 1

THEN. SUBSTITUTING INTO ED. (2).

-2(75)-(-25)+4ac=0 DR QC = 3 1 12 1

(b) SUBSTITUTING INTO EQ (1) AT 1=0 .. -2(0)-(-81/5)+4(15)=0 OR (15)=-25

Now .. We = (UE) + act WHEN N=0: 0= (-2 13)+ (3 5)+ OR 1 = 35 t=06675

(C) HAVE - 4= (4c) + (UC) ot + 2 act? AT 1= 35: 4c-(4c) = (-2 15)(35)+ 2(310)(35)2 -1c-141 = 0.667 in. OR

11.58 GIVEN: (NA) =0, (NE) =0; QA = 32 521; QB = CONSTANT + WHEN HB-(HB) = 3ZIN. 四日日日

FIND: (a) ac

(b) DISTANCE TRAVELED BY C AT 1.35

(a) HAVE .. No = (1502 + 200 (40- (40)) WHEN YR-(48)= 32 IN.: (8 5)2 = 208 (32 IN.)

THEN, SUBSTITUTING INTO EQ.(2) .. -2(-3世岁)-(1岁)+40=0 OR ac = 4 (1-62) 11 -

(b) SUBSTITUTING INTO EG. (1) AT t=0 .. -2(0)-(0)+4(NE)=0 OR (NE)=0

Now. de ac = \$ (1-62) AT t=0, 5=0: | bd de = 1 \$ (1-62)dt OR No= \$ (t-2+3)

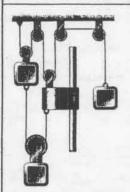
THUS, No =0 AT \$ t(1-2t)=0 OR t=0, t= 755 THEREFORE, BLOCK (INITIALLY MOVES DOWNWARDS (15,00) AND THEN MOVES UPWARD (VC 40). Now - Etc = 4(1-213)

AT t=0, y= (40) : | (40) dye = | t + (t-zt) dt OR 40-(40) = 10 (t2-t4)

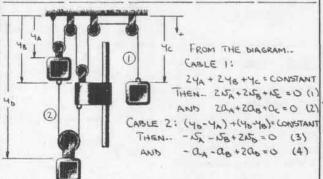
AT t= (\$): 4c-(4c)= \$[(1/2)2-(1/2)]= \$ 10. AT t=35: 4c-(4c)= \$[(3)2-(3)4]=-9 IN. ". TOTAL DISTANCE TRAVELED = (32)+1-9-32 = 916 IN.

= 9.06 IN.

11.59 and * 11.60



GIVEN: BLOCKS A. C. AND D. COLLAR B, AND THE PULLEY CABLE SYSTEM SHOWN



11.59 GIVEN: AT t=0, N=0; ALL ACCELERATIONS CONSTANT; Qc/8 = 60 mm/4, Qc/8 = 110 mm/4, FIND: (a) NE AT t = 35

(b) 40- (40) AT t=55

(a) HAVE - acre = ac-ae = -60 or ae = ac+60 AND adja = ab-aa = 110 OR aa = ab-110 SUBSTITUTING INTO EQS (2) AND (4) ..

Ea (2): Z(ab-110)+Z(ac+60)+ac=0

OR 3ac + 2a0 = 100 (5)

Ea. (4): - (ab-110) - (ac+60) + 2ab=0 OR - ac + ab = - 50

SOLVING EQS. (S) AND (6) FOR QC AND QD ... ac = 40 mm an = -10 mm

Now .. No = (Not) + act AT t= 35: 5= (40 52)(35)

OR 15=120 51

(b) HAVE .. 40 = (40) + 450 2+ 200 th AT t=55: 40-(40) = 2(-10 50)(55)2

OR 46-(46) = 125 mm

GIVEN: AT t=0, J=0, (YA) = (YB) = (YC); 11.60 ALL ACCELERATIONS CONSTANT; AT t=ZS 4014 = 580 mm ; WHEN 1281 = 80 mm 4x-(4x)=160 mm+, 4x-(4x)=320 mm+;

FIND: (a) QA AND QB

(a) HAVE - 4 = (4) + (4) = 4 & 2 = 600 5 AND 40 = (40) + (40) t+ 200 t2

(CONTINUED)

11.60 CONTINUED

THEN - YOU = Ye - YA = 2 (ac-ax)t2 AT t= 25, yea = -280 mm: -280 mm = 2 (ac-an)(25)2 OR ac = 04-140 (5) SUBSTITUTING INTO EQ. (2) __ ZOA + ZOB + (OA-140) = 0 OR ax = 3(140-200) (6)

Now. No = (Not + ast NA = LUXTO + ant

:. NBIA = NB - NA = (08-0A) t ALSO - 48 = (48) + WETO + 2 ast WHEN SBIA = 80 = 1: 80 = (08-0A) t

47x=160 mml: 160 = 20x22 448 = 320 mm+: 320 = 2 ast2

THEN 160 = 2 (ag-an) t2 USING EQ. (7) .. 320 = (80) t OR 1= 45 160 = 2 an (4)2 OR an = 20 52 THEN 320 = 2 ag (4)2 OR Q8 = 40 51 6 NOTE THAT ED. (6) IS NOT USED; THUS, THE PROBLEM IS OVER DETERMINES.

ALTERNATIVE SOLUTION: HAVE .. 152 = (154) + 204 (44-(41)) 150 = (156) + 208 (46-(48)) THEN. JOHN = NO-NA = 1200 [40-(40)] - 1200 [40-(40)] WHEN NOBLE - 80 5 +: 80 5 = 12[(ag(320 mm) - 10x (160 mm)] OR 20 = 12 (1200g - 1100g)

SOLVING EWS (6) AND (8) YIELDS QUE AND QUE. (b) SUBSTITUTING INTO EQ. (5).

ac = 20- 140 = - 120 3

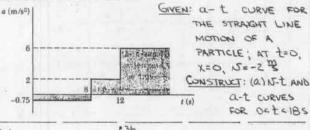
AND INTO EQ. (4) - (20 52) - (40 52) + 200 =0 OR as = 30 52

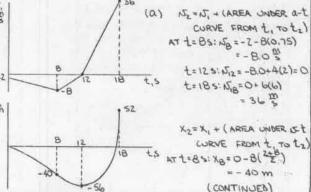
Now. No = (NET 0+ act -600 5 = (-120 52) t MHEN TE =- 600 2:

on tess ALSO -- 40 = (40)0 + (40)0 t + 200 t2 AT t= 55: 46-(46) = 2 (30 50) (55)2

OR 46-(40)=375 mm

11.61 and 11.62





11.61 and 11.62 CONTINUED AT t= 125: X12=-40- 2(4)(8)=-56m 11.63 t=185: X18=-56+2(6)(36)=52m 11.61 FIND: (b) X, N, AND TOTAL DISTANCE TRAVELED AT 1=185 (b) READING FROM THE CURVES. XIB=52m NS18=36 3 FROM t=0 TO t=125: DISTANCE TRAVELED = 56M t=125 TO t=185: DISTANCE TRAVELED = 52-(-56) = 108 m " TOTAL DISTANCE TRAVELES = (56+108)m = 164 m 11.62 FIND: (b) NMIN (C) Xmin (b) READING FROM THE IJ-L CURVE. ISMIN =- 85 (C) READING FROM THE X-T CURVE .. XMIN =- 56m 11.63 and 11.64 GIVEN: IST CURVE FOR THE STRAIGHT LINE MOTION OF A PARTICLE; AT t=0, X=- S40 ft v (ft/s) CONSTRUCT: (a) a-t AND X-L 60 CURVES FOR OCTUSOS (a) a slope of N-t curve AT TIME t FROM t=0 to t=105: N=CONSTANT => Q=0 t=105 to t= 265: a= -20-60 =- 5 st t = 265 TO t= 415: N= CONSTANT => Q=0 t=415 To t=465: Q= -5-(-20) = 3 52 t >465: 15= CONSTANT = Q=0 a, ft X2 = X, + (AREA UNDER IS-t CURVE FROM t, TO tz) AT t=105: X10=-540+10(60)=60 ft

 $X_2 = X_1 + (AREA UNIVER 15-t CURVE FROM t, To t_2)$ AT $t = 10 \text{ Si}: X_{10} = -540 + 10(60) = 60 \text{ ft}$ NEXT FIND TIME AT WHICH N=0. Using SIMILAR TRIANGLES... $\frac{t_{150} - 10}{60} = \frac{2t_0 - 10}{80} \text{ or } t_{150} = 22.5$ AT t = 22.5: $X_{22} = 60 + \frac{1}{2}(12)(160) = 42.0 \text{ ft}$ t = 24.5: $X_{24} = 42.0 - \frac{1}{2}(4)(12.0) = 38.0 \text{ ft}$ t = 44.5: $X_{41} = 380 - 15(20) = 80 \text{ ft}$ t = 50.5: $X_{50} = 17.5 - 4(5) = -2.5 \text{ ft}$ X_{50} X_{50}

-540

11.63 and 11.64 CONTINUED

(b) From t=0 to t=225: DISTANCE TRAVELED AT t=505

(C) t WHEN X=0

(b) From t=0 to t=225: DISTANCE TRAVELED = 420-(-540)

= 960 H

t=225 to t=505: DUTANCE TRAVELED=1-2.5-4201 = 422.5 ft

.. TOTAL DISTANCE TRAVELED = (960+422,5) A = 1382,5 H

BETWEEN 465 AND SOS: $\frac{(t_{x=0})_2-96}{(t_{x=0})_2-46} = \frac{4}{20}$

or (1x=0)=49.55

11.64 FIND: (b) Xmax

(c) t WHEN X=100 St

(b) READING FROM THE X-t CURVE ... Xmm = 420 ft

(C) BETWEEN IDS AND ZZS_

100 ft = 420 ft - (AREA UNDER LET CURVE FROM t, TO 225) HT

OR 100=420-2(22-1,)(15) OR (22-1,)(15) = 640 USING SIMILAR TRIANGLES...

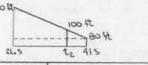
60 \$ N, 225

 $\frac{S_1}{22-t_1} = \frac{60}{12}$ or $S_1 = 5(22-t_1)$ THEN ... $(22-t_1)[5(22-t_1)] = 640$

OR 1,=10.695 AND 1,= 33.3 5

HAVE 1054, 225 \$ t = 1069 \$

BETWEEN 265 AND 415



 $\frac{41-t_z}{20} = \frac{15}{300}$ or $t_z = 40.5$

11.65

GIVEN: AT t=0, N=200 \$\frac{1}{2}, X=600m;

FOR 600 M & X & 586M,

Q=CONSTANT; FOR

586 M < X & 30 M, N=50 \$\frac{1}{2};

WHEN X=0, N=0

FIND: (a) trotal
(b) QINITIAL

ASSUME SECOND DECELERATION IS

CONSTANT. ALSO, NOTE THAT

ZOO H= SS.SSS S, SO H=13.888 S

AT S

-13.888

(CONTINUED)

(CONTINUED)

11.65 CONTINUED

(a) NOW. AX = AREA UNDER J-L CURVE FOR GIVEN TIME INTERVAL

THEN. (586-600) m = -t, (55,555+13.888) m OR t, = 0.4032 S (30-586)m=-t2(13,888 m) OR t2 = 40 0346 5 (\$ 888.81)(et) == m(08-0)

OR t3 = 4.3203 5 : trotal = (0.4032+40.0346+4.3203)5

OR THOTAL = 44.85 (b) HAVE .. QINITIAL = AVINITIAL [-13.888-(55.555)] \$ 0.4032 5

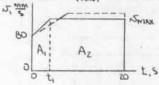
= 103.3 m OR QINITIAL = 103.3 52 1

11.66

GIVEN: AT t=205, X=4m; 5=80 5, amax = 60 mm; town = 155,

JPAINT = CONSTANT FIND: (Smx) MIN

FIRST NOTE THAT (80 5)(205) 4000 mm, 50 THAT THE SPEED OF THE PALLET MUST BE INCREASED. SINCE NEAINT = CONSTANT, IT FOLLOWS



THAT NPAINT = NMAX AND THEN t, = 55. FROM THE N-1 CURVE, A, + A2 = 4000 mm AND IT IS SEEN THAT (WMAX) MIN OCCURS WHEN (= 15mm - 80) 15 MARINE = 60 mm -BO) is MAXIMUM. THUS

1, = (15mm - 80)/60 AND 1, (80+ 17max) + (20-1,)(15max)= 4000 SUBSTITUTING FOR t ...

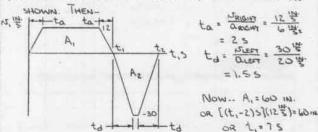
(15max - 80) (80+ 15max) + (20 - 15max - 80) Nmax = 4000

SIMPLIFYING -- 15max - 25605max + 486 400 =0 SOLVING -- UMAX = 207 5 AND UMAX = 2353 5 t, (55 t, > 55 NTMAX = 2353 "15" ". (NSMAX) MIN = 207 MM

11.67 GIVEN: (Nonx) RIGHT = 12 5 (UMAX) LEFT = 30 5; ARIGHT = ± 6 182 QLEFT = = ZO SE FIND: (a) torcle CONSTRUCT (b) No-t AND X-L CURVES

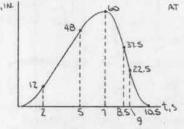
11.67 CONTINUED

(a) AND (b) THE Not CURVE IS FIRST DRAWN AS



AND A = 60 IN. OR { (tt_-7)-1.5) s}(30 5)=60 in. OR t2=10.55

: tarce = 10.55 NOW .. torche = 12 HAVE .. X .: = X; + (AREA UNDER J-1 CURVE FROM L; TO L .:) X. IN. | AT 1=25: X2=2(2)(12)=12 IN.



t=55: X5=12+(5-2)(12) = 4B IN. t=75: X7=60 14. t=8.55: x =60-2(1.5)(30) = 37.5 IN. t=95: 14=375-(05)(30)

= 22.5 IN. t=10.55: X10.5=0

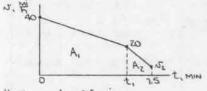
11.68

40 mi/h ional becomeda. STATE THE CHAPTER STATE COMMISSION AND ADDRESS OF

GIVEN: AT 1=0, J=40th, X=0; WHEN X=2.5 mi, L= 20 H; AT t= 7.5 MIN, X=3 mi; CONSTANT DECELERATIONS

FIND: (a) t WHEN X= 2.5 Mi (6) IT WHEN X = 3 mi (C) afinal

THE IJ-t CURVE IS FIRST DRAWN AS SHOWN.



(a) HAVE .. AVE .. A = 2.5 mi OR (t, min) (40+20) mi x 1h Lomin = 2.5 mi OR t, = 5 MIN

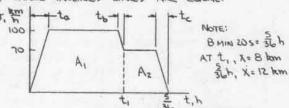
(b) HAVE.. Az = 0.5 mi 20+15 mi 1 1 1 60 min = 0.5 mi OR 15 = 4 TH

(C) HAVE OFINAL = Q12 = (4-20) h , szeuft, 1 mm , 1 h (7.5-5) min mi

OR OFINAL = - O. ISLA 52

11.69 GIVEN: (NAME) AC = 100 H, (NAME & = 70 H; NT = NB =0; tAB = B MIN, 205; ICI = CONSTANT; N= NTMAX AS MUCH AS POSSIBLE FIND: a

THE J-t CURVE IS FIRST DRAWN AS SHOWN, WHERE THE MAGNITUDES OF THE SLOPES (ACCELERATIONS) OF THE THREE INCLINES LINES ARE EGUAL.



DENOTING THE MAGNITUBE OF THE ACCELERATIONS HAVE ...

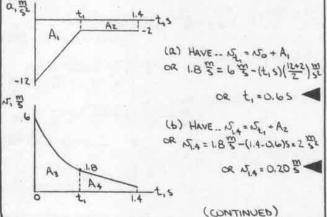
BY a H WHERE Q IS IN AMY THE TIMES ARE IN h. Now. A = 8km: (t,)(100)-2(ta)(100)-2(th)(30)=8 SUBSTITUTING .. 100t, - 2 (180)(100) - 2 (30)(30) = 8 OR t, = 0.08+ 54.5

ALSO. Az=4 km: (36-t,)(10)-2(tc)(10)=4 SUBSTITUTING .. (36-1,)(10)-2(2)(10)=4

OR t = 103 - 1260 - 3 OR Q = SI 259 12 1000m x (3000 5)2 OR a=3,96 52

11.70 GIVEN: 50=6 5; FOR 05+5+, 0x+;
FOR t, 6+51.45, 0=-2 52;
AT t=0 0=-12 52; AT t=1,
0=-2 52, 5=1.8 5 FIND: (a) t, (b) N AND X AT t=1.45

THE Q-t AND W-T CURVES ARE FIRST DRAWN AS SHOWN.



11.70 CONTINUED

NOW .. XI.4 = A3+A4, WHERE A3 IS MOST EASILY DETERMINES USING INTEGRATING. THUS... FOR 0= +++ : a = -2-(-12) + -12 = 3+-12 Now .. dt = a = 3t-12 At t=0, v=6 5: 16 dv = 16 (3t-12)dt

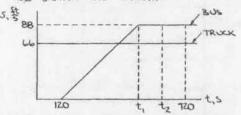
HAVE ... $\frac{dx}{dt} = N = 6 + \frac{2s}{2}t^2 - 12t$ HAVE ... $\frac{dx}{dt} = N = 6 - 12t + \frac{2s}{2}t^2$ HEN... $A_3 = \begin{cases} x_{11} & 0 \\ 0 & 0 \end{cases}$ $= \left[6t - 6t^2 + \frac{2s}{2}t^2 \right] = 2.04m$ ALSO... $A_4 = (1.4 - 0.6) \left(\frac{1.6 + 0.2}{2} \right) = 0.8m$ THEN ..

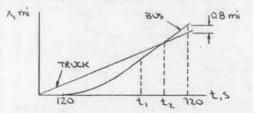
X1.4=(2.04+0.8)m THEN .. OR X1.4 = 2.84 m

11.71 GIVEN: NT = 45 1 ; AT 1=0, XT =0, XB =0; FOR O = 1 = 2 MIN, NB =0; FOR 172 MIN, QB = CONST UNTIL 15 = 60 H, THEN QB = 0; AT 1=12 MIN, X8- XT = 0.8 mi

FIND: (a) t AND X WHEN XB = XT (b) as

FIRST NOTE .. 45 Th = 66 5 (a) Assuming that the BUS REACHES 60 \$ (AT TIME () BEFORE IT PASSES THE TRUCK (AT TIME tz), THE UT AND X-T CURVES CAN THEN BE BRAWN AS SHOWN.





AT t=720 5 (12 MIN): X5-X7 = 0.8 mi or [\frac{1}{2}(4,-120) \(\sigma (88\frac{1}{2}) + (120-\frac{1}{2}) \(\sigma \frac{1}{2} \) = \(\

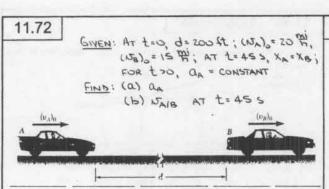
or t = 144 s

AT t=tz: XB = XT OR = (144-120) 5 + (88\$) + (t2-144) 5 + (88\$) = (t2 5) (66 \$) or tz = 5285

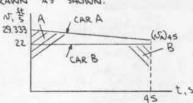
THEN to >t, => ASSUMPTION CORRECT 1. tz = 528 S OR t -8 MIN 485 AT L= t2: XB = XT = (528 5)(66 \$) = 34 BAB St X2 = 6.60 mi

(b) HAVE .. a= (58)4,-0 - (144-120)5

OR 00 = 3.67 \$\$



(a) FIRST NOTE... SO THE 29.333 \$\frac{15}{22}\$ \$\frac{15}{22}\$\$
THE Not corved for the two cars are then DRAWN as SHOWN.



AT t=455, $X_A = X_B$: (AREA), = (AREA), + 200 ft OR (455)($\frac{29.333+15}{2}$) $\frac{45}{5}$ = (455)(22 $\frac{4}{5}$)+200 ft OR (NA), = 23.555 $\frac{45}{5}$ THEN $Q_A = \frac{(NA)s_5 - (NA)}{t_{45}} = \frac{(23.555 - 29.333)}{45.5}$

(b) HAVE - SAIB = 5- SB = (23.555-22) \$\frac{1}{5}\$.

= 1.555 \$\frac{1}{5}\$

OR SAIR = 1.060 \$\frac{1}{15}\$.

11.73 GIVEN: (15) = 36 th (15) = 27 th;

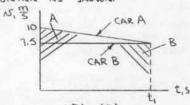
Oh =-0.042 th; CAR A JUST AVOIDS

COLLIDING WITH CAR B

FIND: d



FIRST NOTE.. 36 \$\frac{m}{2} : 10 \frac{m}{2} = 7.5 \frac{m}{2} =



Now.. QA = (NA)4. (NA)0 OR -0.042 = (7.5-10)M/s OR +1 = 59.524 s

OR 1, 59.524 S

AT t=t, XA = XB: (AREA)A = (AREA)B +d

OR (59.524 S)(15+15) = (59.524 S)(1.5 =)+d

OR d= 14.4 m

12 m

11.74

GIVEN: AT t=0, SE=0; FOR

0 \$ SE = 1.8 50; FOR

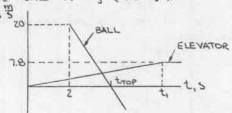
SE=7.8 50; OE=0;

AT t=25, Ne=205;

FIND: THE BALL

HITS: THE ELEVATOR

THE 15-2 CURVES OF THE BALL AND THE ELEVATOR ARE FIRST DRAWN AS SHOWN. NOTE THAT THE INITIAL SLOPE OF THE CURVE FOR THE ELEVATOR IS 1.2 52, WHILE THE SLOPE OF THE CURVE FOR THE BALL 15 -9 (-9.81 52).



THE TIME to IS THE TIME WHEN UE REACHES

7.8 %. THUS... NE = (UE) + Qet

OR 7.8 % = (1.2 %) to or to 6.5 \$

THE TIME LOP IS THE TIME AT WHICH THE
BALL REACHES THE TOP OF ITS TRAJECTORY.

THUS.. Us = (Us) - 9(t-2)

OR 0 = 20 3 - (9.81 32)(4 top - 2)S

OR trop = 4.0387 S

Using THE COORDINATE SYSTEM SHOWN, HAVE...

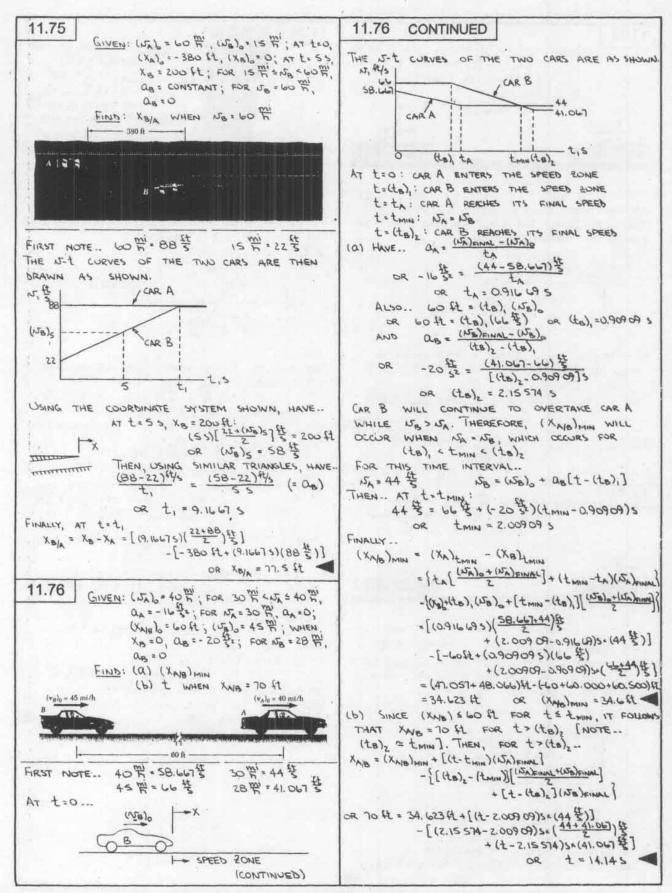
O \(\pm \) \(\pm \)

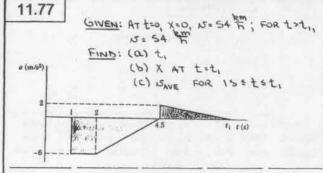
At t = [2 + 2(4.0387 - 2)]5 = 6.07745, $y_8 = 0$ AND AT $t = t_1$, $y_E = -12m + \frac{1}{2}(6.55)(7.83) = 13.35m$... THE BALL HITS THE ELEVATOR $(y_8 = y_E)$ WHEN t = 0

FOR t2trop: 48 = 20.387m-[29(t-trop)2]m
THEN. WHEN 48 = 46...

 $20.387 *1 - \frac{1}{2}(9.81 \frac{m}{5}^{2})(t - 4.0387)^{5}$ $= -12 m + \frac{1}{2}(1.2 \frac{m}{5}^{2})(t + 5)^{2}$ $5.505 t^{2} - 39.6196t + 47.619 = 0$

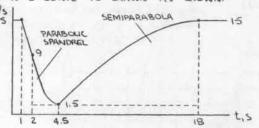
SOLVING.. t=1.525 S AND t=5.675
NOW .. trop < t < t, => t=5.675





FIRST NOTE .. SA = 155 (a) HAVE. IS - No + (AREA UNBER a-t curve FROM ta to tb) THEN .. AT L= 25: N= 15- (1)(6)=9 5 t=4.55: 15=9-2(2.5)(6) + 1.53 t=t: 15=1.5+2(t,-4.5)(2) OR. t1=185

(b) Using THE ABOVE VALUES OF THE VELOCITIES THE N-L CURVE IS DRAWN AS SHOWN. 5, m/s



NOW .. X AT L = 18 5 ... XIA = X30+ E (AREA UNDER THE 15-1 CURVE = (15)(15 m)+(15)(15+9) to t:185) +[(2.55)(1.53)+3(2.55)(7.53)] +[(13.55)(1.5 =)+ }(13.55)(13.5 =)] =[15+12+(3.75+6.25)+(20.25+121.50)]m = 178.75 m OR X18 = 178.8 m (C) FIRST NOTE - X = 15 m X18=178.75 m NAVE = AX = (178.75-15) m = 9.6324 m

DA NAVE = 34.7 5 -

(18-1)5

11.78 GIVEN: AT t=0, X=0, N= B & a (m/s2) CONSTRUCT: (Q) N-t AND t (s) X-L CURVES FOR OKt 44S FIND: (b) X AT t=35 $=-3(t-2)^2$ m/s²

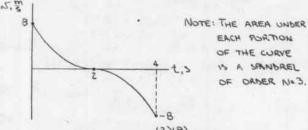
(a) HAVE .. IT = IT + (AREA UNBER a-t curve from t, TO ta) AND .. X2 = X, + (AREA UNDER UST CURVE FROM t, TO (2) (CONTINUED)

11.78 CONTINUED

THEN, USING THE FORMULA FOR THE AREA OF A PARABOLIC SPANDREL HAVE ... AT t=25: 5=8-\$(2)(12)=0

t = 45: 5=0- =(2)(12)=- B=

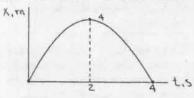
THE J-t CURVE IS THEN BRAWN AS SHOWN.



EACH PORTION OF THE CURVE 15 A SPANDREL OF ORDER NA3.

Now .. AT t=25: X=0+ (2)(B) = 4 m t=45: X=4- (2)(8)=0

THE X-L CURVE IS THEN DRAWN AS SHOWN



(b) HAVE .. AT t=35: Q=-3(3-2)2 -3 52 5=0-3(1)(3)=-1 = X= 4 - (1)(1)

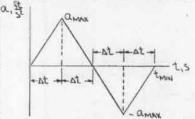
OR X3=3.75 m

11.79

GIVEN: AT t=0, X=0, N=0; XMX=1.2 St; WHEN X= XMAX, N=0; (3) max =4.85

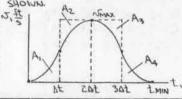
FIND: (a) thin FOR XMAX = 1.2 ft (b) ISMAX AND NAVE FOR OSE STMIN

(a) OBSERVING THAT WMAX MUST OCCUR AT t= 21min, THE Q-t CURVE MUST HAVE THE SHAPE SHOWN. NOTE THAT THE MAGNITUDE OF THE SLOPE OF EACH PORTION OF THE CURVE IS 4.8 45/5.



HAVE .. AT 1= D1: 15=0+2(D1)(amax) = 2 amax st t=2 st: 15max = 2 amax st + 2 (st) (amax) = amou st

USING SYMMETRY, THE UST IS THEN DRAWN AS SHOWN.



(CONTINUED)

11.79 CONTINUED

NOTING THAT A : A : A : A A AND THAT THE AREA UNDER THE 15-2 CURVE IS EQUAL TO XWIAX , HAVE ...

(2 At) (JMAX) = XMAX

Nmax = amax at => 2 amax at = xmax Now ... At = 4.8 fl/s/s so that

2 (4.8 at \$3) at = 1.2 st or at = 0.5 s

THEN tmin = 4 at OR tmin = 2005 ta(ta* 2 t 8.4) = amax at = (4.8 t 8.4) ta(ta* 2 t 8.4) = 4 t 8.4 (4.8)

Nomax = 1.2 15 OR NAVE . O. 6 \$

11.80

GIVEN: XMAX = 1.6 mi; 12max = 4 52, 1(3)max = 0.8 455; Jmax = 20 mi

FIND: (a) thin FOR XMAX = 1.6 mi (b) NAVE

FIRST NOTE .. 20 1 = 29.333 1.6 mi = 8448 ft (a) To OBTAIN thin, THE TRAIN MUST ACCELERATION AND DECELERATE AT THE MAXIMUM RATE TO MIXIMIZE THE TIME FOR WHICH WE DIMAN. THE TIME AT REQUIRES FOR THE TRAIN TO HAVE AN ACCELERATION OF 4 44/52 IS FOUND FROM -- (JE) MAX = CHIAX OR At = 4 tyse 0.BA/3/

NOW .. AFTER 5 5 THE SPEED OF THE TRAIN 15 .. , NS = \(\frac{1}{2}(\text{0}\frac{1}{2})(\text{0}\text{mix})\) (SINCE OF OR \(\frac{1}{2}\frac{1}{2}(5)(4\frac{1}{2}\frac{1}{2})=10\frac{1}{2}\frac{1}{2}\) = CONSTA = CONSTANT)

THEN, SINCE IS LUTIMA, THE TRAIN WILL CONTINUE TO ACCELERATE AT 4452 UNTIL IS I JMAX. THE Q-t CURVE MUST THEN HAVE THE SHAPE SHOWN. NOTE THAT THE MAGNITUSE OF THE SLOPE OF EACH INCLINED PORTION OF THE CURVE IS 0.8 \$15/5.

a, # 1 Ati 55 55 At, 55

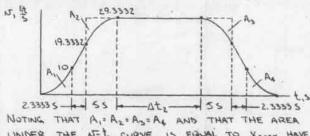
Now .. AT t= (10+ at,) s, 15 = 15mm; :: 2[2(5 s)(4 \$=]+ (at,)(4 \$=) = 29.333 \$= OR At, = 2.3333 S

THEN .. AT t=55: N=0+2(5)(4)=10 \$ t=7.33335: N= 10+(2.3333)(4)=19.3332 5 t=12.3333 5: 15=19.3332+2(5)(4)=29.3332 ft

USING SYMMETRY, THE UT CURVE IS THEN DRAWN AS SHOWN.

(CONTINUES)





UNIDER THE N-L CURVE IS EQUAL TO XMAX, HAVE .. 2 (2.33335) (10+19.3332) 157

+ (10+Atz)s + (29.3332 15) = 844B ft

OR Atz = 275.67 5

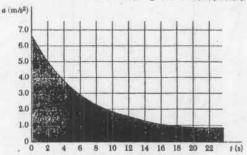
THEN .. + MIN = 4(55) + 2(2.33335) + 275.675

= 300.345 OR tmin = 5.01 Min (b) HAVE .. NAVE = At = 1.6 mi x 36005

OR WAVE : 19.18 1

11.81

GIVEN: Q-t CURVE; AT t=0, X=0, N=0 FIND: (a) IS AT t=85 BY APPROXIMATE MEMS (b) X AT t=205 BY APPROXIMATE MEANS



SOLUTION PROCESURE

1. THE a-t curve is first approximated with a series OF RECTANGLES, EACH OF WINTH At = 25. THE AREA (At)(QAVE) OF EACH RECTANGLE IS APPROXIMATELY EQUAL TO THE CHANGE IN VELOCITY AUT FOR THE SPECIFIED INTERVAL OF TIME, THUS DNS aAVE At

WHERE THE VALUES OF DAVE AND DIS ARE GIVEN IN COLUMNS I AND 2, RESPECTIVELY, OF THE FOLLOWING TABLE.

7 TAHT DIA O = 2 THAT DAITON .S Nz = N, + ANIZ

WHERE DUTY IS THE CHANGE IN VELOCITY BETWEEN TIMES LI AND LL, THE VELOCITY AT THE END OF EACH 25 INTERVAL CAN BE CONIPJIED; SEE COLUMN 3 OF THE TABLE AND THE 15-1 CURVE.

3. THE JULY CURVE IS NEXT APPROXIMATED WITH A SERIES OF RECTANGLES, EACH OF WINTH IS = 25. THE AREA (At) (NAVE) OF EACH RECTANGLE IS APPROXIMATELY EQUAL TO THE CHANGE IN POSITION AX FOR THE SPECIFIED INTERVAL OF TIME. DX = JANE At

(CONTINUES)

11.81 CONTINUED

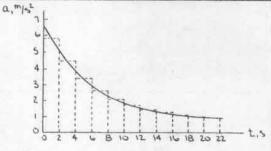
WHERE NAVE AND DX ARE GIVEN IN COLUMNS 4 AND S, RESPECTIVELY, OF THE TABLE.

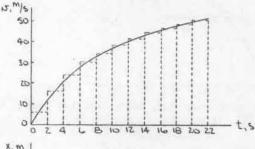
4. WITH X = O AND NOTING THAT

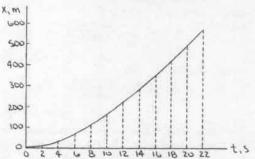
. X2 = X, + DX12

WHERE DX12 IS THE CHANGE IN POSITION BETWEEN TIMES t, AND to, THE POSITION AT THE END OF EACH 25 INTERVAL CAN BE COMPUTED; SEE COLUMN 6 OF THE TABLE AND THE X-L CURVE.

		- 1	-	3	9	2	6
2,5	a, m/s2	GAVE, 1/3	105, 1/5	5, 1/3	SAVE, MYS	Ax,m	X.m
0	6.63	5 21	11/11	0	1777	7777	0
2	5.08	5.86	11.72	11.72	5.86	11.72	11.72
4	3.86	4.47	8.94	20.66	16.19	32.38	44.10
		3.38	6.76		24.04	48.08	
6	2.90	2.58	5.16	27.42	30.00	60.00	92.18
8	2.25	2.06	4.12	32.58	34.64	69.28	152.18
10	1.87	1.21	3.42	36,70	38.41		221.46
12	1.54			40.12		76.82	298.28
14	1.29	1.42	2.84	42.96	41.54	83.08	381.36
-		1.23	2.46	45.42	44.19	86.88	
10	1.10	1.10	2.20		46.52	93.04	469.74
18	1.03	1.00	2.00	47.62	48.62	97.24	562,78
20	0.97	0.94	1.88	49.62	50.56		50.02
22	0.90	17777	/////	51.50	1////		761.14

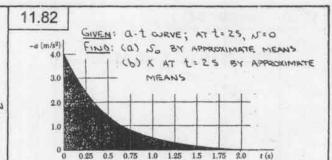






(a) AT t=85, 5=32.58 5 OR 5=17.3 h (b) AT t=205.

X=660 m



SOLUTION PROCEDURE

I. THE Q. I CURVE IS FIRST APPROXIMATED WITH A SERIES OF RECTANGLES, EACH OF WISTH At = 0.25 S. THE AREA (At) (DAVE) OF EACH RECTANGLE IS APPROXIMATELY EQUAL TO THE CHANGE IN YELDCITY AND FOR THE SPECIFIES INTERVAL OF TIME. THUS, AUT = DAVE AT

WHERE THE VALUES OF DAVE AND AN ARE GIVEN IN COLUMNS I AND 2, RESPECTIVELY, OF THE FOLLOWING TABLE.

2. Now .. 5(2) = 50+ 1 adt = 0 AND APPROXIMIATING THE AREA LOUTER THE a-1 CURVE BY EDAVE ST = EDIS, THE INITIAL VELOCITY IS THEN EQUAL TO No=- E AN

FINALLY, USING

No = No + ANG2

WHERE AND IS THE CHANGE IN VELOCITY BETWEEN TIMES t, AND tz, THE VELOCITY AT THE END OF EACH 0.25 INTERVAL CAN BE COMPUTED; SEE COLUMN 3 OF THE TABLE AND THE N. + CURVE.

3. THE JUT CURVE IS THEN APPROXIMATES WITH A SERIES OF RECTANGLES, EACH OF WIDTH 0.25 S. THE AREA (Ot)(NAVE) OF EACH RECTANGLE IS APPROXIMATELY EQUAL TO THE CHANGE IN POSITION AX FOR THE SPECIFIED INTERVAL OF TIME. THUS .. AX = NAVE AT WHERE JAVE AND AX ARE GIVEN IN COLUMNS 4 AND S, RESPECTIVELY, OF THE TABLE.

4. WITH X0 = 0 AND NOTING THAT

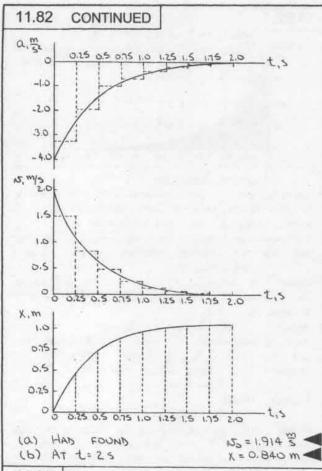
X2 = X, + AX12

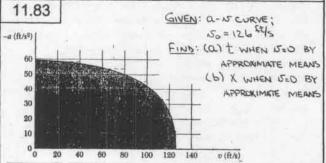
WHERE AXIZ IS THE CHANGE IN POSITION BETWEEN TIMES t, AND tz, THE POSITION AT THE END OF EACH 0.25 5 INTERVAL CAN BE COMPUTED: SEE COLUMN 6 OF THE THBLE AND THE X-T CURVE.

			Z	3	4	5	6
t,s	a, m/sz	DAVE, TYS	AU, m/s	15, m/s	SAVE, MYS	DX, M	X,m
0	- 4.00	2205	11111	1.914	77777	7777	0
0.25	-2.43	-2.772	-0.804	1.110	1.512	0.378	0.378
0,50	-1.40	-1.915	-0.479	0.631	0.871	0.518	0.596
0.75	-0.85	-1.125	-0.281	0.350	0.491	0.153	0.719
		-0.675	-0.169		0.266	0.067	
1.00	-0,50	-0.390	-0.09B	0.181	0.132	0.033	0.786
1.25	-0.28	-0.205	-0.051	0.063	0.058	0.015	0.819
1.50	-0.13	-0.095	-0.024	0.032	0.020	0,005	0.834
1.75	-0.06	-0.030	-0.008	0.008	0.004	0.001	0.839
2.00	0	77777	77777	0	77777	77777	0.840

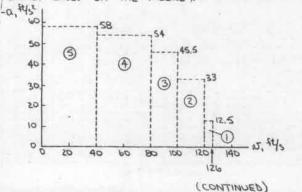
ELAN =- 1.914 m/s

(CONTINUES)





THE GIVEN CURVE IS APPROXIMATED BY A SERIES OF UNIFORMLY ACCELERATED MOTIONS (THE HORIZONTAL DASHED LINES ON THE FIGURE).



11.83 CONTINUED

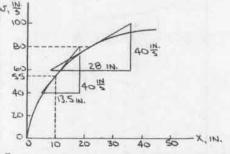
FOR UNIFORMLY ACCELERATED MOTION.. $J_2^L = J_1^L + 2Q(X_2 - X_1)$ $J_2 = J_1 + Q(t_2 - t_1)$ $J_3 = J_4 + Q(t_2 - t_1)$ $J_4 = J_4 + J_4 +$

FOR THE FIVE REGIONS SHOWN ABOVE, HAVE .. a, 4432 | Ax, 42 | At, 5 REGION NI, HIS No, HYS 126 120 -12.5 59.0 0.480 100 120 -33 66.7 3 100 80 -45.5 39.6 0.440 80 - 54 4 40 44.4 0.741 -58 40 13.8 0.690 223.5 2.957

(a) FROM THE TABLE, WHEN N=0 t=2965
(b) FROM THE TABLE AND ASSUMING X=0, WHEN

11.84

| Column | Col

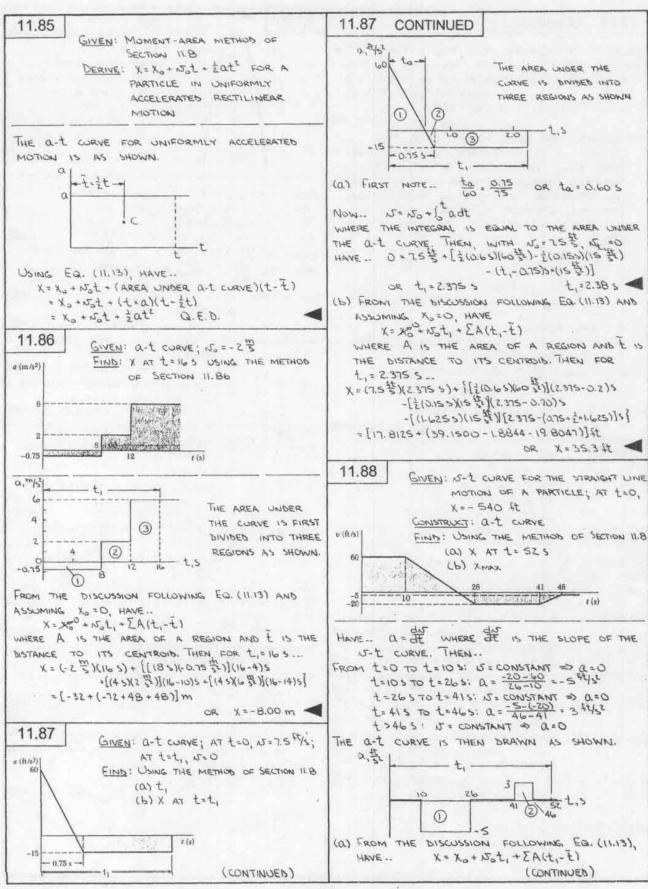


FIRST NOTE THAT THE SLOPE OF THE ABOVE CURVE IS dx. Now... dus Q = D dx

(a) WHEN X=10 IN., $15 = 55 \frac{10}{5}$ THEN.. $a = 55 \frac{10}{5} (\frac{40 \frac{10}{13.5 \cdot 10.}}{13.5 \cdot 10.})$

(b) WHEN $J = 80 \frac{\text{M}}{5}$, HAVE $Q = 163.0 \frac{\text{M}}{5}$. $Q = 80 \frac{\text{M}}{5} \left(\frac{40^{\text{M}}/\text{S}}{28 \text{ M}} \right)$

NOTE: TO USE THE METHOD OF MEASURING THE
SUBNORMAL OUTLINED AT THE END OF SECTION
11.8, IT IS NECESSARY THAT THE SAME SCALE
BE USED FOR THE X AND N AXES (e.g., 1 IN.: SOIN,
1 IN. = SOIN/S). IN THE ABOVE SOLUTION, AN AND
AX WERE MEASURED DIRECTLY, SO DIFFERENT
SCALES COULD BE USED.



11.88 CONTINUED

WHERE A IS THE AREA OF A REGION AND I IS THE DISTANCE TO ITS CENTROID, THEN, FOR 1 = 525 ... X =- 540 ft + (60 \$)(525)+ {-[(165)(5 \$))(52-18)5 +[(5 5)(3 1/1)(52-43.5)5}

= (-540+(3120)+(-2720+127.5)) ft

OR X=-12.50 ft

(b) NOTING THAT XMAX OCCURS WHEN 15=0 (第=0), IT IS SEEN FROM THE UT CURVE THAT XMAY OCCURS FOR 10 5 < £ < 26 5. ALTHOUGH SIMILAR TRIANGLES COULD BE USED TO BETERMINE THE TIME AT WHICH X= XMAX (SEE THE SOLUTION TO PROBLEM 11.63), THE FOLLOWING METHOD WILL BE USED.

FOR 10 SET, 626 S, HAVE $X = -540 + 60t_1 - [(t_1 - 10)(5)][\frac{1}{2}(t_1 - 10)]$ (4t) = -540 + 60t_1 - \frac{5}{2}(t_1 - 10)^2 WHEN X: XMAX: # = 60-5(t,-10)=0 OR (t,) x = 22 S

THEN .. XMAX = - SAD + 60(22) - 2(22-10)2 OR XMAX = 420 St

11.89

GIVEN: X=4t - bt, y=6t - 2t2 x,y-mm, t~5

FIND: IT AND Q AT (a) t=15

(b) t=25

(C) 1=45

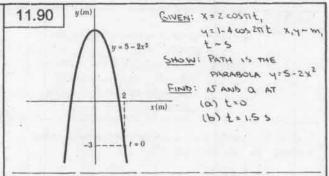
HAVE. X=4t4-6t THEN NX= \$X=16t3-6 ax = disx = 48t2 OUA

4=6t3-2t2 Sy = 27 = 18t-4t ay = dis = 36t - 4

OR N= 17.20 5 54.50 ax = 48 (1)2 = 48 mm ay = 361)-4 . 32 mm OR Q = 57.7 7 1 53.7°

(b) AT t=25: 15x = 16(2)3-6 = 122 mm 5 154 = 18(2)2-4(2)= 64 mm OR 5=137.8 5 4 27.7° ax = 48(2)2 = 192 mm ay=36(2)-4=68 52

OR Q = 204 19 19.50° 0x = 48(4)2=7685 04=36(4)-4=140 55 OR Q=781 5 10.33



HAVE - X=2 COSTIT 4=1-4 COSZTIT 4=1-4(2cosent-1) THEN .. = 5-8(x)2 4=5-5X2 Q.E.D.

trsuierr8= 第= Zu STAMIETTS-=発= Zu ...wall AND $\Omega_{x} = \frac{dux}{dt} = -2\pi^{2} \cosh t$ $\Omega_{y} = \frac{dux}{dt} = 16\pi^{2} \cosh t$

(a) AT t=0: 15x=0 54=0 : U:0 0x = -277 2 m Dy = 1671 = W OR Q = 159.1 52 182.9°

(b) AT t=1.55: 15x=-27 SIN(1.517) 154=87 SIN(271=1.5) = 271 m

DR N= 6.28 # -0x=-272 cos(1.571) Qy=1672 cos(271=1.5) =-16TE OR Q = 157.9 52 1

11.91

GIVEN: X = 12(t-2)++2, 4=12-2(t-1)2 x,4-1t, t-5

FIND: (a) Nomin

(b) t, x, y, AND DIRECTION OF IS

WHEN U= WMIN

(a) HAVE X= 12(t-2) + + 2 4= 12 - 2(t-1)2 THEN .. 5x = \$x = \$(t-2) + 2t 5x = \$x = \$t2 - (t-1) = \ t^2 + t + 1 = \pm t^2 - t + 1 = \$(t+z)2 = 4(t-2)2 12= 12x2+ 12x2 = 1/2 [(++5)4+ (+-5)4]

NOTING THAT IS MINIMUM WHEN IS I'VE MINIMUM, HAVE .. dus = 4[(+2)3+(+-2)3]=0

EXPANSING .. (++6++12++8)+(+3-6++12+-8)=0 OR 2(t3+12t)=0

THE ONLY REAL ROOT OF THIS EQUATION IS \$ =0. : 12 10-5) + (10-5) + (0-5) = 5

OR 15min = 1.414 5 (b) WHEN W= WMIN t=0 X= 12(0-2)3+(0)2 DR. X=-0.667 ft. 4=12(0)3-2(0-1)2 カーキーのこのサ サーキ(0-2)と1世 AND Nx = \$ (0+2)2=1 \$ THEN TAND = 12 = 1 OR Pynn=45 &

11.92

GIVEN: X=4t-2 SINT, Y=4-2 COST X,4 - IN. , t-5 SKETCH: PATH OF THE PARTICLE FIND: (a) Nome AND NOWAX (b) t, x, y, AND DIRECTION OF IT XAMEL I ONA WIME . IL WAHW

HAVE .. X=At- Z SINT 4= 4-2 cost t, s x, m. 4, m. 4, m. 0.5 0 4.28 4.0 12.57 6.0 20.8 4.0 25.1

(a) HAVE .. X= 4t-2 SINT THEN .. Nx = 2 = 4 - 2 cost N = 2 = 2 SINT

Now. 52 = 5x2 + 5x2 = (4-2 cost)2+ (2 sint)2 = 20 - 16 cost

By observation. FOR ITMIN, cost = 1 so that Nmin = 4 OR Nomin = Z 5

FOR JMAX, cost :- 1 TAHT OZ

OR NAME = 6 5 15 MIAX = 36 (b) WHEN IS = WMIN : cost = 1

> OR t= 2NT S WHERE N= 0, 1,2, ...

THEN . X=4(2NT)-25IN(2NT) OR X=BNT IN. 4=4-261) OR 4 = 2 IN.

ALSO - 15x = 4-2(1)=2 13 54 = 2 SIN(2NT)=0

: Bu= 0 WHEN N= NMAX! cost = -1

OR 1=(2N+1)71 5.

(CONTINUED)

WHERE N=0, 1, 2, ...

THEN X=4(2N+1)71-25IN(2N+1)71 OR X=4(2N+1)71 IN 4 = 4 -2 (-1) OR 4=6 IN.

Jy = 2 SIN (2N+1) 17 = 0 ALSO. Nx = 4-2(-1)=6 3 : 0 = 0 -

11.93 GIVEN: [= A(cost + t sint)i +A(sint-tcost); FIND: (a) t SO THAT I AND Q ARE PERPENNICULAR (b) t so THAT I AND a ARE PARALLEL

HAVE. [= A(cost + t sint) i + A(sint - t cost) j 立= = A(-sint+sint+t cost) + A(cost-cost+tsint); = A (t cost) i + A(t sint) j Q = dy = A (cost - t sint) i + A (sint + tcost) j

(a) WHEN [AND & ARE PERPENDICULAR, [.Q.O.

:. A[(cost+t sint) + (sint-t cost) i] · Al (cost -t sint) j + (sint + t cost) j]=0 OR (cost+tsint)(cost-tsint)+(sint-tcost)(sint+tcost)=0 11.93 CONTINUED

OR (cost - t'smit) + (sint - t'cost) = 0 1-t2=0 OR t=15 OR

(b) WHEN I AND Q ARE PARALLEL, IXQ =0 :. A[(cost +t sint) i + (sint -t cost))]

* A[(cost-tsint) + (sint+tcost)]=0

or [(cost+tonit) sint+toot)-(sint-toot)(cost-tonic) Expansing. (sint cost +t +t2 sint cost)

- (sint cost - t + t2 sint cost) = 0

24=0 セ=0

GIVEN: [= X,(1-+1)]+(4, 2 2 cos 27t) 11.94 t-5; X,=30 mm, 4,= 20 mm 9/91 FIND: (a) [, I, AND Q AT LO (b) 1 ,5 , AND Q AT t=1.5 5 1.0 0.5 -0.5 -1.0

HAVE .. 5 = 30(1- ++1) + 20(= = cos 271t); THEN .. I = dE = 30 (+1) = 1 + 20 (- 70 = 200 s 2nt = 5 sins 2nt) = 30 (++1) = 1 - 201 [= 1/2 (2 cosant + 2 su 201)]

 $\bar{\sigma} = \frac{q_1}{q_2} = -30 \frac{(4+1)^2}{5} \bar{i} - soul_{-\frac{1}{2}} e^{\frac{2}{3}} (\frac{1}{5} cos suf + 5 sin suf)$ AND

+ LT(-TISINETT++4cosent)); = - (++1) = +107 2 = 107 2 = (++1) = -1.5 cos 27t) j

(a) AT 1=0: [=30(1-1)1+20(1)1

OR [= 20 mm] 5=30(+)j-2011((1)(+0))j

OR 15 = 43.4 15 7 46.3°

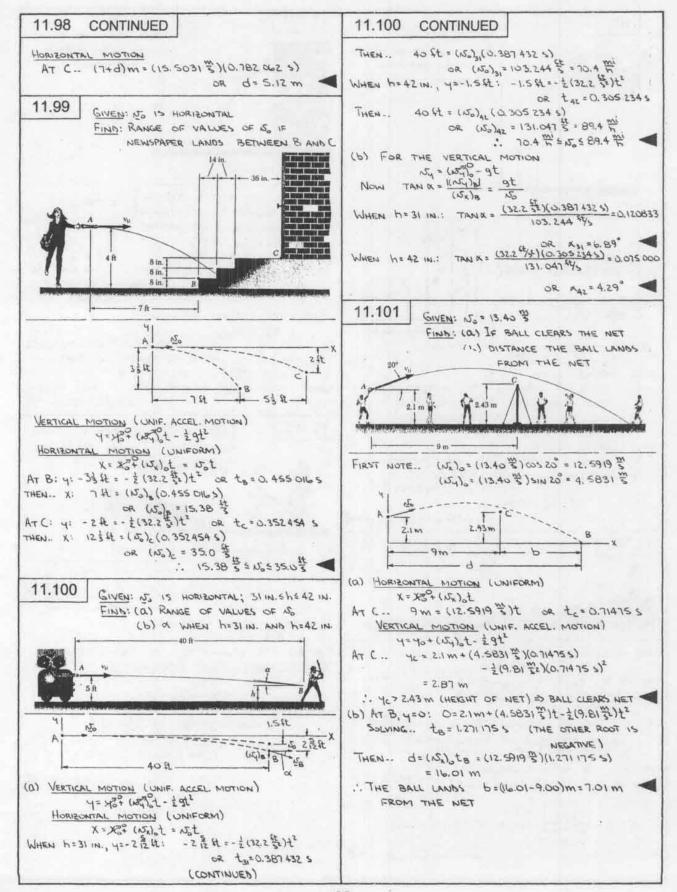
= (18 mm) + (-1.8956 mm))

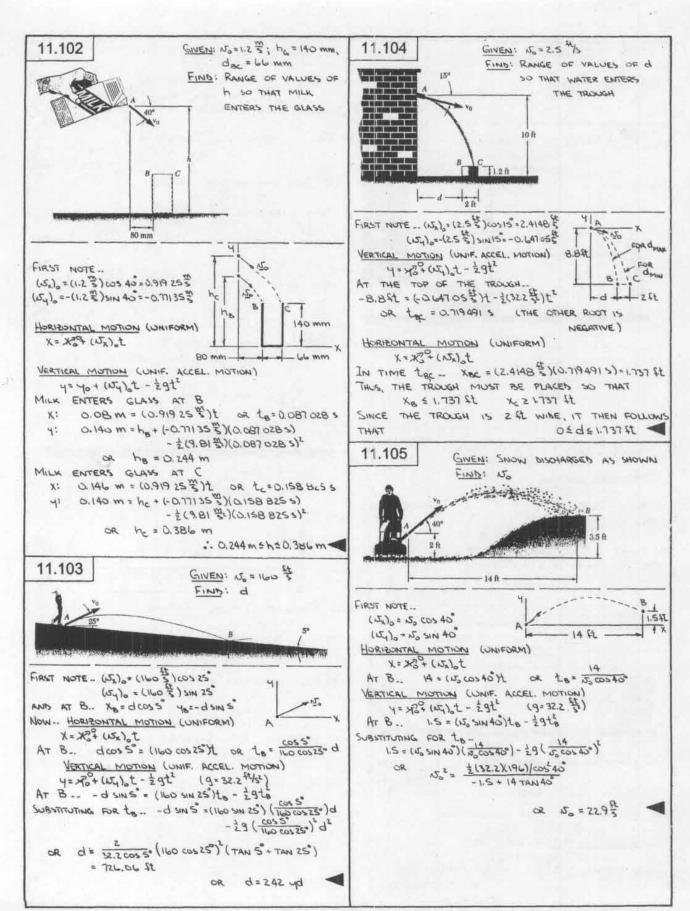
= (4.80 =)+ (2.9778 =);

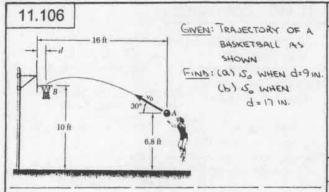
OR N= 5.65 5 1318 a = - (2.5) = 1 + 10 n = 0.75 (0 - 7.5 cos 37) = (-3.84 50) + (70.1582 50)

OR Q = 70.3 5 186.9

11.95 11.96 CONTINUED GIVEN: [= (Rt cos Wat) 1 + ct 1 + (Rt sin wat) k AND Q = -3(0) = +3(1) = +(2-0) k FIND: NE AND a Q= (0)2+ (3)2+ (2)2 = 13 OR . Q= 3.61 55 HAVE .. [= (Rtcosunt) + ctj + (Rt sinunt) & (b) IF I'AND I ARE PERPENDICULAR, I.V.O : [(3t cost); +(3(t+1)) + (t sint) k] THEN .. U = # = R(cos wat - wat sinuat) i + cj ·[3(cost-tout)i+(3(tim))+(sint+tcost)]=0 +R(sin whit + wat coswat) k OR (3t cost)[3(cost-t sint)]+(3 this)(3 (this) a = # = R (- was in out - ou sincut - out cos out) AND + (t sint) (sint + toost) = 0 + R(un cosunt + un cosunt-unt sin unt) E EXPANDING .- (9t cost - 9t sint cost) + (9t) = R (-zww since t - wet cos wat)i + (toin't+te sint cost)=0 OR (WITH t+0) 10+8cost-8t sint cost =0 + R(2Wn cosult - Wit SINWAT) E Now .. No + No + No + Not 7+2 cos 2t - 2t sin 2t = 0 = [R(cosunt-unt sinunt)]2+(c)2 USING TRIAL AND ERROR OR NUMERICAL METHODS, THE +[R(SINWAL+WALCOSWAL)]E ROUT IS SMALLEST In we so I was the same to so I'm NOTE: THE NEXT ROOT IS t= 4.38 S. + wit 2 sin wat) + (sin wat 11.97 1 (the fort smust cosunt + wet cosunt) GIVEN: No = 315 h; h= 80 m FIND : d = R2(1+ W2 t2) + C2 12= 185 (1+W, 75 HCz OR ALSO .. Q = Qx + Qx + Q2 = [R(-204 SHULT - WAT COSUNT)]2 + (0)2 +[R(zwn cosunt-wit sinunt)]2 = RY(4w2 sin wat+4w2 t since t coswat + wht2 cos wat)+ (4 wh cos wat FIRST NOTE .. No : 315 F -Aunt sinunt cosunt + wit 2 sin wint)] = 87.5 % = R2 (4W2+W4+2) VERTICAL MOTION OR a= RWW 14+Witz (UNIFORMLY ACCEL. MOTION) 80 m 4=43+(12762-29th 11.96 GIVEN: [= (At cost); AT B .. - 80m = - 2 (9.81 \$2) +2 OR to = 4.038 555 + (A 12+1) j+(Bt sint) k ~~ ft, t~s; A=3. HORIZONTAL MOTION (UNIFORM) K=X3+(Nx)+ B=1 SHOW: (7)2-(7)2-(8)2=1 AT B .. d = (87.5 %)(4.038 555) FIND: (a) IS AND a AT t=0 d = 353 m (b) tmin (t+0) so THAT 11.98 I AND I ARE GIVEN: 5 IS HORIZONTAL; PATH OF SHOWBALL PERPENDICULAR FIND: (a) Jo (b) d HAVE [= (At cost)] + (A(t2+1)] + (Bt sint) & X = At cost 4 = A VE+1 2 = Bt SINT fz = (5)2-1 THEN cost = At SINT = BE Now .. cost + sint = 1 => (At) + (Bt) = 1 $(\frac{1}{A})^{2}-1=(\frac{1}{A})^{2}+(\frac{1}{B})^{2}$ OR (A)2-(A)2-(B)2=1 Q.E.D. (a) WITH A=3 AND B=1, HAVE .. (a) VERTICAL MOTION 1 = # = 3(cost -t sint) 1 + 3 t 1 + (sint + t cost) 1 (UNIFORMLY ACCEL MOTION) 4= 484 (1578t - 29t2 AND Q = de = 3 (- sint - sint - t cost) = +3 (+2+1-t((2+1)) AT B .. - 1 m = - 2 (9. B1 52) L2 OR to = 0.451 524 5 + (cost + cost - t sint) te HORIZONTAL MOTION (UNIFORM) = -3(2514t+tcost) +3 (+2+1)3/2) x= x2+ (15x)-t AT B .. 7 m = No (0.451 524 5) + (2 cost - t sint) to No=15.503 OR No = 15,5031 3 AT t=0: 15 = 3(1-0) =+ (0) + (0) } (b) VERTICAL MOTION: AT C .. -3 m = - 2 (9.81 32) 12 OR 15= 3 15 OR tc= 27820625 (CONTINUES) (CONTINUED)







3.2 ST B

FIRST NOTE ..

11.107

(U) = No cos 30

(Jy) = J SIN 30 (16-d)ft HORIZONTAL MOTION (UNIFORM) X= X2+ (5,) t AT B .. (16-d) = (15 cos 30) t OR to = 15 cos 30 VERTICIAL MOTION (UNIF. ACCEL. MOTION) 4= x3+ (15,1) t - 29t2 (9= 32.2 52) AT B .. 3.2 = (No SIN 30") 1 8 - 29th SUBSTITUTING FOR to ... 16-d ... 29 (16-d ... 15 cos 30) = 29 (15 cos 30) 29(16-d)2 OR No = 3[(16-d)-3.2] 12 ~ b (a) d=9111.: 5= 2(32.2)(16-12)2 3[\$(16-3)-3.2] OR No = 29,8 \$ 2(32.2)(16-12)2 (P) G= 1) IN: 5[病(16-2)-32]

OR No = 29.6 5

GIVEN: TRAJECTORY OF A BALL AS SHOWN FIND: RANGE OF VALUES OF DO SO

THAT BALL GOES THROUGH THE

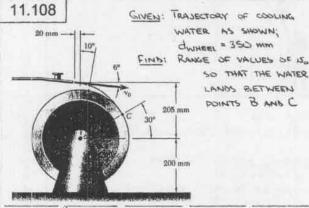
TIRE 3° VIII 1.5° 0.72 m 0.72° 0.72 m

FIRST NOTE: $(x_1)_{\circ} = x_{\circ} \cos 3^{\circ}$ ($(x_1)_{\circ} = x_{\circ} \cos 3^{\circ}$ ($(x_1)_{\circ} = x_{\circ} \sin 3^{\circ}$ ($(x_1)_{\circ} = x_{\circ} \cos 3^{\circ}$) $(x_1)_{\circ} = x_{\circ} \cos 3^{\circ}$ ($(x_1)_{\circ} = x_{\circ} \cos 3^{\circ}$)

When x = 6 m: $(6 = (x_1 \cos 3) + x_1 \cos x_1 \cos x_2 \cos x_2 \cos x_1 \cos x_2 \cos x$

11.107 CONTINUED

OR $50^2 = \frac{177.065}{0.314447 - 40.0}$ AT B, 4 = -0.53 m: $150^2 = \frac{177.065}{0.314447 - (-0.53)}$ OR $(150)_0 = 14.48 \frac{\text{m}}{5}$ AT C, 4 = -1.25 m: $150^2 = \frac{177.065}{0.314447 + (-1.25)}$ OR $(150)_0 = 10.64 \frac{\text{m}}{5}$ $10.64 \frac{\text{m}}{5} \leq 10.64 \frac{\text{m}}{5}$



205 min 10 1/8 C

FIRST NOTE .. $(U_{x})_{o} = U_{s} \cos 6$ $(U_{y})_{o} = U_{o} \sin 6$

HORIZONTAL MOTION (UNIFORM) $X = X_0 + (\sqrt{x}_0)_0 t$

VERTICAL MOTION (WHIF. ACCEL MOTION)

Y=Y=+(47)t-29t2 (9=9.813)

AT POINT B: X = (0.175 m) sin 10, Y = (0.175 m) cos 10

X: 0.175 sin 10 = -0.020 + (15 cos 6) t

OR to = 0.050 388

4: 0.175 cos10 = 0.205 + (-15, 514) * X = - 29th

Substituting for t_{e} . $-0.032 \text{ WS9} = (-N_0 \text{SING}^*)(\frac{0.050388}{J_0 \cos 6^*}) - \frac{1}{2}(9.81)(\frac{0.050388}{J_0 \cos 6^*})$ $DR \qquad N_0^2 = \frac{\frac{1}{2}(9.81)(0.050388)^2}{\cos^2 6^*(0.032 \text{WS9} - 0.050388 \text{TANG}^*)}$

OR (50)8 = 0.678 7

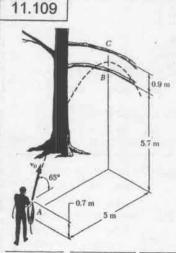
AT POINT (: X=(0.175 m) cos30, 4=(0.175 m) sin 30

X: 0.175 cos30=-0.020+(15, cos6) t

OR tc=0.171 554

4: 0.175 SIN 30 = 0.205 + (-15 SIN 6") $\frac{1}{15} - \frac{1}{2}91^{\frac{2}{5}}$ Substituting for $\frac{1}{15} - \frac{1}{5} - \frac{1}{5}91^{\frac{2}{5}}$ -0.117 SOD = $(-15 - \frac{1}{5} -$

:. 0.678 75 55 511 211 5 -



GIVEN: TRAJECTORY OF A ROPE AS SHOWN. FIND: RANGE OF VALUES OF US SO THAT THE ROPE GOES OVER DNLY THE LOWEST LINIB.

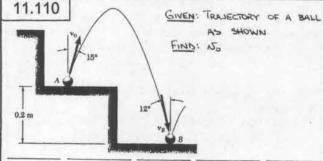
FIRST NOTE .. (N) = N (0365 (N) = N 51N 65 HORRONTAL MIOTION (UNIFORM) X= x= + (5x) t AT EITHER B OR C, X=5 m 5 = (No cos 65) 1 ec OR tax = (15 cos 65°)

VERTICAL MICTION (UNIF. ACCEL. MOTION) 4= x3+ (15/1)- 29t2 (9=9.81 mg) AT THE TREE LIMBS, t=tac YAC = (15 SIN 65") (5 cos 65") - 29 (15 cos 65")" (9.BI)(25)

5 TAN 65 - 48.C

5 - 686.566 5TAN 65 - 5 AT POINT B: OR (US) = 10.95 \$ 5= 686.566 OR (5) =11.93 5 AT POINT C:

: 10.95 5 5 5 5 11.93 5 -



FIRST NOTE .. (NX) = NS SIN IS (154) = 15 cos 15 HORIZONTAL MOTION (UNIFORM) NX = (NX) = NO SIN 15 (CONTINUED) 11.110 CONTINUED

VERTICAL MOTION (UNIF. ACCEL MOTION) 154=1549-9t 4=x1210+1==gt2 = N5 WS15 - 9t = (5, cos 15°)t - 2 9t2 AT POINT B, JULY THEN .. TAN 12° = (NX)8 = No SIN 15°

THEN .. TAN 12° = (NX)8 = 928 - NO COS 15° OR to = 150 (SIN 150 + COS 150) 9=9.81 Th = 0,222 59 50 Notine THAT Y8 = - 02 m, HAVE .. - 0.2 = (No cosis)(0.222 59NG) - 2 (9.81) (D. 222 595)2 OR 15=2.67 TM

11.111 GIVEN: NS = 280 \$ FIND: (a) X (b) h (c) to 340 ft

FIRST NOTE .. (NX) = NO SNK = (280 \$) SINK (Ny)= N3 COSK = (280 #) COSK

(a) HORIZONTAL MOTION (UNIFORM) X= X3+ (Jx) t = (280 SINK) t

AT POINT B: 340 = (280 SINK) +

OR TB = 14 SINX

VERTICAL MOTION (UNIF. ACCEL, MOTION) 4= x2+ (4) st - 29t2 + (280 cosa) t - 29t2 (9=32.2 52)

AT POINT B, tate, 420: 0= (280 COSK) (14 SINK) - 29 (17 SINK)2

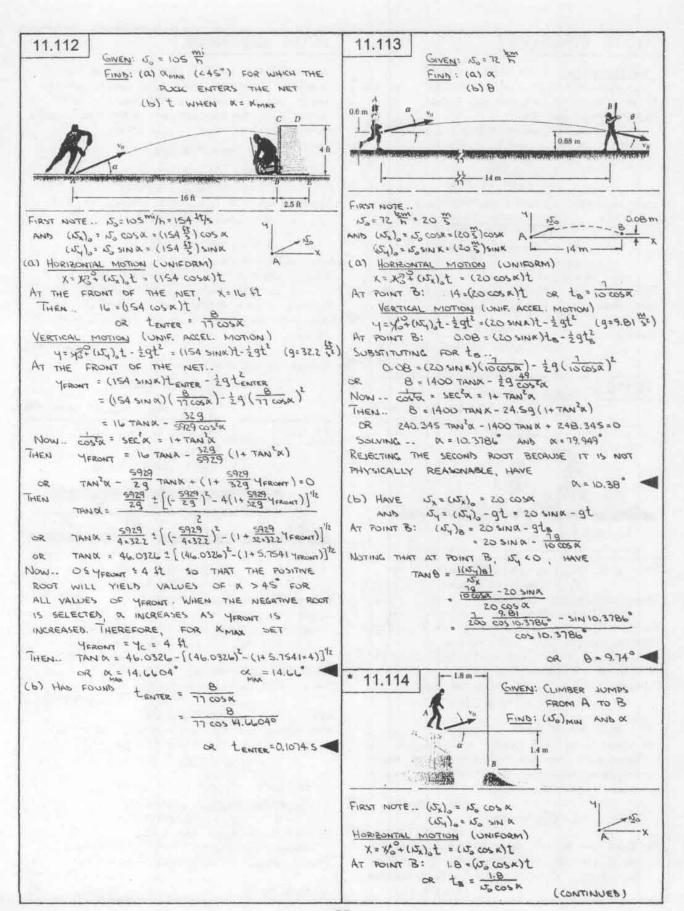
OR 280 SINK COSK - $\frac{1}{2}9(\frac{11}{140})$ OR SINZ N = $\frac{1}{2}(\frac{11}{140})(\frac{322}{140})$ OR N = 4,013 59°

N=4.01 (b) HAVE -- 154 = (154) - gt = 280 cosa - gt WHEN Y= YMAX = h, LY=0: 0=280 cos 4-9t OR th = 280 cos 4-013590 = 8.67433 S

THEN .. h = (280 cosx)th - 29th = (280 cos4.01359°)(8.67433)-2(32.2)(8.67433)2 OR h=1211 ft

(C) HAD FOUND .. to = 14 SING

14 SIN 4. 013 595 OR to =17.35 5



11.114 CONTINUED

VERTICAL MOTION (UNIF. ACCEL. MOTION) $y = y_0^{N+} (Dy_0)_0^{N+} - \frac{1}{2}gt^2 = (D_0 \sin n)t - \frac{1}{2}gt^2 = (g = 9.81\frac{n}{2})$ AT POINT B: $-1.4 = (D_0 \sin n)t - \frac{1}{2}gt^2$ SUBSTITUTING FOR tB... $-1.4 = (D_0 \sin n)(\frac{1.8}{150000}) - \frac{1}{2}g(\frac{1.8}{150000})^2$ OR $D_0^2 = \frac{1.62g}{1.62g}$ $0.9 \sin 2a + 1.4 \cos^2 a$

Now MINIMIZE J_{s}^{2} with RESPECT to α . Have... $\frac{dJ_{s}^{2}}{d\alpha} = 1.629 \frac{-(1.8 \cos 2\alpha - 2.8 \cos 3\alpha \sin \alpha)}{(0.9 \sin 2\alpha + 1.4 \cos^{2} x)^{2}} = 0$ or $1.8 \cos 2\alpha - 1.4 \sin 2\alpha = 0$

OR TAN 2K = 17 OR K = 26.0625° AND K = 206.06° REJECTING THE SECOND VALUE BECAUSE IT IS NOT PHYSICALLY POSSIBLE, HAVE...

FINALLY, 1502 = (1.62 + 9.81)

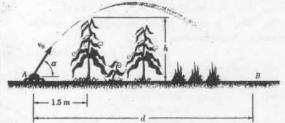
COS 26.0625 (1.8 TAN 26.0625 + 1.4)

OR (No)min = 2.945

11.115

GIVEN: No = 8 5

FIND: (a) dmax AND & WHEN H=0
(b) dmax AND & WHEN H=1.8 M



FIRST NOTE.. $(N_X)_0 = N_0 \cos \alpha = (8\frac{M}{2}) \cos \alpha$ $(N_X)_0 = N_0 \sin \alpha = (8\frac{M}{2}) \sin \alpha$ $(N_0)_0 = N_0 \sin \alpha = (8\frac{M}{2}) \sin \alpha$ HORIZONTAL MOTION (UNIFORM)

AT POINT B, X = d: $d = (8 \cos \alpha)t$ or $t_0 = \frac{d}{8 \cos \alpha}$ VERTICAL MOTION (UNIF. ACCEL. MOTION) $Y = y(8 + (N_0)_0 t - \frac{1}{2} yt^2 = (8 \sin \alpha)t - \frac{1}{2} yt^2$ ($g = 9.81\frac{M}{2}$)

AT POINT B: $O = (8 \sin \alpha)t_0 - \frac{1}{2} yt^2$ SIMPLIFYING AND SUBTITUTING FOR t_0 .

OR d = 9 SINZX (1)

(a) When h=0, THE WATER CAN FOLLOW MAY

PHYSICALLY POSSIBLE TRAJECTORY. IT THEN

FOLLOWS FROM EQ. (1) THAT d IS MAXIMUM

WHEN ZK=90

0 = BSINA - 29 (BCOSA)

THEN d= 64 SIN(2x45°)

OR CHMA = 6.52 m

(b) Based on Eq. (1) AND THE RESULTS OF PART

Q, IT CAN BE CONCLUDED THAT d INCREASES

IN VALUE AS & INCREASES IN VALUE FROM

(CONTINUES)

11.115 CONTINUED

O TO 45° AND THEN I DECREASES AS K IS
FURTHER INCREASED. THUS, I MAN OCCURS FOR THE
VALUE OF K CLOSELT TO 45° AND FOR WHICH THE
WATER JUST PASSES OVER THE FIRST ROW OF CORN
PLANTS, AT THIS ROW XCORN = 1.5 M

SO THAT town = BOOK

ALSO, WITH YCORN=h, HAVE $h = (8 \sin \alpha) t_{corn} - \frac{1}{2} 9 t_{corn}^2$ Substituting for teorn and nuting h=1.8 m, $1.8 = (8 \sin \kappa) (\frac{1.5}{8 \cos \kappa}) - \frac{1}{2} 9 (\frac{1.5}{8 \cos \alpha})^2$

Now.. $cos^2k = sec^2\alpha = 1 + tan^2\alpha$ Now.. $cos^2k = sec^2\alpha = 1 + tan^2\alpha$ Then $1.8 = 1.5 tank - \frac{2.25(9.81)}{12B}(1 + tan^2\kappa)$ or 0.172441 tan^2\alpha - 1.5 tan\alpha + 1.972441 = 0 Solving.. $\alpha = 58.229^\circ$ and $\alpha = 81.965^\circ$ From the above discussion, it follows that $d = d_{max}$ when

FINALLY, USING EQ (1)

d= 64 SN (2x 58.229°)

OR dmax = 5.84 m

11.116

GIVEN: No = 11.5 \$

FIND: (a) dmax

(b) a WHEN d=dMAX



FIRST NOTE .. (NI) = NO COSK = (11.5 %) COSK (NI) = NO SING = (11.5 %) SING

BY OBSERVATION, dMAX OCCURS

WHEN YMAX = 1.1 W.

WHEN Y = YMAX AT B, (NY) B = 0

THEN $(N_{1})_{8} = 0 = (11.5 \text{ SINK}) - 9t$ OR $t_{8} = \frac{11.5 \text{ SINK}}{9}$ AND $t_{8} = (11.5 \text{ SINK}) + \frac{9}{2} + \frac{1}{2} + \frac{1$

SUBSTITUTING FOR to AND NOTING 45=1.1 m. -1.1 = (11.5 SINA)(11.5 SINA) - 19(11.5 SINA)2

= = = (11.5) 2 SINZX

OR SIN'SK = 2.2 x 9.81 \(\alpha = 23.8265^\circ\)

(a) HORIZONTAL MOTION (UNIFORM) $X = X_0^2 + (D_X)^2 = (11.5 \cos x)^2$

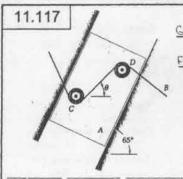
AT POINT B, X=dmax AND t=tB WHERE tB = \$\frac{11\S}{9.81}\$\sin 23.8265 = 0.47356 \$

THEN -- dmax = (11.5)(cos 23.8265)(0.473 56)

OR dmax = 4.98 m

(b) FROM ABOVE

X = Z3.8°



(P) TO MHEN 0= 450 (P) TO MHEN 0= 450 (P) TO MHEN 0= 450

HAVE .. $L_{CO} = L_{A} + L_{CO|A}$ WHERE $L_{A} = (0.5 \frac{10}{8})(-\cos 6)\frac{1}{2} - \sin 6 \frac{1}{2})\frac{1}{2}$ AND $L_{CO|A} = (2 \frac{10}{8})(\cos 6)\frac{1}{2} + \sin 6 \frac{1}{2})\frac{1}{2}$ HEN. $L_{CO} = [(-0.21131 + 2\cos 9)\frac{10}{8}]\frac{1}{2}$ $+ [(-0.45315 + 2\sin 9)\frac{10}{8}]\frac{1}{2}$ $+ (-0.45315 + 2\sin 45)\frac{1}{2}$ $+ (-0.45315 + 2\sin 45)\frac{1}{2}$ $+ (-0.45315 + 2\sin 45)\frac{1}{2}$ OR $L_{CO} = (-0.21131 + 2\cos 6)\frac{1}{2}$ OR $L_{CO} = (-0.21131 + 2\cos 6)\frac{1}{2}$ (b) HAVE... $L_{CO} = (-0.21131 + 2\cos 6)\frac{1}{2}$ $+ (-0.45315 + 2\sin 6)\frac{1}{2}$

= (0.788 49 3)i+ (1.278 90 3)j

11.119

GIVEN: NE = 9.8 KHOTS A 30° FIND: NE = 9.8 KHOTS A 30°

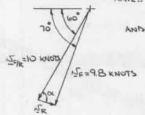
NOTE: F DENOTES THE FERRY AND "R"
DENOTES THE RIVER.

HAVE. UP = UR + UFIR OR UF = UFIR + UR

THE GRAPHICAL REPRESENTATION OF THE SECOND EQUATION

15 THEN AS SHOWN.

HAVE. UR = 9.82+10^2-2(9.8)(10)(00) 10



OR $L_{R} = 1.737 147 \text{ KNOTS}$ AND $\frac{9.8}{510.8} = \frac{1.737 147}{510.10^{\circ}}$ OR $K = 78.41^{\circ}$ OTS NOTING THAT

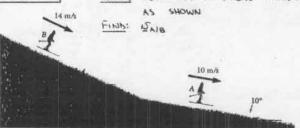


14.81 T STOWN TET. 13.41

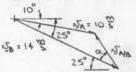
11.118

GIVEN: VELOCITIES OF SKIERS A AND B

NED = 1.503 T 1 58.3



HAVE .. LA = LB + LNB
THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS
THEN AS SHOWN.



THEN.. $J_{A|B}^2 = 10^2 + 14^2 - 2(10)(14)\cos 15^\circ$ OR $J_{A|B} = 5.05379 \%$ AND $\frac{10}{5100} = \frac{5.05379}{51015^\circ}$

OR X = 30,8

:. UNB : 5.05 \$ 1558°



GIVEN: NEW = 235 Th 75°

(6) 54 (6) 54 (6) 454

(C) ATUB FOR At= 15 MIN



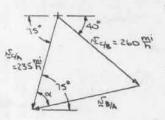
11.120

(a) HAVE. IS = IS + ISIA AND IS = IS + ISIB

DR 128 - 124 - 12618

NOW .. UB - UA - USWA SO THAT

ISUA = ISUA - ISUB OR ISUA = ISUB + ISUA THE GRAPHICAL REPRESENTATION OF THE LAST EQUATION IS THEN AS SHOWN.



HAVE...

NB/A = 2352 + 2602

-2(235)(26)(25565

OR NB/A = 266.798 MI

260 = 266.798 SIN & SIN 65° OR & = 62.03°

(b) HAVE.. LE = LE + LELA W) (- COSTS 1 - SINTS 1)

CR LE (24 W) - (235 W) (- COSTS 1 - SINTS 1)

(CONTINUES)



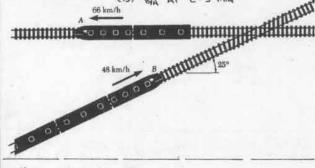
THEN. $\Delta \underline{\Gamma}_{clp} = (\underline{\Gamma}_{clp})t_2 - (\underline{\Gamma}_{clp})t_1 = \underline{M}_{clp}(t_2 - t_1)$ $= \underline{M}_{clp} \Delta t$ $= \underline{M}_{clp} \Delta t$

FOR At = 15 MIN: OFCIB = (260 m) (4 h) = 65 mi

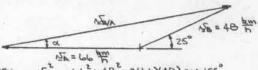
11.121

GIVEN: CONSTANT VELOCITIES OF TRAINS A MIS B; AT t=0, A is AT THE CROSSING; AT t=10 MIN, B is AT THE CROSSING

(b) For AT t=3 MIN



(a) HAVE.. ISB = ISA + ISBA
THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS
THEN AS SHOWN.



THEN -. Not = 662+482-2(66)(48) cos 1550 OR NOW = 111.366 km/h

 $\frac{48}{\sin \alpha} = \frac{111.366}{\sin 1550}$ OR $\alpha = 10.50$

: Non=111.4 Em 2 10.50

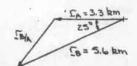
(b) FIRST NOTE THAT

AT 1=3 MIN, A 15 (66 1) (30) = 3.3 km WEST OF

THE CROSSING.
AT t=3 MIN, B 15 (48 m)(10) = 5.6 km

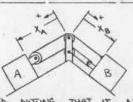
t = 3 min, B is $(48 \frac{\pi}{11})(60) = 5.6$ km southwest of the crossing.

NOW .. IS = IA + IBA THEN AT L= 3 MIN HAVE ..



ToyA : 3.32+5.62 -2(3.3)(5.6)co>25° OR ToyA = 2.96 km ■





FROM THE DIAGRAM... $2X_A + 3X_B = CONSTANT$ THEN.. $2X_A + 3X_B = O$ OR $|X_B| = \frac{3}{3}X_A$

NOW.. IN E = IN + INAL

AND NOTING THAT IN AND IN MUST BE PARALLEL

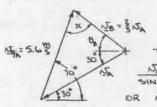
TO SURFACES A AND B, RESPECTIVELY, THE GRAPHKAL

REPRESENTATION OF THIS EDUCATION IS THEN

AS SHOWN NOTE: ASSUMING THAT IN IS DIRECTED

OP THE INCLINE LEADS TO A VELOCITY DIAGRAM

THAT DOES NOT CLOSE.



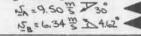
FIRST NOTE .. α = 180-(40+30+θ₆) = 110-θ₆

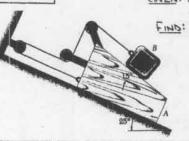
 $\frac{D_{A}}{\sin(110^{\circ}-\theta_{0})} = \frac{\frac{3}{3}D_{A}}{\sin(40^{\circ})} = \frac{5.6}{\sin(50^{\circ}+\theta_{0})}$ OR D_{A} $\sin(40^{\circ}) = \frac{3}{3}D_{A}$ $\sin(110^{\circ}-\theta_{0})$ OR $\sin(110^{\circ}-\theta_{0}) = 0.96418$

OR $\theta_8 = 35.3817^\circ$ AND $\theta_8 = 4.6183^\circ$ FOR $\theta_8 = 35.3817^\circ$: $N_8 = \frac{2}{3}N_A = \frac{5.651N.40^\circ}{51N.(30+35.3817^\circ)}$

OR NA = 5.94 \$ NB = 3.96 \$ 30

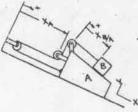
FOR $\theta_{B} = 4.6183^{\circ}$: $N_{B} = \frac{3}{3}N_{A} = \frac{5.16 \times 10.40^{\circ}}{510(30^{\circ} + 4.6183^{\circ})}$ OR $N_{A} = 9.50 \times 10^{\circ}$ $N_{B} = \frac{3.96 \times 10^{\circ}}{510(30^{\circ} + 4.6183^{\circ})}$ $N_{B} = \frac{3.96 \times 10^{\circ}}{510(30^{\circ} + 4.6183^{\circ})}$





11.123

EIND: (0) TO 520, CP, 520, CP,



FROM THE BIAGRAM... $2X_A + X_{B/A} = CONSTANT$ THEN... $2N_A + N_{B/A} = 0$ OR $|N_{B/A}| = |b| \stackrel{\text{IN}}{=}$ AND $2Q_A + Q_{B/A} = 0$ OR $|Q_{B/A}| = |2| \stackrel{\text{IN}}{=}$

(CONTINUED)

11.123 CONTINUED

Note that It and Que must be parallel to the top surface of block A.

(a) HAVE .. UB = UA + DEIA

THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS THEN AS SHOWN. NOTE THAT BECAUSE A IS MOVING DOWNWARD, B MUST BE MOVING UPWARD RELATIVE TO A.



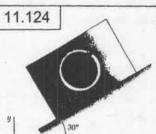
HAVE... $J_{a}^{2} = B^{2} + 16^{2} - 2(B)(16)(05)(15)$ OR $J_{a}^{2} = B.527B \frac{15}{5}$ AND B = B.527BSINK SIN 15°

OR $K = 14.05^{\circ}$ $J_{a}^{2} = B.53 \frac{15}{5} \frac{A}{5} 54.1^{\circ}$

(b) THE SAME TECHNIQUE THAT WAS USED TO DETERMINE QB. AN ALTERNATIVE METHOD IS AS FOLLOWS.

HAVE .. QB = QA + QB/A = (6 1) + 12 (- COS 15 1+ SIN 15 1) * = - (5,5911 5 1) + (3,1058 1) 5 541° OR QB = 6,40 1 541°

* NOTE THE ORIENTATION OF THE COORDINATE AXES ON THE SKETCH OF THE SYSTEM.

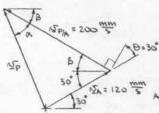


GIVEN: NFIA = 200 mm 3 NA = 120 5 1 30 FIND: (0) NF WHEN 8=30 (b) NF WHEN 8=135°

NOTE: RATHER THAN APPLY THE SAME METHOD OF SULUTION TWICE, TWO EQUALLY APPLICABLE TECHNIQUES WILL BE USED.

(a) METHON 1.

HAVE - UP = UA + UPA
THE GRAPHICAL REPRESENTATION OF THIS EMPATION
IS THEN AS SHOWN.



FIRST NOTE.. $\beta = 90^{\circ} - (30 + 30^{\circ}) = 30^{\circ}$ THEN.. $J_p^2 = |20^2 + 200^{\circ}$ $-2(|20)|(200)(25)(60^{\circ})$ OR $J_p = |74.35|(60^{\circ})$ SING = $\frac{174.35}{5}$ OR $K = 36.6^{\circ}$

(b) МЕТНОВ Z. 9=135° 15° 15°

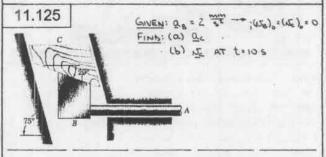
(CONTINUED)

11.124 CONTINUED

HAVE .. No = NO + NO | + NO | + 200 (- COSTS : - SINTS ;)

= (52.159 mm) : - (133.185 mm) \ (80.00)

OR NO = 143.0 mm \ (80.00)



(a) Have.. Qc = QB + Qc|B

The graphical representation of this equation is then as shown.

FIRST NOTE .. K = 180- (20+ 105°)

= 55°

15°

Qc/B

THEN- Qc

SIN 20° = 2 NMM

Qc = 0.835 06 NMM

Qc = 0.835 06 NMM

Cc = 0.835 06 NMM

AB = 2 NMM

Cc = 0.835 06 NMM

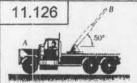
AB = 2 NMM

Cc = 0.835 06 NMM

Cc =

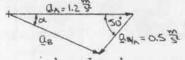
(b) FOR UNIFORMLY ACCELERATED MOTION ..

ルモ= (なり) + act AT t=10 s: ルモ= (0.83506 変型)(10 s) = 8.3506 変型 OR ルモ= 8.35 変型 入75° ◆



SIVEN: Qx = 1,2 m = ; (LB) = 0 Qen = 0.5 m y 50 FIND: (a) QB (b) LB AT t= 25

(a) Have.. QB = QA + QBA
THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS
THEN AS SHOWN.



HAVE .. Q = 1.22+ 0.5- 2(1.2)(0.5) cos 50°

OR QB = 0.958 46 \$4

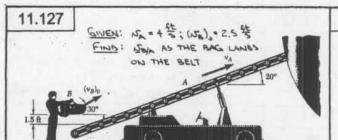
AND 0.5 = 0.95846

SINK SIN 50"

1. 28 = 0.958 \$ 7 23.5

(b) FOR UNIFORMLY ACCELERATED MOTION ...

AT t= 25: No = (0.95846 32)(25) = 1.91692 5 COR NO = 1.917 5 7 23.6



FIRST DETERMINE THE VELOCITY

OF THE BAG AS IT LANDS

ON THE BELT. Now...

[(1/2)_x]_0 = (1/2)_0 (0530 = (2.5 \frac{5}{5}) (0530 = \frac{5}{5}) \frac{5}{5} \frac{

* (2.5 cos 30") t = 2.5 cos 30"

VERTICAL MOTION (UNIF. ACCEL MOTION)

7=40+(156)4] t- 29t2 (156)4] - 9t =1.5+(2.5 51N30)t- 29t2 = 2.5 51N30 - 9t

THE EQUATION OF THE LINE COLUMEAR WITH THE TOP SURFACE OF THE BELT IS

THUS, WHEN THE BAG REACHES THE BELT..

1.5+(2.5 SIN 30)t-29t = [(2.5 COS30)t] TAN 20

OR 2(32.2)t+2.5(COS 30 TAN 20-SIN 30)t-1.5=0

OR 16.1t-0.46198t-1.5=0

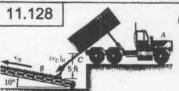
SOLVING -- t = 0.31992 s AND t =- 0.29122 s (REJECT)
THE VELOCITY US OF THE BAG AS IT LANDS ON
THE BELT IS THEN --

収象 * (2.5 cos30) j + (2.5 51N30-32.2(0.31992)] j = (2.16 51 生) j - (9.0514 生) j

FINALLY - UE = UZ + UE/A

OR UE/A = (2.1651 \(\) - 9.0514 \(\) - 4(cos 20 \(\) + 51N 20 \(\) \)

=-(1.593 67 \(\) \(\) - (10.4195 \(\) \(



GIVEN: (II) = 6 \$ 750° FIND: (a) IS IF IS/8 IS VERTICAL (b) IS IF IS/8 = (IS/8) MIN

FIRST DETERMINE THE VELOCITY OF THE COAL AS IT LANDS ON THE BELT. Now.. $\{(IS_c)_x\}_0 = (IS_c)_0 \cos 50^\circ = (IS_c)\cos 50^\circ$ $\{(IS_c)_y\}_0 = -(IS_c)_0 \sin 50^\circ = (IS_c)_0 \sin 50^\circ$ $\{IS_c)_x = \{(IS_c)_x\}_0 = -IS_c \cos 50^\circ$ = -3.8567

VERTICAL MOTION (UNIF. ACCEL. MOTION) $(152)_{1}^{2} = [(152)_{1}^{2} - 29(4-46) (9=32.2 \%)$ AT THE BELT: $(152)_{1}^{4} = (-6 \sin 50^{-2} - 2(32.2)(-5))$ OR $(152)_{1}^{4} = -18.5237 \%$

(CONTINUED)

11.128 CONTINUED

THEN UC =- (3.8567 \$) 1- (18.5257 \$) 1 = 18.9209 \$ 7 18.239

(a) HAVE.. IZ = IZB + IZOB

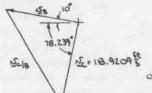
IF IZOB 15 TO VERTICAL, THEN (IZOB) = 0 WHICH

IMPLIES (IZO) = (IZO) = 3.8567

OR 15 = 3.92 \$ 100

(b) HAVE .. LE = LE + LEUB

THE GRAPHICAL REPRESENTATION OF THIS EGULATION IS
THEN AS SHOWN.



FOR JUB TO BE MINIMUM,
JUB MUST BE
PERPENDICULAR TO 158.

.: 158 = 18,9209 cos 88,239

OR NE = 0.581 \$ 10

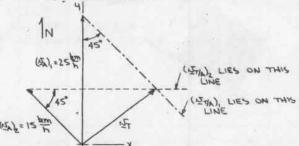
11.129

GIVEN: (ISA) = 25 km , (ITIM), \$\forall 450

(ISA) = 15 km \(ITIM), \$\forall 450

FIND: IST, WHERE IST IS CONSTANT

HAVE .. IST = ISA + ISTIA USING THIS EQUATION, THE TWO CASES ARE THEN GRAPHCALLY REPRESENTED AS SHOWN.



FROM THE BIAGRAM .. (UT) x = 25-15 SIN 45° = 14.3934 h

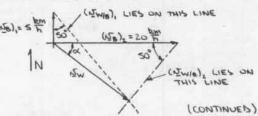
:. 15. 17.88 H 136.4° €

11.130



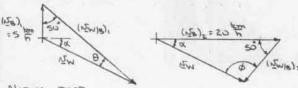
(THE) = 50 PM (THE) A 50,

HAVE.. IN = IS + INIB USING THIS EQUATION, THE TWO CASES ARE THEN GRAPHICALLY REPRESENTED AS SHOWN.



11.130 CONTINUED

WITH UN NOW DEFINED, THE ABOVE DIAGRAM IS REBRAWN FOR THE TIND CASES FOR CLARITY.



NOTING THAT 8 = 180- (50+90+K) = 40 - K

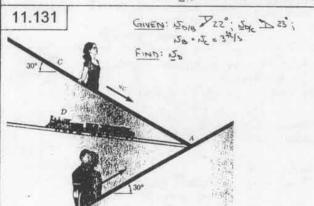
HAVE THEREFORE SIN (40°-X)

OR SIN 130 COSX - COS 130 SINA = 4(SIN 40 COSX - COS 40 SINA) TANA = SIN130 - 4 5140 cos 130 - 4 cos 40

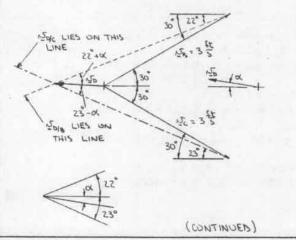
OR X = 25.964"

NW = 5 51N 50 THEN SIN(40-25.964") = 15.79 km

: 15W = 15.79 H 450.0.



HAVE .. No = No + NO 18 12 = VE + 120F THE GRAPHICAL REPRESENTATIONS OF THESE EQUATIONS ARE THEN AS SHOWN.



11.131 CONTINUED

THEN -51N7" = 3 SINBO = SIN(22°+ a) EQUATING THE EXPRESSIONS FOR

SINB (SIN 23 COSX - COS 23 SINK)

THEN
$$N_b = \frac{3 \sin 8^{\circ}}{\sin (22^{\circ} + 2.0728^{\circ})} = 1.024 \frac{5t}{5}$$

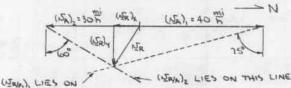
: 5=1.024 St = 2.07

11.132

GIVEN: (152) = 40 H N, (500), 175 (UTA) = 30 Th S, (UTA) , GO WITH THE VERTICAL

FIND: NR

HAVE .. UZ = (UZA), + (UZALA), UZA = (UZA)2 + (UZALA)2 THE GRAPHICAL REPRESENTATIONS OF THESE EDUCTIONS ARE THEN AS SHOWN. NOTE THAT THE LINE OF ACTION OF (LENA) MUST BE DIRECTED AS SHOWN SO THAT THE SECOND VELOCITY DIAGRAM CLOSES.



THIS LINE FROM THE DIAGRAM.. (NE), = [40+(NE)x] TAN IS" AND (UR)4 = (30 - (SR)x) TAN 30

EQUATING THE EXPRESSIONS FOR (NEX) --[40+(NR)x] TAN 15 = [30-(NA)x] TAN 30

OR (NR) = 7.8109 # THEN .. (NR)4 = (40+7, 8109) TAN 15 = 12, 8109 h : 15.00 H 758.6 .

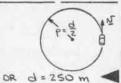
11.133 GIVEN: J= 18 5. P= 14 m FIND: an HAVE ... (18 E)

OR an = 23.1 1

11.134

GIVEN: CIRCULAR TRACK OF DIAMETER & FIND: (a) & WHEN 5=72 #, an=3.2 5 (b) IT WHEN d= 180 m, an= 0.69

(a) FIRST NOTE.. 5=72 = 20 = Now ... an = 152

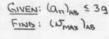


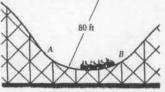
(b) HAVE an = 152

THEN - 152 = (0.6x9.81 =)(2x180m)

N= 82,9 km

11.135





HAVE _

THEN .. (LIMIN) AS = (3,32.2 5=)(80 ft) OR (JMAX) = 87,909 \$ OR (Noman) = 59.9 1

11.136



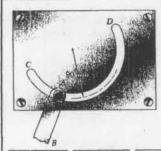
GIVEN: [(ac)n] = 26 50 [(ac)] = 267 5 B ROLLS ON A FIND: da

FIRST NOTE THAT ROLLING WITHOUT SLIPPING IMPLIES (No) = (No) = No

Now .. [(ac)n] = NE AND [(ac)n] = NE

WHERE PB = Z PA[(ac)n] = [(ac)n] (de) THEN -

SUBSTITUTING .. (2.6 IN.)(26 52) = (267 52)(de) OR do = 0.506 in. 11.137



GIVEN: (UL) = 0; FIND: (a) an AT too (b) an AT 1:25

(a) AT t=0, J=0 WHICH IMPLIES (QA) =0 1. an = (an)+

OR 04 = 20 52 (b) HAVE UNIFORMLY ACCELERATED MOTION ...

1. $\Delta x = (\Delta x_0^{70} + (\alpha_x)_{t}t)$ At t = 2s: $\Delta x = (20 \frac{mm/s}{s})(2s) = 40 \frac{mm}{s}$ Now. $(\alpha_x)_n = \frac{N x_0^2}{R} = \frac{(40 \frac{mm/s}{s})^2}{90 \frac{mm}{s}} = 17.7778 \frac{mm}{s}$

Qx = (Qx) = + (Qx) = .= (20)2 + (17.7778)2 FINALLY.

OR an = ZL. 8 ST

11.138

GIVEN: 0=250 mm; N= 45 5; a, = CONSTANT; AT 1=95, 5=0 FIND: + WHEN Q= 40 52

HAVE UNIFORMLY DECELERATED MOTION ... 1. 5 = 50+ at AT t=95: 0=45 \$ + Q_(95) OR at =- 5 3

Now .. a = a + a 2 WHEN 0=40 52 : 402 (-5)4 00 an = p an = 39.686 52

N= (39.686 32)(0.125m) THEN

FINALLY .. 2.227 = 45 3+(-5 3) t

OR t= 8,55 5

11.139

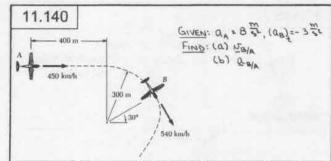
GIVEN: d = 420 A; Q = CONSTANT; 15 = 14 5, 52 = 24 st, 4512 = 95 ft FIND: a AT 1=25

HAVE UNIFORMLY ACCELERATED MOTION .. : 152 = 15,2 + 20x D512 SUBSTITUTING. (24 \$1)2 = (14 \$1)\$ 20, (95 ft)

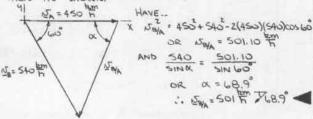


OR Q = 2 52 ALSO. $S = N_1 + Q_1 t_{\frac{1}{5}}$ AT t = 25: $S = 14 t_{\frac{1}{5}} + (2 t_{\frac{1}{5}})(2 t_{\frac{1}{5}}) = 18 t_{\frac{1}{5}}$ NOW... $Q_n = \frac{15}{6} \frac{(18 t_{\frac{1}{5}})^2}{210 t_{\frac{1}{5}}} = 1.54286 t_{\frac{1}{5}}^{\frac{1}{5}}$

 $a^2 = a_1^2 + a_n^2$ AT t= 25: Q2 = 22+ 1.54286



FIRST NOTE... $d_A = 450 \frac{km}{h}$ $d_B = 540 \frac{km}{h} = 150 \frac{m}{s}$ (a) Have... $d_B = d_A + d_{BA}$ The graphical representation of this equation is then as shown.



(b) First NOTE. $Q_{A} = \theta \frac{32}{5} \rightarrow (Q_{B})_{c} = 3\frac{52}{5} \Delta 60^{\circ}$ NOW. $(Q_{B})_{n} = \frac{\sqrt{5}e^{2}}{\beta B} = \frac{(150^{\circ}M)_{2})^{2}}{300^{\circ}M} OR (Q_{B})_{n} = 75\frac{32}{5} \sqrt{2}30^{\circ}$

THEN. QB = (QB) + (QB) = 3(-cos601+ SIN601)+75(-cos301- SIN301)
= -(66.452 7)1-(34.902 7)1

FINALLY. QB = QA + QB/A

OR QB/A = (-66.452 i - 34.902 i) - (8i)

= - (74.452 i) i - (34.902 ii)

OR QB/A = 82.2 ii > 7251°

11.141

GIVEN: Q STRAIGHT = Q_{L} = CONSTRUT; AT L=0, CAR ENTERS EXIT RAMP; FOR L>10 S, J= 20 $\frac{1}{12}$, Q= $\frac{1}{4}$ Q



FIRST NOTE. NO: 20 TH = 39 TE WHILE THE CAR IS ON THE STRAIGHT PORTION OF THE HIGHWAY

 $\begin{array}{c} \alpha = \alpha_{\text{STRAIGHT}} = \alpha_{\text{L}} \\ \text{AND} \quad \text{FOR} \quad \text{THE} \quad \text{CIRCULAR} \quad \text{EXIT} \quad \text{RAMP} \\ \alpha = \sqrt{\alpha_{\text{L}}^2 + \alpha_{\text{L}}^2} \\ \text{WHERE} \quad \alpha_{\text{R}} = \frac{\nabla^2}{\Omega} \end{array}.$

BY OBSERVATION, QMAX OCCURS WHEN IS IS MAXIMUM, WHICH IS AT \$\frac{1}{2} = 0 WHEN THE CAR FIRST ENTERS THE RAMP.

FOR UNIFORMLY BECELERATES MOTION

IS = 15 + 0, \$\frac{1}{2}\$

(CONTINUES)

11.141 CONTINUED

AND AT t 105: $D = CONSTANT \Rightarrow Q = Q_n = \frac{N_0^2}{P}$ $Q = \frac{1}{4}Q_{ST}$ THEN $Q_{ST} = Q_E$ $\frac{1}{4}Q_E = \frac{N_0^2}{P} = \frac{(Q_S + 1)^2}{S_{\mu\nu}(R)}$

OR $Q_{\xi} = -6.1460 \frac{\xi\xi}{5}$.

(THE CAR IS DECELERATING; HENCE, THE MINUS SIGN).

THEN AT $\xi = 0.195$:

OR $\xi = 90.193 \frac{\xi\xi}{5}$.

OR $\xi = 90.193 \frac{\xi\xi}{5}$.

THEN AT t=0: amm = \(a_t^2 + (\frac{15^2}{5^2})^2 \)
= \((-6.1460\frac{45}{5})^2 + \frac{(90.793\frac{45}{5})^2}{540\frac{45}{5}} \)

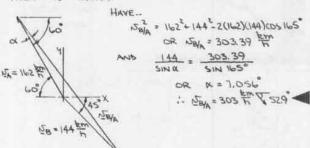
OR amm = 15.95\frac{45}{5}

11.142

| GIVEN: (Oalt=-7 54 (

FIRST NOTE.. NA = 162 TH = 45 TH US = 144 TH = 40 TH CO.) HAVE... NA = NA + NANA

THE GRAPHICAL REPRESENTATION OF THIS EQUATION IS THEN AS SHOWN.



(b) FIRST NOTE ... (QA) = 75 760 (QB) = 25 745 NOW ... Qn = 52 76 760 (QB) = 25 76 745 THEN ... (QA) = (45 8)2 (QB) = (40 8)2

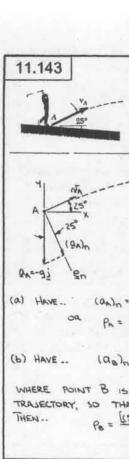
 $(\Omega_{A})_{n} = \frac{(70.5)^{n}}{300m}$ $(\Omega_{B})_{n} = \frac{(70.5)^{n}}{250m}$ or $(\Omega_{A})_{n} = 6.75 \frac{5}{5} \times 730^{\circ}$ $(\Omega_{B})_{n} = 6.40 \frac{5}{5} \times 4.45^{\circ}$

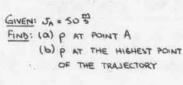
NOTING THAT $Q = Q_{k} + Q_{n}$ HAVE... $Q_{k} = 7(\cos 60\dot{1} - \sin 60\dot{1}) + 6.75(-\cos 30\dot{1} - \sin 30\dot{1})$

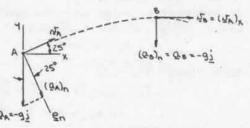
=-(2.3457 3)1-(9.4372 3)1 AND Q8=2(05451-51452)+6.40(05451+514452)

= (5.9397 \$\frac{1}{2})\dots + (3.1113 \frac{113}{52})\dots
FINALLY... QB = QA + QB/A
OR QB/A = (5.9397\dots + 3.1113\dots) - (-2.3457\dots - 9.4372\dots)

= (8.2854 \) i + (12.5485 \) i OR QB = 15.04 \) 15 56.6







(a) HAVE..
$$(a_{A})_{n} = \frac{\sqrt{x}}{P_{A}}$$
or $P_{A} = \frac{(50 \text{ m})^{2}}{(9.81 \text{ m})\cos 25}$

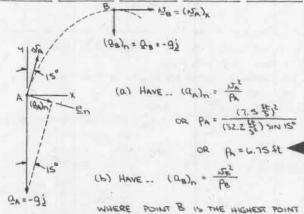
WHERE POINT B IS THE HIGHEST POINT OF THE TRAJECTORY, SO THAT $N_B = (N_A)_X = N_A \cos 25^\circ$ THEN... $P_B = \frac{[(50\%)\cos 25^\circ]^2}{9.91\%}$

DR PB = 209 m

11.145



GIVEN: NT = 7.5 \$ FIND: (a) P AT POINT A (b) P AT THE HIGHEST POINT OF THE TRAJECTORY

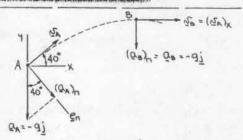


OF THE TRAJECTORY, SO THAT No = (NA) = NA SIN 15°
THEN -- (17.5 15) SIN 15')2

OR P6 = 0.1170 ft

11.144

GIVEN: A . B.Sm FIND: (a) NA (b) P AT THE HIGHEST POINT OF THE TRAJECTORY



(QA) = 15A (a) HAVE .. OR 152 = (9.81 cos 40)(8.5 m) = 63, 8766 m/s2

(0,0)n = 150 00 15x=7.99 € 140 € (b) HAVE ..

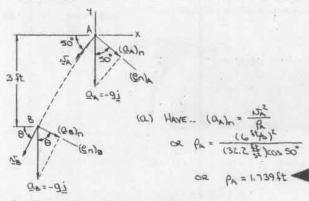
WHERE POINT B IS THE HIGHEST POINT OF THE TRAJECTORY, SO THAT UT = (UT) = UT COS40"

PB = (63,8766 32) cos240

OR Pa = 3.82 m



GIVEN: No 6 5 FIND: (a) P AT POINT A (b) P AT THE POINT ON THE TRAJECTORY 3 St BELOW A



(b) Horizontal motion (Uniform) $(J_B)_{\lambda} = (J_A)_{\lambda} = (L_A^{\frac{1}{2}})\cos 50 = 3.8567$ VERTICAL MOTION (UNIF. ACCEL. MOTION) HAVE.. Ny = (NA) + 29 (4-1/2) WHERE (NA) + (65) SIN 50 = 4.5963 5 AT POINT 8, 4=-3 H: (58)2 = (4.51)2 (52.2 \$)(-3 A) THEN .. No = (1/5) + (1/5) + (1/5) + (1/5) 850) + (1/4, 6399) = 15,1394 \$ AND TANG - 14 6399 OR 0- 75.24° (CONTINUED)

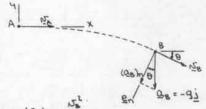
11.146 CONTINUED

(as) = 1500 Now .. (15.1394 5)2 P8 = (32.2 \$) cos 75.240

OR PB = 27.9 St

11,147





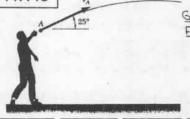
HAVE .. (as) = B

 $(a_B)_n = a_B \cos \theta = g \cos \theta$ WHERE NOTING THAT THE HORIZONTAL MUTION IS UNIFORM.

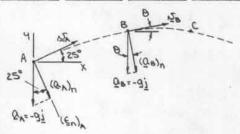
HAVE --(NB)x = NA (UB) x = NB COSB .. cos8 = 5A

THEN --

11.148



GIVEN: 15 = 20 5 FIND: IS AT THOSE POINTS WHERE P= 3PA



ASSUME THAT POINTS B AND C ARE THE POINTS OF INTEREST, WHERE YE = YC AND No = No. Now ...

P= 9 cos 25

Po = 4 Ph = 3 NAZ THEN

(CONTINUED)

11.148 CONTINUED

(ag) = 150 HAVE WHERE (a.) = 9 cos 0 \$ 9 cos 250 . SO THAT OR No = \$ (058 . NA

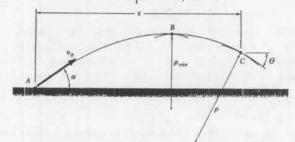
NOTING THAT THE HORIZONTAL MOTION IS UNIFORM, HAVE -(NA)x = (NB)x (NA)x = NA COS 25° (NB) x = NB COSB WHERE NT COS 25 = NT 6 COS B DR COS D = Us cos 25°

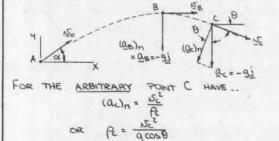
SUBSTITUTING FOR COST IN EQ. (1), HAVE. NB = \$ (NA cos 25°) NA JB = 3 JA = 3 (20 5) OR UB = NE = 18.17 3

11.149

GIVEN: THE INITIAL VELOCITY IS AND THE TRAJECTORY OF THE PROJECTILE AS SHOWN

SHOW: (a) PB = PMIN, WHERE YB = YMAX (b) Pc = Pmin/cos 3B





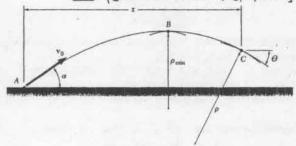
NOTING THAT THE HORIZONTAL MOTION IS UNIFORM, HAVE -(Na) = (Nc)x (NA) = NO COSA WHERE (NE)x = NE COSB THEN No cosA = NE cosA NE = COSK NO

SO THAT $\beta = \frac{1}{9\cos\theta} \left(\frac{\cos\theta}{\cos\theta} \, I_{\sigma}\right)^2 = \frac{9\cos^2\theta}{9\cos^2\theta}$ (a) IN THE EXPRESSION FOR PE. 15, K. AND 9 ARE CONSTANTS, SO THAT A IS MINIMUM WHERE COS B IS MAXIMUM. BY OBSERVATION, THIS OCCURS AT POINT B WHERE 0=0. 1. PMIN = PB = 150 COSTA

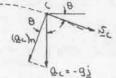
(P) be = co20 (2 co2x) Pc = PMIN Q.E.D. 11.150

GIVEN: THE INITIAL VELOCITY IS AND THE TRAJECTORY OF THE PROJECTILE AS

FIND: PE IN TERMS OF X, No, a, AND 9







HAVE.. $(Q_c)_n = \frac{Nc^2}{R}$ or $R = \frac{Nc^2}{Q_c Q_c Q_c}$

Noting that the horizontal motion is uniform, have $(N_A)_X = (N_C)_X$ $X = X_0^2 + (N_C)_X t = (N_C \cos \alpha)t$ where $(N_A)_X = N_C \cos \alpha$ $(N_C)_X = N_C \cos \alpha$ (1) Then $N_C \cos \alpha = t \cos \theta$ and $(N_C)_X = N_C \cos \alpha$ (1)

OR 600 = 150 000 K

SO THAT PE = 9150 000 K

FOR THE UNIFORMLY ACCELERATED VERTICAL MOTION HAVE

FROM ABOVE - $X = (U_0)_{y} - gt = U_0 \sin x - gt$ THEN... $(U_0)_{y} = U_0 \sin x - g \frac{x}{x^2 \cos x}$ (2)

NOW .. No = (NE) + (NE)

SUBSTITUTING FOR $(NE)_{\chi}$ [Eq.(1)] AND $(NE)_{\chi}$ [Eq.(2)] $K_{\xi}^{2} = (N_{0} \cos \kappa)^{2} + (N_{0} \sin \alpha - 9 \frac{\kappa}{N_{0} \cos \alpha})^{2}$ $E^{2} = (N_{0} \cos \kappa)^{2} + (N_{0} \sin \alpha - 9 \frac{\kappa}{N_{0} \cos \alpha})^{2}$

= Not (1 - 29x TANA + 92x2 Notes

or $N_c^3 = N_o^3 \left(1 - \frac{29 \times TANR}{N_o^2} + \frac{9^2 \times V}{N_o^4 \cos^2 N} \right)^{3/2}$

FINALLY, SUBSTITUTING INTO THE EXPRESSION FOR A, OBTAIN-

ρ = $\frac{g\cos\alpha}{3\cos\alpha} (1 - \frac{23χτανικ}{23χτανικ} + \frac{3χχ}{23γς} + \frac{3λχ}{23γς}$

* 11.151

GIVEN: C=(Rt cos wat) i+ctj+(Rt sin wat) & Find: P AT t=0

HAVE. $\underline{U} = \frac{d\underline{U}}{d\underline{U}} = R(\cos \omega_n t - \omega_n t \sin \omega_n t)\underline{i} + C\underline{i}$ $+ R(\sin \omega_n t + \omega_n t \cos \omega_n t)\underline{k}$ $+ R(\omega_n \cos \omega_n t + \omega_n \cos \omega_n t - \omega_n^2 t \cos \omega_n t)\underline{i}$ $+ R(\omega_n \cos \omega_n t + \omega_n \cos \omega_n t - \omega_n^2 t \sin \omega_n t)\underline{k}$

(CONTINUED)

11.151 CONTINUED

or Q = What consist = or the control of the control

Now.. $15^{2} = R^{2}(\cos \omega_{n}t - \omega_{n}t \sin \omega_{n}t)^{2} + C^{2}$ + $R^{2}(\sin \omega_{n}t + \omega_{n}t \cos \omega_{n}t)^{2}$

= R2(1+W)2+2)+C2

AND dt = (R2(1+C2t2)+C2)/6

= (\$\frac{AF}{2\pi_{2}} \right)_{5} + (\frac{B}{2} \right)_{5} \right)_{5}

AT t=0: 2 =0

 $\overline{\alpha} = \Omega^{M} S(S \overline{F}) \quad \text{os} \quad \alpha = S \Omega^{M} S$

THEN, WITH SE = 0, HAVE .. a = 6

OR P= R2+C2

* 11.152

GIVEN: [= (At cost) = (A 4+1) = + (8t swt) k, r-ft, t-s;

A=3, B=1 FIMB: P AT t=0

WITH A=3, B=1 HAVE ...

[- (3t cost) + (3 (t2+1) + (t sint) +

NOW - 1 = 9 = 3 (cost - t smt) = + (35)

Mb. Q = de = 3(-sint - sint - t cost) = 1 (ten - t (ten)) = +(cost + cost - t sint) =

=-3(2 sint + 2 cost) = +3 (+2+1)92 =

+ (zcost - t swt) E.

THEN ... It = 9(cost-t smt)2+9 + + (sint+tcost)2

Expanding and SIMPLIFYING TELDS.- $S^2 = t^4 + 19t^2 + 1 - 8(\cos^2 t + t^4 \sin^2 t) - 8(t^3 + t) \sin 2t$ Then $S = (t^4 + 19t^2 + 1 + 8(\cos^2 t + t^4 \sin^2 t) - 8(t^3 + t) \sin 2t)^{1/2}$

AND

4.5 4t3+38t+81-2costsint+4t3 sin2t+2t3 sint cost)-8(19t2+1)5m2ta2(122txost)

45 2[t4+19t2+1+8(cos2t+t4sin2t)-8(t3+t1)=112+t)=112+t)

2[t4+19t2+1+8(cos2t+t4sin2t)-8(t3+t1)=112+t)=112+t)

Now - a= at + an = (dt)2+ (p)2

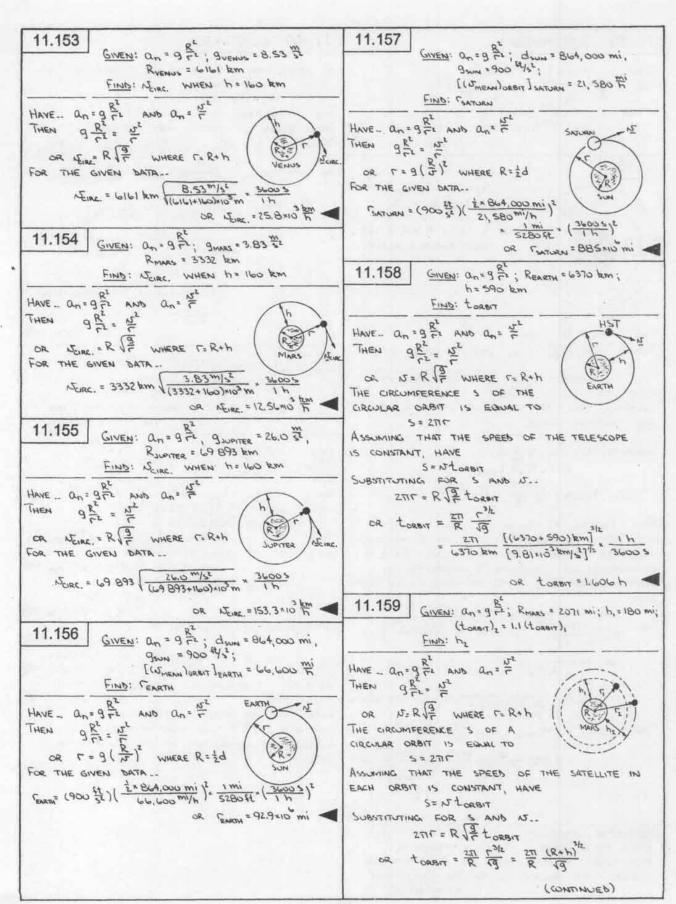
AT t=0: Q=31+2k or Q= 113 5

元=0

THEN, WITH \$# =0 HAVE .. a = P

OR P= 954

or p = 2.50 ft



11.159 CONTINUED

Now... $(torry)_2 = 1.1 (torry)_1$ or $\frac{27}{R} \frac{(R+h_2)^{3/2}}{\sqrt{q}} = 1.1 \frac{27}{R} \frac{(R+h_1)^{3/2}}{\sqrt{q}}$ or $h_2 = (1.1)^{2/3} (R+h_1) - R$ $= (1.1)^{2/3} (2071+180) mi - (2071 mi)$ or $h_3 = 328 mi$





GIVEN: an = 9 to ; ha = 120 mi,
ha = 200 mi; REARM = 3960 mi;
AT t=0, A AND B

ALIGNED AS SHOWN
FIND: t WHEN A AND B ARE
NEXT RADIALLY ALIGNED

HAVE.. $a_n = g \frac{R^2}{r^2}$ AND $a_n = \frac{\sigma^2}{r^2}$ THEN $g \frac{R^2}{r^2} = \frac{\sigma^2}{r^2}$ OR $\sigma = R \sqrt{\frac{g}{r^2}}$ WHERE r = R + h

The circumference s of a circular orbit is equal to $s=2\pi \Gamma$ Assuming that the speeds of the satellites are constant, have

SUBSTITUTING FOR S AND No.

or torbit =
$$\frac{2\pi}{R} \frac{r^{3k}}{\sqrt{a}} = \frac{2\pi}{R} \frac{(R+h)^{3k}}{\sqrt{a}}$$

NOW he > ha => (toreit) &> (toreit) a

NEXT LET TIME TYDIAL BE THE TIME AT WHICH THE

DATELLITES ARE NEXT RADIALLY ALIGNED. THEN, IF

IN TIME TYDIAL SATELLITE B COMPLETES N

ORBITS, SATELLITE A MUST COMPLETE (N+1) ORBITS.

THUS,

Thus,
$$t_{TOTAL} = N(t_{ORBIT})_B = (N+1)(t_{ORBIT})_A$$

$$N\left[\frac{2\Pi}{R}\frac{(R+h_B)^{3|z}}{(g^2)^3}\right] = (N+1)\left[\frac{2\Pi}{R}\frac{(R+h_B)^{3|z}}{\sqrt{g}}\right]$$

$$\Rightarrow R N = \frac{(R+h_B)^{3|z}}{(R+h_B)^{3|z}} = \frac{1}{(\frac{R+h_B}{R+h_A})^{5|z}-1}$$

$$= \frac{1}{\left(\frac{39LO+200}{396O+120}\right)^{3|z}} = 33.035 \text{ or Bits}$$

THEN - $t_{TOTAL} = N (t_{ORBIT})_B = N \frac{2\pi}{R} \frac{(R+h_B)^{3/2}}{\sqrt{9}}$ = 33.83S $\frac{2\pi}{3960 \text{ mi}} \frac{[(3960+200)\text{mi}]^{3/2}}{(32.2 \frac{f_L}{52.80 \text{ ft}})^{1/2}}$ × $\frac{1}{3600 \text{ ft}}$

OR THOTAL SIZH

ALTERNATIVE SOLUTION

FROM ABOVE HAVE (LOBBIT) > (LOBBIT) A
THUS, WHEN THE SATELLITES ARE NEXT RADIALLY
ALIGNED, THE ANGLES BA AND BB SWEPT OUT
(CONTINUED)

11.160 CONTINUED

BY RADIAL LINES DRAWN TO THE DATELLITES MUST DIFFER BY 271. THAT 15, $\theta_A = \theta_B + 2\pi$ FOR A CIRCULAR ORBIT.. $S = \Gamma \theta$ FROM ABOVE.. S = NST AND $NS = R\sqrt{\frac{n}{2}}$ THEN $\theta = \frac{s}{\Gamma} = \frac{NST}{\Gamma} = \frac{1}{\Gamma} \left(R\sqrt{\frac{n}{2}}\right) \frac{1}{\Gamma} - \frac{R\sqrt{n}}{\Gamma} \frac{1}{\Gamma} + \frac{R\sqrt{n}}{\Gamma} \frac{1}{\Gamma} \frac{1}{\Gamma}$

* 1 h SERO 5

OR tropal = 51.2 h

11.161

GIVEN: $\Gamma = 3(2 - Q^{\frac{1}{2}})$, $\theta = 4(1 + 2Q^{\frac{1}{2}})$ $\Gamma = M$, t = s, $\theta = RAS$ FIND: (Q) $L^{\frac{1}{2}}$ AND $L^{\frac{1}{2}}$ AT L = D(b) $L^{\frac{1}{2}}$ AND $L^{\frac{1}{2}}$ AS L = D; THE FINAL PATH OF THE PARTICLE

HAVE.. $r = 3(2 - e^{-t})$ $\theta = 4(t - 2e^{-t})$ THEN $r = 3e^{-t}$ $\theta = 4(1 - 2e^{-t})$ AND $r = -3e^{-t}$ $\theta = 8e^{-t}$ NOV... $v = re_r + r\theta e_\theta = 3e^{-t}e_{r} + 12(2 - e^{-t}) e_\theta$ AND $q = (r - r\theta^2)e_{r} + (r\theta - 2r\theta)e_{\theta}$ $= (r\theta^2 - 4\theta)e_{r} + (r\theta - 2r\theta)e_{r}$ $= (r\theta^2 - 4\theta)e_{r} + (r\theta^2 - 2r\theta)e_{r}$

(a) At t=0: 15 = 3 gr + 12(2-1)(1-2) gb

OR 5 = (3 \frac{m}{2}) gr - (12 \frac{m}{2}) gb =

Q = [-3-48(2-1)(1-2)^2] gr

+ [24(2-1)+24(1-2)] gb

OR 0 = -(51 \frac{m}{2}) gr

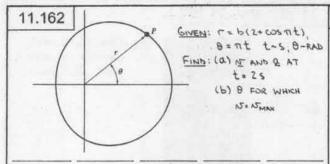
(b) As t-oo: N= (0)er+12(2-0)(1-0)eg

OR N=(24 = 1)eg

Q= (0+0)(1-0)^2]er+(0+0)eg

OR Q=-(96=1)er+

As \$\pmo_, \(\Gamma_\) \component. A constant. Thus, the final path is a circle of rabius 6 m. The speed of the particle is constant (24 \frac{\pi}{2}); thus, the transverse (tangential) component of the acceleration is zero.

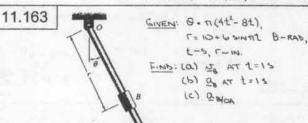


(= b (2+ cos Tt) B. nt HAVE .. Truizen -= i n=8 THEN i = -n2 b cosnt 8.0 AND Now. v= re-+ree =- (upsinut)e+up(s+cosut)e Q=(=-102)e++(+0+2+0)e0 = [-17 b cos nt - n2 b (2+cos nt)]eas(Inuice TTS - 0)+ =-272b[(1+cosnt)e+(sinntle)]

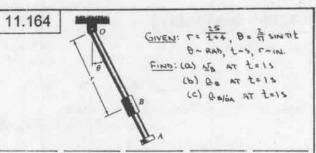
(a) AT t=25: 15=-(0)e+ Tb(2+1)e8 OR W = 377 BEB a=-2712 b[(1+1)er+(0)e)] OR Q = -412ber

(b) HAVE. N= TID ((-SINTE)2+(2+COSTE)2 = 716 (5+4 cos 11 t = TIB (5+4005B

BY OBSERVATION, No. JAMES WHEN COSO = 1 OR 0= ZNA, N=0,1,Z,...



B=71(42-81) HAVE .. F = 10+6 SINTIL i = un cosnt 0 = BT (t-1) THEN r = - 67 SINT 118 + 0 AT L=1 5: T=10 IN. 8 = - 4TT RAS 0 = 0 B= BT RADIE F = 0 (a) HAVE. I = FEr + FOEB アニー(アルア)ら SO THAT (b) HAVE .. Q= ("- rot)er + (ro+2+0)e0 = (10)(811)20 or a = (800 5)es 0=-4TI RAD (C) HAVE .. aBlog = " SO THAT ab/04 = 0



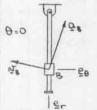
D= FSINAt HAVE .. THEN AND

B = 2 cosnt ITMIZETS = 0

AT t=15: (=5 m. 8=0 0 =- 2 eno/s F=-1 14/5 F = 0.4 IN. 152 8 =0

(a) HAVE .. Us= rer+ + PeB + (-1)e+ (5)+2)eB OR 15=-(15)e--(105)e

(b) HAVE .. Q= ("-+ 62) = + (++2+6) =0 = (0.4-(5)(2)2)e+(0+2(-1)(2))ea OR Q=-(19.63)e-+(41/2)ep



(C) HAVE abox = T SO THAT QB/OA = (04 TE)E

11.165 GIVEN: 7 = 2-WATE, DEAT r-m, t-s, 0-RAD FIND: (a) IT AND Q AT (b) IS AND Q AT t. 0.55

C= 2-cosnt HAVE ... D=nt i = - 27 51 m t THEN

= -27 Tcosnt(2-cosnt)-smnt(2nsmnt) AND (2- cos 11)3

1 - 2 - 2 - 2 - 1 - 1 - 5 - 2 - 2 - 2 - 2 - 2 - 2 (z-cos 11t)3

(a) AT t=0: F=2 m B=0 B=T RAB r=0 =- 2772 mg 8=0

Now .. 5 = ig + raga = (2)(17) ea OR 15=(211 5) CB Q = ("-r\0)er + (r\0+2\0)e0

AND .. = [-2112-(2)(11)2]e-

OR a = - (477 52)er 0=0 (CONTINUED)

11.165 CONTINUED

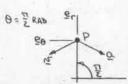
(b) At t=0.55: r=1m $\theta=\frac{\pi}{2}$ RAD $\dot{r}=-\frac{2\pi}{2}\frac{M}{2}$ $\dot{\theta}\approx\pi$ $\frac{\pi}{2}$

 $\frac{1}{11} = -3U_S \frac{(3/2)}{-1-1} = \frac{5}{U_S} \frac{3}{M}$ $\frac{6}{9} = 0$

OR 12 = - (2 m) 6 + (11 m) 50

AND.. $\bar{Q} = (\ddot{r} - r\dot{\theta}^z)\bar{e}_r + (r\ddot{\theta} + 2\dot{r}\dot{\theta})\bar{e}_\theta$ $= [\frac{\pi^z}{2} - (1)(\pi)^z]\bar{e}_r + [2(-\frac{z}{2})(\pi)]\bar{e}_\theta$

OR Q =- (12 32) 2- - (n2 12) 20



11.166

GIVEN: (= 20 COSO, 0= 2 bt2

(b) p; PATH OF THE PARTICLE

(a) Have.. $\Gamma = 2a \cos \theta$ $\theta = \frac{1}{2}bt^2$ Then $\dot{\Gamma} = -2a \dot{\theta} \sin \theta + \dot{\theta}^2 \cos \theta$ $\dot{\theta} = bt$ And $\dot{\Gamma} = -2a (\dot{\theta} \sin \theta + \dot{\theta}^2 \cos \theta)$ $\dot{\theta} = b$ Substituting for $\dot{\theta}$ And $\dot{\dot{\theta}}$

i = - 2abt sin 0 i = - 2ab(sin 0 + bt2 cos 0)

Now. $5r = \dot{r}$ $9 = r\dot{\theta}$ 9 = 70 9 = 20 sabt 6 = 8

THEN .. IS = \(\sigma_{\text{t}}^2 + \sigma_{\text{b}}^2 = 2abt[(-\sin \theta)^2 + (\cos \theta)^2]^{1/2}

ALSO... ar= "-rb2 = - 2ab(sIND+bt2cos0)-2ab2t2cos0

=- $2ab(sin\theta+zbt^2cos\theta)$ AND $a_\theta=r\theta+zr\theta=2abcos\theta-4ab^2t^2sin\theta$

 $= 2ab(cos\theta - 2abcos\theta - 4abt$ $= 2ab(cos\theta - 2abt^2 sin\theta)$

THEN. $a = \sqrt{\alpha_1^2 + \alpha_0^2}$ = $2\alpha b [(\sin\theta + 2bt^2\cos\theta)^2 + (\cos\theta - 2bt^2\sin\theta)^2]^{1/2}$

(b) Now. $a^2 = a_t^2 + a_n^2 = (\frac{ds}{dt})^2 + (\frac{s^2}{P})^2$

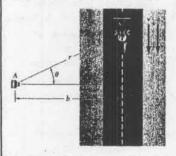
THEN -. df = df (2abt) = 2abSO THAT $(2ab (1+4b^2t^4)^2 = (2ab)^2 + a^2 +$

INALLY.. $a_n = \frac{\sqrt{2}}{\rho} \Rightarrow \rho = \frac{(2abt)^2}{4ab^2}$

SINCE THE RADIUS OF CURVATURE IS A CONSTANT, THE PATH IS A CIRCLE OF RADIUS Q.

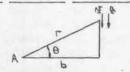


11.167 and 11.168



GIVEN: THE RECTILINEAR MOTION OF A RACE CAR AS SHOWN

HAVE.. $\Gamma = \frac{b}{\cos \theta}$ THEN $\Gamma = \frac{b\theta \sin \theta}{\cos^2 \theta}$

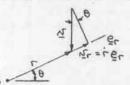


1.167 FIND: N IN TERMS OF b. D. AND B

HAVE.. $D^2 = N_C^2 + N_B^2 = (\dot{r})^2 + (\dot{r}\dot{\theta})^2$ $= \left(\frac{b\dot{\theta}\sin\theta}{\cos^2\theta}\right)^2 + \left(\frac{b\dot{\theta}}{\cos\theta}\right)^2$ $= \frac{b^2\dot{\theta}^2}{\cos^2\theta} \left(\frac{\sin^2\theta}{\cos^2\theta} + 1\right) = \frac{b^2\dot{\theta}^2}{\cos^2\theta}$ OR $N = \pm \frac{b\dot{\theta}}{\cos^2\theta}$

FOR THE POSITION OF THE CAR SHOWN, θ is decreasing; Thus, the negative root is chosen. $\frac{\dot{\theta}}{\dot{\theta}}$

ALTERNATIVE SOLUTION



FROM THE DIAGRAM.. $\dot{\Gamma} = -N \sin \theta$ OR $\frac{\dot{b}\dot{\theta}\sin \theta}{\cos^2 \theta} = -D \sin \theta$ OR $D = \frac{\dot{b}\dot{\theta}}{\cos^2 \theta} = \frac{\dot{\theta}\dot{\theta}}{\cos^2 \theta} = \frac{\dot{b}\dot{\theta}}{\cos^2 \theta} = \frac{\dot{b}\dot{\theta}}{\cos^2 \theta} = \frac{\dot{\theta}\dot{\theta}}{\cos^2 \theta} = \frac{\dot{\theta}\dot{\theta}}{\cos^2 \theta} = \frac{\dot{\theta}\dot{\theta}}{\cos^2 \theta} = \frac{\dot{\theta$

11.168 FIND: Q IN TERMS OF D, B, B, AND B

FOR RECTILINEAR MOTION Q = AT FROM THE SOLUTION TO PROBLEM 11.167

THEN $\alpha = \frac{\partial^2 \cos^2 \theta}{\partial \theta} - \frac{\partial^2 \cos^2 \theta}{\partial \theta} - \frac{\partial^2 \cos^2 \theta}{\partial \theta} = 0$

OR Q=- b (H+20 TAND)

ALTERNATIVE SOLUTION $\frac{b}{cos\theta}$ $\dot{r} = \frac{b\dot{\theta} \sin \theta}{\cos \theta}$

THEN. $\ddot{r} = b \frac{(\ddot{\theta} \sin \theta + \dot{\theta} \cos \theta)(\cos^2 \theta) - (\dot{\theta} \sin \theta)(-2\dot{\theta} \cos \theta \sin \theta)}{\cos^2 \theta}$

 $= p \left[\frac{\cos_2 \theta}{\cos_2 \theta} + \frac{\cos_2 \theta}{\cos_2 \theta} \right]$

Now. $\alpha^{2} = \alpha_{r}^{2} + \alpha_{\theta}^{2}$ $where \quad \alpha_{r} = \ddot{r} - r\dot{\theta}^{2} = b \left[\frac{\ddot{\theta} \sin \theta}{\cos^{2}\theta} + \frac{\dot{\theta}^{2} (1 + \sin^{2}\theta)}{\cos^{2}\theta} \right] - \frac{b\dot{\theta}^{2}}{\cos\theta}$ $= \frac{.b}{\cos^{2}\theta} \left(\ddot{\theta} \sin \theta + \frac{2\dot{\theta}^{2} \sin^{2}\theta}{\cos\theta} \right)$ (continues)

11.168 CONTINUED

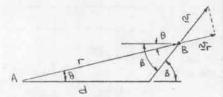
$$\begin{aligned} & \alpha_r = \frac{b \sin \theta}{\cos^2 \theta} \left(\ddot{\theta} + 2 \dot{\theta}^2 \tan \theta \right) \\ & \text{and} \quad & \Omega_\theta = r \ddot{\theta} + 2 \dot{r} \dot{\theta} = \frac{b \ddot{\theta}}{\cos^2 \theta} + 2 \frac{b \dot{\theta}^2 \sin \theta}{\cos^2 \theta} \\ & = \frac{b \cos^2 \theta}{\cos^2 \theta} \left(\ddot{\theta} + 2 \dot{\theta}^2 \tan \theta \right) \\ & \ddot{l} \text{HEN} \quad & \Omega = \frac{1}{2} \frac{b}{\cos^2 \theta} \left(\ddot{\theta} + 2 \dot{\theta}^2 \tan \theta \right) \left[(\sin \theta)^2 + (\cos \theta)^2 \right]^{1/2} \end{aligned}$$

FOR THE POSITION OF THE CIAR SHOWN, θ is NEGATIVE; FOR α to be positive, the negative root is chosen. $\alpha = -\frac{b}{\cos 2\theta} (\ddot{\theta} + 2\dot{\theta}^2 \tan \theta)$

11.169

GIVEN: STRAIGHT LINE TRAJECTORY OF THE HELICOPTER SHOWN
FIND: N IN TERMS OF d, b, B, AND B





FROM THE DIAGRAM ..

THEN $f = d \tan \beta \frac{-(\tan \beta \sin \theta - \cos \theta)}{(\tan \beta \cos \theta - \sin \theta)^2} \theta$ $= d \theta \tan \beta \frac{-(\tan \beta \cos \theta - \sin \theta)^2}{(\tan \beta \cos \theta - \sin \theta)^2}$

FROM THE DIAGRAM

THEN

 $d\theta \tanh \beta \frac{\tanh \theta \sinh \theta + \cos \theta}{(\tanh \theta \cosh \theta - \sin \theta)^2} = 15(\cosh \cosh \theta + \sin \theta + \sin \theta)$ $= 15(\cosh \theta + \sin \theta + \cos \theta)$

OR IS = do TANB SEC B (TANB COOR - SIND)?

ALTERNATIVE SOLUTION

HAVE .. N2 = NF2 + NB2 = (1)2 + (18)2

USING THE EXPRESSIONS FOR T AND ' FROM ABOVE ...

NS = [dBTANB TANBSINB+COSB]2

+ (di TANBCOSO-SINO)2

(CONTINUED)

11.169 CONTINUED

OR
$$N = \pm \frac{d\theta \tan \theta}{(\tan \theta \cos \theta - \sin \theta)^2} \left[\frac{(\tan \theta \sin \theta + \cos \theta)^2}{(\tan \theta \cos \theta - \sin \theta)^2} + 1 \right]^{1/2}$$

$$= \pm \frac{d\theta \tan \theta}{(\tan \theta \cos \theta - \sin \theta)^2} \left[\frac{\tan \theta \cos \theta - \sin \theta}{(\tan \theta \cos \theta - \sin \theta)^2} \right]^{1/2}$$

NOTE THAT AS 8 INCREASES, THE HELICOPTER MOVES IN THE INDICATED DIRECTION. THUS, THE POSITIVE ROOT IS CHOSEN.

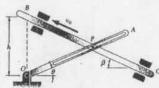
.. N= do TANB SECB

* 11.170

GIVEN: No = CONSTANT

FIND: B IN TERMS OF

No, h, B, AND B



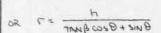
FROM THE DIAGRAM...

T

SIN(30-B) = SIN(4+B)

OR r(SINBCOSB+COSBSINB)

= hcosB



ALSO -- $N_{\theta} = N_{\phi} \sin(\beta + \theta)$ WHERE $N_{\phi} = N_{\phi} \left(\sin \beta \cos \theta + \cos \beta \sin \theta \right)$

OR $\dot{\theta} = \frac{155\cos\beta}{h} (\pi M \beta \cos\theta + \sin\theta)^{\frac{1}{2}}$

ALTERNATIVE SOLUTION

FROM ABOVE ... T= TANBOOSB+SINB

THEN is + THE GIVE - COSO : 0

Now. 12 = 12 + 12 = (+)+ (-0)+

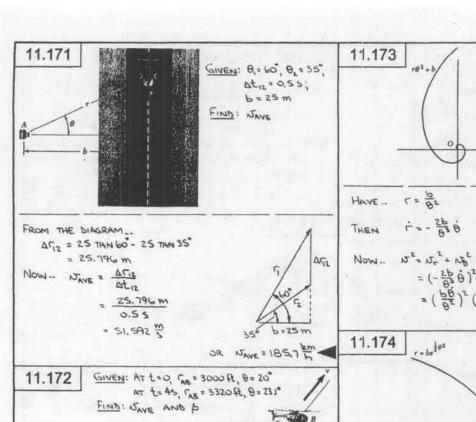
OR $N_0^2 = \left[H \dot{\theta} \frac{\text{TANBEIND} - \cos \theta}{(\text{TANB } \cos \theta + \sin \theta)^2} \right]^2 + \left(\frac{H \dot{\theta}}{\text{TANB } \cos \theta + \sin \theta} \right)^2$

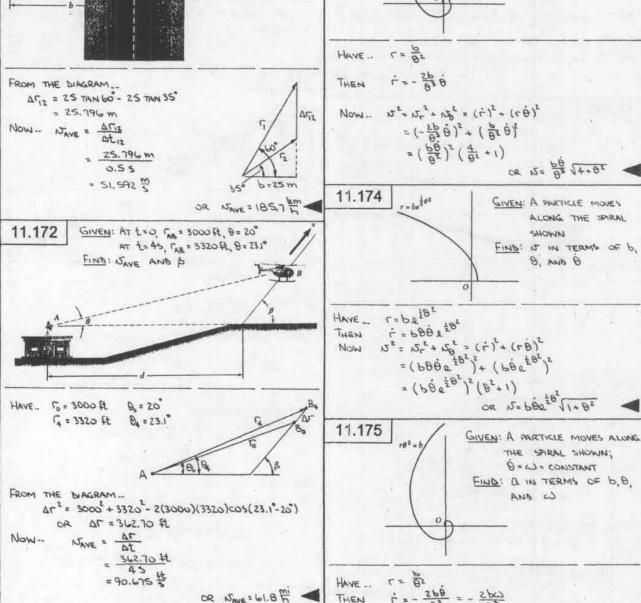
OR No = + TRUBCOSON SIMO [TRUB COSON + 1] 1/2

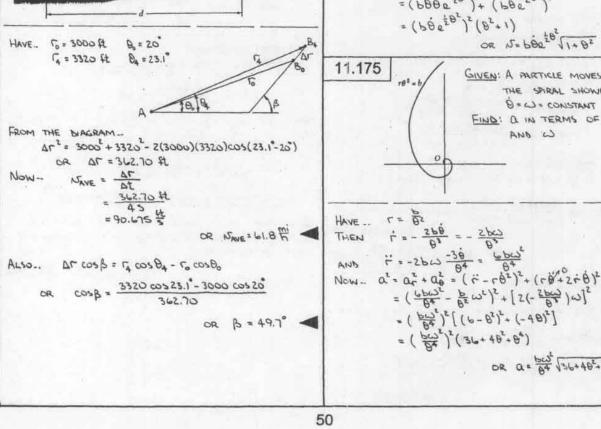
 $= \pm \frac{h\dot{\theta}}{TANBCOS\theta + SIN\theta} \left[\frac{TANBCOS\theta + IN\theta}{(TANBCOS\theta + SIN\theta)^2} \right]^{\frac{N}{2}}$

NOTE THAT AS O INCREASES, MEMBER BC MOVES IN THE INDICATED DIRECTION. THUS, THE POSITIVE ROOT IS CHOSEN.

: $\dot{\theta} = \frac{N_0 \cos \theta}{h} (\pi m \beta \cos \theta + \sin \theta)^2$





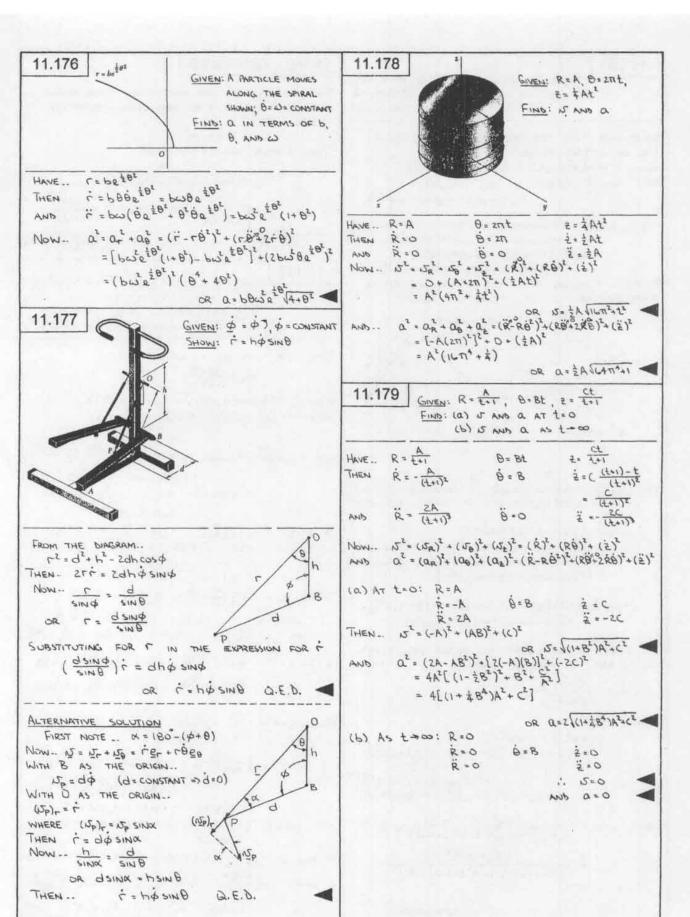


GIVEN: A PARTICLE MOVES

FIND: IT IN TERMS OF b. O. AND B

SHOWN

ALONG THE SPIRAL



11.180

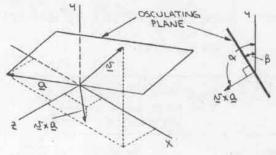
GIVEN: [= (Rt coscy t) 1 + ct 1 + (Rt sincy t) } FIND: THE ANGLE THAT THE OSCULATING PLANE FURMS WITH THE Y AXIS

FIRST NOTE THAT THE VECTORS IT AND Q LIE IN THE OSCULATING PLANE. Now. [= (Rt coscult) + ct + (Rt sincult) & I= de = R(cosunt-unt sinunt)i+ci

+R(sincht+ whitcosunt) 12 B = A = R (-WA SINGAT - WA SINGAT - WATER COSWAL) +R(who cosupt + who cosupt-wat smart)k

> = WR[-(251NOUT+WAT COSWAT)i +(2 coscunt-unt sincut) &]

IT THEN FOLLOWS THAT THE VECTOR (\$ + Q) IS PERPENBICULAR TO THE OSCULATING PLANE.



(Uxa) = WAR R(cos unt-unt sinuat) C RISHWAT + WITCOSWIZ) -(2 sinult+out coscult) 0 (2000 wit-out sinult)

= WAREC(ZCOSWAT-WATSINWAT)i +R[-(sincut+watcoswatXzsincat+watcoswat) - (cos unt-write munt)(zcosunt-unternunt)] +c(zsincht+watcoswat) =

= WR[c(2coswnt-wnt sinwnt)i - R(2+w2t2)j + C(251N WAT + WAT COS WAT) E]

THE ANGLE & FORMED BY THE VECTOR (15xQ) AND THE 4 AXIS IS FOUND FROM ..

WHERE 121=1 (15xQ). i = - WAR2 (2+ WZ t2) 1(= x = 1) = wh & (c (s cosont - out sinont) =

 $+ c^{2}(251100_{H}t + \omega_{1}t\cos\omega_{1}t)^{2})^{1/2}$ $= \omega_{1}R[c^{2}(4 + \omega_{1}t^{2}) + R^{2}(2 + \omega_{1}t^{2})^{2}]^{1/2}$

- WNR(Cz(4+Wztz)+Rz(2+Wztz)z)1/2 THEN $-R(2+\omega_{1}^{2}t^{2})$ $[c^{2}(4+\omega_{1}^{2}t^{2})+R^{2}(2+\omega_{1}^{2}t^{2})^{2}]^{1/2}$

(CONTINUES)

11.180 CONTINUED

THE ANGLE B THAT THE OSCULATING PLANE FORMS WITH THE Y AXIS (SEE THE ABOVE DIAGRAM) IS EGNAL TO

B= x-90" THEN COSK = COS(B+90) = - SIND - R(2+W2+2) :. - SIMB = [(2(4+0)2t2)+R2(2+0)2t2)2]118

R(2+wite) THEN TANB= R(2+wite) OR B= TAN' RECEIVANT

* 11.181

GIVEN: [= (At cost) +(A (ti+1)) + (Bt sint) } T- ft, t-s; A=3, B=1

FIND: (a) DIRECTION OF EL AT t=0 (b) DIRECTION OF CB AT t= 25

FIRST NOTE THAT QL IS GIVEN BY EP = MxO

Now .. [= (3t cost) + (3/22+1) + (t sint) } THEN == == == 3(cost-tsint) + ==]

+ (sint+tcost) + Q = de = 3(-sint-sint-tcost) 1+3 (text-t(qua)

+ (cost + cost - t sint) & =-3(2 sint + t cost) + 3 (+2+1) = 1

+ (zcost-t sint) k

(a) AT t=0: 15=(3\$) = (3\$) + (2\$) }

15 x Q = 31 x (3) + 2k) THEN = 3 (-2)+31)

14.01=3 1(-2)2+(3)2 = 313 THEN $Q_b = \frac{3(-2\frac{1}{2} + 3\frac{b}{2})}{3(13)} = \frac{1}{\sqrt{15}}(-2\frac{1}{2} + 3\frac{b}{2})$

 $\theta_{x} = 90^{\circ} \quad \theta_{y} = -\frac{2}{13} \quad \cos \theta_{z} = \frac{3}{13}$ $\theta_{x} = 90^{\circ} \quad \theta_{y} = 123.7^{\circ} \quad \theta_{z} = 33.7^{\circ}$: cos 0x = 0 OR

(b) AT t= 25: 15= - (37 5) + (37 54 4) + (1 5) 1 Q=-(6 \$) 1 + [(7 24) 3/ 5] 1 - (7 5/ 1) E

THEN .. UZ x Q = - 2T

 $= -\left[\frac{3\pi_{F}}{3\pi_{F}} + \frac{(\pi_{F} + 4)_{H}}{54} \right] \bar{j} - (P + \frac{3\pi_{F}}{4}) \bar{j}$ +[- 3617 + 1817] k

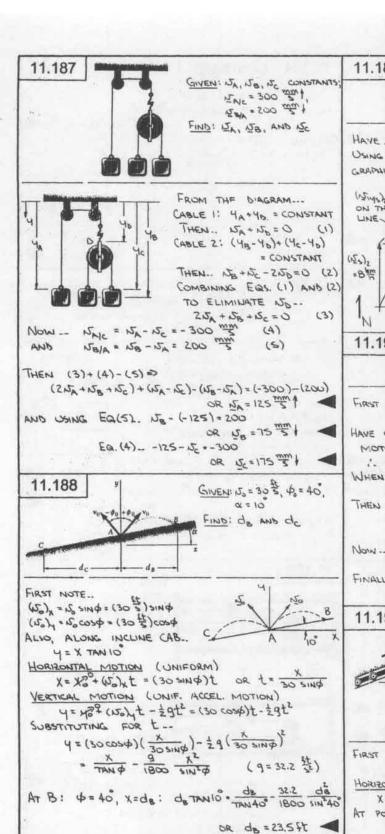
= -4.439841-13.40220+12.99459 k

AND 1 17 x Q 1 = [(-4.43984)2+ (-13.40220)2+(12.994 59)2]12 = 19.188 29

THEN ... 96 = 19.18829 (-4.439841-13.402201+12.994591) :. cosbx = - 4.439 84 cosbx = - 13.402 20 cosbx = 12.994 59

0x=103.4° By=134.3° B=47.4° OR

11.182 11.184 CONTINUED GIVEN: X = 2t3-15t2+24t+4 X-m. t-5 FIND: (a) + WHEN 5=0 OR Q = - k (No - K X) AT 1:0: Q = - 2700 5 (900 5 -0) (b) X AND TOTAL DISTANCE OR 0 = -2.43 10 12 TRAVELED WHEN Q = 0 (b) HAVE _ dx = N = No- KX AT t=0, X=0: 1 0 1 - 1 0 dt X= 2t3-15t2+24t+4 HAVE --THEN N= # 6+2-30++24
AND Q= # 12+-30 or - = [[[N(15-kx)] = t OR t= 1 LN (150 - 1) = 1 LN (1- 1 X) (a) WHEN N=0: 6t2-30t+24=0 OR (t-1)(t-4)=0 WHEN X= 3.9 IN .: t = 27005 LN OR t=15 AND t=45. OR 1=1.366x10's (b) WHEN Q=0: 12t-30=0 OR t=2.55 AT t= 2.55: X2.5 = 2(2.5)3-15(2.5)2+24(2.5)+4 11.185 GIVEN: NE = 6 31; AT 1=0, 4=40=0; FOR OR X25=1.5 m FIRST OBSERVE THAT OSECIS NO > 0 t = 45, Sp =0; Qp = 2,4 4/51 15<+\$2.55 NCO FIND: (a) t AND Y WHEN YE = YO Now .. AT t=0: X = 4 m (b) No WHEN YE = YP t=15: X1 = 2(1)3-15(1)2+24(1)+4=15 m (a) FOR t20: 4= (48) + 15=t t245: 4= (48) + (48)(t-4) a=2.45 + 2 ap (1-4)2 WHEN 4=4p-- # (1-4)2 (2.5 5) (0) EXPANSING AND SIMPLIFYING .. THEN -. X1-X0 = 15-4 = 11 m 1x25-X1= 1.5-15 = 13.5 m t - 13+ +16 = 0 SOLVING .. t = 1.3765 \$ AND t=11.6235 \$ 1. TOTAL DISTANCE TRAVELED=(11+13.5)m=24.5m MOST REQUIRE to 45 1. t=11.625 AT t= 11.6235 5: 4= (6 \$)(11.6235 5) 11.183 GIVEN: Q =- 60x -1.5 Q - 52, X-m; AT t=0 OR 4=4p=69.7 St 5=0, X=4m (b) FOR t = 45: Np = (487) + ap(t-4) FIND: (a) IJ WHEN X=2 M (b) I WHEN X=1 m AT t=11.6235 >: 15p=(2.4 \$)(11.6235-4)5 (C) N WHEN X=0.1 m OR No = 18,30 5 HAVE .. NON = 0 = - 60 x 1.5 11.186 GIVEN: UB = 150 MM -WHEN X=4m, 15=0: 10 500 = 1x (-60x -1.5)dx FIND: (a) JA OR \$152 = 120[x -0.5]X (P) N. (c) 248 OR N3 = 240 (1 - 2) (a) WHEN X=2 m: 152=240(12-2) OR N=-7,05 % (a) FROM THE DIAGRAM 5= 240 (1- 2) HAVE .. DR N=-10.95 \$ (xA-XB)+(-XB)+2(-XA) (C) WHEN X=0.1 m: 52=240(101-2) = CONSTANT OR 15=-25.3 5 THEN .. NA + 2NB = 0 SUBSTITUTING .. 11.184 GIVEN: N= No- tx No- tx X-ft; AT t=0 NA+2(-150 5)=0 X=0, 5= 900 5; WHEN X=4 IN. - Xc-OR 1 = 300 5 +4 5=0 (b) FROM THE DIAGRAM HAVE ... FIND: (a) a AT t=0 (XA-XB) + (XC-XB) = CONSTANT (b) + WHEN X = 3.9 IN. THEN .. WA - 2NB+NC = 0 SUBSTITUTING .. 300 " - 2(-150 ")+ 4 = 0 FIRST NOTE .. WHEN X= 12 St. W=0: 0=(900 5)-k(24) OR NE = 600 5 (C) HAYE -- US/B = NE - USB mm - (-150 mm) OR k = 2700 \$ (a) HAVE - 5=5- KX THEN a=#= &(16-kx)=-ku OR NE = 450 5 (CONTINUED)



AT C: \$=-40, X=-de: -de TANIO = -de

11.189 GIVEN: (US) = 8 5 - (NEWIS) (5) - 8 mt, (5 ms) 2 45 FIND: UN, WHERE UN IS CONSTANT HAVE - Now = NS + Nowing USING THIS EQUATION, THE TWO CASES ARE THEN GRAPHICALLY REPRESENTES AS SHOWN. ON THIS (TWIS) LIES ON THIS LINE FROM THE DIAGRAM .. 50 = (8)2+ (8+8)2 OR NW = 17.89 # AND TANK = 16 OR X = 63.4" :. 5 = 17.89 FM 634 (m2)=8 FM 11.190 GIVEN: P=1500 ft; N=45 H, N=30 H, A Siz = 750 St; Qt = CONSTANT FIND: a WHEN 45 = 500 ft FIRST NOTE .. J = 45 # = 665 坂=30 円=44 等 HAVE UNIFORMLY DECELERATED MOTION .. 1. 15 = 15, + 20x (5-5,) WHEN 5= 52: (44 \$)2= (66 \$)2+20x (750 \$2) OR at = - 1.61333 ths THEN WHEN 45=500 ft: an = 27 = 2742.67 452 - 1.82845 45 Q2 = Q2 + Q2 = (-1.6133332)2 (1.82845 12)2 FINALLY .. OR a = 2.44 5 11.191 GIVEN: No = 24 \$ FIND: 0 FIRST NOTE.. (U) = 5 COSK = (24 5) COSK (Ly)= 5 514 x = (24 5) 514 x HORIZONTAL MOTION (UNIFORM) X=X3+ (15x) = (24 cos x)t AT POINT B: 25 = (24 cosm)t OR LB = ZA COSOL VERTICAL MOTION (UNIF. ACCEL. MOTION) 4= 480+ (My) ot - 292= (24 SINN) t - 2922 (9=32.2 52)

AT POINT B: -15 = (24 SINX)tg - 29tg

(CONTINUED)

SUBSTITUTING FOR to ..

-de 32.2 (-de) TAN(-40) 1800 SIN2 (-40)

OR de = 31.6 ft

11.191 CONTINUED

-15 = (24 SING) (25) - 129 (25 COSK)2 -3 = 5 TANK - 1152 cos &

NOW - COSTA = SECTA = 1+ TANZA

THEN - - 3 = STAN & - 125x32.2 (1+ TAN &)

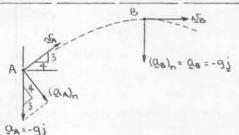
OR 3.4939 TAN2 - 5 TANA + 0.49392 = 0

SOLVING -- TANα · 0.106746 AND TANα = 1.32432

THEN -- α = 6.09° AND α • 52.9°

11.192

GIVEN: PA = 25 m FIND: (a) UTA (b) PB, WHERE 48 = 4MAX



(a) HAVE .. (ax) = 152

OR 152 = [\$(9.81 \$2)](25 m)
OR 152 = [4.0071 \$ 1. 5/- 14.01 \$ 1369° €

(b) HAVE .. (a8) = 1582

WHERE No = (NA) = 4 NA

As = (\$ × 14,0071 \$)2

OR A = 12.80 m

11.193



GIVEN: 50 = 50 - , 5= CONSTANT

FIND: B AND B IN TERMS OF U. h.

FROM THE DIAGRAM T= h Ng = No SINB

Now .. No = 18 SUBSTITUTING FOR US AND F .. 5 (BING = (SINB) B

(CONTINUED)

11.193 CONTINUED

OR B= LSINZO

HAVE 0 = FSIN B

THEN. B = # (20 sind cos 8)

SUBSTITUTING FOR D ... θ = 10 (2 SIN B COSB) (15 SIN B)

OR 0 = 2 12 5 12 0 cos 0

ALTERNATIVE SOLUTIONS

HAVE .. T = SINB

THEN $\dot{r} = -\frac{h\cos\theta}{\sin^2\theta}\dot{\theta}$ Now. $\mu^2 \cdot \nu_r^2 + \nu_\theta^2 = (\dot{r})^2 + (r\dot{\theta})^2$

OR $N_0^2 = \left(-\frac{h\cos\theta}{\sin^2\theta}\dot{\theta}\right)^2 + \left(\frac{h}{\sin\theta}\dot{\theta}\right)^2$ $= \left(\frac{h\theta}{\sin^2\theta}\right)^2 \left(\frac{\cos^2\theta}{\sin^2\theta} + 1\right)$ $= \left(\frac{1h\theta}{\sin^2\theta}\right)^2$

OR B = + TO SINED

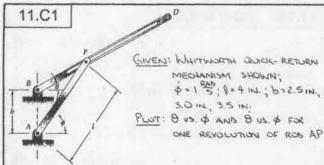
NOTE THAT AS B INCREASES, THE AIRPLANE MOVES IN THE INDICATES DIRECTION. THUS, THE POSITIVE ROOT IS CHOSEN. 1. 8 = 5 5IN2B

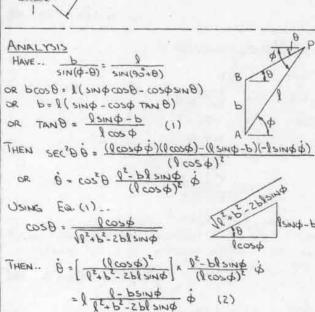
HAVE .. a = ar + ag

NOW NO = CONSTANT => Q=0

: ap = r 8+2 + 0 = 0

OR B = -S (- HORD + H SING) (= SINGB)





Eq.(1) => -905860 Thus, for these values of ϕ must use $\theta = \tan^{-1}\left(\frac{\sin\phi - b}{\cos\phi}\right) + 360^{\circ}$

WHEN PLOTTING THE GRAPH.

NOTE: FOR OSOK TAN' (102 LE)

SIMILARLY, FOR 90 < ϕ < 270, Eq. (1) => -90 < θ < 90 ... θ = TAN' ($\frac{1}{1}\cos\phi$) + 180

FOR 270° < \$ \(\phi \) 360°, Eq. (1) => -90° < \(\theta \) 0 : \(\theta = \text{TAN''} \left(\frac{1 \sin \phi - b}{1 \cos \phi} \right) + 360°

OUTLINE OF PROGRAM

INPUT VALUE OF B

CONSTRUCT BORDER FOR GRAPH OF 8 VS. 4; LABEL AXES

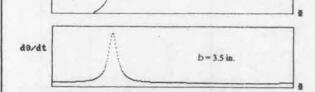
FOR VALUES OF \$ FROM 0 TO 360 IN INCREMIENTS OF 1

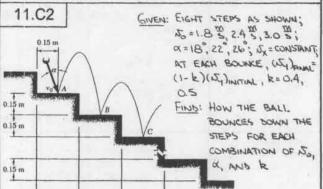
COMPUTE 8:

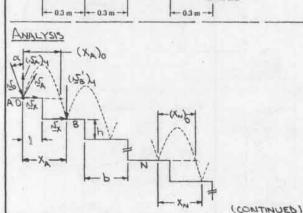
FOR $0 \le \phi < \tan^{-1}\left(\frac{b}{(1b-b^2)}\right)$, $\theta = \tan^{-1}\left(\frac{4\sin\phi - b}{4\cos\phi}\right) + 3bo$ FOR $\tan^{-1}\left(\frac{b}{(1b-b^2)}\right) \le \phi < 9o$ (CONTINUES)

11.C1 continued

PLOT (4, 8)







11.C2 continued

FIRST NOTE... \sqrt{x} , \sqrt{x} SINA $(\sqrt{x}_{A})_{+} = (1-k)\sqrt{x}$, $\cos x$ WITH THE ORIGIN OF A RECTANGULAR COORDINATE SYSTEM AT POINT 0...

HORSONTAL MOTION (UNIFORM) $X = X_{3}^{2} + \int_{x_{A}} t$ OR $t = \sqrt{x}_{x}$ VERTICAL MOTION (UNIF, ACCEL, MOTION) $y = y_{3}^{2} \cdot (\sqrt{x}_{A})_{+} t - \frac{1}{2} \cdot gt^{2}$ $\int_{x_{A}} = (\sqrt{x}_{A})_{+} - gt^{2}$ SUBSTITUTING FOR t... $y = \frac{(\sqrt{x}_{A})_{+}}{\sqrt{x}_{A}} \times - \frac{1}{2} \cdot g \cdot \frac{x^{2}_{A}}{\sqrt{x}_{A}} = \sqrt{x}_{A} \cdot g \cdot g - g \cdot \frac{x}{\sqrt{x}_{A}}$

CONSIDER THE MOTION OF THE BALL AFTER IT LANDS ON A GIVEN STEP

1. DETERMINE IF THE BALL BOUNCES TWICE ON STEP A: $ON \text{ STEP A, } q=0: O = \frac{(N_A)^q}{N_X}(X_A)_0 - \frac{1}{2}q \frac{(X_A)^q}{N_X^2}$ $OR (X_A)_0 = \frac{2}{q} N_X (N_A)_q$

.. IF (XA) < P, THE BALL BOUNCES TWICE ON STEP A.

IN GENERAL, THE BALL BOUNCES TWICE ON STEP N (N=A,B,C,...,H) IF

(xn) <) + (N-1) b - \(\frac{1}{2} \times \cdot \cdot

WHERE $(X_n)_0 = \frac{7}{9}N_X(N_n)_y$ AND X_n AND $(N_n)_y$ ARE GIVEN BELOW.

2. DETERMINE IF THE LANDS ON STEP B:

ON STEP B, y=-h: $-h=\frac{(N_n)_x}{N_x}X_n-\frac{1}{2}9\frac{X_n}{N_x}$

SOLVING FOR XA AND TAKING THE POSITIVE

ROOT $(X_{R} > 0)$, HAVE... $X_{R} = \frac{(U_{R})_{1}}{U_{R}} + \frac{1}{2} \left[-\frac{(U_{R})_{1}}{U_{R}} \right]^{2} + 4 \left(\frac{9}{2} c_{X_{R}}^{2} \right) (-h)^{\frac{1}{2}}$ $= \frac{1}{9} \left\{ (U_{R})_{1} + \sqrt{\left[(U_{R})_{1} \right]^{2} + 29h} \right\}$

". IF XA & l+b, THE BALL BOUNCES ON STEP B.

In general, after the Ball Bounces on STEP in the N it next Bounces on STEP in $\sum_{i=1}^{N} X_N \leq \frac{1}{2} + (i-1)b$

WHERE $X_N = \frac{U_N}{q} \left\{ (U_N)_q + \sqrt{(U_N)_q}^2 + 2q((i-N)h) \right\}$

Finally, if the Ball Bounces on Step B, have using the expression derived above for n_y - $(N_B^2)_q = (N_A)_q - 9 \frac{X_A}{N_X}$

NOTING THAT $(\sqrt{s})_{\gamma} < 0$ AND THAT THE MAGNITUSE OF THE VERTICAL COMPONENT $(\sqrt{s})_{\gamma}$ OF THE VELOCITY AFTER THE BOUNCE IS $(\sqrt{s})_{\gamma} = (1-k)[q\frac{x}{\sqrt{s}} - (\sqrt{s})_{\gamma}]$

HAVE IN GENERAL -- (5N-1)4]

11.C2 continued

OUTLINE OF PROGRAM

FOR INITIAL ANGLES K: $\alpha = 18^{\circ}, 22^{\circ}, 26^{\circ}$ FOR VALUES OF k: k = 0.4, 0.5FOR INITIAL VELOCITIES $\sqrt{5}: \sqrt{5} = 1.8 \frac{10}{5}, 2.4 \frac{10}{5}, 3.0 \frac{10}{5}$ FOR EACH COMBINATION OF $x, k, and \sqrt{5}$ (DMPUTE $\sqrt{5}$ and $(\sqrt{5}a)_4$:

LX = 152 SIN A (LX) = (1-k) 15 COSK

SET INITIAL CONDITIONS: N=1, 1=2, XTOTAL = 0

WHERE 1, 2, 3, ..., 8 CORRESPOND TO STEPS

A, B, C, ..., H AND XTOTAL IS THE SUM OF

THE HORIZONTAL DISTANCES BETWEEN

SUCCESSIVE POINTS OF IMPACT.

DETERMINE IF THE BALL BOUNCES TWICE ON STEP N:

IF \(\frac{2}{9} \langle \times \langle \times \langle \times \langle \langle \times \langle \langle \times \langle \langle \times \langle \langle \times \langle \langle \times \langle \langle \times \langle \langle \times \langle \times \langle \times \langle \times \langle \langle \times \langle \langle \times \langle \langle \times \langle \langle \times \langle \times \langle \times \langle \times \lang

CONSIDER THE NEXT COMBINATION OF K, k, AND IS.

DETERMINE THE NEXT STEP ON WHICH THE

WHERE XN = \frac{1}{9} \left\{ (\dagger_n)_y + \left\{ (\dagger_n)_y\right\}^2 + \frac{1}{9} \left\{ (\dagger_n)_y

DETERMINE IF THE BALL BOUNCES ON CONSECUTIVE STEPS

IF X-TOTAL > 0.15+(i-1)(0.3) AND
is 8 PRINT: BALL MISSES STEP i."
RESET X-TOTAL: X-TOTAL = X-TOTAL - X-N
UPDATE i: i=i+1
IF i < B, COMPUTE NEW XN AND

IF 128, COMPUTE NEW XN AN XTOTAL AND REPEAT CHECK IF 128, CONSIDER THE NEXT

COMBINATION OF M, E, AND NS.

DETERMINE HOW THE BALL BOUNCES

DOWN THE REMAINING STEPS

IF N > B PRINT: BALL CONTINUES

TO BOUNCE DOWN THE STEPS.

IF N < B, UPDATE VALUES FOR

THE NEXT STEP: $S_{4}: (S_{1})_{4} = (1-k) [9 \frac{X_{N}}{S_{K}} - (S_{N})_{4}]$

N: N=1

PROGRAM OUTPUT

α k V₀

18° 40% 1.8 m/s Ball first bounces twice on step A

2.4 m/s Ball first bounces twice on step C

3.0 m/s Ball misses step D

Ball continues to bounce down the

steps

50% 1.8 m/s Ball first bounces twice on step A 2.4 m/s Ball first bounces twice on step B

3.0 m/s Ball first bounces twice on step H 22° 40% 1.8 m/s Ball first bounces twice on step A

22 40% 1.8 m/s Ball first bounces twice on step A 2.4 m/s Ball continues to bounce down the steps

(CONTINUED)

(CONTINUED)

11.C2 continued 3.0 m/s Ball misses step B Ball misses step E Ball misses step G 50% 1.8 m/s Ball first bounces twice on step A 2.4 m/s Ball first bounces twice on step C 3.0 m/s Ball misses step C Ball misses step H 26° 40% 1.8 m/s Ball first bounces twice on step B 2.4 m/s Ball misses step D Ball misses step G 3.0 m/s Ball misses step B Ball misses step D Ball misses step F Ball misses step H 50% 1.8 m/s Ball first bounces twice on step A 2.4 m/s Ball continues to bounce down the steps 3.0 m/s Ball misses step B Ball misses step E Ball misses step G

11.C3

GIVEN: LOB = 10 m; aprace = - RUZ, R = 0, ZNOT m; 4×102 m; 0=70, 100; 130

FIND: NAME AND THE FIRST TWO VALUES OF 0
FOR WHICH N=0 FOR EACH COMBINATION OF 0 AND R

ANALYSIS

IN THE TANGENTIAL

DIRECTION, THE TANGENTIAL

COMPONENT OF THE

ACCELERATION OF THE

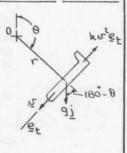
AIRPLANE IS

= 9 SIN(180-8) - Rus

RECALLING THAT Q= JE

HAVE JE 9 SIN 8 - Rus

HAVE JE 9 SIN 8 - Rus



Now, since reconstant, have 15=10 Therefore, the differential equations

and - Has

第=卡丸

DEFINE THE MOTION OF THE AIRPLANE.

OUTLINE OF PROGRAM

INPUT VALUE OF R

CASE 1: DETERMINE THE VALUE OF THE

VELIXITY AT THE SPECIFIED ANGLE OF

INPUT OF

USE, FOR EXAMPLE, THE MODIFIED EULER

METHOD (SECOND-ORDER RUNGE-KUTTA

METHOD - SEE CHAPRA AND CANALE,

NUMERICAL METHODS FOR ENGINEER'S, 2d (CONTINUES)

11.C3 continued

ED, McGRAW-HILL, 1988.) WITH A STEP SIZE At=0.008 5 TO NUMERICALLY INTEGRATE THE EQUATIONS

du = (951108-km² 8 ±180°

dt = 2-951108-km² 87180°

de = 1.00°

UNTIL $\theta_i' \in \theta_i \in \theta_c$, where θ_i' and θ_i are the values of θ at the midpoint and end, respectively, of the final time interval.

USE LINEAR INTERPOLATION TO DETERMINE THE FINAL VELOCITY IS:

LITE EINAL VELOCITY \mathcal{L}_{L}^{2} :

PRINT THE VALUES OF k, θ_0 , θ_4 , and σ_5 (ase 2: Determine the value of θ for

WHICH THE VELOCITY IS FIRST ZERO USE THE MODIFIED EULER METHOD WITH A STEP SIZE LE: 0.008 S TO NUMERICALLY INTEGRATE THE EQUATIONS

1 = 45

WHERE θ_i is the value of θ at the beginning of a time interval, until $N_2 < 0$, where N_2 is the velocity at the end of a time interval. Use linear interpolation to determine the final angle θ_L :

THE FINAL ANGLE θ_t : $\theta_t = \theta_1' + \frac{O - \Sigma_1'}{\sqrt{\Sigma_2 - \Sigma_1'}} (\theta_2 - \theta_1')$

PRINT THE VALUES OF R, BO, AND BE

SUMMARY OF PROGRAM OUTPUT

Maximum velocity attained for a release angle θ_0

$\theta_{\rm b}$	V _{nes} , m/s						
	k = 0	k =2 × 10 ⁻⁴ m ⁻¹	k =4 × 10-2 m-1	k = 0, theory			
70*	16.23	16.19	11.67	16.23			
100*	12.73	12.71	9.78	12.73			
1300	8.37	8.36	6.97	8.37			

First $[(\theta_0)_1]$ and second $[(\theta_0)_2]$ rest positions for a release angle (θ_0)

eb	k = 0		k =2 × 10 ⁻¹ m ⁻¹		k = 4 × 10 m 1	
	(B ₀),	(B ₀) 2	$(\theta_0)_1$	(Oo) 2	(B ₀),	(0)2
70°	290.0°	70.0°	289.2°	71.6*	229.4°	146.7°
1000	260.0	100.00	259.7°	100.6	223.7°	149.3°
130°	230.0°	130.0	229.9°	130.2	213.6	154.6°

GIVEN: (AR TRAVELING ON AN EXIT

RAMP; NS=60 F, NF, INAL = 0;

IO, MAX 1 = 10 1/3; RAMP 15

EITHER STRAIGHT OR CURVED

(P=800 ft); df 15 EITHER

CONSTANT OR VARIES

LINEARLY DURING TIME

INTERVALS OF 1 S

FIND: LSTOP AND DISTANCE TRAVELED

ON THE RAMP FOR EACH

COMBINATION OF RAMP TYPE

ANALYSIS

CASE 1: STRAIGHT RAMP, $\frac{dv}{dt}$ = constant

FOR THIS UNIFORMLY DECELERATES RECTILINEAR

MOTION HAVE... $\frac{ct}{dt}$ = $0 = -10 \frac{ct}{st}$ THEN $v = v_s + (-10)t$ AND $v = v_s + (-10)(x - v_s)$ Noting that $v = v_s + (-10)(x - v_s)$ HAVE $v = v_s + (-10)(x - v_s)$ $v = v_s + (-10)(x - v_s)$ $v = v_s + (-10)(x - v_s)$ And $v = v_s + (-10)(x - v_s)$ $v = v_s + (-10)(x - v_s)$ $v = v_s + (-10)(x - v_s)$ And $v = v_s + (-10)(x - v_s)$ $v = v_s + (-10)(x - v_s)$

AND OF

WHERE tSTOP AND XTOTAL ARE THE TIME FOR THE CAR TO COME TO REST AND THE TOTAL DISTANCE TRAVELED BY THE CAR ON THE RAMP, RESPECTIVELY. ALSO, UT = 60 M/h = 88 ft/s

CASE 2: STRAIGHT RAMP, OF LINEARLY VARYING HAVE a= gr AND ASSUMING THAT FOR ANY TIME INTERVAL a1=0 a2=-104/5 #= a = - 10 (+-+,) () AT t=t, N=N; (dN = [+ 10 (t-t,)dt Now - & = 5, - & (++1,)2 (1) AT t=1, X=X, : x dx = [(J, - At(1-1,)2) dt OR X=X,+15,(t-t1)-30t(t-t1)3 (2) FOR At=15 AND WHEN t=tz, HAVE .. $(1) \Rightarrow N_2 = N_1 - S (\frac{11}{5})$ $(2) \Rightarrow X_2 = X_1 + N_1 - \frac{1}{3} (\frac{11}{5})$ FOR THE FINAL TIME INTERVAL (STEINAL < 15) 5=0 AT t= triNAL. THEN, ASSUMING t,=0 (FOR CONVENIENCE) HAVE .. (1) => 0= N - A (FEINAL) At=15

OR LEWIN = (S)

AND (2) => XFINAL = X, + 5, TFINAL - 3TFINAL (ft)
WHERE XFINAL IS THE TOTAL DYSTANCE, TFINAL IS
THE TIME BURATION OF THE FINAL TIME

(CONTINUES)

11.C4 continued

INTERVAL, AND N AND X, ARE THE VELOCITY AND DISTANCE, RESPECTIVELY, AT THE BEGINNING OF THE FINAL TIME INTERVAL.

CASE 3: CURVED RAMP,
$$dt = constant$$

HAVE... $a_t = dt = constant$

NOW... $a^2 = a_t^2 + a_n^2$
 $= a_t^2 + (\frac{n^2}{p})^2$
 $a_t = a_t - a_t$

WHERE P= BOD Ft AND | amax |= 10 ft)s²
FOR EACH TIME INTERVAL, Q₁ is CONSTANT AND Q₁₇ is MAXIMUM AT TIME t, SINCE THE VELOCITY DECREASES FROM t, TO t₂.

$$\therefore \ \, \alpha_{\text{MAX}}^{2} = \alpha_{\xi}^{2} + \left(\frac{\sigma_{\xi}^{2}}{\rho_{\xi}}\right)^{2}$$
or $\alpha_{\xi} = -\sqrt{\alpha_{\text{MAX}}^{2} - \frac{\sigma_{\xi}^{4}}{\rho_{\xi}}} \quad \left(\frac{\delta \xi}{\delta^{2}}\right)$

FOR EACH TIME INTERVAL.

NOW. $a_t = constant$ (UNIF. ACCEL. MOTION)

THEN. $A_t = K_1 + a_t(t-t_1)$ (3)

AND $X = X_1 + \kappa_1(t-t_1) + \frac{1}{2}a_t(t-t_1)^2$ (4)

FOR $\Delta t = 1$ S AND WHEN $t = t_2$, HAVE...

$$(3) \Rightarrow \Omega^{r} = -\sqrt{0^{4482} - \frac{2}{6}} \left(\frac{4r}{2}\right)$$

$$(3) \Rightarrow \Omega^{r} = -\sqrt{0^{482} - \frac{2}{6}} \left(\frac{4r}{2}\right)$$

(4) => $x_2 = x_1 + x_1 + \frac{1}{2}a_{\pm}$ (ft) FOR THE FINAL TIME INTERVAL, $x_2 = 0$ AT $t = t_{\text{FINAL}}$. THEN, ASSUMING $t_1 = 0$ HAVE...

$$a_{\xi} = -\sqrt{a_{max}} - \frac{v_{1}^{-4}}{p\xi} \qquad (\frac{\xi\xi}{s\xi})$$

$$0 = 5\zeta + a_{\xi}(t_{max}) \frac{v_{1}}{s\xi} \qquad (5)$$
or $t_{final} = a_{\xi} \qquad (5)$

(4) => XENNE = X1+ NITEINAL + 202 TEINNE (42)
WHERE NI AND X1 ARE THE VELOCITY AND
DISTANCE, RESPECTIVELY, AT THE BEGINNING OF
THE FINAL TIME INTERVAL.

CASE 4: CURVED RAMP,
$$\frac{dJ}{dt}$$
 LINEARLY VARYING

ASSUMING FOR ANY TIME $\frac{dz}{dt}$ to the thing interval $\frac{dz}{dt}$ and $\frac{dz}{dt}$ to $\frac{dz}{dt}$.

HAVE:

 $\frac{dz}{dt} = \frac{(a_t)_z}{\Delta t}(t-t_1)$ $\frac{dz}{dt}$

NOW: $\frac{dz}{dt} = a_t = \frac{(a_t)_z}{\Delta t}(t-t_1)$ $\frac{dz}{dt}$

AT $t=t_1$, $z=z_1$: $\frac{z}{z}$ $\frac{dz}{dt}$ $\frac{dz}{dt}$ $\frac{dz}{dt}$ (t-t₁) $\frac{dz}{dt}$

OR $z=z_1+\frac{(a_t)_z}{z}$ (t-t₁) (S)

ALSO,
$$\frac{dx}{dt} = 15$$

AT $t = t_1, x = x_1$: $\begin{cases} x \ dx = \int_{t_1}^{t} [x_1 + \frac{(a_t)_t}{2at}(t - t_1)^2] dt \end{cases}$
OR $x = x_1 + x_1(t - t_1) + \frac{(a_t)_t}{6at}(t - t_1)^3$ (6)
NOW... $a^2 = a_t^2 + a_n^2$
 $= a_t^2 + (\frac{x_1}{p})^2$. (CONTINUED)

11.C4 continued

WHERE P= 800 ft AND | QMAX | = 10 ft/st.

NOW, FOR ANY TIME INTERVAL,

(an) MAX OCCURS AT t=t, (WHEN THE

VELOCITY IS MAXIMUM)

(at) MAX OCCURS AT t=tz

(an) MAX occurs AT ALL TIMES (NOTE...

: Assume a= amax at t=t2. Then- $a_{\text{max}}^{z} = (a_{t})_{z}^{z} + (\frac{\sqrt{2}z}{2})^{z}$ (7)

FOR $\Delta t = 1.5$ AND WHEN $t = t_2$, HAYE $(S) \Rightarrow N_2 = N_1 + \frac{1}{2}(Q_t)_2$ $OR (Q_t)_1 = 2(N_2 - N_1) \qquad (8)$

(6) $\Rightarrow x_2 = x_1 + x_2 + \frac{1}{6}(a_1)_2$ (ft) (DMBINING EGS. (7) AND (8) TO ELIMINATE $(a_1)_2$... $a_{max} = \left[2(a_2 - a_1)\right]^2 + \frac{a_2^4}{p^2}$ OR $\frac{a_2^4}{p^2} + 4a_2^2 - 8a_1^2a_2 + (4a_1^2 - a_{max}^2) = 0$ -- A GUARTIC EQUATION WHICH DEFINES a_2^2 . FOR THE FINAL TIME INTERVAL, $a_2^2 = a_1^2$. $a_2^2 = a_2^2 + a_2^2 = a_2^2$. WHERE $a_2^2 = a_2^2 = a_2^2$. $a_2^2 = a_2^2 + a_2^2 = a_2^2$. WHERE $a_2^2 = a_2^2 = a_2^2$. $a_2^2 = a_2^2 + a_2^2 = a_2^2$.

(6) \$\(\times \) \times \(\times \) \(\ti

OR trive = \- (a)

OUTLINE OF PROGRAM

INPUT INITIAL VELOCITY UT,

CONSIDER EACH CASE:

CASE 1: STRAGHT RAMP, ST = CONSTANT

COMPUTE TIME LSTOP: LSTOP = 15

COMPUTE DISTANCE XTOTAL: XTOTAL = $\frac{N_1^{-}}{20}$ PRINT THE VALUES OF LINEARLY VARYING CASE 2: STRAIGHT RAMP, OF LINEARLY VARYING FOR EACH SUCCESSIVE TIME INTERVAL COMPUTE $N_2: N_2 = N_1 - S$ WHILE $N_2 > 0$

UPDATE DISTANCE X_i : $X_i X_i + N_i - \frac{5}{3}$ UPDATE TIME AND SPEED: $t = t + 1 \; ; \; N_i = N_i$

FOR THE FINAL TIME INTERVAL
COMPUTE TRINAL : (\$15)

COMPLIE TIME ESTOP: ESTOP = E+ ENNAL

COMPLIE DISTANCE XTOTAL:

XTOTAL = XI + DITENUA - 3 TEINAL

PRINT THE VALUES OF ESTOP AND XTOTAL

CASE 3: CURVED RAMP, SE CONSTANT

FOR EACH SUCCESSIVE TIME INTERVAL

(CONTINUES)

11.C4 continued

COMPUTE S_2 : $S_2 = -(100 - \frac{S_1^4}{64 \times 10^4})^{1/2}$ COMPUTE S_2 : $S_2 = S_1 + S_2$ While $S_2 > 0$

UPDATE DISTANCE X: X=X+15+20; UPDATE TIME AND SPEED: t=t+1; 15=15;

FOR THE FINAL TIME INTERVAL

COMPUTE Q: Q= - (100 - 154)12

COMPUTE TENNE: TENNE = - OT COMPUTE TIME TSTOP: TSTOP = T+ TENNEL COMPUTE DISTANCE ATOTAL!

RTOTAL = X1 + NT TEINAL + 2 OF TEINAL

PRINT THE VALUES OF TSTOP AND XTOTAL

CASE 4: CURVED RAMP, ST LINEARLY VARYING

FOR EACH SUCCESSIVE TIME INTERVAL

SOLVE THE EQUATION $\frac{N_{2}^{4}}{L4\times10^{4}} + 4N_{2}^{2} - 8N_{1}N_{2} + (4N_{1}^{2}-100) = 0$

FOR AS USING NEWTON'S METHOD
(SEE, FOR EXAMPLE, CHAPRA AND
CANALE, NUMERICAL METHODS FOR
ENGINEERS, 2d ED., McGRAW-HILL,
1988.)

MAILE 452>0

COMPUTE $(Q_1)_2$: $(Q_2)_2$: $2(N_2-N_1)$ UPDATE DISTANCE X_1 : X_1 : X_1 + X_1 + X_1 + X_1 + X_2 + X_3 + X_4 + $X_$

FOR THE FINAL TIME INTERVAL

COMPUTE $(a_t)_2$: $(a_t)_2 = 2(a_2 - a_1)$ COMPUTE TENNA: $t_{FINAL} = [-2, \frac{a_1}{(a_1)_2}]^{1/2}$ COMPUTE TIME t_{STOP} : $t_{STOP} = t_{T_{STOP}}$

COMPUTE DISTANCE XTOTAL:

XTOTAL = X1 + N, + FINAL + & (Q4) & FINAL

PRINT THE VALUES OF TSTOP AND XTOTAL

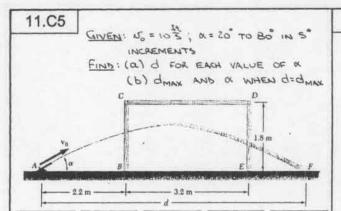
PROGRAM OUTPUT

For a straight highway and a constant rate of change of the speed, time to stop = 8.80 s distance traveled = 387.2 ft

For a straight highway and a uniformly varying rate of change of the speed, time to stop = 17.77 a distance traveled = 789.2 ft

For a curved highway and a constant rate of change of the speed, time to stop = 11.29 s distance traveled = 581.4 ft

For a curved highway and a uniformly varying rate of change of the speed,
time to stop = 20.71 s
distance traveled = 1015.3 ft



ANALYSIS

HURIZONTAL MOTION (UNIFORM) 4 OR t = x cosx

VERTICAL MOTION (UNIF. ACCEL. MOTION) 4= x3+ (5= SINX)t- 29t2 N= 5= SINX-9t SUBSTITUTING FOR t. X2 4 = (TANA) X - 29 152 COSZA

AT POINT F, X=d AND Y=0: 0= (TANA) d - 29 15 205 2

OR d= 5 SINZX

AT THE MAXIMUM THEORETICAL HEIGHT YMAX OF THE WATER, Ny = 0. THEN ... 0= 55 SINK-9tymax OR tymax = 35 SINK THEN YMM = $(\sqrt{5}\sin\alpha)(\frac{\sqrt{5}}{9}\sin\alpha) - \frac{1}{2}9(\frac{\sqrt{5}}{9}\sin\alpha)^2$

= 1 No SINSX Xymu = (15,000K) (3,000 M)

IF THE WATER HITS THE ARBOR, 4=1.8 m AT THE POINT OF IMPACT. THE CORRESPONDING VALUE OF X IS THEN .. X

1.8 = (TANA) - \$ - X (DIAT) = 8.1 OR XARBOR = TANK + 1(-TANK)2 - 3.69 - 105 to 25 COSEK

WHERE THE (+) AND (-) SIGNS CORRESPOND TO THE WATER HITTING THE ARBOR FROM ABOVE AND FROM BELOW, RESPECTIVELY.

OUTLINE OF PROGRAM

INPUT MINIMUM AND MAXIMUM VALUES OF A INPUT SIZE OF INCREMENT OF A FOR EACH VALUE OF &

COMPUTE Y AT X= 2.2 M:

42.2 = 2.2 TANA - 0.02429

COMPUTE Y AT X= S.4 W. 45.4 = 5.4 TANK - 0.14589

(CONTINUED)

11.C5 continued

- IF 42.2 > 1.8 m AND 45.4 > 1.8 m COMPUTE d: d = 9 SIN 2A (1) PRINT THE VALUES OF & AND of NEXT VALUE OF A
- (2) IF 42.2 > 1.8m AND 45.4 5 m COMPLIE (XHABOR) MEDILE:

(XARBOR)ABOVE = 100 COSX (SINK + VSIN2X-0.0369)

PRINT THE VALUES OF K AND (XARBOR) ABOVE NEXT VALUE OF A COMPUTE YMAX : YMAX = 50 SINGA

COMPUTE XYMAX: XYMAX = 50 SINZX (3) IF YMAX & 1.8 m COMPUTE d: d= 100 SINZK

PRINT THE VALUES OF X AND d NEXT VALUE OF X (4) IF 2.2 m = X4max = 5.4 m

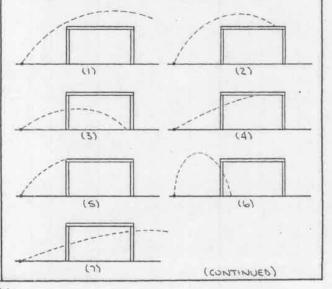
COMPUTE (XARBOR) BELOW: (XARBOR) BELOW = 100 COSK (SINK-SIN 2-0.0369

> PRINT THE VALUES OF & AND (XARBUR) BELOW NEXT VALUE OF &

(5) IF 42.2 = 1.8 m PRINT THE WATER HITS THE ARBOR AT CORNER C. NEXT VALUE OF K

IF Xymax < 2.2 m OR IF YS.4 < 1.8 m (6),(7) COMPUTE d: d= 100 SIN ZA PRINT THE VALUES OF A AND & NEXT VALUE OF X

THE SEVEN POSSIBLE TRAJECTORIES TESTED FOR IN THE PROGRAM ARE ILLUSTRATED BELOW.



11.C5 continued

PROGRAM OUTPUT

(a)

For $\alpha=20.00^\circ$, the water hits the ground at d=6.552 m For $\alpha=25.00^\circ$, the water hits the ground at d=7.809 m For $\alpha=30.00^\circ$, the water hits the ground at d=8.828 m For $\alpha=35.00^\circ$, the water hits the ground at d=9.579 m For $\alpha=40.00^\circ$, the water hits the top of the arbor from below at x=3.106 m For $\alpha=45.00^\circ$, the water hits the top of the arbor from below at x=2.335 m For $\alpha=50.00^\circ$, the water hits the ground at d=10.039 m For $\alpha=50.00^\circ$, the water hits the ground at d=9.579 m For $\alpha=50.00^\circ$, the water hits the ground at d=8.828 m For $\alpha=60.00^\circ$, the water hits the ground at d=8.828 m For $\alpha=60.00^\circ$, the water hits the ground at d=7.809 m For $\alpha=70.00^\circ$, the water hits the ground at d=6.552 m For $\alpha=70.00^\circ$, the water hits the top of the arbor from above at x=4.557 m

(b)

For α = 46.20°, the water hits the top of the arbor from below at x = 2.202 m

For α = 46.21°, the water hits the top of the arbor from below at x = 2.201 m

For α = 46.22°, the water hits the top of the arbor from below at x = 2.200 m

For α = 46.23°, the water hits the ground at d = \$10.184 m

For α = 46.24°, the water hits the ground at d = \$10.184 m

For α = 46.25°, the water hits the ground at d = \$10.184 m

For α = 46.26°, the water hits the ground at d = \$10.184 m

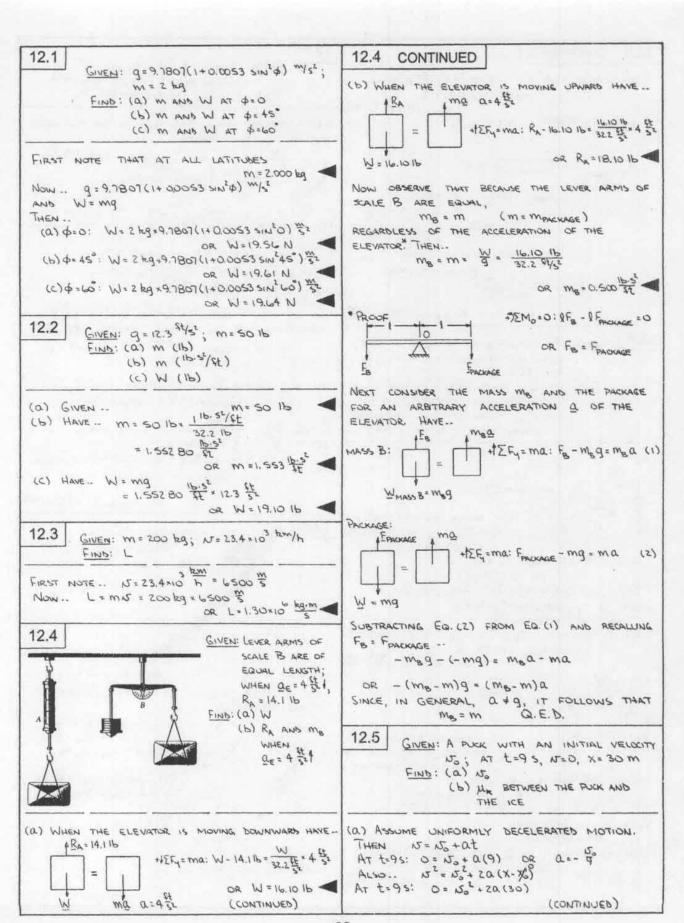
For α = 46.27°, the water hits the ground at d = \$10.184 m

For α = 46.28°, the water hits the ground at d = \$10.184 m

For α = 46.28°, the water hits the ground at d = \$10.184 m

For α = 46.28°, the water hits the ground at d = \$10.183 m

For α = 46.30°, the water hits the ground at d = \$10.183 m



12.5 CONTINUED

SUBSTITUTING FOR Q ... 0=52+2(-50)(30)=0 OR No = 6. 6667 \$

OR 15:6.67 3 -AND Q = - 6,6667 = -0.74074 Th (P)

HAVE - + 1 EF = 0: N-W=0

OR N=W=ma

SLIBING: F=HRN = Mk mg

+> EFx = ma: - F = ma

OR - ME mg = ma

OR Mr = - 3 = - -0.140 JAM/2

OR 11 = 0.0755

12.6

GIVEN: AN AUTOMOBILE INITIALLY AT REST; US : 0.80 BETWEEN THE TIRES AND THE PAVEMENT

FIND: (a) JMAX WHEN X= 400 M FOR FRONT-WHEEL DRIVE, WERONT/W = 0.62 (b) Nomax WHEN X=400 M FOR REAR-WHEEL DRIVE WREAR/W = 0.43

(a)

FOR MAXIMUM ACCELERATION .. FF = FMAX = US NF = 0.8 (0.62W) = 0.496 W = 0.496 mg

Now .. + EFx = ma: FF = ma OR 0.496 mg = ma

THEN a = 0.496 (9.81 52) = 4.86576 52

SINCE a 15 CONSTANT, HAVE --

WHEN X= 400 m: NMAX = 2(4.86576 \$)(400 m)

OR NEMAX = 62.391 3 OR 5 1 225 Th

FOR MIDNIMUM ACCELERATION .. FR = FMAX = U.SNR = 0.8(0.43W) NAB

= 0.344 W = 0.344 mg

Now -- EFx = ma: Fx = ma

OR 0.344 mg = ma

THEN Q = 0.344 (9.81 52) . 3.37464 52

SINCE Q IS CONSTANT, HAVE --

WHEN X = 400 m: 52 = 2 (3.374 64 52) (400 m)

OR Nomes = 51.959 5

OR NMAX = 187.1 Em

12.7

GIVEN: (Q) LEVEL = 3 \$ 4/5 ! BURGADE = 7, (5) URGADE = 60 MI/h; P = CONSTANT FIND: XURGRADE WHEN N= 50 mi/h

FIRST CONSIDER WHEN THE BUS IS ON THE LEVEL SECTION OF THE HIGHWAY.

HAVE .. - EF = ma: P = a a LEVEL NOW CONSIDER WHEN THE BUS IS ON THE URGRADE

HAVE. I EF = ma: P- WSINT = Qa SUBSTITUTING EOR P. . GaLEVEL - WSIN7 = & a OR a' = aLEVEL - 9 SINT = (3-32,2 SINT) } = - 0.924 19 5

FOR THE UNIFORMLY DECELERATED MOTION .. LZ = (No) UTGRADE + 20' (XUMBRADE - XS)

NOTING THAT 60 THE SHIT SHEN WHEN N= 50 mi (= & No) HAVE ... (= x88 \$)2 = (88 \$)2+2(-0.924 19 \$) XURGRADE OR XUBGRADE = 1280.16 St

IM SAS, O = 3 dassoruX SO

GIVEN: QAB = 18 A/50); 12.8 (HE)AB = (HE)BC = MR FIND: QR

FIRST CONSIDER THE MOTION OF THE PACKAGE ON SECTION AB.

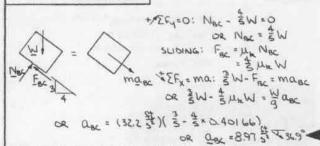
1/2 Ex=0: NAB - 3W=0 SLIBING: FAB " ME NAB \$EFx=ma: &W-FAB = Mans OR &W- & MEW = GaAB - 18 4/52) THEN

= 0,401 66

NOW CONSIDER SECTION BC.

(CONTINUED)





12.9 GIVEN: AN AUTOMOBILE'S BRAKING DISTANCE,

FIND: (a) XBR FROM 90 bm/h FOR A

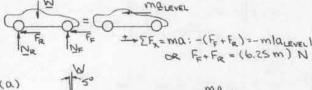
S° INCLINE - UP

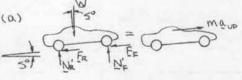
(b) XBR FROM 90 bm/h FOR A

3% INCLINE - DOWN

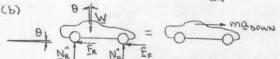
FIRST CONSIDER BRAKING ON LEVEL PAVEMENT.

ASSUMING UNIFORMLY DECELERATED MOTION, HAVE... $S^{2} = (N_{0})^{2} + 2Q_{LEVEL}(X - N_{0})^{2}$ NOTING THAT $90^{tom}/h = 25^{m}/s$ HAVE... $0 = (25^{m}/s)^{2} + 2Q_{LEVEL}(50^{m}/s)$ OR $Q_{LEVEL} = -6.25^{m}/s$





ASSUMING THAT THE BRAKING FORCE (F_F+F_R) IS INDEPENDENT OF THE GRADE, HAVE... $\Sigma F_X = MQ: -(F_F+F_R) - W \sin S^2 = MQUP$ OR $-6.25 M - MQ \sin S^2 = MQUP$ THEN $Q_{UP} = -(6.25 + 9.81 \sin S^2) = -7.1050 <math>\frac{M}{3}$ FINALLY.. $D^2 = (N_S)^2 + 2Q_{UP}(N_BR - N_O)$ SUBSTITUTING.. $D = (25 \frac{M}{3})^2 + 2(-7.1050 \frac{M}{3})^2 N_BR$ OR $N_{BR} = 44.0 M$

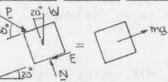


HAVE .. $^{4}\Sigma F_{x} = ma$: $W \sin \theta - (F_{F} + F_{R}) = ma_{bown}$ Now .. $TAN \theta = 0.03 \Rightarrow \theta$ smiall $\Rightarrow \sin \theta \approx TAN \theta$ THEN .. $mq TAN \theta - 6.25 m = ma_{bown}$ $or a_{bown} = 9.81(0.03) - 6.25 = -5.9557 32$ Finally .. $\beta^{2} = (n_{0}^{2})^{2} + 2a_{bown}(x_{BR} - x_{0}^{2})$ SUBSTITUTING .. $0 = (25 \frac{\pi}{3})^{2} + 2(-5.9557 \frac{\pi}{3})^{2} \times 3(-5.957 \frac{\pi}{3})^$ 12.10



GIVEN: M = 20 kg; & =0; AT 1 = 105, AX=5 m; M= 0.4, Mk= 0.3

FIND: P



FIRST OBSERVE THAT THE PACKAGE IS UNIFORMLY ACCELERATED SINCE ALL OF THE FORCES ARE CONSTANT. THEN...

CONSTANT. THEN...

X = X + 1/5 + 2 at 2

AT L = 10 5: S m = 2 a (10 5) 2

OR a = 0.10 52

NOW. + 1 EFy=0: N-Was 20-P SIN 50=0
OR N=mg cos 20+P SIN 50

SLIDING: F=UEN

** LIE (mg cos 20+ P sin 50°)

THEN... PCOS 50 - mg sin 20 - Lie (mg cos 20+ P sin 50°) = ma

OR

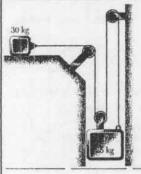
P= m[a+g(sin 20+Lie cos 20)]

20 50 - Lie 30 50°

20

COS 50° - 0.3 SIN 50° OR P = 301 N

12.11 and 12.12



GIVEN: BLOCKS A AND B

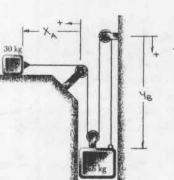
AND THE PULLEY/

CABLE SYSTEM, WHICH

15 OF NEGLIGIBLE

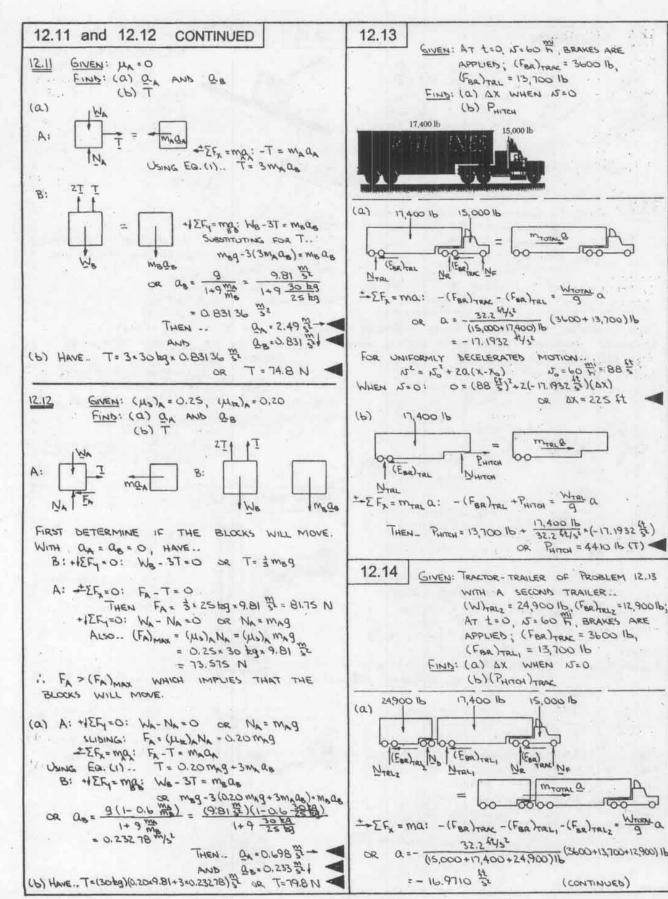
MASS, SHOWN;

(UZ) = (UB) = 0



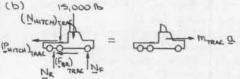
FROM THE DIAGRAM.. $X_A + 3Y_B = CONSTANT$ THEN .. $N_A + 3N_B = 0$ AND $Q_A + 3Q_B = 0$ OR $Q_A = -3Q_B$ (1)

(CONTINUED)



12.14 CONTINUED

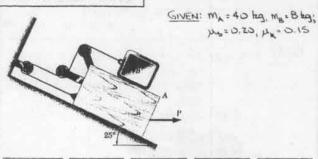
FOR UNIFORMLY DECELERATED MOTION .. Nº = PO H = 88 2 12= 20+50 (X-X9) WHEN 15=0: 0= (88 \$1)2+2(-16.9710 \$1)(0X) 12 BSS = XA SO

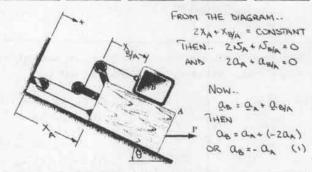


- EFX = MTRAC Q: - (FB) TRAC - (PHITCH) TRAC = WTRAC Q THEN - (PHITCH) TRAC = - 3600 16 - 15,000 16 (-16,9710 52)

OR (PHITCH) TRAC = 4310 16 (T)

12.15 and 12.16





12.15 GIVEN: P=0, 0= 25° FIND: (a) QB (b) T

FIRST DETERMINE IF THE BLOCKS WILL MOVE FOR THE GIVEN VALUE OF 8. THUS, SEEK THE VALUE OF B FOR WHICH THE BLOCKS ARE IN IMPENDING MOTION, WITH THE IMPENSING MOTION OF A DOWN THE INCLINE.



4/2 Fy = 0: NAB - WE COS 0 = 0 OR NAS = mg g cos 8 Now. FAB = MS NAB = 0.2 mgq cost OR T = mgq (0.2 cos0 + sin 0)

(CONTINUED)

12.15 and 12.16 CONTINUED

NAB 4/EF4 = 0: NA - NAB - WA COSO = 0 OR Na = (ma+ma)q cosB Now .. FA = Ms NA = 0.2 (mx+mx)9 cos0 \$2Fx=0:-T-Fx-FAB+WASINGO BROOF (BM+MB)S.O-BUILDEM = T RO

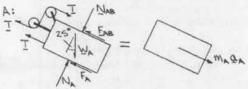
- 0.2 mag cos A

= 9[m_sin0-0.2(m_+2m_e)cos0] EQUATING THE TWO EXPRESSIONS FOR T... mag (J.Z cos0 + sin B) = g[masin B - O.Z(ma+2ma)cos B] OR B(02+TAND) = [40 TAND-02(40+2+8)]

OR TAND: 0.4 OR 8.21.8° FOR IMPENDING MOTION, SINCE 0 < 25°, THE BLOCKS WILL MOVE. NOW CONSIDER THE MOTION OF THE BLOCKS.

(a) 750 B: I +12 Fy=0: NAB - WB COS 25 = 0 OR NAB = mag cos 25° SLIBING: FAB = ME NAS

*EFX = MBaB: -T+FAB+ WB SINZS = MBaB OR T = mg [9 (0.15 cos 25 + sin 25) - aB] = 8[9.81(0.15 cos 25+ SIN 25)- as] = 8(5.47952 - 08) (N)



+/EK=0: NA-NAB-WA COSES=0 OR NA = (MA+MB) 9 cos 25° SLIDING: FA = MR NA = 0.15 (MA+ MB) 9 00525 = EFx=Maa: -T-Fa-FAB+WaSIN 25= Ma QA SUBSTITUTING AND USING EQ. (1) .. T = MAG =14 25 - 0.15 (MA+MB) g cos25 - 0.15 MB g cos25 - MA (- CB)

= 9[m451N25-0.15(M4+2MB)cos25"]+ MAQB = 9.81(40 SIN 25-0.15(40+2x8) cos25°]+40 as = 91.15202+40 aB (N)

EQUATING THE TWO EXPRESSIONS FOR T ... 8(5.47952- a8) = 91.15202+40 a8 OR an =- 0.985 75 52

:. ag = 0.986 32 125° (b) HAVE - T=8[5.47952- (-0.98575)] OR T=51.7 N

(CONTINUED)

12.16 CONTINUED

12.16 GIVEN: P = 40 N - B = 25° FIND: (a) aB (b) T

FIRST DETERMINE IF THE BLOCK'S WILL MOVE FOR THE GIVEN VALUE OF P. THUS SEEK THE VALUE OF P FOR WHICH THE BLOCKS ARE IN IMPENBING MOTION, WITH THE IMPENDING MOTION OF A DOWN THE INCLINE.

4/ [Fy = 0: NAB - WB COS 25 = 0 OR NAS = MBq COS 25 NOW -. FAB = MS NAB = 0,2 mag cos25° EFx = 0; -T+FAS+ WB SIN 25 =0

OR T = D. 2 mag cos25 + mag sin25 =(8kg)(9.81 3)(02cos25+SIN25) = 47.39249 N

+/EFy=0: NA - NAS - WA COS 25 +P SIN25 = 0 OR NA = (mA+MB) q COS25-P SIN25 Now -- Fa = Ma NA OR FA = 0.2 [(MA+MB) g cos 25" -PSINZS]

= EFx = 0: -T - FA - FAB + WA SINSS + PCOSSS = 0 OR -T-0.2[(MA+MB)gcos25-PSIN25]-0.2MBgcos25 + MA9 SIN 25+ P cos 25° = 0

OR P(0.2 SIN 25+ COS 25") = T+0.2[(MA+2MB)g COS 25"] - MAGSINZS

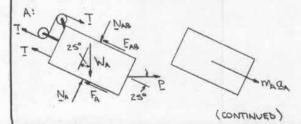
THEN P(0.2 SIN25+COS25) = 47.39249 N +9.81 = 10.2 (40+2+B)cos25-40 sin25] by OR P = - 19.04 N FOR IMPENDING MOTION. SINCE PC 40 N, THE BLOCKS WILL MOVE. NOW

CONSIDER THE MOTION OF THE BLOCKS. (a)

- We cas 25 = 0 OR NAB = MBQCOS25 SUDING: FAB : MR NAB

= 0.15 Mag cos 25 = EFx = MB QB: -T + FAB + WB SIN 25 = MB QB OR T = mg [g(ais cos 25+ sin 25) - ab] = 8[9.81 (0.15 cos 25 + SIN 25) - ab]

=8(5.47952-as) (N)



12.16 CONTINUED

+/EFy=0: NA -NAS - WA COSZS"+ PSIN 25" = 0 DR NA = (MA+MA) 9 COSES - PSINZS FA = MENA = 0.15[(mx+ma)g cos 25- P sin 25] SUDING: = E Fx = MA Qx : - T - FA - FAS + WA SIN 25 + PCOS 25 = MA QA SUBSTITUTING AND USING EQ. (1) .. T: MAQ SIN 25 - 0.15 [(MA+MB) g COSES-PSIN 25) DISMOG COSES+ PCOSES- MA (- as) = 9[masin25-0.15(ma+2ma)cos25] + P (0.15 5N 25 + 605 25°) + MA QB

= 9.81 (40 SIN 25-0.15 (40+2+8) COS 25") + 40 (0.15 MN 25+ cos 25")+ 40aa = 129,940 04 + 40 ag (N)

EWATING THE TWO EXPRESSIONS FOR T ... 8(5,47952- a8) = 129,94004 + 40 a8 OR ag =- 1,79383 52

: as = 1.794 52 1 25° . (b) HAVE .. T= B[5.47952-(-1.79383)]

OR T = 58.2 N

12.17

GIVEN: AT t=0, 5, 5, 5, 0, BELT BEGINS TO MOVE - SO THAT SLIPPING OF BOTH BOXES OCCURS; (ME)A = 0.30, (Ha) = 0.32

FIND: QA AND QB



ASSUME THAT QB > QA SO THAT THE NORMAL FORCE NAS BETWEEN THE BOXES IS BERO.

+1EF4=0: NA-WA COSIS=0 WA OR Na = Wa cos IS SLIPPING: FA = (MA) NA = 0.3 W cos 15 INEFX = MAQA: FA - WA SINIS = MAQA

OR 0.3 WA COSIS - WA SIN 15 = WA QA an = (32.2 \$)(0.3 cos 15 - SIN 15") = 0.997 \$

8: +12 Fy =0: Na - We cos 15 =0 OR No : We cos 15° SLIPPING: FB = (MR)BNB = 0.32 Wa cas 15° += EFx = maa: Fa - Wasin 15 = maa

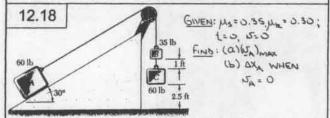
OR 0.32 W COSIS - WB SINIS = 4 AB OR a= (32.2 (1)(0.32 cos15-51415)=1.619 (

OB > OA => ASSUMPTION IS CORRECT : Q=0.997 \$ ZIS Q8=1.619 12 15° -

NOTE: IF IT IS ASSUMED THAT THE BOXES REMAIN IN CONTACT (NAS + 0), THEN (CONTINUED)

12.17 CONTINUED

an= as AND FIND (EFx = ma) A: 0.3 WA cosis - NA SINIS - NAB = ga 8: 0.32 Wa cos 15 - Wa SNIS + NAS . We a SOLVING YIELDS Q=1.273 4452 AND NAS =- 0.859 16, WHICH CONTRADICTS THE ASSUMPTION.



FIRST DETERMINE THE COMBINES MINIMUM WEIGHT OF BLOCKS B AWS C: FOR IMPENDING MOTION OF PACKAGE A UP THE INCLINE.

301WA - T + EF4=0: NA - WA COS 30 =0 OR NA = WA COS 30 Now .. FA = Ms NA = 0.35 WA cos 30 * == = 0: T-FA-WA SIN30 = 0 OR T - WA (0.35 COS30 + SN 30)

WA = 60 16 0 T = 482 16 THEREFORE SINCE THIN IS LESS THAN THE (9516) PACKAGE A WILL MOVE UP THE INCLINE WHEN BLOCKS B AND C ARE RELEASED.

(a) "MOTION I" .. A, B, AND C MOVE TOGETHER THROUGH 2.5 St.

B+C:
$$T_i$$
 +\(\Sigma_i\): $W_B + W_C - T_i$

$$= \frac{W_B + W_C}{9} \alpha_i$$

$$DR T_i = 95(1 - \frac{\alpha_i}{9}) \quad (1)$$

$$W_B + W_C \quad (M_B + m_C) \alpha_i$$

= 0.3 WA COS 30 + EFx = maa, : T, - Fa - Wa sin 30 = 3 a, OR T = 60 (0.3 cos30 + sin30 + g) = 60(0.759 808 + a) (2)

EQUATING THE TIMO EXPRESSIONS FOR TZ ... 95(1- 322) = 60 (0.759 BOB + 322)

a, = 10, 26 48 \$

MOTION 2".. C IS AT REST. A AND B MOVE TOGETHER THROUGH I St. FOR THIS CASE, Eqs. (1) AND (2) BECOME ... $T_2 = 35(1 - \frac{0.2}{9})$ (1') (CONTINUED)

12.18 continued

T2 = 60(0.759 808 + 9) (2') THEN 35 (1- 32) = 60(0.759808 + 21)

OR Qz =- 3,5889 52 . SINCE Q2 CO, A BEGINS TO DECELERATE AFTER BLOCK C REACHES THE GROWN; THUS, (UT) MAX OCCURS AT THE END OF "MOTION I." FOR THE UNIFORMLY ACCELERATION OF "MOTION I, HAVE ... 15x2 = (15x3)2 + 2a, (x-X0)

WHEN AX=2.5 ft: (15) 1 = 2(10.2648 \$1)(2.5 ft) OR (4) max = 7.16 \$ 20

(b) FIRST NOTE THAT AT THE END OF "MOTION Z,

THE SPEED OF PACKAGE A 15.. $(N_{A})_{2}^{2} = (N_{A})_{MAX}^{2} + 2Q_{2} \Delta X_{2}$ $= (7.1641 \pm 1)^{2} + 2(-3.5889 \pm 1)(144)$ OR $(N_{A})_{2}^{2} = 6.6443 \pm 1$

MOTION 3 ... B AND C ARE AT REST, A CONTINUES UP THE INCLINE AND FINALLY COMES TO REST. FOR THIS CASE, T=0 SO THAT EQ (2) BECOMES $(60(0.759808+\frac{0.5}{9})=0$ (2"),

THEN - 03 = - 0.759808 (32.2) = - 24,466 52

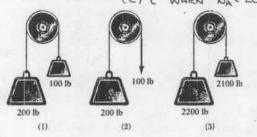
THEN .. UZ = (UZ)2+2Q3(X-X3)3 WHEN UZ=0: 0= (6.6443 \$)+2(-24.466 \$2) 0X3 OR AX= 0.9022 St

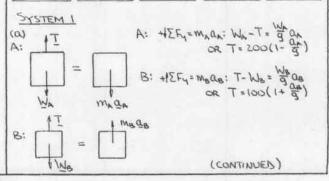
THE TOTAL DISTANCE DXA TRAVELED BY A UP THE INCLINE BEFORE COMING TO REST IS

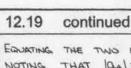
> 11(550P.0 + 1+2.5) = (XA+AX2+XA+1XA OR DXA = 4.40 ft

12.19 GIVEN: THE THREE SYSTEMS SHOWN; 15=0 FIND (FOR EACH SYSTEM): (a) QA

(b) NA WHEN DYA = 10ft (C) t WHEN NA = ZO \$







EQUATING THE TWO EXPRESSIONS FOR T AND NOTING THAT |QA = 1 as 1 ...

 $ZOO(1-\frac{QA}{3}) = 100(1+\frac{QA}{3})$ or $Q_A = \frac{1}{3}9 = \frac{1}{3}(32.2 \frac{1}{5}\epsilon) \cdot 10.7333 \frac{1}{3}$: a = 10.73 52

(b) HAVE - 152 = (18) + 20, (4-40) WHEN AYA = 10 ft: 152 = 2(10.7333 \$)(10 ft) OR NA = 14.65 1

(C) HAVE .. 15= (15) - ant WHEN No : 20 15: 20 4/2 . (10.7333 52)} OR t = 1.863 5

SYSTEM 2 + EFy= maa: Wa-T. Taan (0) I =10016 A: OR .. On: (32.2) (1- 100) OR Q4 = 16,1 321 MAQA

(b) HAVE -. 52 = (52) + 20x(4-40) MHEN AYA = 10 ft: 52 = 2(16.1 5t)(10 ft) OR 5 = 17.94 \$

(C) HAVE .. IS = (A) + OAT STATE OF STA OR t - 1.242 S

SYSTEM 3 (0)

A: +1 EFy = maa: Wa-T = gaa OR T= 2200(1- a) Ai B: + EK = MBas: T- Ws = q as

& MB QB

EQUATING THE TIMO EXPRESSIONS FOR T AND

OR T= 2100 (1+ 00)

NOTING THAT $|a_{k}| = |a_{k}|$. $|a_{k}| = |a_{k}| = |a_{k}|$ $|a_{k}| = |a_{k}|$ 1. Qx = 0.749 51

(b) HAVE .. 152 = (\$\bar{D}_{A}^{2} + 20_{A} (4-40) \\
WHEN \$\Delta y = 10 \text{ ft}; \Delta x = 2(0.748 \text{ ft})(10 \text{ ft}) OR NA = 3.87 1

(C) HAVE .. 5= (M) + ant WHEN 15=20 \$: 20 \$= (0.748 84 \$) } OR t = 26.7 5 12.20

GIVEN: Q . CONSTANT; ms : 3 kg; motion OF B IS IMPENDING Ms = 0.30, M= = 0.25 FIND: (a) a EL WHEN GEL! AND NAB=NOC=ZWB (b) NAS AND NOC WHEN QEL = 2.0%

FIRST OBSERVE THAT BECAUSE B IS NOT MOVING RELATIVE TO A AND TO C THAT QB . QEL.

(0) WBOG

HAVE - F= MSN = 0.30 (2WB) = 0.6 MB = 0.6 MBQ FOR QEL TO BE! THE NET YERTICAL FORCE MUST

BE ! WHICH REQUIRES THAT THE FRICTIONAL FORCES BE ACTING AS SHOWN. IT THEN FOLLOWS THAT THE IMPENSING MOTION OF B RELATIVE TO A AUD C IS BOWNWARD. THEN ..

+ EFY = MB aEL: 2F- WB = MBAEL OR 2(0.6 Mgg) - Mgg = Mgael DEL = 0.2 + 9.81 52

CR QEL = 1,962 521

(b) ? E +

HAVE -- F = MSN = 0.30 N

NOW OBSERVE THAT BECAUSE THE BIRECTION

OF THE IMPENDING MOTION IS UNKNOWN, THE DIRECTIONS OF THE FRICTIONAL FORCES IS ALSO UNKNOWN (ALTHOUGH FIRET MUST BE DOWNWARD). + EFy = MB aEL: ± ZF - WB =-MB | aEL | OR

1 2F = MB (9-10EL1) = 3kgx (9.81-2) 52

SINCE THE MAGNITUDE OF F MUST BE POSITIVE, IT THEN FOLLOWS THAT E AND THAT THE IMPENDING MOTION OF B RELATIVE TO A AND C IS DOWNWARD. FINALLY .. 2 (0.30 N) = 3 kg x (9.81-2) ==

OR NAB = NBC = 39.1 N





GIVEN: AT t=0, 15=0; FOR

O< t = 1.3 5, Q BELT = 2 51 ->

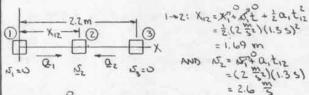
FOR t > 1.3 5, Q BELT = Q 2 ->

WHEN AXBELT = 2.2 M,

JEELT = 0; 11=0.35, 11=0.25

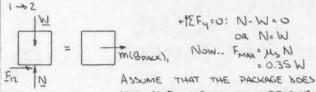
FIND: (a) Q2 (b) XARCHAGE/BELT WHEN WBELT = 0

(a) FOR THE UNIFORMLY ACCELERATED MOTION OF A POINT ON THE BELT HAVE ..



 $2 \rightarrow 3$: $A_3^2 = A_2^2 + 2a_2(x_3 - x_2)$ $0 = (2.6 \frac{\pi}{3})^2 + 2a_2(2.2 - 1.69) \text{ M}$ or $a_2 = -6.62745 \frac{\pi}{3}^2$

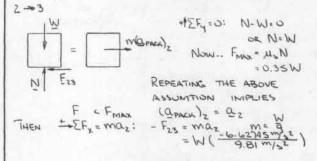
(b) NOW CONSIDER THE PACKAGE FOR EACH PORTION OF THE MOTION



MOT SUP MOR IS IN IMPENDING MOTION RELATIVE TO THE BELT.

THEN FIZEFX = Ma; (AMCK), = a, W AND .. = EFX = Ma; FIZ = Ma, 2 m/s² = W = W(9.81 m/s²)

= 0.204 W



OR F23 = 0.676 W

.. F23 (0.676 W) > FMAX (0.35 W) => ASSUMPTION IS
INCORRECT, SO THAT THE PACKAGE SUPS ON THE
BELT AS THE BELT COMES TO REST.

(CONTINUED)

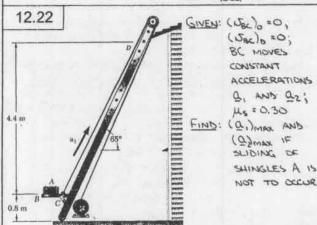
12.21 continued

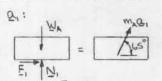
THEN. - SUPPING: $F_{23} = \mu_B N$ = 0.25 mg $^{\pm}\Sigma F_X = m(\alpha_{PMX})_2 : - F_{23} = m(\alpha_{PMX})_2$ $OR - 0.25 mg = m(\alpha_{PMX})_2$ $OR (\alpha_{PMX})_2 = -0.25(9.81 m/s^2) = -2.4525 m/s^2$ NOW. $(\alpha_{PMX})_2 = \alpha_2 + (\alpha_{PMX})_{BELT})_2$ $OR (\alpha_{PMX})_2 = \alpha_2 + (\alpha_{PMX})_{BELT})_2$ $OR (\alpha_{PMX})_2 = \alpha_2 + (\alpha_{PMX})_{BELT}$ $OR (\alpha_{PMX})_2 = \alpha_2 + (\alpha_{PMX})_{BELT}$ $OR (\alpha_{PMX})_2 = \alpha_2 + (\alpha_{PMX})_2 = \alpha_2 + (-6.62745 \frac{m}{5}^2)$

FOR THE BELT.. $d_3^2 = M_2 + \alpha_2 t_{23}$ OR $0 = 2.6 \frac{\pi}{5} + (-6.62745 \frac{\pi}{5}) t_{23}$ OR $t_{23} = 0.392315$ THEN.. (XPACUIBELT) $t_{23} = 3.6 t_{23} + \frac{1}{2} (\alpha_{maximent})_2 t_{23}^2$ $t_{23} = \frac{1}{2} (4.17495 \frac{\pi}{5}) (0.392315)^2$ $t_{23} = 0.321 m$

FINALLY .. XPACUAGE/BELT = (XPACK/BELT)12+ (XPACK/BELT)23

OR XPACUAGE/BELT = 0.32/M-

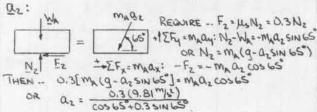




NOTE THAT THE DIRECTION OF Q_1 FIXES THE DIRECTION OF F_1 AND THAT FOR (Q_1) max, $F_1 = (F_1)$ max

THEN ... $F_1 = |J_2N_1| = 0.3N_1$ $+1 \sum_{i=1}^{n} F_{i} = m_{i} \alpha_{i} : N_1 - W_{i} = m_{i} \alpha_{i} : sin 65^{\circ}$ $OR N_1 = m_{i} (9 + \alpha_{i} : sin 65^{\circ})$ $+ \sum_{i=1}^{n} F_{i} = m_{i} \alpha_{i} : cos 65^{\circ}$ THEN ... $0.3 [m_{i} (9 + \alpha_{i} : sin 65^{\circ}) = m_{i} \alpha_{i} : cos 65^{\circ}$ $OR \alpha_{i} = \frac{0.3 (9.81^{m}/5^{\circ})}{cos 65^{\circ} - 0.3 : sin 65^{\circ}}$

OR (Q1)max=19.53 \$2 \$265°



OR (Q2)MAX: 4,24 51- 765°



GIVEN: 1=0, 5=0; 46=0.40. - MR=0,30

FIND: (a) (QTRIXX) MIN SO THAT PLYWOOD SUDES

(b) arever so THAT AXPENDONITAXXX = 2 m AT t= 0.95

(a) SEEK THE VALUE OF QTRUCK SO THAT RELATIVE MUTION OF THE PLYWOOD WITH RESPECT TO THE TRUCK IS IMPENDIUS. NOTE .. Q PLYWOOD = Q TRUEK

+1EFy = mply ay: N- Wply cos 20 = - mply are sinzo

OR N=MPLY (9 COS 20 - aTR SIN 20) + EFx = MALY Qx: F-WALY SINZO = MALY aTR COS 20 OR F = MPLY (9 SIN 20 + QTR COS 20)

SUBSTITUTING INTO EQ. LI) .. MPLY (9 SINZO+ aTR COSZÓ)= 0.4 MPLY (9 COSZO- CATR SINZO) OR OTR = 9(0.4 cos 20 - SIN 20) = (9.81 =) 0.4 - TANZO 1+0.4 TANZO

OR (QTRUCK) MIN = 0.309 52 -

(b) FIRST NOTE THAT BECAUSE ALL OF THE FORCES ARE CONSTANT, THE ACCELERATIONS ARE ALSO CONSTANT. THEN ...

XPLYITR = (XPLYITR) + 15t + 2 aprylta t2 AT t= 0.9 5: 2m = \(\frac{1}{2} april \tag{TR} (0.9 5)^2

QPLYTTR = 4.93827 5 720 DR.

apry = ate + aprylTR NOW --THEN

MALY Q PLYTTR

F= MEN HAVE -.

1- EFX = MPLY ax: F- Way SIN 20 = MPLY (att costo-apylite) OR F = mpy (9 SIN 20 + atracos 20 - apyltra) +12 Fy = MARY Qy: N- WALY COSZO = - MARY OTR SINZO OR N= MPLY (gcos 20 - at sin 20)

SUBSTITUTING INTO EQ. (1) --

MALY (9 SIN 20 + aTR COS 20 - aprilTR)

= 0.3 (MPLY (9 COS 20 - QTR SIN 20) COS 20 + 0.3 SIN 20

= (9.81 =)(0.3cos20- SIN20) + 4.938 27 == COS 20°+ 0.3 SIN 20°

OR Graver = 4.17 52-

12.24

GIVEN: SHIP OF WEIGHT WI HAVING A PROPULSIVE FORCE FO: AT t=0, 15 = No (" Nomas), FORWARD, ENGINES ARE REVERSED; FWATER & 152

FIND: X WHEN 5=0

FIRST CONSIDER WHEN THE SHIP IS MOVING FURWARD.

FUNTER = KU WHERE K IS A CONSTANT Fo - kNo = 0. or k = For

NOW CONSIDER WHEN THE SHIP IS DECELERATING.

N | EWATER

OR Q = - \frac{9}{W} (F_0 + \frac{F_0}{45} U^2) = - \frac{9}{45} \frac{F_0}{W} (U^2 + U_0^2)

Now .. 17 du = a = - 25 to (12 + 12)

AT too, X=0, 5=50: 1 dx = - 2 F 1 0 2 dx

 $X = -\frac{d}{n_2} \frac{E}{M} \left[\frac{1}{7} \Gamma M (\Omega_5 + \Omega_5) \right]_0^2 = -\frac{5}{7} \frac{d}{n_2} \frac{E}{M} \Gamma M$

12.25



GIVEN: CONSTANT FORCE P; PISTON AND ROD OF MASS M; Foil = EN; AT 1=0, X=0, 5=0 SHOW: f(x, v,t)=0 IS LINEAR

IN X, W, AND I

+ EFx = ma: P-Foil = ma or a=m(P-ks)

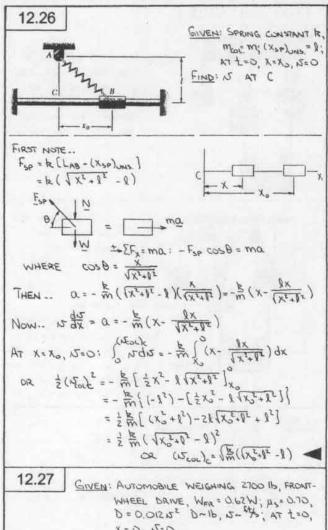
Now - # = 0 = # (P- F2) AT t=0, 5=0: [tdt = m] P-ks

OR t=m[- + LN(P-ku)] OR t=- # W P- ENT ALSO. 15 du = a = 1/2 (P- ku)

AT X=0, 5=0: [Xdx = m] " 10 do

OR X=m [[- k + k(P-ku)]du] = m[- = - = LE LN(P-FR)]" =- m(= + B= M P-FR)

USING EQ. (1) -- X=- TE+ Et Xk+mv-Pt = 0 WHICH IS UNEAR IN X, N, AND t.



X=0, 5=0 FIND: NAMAX WHEN X= 0.25 mi

F = Fmax FOR 15 = Smax ... F = 115 NF = 0.70 (0.62 W) NF = 0.62/W

 $\pm \sum_{x = ma} F - D = \frac{w}{q} a$ or $a = \frac{q}{w} (0.434W - 0.012.15^2)$ $= 0.002 \frac{q}{w} (211W - 6.65^2)$

NOW. 15 dx = a = 0.002 d (217W-622)
AT X=0, 5=0: 0.002 d / dx = 10 217W-652

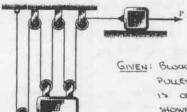
OR 0.002 \$\frac{A}{2} \times = -\frac{1}{12} \left[\text{LN} (21) M-602) \right] = -\frac{1}{12} \text{LN} \left(\frac{217 M-602}{217 M} \right)

OR 217W-652 -0.024 WX OR N=[21]W(1-0.02+ Wx)]/12

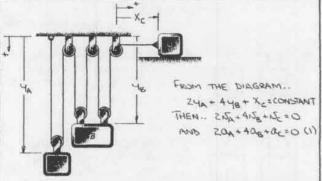
WHEN X=0.25 mi = 1320 ft: NMAX = \[\frac{217}{6} (2700)(1-\frac{2}{6}0.024\frac{32.2}{2700} \cdot 1320) \]^{1/2} = 175.285 \frac{47}{5}

OR JMAX = 119.5 Th

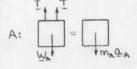




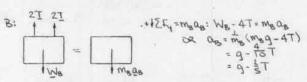
GIVEN: BLOCKS A, B, AND C AND THE PULLEY/ CABLE SYSTEM WHICH IS OF NEGLIGIBLE MASS.

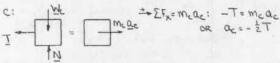


12.28 GIVEN: mx = 4 kg, mg = 10 kg, mc = 2 kg; P=0 FIND: (a) QA, QB, AND QC



+IEFy=maa: Wh -2T= maan OR an = ma (mag - 2T)





SUBSTITUTING THE EXPRESSIONS FOR QA, QB, AND QC INTO Ea. (1) --

2(9-27)+4(9-37)+(-27)=0 OR T = 379 = 57 (9.81) = 18.9871 N (a) THEN .. ax = 9.81 - 2(18.9871)

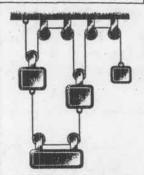
OR QA = 0.316 521 . an = 9.81 - 3 (18.9871)

OR QB = 2.22 521 ac = - 2(18,9871)

OR ac=949 51-(b) HAVE ... T=18.99 N

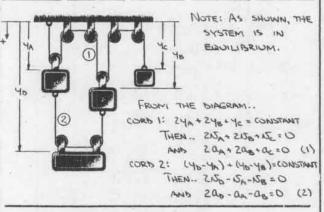
12.29 continued 12.29 GIVEN: MA = 8 kg, Mg = 16 kg, Mc = 10 kg; 11 = 0.30, Ma = 0.20; AT t=0, N=0; AT 1=0.85, Ay8=2m+ FIND: (a) QA, QB, AND QC (b) T (c) P (a) FIRST NOTE THAT BECAUSE ALL OF THE FORCES ARE CONSTANT, ALL OF THE ACCELERATIONS ARE CONSTANT. THEN ... 48 = (48) + (48) t + \$ ast2 AT t=0.85: 2m=2a8(0.85) OR QB = 6.25 52 08 = 6.25 52 1 + Ety=maa: Wa-2T= maaa OR MAG- 2T = MAGA OR B9-27-Bax (2) maga +125,= mg ag: Ng - 4T = mg ag or mag-4T = made OR 169-4T=160g (3) COMPARING EQS (2) AND (3), IT FOLLOWS THAT an = as :. QA = 6.25 To 1 SUBSTITUTING INTO EQ. (1). 216.25) + 4(6.25) + ac = 0 ac =- 37.5 52 ac=37.5 \$ -(b) FROM EQ (2) - T= 4(9-QA) = 4 (9.81-6.25) . OR T = 14.24 N (0) C: +1EK=0: W-N=0 OR N=meg SLIDING: F = MEN == EFx=mea: P+F-T=meae OR P=T+mc(ac-0.29) = 14.24 N + (10 kg)(-37.5-0249.81) == OR P=380 N-

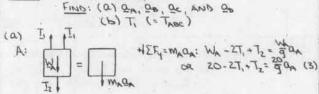
12.30 and 12.31



12.30

GIVEN: BLOOKS A, B, C, AND B AND THE PULLEY/CABLE SYSTEM, WHICH IS OF NEGLICIBLE WEIGHT, SHOWN; WA = WB = 20 16, WC = 14 16, Wb = 16 16



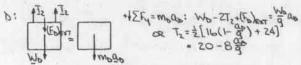


GIVEN: (Fb) EXT = 24 16 +

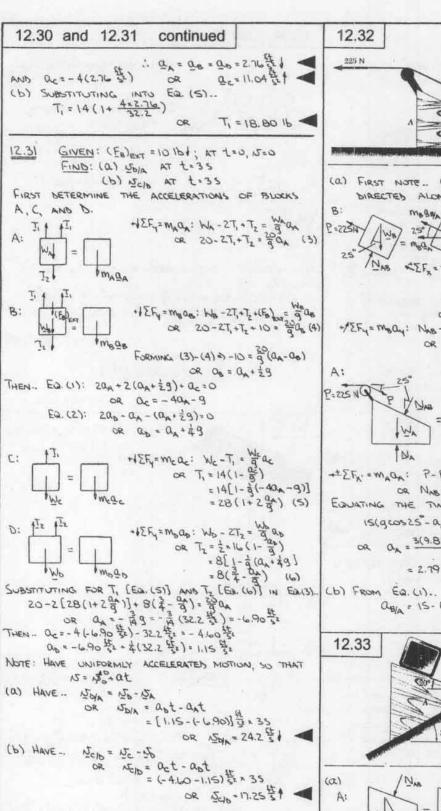
B:
$$\frac{T_1}{W_0} = \frac{1}{1} + 1 \sum_{m_0, m_0} \frac{1}{m_0} \frac{$$

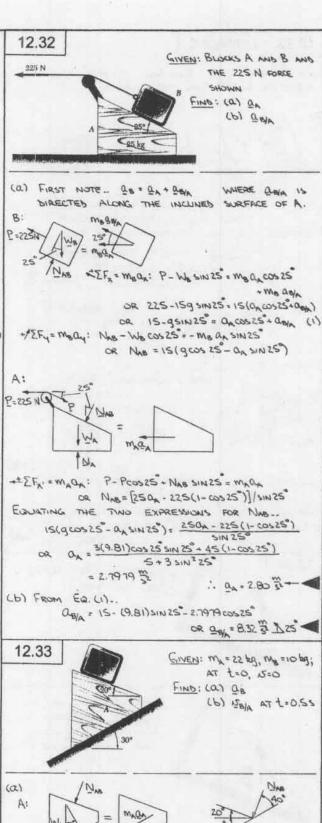
THEN .. EQ(1) => ac=-4ax EQ(2) => ao=ax

C:
$$\frac{T_{3}}{1}$$
 + $\frac{1}{2}$ $\frac{1}{2}$ + $\frac{1}{2}$ $\frac{1$



SUBSTITUTING FOR T, [EQ. (S)] AND T_2 [EQ. (6)] IN EQ. (3)... $ZO - 2 + 14(1 + \frac{4\alpha_0}{3}) + (20 - 8\frac{\alpha_0}{3}) = \frac{20}{3}\alpha_0$ $CR \qquad Q_4 = \frac{3}{35}q = \frac{3}{35} \times 32.2\frac{44}{52} = 2.76\frac{44}{52}$





WASIN 30 + NAB COSAO - MY QA

(CONTINUED)

OR NAB = 22 (QA - 29)

OR 546-17.25 \$1

12.33 continued

. Now NOTE: QB = QA + QBIA WHERE QUA IS DIRECTED ALUNG THE TOP SURFACE OF A.





+/EFy = mg ay: NAS - Wg cos 20 = - mg ax sin 50 OR NAS = 10 (9 cos 20 - Qu SIN 50°) EQUATING THE TWO EXPRESSIONS FOR NAS. 22(QA - 29) = 10(900520 - QA SIN 50) OR Qx = (9.81)(1.1+ cos 20 costo) = 6.4061 m

*EFx = mgax: Wg sin20 = mgagy - mgag cosso OR agy = 9 51N 20+04 COS 50° = (9.81 51N 20+6.4061 COS 50°) 52 = 7.4730 3

FINALLY. QB = QA + QB/A

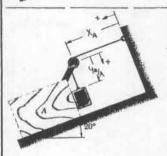
HAVE _ QB = 6.4061 + 7.4730 -2(6.4061X7.4730)cos 50 an = 5.9447 12 AND OR 0 = 74.4°

: QB = 5.94 \$ \$ TIS.6 (b) NOTE: HAVE UNIFORMLY ACCELERATED MOTION. SO THAT N= 15+ Oct

Now .. ISBN = ISB - ISA = Cat - Cat = CB/At AT t=0.55! NEW = 7.4730 52 x 0.55 OR 54 = 3.74 5 \$ 20 4

12.34

GIVEN: WA = 50 16, WB = 30 16 FIND: QA AND T



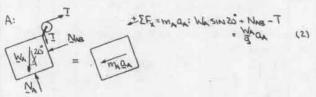
FROM THE DIAGRAM .. XA + YBA = CONSTANT THEN .. NA + NB/A = 0 an+ any =0 OR aby =- a (1)

FIRST NOTE : QB = QA + QBYA WHERE GAY IS DIRECTED ALDING THE SIDE OF A

(CONTINUED)

12.34 continued

IEFx = maax: Wasin 20 - NAB = MBOA OR NAS = WB (SIN 20 - an) B: +12 Fy = mgay: We cos 20 - T = mg agg USING ED. (1) -- A.) T= WB (COS 20 + A.) megala

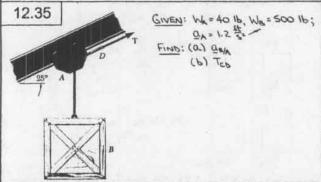


NOW SUBSTITUTE THE EXPRESSIONS FOR Now AND T INTO EQ. (2) ... 50 51N 20 + 30(51N 20 - 9)-30(coszo+ 0) = 50 9 OR CA = # (32.2 \$)(B SINZO - 3 COS 20)

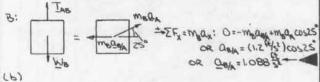
= -0.24272 英 i. a = 0.243 5 1 120 € USING THE ABOVE EXPRESSION FOR T.

T = (30 16)(cos 20 + -0.24272 14/32) 32.2 44/52

OR T = 28.0 1b



(a) FIRST NOTE: QB = QA+QBA WHERE QUA IS DIRECTED PERPENDICULAR TO CABLE AB



A: N.

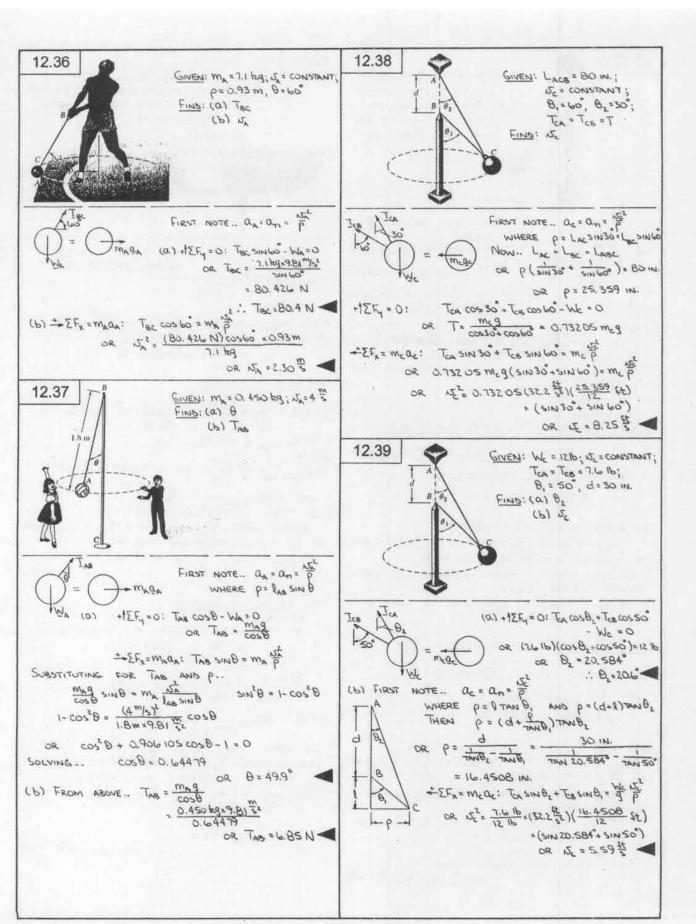
FOR CRATE B .. + PEFy = MORY: THE - WB = GAR SIN 25 OR TAB = (500 16)[1+ (1.24/5)5W25"

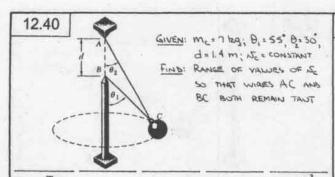
= 507.87 lb FOR TROLLEY A ...

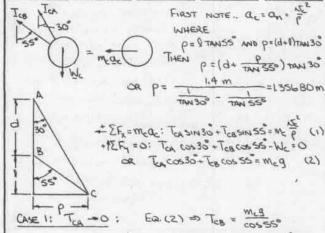
+ EFX = MAQ: TOO-TABSINZS-WASINZS - GOA

Tcb = (507.87 16) SIN25 OR. + (4016)(51425+ 1.2 tys.)

OR Tob = 233 16







SUBSTITUTING INTO EQ. (1) ... \(\frac{weg}{\cos 550} \sin 550^2 = We \frac{p^2}{p}\)

OR (\(\varphi_c^2\)\)

OR (\(\varphi_c^2\)\

Now FORM (cosso)(1) - (sin30)(2)...

To sin55 cos 30 - To cossos sin30 = m to cosso - mgsin30

OR Top sin25 = m cost cosso - m g sin30

... (No)max occurs when Top = (Tob)max, which occurs when Top = 0.

... (No)max = 4.36 & AND WIRE AC WILL BE TAUT IF No 4.36 &...

CASE Z: ToB → O: EQ.(2) => ToA = Mcg cos 30

Substituting into Eq. (1) ... $\frac{meg}{\cos 30}$ $\sin 30 = me \frac{\sqrt{2}}{p}$ OR $(\sqrt{2})_{\cos 30} = (1.35680 \, \text{m})(9.81 \, \text{m})$ OR $(\sqrt{2})_{\cos 30} = 2.77 \, \text{m}$

NOW FORM (cos 55)(1)-(sin 55)(2)...

To sin 30 cos 55°- Top cos 50 sin 55°= mc \$ cos 55°- mc g sin 55°

OR - Top sin 25° = mc \$ cos 55°- mc g sin 55°

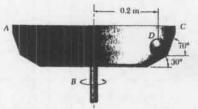
If min occurs when Top = (Top) max, which occurs when Top = 0.

If (UE) min = 2.77 \$ AND WIRE BC WILL BE TAUT IF NE > 2.77 \$.

: BOTH WIRES ARE TAUT WHEN 2.77 " C. 4.36 "

12.41

CIVEN: Mo = 0.1 kg; NS = CONSTANT
FIND: RANGE OF VALUES OF NS SO THAT
NEITHER OF THE NORMAL FORCES
EXCEEDS 1.1 N



No No mo Qu

FIRST NOTE - ab = an = p

 $+\sum F_{X} = m_{b}a_{b}: N_{i}\cos b_{0}^{2} + N_{i}\cos 2\dot{o} = m_{b}\frac{\sigma^{2}}{P}$ $+\sum F_{X} = m_{b}a_{b}: N_{i}\sin b_{0}^{2} + N_{i}\sin 2\dot{o} - M_{b} = 0$ $OR N_{i}\sin b_{0}^{2} + N_{i}\sin 2\dot{o} = m_{b}g$ (2)

CASE 1: N IS MAXIMUM

Eq. (2).. (1.1 N) sure + Nz sin 20 = (0.1 kg)(9.81 $\frac{M}{3^2}$)
or Nz = 0.082954 N
.: (Nz)(N)(Mx < 1.1 N ... O.K.

EQ. (1) ... 3.2 m (1.1 cos 60+0.082954 cos 20) N

DR (μο (Ν.) max = 1.121 = 1.1

CASE 2: N_2 is maximum LET $N_2 = 1.1 N$

Ea. (2) ... N, SINGO + (1.1 N) SINZO = (0.1 kg)(9.81 x2)

OR N, = 0.698 34 N

... (N,)(N,)MAX < 1.1 N ... O.K.

Ea. (1) -- (50) Nelma = 0.2 m (0.69834 cos60+1.1 cos20") N

ER (No)(No)ma = 1.663 5

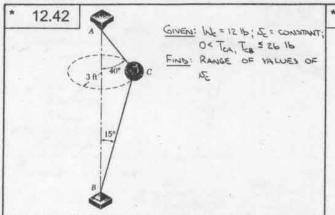
NOW FORM (SINGO)(1) - (COSGO)(2) -
NECOSZO SINGO - NESINZO COSGO = MB JE SINGO - MB G COSGO

OR NECOS 40 = MB JE SINGO - MB G COSGO

L. (UB)MAX OCCURS WHEN NE = (NE)MAX

L. (UB)MAX = 1.663 35

.. FOR N, N2 = 1.1 N



FIRST NOTE.. Q= an = D=

WHERE P= 3 ft

WHERE P= 3 ft

TEB + EFx=mcac: Tex sin40+Tex sin15 = Me g=

HEFy=0: Tex cos40-Tex cos15-INc=0 (2)

NOTE THAT EQ. (2) IMPLIES THAT

NOTE THAT EQ. (2) IMPLIES THAT
(a) WHEN TOB = (TOB)MAX, TOLK = (TOB)MAX
(b) WHEN TOB = (TOB)MIN. TOR = (TOB)MIN

NOW FORM (COS 15°)(1)+ (SIN 15°)(2)...

TEA SIN 40 COS 15°+ TEA COSAO SIN 15° = WE JE COS 15°+ WE SIN 15°

OR TEA SIN 55° = WE JE COS 15°+ WE SIN 15° (3)

... (UE) MAX OCCURS WHEN TEA = (TEA) MAX

... (UE) MAX = 12.31 ½

CASE 2: TOA IS MINIMUM

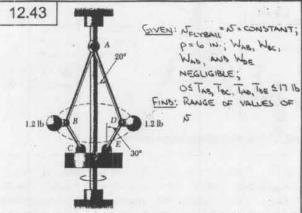
BECAUSE (TCA) MIN OCCURS WHEN TCB = (TCB) MIN, LET TCB = 0 (NOTE THAT WIRE BC WILL NOT BE TAUT). EQ. (2). TCA COS 40 - (12 Ib)=0

OR TCA = 15.6649 Ib < 26 Ib.. OX

NOTE: EQ (3) IMPLIES THAT WHEN TCA = (TCA) MIN, NZ = (NZ) MIN. THEN.

EQ. (1)-- (Λ_c^2) MIN = $\frac{(32.25)(34)}{12 Ib}$ (15.6649 Ib) SIN 40

OR $(NZ)_{MIN} = 9.005$



FIRST NOTE .. a : an = pt where p = 0 5 ft where p = 0.5 ft where p = 0.5

(ASE 1: TOA IS MAXIMUM

LET TOA = 17 16

EQ (2).. (17 16) COS 20 - TOE COS 30 - (1.2 16) = 0

OR TOE = 17.06 16... UNACCEPTABLE (>17 16)

NOW LET TOE = 17 16

EQ (2)... TOA COS 20 - (17 16) COS 30 - (1.2 16) = 0

CR TON = 16.9443 1b ... OK (K 17 1b)

... (TON) MAX = 16.9443 1b (The) MAX = 17 1b

EQ. (1) -
(U2) (TON) MAX = (32.2 \$\frac{4}{5}\$ (0.5\frac{6}{5}\$) (16.9443 SIN 20 + 17 SIN 30) 1b

OR NS(TOA) MAK = 13.85 5

Now Form (cos30)(1) + (sin30)(2)...

Top sin20 cos30 + Top cos20 sin30 = Wyz cos30 + Wsin30

OR Top sin50 = Wyz cos30 + Wsin30

(3)

Mark occurs WHEN Top = (Top)max

... DMAX = 13.85 5

CASE 2: TOM IS MINIMUM

BECAUSE (TOM)MIN OCCURS WHEN TOE = (TOE)MIN,

LET TOE = 0.

EQ. (2) .. TOM COS 20 - (1.2 1b) = 0

OR TOM = 1.27701 1b < 17 1b .. OK

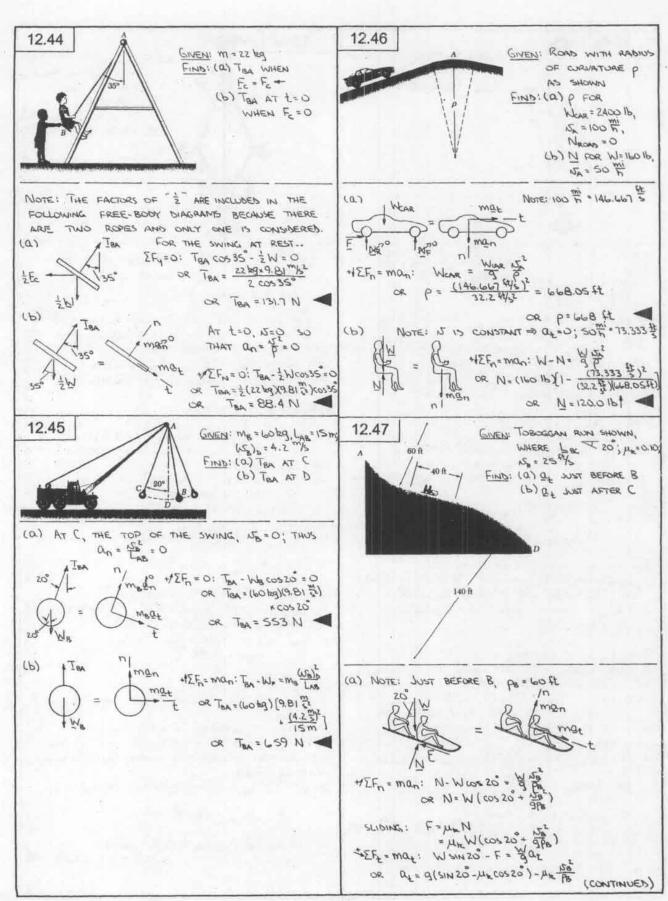
NOTE: EQ. (3) IMPLIES THAT WHEN TOM = (TOM)MIN,

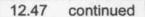
N = NOMIN. THEN.

EIG. (1) (N²)MIN = (52.2 \$\frac{1}{2}\)(0.5 \$\frac{1}{2}\)(1.27701 1b) SIN 20

OR NMIN = 2.42 \$

:. 05 TAB, TBC, TAD, TOE \$ 17 16 WHEN 2.42 15 5 5 5 13.85 15





THEN. at = (32.2 \$)(31N 20-0.1 cos20)-0.1 (25 \$)2 60 \$ \$

OR at = 6.95 \$ \$ \forall 20

(b) It is first necessary to determine it.



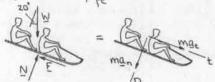
4/ΣFy=0: NBC-Wcos 20° =0 OR NBC=Wcos 20°

SLIDING: FBC = MR NBC = MR WCOS 20° = EFX = MaBC: WSIN20 - FBC = 9 ABC OR OBC = 9(SIN20 - MR COS 20°) = (32.2 fYs²)(SIN20 - 0.1 COS 20°) = 7.9B72 fYs²

FOR THIS CHIFORMLY ACCELERATES MOTION HAVE ...

= $(25\frac{4}{5})^2 + 2(7.9872\frac{4}{5})(40\%)$ OR $\sqrt{2} = 35.552\frac{4}{5}$

NOW .. JUST AFTER C, R = 140 ft

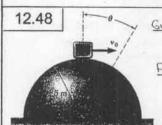


+/ΣFn=man: Wcoszo-N= 9 R 2 2 3 2)

SLIDING: F=μ_EN =μ_EW(co=20- 4FE) ±Σξ=ma_E: Wsinzo-F= 4 at OR at = g(sinzo-μ_E coszo)+μ_E a

NOTE: 9(SINZO- ME COSZO) = asc THEN- Q= 7.9872 \$ + 0.1 (35.552 %)2

OR Qt = 8.89 \$ T Zo €



<u>CIVEN</u>: M=0.5 bg; AT t=0, Ω=Ω; WHEN θ=30,

FIND: (a) No

(b) FORCE EXERTED ON

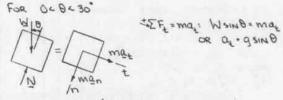
THE SURFACE BY

THE BLOCK WHEN

(a) WHEN $\theta = 30^{\circ}$.

WEF = man: Wcox 30 = m $\frac{5^{\circ}}{2^{\circ}}$ or $\frac{5^{\circ}}{2^{\circ}} = pq \cos 30^{\circ}$ N=0 man $\frac{5^{\circ}}{2^{\circ}} = pq \cos 30^{\circ}$ (CONTINUED)

12.48 continued

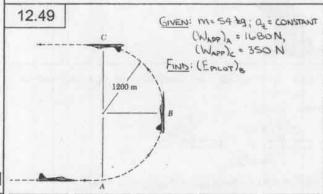


THEN $U_0^2 = pq \cos 30^2 - 2pq (1 - \cos 50^2)$ = $pq (3 \cos 30^2 - 2)$ = $(1.5 m)(9.81 \frac{m}{5}^2)(3 \cos 30^2 - 2)$ = $8.800 + 9 \frac{m}{5}^2$

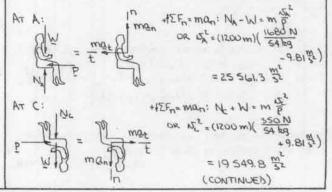
(b) WHEN $\theta = 0$.. V = W = 0 V = 0

:. THE FORCE EXERTED ON THE SURFACE BY THE BLOCK
IS

1.971 N



FIRST NOTE THAT THE PILOT'S APPARENT WEIGHT IS EQUAL TO THE VEXTICAL FORCE THAT SHE EXERTS ON THE SEAT OF THE JET TRAINER.



12.49 continued

SINCE Q = CONSTANT, HAVE FROM A TO C .. DR 19 549.8 % = 25 561.3 % + 20 & (Tx 1200 m) THEN FROM A TO B ...

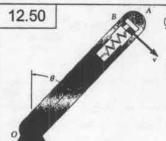
158 = 15A + 2QL ASAB = 25 561.3 1 +2 (-0.797 30 52)(2x1200 m) = 22 555 mE

AT B: OR NB= 1014.98 N=

+1EF+ = max: W+ == m1a+1 or PB = (54 trg) (0.79730-9.81) \$2 or PB = 486.69 N1

FINALLY - (FPILOT) = \ NB + PB2 = \((1014.98)^2 + (486.69)^2 = 1126 N

OR (FPILOT) = 1126 N 1 25.6



GIVEN: WR = 0.5 16; 0 = CONSTANT; WHEN 8=180°, NEACE = 0.8 16 FIND RANGE OF VALUES OF B SO THAT NEACE = 0

FIRST NOTE THAT 8 = CONSTANT => No = CONSTANT => at = 0 WHEN 8=180:

MI (FSP)180°

+12Fn = man: (NFACE) 180 + (FSP) 180 - W = m 18

FOR AN ARBITRARY VALUE OF 0: \$EFn=man: NEACE + FSP + WOOSE= m D

NOW. AS BLOCK B LOSES CONTACT WITH THE CAVITY NEACE -O, FED = (FEP) BO, P= PMAN AT A, NFACE = 0, FED = (FED) 185, PER PMAN STATEM ... (FED) 185 + WCOSO = (NFACE) 185 + (FED) 185 - W (= MFAME) cos 0 = (NFACE) 180 -1 = 0.816 -1 = 0.6 B= ± 53.1°

: BLOCK B IS NOT IN CONTACT WITH THE FACE OF THE CAVITY AT END A WHEN -53.1 5 8 5 53.1°

12.51

GIVEN: CAR TRAVELING AT A CONSTANT SPEED NO ON A ROAD BANKED AT AN ANGLE

FIND: RANGE OF VALUES OF IT SO THAT THE CAR DOES NOT SKID; N= f(r, 0, 45)

CASE 1: N= Nmax

NOTE: R = F + N

+ ΣFn=man: RSIN(0+A)= m - (1) +1 EFy = 0: RCOS(8+45) - W= 0

 $\frac{R \sin(\theta + \phi_s)}{R \cos(\theta + \phi_s)} = \frac{R \cos(\theta + \phi_s) = mq}{mq}$ (2) FORMING (2) -OR Nomax = 195 TAN (8+ 45)

CASE 2: N= SMIN

NOTE: R = F+N

+ ΣFn = man: R SIN (0-4)= m mn (3)

+ (EFy = 0: RCOS(0-45)-W=0 OR RCOS(B-A). mg FORMING (4) --R SIN(B-AS) = M SMIN R COS(B-AS) = mg OR JMIN: 195 TAN(8-4)

". FOR THE CAR NOT TO SKIB ..

1 OF TAN (8-45) = U = 195 TAN (8+45)

12.52

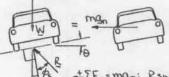
GIVEN: 15=95 T; 1=40m; 11=0.70 FIND: (a) ON FOR NO SKIDDING WHEN 8=10 (b) DU FOR NO SKIBBING WHEN 8=-5°



FIRST NOTE .. TIAN \$= 0.70 (=Us) OR \$ = 34.992°

ALSO, REQUIRING THAT THE SPEED OF THE GAR BE DECREASED TO AVOID SKIDDING, IMPLIES THAT IMPENDING SCIDDING IS "OUTWARD."

(a) 8=10°



+ EFn = man: RSIN(8+\$) = m = +12Fy=0: Rcos(0+4s)-W=0

(1)

OR RCOS(B+ \$4) = mg (2) FORMING (2) ROS(0+45) = mg (CONTINUES)

12.52 continued

OR 152 = 9 = TAN (0+45) = (9.81 = X40 m) TAN (10+34.992)

OR 15 = 19.8063 = 71.302 = M/h

THEN ... AN = NS - NS = (95-71.302) = M/h

OR AN = 25.7 = M

(b) $\theta = -5^{\circ}$ $191 - R \xrightarrow{\Delta \Sigma F_{n} = ma_{n}} R \sin(\Delta - 101) = m\frac{\Delta}{\Sigma}$

FORMING $\frac{(3)}{(4)} = \frac{R \sin(\Phi_8 - 101)}{R \cos(\Phi_8 - 101)} = \frac{m \Phi_8}{m \Phi_8}$ (4)

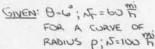
R cos(\$\delta_{5}-181) mg

OR \(\sigma_{=}^{2} = 9\tau \tau (\delta_{5}-181) = (9.81 \frac{\pi}{2})(40 m) \tau (34.992 - 5°)

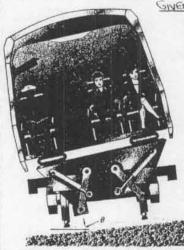
OR \(\sigma_{=}^{2} = 15.0492 \frac{\pi}{2} = 54.177 \frac{\tam}{2} m/h

THEN \(\tau \sigma_{=} \sigma_{5} - \sigma_{=} (95 - 54.177) \frac{\tam}{2} m/h

12.53 and 12.54



OR AN=40.8 Th



 $F_{S} = W \left[\begin{array}{c} W_{S} \\ W_$

12.53 and 12.54 continued

12.53 GIVEN: A PASSENGER OF WEIGHT W FIND: (a) Fo WHEN \$0:0 (b) \$\Phi\$ FOR Fo = 0

(a) Substituting THE KNOWN VALUES INTO EQ. (1)... $F_S = W \Big[\frac{(100 \, \text{m/h})^2}{(60 \, \text{m/h})^2} \, \text{TANG} \, \cos 6^{\circ} - \sin 6^{\circ} \Big]$ $= W \Big(\frac{2}{9} - 1 \Big) \sin 6^{\circ}$

OR F5 = 0.1858 W .

(b) SETTING $F_S = 0$ IN EQ. (1).. $0 = W \Big[\frac{(100 \text{mi/h})^2}{(60 \text{my/h})^2} \text{ TRNG} \cos(6 + \phi) - \sin(6 + \phi) \Big]$ OR $\text{TRN} (6 + \phi) = \frac{25}{9} \text{ TRNG}$ OR $6 + \phi = 16.28$

OR \$=10.28°

12.54 GIVEN: F. = 0.1 W FIND: \$

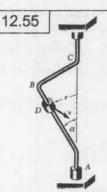
SUBSTITUTING THE KNOWN VALUES INTO EQ. (1)... $O(1 N) = M \left[\frac{(100^{mi}h)^2}{(60^{mi}h)^2} TANG \cos(6+\phi) - \sin(6+\phi) \right]$

OR $[0.1+\sin(6+\phi)]^2 = [\frac{25}{9} \tan 6^{\circ} \cos(6+\phi)]^2$ OR $0.01 + 0.2 \sin(6+\phi) + \sin^2(6+\phi)$ $= 0.08 \le 238[1-\sin^2(6+\phi)]$

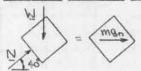
 $= 0.085238[1-5in(6+\phi)]$ OR $1.0852385in^{2}(6+\phi)+0.25in(6+\phi)-0.075238=0$ SOLVING FOR THE POSITIVE ROOT.- $5in(6+\phi)=0.186816$

OR 6°+0 = 10.77°

OR \$=4.77"



GIVEN: Mo = 0.3 kg; & = 40 BASC = 5 PE (CONSTANT) FIND: F IF F = CONSTANT



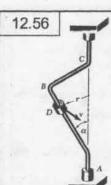
FIRST NOTE __ No TO PRO + 12 Fy = 0: NSIN 40 - W = 0 OR N = mg

 $\Sigma F_n = man: N\cos 40 = m \frac{s_b^2}{r}$ or $\frac{mg}{\sin 40} \cos 40 = m \frac{(r\theta_{ABC})^2}{r}$

OR $r = \frac{9}{82} \frac{1}{14040}$ $= \frac{9.81 \, \text{m/s}^2}{(5 \, \text{RAD/s})^2} \frac{1}{14040}$

=0.468 m

DR 5 = 468 mm



GNEN: Mo=0.2 kg; a=30, C=0.6 m, Mo=0.30; Bacc = CONSTANT FIND: RANGE OF VALUES OF IS SO THAT COLLAR D BOES NOT SLIDE ON THE ROA

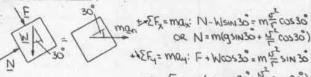
CASE 1: IT = JMIN, IMPENDING MOTION DOWNWARD

= EFx=max: N- Wsin30=m=cos30 OR N=m(9511323+4 cos36) + EFy= may. F- Wcos30=-m = sw30 OR F=m(90530-15 51430)

F= MSN Now -W(qcos30- 5 sin30)= 115 x m(qsin30+ 5 cos30) IHEN .. OR 52= 91 1-115 TAN 30 MS+ TANSO = (9.81 52)(0.6 m) 1-0.3 TAN 30 0.3 + TAN 30

OR JMIN = 2.36 5

CASE 2: N=NMM, IMPENDING MOTION DAWARD



OR F = m (-qcos30+ = SIN30) F= Ms N Now ..

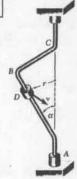
m(-gcos30+ = sin30)= H=x m(gsin30+ = cos30) 1 = 95 1+ 11= TAN 30 - 115 = (9.81 m/s) (0.6m) 1+0.3 TAN30 -0.3

CR JMAX = 4.99 5

FOR THE COLLAR NOT TO SUBE. 2.36 € 55 4.99 €

12.57

THEN



GIVEN: WD = 0.616; F=8 IN. B = 10 RADYS (CONSTANT); COLLAR D DOES NOT SLIDE ON THE ROD FIND: (a) (US) MIN WHEN \$\alpha = 150

(P) (A) WIN MHEN 0 = 45°

(CONTINUED)

12.57 continued

FIRST NOTE THAT N=(BAR = (BEA)(10 PM), 30 A AND THAT REQUIRING US=(US)MIN IMPLIES THAT SLIDING OF COLLAR D IS IMPENDING. ALSO, US = TAN As

NOW CONSIDER THE TWO RUSSIBLE CASES OF INPENDING MOTION.

CASE 1: IMPENDING MOTION DOWNWARD

$$\frac{1}{N} = \frac{1}{N} \sum_{k=1}^{N} \sum_{k=1}^{$$

Now -- F= MSN THEN - W(COSK - 5 SINK) = MAXW(SINK + OF COSK) 1- US TANK 1- TANDS TANK TANK + TANGS = TAN(M+PS)

CASE 2: IMPENDING MOTION UPWARD

$$\frac{1}{N} = \frac{1}{N} \sum_{k=1}^{N} \sum_{k=1}^{N$$

Now .. F= us N THEN .. W(-COSK+ OF SINK) = MOX W(SING+ OF COSK)

TAN(X-\$) 35 = (32.2 ft/s2)(8 ft) = 0.483 (39 th/)2

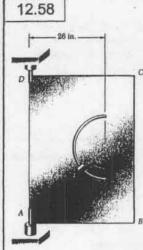
THEN TAN(a + \$) = 0,483 x = \$ = 25.781° . \$ 20 AND WHERE THE "+ CORRESPONDS TO IMPENDING DUIDUNG OF - THE GUA GRAWING WOITOM MOTION UPWARD.

(a) x=15°: HAVE 15° + \$= 25.781° \$ 18.00 \$ "+" SO THAT \$ 10.781" THEN (US) THEN TAN 10.781°

OR (M2) MIN = 0.1904, MOTION IMPENDING BONDWARD (6) x = 45°: HAVE 45° + \$= 25.781° 45° + \$= 25.781°

\$ 20 \$ " -" SO THAT \$ = 19.219" THEN (Us) min = TAN 19.219°

OR (US)min = 0.349, MOTION IMPENDING UPWARD



GIVEN: T=10 IN, \$\delta_{ABCD} = 14 \frac{RAD}{5};
WE = 0.8 16; \mu_5 = 0.35;
\[
\mu_K = 0.25
\]

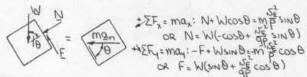
C FIND: (a) E AND IF THE BLOCK
SLIDES IN THE SLOT AT

t=0 WHEN B. BS

(b) F AND IF THE BLOCK
SLIDES IN THE SLOT
AT t=0 WHEN B=40

FIRST NOTE.. $\rho = \frac{1}{12}(26 - 10 \sin \theta) \text{ St}$ THEN $\Omega_n = \frac{15^2}{98}(13.5 \sin \theta)^{\frac{1}{2}} = \frac{1}{12}(26 - 10 \sin \theta) \text{ St}(14 - \frac{845}{5})^2$

ASSUME THAT THE BLOCK IS AT REST WITH RESPECT TO THE PLATE.



(a) HAVE $\theta = 80^{\circ}$. THEN $N = (0.8 \text{ lb})[-\cos 80^{\circ} + \frac{32.2 \text{ fg}}{32.2 \text{ fg}}; *\frac{98}{3}(13-5\sin 80^{\circ})]^{\frac{1}{4}} \cdot \sin 80^{\circ}]$ $= (0.8 \text{ lb})[\sin 80^{\circ} + \frac{1}{32.2 \text{ fg}}; *\frac{98}{3}(13-5\sin 80^{\circ})]^{\frac{1}{4}} \cdot \cos 80^{\circ}]$ = 1.92601 lb

NOW- FMAX = ALSN = 0.35(6.3159 16) = 2.2106 16

.. THE BLOCK BOES NOT SLIDE IN THE SLOT F=1926 16_1 80°

(b) HAVE $\theta = 40^{\circ}$ THEN

N = (0.8 lb)[- $\cos 40^{\circ} + \frac{32.246}{32.2465} \times \frac{98}{3}(13.5 \sin 40^{\circ})\frac{15}{3} \times \sin 40^{\circ}]$ = 4.4924 lb

F = (0.8 lb)[$\sin 40^{\circ} + \frac{1}{32.2465} \times \frac{98}{3}(13.5 \sin 40^{\circ})\frac{15}{3} \times \cos 40^{\circ}]$

= 6.5984 B

Now - Fmax = USN FROM WHICH IT FOLLOWS

THAT F > Fmax

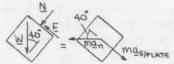
.. BLOCK E WILL SLIDE IN THE SLOT AND QE = Qn + QEIPLATE

= Qn + (QE/PLATE)+ (QE/PLATE)

AT \$\frac{1}{2}\$, THE BLOCK IS AT REST RELATIVE TO THE PLATE, THUS, (QEIPLATE) = 0 AT \$\frac{1}{2}\$ SO THAT QEIPLATE MUST BE DIRECTED TANGENTIALLY TO THE SLOT.

(CONTINUED)

12.58 continued



+ EFx = max: N+Wcos40 = mp sin40 OR N=W(-cos40+ ye sin40) (AS ABOVE)
= 4.4924 16

SLIDING: F= MEN = 0.25 (4.4924 16) = 1.123 16

NOTING THAT E AND $Q_{E/PLANE}$ MUST BE DIRECTED AS SHOWN (IF THEIR DIRECTIONS ARE REVERVED, THEN Σ_{X}^{F} is a while max is a) have ... The Block subes downward in the slot and F:1.123 ib Λ 40°

ALTERNATIVE SOLUTIONS

(Q) Assume that the block is at rest with respect to the plate.

THEN.. TAN $(\phi-10^{\circ}) = \frac{W}{ma_n} = \frac{W}{W} \frac{JE}{P} = \frac{9}{P(\phi_{ABCB})^2}$ $= \frac{98}{32.2} (13-5 \sin 80^{\circ}) \frac{11}{52} \quad (FROM AB)$

OR \$-10° = 6,9588° AND \$=16,9588°

NOW -. TANDS US US= 0.35 SO THAT \$= 19.29

SO THAT \$5=14.29 . O < \$ < \$ ⇒ BLOCK BOES NOT SUBE AND R IS DIRECTED AS SHOWN.

Now- $F = R \sin \phi$ AND $R = \frac{W}{\sin(\phi - 10^{\circ})}$

THEN.. $F = (0.8 \text{ Ib}) \frac{\sin 16.9588^{\circ}}{\sin 6.9588^{\circ}} = 1.926 \text{ Ib}$.' The <u>Black does not slibe</u> in the slot and F = 1.926 Ib 1.080°

(b) Assume that the block is at rest with respect to the plate.

EE= ma: W+R= man FROM PART a (ABOVE), IT THEN FOLLOWS THAT

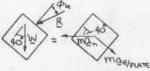
TAN (0-50) = 9 = 32.2 1t/52 = 32.2 1t/52 = 98 (13-5 SIN 40) 14/52

OR \$-50 = 5.752°
AND \$=55.752°

Now $\phi_s = 19.29^\circ$ so that $\phi > \phi_s$.'. THE BLOCK WILL SUDE IN THE SLOT AND THEN $\phi = \phi_k$ WHERE TAND = $\mu_k = 0.25$ OR $\phi_k = 14.0362^\circ$

12.58 continued

TO DEFERMINE IN WHICH DIRECTION THE BLOCK WILL SLIBE, CONSIDER THE FREE-BODY DIAGRAMS FOR THE TWO POSSIBLE CASES.



Maryguate

DOWNWARD

Now - EE = ma: W+R = man + marphire FROM THE DIAGRAMS IT CAN BE CONCLUDED THAT THIS EDUCATION CAN BE SATISFIED ONLY IF THE BLOCK IS SLIDING DOWNWARD. THEN .. + EFx = max: Wcos 40 + R costx = m = sin 40 Now .. F = R sinter THEN - Woosto + THIRE = 3 P SIN 40 F = M& W (-cos 40 + WE SIN 40) = 1.123 Ib (SEE THE FIRST SOLUTION)

". THE BLOCK SLIDES DOWNWARD IN THE SLOT AND F=1.123 1b 1 40°

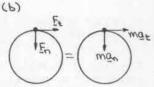


GIVEN: d=0.225 m; 5=0, Q=4 32; m=1.6=0, bg FIND: (a) N AT t= 35 (b) FUET AT 1=35

(a) a = constant => UNIFORMLY ACCELERATION MOTION THEN .. 15 = 557 04 t

AT 1=35: S= (4 m/3)(35)

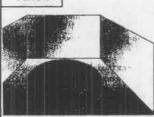
OR 15=12 3



EF = max: Fx = max OR F = (1.6x10 kg)(4 th) EFn= Man: Fn= m }

AT t= 35: Fn = (1.6x10 kg) (12m/s)2 m) = 2.048 x10 N

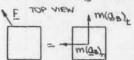
FINALLY - FTUFT = JF2+F2 = \((6.4x10 N)2+(2.048x103 N)2 OR FTUET = 2.05 110 N 12.60



GIVEN: 50=0, (as) = 0, ZA 52 TRUNK B BEGINS TO SUDE AT t= 105 FIND: MS

FIRST NOTE THAT (QB) = CONSTANT IMPUES UNIFORMLY ACCELERATED MOTION.

: N= NS+(OB) t AT t=105: 5 = (0.24 5)(105) = 2.4 5



IN THE PLANE OF THE TURNTABLE ...

EF=maao: F=malable



THEN .. F = m ((a))2+(a)2 = m ((a) 2+ (45)2 +125,=0: N-W=0

OR N=mag AT t= 10 s: F= 11, N= 11, mag

THEN ... Ms mag = ma ((ag/2+(156))2 Ms= 9.81 m/s [(2.4 m/s)2 + [(2.4 m/s)2]2 7 1/2

OR Ms = 0236

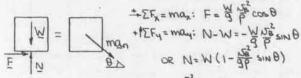
12.61

GIVEN: PARALLEL-LINK MECHANISM ABCD; 58 = 2.2 ft/s

FIND: (a) (US) MIN IF COMPONENTS ARE NOT TO SLINE

> (b) O FOR WHICH SLIDING IS INIPENDING

10 in 20 in. 20 in



NOW .. FMAX = USN = USW (1- 30 SINB) AND FOR THE COMPONENT NOT TO SLIBE F & FMAX

M 28 cos 8 = MSM (1 - 45 sin 8)

12.61 continued

.. MUST DETERMINE THE VALUES OF B WHICH MAXIMIZE THE ABOVE EXPRESSION. THUS ..

$$\int_{\Omega} \left(\frac{ds}{ds} - \sin \theta \right) = \frac{-\sin \theta \left(\frac{ds}{ds} - \sin \theta \right) - (\cos \theta)(-\cos \theta)}{\left(\frac{ds}{ds} - \sin \theta \right)^2} = 0$$

OR SIND = 39 FOR US = (MS)MIN OR SIND 373 NOW - SIND = (35.2 (1/2) = 0.180373

OR 0=10.3915" AND 0=169.609"

(a) From ABOVE,

$$(\mu_s)_{min} = \frac{3\rho}{3\rho} - \sin\theta = \frac{169.609}{169.609}$$

$$(\mu_s)_{min} = \frac{3\rho}{3\rho} - \sin\theta = \frac{169.609}{109.609} = \frac{500}{3\rho}$$

$$(\mu_s)_{min} = \frac{\cos\theta}{\sin\theta} - \sin\theta = \frac{\cos\theta}{1 - \sin^2\theta} = \tan\theta$$

OR (Us)my = 0.1834

(b) HAVE IMPENDING MIDTION TO THE LEFT FOR 0 = 10:39

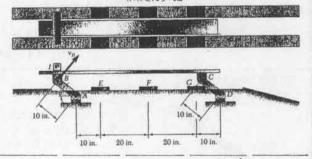
= TAN 10,3915°

TO THE RIGHT FOR 8=169.6°

12.62

GIVEN: PARALLEL-LINK MECHANISM ABCD; Ms = 0.35, Ma = 0.25 FIND: (a) (158) MAX IF COMPONENT I IS NOT TO SUBE ON MEMBER BC

(b) B FOR WHICH SUDING IS IMPENDING



== Fx=max: F= W = cos 8 man +1ΣFy=may: N-W=- 4 5 5 5 NB OR N=W(1- 40 SIN B) DA Now - Fmax = M = M = M (1 - 3/8 5/10) AND FOR THE COMPONENT NOT TO SLIDE .. F & FMAX OR W 15 COSD & MAN (1- 43 SIND)

TO ENSURE THAT THIS INEQUALITY IS SATISFIED, US) MIX MUST BE LESS THAN OR EQUAL TO THE MINIMUM VALUE OF MGGP/(COSD+MGSIND), WHICH OCCURS WHEN (COS B+ ILS SIN B) IS MAXIMUM. THUS ... (CONTINUED)

12.62 continued

0= 0 200 24+ BAIC - = (BAIC 24+ BEO) BE OR TAND = Ms Ms= 0.35 OR 8 = 19,2900°

(a) THE MAXIMUM ALLOWED VALUE OF US IS THEN .. EN - B MAX = US COS B+ US SING WHERE TAN B = US = 9P TONE (BURT) = 9PSINB

OR (NB) MAX = 2.98 \$ (b) FIRST NOTE THAT FOR 90° < 0 = 180°, EQ. (1) BECOMES No Se & MS COSA + MS SING

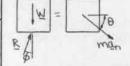
WHERE OL : 180 - B. IT THEN FOLLOWS THAT THE SECOND VALUE OF B FOR WHICH MOTION IS IMPENDING 15 ..

0=180-19.2900 = 160.7100

.. HAVE IMPENDING MOTION TO THE LEFT FOR 8=19.29" TO THE RIGHT FOR B=160.7°

ALTERNATIVE SOLUTION

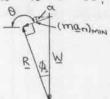
EF=ma: W+R=man THEN



FOR IMPENDING MOTION, ϕ : ϕ_s , ALSO, as shown above, the values of θ for which motion is IMPENDING MINIMIZE THE VALUE OF US, AND THUS THE VALUE OF an (an = No). FROM THE ABOVE BIAGRAM IT CAN BE CONCLUDED THAT an is minimum when man and B ARE. PERPENDICULAR. THEREFORE ...

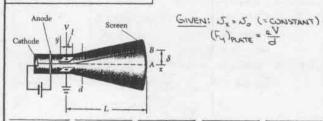
TIMENIMIN FROM THE DIAGRAM... AND $ma_n = W \sin \phi_s$ $ce m \frac{\sqrt{s^2}}{\rho} = mg \sin \theta$ OR JB = 9P SINB (AS ABOVE)

FOR 90 & 0 = 180 HAVE ...



FROM THE DIAGRAM .. $\alpha = 180^{\circ} - \theta$ (avaga ca) $\phi = \phi$ AND man = W SINAS OR JB = 9P SIND (AS ABOVE)

12.63 and 12.64



FIRST NOTE THAT THE HURIZONTAL COMPONENT OF THE VELUCITY OF AN ELECTRON IS A CONSTANT (NJ) REGARDLESS OF THE VALUE OF THE PUTENTIAL V. THEN...

1= X30+ 55t

THE TIME truste FOR AN ELECTRON TO TRAVEL BETWEEN THE PLATES IS THEN.

OR topate = 15

AND THE TIME TYPEEN TO TRAVEL FROM THE END OF THE PLATES TO THE SCREEN IS.. $(L-\frac{1}{2}\ell)$ = No $(\frac{1}{2}x_{REEN})$

OR threen = L-1/2

NEXT CONSIDER THE VERTICAL MOTION OF AN ELECTRON AS IT MOVES BETWEEN THE PLATES.

$$\frac{1}{(E_{Y})_{PLATE}} = \frac{1}{100} \frac{1}{100}$$

THEN, FOR THE UNIFORMLY ACCELERATED MOTION IN THE Y DIRECTION HAVE $N_{\gamma} = (N_{\gamma}^{2})^{2} + \alpha_{\gamma}^{2} t \qquad \qquad \gamma = N_{\gamma}^{2} + (N_{\gamma}^{2})^{2} t + \frac{1}{2}\alpha_{\gamma}t^{2}$ $= (N_{\gamma}^{2})^{2} t \qquad \qquad \qquad = \frac{1}{2}(\frac{\alpha_{\gamma}}{\alpha_{\gamma}})^{2}t^{2}$

At the END OF the Plates... $(LC_1)_0 = (\frac{QV}{md})(\frac{1}{U_0}) \qquad H_0 = \frac{1}{2}(\frac{QV}{md})(\frac{Q}{G_0})^2$ $= \frac{QVQ}{mdU_0} \qquad = \frac{QVQ^2}{2mdU_0^2}$

12.63 FIND: & IN TERMS OF V. 150, R, M, d. P. L

FIRST NOTE THAT THE VELOCITY OF AN ELECTRON IS CONSTANT AFTER IT LEAVES THE PLATES.



THEN, FROM THE END OF THE PLATES TO THE SCREEN ...

 $f(x) = A = A + (x^2)^2 + (\frac{\delta A}{\delta x}) + (\frac{\delta A}{\delta x}) + (\frac{\delta A}{\delta x})$

At the screen: $\delta = \frac{eVl^2}{2mdv_o}t + (\frac{eVl}{mdv_o})(\frac{L - \frac{1}{2}l}{v_o})$

OR S= QVIL MdJ.s (CONTINUED) 12.64 continued

12.64 GIVEN: AT X=1, (= -4) = 0.05d FIND: () MIN IN TERMS OF 0, M, JS, V

2 --- T AT X= 1, HAVE
Y= Y1 = 2 Md x 2.

OR 0.45d 2 eV/2 OR 0.45d 2 eV/2 2 mdv2 0.9 ms2

THE MINIMUM VALUE OF $\frac{d}{2}$ is THEN $\left(\frac{d}{2}\right)_{\text{min}} = \left(\frac{eV}{2 \text{mis}_{2}^{2}}\right)^{1/2}$

12.65

GIVEN: $J_{x} = J_{0}$ (= CONSTANT);

Cathode

Screen

Anode

Screen

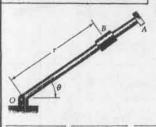
A J_{0}

From the solution to Problem 12.43 HAVE $8 = \frac{\text{eVVL}}{\text{mdJ}_{s}^{2}}$ Then the Solution to Problem 12.43 HAVE

THEN, SINCE δ is unchanged, have $(\delta)_{\text{MODIFIED}} = (\delta)_{\text{ORIGINAL}} : \frac{\text{QVI'L'}}{\text{md'} \text{s.}^{\text{c}}} = \frac{\text{QVIL}}{\text{md'} \text{s.}^{\text{c}}}$ OR $\frac{\text{I'}(0.6\text{L})}{0.8\text{d}} = \frac{\text{IL}}{\text{d}}$

OR 1'=1.3331

12.66 and 12.67



GIVEN: mg = 0.2 kg; r = 250+150 SINTE, θ = T(4t2-8t) r~mm, t~s, θ~RAB

HAVE. $\Gamma = (0.25 + 0.15 \sin \pi t) m$ $\theta = \pi (4t^2 - 8t) RAV$ THEN $\dot{\Gamma} = (0.15\pi \cos \pi t) \frac{m}{2}$ $\dot{\theta} = \pi (8t - 8) \frac{\cos \pi t}{2}$ AND $\ddot{\Gamma} = -(0.15\pi^2 \sin \pi t) \frac{m}{2}$ $\dot{\theta} = 8\pi \frac{\cos \pi t}{2}$

12.66 FIND: (a) Fr AND FO AT t=0.55

(a) AT t=0: T=0.25m : 0.15n = 0

0 = - 817 848 0 = 817 848

12.66 and 12.67 continued

Now. a==--102 = 0-10.25 m)(-BT RAD)2=-16TT2 5 AND Q = (B+2+ B = (0.25 m)(81) (81) + 2(0,151) (5) (-81) (84) = T1(2-2.4T1) ==

OR F= = - 31.6 N .

B=mag=(0.2kg)[T1(2-2.4T1) =] OR FB = - 3.48 N

(b) AT t=0.51: T=0.40 m

8 = - 4TI RAD r=0 "=-0.15T2 X B = BTI RAD

Now - ar = "-102 =- (0.1572 52) - (0.40 m) (-471 825)2 =-6.55 The The

AND Q = TB+2+B = (0.40m)(87 (32) + 0=3.2017 5

FINALLY -- Fr = mar = (0.2 tg)(-6.55 Ti 32)

OR F =- 12.93 N . Fa = maa = (0.2 kg)(3.2017 TE)

OR FB = 2.01 N

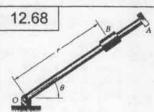
12.67 FIND: FO AND FO AT t=1.55

AT t=1.55: F=0.10m B = 4TT RAB L = 0. B = BTT SAB " = 0,15 TZ 32 Now - a===++++ = (0.15 n2 =) - (0.10 m)(472 RAD)2

00 = 18+2+8 = (0.10 m)(81 = 1+0 = 0.811 5

FINALLY -- Fr= mar = (02 kg)(-1.45 72 52) OR F == - Z. 86 N F8 = mag = (0.2 kg)(0.8 17 32)

OR FA . 0.503 N



GIVEN: WB = 516;

raft tos O-RAD FIND: (a) Fr AND FA AT t=15

Lb) F AND FA AT to 65

HAVE .. (= 10 St 0 = (A SINAT) RAD THEN .. 1 = - (+4) = 5 B=(zcosnt) & (+4)3 ft B = - (27 SINAT) RAD dus

(a) AT t=15: r=2 St 下=-0.4 装 8 = -2 840 = 0.16 the Now. ar = "-182 = (0.16 50) - (2 ft)(-2 25)2 =-7.84 \$4/52

a= +2+0= 0+2(-0.4 5)(-2 3)=1.6 5

FINALLY .. Fr= ma ar = 522 4/52 (-7.84 52)

OR F =- 1.217 16 4 Fo = MB ao = 516 (1.6 452) OR F8 = 0. 248 16

(CONTINUES)

12.68 continued

(b) AT t= 65: F= 1 St ドョーロー生 0 = 2 T " = 0.02 E B = 0

Now. a== =- r62 = (0.02 \$) - (14X2 \$40)2 =- 3.98 \$ AND Q= (B+ 2 + B = 0 + 2(-0.1 +)(2 PAB) = -0.4 \$=

FINALLY .. Fr = Waar = 32.25432 (-3.98 52)

DR F =- 0.618 16 Fo = Mo ab = 516 (-0.4 52)

DR F =-0.0621 16

12.69

GIVEN: B = Ct C = CONSTANT, + =- k; AT t=0, F= F0

FIND: (a) T IN TERMS OF m, c, k, 5, t

> (b) Q, FORCE EXERTED ON B

BY ARM AA'

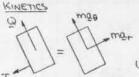
KINEMATICS

HAVE . OF = = - k AT t=0, r. Ts: 15 dr= 13-kdt OR T=To-kt

ALSO .. " = 0

Now. ar= "-r= = 0-(ro-kt)(ct)=-c2(ro-kt)t2 a===0+2+0=(5-kt)(c)+2(-kxct)

= c(10-3kt)



(a) to EF= mar: -T=m[-c/r-kt)t] OR T=mc2(r-kt)t2

(b) AEFo= mag: Q= m[c(ro-3kt)]

OR Q=mc(ro-3kt)

12.70

GIVEN: MB = 3 kg; 8=0.75t 1:0.5 % AT 1:0

FIND: L WHEN T=Q. WHERE Q IS THE

FORCE ON B FROM AA

KINEMATICS

#= r= 0.5 g HAVE

AT t=0, r=0: | dr = | to.sdt OR 1=(0.5t) m

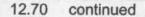
ALSO .. " = 0

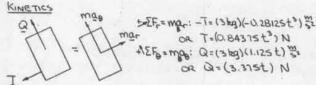
0 = (0.752) RAD 0 = 0.75 RAD

Now. Qr= "-18 = 0- ((0.5t)m) ((0.75t) 846)2

=-(0.28125t3) 52 ap = r0+zr0=[(0.5t)m](0.75 32) AND

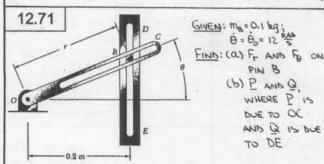
+2(0.5 8) (10.75t) PM) = (1.125 t) 52





NOW REQUIRE THAT . T= Q OR (0.84375t3)N= (3.375t)N DR t2= 4.000

OR t= 2.00 s



KINEMATICS

FROM THE DRAWING OF THE SYSTEM HAVE... $C = \frac{O.2}{\cos \theta} \text{ M}$

D = 12 RAB

THEN += (0.2 SIND 0) 5

"= 0.2 0000(0000)-5100(-20000 mB) AZ

= (0.2 T+ SINZA 02) m

SUBSTITUTING FOR 0 ... F = 0.2 - SIND (12) = (2.4 - SIND) M " = 0.2 (15) (12) = (28,8 1+5) 10) 10 = " = "

Now. $\alpha_r = r - r\dot{\theta}^2 = (28.8 \frac{1 + \sin^2 \theta}{\cos^2 \theta}) - (\frac{0.2}{\cos \theta})(12)^2$

a= 10+210 = 0+2(2.4 = 10)(12) = (57.6 SIND) m

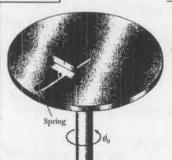
(a) HAVE .. Fr = mgar = (0.1 kg)[(57.6 cos = 0) =] AND FB = MBQB = (0.1 kg)[(57.6 \(\frac{51NB}{52}\)] \(\frac{52}{52}\)
OR FB = (5.76 N) THUB SECB

NOW .. + EF4: FA COSB + FF SINB = P COSB OR P = 5.76 TAND SECB+(5.76 TAND SECB) TAND OR P=(S.76N)THUBSER'S YB (CONTINUED)

12.71 continued

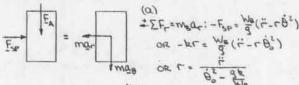
= EF: Fr = Q cost OR Q = (5.76 TAN & SECO) COSB OR Q = (S.76N)TAN'S SEC'B -

12.72



GIVEN: 0 = 15 84 WB = 0.5 16 K=4 16t; WHEN F=0, X=p=0; F=-40 84/58 FA = 216 FIND: (a) F (b) No

FIRST NOTE ... WHEN FOO XSPED => FSPERT 0 . 0 = 15 BAD THEN 9 = 0

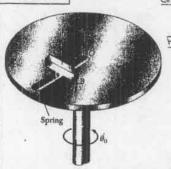


-40 845 (15 RAD)2- (32.2 4452)(4 1648)

(b) +12F6 = map: Fx = Wa (+8+2+0) Now .. No = r 2 fx = (32.2 52)(21b) 2 WB B = 2(0.51b)(15 RAB)

OR N=4.29 5

12.73



GIVEN: 8 = 12 5 WB = 8,05 04; WHEN TED, XLP +O; AT 1=0, F=0, F= 15 IN. FIND: (a) T AND FA AT \$=0.15

(b) T AND F AT 1-0.15 R=3.25 16/81 WHERE FA IS THE HORIZONTAL FORCE EXERTED ON THE SLIBER BY THE DISK

FIRST NOTE --WHEN (=0, XSP = 0 => FSP = RT 8 = 80 = 12 RAD To = 15 IN. = 1.25 ft and 0 = 0 THEN

+ΣF=mga=: - F== = = = ("-- - b)2) OR "+(\frac{kg}{vh} - \theta^2) \(\text{r} = 0 \) +1 EFg=meap: Fx = (1842+2+3) (2)

12.73 continued

(a) k = 2.25 16/st

Substituting the given values into EQ (1).. $\ddot{r} + \left[\frac{2.25^{6}/4! \times 32.2^{47} s^{2}}{8.05^{6}} - \left(12 \frac{Rab}{5} \right)^{2} \right] r = 0$

THEN .. # = " = 0 AND AT t=0, +0:

0=70

'OR

:. r=1.25 ft ◀

NOTE: FOO IMPLIES THAT THE SLIBER REMAINS AT 175 INITIAL RABIAL POSITION.

WITH F=O, EQ. (2) IMPLIES

F4=0

(b) k=3.25 16/ft

SUBSTITUTING THE GIVEN VALUES INTO EQ (1) ...

" + [3.25 | 1/24 - 32.2 | 1/2 - (12 = 20) 2] [= 0

OR ++645=0 ·= 5 # = # 3 = 5 = 5 Now .. " = \$ (+)

THEN "= 15- dist

SO THAT No 95 + 645 = 0 AT t=0, 5=0, C=6: 105-d5= -64/6-de

OR 15- = B 152-FE Now - 12 = # = 8/23-65 AT t=0, r=r: [di 150-r= =] + 8dt - 8t

(=10 214) CCOSAGA - OF LET

12 125-125211150 = 8F

18 = 8t OR

OR SINT (=) - P = 8t

 $r = r_0 \sin(8t + \frac{n}{2}) = r_0 \cos 8t = (1.25 \pm 1) \cos 8t$ THEN ==-(10 5) SIN BE

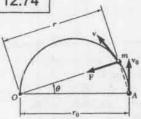
FINALLY .. AT t= 0.15:

T= (1.25 ft) cos (8x0.1)

OR F= 0.871 ft EQ(2).. FA = (8.05/16)16 + 2. [-(10 =) SIN (8x0.1)](12 = 0.81)

OR Fx=-2.69 16

12.74



GIVEN: CENTRAL FORCE F AND SEMICIRCULAR PATH SHOWN; AT too, 8=0, 5-5 SHOW: N= 15/cos 0

HAVE - 12 - LEL+LBEB SO THAT NE - + 2 + PE BE FROM THE DIAGRAM. T= 5 COS B ((Buis 2) - = (C SINB) SUBSTITUTING INTO EQ (1).

15 = (-100 SINB) + (15 COSB) & = 12 (SINZ 8 + COS 8) BE

OR 5= 50

AT t=0: 15= 100 L5B = C5B FROM Ea. (12.27):

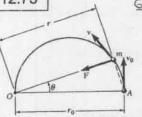
OR $\dot{\theta} = \frac{r_0 Z_0}{r_0 Z_0} = \frac{r_0 Z_0}{r_0 Z_0} = \frac{r_0}{r_0} \frac{r_0}{r_0}$

SUBSTITUTING FOR B IN EQ. (2) ..

2= 50 (10000)

N = COSE B Q.E.D. 02

12.75



GIVEN: CENTRAL FORCE F AND SEMICIRLULAR PATH O=+ TA ; MWOHE. 8=0 N=Nº ' L = Cº FIND: (a) & WHEN 8=0 (b) FE WHEN 8.45°

r= 5 cos 8 FROM THE DIAGRAM ..

BIBNIE TI-= " WITHT Now .. 5= + 2-+ + 8 20

SO THAT AT t=0 .. FROM EQ. (12.27): r20= 526

OR $\dot{\theta} = \frac{r_0 N_0}{r^2} = \frac{r_0 N_0}{(r_0 \cos \theta)^2} = \frac{N_0}{r_0} \frac{r_0 N_0}{\cos^2 \theta}$

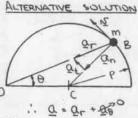
FROM PROBLEM 12.74:

Now .. Q = dt = dt (wo) = No 2 cos4 B = 250 SINB (10 (10 (10 (10) = 2 No SINB

FINALLY.. Ft = mat = 2m 5 cos 5

(a) WHEN 8=0 (b) WHEN 8=45°: F=2m= 5 5145° F=0 OR FL . 8m (CONTINUED)

12.75 continued



FIRST NOTE THAT TRIANGLE OBC IS AN ISOSCELES TRIANGLE.

: 4 OBC = 8 FOR CENTRAL FORCE ag = 0 MOTTON

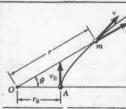
OR at+an=ar Now - a = at+an FROM THE ABOVE DIAGRAMI ...

at = an TAND P= 50 WHERE On = 15

AND FROM PROBLEM 12.74

5 = 50 COS 28 $Qf = \frac{(\sqrt{2}\sqrt{\cos_2\theta})_s}{(\sqrt{2}\cos_2\theta)_s} \times \frac{\sin\theta}{\sin\theta} = S\frac{2}{\sqrt{2}}\frac{2}{\cos\theta}\frac{\sin\theta}{\sin\theta}$ FINALLY -- Ft= mat = 2m 15 SINB (AS ABOVE)

12.76 and 12.77



GIVEN: CENTRAL FORCE F AND PATH SHOWN; C= 5 TA : 0500 V =7 8=0, N=No, T= To

HAVE T= (cos 20) = 6 (cos 20) } THEN += 10 [-1 (COSZB)] (-251MZB) = Fo SINZA B

Now .. N= re+ + r8 ep SO THAT AT t=0.. 5= 500 FROM EQ. (12.27): 120 = 520. OR B = \(\frac{12}{\chi_0 \chi_0} = \frac{(\frac{12}{\chi_0})^2}{(\frac{12}{\chi_0})^2} = \frac{2}{\chi_0} \cos 2\text{B}

12.76 FIND: No AND NO AS FUNCTIONS OF 8

HAVE. No = 1 = 10 COSTESD x 15 COS 28 OR NE = NO TOOSED

AND No = 10 = 100 100 20

OR 15 = No 1 cos 20

(CONTINUED)

SHOW: (a) SAT AND FAT

(a) FROM THE SOLUTION TO PROBLEM 12.76 HAVE NT = No VCOSZB No = No VCO=20

12.77 continued

Now.. $N_{z} = N_{z}^{2} + N_{\theta}^{2}$ $= (N_{\theta} \frac{(\cos 5\theta)}{(\cos 5\theta)})_{z} + (N_{\theta} \sqrt{\cos 5\theta})_{z}$ $= N_{\theta}^{2} + N_{\theta}^{2} + N_{\theta}^{2}$ 02 N= 150520

RECALLING THAT T' TOURS

IT FOLLOWS THAT IS= 100 DR WAT Q.E.D.

NOW - F = NT = No 514 20 AND F = TO TOOS 28

COMBINING - + = 5 FSIN 28

THEN - "= # (12 r sin 20) = 12 [+ sin 20 + + (2 cos 20) 6]

NOTING THAT $\vec{r} = \vec{h} = \vec{$ = = (1+ cos 20) =

Now - ar = 1 - 182 &= 5 cos28 (FROM ABOVE) = = (1+00520)7-7(= 00+20)2

FINALLY .. F = FF+ FB AND FOR CENTRAL FORCE MIOTION, ED = O. THEN -.

F = Fr = mar = m 50 7

OR FAT Q.E.b. (b) FIRST NOTE .. N= 50 (PART a)

AND QB = O (CENTRAL FORCE MOTION)

· 是· 是· 是· 是· 是· = 15 (15 (SIN 28) (FROM PART a)

= No L SINSA HAVE .. $a^2 = a_t^2 + a_n^2 = a_r^2 + a_0^2$ $a_r = \frac{s_0^2}{r_0^2} r$ (Aug a)

THAT $Q_n^2 = \left(\frac{\sqrt{c^2}}{\sqrt{c^2}}r^2\right)^2 - \left(\frac{\sqrt{c^2}}{\sqrt{c^2}}r^2\right) \times \frac{\sqrt{c^2}}{\sqrt{c^2}}$ $= \frac{\sqrt{c^2}}{\sqrt{c^2}}r^2\cos^2 2\theta \qquad r = \frac{\sqrt{c^2}}{\sqrt{c^2}}$

OR Q = 50

FINALLY... $a_n = \frac{u^2}{\rho}$ $u = \frac{u^2}{\rho}$ (FROM PART a)

OR $\frac{u^2}{\rho} = \frac{(\frac{u^2}{\rho}r)^2}{\rho}$

OR P = 12 13

OR PARS Q.E.D.

12.78 GIVEN: A PLANET OF RADIUS R AND OF DENSITY P; MOON HAVING ORBITAL RABIUS F= 2R SHOW: T= (24716P)12 HAVE .. F=G Mm [Ea. (12.281) AND F=Fn=man=m= CMm = m E OR Nº = GM FOR THE PLANET .. M=PV (47183) ひょう き(らまりなる)= まかのちゃ THE TIME I FOR THE MOON TO COMPLETE ONE FULL REVOLUTION IS T = 277 = 277 (376 P =)-2 = 120 (=)3/5 FOR r= 2R .. I = (3/7) (2R) 1/2 OR T= (ZATT Q.E.D. 12.79 GIVEN: A PLANET OF RABIUS R HAVING AN ACCELERATION OF GRAVITY 9 AT ITS SURFACE; T, THE ORBITAL PERIOD OF A MOON SHOW: r=f(R,g,E), WHERE F IS THE ORBITAL RABIUS OF THE MOON FIND: 9 FOR JUPITER; R=71492 km. TEUROPA = 3.551 DAYS, TEUROPA = 670.9 HO3 km HAVE - F = G Mm [Ea. (12.28)] AND F= Fn = Man = ME THEN GME : WE OR 52 GM SO THAT 52 = 982 EQ. (12.30) OR N= R 19 FOR ONE ORBIT .. T = 2TT = 2TT OR F= (\frac{9\tau^2}{4\tau^2}) 13

HAVE .. $F = G \frac{Mm}{F^2}$ [EQ. (12.28)]

AND $F = F_n = MQ_n = M \frac{S^2}{F^2}$ THEN $G \frac{Mm}{C^2} = m \frac{S^2}{F^2}$ OR $J^2 = \frac{GM}{GM}$ NOW $GM = gR^2$ SO THAT $J^2 = \frac{gR^2}{F^2}$ OR $J^2 = \frac{2\pi r}{F^2} = \frac{2\pi r}{R} = \frac{2\pi r}{R} = \frac{2\pi r}{R} = \frac{g}{R} = \frac{2\pi r}{R} = \frac{2\pi r}{R} = \frac{g}{R} = \frac{g}{R}$

12.80 GIVEN: SATELLITE IN A GEOSTNICHRONOUS EARTH ORBIT : T = 23.934 h FIND: (a) ALTITUDE IN OF THE SATELLITE (b) VELIXITY IS OF THE SATELLITE FIRST NOTE .. T= 23.934 h = 86, 1624+10 5 AND REARTH = 3960 mi = 20,9088 x10 St REARTH = 6.37×10 M (a) FROM THE SULUTION TO PROBLEM r= (322 R2)1/3 Now .. h = r-R THEN .. SI: H= [9.81 = (86.1624 x10 5) = (6.37 x10 m)2 113 - 6.37x10 W = (42.145-6.37) x10 m OR h=35,7740 km h=[32.2 \$ ~ (86.1624 no 3) 2 (20.9088 no H) 1) 12 O1 880P.05 -= (138.3343-20.9088) =10 ft OR h=22,240 mi (b) HAVE .. 5 = 271 THEN .. SI: 15=27 42.145 x10 m 86.1624 x10 s OR 15=3070 5 U.S. UNITS: 15= 27 138.3343 x10 ct OR 5=10,090 \$ 12.81 GIVEN: FMOON = 238, 910 mi, TMOON = 27.32 DAYS FIND: MASS M OF THE EARTH HAVE .. F = 6 TE [Ea. (12.28)] F = Fn = man = m + THEN GMM = M MTZ OR M = = 552 NOW .. No ET 50 THAT M= [(2717) = 1 (277) 2 = M NOTING THAT T = 27.32 DAYS = 2.3604 x10 5 T = 238, 910 mi = 1. 26144 x109 ft HAVE .. M = 1 (271 (2.3604 x 10 5) (1.261 44 x 13 ft) 3

OR M= 413×1021 16.52

12.84 12.82 GIVEN: FOR THE MOONS JULIET AND TITANIA OF GIVEN: ALTITUDE h= 380 km OF SPACECRAFT IN ORBIT ABOUT MARS; PMARS=3.94 Mg/m3 CRANUS, ZJ = 0,4931 DAYS. TT = 8.706 DAYS, 5 = 49,000 mi RMARS = 3397 km FIND: (a) MASS M OF URANUS FIND: (a) TIME I OF ONE ORBIT (P) C (b) VEWATY IS OF THE SPACECRAFT HAVE. F. G FT (a) FROM THE SOLUTION TO [ED. (12.28)] AND F=Fn=man=mx PROBLEM 12.78 HAVE THEN G MM = M 52 I = (F) 3/2 OR M= ENL WHERE T= R+h = (3397+380)km Now N = ZTIC = 3777 km SO THAT M. E(=)2 = E(=)20 T=[(66.73x10" m3) (3.94x10" m3)] Now .. Is = 0.4931 DAYS = 42,604 5 AND 5 = 40,000 mi = 211.2 x10 ft (a) Using Ea. (1). = 7019.55 $M = \frac{1}{G} \left(\frac{2\pi}{T_2} \right)^2 \zeta^3 = \frac{1}{34.4 \times 10^9 \frac{114}{16.54}} \left(\frac{2\pi}{42,6045} \right)^2 (211.2 \times 10^6 \text{ ft})^2$ OR T=1 h ST MIN (b) HAVE - 5= 2717 OR M=5,96×10 1 54 (3777 x10 m) (b) REWRITING EQ (1) .. MG = T3 AND THEN TT . TS OR N= 3380 5 12.83 GIVEN: ALTITUDE hs = 3400 km of SATELLITE OR IT = (B.706 BAYS) = (40,000 mi) IN ORBIT ABOUT SATURN; UT = 24.45 PM; FOR MOON ATLAS, F= 137.64x103 km, DR F = 271.2+10 mi TATLAS = 0.6019 DAYS 12.85 FIND: (a) RADIUS R OF SATURN GIVEN: SPACECRAFT OF WEIGHT W= 1200 lb; (b) MASS M OF SATURN he = 2800 mi; Mmoon = 0,01230MEATH, RMOON = 1080 mi HAVE .. F= G Mm [EQ.(12.28)] ATLAS M FIND: (a) GRAVITATIONAL FORCE F ON THE SATELLITE AND F=Fn=man=m = SPACECRAFT, EARTH ORBIT THEN GMM = ME (b) Fm, TE = Im (c) 9moon OR GM = FIS2 Ea. (12.29): 9= GM FIRST NOTE THAT RE = 3960 mi THEN TE = RE+hE = (3960+2800) mi AND THEN GREETSE = 6760 mi (a) HAVE .. F = G MM EQ. (12.28) AUD GM = gR2 EQ. (12.29) he OR N= R/\$ (1) AND R19=11/1 = N5 (R+ h (2) Now .. T = ZTT = ZTT [USING EQ. (1)] THEN .. F = gR2 m = W(R)2 or $R\sqrt{g} = \frac{2\pi r^{3/2}}{T}$ (5) FOR THE EARTH ORBIT .. F = (1200 Ib)(3960 ml) OR F = 412 16 (a) Using Eas. (2) AND (3) .. RSATURN (SATURN = No (R+hs = ZTI TA 42 (6) FROM THE SOLUTION TO PROBLEM 12.81 HAVE M== = (=)2 -3 OR R = (271 /2 3/2)2- hs NOTING THAT $T_k = 0.6019 \text{ bays} = 52.0042 \times 10^3 \text{ s}$ HAVE. $R = \left[\frac{2\pi \left(137.64 \times 10^6 \text{ m} \right)^{312}}{(24.45 \times 10^5 \text{ m})(52.0042 \times 10^5 \text{ s})} \right]^2 - 3400 \times 10^3 \text{ m}$ Now.. TE = Tm => 2717 12 = 271 542 OR I'M = (MM)13 = (0.0130) (6760 mi) = 60. 273x10 m OR R=60,3x10 km Tm = 1560 mi (C) HAVE .- GM = gR2 Ea. (12.29) LA) FROM ABOVE - GM = CS2 SUBSTITUTING INTO EQ. (1) THEN - M = 5 1 = 5 (R+ho) = (24.45×103 m) (60.273×10 , 3400×103) m RE TE RM TIM OR M = 570x10 bg (CONTINUED)

12.85 continued

OR $g_{M} = \left(\frac{Re}{R_{M}}\right)^{2} \left(\frac{r_{M}}{r_{E}}\right)^{3} g_{E} = \left(\frac{Re}{R_{M}}\right)^{2} \left(\frac{M_{M}}{M_{E}}\right) g_{E}$ USING THE RESULTS OF PART (b). THEN... 9m = (3960 mi) (0.01230) (32.2 \$)

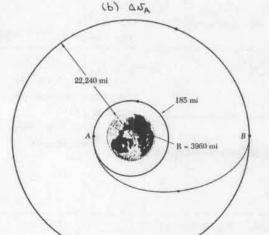
OR 9moon = 5.32 52

NOTE: 9 MOON = to GEARTH

12.86

GIVEN: CIRCULAR ORBITS AND ELLIPTIC TRANSFER DRBIT AB SHOWN;

AUS = 4810 A/s FIND: (a) (58)TR



FIRST NOTE .. R = 3960 mi = 20.908840 A T= (3960+185)mi = 4145 mi = 21.8856400 ft rg = (3960+22,240) mi=26,200 mi=138,336+10 ft (b) Now- (NSB)cac = (NB)rg + ΔLG

FOR A CIRCULAR ORBIT .. EF = Man: F= m =

EQ. (12.28): F=6 MM THEN G MM= M T OR 52 GM

Eq. (12.29): $GM = gR^2$ so THAT $U^2 = gR^2$

FOR A CIRCULAR ORBIT

THEN. (NA Lirc. = 32.2 Ft/s x (20.9088 x 10 ft)

OR (NA) CIAC = 25, 362 5

AND $(N_B)_{circ}^2 = \frac{32.2 \text{ ft/s}^2 \times (20.9088 \text{ no}^6 \text{ ft})^2}{138.336 \times \text{no}^6 \text{ ft}}$ OR $(N_B)_{circ}^2 = 10,088 \frac{\text{ft}}{5}$

(a) HAVE. (No bige = (No)TR + ONE

OR (48/18 = (10,088-4810) = 5278 \$ OR (NB) = 52805

(b) CONSERVATION OF ANGULAR MIDMENTUM REQUIRES THAT GM (NA)TR = GM (NB)TR

OR (NA) TR = 26,200 mi x 5278 \$\frac{5}{5}\$
= 33,362 \$\frac{5}{5}\$

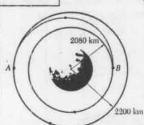
(CONTINUED)

12.86 continued

Now. (NA)TR = (NA) CIRC + DNA ANT = (33, 362 - 25, 362) \$ OR

OR ANT = 8000 5

12.87



GIVEN: CIRCULAR ORBITS ABOUT THE MOUN AND ELLIPTIC TRANSFER ORBIT AB AS SHOWN; DUS = - 26.3 5; mmoon = 73, 49 x 1021 kg

FIND: (a) (58) TR (b) aug

FOR A CROWLAR ORBIT. IF = Man: F= M FT

EQ. (12.28): F= G Mm

FL

THEN .. G Mm = m & OR WE GM

THEN .. (WA CIRC = 66.73×10 trays2 × 73.49×1021 bg

OR $(LS_A)_{C,RC} = 1493.0 \frac{18}{5}$ AND $(LS_B)_{C,RC}^2 = \frac{(66.73 \times 10^{28} \frac{189.52}{69.52} \times 73.49 \times 10^{21})}{20.900 \times 10^{2}}$

OR (NB)CIRC = 1535,5 5

(a) Have.. $(\sqrt{A})_{TR} = (\sqrt{A})_{CRC} + \Delta \sqrt{A} = (1493.0 - 26.3)^{\frac{10}{5}}$ = 1446.7 $\frac{10}{5}$ = 1466.7

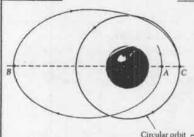
CONSERVATION OF ANGULAR MOMENTUM REGICIRES THAT TO MILETATE . TEM (NE)TE

OR (NB) = 2200 km x 1466.7 5 = 1551.3 5

OR (SB)TR = 1551 5 OR DUG = (1535.5-1551.3) "

DR 45=-1585

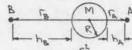
12.88



GIVEN: CIRCULAR ORBIT ABOUT VENUS AND ELLIPTIC TRANSFER ORBITS AB AND BC; E=6420 km; 5 + 7420 5 ha= 288 km DN = - 264 3; MVENUS = 4. BL9x10 ha

Ruenus: 6052 km Circular orbit FIND: (a) (NE) TRAB (b) hB

FIRST NOTE .. TA = R+ hA = (6052+288)km



(CONTINUED)

= 6340 km FOR A CIRCULAR ORBIT - EF = Man: F= ME Ea (12.28): F=G F

THEN ... G Mm = m 52 m OR 52 GM
THEN (AE) 2 E 6.73 × 10 12 m C × 4 869 × 10 2 kg
(C) 2 (AZO × 10 2 m)
(C) 2 (AZO × 10 2 m)

12.88 continued

(NE) CIRC = 7114.0 3 OR Now. (NE) CAR = (NE) TRAC + ANE

(NE) TRE = [7114.0 - (264)] = 7378.0 5

(a) CONSERVATION OF ANGULAR MOMENTUM REDUKES THAT .. AB: FAM(USA) = FBM(USA)TRAS

BC: 18 m (NB)TRAC = 12 m (NE)TRAC (2)

THEN (1) => TB (NB)TRBC TC (NE)TRBC (S(JE)TRAS TA (JA)

(NB)TREE = (NB)TRAR + ANB

(NB)TRAB + AND TO CHETTER THEN ... (NE)TRAB

OR (NB) TRAB = (NE) TRAB -1 (420 km) 7420 5 =3557.7 %

OR (NB) TRAB = 3560 5 (b) FROM EQ. (1) ...

FB = NS/ FA = 7420 M (15) TRAB = 1760 5 + 6340 km = 13 223 km

TB = R+hB Now. or hg = (13223-6052) km

0x he=7170 km

12.89

B ABOUT THE EARTH AND ELLIPTIC TRANSFER ORBITS BC AND CD; AUG = 280 \$, AUG=260 E = 4289 mi FIND: AND

FIRST NOTE __ R= 3960 mi = 20,9088MO St TA = (3960+380) mi = 4340 mi = 22.9152x10 Ft 18= (3960+180) mi = 4140 mi = 21.8592=10° ft FOR A CIRCULAR ORBIT .- IF = man: F=m F

THEN - G Mm = m 5

52 = GM = gR2 USING EQ. (12.29)

THEN - (NA) CIRC = 32.2 4/52 x (20,9086 x 10 ft)2

OR (15A) CIRC = 24, 785 4/3 (158) CIRC = 32.24/32 x (20.9088 MO AL)2 21.8592 NO ft AND

OR (UB) CIRC = 25,377 fys

HAVE .. (NB) TRBC = (NB) CIRC + DINB = (25,377+280) \$ = 25,657 445

(CONTINUED)

12.89 continued

CONSERVATION OF ANGULAR MOMENTUM REDURES THAT ...

BC: To m (Notrace = To m (Notrace CD: To m (Notrace = To m (Notrace

FIROM EQ (1).. (NE) + Rec = (15) + Rec = 4140 mi × 25,657 5

Now.. (NE) + (NE) + are + are = (24,766+260) \$

FROM EQ. (2) .. (50) TRED = TE (50) TRED = 4289 mi - 25,026 5 = 24,732 5

FINALLY. (NA) CIRC = (NO)TROB + AND OR AND = (24, 785-24, 732) 44/5 OR OND = 53 ths

12.90

GIVEN: 10=1, 0 = 16 5 (x==) = 0; k=2" HE: NEGLECT FRICTION AND

MADA; W=3 16 -18 in.-FIND: (a) (an), was (an) 6 in. --(b) (acounsises) (C) (NB)A

man GIVEN: CIRCULAR ORBITS A AND FIRST NOTE. FOR = K(r-17)

(a) F0 = 0 AND AT A, F= - F50 = 0

(QA) =0 (QA)8=0

(b) 1. EF = mar: - Fsp = m (i - r b)

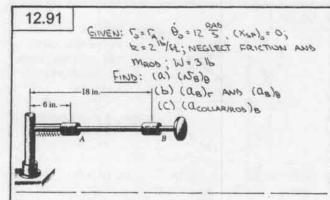
NOTING THAT acollarized = ", HAVE AT A ...

O = m[acollarized - (6 in.)(16 5)2]

OR (ACOLLAR/ROD)=1536 3 (C) AFTER THE CORD IS CUT, THE ONLY HORIZONTAL FORCE ACTING ON THE COLLAR IS DUE TO THE SPRING. THUS, ANGULAR MOMENTUM ABOUT THE SHAFT IS CONSERVED.

". TA MULA DE = TO M (NOB) WHERE (NA) = TA BO THEN - (NB) = 6 IN. [(6 IN.)(16 RAD)]

OR (158) = 32.0 5



FIRST NOTE .. FSP = k(r-r)

AT B: (FSP) = 2 Ft (18-6)IN. 151 = 2 16

Mag

Mag

ESP

Mac

(a) AFTER THE CORD IS CUT, THE ONLY HORIZONTAL FORCE ACTING ON THE COLLAR IS DUE TO THE SPRING. THUS, ANGULAR MOMENTUM ABOUT THE SHAFT IS CONSERVED.

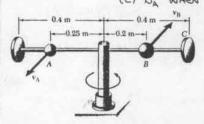
THEN: [A M (UA)] = [B M (UB)] WHERE (UA)] = [C M).

THEN: (NB) = [B IN.] [(6 IN.)(12 RAD)]

(b) HAVE... $F_0 = 0$ Now. $\stackrel{*}{=} \sum_{r} F_r = m\alpha_r : -(F_5 e)_g = \frac{1}{3}(\alpha_g)_r$ OR $(\alpha_g)_r = -\frac{2 \ln \pi}{3 \ln \pi} \times 32.2 \frac{1}{32} = -21.467 \frac{12}{32} = -257.6 \frac{10}{32}$

(C) HAVE.. $Q_{r} = \vec{r} - r \vec{\theta}^{2}$ NOW.. $Q_{r} = \vec{r} - r \vec{\theta}^{2}$ NOW.. $Q_{r} = \vec{r} - r \vec{\theta}^{2}$ THEN.. AT B: $(Q_{r} = r \vec{\theta}^{2} + r \vec{\theta}^{2})$ OR $(Q_{r} = r \vec{\theta}^{2} + r \vec{\theta}^{2})$ OR $(Q_{r} = r \vec{\theta}^{2} + r \vec{\theta}^{2} +$

12.92 GIVEN: $M_{A}=0.2 \, kg$, $M_{B}=0.4 \, kg$, $M_{ROD}=0$; $(S_{A})_{a}=2.5 \, g$; NEGLECT FRICTION; AT t=0, BALL B BEGINS TO MOVE FROM B TO CFIND: $(a) (a_{B})_{r}$ AND $(a_{B})_{B}$ AT t=0 $(b) (a_{B|ROD}$ AT t=0 $(c) S_{A}$ WHEN BALL B IS AT C



(a) WHEN THE PIN HOLDING BALL B IS REMOVED,
THERE ARE THEN NO HORIZONTAL FORCES
ACTING ON THE BALL! THEREFORE,
AT \$\frac{1}{2}\$, \quad \text{Fr}=0 \quad \text{AND} \quad \text{Fg}=0

(CONTINUED)

12.92 continued

SO THAT

(ag)=]=0 (ag)=)=0

(b) HAVE... $Q_{r} = (-r\dot{\theta}^2)^2$ NOW... $Q_{B|ROS} = (-r\dot{\theta}^2)^2$ THEN, AT t = 0... $(Q_{B|ROS})_0 - (r_B)_0 [(r_S)_0]^2 = 0$ OR $(Q_{B|ROS})_0 = 0.2 \text{ m} \cdot (\frac{2.5 \text{ m}}{0.25 \text{ m}})^2$

OR (QBIRDD) = 20.0 \$\frac{1}{2}\$ \(\)

(C) NOW, Fr = 0 AND FB = 0 WHILE B IS MOVING

FROM ITS INITIAL TO ITS FINAL POSITION. THUS,

ANGULAR MOMENTUM ABOUT THE SHAFT IS

CONSERVED. THUS...

TAMA (NTA) + (TO)MB (NOB) = TAMA NA + TO MB NOB

WHERE () BENOTES THE STATE WHEN BALL B IS

AT C. NOW...

 $(N_{\mathcal{B}})_{\circ} = (\Gamma_{\mathcal{B}})_{\circ} \stackrel{\circ}{\Theta} = (\Gamma_{\mathcal{B}})_{\circ} \left[\frac{(N_{\mathcal{B}})_{\circ}}{\Gamma_{\mathcal{B}}} \right]$ $AND \quad N_{\mathcal{B}} = \Gamma_{\mathcal{B}} \stackrel{\circ}{\Theta} = \Gamma_{\mathcal{B}} \left(\frac{N_{\mathcal{B}}}{\Gamma_{\mathcal{B}}} \right)$

THEN - $\Gamma_A M_A (J_A)_o + (C_0)_o M_B \left[\frac{\Gamma_A}{\Gamma_A} (J_A)_o \right] = \Gamma_A M_A J_A' + C_0' M_B \left(\frac{\Gamma_B}{\Gamma_A} J_A' \right)$ OR $\left\{ 1 + \frac{m_B}{m_A} \left[\frac{(\Gamma_B)_o}{\Gamma_A} \right]^2 \right\} (J_A)_o = \left\{ 1 + \frac{m_B}{m_A} \left(\frac{\Gamma_B'}{\Gamma_A} \right)^2 \right\} J_A'$

 $\left[1 + \frac{0.4 \text{ kg}}{0.2 \text{ kg}} \left(\frac{0.2 \text{ m}}{0.25 \text{ m}} \right)^2 \right] (2.5 \frac{\text{m}}{5}) = \left[1 + \frac{0.4 \text{ kg}}{0.2 \text{ kg}} \left(\frac{0.4 \text{ m}}{0.25 \text{ m}} \right)^2 \right] 454$

OR 5/ = 0,931 5

12.93

GIVEN: INITIAL STATE OF THE BALL

DEFINED BY 1, 8, FIND

THE FINAL STATE DEFINED

BY 12, 82

Find: (a) RELATION AMONG $\begin{cases}
l_1, \theta_1, l_2, \text{ AND } \theta_2 \\
l_2 = 0.6 \text{ M}, \theta_1 = 35^\circ
\end{cases}$

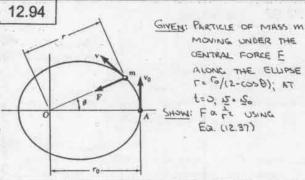
(a) For STATE 1 OR 2... $\frac{1}{1} \sum_{k=0}^{1} C_{k} = 0 : T_{k} = 0 = 0$ OR $T = \frac{mg}{\cos \theta}$ $\frac{1}{1} \sum_{k=0}^{1} C_{k} = ma_{n} : T_{k} = ma_{n} = ma_{n}^{2}$ WHERE $T = 1 \le \sin \theta$ Then $\left(\frac{mg}{\cos \theta}\right) \sin \theta = m \frac{s^{2}}{1 \sin \theta}$ OR $S^{2} = 91 \sin \theta \tan \theta$ It then Follows that $\frac{s_{2}^{2}}{s_{1}^{2}} = \frac{1}{1} \sin \theta : \tan \theta : \frac{s_{2}^{2}}{1 \sin \theta} : \cot \theta : \cot \theta : \frac{s_{2}^{2}}{1 \sin \theta} : \cot \theta : \cot$

Now. $\Sigma M_q = 0 \Rightarrow H_q = constant$ Thus.. $\Gamma_1 m S_1 = \Gamma_2 m S_2$

OR IS = RISINBI (2)

COMBINING EGS. (1) AND (2) - (\(\frac{l_1 \sin\theta_1}{\lambda_2 \sin\theta_2} \)^2 \(\frac{l_2 \sin\theta_2}{\lambda_1 \sin\theta_1} \), sin\theta_1 \(\frac{r_1}{\lambda_2} \sin\theta_1 \)

(b) HAVE... (O.B.m) 3 SIN 3 3 TAN 3 = (O.6m) 3 SIN 3 3 TAN 3 S = (O.6m) 3 SIN 3 3 TAN 3 S = (O.313197 OR 3 OR OR 3 OR 3



HAVE $\frac{d^2u}{d\theta^2} + u = \frac{F}{mh^2u^2}$ Eq. (12.37)

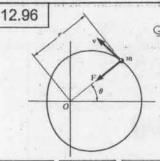
WHERE $u = \stackrel{?}{t}$ AND $mh^2 = constant$... $F \propto u^2 \left(\frac{d^2u}{d\theta^2} + u \right)$ NOW... $u = \stackrel{?}{t} = \frac{1}{t^2} (2 - cos \theta)$ THEN ... $\frac{du}{d\theta} = \frac{d\theta}{d\theta} \left[\frac{1}{t^2} (2 - cos \theta) \right] = \frac{1}{t^2} sin \theta$ AND $\frac{d^2u}{d\theta^2} = \frac{1}{t^2} cos \theta$ THEN ... $F \propto (\frac{1}{t^2})^2 \left[(\frac{1}{t^2} cos \theta) + \frac{1}{t^2} (2 - cos \theta) \right] = \frac{7}{t^2} \frac{1}{t^2}$ NOTE: F > 0 IMPLIES THAT $F = \frac{1}{t^2} = \frac{1}{t^2} \frac{1}{t^2} = \frac{1}{t^2} \frac{1}{t^2} \frac{1}{t^2} = \frac{1}{t^2} \frac{1}{t^2$

12.95 GIVEN: PARTICLE OF MASS IN MOVING UNDER A CENTRAL FORCE E ALONG THE PATH (= 65 IN 8 SHOW: F & 75 USING Ed. (12.37)

HAVE $\frac{d^2u}{d\theta^2} + u = \frac{F}{mh^2u^2} \qquad EQ. (12.37)$ WHERE $u = \frac{F}{mh^2u^2} \qquad EQ. (12.37)$ $\therefore F \times u^2 \left(\frac{d^2u}{d\theta^2} + u \right)$ $\therefore F \times u^2 \left(\frac{d^2u}{d\theta^2} + u \right)$ $\text{THEN --} \qquad \frac{du}{d\theta} = \frac{1}{d\theta} \left(\frac{1}{C \sin \theta} \right) = -\frac{1}{C} \frac{\cos \theta}{\sin^2 \theta}$ $\text{AND} \qquad \frac{d^2u}{d\theta^2} = -\frac{1}{C} \left[\frac{-\sin \theta(\sin^2 \theta) - \cos \theta(2 \sin \theta \cos \theta)}{\sin^4 \theta} \right]$ $= \frac{1}{C} \frac{1 + \cos^2 \theta}{\sin^3 \theta}$

THEN -. $F \propto (\frac{1}{r})^2 \left(\frac{1}{r} \frac{1 + \cos^2 \theta}{\sin^3 \theta} + \frac{1}{r^2 \sin^3 \theta} \right)$ = $\frac{1}{r^2} \frac{1}{r^2} \left(\frac{1 + \cos^3 \theta}{\sin^3 \theta} + \frac{\sin^3 \theta}{\sin^3 \theta} \right)$ = $\frac{2}{r^2} \frac{1}{r^2} \frac{1}{\sin^3 \theta} + \frac{\sin^3 \theta}{\sin^3 \theta} = (\frac{r}{r^2})^3$ = $\frac{2r^2}{r^2}$: $F \propto \frac{1}{r^2} \leq G.E.b.$

Note: FOO IMPLIES THAT E IS ATTRACTIVE.



GIVEN: PARTICLE OF MASS

M MOVING UNDER

THE CENTRAL FORCE

E ALONG THE

CARDIOID $\Gamma = \frac{r_0}{2}(1+\cos\theta)$ SHOW: F.G. $\Gamma = \cos\theta$ EQ. (12.37)

HAVE $\frac{d^2u}{d\theta^2} + u = \frac{F}{mh^2u^2}$ Eq. (12.37)

WHERE $u = \frac{1}{h}$ AND $mh^2 = constant$ $\therefore F \propto u^2 \left(\frac{d^2u}{d\theta^2} + u \right)$ NOW - $u = \frac{1}{h} = \frac{2}{h} \frac{1}{14 \cos \theta}$

THEN - $\frac{du}{d\theta} = \frac{d}{d\theta} \left(\frac{z}{16} \frac{1}{1+\cos\theta} \right) = \frac{z}{16} \frac{\sin\theta}{(1+\cos\theta)^2}$ AND $\frac{d^2u}{d\theta^2} = \frac{z}{16} \frac{\cos\theta(1+\cos\theta)^2 - \sin\theta[z(1+\cos\theta)](-\sin\theta)}{(1+\cos\theta)^4}$ $= \frac{z}{16} \frac{1+\cos\theta + \sin^2\theta}{(1+\cos\theta)^2} = \frac{z}{16} \frac{1}{(1+\cos\theta)^2} \frac{\cos\theta}{(1+\cos\theta)^2}$

 $\begin{aligned}
& = \frac{1}{C} \frac{(1 + \cos \theta)^{2}}{(1 + \cos \theta)^{2}} = \frac{2}{C} \left(\frac{1}{C} (\frac{1}{C} - 1) \right) \\
& = \frac{2}{C} \frac{(1 + \cos \theta)^{2}}{(1 + \cos \theta)^{2}} = \frac{2}{C} \left(\frac{2C}{C} \right) \left[2 + \left(\frac{1}{C} - 1 \right) \right] \\
& = \frac{2}{C} \frac{(1 + \cos \theta + \sin^{2} \theta)}{(1 + \cos^{2} \theta)^{2}} = \frac{2}{C} \left[\frac{1}{C} (\frac{1}{C} - 1) \right]
\end{aligned}$ Then $F \times \left(\frac{1}{C} \right)^{2} \left[\frac{2C}{C} \left(3 - \frac{2C}{C} \right) + \frac{1}{C} \right] = \frac{3}{2} \frac{C}{C}$

NOTE: FOO IMPLIES THAT F IS ATTRACTIVE.

12.97

GIVEN: PARTICLE OF MASS M

MOVING UNDER THE

CENTRAL FORCE F

ALDNG THE PATH

T=G/VCOSZB; AT 1=0

SHOW: FXT USING EQ.(12.37)

HAVE $\frac{d^2u}{d\theta^2} + u = \frac{F}{mh^2u^2}$ Eq. (12.37) WHERE $u = \frac{1}{F}$ AND $mh^2 = constant$ $\therefore F \times u^2 \left(\frac{d^2u}{d\theta^2} + u\right)$

Now.. $u = \frac{1}{r} = \frac{1}{r^2} \sqrt{\cos 2\theta}$ THEN .. $\frac{d\theta}{d\theta} = \frac{d\theta}{d\theta} \left(\frac{1}{r^2} \sqrt{\cos 2\theta} - \frac{1}{r^2} \frac{3 \ln 2\theta}{\sqrt{\cos 2\theta}} \right) = \frac{1}{r^2} \frac{3 \ln 2\theta}{\sqrt{\cos 2\theta}} \sqrt{\cos 2\theta}$ $= -\frac{1}{r^2} \frac{1 + \cos^2 2\theta}{(\cos^2 2\theta - \cos^2 2\theta)} = -\frac{1}{r^2} \left(\frac{r^2}{r^2} \right)^3 \left[1 + \left(\frac{r^2}{r^2} \right)^4 \right]$ $= -\frac{r^3}{r^3} \left[1 + \left(\frac{r^2}{r^2} \right)^4 \right]$

THEN .. FX (+)2 \ - \frac{r^3}{5^4} \[1+ (\frac{r}{2})^4 \] + \frac{r}{7} = -\frac{r}{5^4}.

FX \ Q.E.D.

NOTE: FKO IMPLIES THAT E IS REPULSIVE.

12.98

GIVEN: PARABOLIC TRAJECTURY OF GALILEO SPACECRAFT ABOUT THE EARTH; MY ONE = SOUTITUA MUMINIM

FIND: NMAX

FIRST NOTE .. R= 6.37 10 m SO THAT 6= (6.37 -10+960 -103) m

NOW - JMAX = No

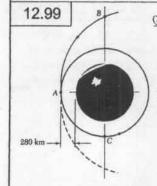
AND FROM PAGE TOP OF THE TEXT

No = [2GM = [29R2 USING EQ. (12.30)

THEN .. Nonax = [2 49.81 m/52 x (637 x 10 m)2]1/2

= 10 421.7 2

OR 5,00 = 10.42 5



GIVEN: TARABULIC APPROACH TRAJECTORY AND CIRCULAR ORBIT ABOUT VENUS; MUENUS= 4.87 HIS by R = 6052 km

FIND: (a) (NA) PAR (b) IDNAI

FIRST NOTE .. FA = (6052+280) km = 6332 km (a) FROM PAGE 709 OF THE TEXT, THE VELOCITY AT THE POINT OF CLOSEST APPROACH ON A PARABOLIC TRAJECTORY IS GIVEN BY

THUS, (NA) PAR = [2x66.73 x102 kg 52 x 4.87 x102 kg] 1/2 = 10 131.4 2

OR (U) PAR = 10.13 EM

(b) HAVE .. (NA) CIRC = (NA) PAR + ANA

Now. (NA) CIRC = (GM EQ. (12.44) = 1 (NA) and

THEN. DUTA = ((UTA) PAR - (UTA) PAR = (= 1) (10. 1314 Em) = -2.97 Em

": 125=1=2.97 Em

12,100

GIVEN: TRAJECTORY OF GALILEO SPICECRAFT ABOUT THE EMRTH; AT THE POINT OF CLOSEST APPROACH, IJ = 46.2410 \$ ALTITUDE = 188.3 mi

FIND : E AT POINT OF CLOSEST APPROACH

FIRST NOTE .. R = 3960 mi = 20,9088 mo # AND T = (3960+188.3) mi = 4148.3 mi = 21.9030×10 H

HAVE - + = GM (1+ E cos 8) Es. (12.39')

AT POINT O, F=Fo, 8=0, h=h=Fo. 50 ALSO.. GM=gR2 EQ. (12,30)

THEN - $\frac{1}{L^2} = \frac{gR^2}{gR^2}(1+\epsilon)$

OR E = \(\frac{6.5^2}{9.8^2} - 1 = \frac{(21.9030 no \text{ ft})(46.2 no \frac{6.5}{5})^2}{(32.2 \frac{6.5}{5.2})(20.9088 no \text{ ft})^2}

188.3 mi

12.101

GIVEN: TRAJECTORY OF GALILED SPACECRAFT ABOUT IO: AT THE POINT OF CLOSEST APPROACH (= 1750 mi)

N= 49.470 \$; MID: 0.01496 MENE FIND: E AT POINT OF CLOSEST AFFROACH

FIRST NOTE .. G= 1750 mi = 9.24×10 ft REASTH = 3960 mi . 20.908840" FL

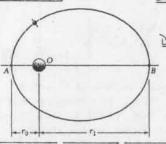
HAVE .. L = GM (1+ € 0058) Ea. (12.39) AT POINT O, r=10, B=0, h=h== 15 US

ALSO .. GM TO = G (0.01496 MEARTH) = 0.01496 9 REARTH USING ED (12.30) 1 = 0.014969 REARTH (1+E)

6 50° (9.24×10 CE)(49,4×10 = 12 OR E = (9.24 x10" EE)(49.4 x10" E) 1 -1

OR E=106.1

12.102



GIVEN: ELLIPTIC ORBIT OF A SATELLITE ABOUT A PLANET OF MASS M DERIVE: + + = ZGM

HAVE. $\frac{1}{h} = \frac{GM}{h^2} (1 + \epsilon \cos \theta)$ Ea. (12.39') Now... AT A: $f = f_0$ (1+ \epsilon) (1)

AT B: F= F, B= 180°: : + = GM (1-E)

THEN (1)+(2)=> ++ += EM[(1+E)+(1-E)]

OR = + = = 2GM Q.E.D.

12.103

GIVEN: ELLIPTIC AND CIRCULAR DRBITS OF THE SPACE SHUTTLE ABOUT THE EARTH , NO PERPENDICULAR



FIND: (a) 50 R = 3960 mi (b) Aug

FIRST NOTE .. 1 = (3960+ 40.3) mi . 4000.3 mi \$7 OIX 2151.15.= ra = (3960+336) mi = 4296 mi +22.6829x10 St R = 3960 mi = 20.9088 NO 4

(a) FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR THE ELLIPTIC DABIT AB ..

Now.. h=h= ~~~~~ THEN - 1/A + 1/B = 29 R2 (7,15)2 GM=gR2 [Ea.(12.30)] OR NS = R (29)

= 26,272 5

DR 15 = 26.3×10 \$ 1 (b) FOR THE ELLIPTIC ORBIT AB HAVE.

h= ha= ha: [ANS = [8 (NB)AB THEN .. (NB) AB = 4000.3 mi = 26,272 ft

= 24, 464 \$ FOR THE CIRCULAR ORBIT, USE EQ. (12.44) (128) CIRC = 18 = 50 9088 x10 ft (35.2 4/32

= 24 912 5

FINALLY .. (No) CIRC = (NO) AB + DISB OR AUB = (24,912-24,464)4/5 DR ANg = 448 5

12.104

GIVEN: A PLANET OF RADIUS R AND A SPACE PROBE IN A CIRCULAR DRBIT ABOUT THE PLANET AT AN ALTITUDE OR AND HAVING A VELOCITY So; ELLIPTIC ORBIT, WHERE 5= \$55.

FIND: AMIN SO THAT THE PROBE DOES NOT CRASH

FOR THE CIRCULAR ORBIT -- IS = (GM [EQ.(12.44)] WHERE TA = R+ aR = R(1+x) THEN .. GM = 52 R (1+0x)

(KEUNITHOS)

12,104 continued

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR THE ELLIPTIC ORBIT ...

Now. h= ha = Ta (Sh) AB = [R(1+a)] (\$15)

THEN --

 $\frac{1}{R(1+\alpha)} + \frac{1}{F_B} = \frac{2J_0^2 R(1+\alpha)}{[R(1+\alpha)\beta J_0]^2}$ B2R(1+0x)

NOW .- BMIN CORRESPONDS TO FB -> R. THEN ..

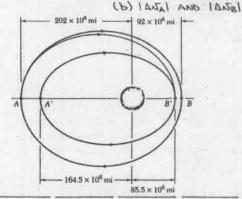
R(1+K) + R = B2R(1+K)

OR BMIN = 12+06

12.105

GIVEN: ELLIPTIC ORBITS AB AND A'B' OF A SPACECRAFT ABOUT THE SUN AND THE ELLIPTIC TRANSFER DRBIT AB; MSUN = (332.8 x103)M

FIND: (a) NA (ON AB)



FIRST NOTE .. REARTH = 3960 mi = 20,9088 x10 ft TA = 202 x 10 mi = 1066,56 x 109 ft TB = 92x10 mi = 485.76x109 ft

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR ANY ELLIPTIC ORBIT ABOUT THE SUN.

(a) FOR THE ELLIPTIC ORBIT AB HAVE ..

1.=14, 12=18, h= ha= 14 1/4 ALSO - GM SUN = G[(332.8×103) MEARTH]

= 9 REARTH (358.84103) USING EQ. (18.30)

THEN .. + + = = 29 REARTH (332.8×103)

OR UN = REARTH (CLS.69x103) 1/2

= 3960 mi (CLS.6x103 x 32.2 ft/s2

= 202x10 mi (1066.56x103 H + 485.76x109 H

= 52, 431 5

OR UT = 52.4×10 \$ (CONTINUED)

12.105 continued

(b) FROM PART (a) HAVE 2GM sun = (To No) (= + =)

THEN, FOR ANY OTHER ELLIPTIC ORBIT ABOUT

THE SUN HAVE -- (TA + TE)

FOR THE ELLIPTIC TRANSFER ORBIT AB HAVE .. 1=1A, 12=1B', h=hTR=1A(JA)TR

THEN .. 1 + 1 = ([[[[]] ([] + 1])] 2

= (52,431 \(\frac{4}{5}\)\(\frac{1+\frac{202}{92}}{1+\frac{202}{202}}\) = 51,113 5

NOW -- hTR = (ha)TR = (he')TR: (LA)TR = (B'(LB))TR

THEN (NB')TR = 202×10 MI x S1, 113 \$ = 120 758 #

FOR THE ELLIPTIC ORBIT A'B' HAVE ...

1 = 1/4, 12 = 18', h = 18' NB' THEN -- 1 + 1 = ((TANA) 2 (TA + (TE))2

= (52, 431 5) 202×10 mi (202×106 + 92×106) 85.5×10 mi (164.5×106 + 85.5×106)

= 116,862 \$

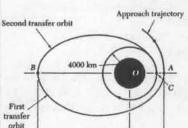
FINALLY .. (NA)TR = NA + ANA OR ANT = (51,113-52431) \$ OR 105/= 1318#

Ng = (Ng) /18 + DNB AND OR ANB' = (116, 862-120,758) \$ = - 3896 =

OR 1ANS 1= 3900 #

12.106

GIVEN: PARABOLIC APPROACH TRAJECTORY. ELLIPTIC TRANSFER ORBITS AB AND BC. AND CIRCULAR ORBIT OF A SPACE PROBE ABOUT MARS; TA = 9x10° km, TB = 180x10° km;



Approach trajectory MMARS = 0.1074 MEARTH
FIND: (Q) | ANA) (b) 10581 (C) 1 AJE1

(CONTINUED)

12,106 continued

(a) FOR THE PARABOLIC APPROACH TRAJECTORY, POINT A IS THE POINT OF CLOSEST APPROACH. THEN, FROM PAGE 709 OF THE TEXT HAVE

NOW- GMMARS = G(0.1074 MENRY) = 0.1074 g REARTH USING Ed. (12.30)

THEN .. (Na) PAR = REARTH (\frac{2x0.10749}{60.2148x9.81 m/s2})12
= (6.37x10 m) (\frac{0.2148x9.81 m/s2}{9x10 m})12 = 30823 3

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR ANY ELLIPTIC ORBIT ABOUT MARS...

1 + 1 = 2GMMARS (1)

FROM ABOVE .. ZGMMARS = TA [(WA) PAR] THEN .. FOR THE ELLIPTIC TRANSFER ORBIT AB ..

TA + TB = TA ((NA) PAR)2

WHERE has = (ha) as = Ta (Na) as THEN - + to = Ta (Na) one]?

DR (NA) AB = (NA) OAR (1 / 12) 1/2 (NA) OAR (1+ TA) = (3082.3 %) (1+ 9×103 km

= 3008.0 3

FINALLY .. (15) AB = (15) PAR + AUG OR AUG = (3008.0-3082.3) S

OR 105/1=74.3 5

(b) FOR THE ELLIPTIC TRANSFER ORBIT AB ..

 $h_{AB} = (h_A)_{AB} = (h_B)_{AB}$: $F_A(N_A)_{AB} = F_B(N_B)_{AB}$ $F_A = \frac{9 \times 10^3 \text{ km}}{180 \times 10^3 \text{ km}} \times 3008.0 \frac{\text{M}}{\text{M}}$ = 150.40 2

NOW APPLY EQ. (1) TO THE SECOND ELLIPTIC TRANSFER ORBIT BC AND USE

hBC = (B(NB)BC THEN - 1 + = Tal (Na) one)2

OR (NB)BC = (NA)DAR (TA + TE) = 3082.3 th (9×103 km + 4×103 km) = 101.62 %

FINALLY .- (158) BE = (158) AB + AUS OR AUS = (101.62-150.40) \$ OR 125/= 48.8 M

(C) FOR THE ELLIPTIC TRANSFER ORBIT BC ..

THEN. $(N_c)_{BC} = (h_c)_{BC} : \Gamma_B(N_B)_{BC} = \Gamma_C(N_c)_{BC}$ $\frac{1}{4 \times 10^3} \frac{1}{8} \times 101.62 \frac{m}{5}$ = 4572.9 7

FOR THE CIRCULAR ORBIT HAVE ..

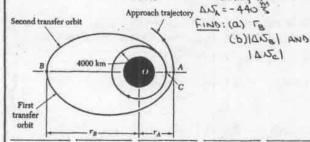
(Schore = GMMARS [EQ. (12.44)] (CONTINUES)

12.106 continued

RECALLING FROM PART (a) THAT $(N_h)_{pag} = \sqrt{\frac{2GM_{mass}}{G_h}}$ HANE $(N_c)_{circ} = (N_h)_{pag} \left(\frac{r_h}{2r_c}\right)^{1/2}$ $= (3082.3 \frac{m}{5}) \left(\frac{9 \times 10^3 \text{ km}}{2 \times 4 \times 10^3 \text{ km}}\right)^{1/2}$ $= 3269.3 \frac{m}{5}$ FINALLY ... $(N_c)_{circ} = (N_c)_{gc} + \Delta N_c$ OR $\Delta N_c = (3269.3 - 4572.9) \frac{m}{5}$ OR $|\Delta N_c| = 1304 \frac{m}{5}$

12.107

GIVEN: PARABOLIC APPROACH TRAJECTORY,
ELLIPTIC TRANSFER ORBITS AS AND
BC, AND CIRCULAR ORBIT OF A
SPACE PROBE ABOUT MARS;
MMARD = 0.1074 MEARTH; [4.9×103 km.,



(a) FOR THE PARABOLIC APPROACH TRAJECTORY, POINT A IS THE POINT OF CLOSEST APPROACH. THEN, FROM PAGE TOS OF THE TEXT HAVE

(UA) PAR = \[\frac{2GM_{MARS}}{G} \]

NOW -. GMMAS = G(0.1074 MEARTH) = 0.1074 gREARTH USING EQ.(12.30)

THEN - (JA) PAR = REARTH (2x0.10749)/2 = (6.37 × 10 m)(0.2148 x 9.81 m/s²)/2 = 3082.3 m/s

NOW - (15) AB = (15) PAR + ANS = (30823-440) \$ = 2642.3 \$

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR ANY ELLIPTIC ORBIT ABOUT MARS...

FROM ABOVE .. 26Mmas = Ta[(Na har) 2
THEN .. FOR THE ELLIPTIC TRANSFER ORBIT AB ..

WHERE has = $(h_m)_{aB} = T_a (J_a)_{AB}$ THEN - $\frac{1}{T_a} + \frac{1}{T_B} = \frac{T_a (J_a)_{AB}}{[T_a (J_a)_{AB}]^2}$ = $\left[\frac{(J_a)_{AB}}{J_a}\right]^2 \frac{1}{J_a}$

OR $\frac{1}{18} = \frac{1}{18} \left\{ \left[\frac{(\sqrt{3} n)_{MR}}{(\sqrt{3} n)_{MR}} \right]^{2} - 1 \right\} = \frac{1}{9 \times 10^{3} \, \text{km}} \left[\left(\frac{3082.3 \, \frac{10}{8}}{2042.3 \, \frac{10}{8}} \right)^{2} - 1 \right]$

OR TB = 24,946 x103 km OR TB = 24,9x103 km

(b) FOR THE ELLIPTIC TRANSFER ORBIT AB..

has = (ha)as = (hs)as : [(Ua)as = [(Us)as (CONTINUED)

12.107 continued

THEN -. (NB)AB = 24.946103 km + 2642.3 5

NOW APPLY EQ. (1) TO THE SECOND ELLIPTIC TRANSFER DRBIT BC AND USE

THEN ..
$$\frac{1}{r_B} + \frac{1}{r_C} = \frac{r_B (N_B)_{BC}}{(r_B (N_B)_{BC})^2}$$

OR $(N_B)_{BC} = \frac{(N_B)_{MB}}{r_B} \left(\frac{r_A}{r_B} + \frac{1}{r_C} \right)^{1/2}$

$$= \frac{3082.3 \frac{3}{5}}{24.946103 \text{ bm}} \left(\frac{9 \times 10^3 \text{ bm}}{24.946 \times 10^3 \text{ bm}} + \frac{1}{4 \times 10^5 \text{ bm}} \right)^{1/2}$$

= 688.2 %

THEN.. (NB) BC = (NB) AB + ANB OR ANB = (688.2-953.3) \$ OR |ANB | = 265 \$ \$

NOW.. FOR THE ELLIPTIC TRANSFER ORBIT BC..

THEC = (he)BC = (hc)BC: FB(UB)BC = FC(UE)BC

THEN.. (NE)BC = 24.946 MO' RM × 688.2 M

= 4292.0 M

= 4292.0 M

FOR THE CIRCULAR ORBIT HAVE ..

RECALLING FROM PART (a) THAT (
$$\sqrt{x}$$
) $= \sqrt{\frac{2GM_{MMRS}}{F_{A}}}$

$$(\sqrt{x})_{CRC} = (\sqrt{x})_{PAR} \left(\frac{x}{2T_{C}}\right)^{\frac{1}{2}} \left(\frac{3082.3 \frac{M}{S}}{5}\right) \left(\frac{2 \times 4 \times 10^{3} \text{ km}}{2 \times 4 \times 10^{3} \text{ km}}\right)^{\frac{1}{2}}$$

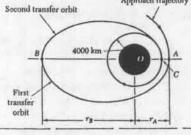
$$= 3269.3 \frac{M}{S}$$

FINALLY -- (NE) circ = (NE) BC + ANE OR ANE = (3269.3-4292.0) 5

OR 125 = 1053 5

12.108

GIVEN: ELLIPTIC TRANSFER ORBIT AB OF PROBLEM 12.106; To =9x10 km,
To = 180x10 km
Approach Inspectory FIND: LAB



FROM THE SOLUTION TO PROBLEM 12.106 HAVE

(UA)AB = 3008.0 \$\frac{18}{3}\$

FROM EQ. (12.45) IT FOLLOWS THAT

FROM EQ. (12.45) IT FOLLOWS THAT

tag = 2 (CELLIPSE) AB = TILL THAT

WHERE Q = \$\frac{2}{\Gamma_{A} + G_{B}} = \frac{2}{\chi}(9\xio^{3} + 180\xio^{3}) = 94.5\xio^{3}\text{ km}

AUD B = \sqrt{\Gamma_{B}} = \frac{2}{(9\xio^{3})(180\xio^{3})\frac{1}{\chi}} = 40.249\xio \text{km}

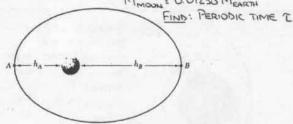
ALSO... has = \Gamma_{G}(\sub_{A})_{AB} = (9\xio^{2}\text{m})(3008.0\frac{\chi}{2}) = 27.072\xio \frac{\chi}{2}



tas = 7 (94.5 x10 m) (40.249x10 m)
27.072x109 m2 = 441.384 x10 5 OR tas= 122 h 36 MIN 245 TE = 83.0 x 106 mi

12,109

GIVEN: ELLIPTIC ORBIT OF THE CLEMENTINE SPACECRAFT ABOUT THE MOON; ha = 400 km, ha = 2940 km; RMOON = 1737 Em. MMOON = 0.01230 MEARTH



FIRST NOTE .. 1 = (1737+400) = 2137 km 13 = (1737 + 2940) = 4677 km

Now .. T = 217ab EQ. (12.45)

WHERE a= E(TA+TB) = 2(2137+4677) km = 3407. tem

b= Trata aug

FROM THE SOLUTION TO PROBLEM 12.102 HAVE .. The + Ta = ZGMMOON

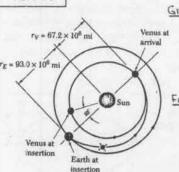
NOW - GMMOON = G(0.01230 MEARTH) = 0.01230 g READTH USING ED. (12.30)

THEN. LZ = 2(0.01230 9 REARTH) 0.01230 9 REARTH

= 52 (SXO. O1230 9 REARTH)

OR h = b REARTH (0.01230 9) 1/2 THEN .. T = 27106 2710 = 2710" | REARTH (0.0.12309) 1/2 = (6.37x10 m)(0.01230x9.81 m)52)112 = 17.8571x103 S OR T= 44 57 MIN 375

12.110



GIVEN: ORBITS OF VENUS AND THE EARTH AND THE ELLIPTIC TRANSFER ORBIT OF A SPACE PROBE: MSUN = 332 BRIOTH

FIND: \$, THE RELATIVE POSITION OF VENUS WITH RESPECT TO THE EARTH AT THE TIME OF INSERTION

FIRST DETERMINE THE TIME LARDRE FOR THE PROBE TO TRAVEL FROM THE EARTH TO VENUS. NOW .. tprose = 2 TTR

WHERE I'M IS THE PERIODIC TIME OF THE ELLIPTIC TRANSFER ORBIT. APPLYING KEPLER'S THIRD LAW TO THE ORBITS ABOUT THE SUN OF THE EARTH AND THE PROBE OBTAIN ...

ITR - atre TEARTH O EARTH

WHERE at = \$ ((=+ () = \$ (93 × 10 + 67.2 × 10) mi

aeARTH = PE (NOTE: EEARTH = 0,0167) THEN ...

 $t_{PROBE} = \frac{1}{2} \left(\frac{\alpha_{TR}}{T_E} \right)^{3/2} T_{EARTH}$ $= \frac{1}{2} \left(\frac{80.1 \times 10^6 \text{ mi}}{93.0 \times 10^6 \text{ mi}} \right)^{3/2} (365.25 \text{ DAYS})$ = 145,977 DAYS

= 12.6124 ×10 5

IN TIME tPROBE, VENUS TRAVELS THROUGH THE ANGLE BY GIVEN BY By = By t PROBE = Sy tPROBE

ASSUMING THAT THE ORBIT OF VENUS IS CIRCULAR (NOTE: EVENUS: 0.0068). THEN, FOR A CIRCULAR ORBIT ..

NV = J GMoun [EQ. (12.44)]

NOW. GMSUN = G(332.8×103 MENRTH)
= 332.8×103 (9 REARTH) USING EQ.(12.30)
THEN. By = \frac{tprobe}{V} \Big[\frac{332.8 \times 10 B REARTH}{V} \] \]

= trade REARTH (332.8 9x103) 1/2

REARTH = 3960 mi = 20.9088x10" (t C = 67.2x10 mi = 354.816x109 ft

THEN. Q = (12.6124×10 5)(20.908B×10 4) (3328×10 432242) (354. BILXIO9 41) 1/2

= 4,0845 RAD = 234.02°

FINALLY .. \$ = 0, - 180°

= 234.02 - 180

OR \$= 54.0°

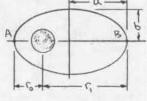
12.111

GIVEN: ELLIPTIC ORBIT ABOUT THE SUN OF THE COMET HYAKUTAKE; E: 0999887, TMIN = 0.230 RE; KE = QEARTH FOR THE EARTH'S ORBIT ABOUT THE

FIND : T FOR THE COMET

Using Eq. (12.39") HAVE FOR ANY ELLIPTIC ORBIT ABOUT THE SUN.

T = GWOON (1+ E COSB)



T = GM con (1+ E) (1)

AT B, 0=180: = GM www (1-E) (2)

FORMING (2) => 1/6 - 1-6 OR 1= 1-6 10

Now .. Q = 2 (10+1,) = 1/6 + 1/6 10) = 1-6

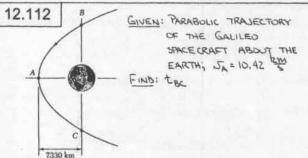
APPLYING KEPLER'S THIRD LAW TO THE ORBITS
ABOUT THE SUN OF THE EARTH AND THE COMET
HAVE.. ISOMET

FROM ABOVE - COMET = (5 komet = (5 komet = 1- Ecomet = 0.230 Re

AND QEARTH = RE

THEN ... $\frac{1 - \text{COMET}}{\text{TLEARTH}} = \left(\frac{0.230 \, \text{Re}}{1 - \text{COMET}}\right)^3 = \left(\frac{0.230}{1 - \text{COMET}}\right)^3$ OR $\frac{0.230}{1 - 0.999881}$

OR TCOMET = 91.8×10 yr



Have.. $\stackrel{\leftarrow}{\leftarrow} = \frac{GM}{h^2}(1+\varepsilon\cos\theta)$ Eq.(12.39')

FOR A PARABOLIC TRAJECTORY, $\varepsilon=1$ NOW.. AT A, $\theta=0$: $\stackrel{\leftarrow}{\leftarrow} = \frac{GM}{h^2}(1+1)$ $\stackrel{\downarrow}{\rightarrow} \frac{1}{1+1}$ AT C, $\theta=90$: $\stackrel{\leftarrow}{\leftarrow} = \frac{GM}{h^2}(1+0)$ $\stackrel{\downarrow}{\rightarrow} = \frac{1}{1+1}$ OR $\frac{1}{1+1} = \frac{1}{1+1}$ OR $\frac{1}{1+1} = \frac{1}{1+1}$

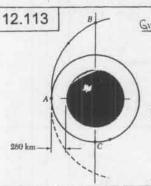
AS THE SPACECRAFT TRAVELS FROM B TO C, THE AREA SWEPT OUT IS THE PARABOLIC AREA BAC. THUS,

(CONTINUED)

12.112 continued

AREA SWEPT OUT = $A_{BK} = \frac{4}{3}(T_{h})(T_{h}) = \frac{8}{3}T_{h}^{2}$ NOW. $\frac{dA}{dt} = \frac{1}{2}h$, WHERE h = CONSTRUTTHEN $A = \frac{1}{2}ht$ OR $t_{BK} = \frac{2A_{BK}}{h} + T_{h}T_{h}$ $= \frac{2A_{BK}}{T_{h}} - \frac{16}{3}T_{h}^{2}$ $= \frac{16}{3}\frac{1530 \text{ km}}{10.42 \text{ km/s}}$ = 3751.85

OR Lec = 14 2 MIN 325



GIVEN: PARABOLIC APPROACH
TRAJECTORY AND
CIRCULAR ORBIT ABOUT
VENUS OF A SPACE
PROBE; R=6052 km
Mybrus = 4.87m034 kg
FIND: tbc

FROM THE SOLUTION TO PROBLEM 12.99 HAVE..

(JA) PAR = 10 131.4 M

AND (JA) CRC = (E (JA) PAR = 7164.0 M

ALSO, TA = (6052+280) Em = 6332 Em

FOR THE PARABOLIC TRAJECTORY BA HAVE

T = GMV (1+ E cos B) [EQ.(12.391)]

WHERE E=1. NOW-- $\frac{GMV}{A} = \frac{1}{h_{BA}^2} (1+1)$ AT A, $\theta=0$: $\frac{1}{h_{BA}} = \frac{h_{BA}^2}{h_{BA}^2} OR$ $\frac{1}{h_{BA}^2} = \frac{h_{BA}^2}{h_{BA}^2} OR$ AT B, $\theta=-90$: $\frac{1}{h_{BA}^2} = \frac{GMV}{h_{BA}^2} OR$ $\frac{1}{h_{BA}^2} = \frac{h_{BA}^2}{GMV} OR$

.. $\Gamma_B = 2\Gamma_A$ As the probe travels from B to A, the area swept out is the semiparabouc area defined by vertex A and point B. Thus,

(Area swept out) $\Gamma_B = A_B = \frac{3}{3}(\Gamma_A)(\Gamma_B) = \frac{4}{3}\Gamma_A^2$ Now - $\frac{1}{24}\Gamma_A = \frac{1}{2}\Gamma_A$, where $\Gamma_A = \frac{1}{2}\Gamma_A = \frac{1}{2}\Gamma_A$

NOW - $\frac{df}{dt} = \frac{2h}{2h}$, where $\frac{2ABA}{hBA} = \frac{2ABA}{hBA} = \frac{2h}{4} \frac{3}{16} = \frac{2h}{3} \frac{3}{16$

FOR THE CIRCULAR TRAJECTORY AC, $t_{AC} = \frac{72 \Gamma_A}{(UN) c_{IRR}} = \frac{71}{2} \frac{6332 \times 10^3 m}{7164.0 \text{ m/s}} = 1388.37 \text{ s}$

FINALLY -- tBC = tBA + tAC = (1666.63+1388.37)s = 3055.0 \$

OR tBC = 50 MIN SSS

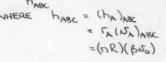
12.114

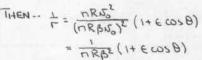
GIVEN: CIRCULAR ORBIT OF RADIUS TIR OF A SPACE PROBE HAVING VELOCITY IS ABOUT A PLANET OF RADIUS R; AT POINT A, VELOCITY IS REDUCED TO BUS (BKI) SO THAT PROBE IMPACTS AT POINT B FIND: 4 AOB IN TERMS OF TI AND B

HAVE FOR THE CIRCULAR ORBIT

No= (GM (EQ. (12.44))

OR GM=nR52 FOR THE ELLIPTIC ORBIT ABC T = GM (1+ E COSB) [EQ. (12.34)] 15





NOTING THAT POINT C IS THE PERIGEE OF THE ELLIPTIC IMPACT TRAJECTURY SO THAT ANGLE B IS DEFINED AS SHOWN, HAVE ..

AT A, 9=180: nR = nRBE (1- E) OR E = 1-B2

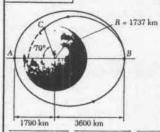
AT B: R = TRBE (1+ E COSB) = TRBE [1+ (1-B) cosb]

OR $COS\theta = \frac{nB^2-1}{1-A^2}$

Now - 4 AOB = 180 - 8 SO THAT COS (180- 4 AOB) = mB2-1 OR - cos (4 AOB) = nB2-1-B2

OR * ADB = cos' 1-18

12.115



GIVEN: ELLIPTIC ORBIT AND ELLIPTIC IMPACT TRAJECTORY OF LONAR ORBITER Z:

MMOON + 0.01230M FIND: I AND FOR IMPACT AT POINT C

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR THE ELLIPTIC ORBIT AB ..

WHERE HAS = (ha) AB = TB (NB) AB

AND GMMOUN = G(0.01230 MEARTH) = 0.012309 REARTH USING EQ. (12.30)

THEN .. 1/4 + 1/8 = 2(0.01230 9 REARTH) (CONTINUED)

12.115 continued

OR
$$(N_8)_{AB} = \frac{R_{EARTH}}{I_8} \left(\frac{0.02469}{I_8} \right)^{1/2}$$

$$= \frac{6.37 \times 10^6 \text{ m}}{3600 \times 10^3 \text{ m}} \left(\frac{0.0246 \times 9.81^{30} / 5^2}{1790 \times 10^5 \text{ m}} + \frac{1}{3600 \times 10^3 \text{ m}} \right)^{1/2}$$

$$= 950.43 \frac{\text{m}}{\text{s}}$$

FOR THE ELLIPTIC IMPACT TRAJECTORY HAVE .. $\frac{1}{L} = \frac{CM_{MOON}}{1} + C\cos\theta$ [Eq. (12.391]

NUMERE has = (ha) as = (ba) as Number that point B is the appelle of this TRAJECTORY, HAVE TO GMMOON - C

AT (, B=-70: R = GMMONH + CEOSLINO) OR C = Tos To (TR - GM MOON)

THEN- GMMOON - 1/8 = COSTO (1/R - GMMOON)

OR hac = GMmoon (1+0570)

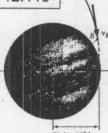
OR (UB) OR = REARTH (0.012309 (1+0570)) 1/2

 $(N_B)_{BC} = \frac{6.37 \times 10^5 \text{ m}}{3600 \times 10^5 \text{ m}} \left[\frac{0.01230(9.81 \frac{5}{5^2})(1 + 0570^6)}{1737 \times 10^3 \text{ m}} + \frac{3600 \times 10^5 \text{ m}}{3600 \times 10^5 \text{ m}} \right]^{1/2}$ = 869.43 3

FINALLY -- (NB)BC = (NB)AB + ANB
OR ANB = (869.43-950.43) \$

OR 14581 = 81.0 =

12.116



GIVEN: HYPERBOLIC TRAJECTORY OF A PROBE, E= 1.031; ALTITUDE AT B = 450 km. SB V 82.9"; FOR JUPITIER R = 71.492 ×103 km, pd 15014 P.1 = M

FIND: (a) & AOB (b) NB

FIRST NOTE... $F_8 = (71.492 \times 10^3 + 450) \text{ km} = 71.942 \times 10^3 \text{ km}$ (a) HAVE... $F_7 = \frac{GM_3}{h^2} (1 + 6 \cos \theta)$ [Eq. (12.39')]

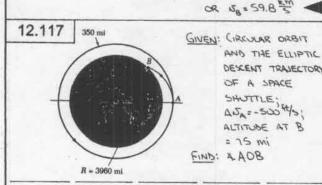
AT A, $\theta = 0$: $\frac{1}{\Gamma_A} = \frac{GM_J}{h^2}(1+E)$ OR $\frac{h^2}{GM_J} = \Gamma_A(1+E)$ AT B, $\theta = \theta_B = AAOB$: $\frac{1}{\Gamma_B} = \frac{GM_J}{h^2}(1+E\cos\theta_B)$ OR $\frac{h^2}{GM_J} = \Gamma_B(1+E\cos\theta_B)$

THEN -- TA (I+E) = TO (I+ E COSOB)

OR COS 08 = = [[[(14 E) - 1] = 1 [70.8403 km (1+1.031)-1] = 0.96873 (CONTINUED)

12.116 continued

OR OB = 14.3661° : 4 AOB= 14.37" (b) FROM ABOVE - h = GM, Fg (1+ & cos08) WHERE h= m | Is x miss = 18 No sind \$= (BB+82.9°) = 97.2661 (18 NB SIND) = GM, FB (1+ E COSBB) OR No = 1 (1+ E cos 80) 1/2 = 1 (66.73×10 2 m3 = 1.9×107 kg *[1+(1.031)(0.96873)]|1/2



GIVEN: CIRCULAR ORBIT AND THE ELLIPTIC DESCENT TRAJECTORY A OF A SPACE SHUTTLE; AUT = - SWAYS; ALTITUDE AT B = 75 mi

FIRST NOTE .. R=3960 mi = 20.9088 x10 ft ra = (3960+350)mi = 4310 mi = 22,7568x10 ft To = (3960+75)mi = 4035 mi FOR THE CIRCULAR ORBIT HAVE

NEIRC = (9RE [ED. (12.44)]

32.2 \$452 = 20.9088 MO ft (32.7568 NO ft) 1/2 = 24 871 5

NOW.. (NA) AB = NCIBE + AND = (24, B71, -500) \$ = 24,371 445

FOR THE ELLIPTIC DESCENT TRAJECTORY HAVE ..

= GM + (cos 8 [EQ. (12.39)]

NOTING THAT POINT A IS AT THE APOGEE OF THIS TRAJECTORY, HAVE.

THIS TRAJECTORY, HAVE ... AT A, B= 180: The GM - C C = GM - TO CR C = THE - TO

AT B, B= BB= 180- 4AOB: 18 = GM + Ccos BB

DR C= COSB (FB - GM)

THEN - GM - TA = COSDA (TB - GMZ) OR COS BB = TB - GM - L

NOW .. h= (ha) as = (Na) as AND GM= gRE EQ. (12.30)

FROM ABOVE - $9R^2 = T_A(UCIRC)^2$ (EQ. (12.44)]

THEN - $GM = \frac{T_A(UCIRC)^2}{[T_A(UA)AB]^2} = \frac{1}{T_A} \left[\frac{NCIRC}{(UA)AB} \right]^2$

(CONTINUED)

12.117 continued

SO THAT 4310 mi - (24,871 44/5)2 (24, B71 445)2-1 = 0,64411

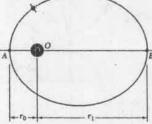
DR Bg = 49.901° FINALLY .. 4 AOB = 180- 49,901°

OR XAOB = 130.1°

12.118

GIVEN: ELLIPTIC ORBIT OF A SATELLITE AS SHOWN ZHOIM: == = = (=+=)

WHERE P= PA = PB



FROM THE SOLUTION TO PROBLEM 12.102 HAVE .. To + to = 2GM WHERE h = h = To No.

CONSIDER THE SATELLITE AT POINT A ..

±EF=man: F= m 5 O = O m(an)n

Now - FA = G Mm THEN - M D = G Mm [ED.(12.28)]

OR GM = & (N2 62) = & h2 EINALLY .. + + + = 3(\$ 45)

OR == = (=+ =) Q.E.D.

12.119

GIVEN: ELLIPTIC ORBIT OF A SATELLITE AS SHOWN; FOR COMET

> HYAKUTAKE, TO = 0.230RE E = 0.999887: Re = 149.6 ×10 km

FIND: (a) E IN TERMS OF TO AND F.

(b) I, FOR COMET HYAKUTAKE

(a) HAVE .. + = GM (+ E cos 8) Eq. (12.39')

THEN .. TO (1+ E) = T, (1-E) OR E = - 17- 10

(b) FROM ABOVE - 1 = 1+E G

WHERE TO: 0.230 RE

12.119 continued

THEN. F = 1+0.999887 x 0.230 (149.6 x 109 m)

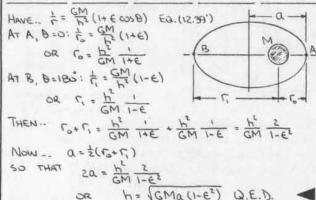
OR (= 609 x10 m

NOTE: I, = 4070 RE OR I, = 0.064 LIGHT YEARS

12,120

GIVEN: ELLIPTIC ORBIT OF SEMIMAJOR AXIS Q AND ECCENTRICITY & OF A SATELLITE ABOUT A PLANET OF MASS M

SHOW: h = 16Ma (1- 62)



12.121

GIVEN: TWO ELLIPTIC DRBITS OF SEMIMAJOR AXES Q, AND Q2 ABOUT A BODY OF MASS M; PERIODIC TIMES I, AND IZ OF TWO SATELLITES IN THE ELLIPTIC ORBITS

DERIVE: KEPLER'S , THIRD LAW ($\frac{T_{c}}{T_{s}^{2}} = \frac{\Omega_{c}}{\Omega_{c}^{2}}$) USING EQS. (12.39) AND LIZ.45)

CONSIDER THE ELLIPTIC DABIT OF SATELLITE 1. NOW == GM + (cos 0 Ea.(1239) - (B) THEN, FOR ORBIT 1... AT A, 8=0: (G) = GM + C, AT B, 0=180: (G) = GM - C,

THEN -- (TA), + (TB), = (GM + C1) + (GM - C1) OR (TA)+(TB) = ZGM (TA), (TO), Now a = 2((a) + ((a))) b, = ((TA), (TB), THEN ...

 $\frac{2a_1}{b_1^2} = \frac{2GM}{b_1^2}$ OR h = b, Jam

Ea. (12.45) FOR DRBIT 1 -- $\Sigma_1 = \frac{2\pi\alpha_1b_1}{b_1\sqrt{GM}} = \frac{2\pi}{\sqrt{GM}}$

(CONTINUED)

12.121 continued

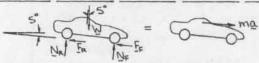
SIMILARLY, FOR THE ORBIT OF SATELLITE 2 ... I2 = 20 a22

THEN ..

12.122

GIVEN: AUTOMOBILE OF WEIGHT 3000 16 MOVING DOWN A 5° INCLINE; 15 = 50 H; AT t=0, FBRAKE=1200 16 IS APPLIED

FIND: X WHEN 15=0



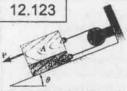
HAVE .. + EFx = ma: Wsins - (Fe+Fe) = Qa

WHERE FE+FR = FRRAKE

THEN -- Q = (32.2 ()(SIN 5 - 1200 16) = - 10 0736 52

FOR THIS UNIFORMLY DECELERATED MOTION HAVE -N2 = N2 + 2a(x-X3)

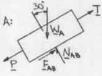
WHERE No = 50 M = 73.333 4/5 THEN WHEN N=0.. 0= (73.333 \$)22(-10.0136\$)X OR X= 267 St



GIVEN: mx = 30 kg, mg = 15 kg; M= 0.15, M= 0.10; 8-30, P= 250 N FIND: (a) QA

(b) T

FIRST DETERMINE IF THE BLOCKS WILL MOVE FOR THE GIVEN VALUE OF P. THUS, SEEK THE VALUE OF P FOR WHICH THE BLOCKS ARE IN IMPENDING MOTION, WITH THE IMPENDING MOTION OF A DOWN THE INCLINE.



AEFy = 0: NAB - WA COS30 = 0 OR NAB = MAG COS30 NOW .. FAB = USNAB = 0.15 mag cos30 DR T=P+m/g(sin30-0.15cos30)

SUBSTITUTING .. T= P+(30 kg)(9.81 52)(51N30-0.15cos30) =(P+108.919) N

T AEFy =0: No - NAS - Wa cos30 = 0 OR No = 90030 (MA+ME) NOW -- FB = MS No = 0.1590030 (MA+ME)

OR T = mg q sin30 + 0.15 magcos30 + 0.15 q cos30 (ma+ma)

12.123 continued

OR T= 9 [mg sin30 + 0.15 (2ma+mg) cos30] = (9.81 52) [(15 kg) sin30+0.15(2=30+15) kg=cos30] = 169.152 N

THEN -- 169.152 N = (P+ 108.919) N

OR P = 60.2 N FOR IMPENDING MOTION OF A DOWNWARD. SINCE P < 250 N, THE BLOCKS WILL MOVE, WITH A MOVING DOWNWARD.

NOW CONSIDER THE MOTION OF THE BLOCKS.

A: WA NAB = MAGA

AΣFy=0: NAB - WA COS30 = 0

OR NAB = MAG COS30

SLIDING: FAB = μA NAB

= 0.1 MAG COS30

ΣΕFx = MQ: -T+P - FAB + WA SIN30 = MAGA

25 Fx = ma: -T+P-FAB+Wa SIN30 = Maga CR T=P+ mag(SIN30-01cos30)-maga

T = 250 N + (30 kg) { (9.81 3 × X 514 30 - 0.1 cos 30) - ax } = (371.663 - 30 ax) N (1)

B: NAS FAST T MGQ8

AΣFy=0: NB-NAB-WB 0530=0 OR NB=9 0530 (MA+MB)

SLIBING: FB = MR NE

= 0.1 g cos 30 (m+m)

** EFx = mg as: T-FAs-Fg-WB 51N30 = mg ag

OR T = mgg 51N30+0.1 mg cos30+0.1g cos30 (ma+mg)+mg ag

= g[mg51N30+0.1(2ma+mg)cos30]+mg ag

= (9.81 52)(15 kg)51N30+0.1(2x30+15) kg x cos 50)

+ (15 kg) ag

= (137. 293 + 15as) N (2)

Equating the expressions for T [Eas.(1) and (2)] and noting that $Q_A = Q_B$.

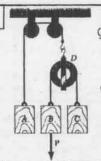
371.663-300A = 157.293+150A OR 0A = 5.2082 52

i. a. = 5.21 ₹ 730

(b) Substituting into Eq (1).. T= (371.663-30×5.2082) N

OR T = 215 N

12.124



GIVEN: WA = 20 16, WB = WC = 10 16; AT t = 0, 550; AT t = 25, 448 = 8 ft & FIND: (a) P (b) TAD

TIB A D THE

FROM THE DIAGRAM..

CORD 1: YA+ YB = CONSTANT

THEM... SA+ SB = O

AND (A+ CB = O

CORD 2: (YB-YB)+ (YC-YB)

= CONSTANT

THEM... JB+NE-2 JB=0

OR .. 20x+0x+0c-20b=0

Now... HAVE UNIFORMLY ACCELERATED MOTION BELOWSE ALL OF THE FORCES ARE CONSTANT. THEN... $48 = (46)_0 + (58)_0^2 + \frac{1}{2} a_0 t^2$

45 = (46) + (158) 2 + 2ast

AT t=25, Ays=8fl: 8fl = 2as (25)2

OR as = 4 442

(a)
PULLEY D: +1ΣFy = PABOD: 2Tex -Tab = D

OR Tab = 2Tex

Tex

Substituting the expressions for Q_A and Q_C into Eq. (1) - $2g(1-\frac{T_{AB}}{W_A})+Q_B+g(1-\frac{T_{AB}}{W_C})=0$

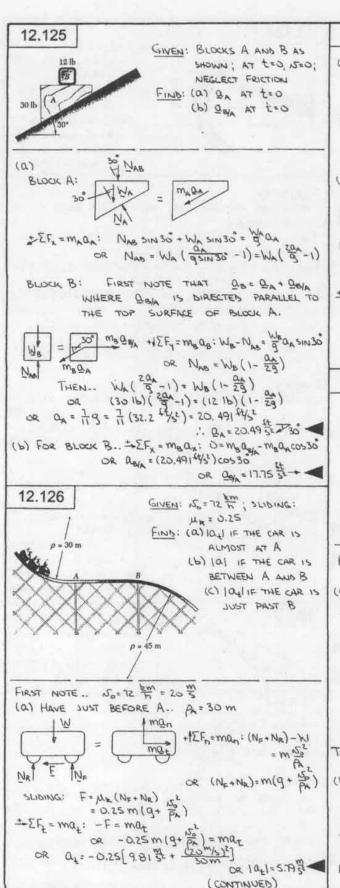
THEN: $(\frac{2}{201b} + \frac{1}{24101b})$ Tab = $3 + \frac{Q_B}{3}$

OR TAS = 20,828 16 AND THEN TBC = 10,414 16

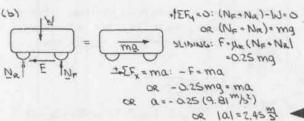
BLOCK B: $\frac{1}{M_B} = \frac{1}{1} + 1 \sum_{k=1}^{N_B} \sum_{k=1}^{$

(b) HAVE FROM ABOVE ..

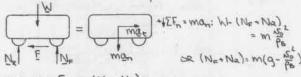
OR P = 1.656 16



12.126 continued

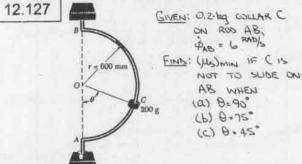


(C) HAVE JUST PAST B .. PB = 45m



SLIDING: $F = \mu_R (N_E + N_R)_{M_R}^2$ = 0.25 m (9 - $\frac{N_R}{R_R}$) $+ \Sigma F_E = m \alpha_E$: $-F = m \alpha_E$ $0R - 0.25 m (9 - <math>\frac{N_R}{R_R}$) = $m \alpha_E$ $0R - 0.25 [9.81]_{M_R}^{M_R} - \frac{(20^m/5)^2}{45m}$]

OR 10+1=0.230 32



FIRST NOTE .. WE = ((SIN 0) DAB = (0.6 m) (6 5) SIN 8 = (3.6 5) SIN 8

(a) WITH θ=90°, ~ Σ=3.6 %

NEF, =0: F-W=0

OR F= meg

Now... F= μs N

OR N=μs meg

OR μς meg= me νε

OR MS = 3/2 = (9.81 /6)(0.6 m)

OR (US)min = 0.454 THE DIRECTION OF THE IMPENDING MOTION IS

DOWNWARD

(6) AND (C)
FIRST OBSERVE THAT FOR AN ARBITRARY VALUE OF θ IT is not known whether the impending motion will be upward or downward. To consider both possibilities for each value of θ , let F_{boun} correspond to impending motion downward (continues)

12.127 continued

FUP CORRESPOND TO IMPENDING MOTION UPWARD THEN, WITH THE TOP SIGN CORRESPONDING TO FDOWN, HAVE ..

EUP

mean

+ 1EFy=0: Nos8 + Fsin8 - Wc = 0 Now. F= us N D=P=M-BMENZU=BEODN NAHT COSO + MSSIND

Buse = usmed

+ EFn = me an: NSINB = FCOSB = ME DE P=rsinb SUBSTITUTING FOR N AND F ... TSING SING = BUILD THE BUILD COSD = ME TSING COSD = ME TSING

TAND Buist = Buntanti + Buntanti

US = + TAND - grand 1+ NEE TAND

152 = (9.81 1/2) (0.6 m) 2 NO 18 (9.81 1/2) = 2.2018 SINB

Ms = + TANB - 2. 2018 SINB

(b) 8=75° TAN 75 - 2,2018 SIN 75 1+2.2018 SINTS TANTS = ± 0.1796

THEN .. DOWNWARD: US=+0.1796 UPWARE: MY CO .. NOT POSSIBLE

(MS)MIN= 0.1796

THE DIRECTION OF THE IMPENDING MOTION IS DOWNWARD -(C) 8= 45°

Ms=± TAN 45-2.2018 SIN 45 = ± (-0.218)

THEN - DOWNWARD : ILS CO - NOT POSSIBLE BIS.0 = 24 CRAW9U

(US) min = 0.218

THE DIRECTION OF THE IMPENDING MOTION IS UPWARK

NOTE: WHEN 0= Buil 8105.5 - BUAT OR 0 = 62.988°

145 = 0. THUS, FOR THIS VALUE OF 8 FRICTION IS NOT NECESSARY TO PREVENT THE COLLAR FROM SLIBING ON THE ROB.

12,128

GIVEN: We = 4 16, b= 20 IN. WHEN 8 : 20, 8 . 15 80 8 = 250 RAD FIND: (a) Fr AND FO ON PIN B WHEN 8=20°

(b) P AND Q WHEN B= 20 WHERE P 15 DOWN DO OT SUD IS DUE TO DE

KINEMATICS

FROM THE BRAWING OF THE SYSTEM HAVE r= 2bcos 8

THEN .. C=-126 SINB)B

" = -26 (BSINB + B2 COSB)

Now. a= "-rb2 = -26(8 sin 8+62 cos8)-(26 cos8) 62

=-26(95m0+202000) --2(12 ft)[(250 2000) = - 1694.56 452

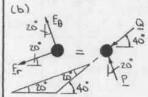
0(04120ds)s+0(0200ds)=075+07=00

= 26(8000 - 2020 sin 8) = 2(12 H)[(300 50) cos20 - 2(15 50) sin 20] - 270.05 Hys

KINETICS

(a) Have.. Fr= mar = \$16 (-1694.56 \$1)=-13,1565 16 DR F =- 13.16 15

\$16 (270.05\$1)=2.0967 16 DR F = 2.10 16

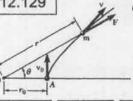


==== - Q cos 20 OR Q = cosso (13.1565 16) = 14.00 09 1b

12F6: F0 = P-Q SMZO OR P=(2.0967 + 14.00009 SIN 20) 16

= 6.89 1b 1. P=6.89 16 à 70 Q=14.00 16 7 40°

12,129



GIVEN: CENTRAL FORCE E AND PATH SHOWN; T= To/cosza; AT t=0, R=2° θ=0 €= 2° FIND: No AND NO AS FUNCTIONS OF 8

COS 28

THEN - C = 25 SINZO O

Now .. 5= 12-+1868 SO THAT AT \$=0... 5 = 500 FROM EQ. (18.27): 120 = 500

OR B = 1015 = 1010 (COS 28) 2 = 15 00 2 50

THEN i= 20520 (500528) = 205 SINZB (CONTINUED)

12.129 continued

Now Not + i

OR No 2NS SINZA

AND 15=10 = 500 = 500 cos 28 BS2000 Th = DA SO

12.130

GIVEN: RADIUS I OF THE MOON'S ORBIT: RADIUS R OF THE EARTH; THE ACCELERATION OF GRAVITY Q AT THE EARTH'S SURFACE; THE PERIODIC TIME TO OF THE MOON SHOW: r=f(R,g,z)

FIND: F KNOWING THAT T= 27.3 DAYS

HAVE -- F = G MM [Ea. (12.28)] F = Fn = man = m 5 AND THEN C WW = W Z N2 = GM



12 = 9R2 EQ. (12.30) GM= 9R2 Now SO THAT

FOR ONE ORBIT .. $T = \frac{2\pi \Gamma}{5} = \frac{2\pi \Gamma}{5}$ OR 7= (972 R2) 13

NOW .. T= 27.3 BAYS = 2.35872×10 5 R = 3960 mi = 20.9088 x10 ft

T=[9.81 32 * (2.35812x10 5)2 (6.37x10 m)2] 15 = 382.81×10° m

OR F= 383×10 km

U.S. CUSTOMARY UNITS:

T=[32.2 32.x (2.35872x10 5) x (20.9088x10 4)2] 13 = 1256, 52×10 ft

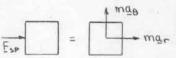
OR F = 238 × 10 mi

12,131 100 mm

GIVEN: m=0.25 kg: k=6 m. (Lo) = 0.5 m; AT t=0, AT A: NEGLECT FRICTION AND MROS

FIND: (a) (NB)B (b) (as), AND (as)8 (C) (acourrisos)8

FIRST NOTE .. FSp = k[(Lo)sp-F] AT B: (For) = 6 m (0.5-0.4) m = 0.6 N



(a) AFTER THE CORD IS CUT, THE ONLY HURIZONTAL FORCE ACTING ON THE COLLAR IS DUE TO THE (CONTINUED)

12.131 continued

SPRING. THUS, ANGULAR MOMENTUM ABOUT THE SHAFT IS CONSERVED.

.. I m (NA) = IB M (NB) B WHERE (NA) = TA O. THEN - (WE) = 0.1 m [(0.1 m)(16 =)]

OR (NE) = 0.4005 (b) HAVE FO = 0 : (as) = 0 Now .. = EF = mar: (Fsp) = m (as), OR (00) = 0.6 N

OR (a0)=2,40 12 (C) HAVE .. OF= "- T B'

NOW .. acollegros = " AND Bo = (156) THEN. AT B: (QOULLEROO) = 2.40 5 + (0.4m) (0.400 5) OR (QUOLURIRON) = 2.80 52

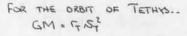
12.132 GIVEN: TRAJECTORY OF THE VOYAGER I SPACECRAFT ABOUT SATURN; AT THE POINT OF CLOSEST APPROACH, T= 185×10 km, 5=21.0 km/s; FOR

THE CIRCULAR DREIT OF THE MOON
TETHYS, T= 295 NO EM,
N= 11.35 NO EM/S

FIND: E AT THE POINT OF CLOSEST APPROACH OF VOYAGER I

FOR A CIRCULAR ORBIT

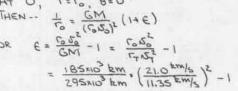
S= GM EQ. (12 Ea. (12.44)





FOR NOVACER'S TRAJECTORY HAVE..

h = 10,50 WHERE r=10, 8=0 AT O.



OR E=1.147



40 mi

150 mi

GIVEN: ELLIPTIC AND CIRCULAR ORBITS OF THE SHUTTLE COLUMBIA ABOUT THE EARTH

FIND: (a) tAB
(b) TCIRC

FIRST NOTE.. R = 3960 mi = 20,9088 x10° ft

FA = (3960 + 40) mi = 4000 mi = 21.120 x10° ft

FB = (3960 + 150) mi = 4110 mi = 21.7008 x10° ft

(a) THE PERIODIC TIME TOF AN ELLIPTIC ORBIT IS

T = 2774 b

[EQ.(12.45)]

: $t_{AB} = \frac{1}{2}\hat{z} = \frac{nab}{h_{AB}}$ WHERE $a = \frac{1}{2}(r_A + r_B) = \frac{1}{2}(21.120 \times 10^6 + 21.7008 \times 10^6)$ ft $b = (r_A r_B) = \frac{1}{2}(21.120 \times 10^6 + 21.7008 \times 10^6)$ ft $b = (r_A r_B) = \frac{1}{2}(21.120 \times 10^6 + 21.7008 \times 10^6)$ ft

FROM THE SOLUTION TO PROBLEM 12.102, HAVE FOR THE ELLIPTIC DRBIT ..

 $\frac{1}{G} + \frac{1}{G} = \frac{2GM}{h_{AB}^2}$

Now .. GM = gR [EQ. (12.30)]

SO THAT $h_{AB} = \left(\frac{29R^2}{\frac{1}{12} + \frac{1}{18}}\right)^{\frac{1}{18}} = \left[\frac{2(32.2\frac{64}{32})(20.9088 + 10^{\frac{1}{18}})^{\frac{1}{18}}}{21.1008 \times 10^{\frac{1}{18}} + \frac{1}{21.1008 \times 10^{\frac{1}{18}}}\right]^{\frac{1}{18}} B: W_{8,30}$ $= 548.95 \times 10^{\frac{1}{18}} \frac{41}{21.1008 \times 10^{\frac{1}{18}}}$

FINALLY -- tab = TT(21.4104NIO ft)(21.4084NIO ft) = 2623.2 S

OR tas = 43 MIN 435

(b) FOR THE CIRCULAR ORBIT

Teire = 2717B

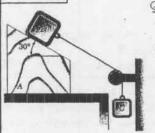
WHERE IS - 9RE

WHERE $\Lambda E_{IRC} = \sqrt{\frac{9R^2}{F_B}}$ [EQ.(12.44)]

THEN .- ICIRC = 271 15 12 = 271 (21.7008 *10 4t) 3/2 (20.9088 NO 4t) (32.244/3)/2 = 5353,55

OR Tare = 1 h 29 MIN 135

12.C1



GIVEN: MA = 20 kg, MB = 10 kg, Mc = 2 kg; t=0, 5=0; M = 0

FIND: QA AND QB/A FOR MED USING QUE 0.01 WHILE QA 20 AND QUE 201 WHILE QB/A 20

ANALYSIS

KINEMIATICS

HAVE.. QB = QR + QBA WHERE QBA IS DIRECTED ALONG THE INCLINED SURFACE OF A. THEN..

 $\Omega_{B} = \Omega_{A} \left(-\cos 30 \frac{1}{2} - \sin 30 \frac{1}{2} \right) + \Omega_{BA} \frac{1}{2}$ ALSO, SINCE THE CORD IS OF

 $\alpha_c = (\alpha_B)_X$, $\alpha_c = (\alpha_B)_X$, $\alpha_c = \alpha_B - \alpha_A \cos 30^\circ$

KINETICS:

+1ΣFy=meae: T-We=-meae

OR T=me(g-ae)
= me(g-aey+axcos30)

 $\frac{1}{1} = \frac{30}{m_B \alpha_A} = \frac{30}{m_B \alpha_{BA}}$

QB/A

NAS + EFx = mBax: T-FAB+ WB SIN30 = mBabk-mBakcos30

OR T-FAB+109 SIN30 = 10 aBk-1004 cos30 (3)

+1/2Fy1 = MBQy1: NAB - WB COS30 = - MBQA SIN30
OR NAB = 109 COS30 - 1004 SIN30 (4)

SLIDING: FAB = JL NAB

OR FAB = 10 M (9 COS30 - QA SIN30) (5)
SUBSTITUTING EQS. (2) AND (5) INTO EQ. (3)...
2 (9-08/A+QACOS30) - 10 M (9 COS30-QA SIN30)+109 SIN30
= 10 QB/A - 10 QA COS30

OR 9 (1-54 cos30+5 SIN30)

= 6 aB/A - aA (5,451N30+600530) (6)

NOTE: BLOCK A WILL NOT MOVE (Q_A = 0) BEFORE

BLOCKS B AND C WILL NOT MOVE (Q_{BA} = Q_B = 0).

THEREFORE, THE SYSTEM WILL REMAIN AT

REST WHEN

g(1-5/20030+551030)=0
OR 1220.808 FOR NO MOTION

FAB NAB = MAQA

= ΣFx = mxax: NAB SIN30 - Fx - FAB COS30 = mxax

OR NAB (SIN30 - μCOS30) - Fx = 20 ax (7)

(CONTINUED)

12.C1 continued

+12Fy = 0: NA - NAB COS30 - FAB SIN30 - WA = 0 OR NA = NAS (COS30+ MSIN30) + 209 (8) SLIDING: FA = LINA

OR FA = MNAB (cos30 + MSIN30) + 20Mg (9)

SUBSTITUTING EQ. (9) INTO EQ. (7) .. NAB (SIN 30- MCOS30) - MNAB (COS30+ MSIN 30) - 20 Mg

OR NAB[(1-122) 51N30 - 212 COS30] - 20119 = 200A SUBSTITUTING FOR NAS [EQ. (41] .. (109 cos30-100, sin30) [(1-12) sin30-24 cos30]

2004 = PUCS -

LET A= (1-42) SIN30 - 240530 THEN .. 9 (A cos30 - 2 pl) = (2+ A SIN30) Ox

ax = Acos30 - 2M 9 (101)

NOTE: BLOCK A WILL REMAIN AT REST WHEN q (A cos30 - 2 m) = 0

OR [(1-12) SIN 30 - 211 COS 30) COS 30 - 211 = 0 OR (\$51060) LL2+ 2(1+ cos 30) L- \$51060 = 0

OR M 2 0.12188 FOR BLOCK A TO REMAIN AT REST

NOW - REWRITE EQ. (6) AS 084 = 6 [9 (1-5 ju cos 30" + 5 sin 30")

+ an (54 SIN 30+6 cos30")] (11) WHICH REDUCES TO

ana = 916 (1-540530+551430) (12) WHEN Qy = 0

OUTLINE OF PROGRAM

DEM : M TO BULLY LAITINI TUGHT COMPUTE A: A = (1-12) SIN 30- 24 COS 30 ax = A cos30 - 211 COMPUTE Qu:

WHILE ax >0

COMPUTE aba:

Oep = 6[9(1-5/20030+5 sin 30) + an (5 m sin 30 + 6 cos 30)]

TRINT THE VALUES OF IL, OA, AND DEVA UPDATE M: M= M+ 0.01

INCREASE IL TO THE NEXT TENTH:

1. = 10 [INTEGER VALUE (10,11)] + 0.1

COMPUTE CLEYA: 3 (1-5/4 COS30+5 SIN 30°)

WHILE aby >0

PRINT THE VALUES OF ILL AND QUA UPDATE M: M=M+0.1

12.C1 continued

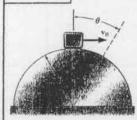
PROGRAMI OUTPUT

ji	accel. of A,	m/s ² accel.	of B wrt A, m/s2	
0.00	1.888		7.358	
0.01	1.742		7.167	
0.02	1.594		6.975	
0.03	1.445		6.780	
0.04	1,295		6.582	
0.05	1.143		6.382	
0.06	0.989		6.179	
0.07	0.833		5.973	
0.08	0.676		5.764	
0.09	0.518		5.553	
0.10	0.357		5.339	
0.11	0.195		5.122	
0.12	0.031		4.901	

For those values of µ for which the wedge is at rest

μ	accel.	of B wrt A	m/s=
0.20		4.307	
0.30		3.599	
0.40		2.891	
0.50		2.183	
0.60		1.475	
0.70		0.767	
0.80		0.059	

12.C2



GIVEN: W=116, 15=105; 0= ML = 0.4 FIND: 8 AT WHICH THE BLOCK LEAVES THE SURFACE; M=0,0.05,0.10, ..., 0.4

ANALYSIS



1ΣFn=man: Woos θ- N= m & OR N=m(gcos8-SLIBING: F= MEN = Mx m(gcos 8 -

* EF = max: WSIND-F = max OR az = 9 SIND - mF

SUBSTITUTING FOR F .. at=g(sIND-MACOSD)+ME & #= 3(21NB-HFCO2B)+HFD (1)

ALSO. 5=10 OR #= = 55 THUS, DIFFERENTIAL EQUATIONS (1) AND (2) DEFINE THE MOTION OF THE BLOCK. AS THE BLOCK LEAVES THE SURFACE, N-DO. THUS, GCOST- 500 DEFINES THE VALUE OF B AT WHICH THE BLOCKS LEAVES THE SURFACE.

OUTLINE OF PROGRAM

FOR EACH VALUE OF ME DEFINE THE INITIAL VALUES OF IT AND B USE THE MONFIED EULER METHOD (SEE THE SOLUTION TO PROBLEM 11. (3) WITH A STEP (CONTINUES)

(CONTINUED)

12.C2 continued

SIZE At = 0.01 5 TO NUMERICALLY INTEGRATE THE EQUATIONS

WHERE $\rho = 5$ ft. COMPUTE N_1 AND N_2 : $N_1 = \cos\theta_1 - \frac{\omega_1^2}{3\rho_{52}}$ $N_2 = \cos\theta_2 - \frac{\omega_1^2}{3\rho_{52}}$

WHERE θ_1 AND u_1^* ARE THE VALUES OF θ_2^* AND u_1^* VELOCITY, RESPECTIVELY, AT THE BEGINNING OF A TIME INTERVAL, AND θ_2^* AND u_1^* ARE THE VALUES AT THE END OF THE TIME INTERVAL.

IF $N_2 > 0$, update it and $\theta: J_1 = J_2; \theta_1 = \theta_2$ If $N_2 < 0$, use linear interpolation to DETERMINE THE VALUE OF θ at which N=0: $\theta = \theta_1 + \frac{O-N_1}{N_2-N_1}(\theta_2-\theta_1)$

PRINT THE VALUES OF IL AND B

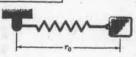
PROGRAM OUTPUT # # # 0.00 29.11* 0.05 29.61*

0.05 29.61° 0.10 30.16° 0.15 30.72°

0.20 31.33° 0.25 31.96° 0.30 32.63°

0.30 32.63° 0.35 33.35° 0.40 34.11°

12.C3



GIVEN: BLOCK OF MASS IM AUD SPRING OF CONSTANT & AT \$=0, 15=0 AND THE SPRING IS HORIBONTAL AUD UNSTRETCHED

FIND: (a) I AND IS WHEN THE BLOCK PASSES UNDER

THE PIVOT O

(b) YM WHEN TO=1 M

SO THAT Y → WHEN

THE BLOCK PASSES

UNDER O

-X FIRST NOTE...
$$\Gamma = \sqrt{x^2 + y^2}$$
 $\cos \theta = \frac{\lambda}{r} \sin \theta = \frac{\lambda}{r}$
 $F_{SP} = \frac{\lambda}{r} (\Gamma - \Gamma_0)$

$$F_{SP} = \begin{cases} -F_{SP} = ma_{x}: -F_{SP} \cos \theta = ma_{x} \\ DR \ a_{x} = -\frac{F_{SP}}{m(r-r_{o})\cos \theta} \end{cases}$$

$$ma_{xy} \quad \frac{du_{x}}{dt} = \frac{1}{m}(\sqrt{x^{2}} + \frac{1}{m^{2}}) (1)$$

12.C3 continued

ALSO... $\frac{dx}{dt} = \mathcal{S}_{x}$ (3) $\frac{dy}{dt} = \mathcal{S}_{y}$ (4)

Therefore, differential equations (1)-(4) define

The motion of the mass.

Now... $\mathcal{S} = [\mathcal{S}_{x}^{2} + \mathcal{S}_{y}^{2}]$ and $\theta_{\mathcal{S}} = \text{Tan}^{-1} \mathcal{S}_{y}^{2}$ Define the magnitude and direction, respectively of the velocity.

OUTLINE OF PROGRAM

INPUT VALUE OF R/M

INPUT UNSTRETCHED LENGTH OF THE SPRING TO INPUT SYSTEM OF UNITS

DEFINE THE INITIAL CONDITIONS:

VIETO, VIETO; (UX), TO, (UV), TO

USE THE MODIFIED FOLER METHOD (SEE THE

SOLUTION TO PROBLEM II.C3) WITH A STEP

SIZE At = 0.001 S TO NUMERICALLY

INTEGRATE THE EQUATIONS

WHEN $X_1 > 0$ AND $X_2 < 0$ COMPUTE T_1 AND T_2 : $T_1 = \sqrt{X_1^2 + y_1^2}$ $T_2 = \sqrt{X_2^2 + y_2^2}$ (Ba) = $TAN^2 \left(\frac{y_1^2}{y_2^2}\right)$

COMPUTE 152 AND BUZ: 5_ = \((Ux)\)_2 + (U4)\)_2 (BU)_2 = TAN \((Ux)\)_2 (Ux)_2

WHERE (), AND ()2 DENOTE VALUES AT THE BEGINNING AND END, RESPECTIVELY, OF A TIME INTERVAL.

USE LINEAR INTERPOLATION TO DETERMINE THE VALUES OF F, J, AND By AT X=0:

$$C = C_1 + \frac{O - X_1}{X_2 - X_1} (C_2 - C_1)$$

$$O = (O_{\sigma})_1 + \frac{O - X_1}{X_2 - X_1} ((O_{\sigma})_2 - (O_{\sigma})_1)$$

$$O = (O_{\sigma})_1 + \frac{O - X_1}{X_2 - X_1} ((O_{\sigma})_2 - (O_{\sigma})_1)$$

PRINT THE VALUES OF BYM, TO, T, J, AND BU

PROGRAM OUTPUT

(α) k/m = 15.00 /s° Unstretched length of the spring = 1 m

X1 = 0.001 m X2 = -.002 m Y = 2.765 m Y = 2.740 m/s

v = 2.740 m/sAngle v forms with the horizontal = -6.19°

k/m = 20.00 /s² Unstretched length of the spring = 1 m

X1 = 0.001 m X2 = -.002 m

r = 2.372 m v = 2.983 m/s Angle v forms with the horizontal = 0.93°

(CONTINUED)

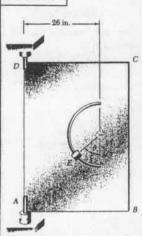
12.C3 continued

 $k/m = 25.00 /s^4$ Unstretched length of the spring = 1 m

X1 = 0.000 m X2 = -.003 m r = 2.121 m v = 3.195 m/s Angle v forms with the horizontal = 4.62°

(b) $k/m = 19.11 /s^2$ Unstretched length of the spring = 1 m

12.C4



GIVEN: MS=0.35; \$\Phi_{ABCD} = 14 \frac{RAD}{5}; \$\Phi_{ABCD} = 10 \text{ in: }; \$\frac{P}{5}\$; \$\Phi_{SUT} = 10 \text{ in: }; \$\frac{P}{5}\$; \$\Phi_{SUT} = 10 \text{ in: }; \$\frac{P}{5}\$; \$\Phi_{SUT} = 0.816 \$\frac{P}{5}\$; \$\Phi_{SUD} = 0.816 \$\Phi_{SUD} =

ANALYSIS

FIRST NOTE.. $\rho = \frac{1}{12}(26-10 \sin \theta) + \frac{1}{12}(3-5 \sin \theta) + \frac{1}{$

NAF. N = W(cos0 - WE SIND)

HAVE. $F = \mu_s N c_s$ THEM. $W(\sin\theta + \frac{\sqrt{\epsilon}}{9p}\cos\theta) = \mu_s \cdot W(\cos\theta - \frac{\sqrt{\epsilon}}{9p}\sin\theta)$ OR $[6q\sin\theta + \phi(13 - 5\sin\theta)\cos\theta]$

CASE Z: THE BLOCK IS RESTING AGAINST THE OUTER SURFACE OF THE SLOT; DOWNWARD MOTION IS IMPENDING (0 & 8 5 90°)

HEF=0: Fsinθ-Nosθ-W=0

+EF=man: Fcosθ+Nsinθ= \(\frac{1}{2} \)

THEN_ F= W(sinθ+ \(\frac{1}{2} \)

AND... N=W(-cosθ+ \(\frac{1}{2} \)

Then Sinθ)

HAVE _ F= MSN

(CONTINUED)

12.C4 continued

THEN.. $W(\sin\theta + \frac{\sqrt{2}}{9p}\cos\theta) = \mu_0 \times W(-\cos\theta + \frac{\sqrt{2}}{9p}\sin\theta)$ OR $[6g\sin\theta + \dot{\phi}^2(13-5\sin\theta)\cos\theta]$ = 0.35[-6gcab+ $\dot{\phi}^2(13-5\sin\theta)\sin\theta]$ (2)

CASE 3: THE BLOCK IS RESTING AGAINST THE OUTER SURFACE OF THE SLOT; DOWNWARD MOTION IS IMPENDING (90°50±180°)

HAVE .. F = μ N THEN .. W(cosα - \(\frac{\sign}{gp}\) sinα) = μ = W(sinα + \(\frac{\sign}{gp}\) cosα)

Now α = θ - 90

Substituting .. [cos(θ - 90) - \(\frac{\sign}{gp}\) sin(θ - 90)]

[cos(θ - 90) - \(\frac{\sign}{gp}\) sin(θ - 90)]

OR (SIND+ of COSO) = M. [SIN (0.90)) + of COSO TO THE DEFINING EGUATION

OF CASE 2

CASE 4: THE BLOCK IS RESTING AGAINST THE OUTER SURFACE OF THE SLOT; UPWARD MOTION IS IMPENDING (90°58:180°)

#ΣΕς=0:-Fωsα+Nsinα-W=0 L

+ΣΕς=man: Fsinα+Ncosα= N DE

THEN. F=W(-cosα+ SE SINα)

AND. N=W(SINα+ NECOSA)

HAVE.. $F = \mu_S N$ THEN.. $W(-\cos n + \frac{\sigma_S^2}{qp} \sin n) = \mu_c \cdot W(\sin n + \frac{\sigma_S^2}{qp} \cos n)$ NOW $\alpha = \theta - 90^\circ$ SUBSTITUTING.. $[-\cos(\theta - 90^\circ) + \frac{\sigma_S^2}{qp} \sin(\theta - 90^\circ) + \frac{\sigma_S^2}{qp} \cos(\theta - 90^\circ)]$ OR $(-\sin \theta - \frac{\sigma_S^2}{qp} \cos \theta) = \mu_S(-\cos \theta + \frac{\sigma_S^2}{qp} \sin \theta)$ WHICH IS THE SAME AS THE DEFINING EQUATION

WHICH IS THE SAME AS THE BEFINING EQUATION OF CASE I AFTER MULTIPLYING BOTH SIDES OF THE EQUATION BY -1.

It is next necessary to solve Eqs. (1) and (2) for θ . Each of these educations can be expressed as $f(\theta)$, and then the values of θ for which $f(\theta)=0$ can be determined. Substituting for $g(3z,z^{\frac{1}{2}})^{\frac{1}{2}}$ and then simplifying, find... $f_1(\theta)=(c_1\cdot c_2-13\dot{\phi}^2)\cos\theta-(4.55\dot{\phi}^2+193.z)\sin\theta$ $f_2(\theta)=-(c_1\cdot c_2+13\dot{\phi}^2)\cos\theta+(4.55\dot{\phi}^2-193.z)\sin\theta$ $f_3(\theta)=-(c_1\cdot c_2+13\dot{\phi}^2)\cos\theta+(4.55\dot{\phi}^2-193.z)\sin\theta$ $f_3(\theta)=-(c_1\cdot c_2+13\dot{\phi}^2)\cos\theta+(4.55\dot{\phi}^2-193.z)\sin\theta$

NOTE: FOR THOSE VALUES OF B FOR WHICH THE BLOCK
IS AT REST WITH RESPECT TO THE PLATE,

FMAX = JUS N 2 F

WHERE N AND F ARE GIVEN ABOVE FOR

WHERE N AND F ARE GIVEN ABOVE FOR EACH OF THE CASES. ALSO, $f(\theta) = F_{Max} - F$ (CONTINUES)

12.C4 continued

OUTLINE OF PROGRAM

INPUT VALUE OF \$

CONSIDER CASES I AND 4

FOR VALUES OF B FROM O TO 179 IN INCREMENTS OF 1°

COMPUTE f, (8):

f, (0) = (67.62-130) cos0-(4.550+1952) sin0 41.75 \$ 5 m2 0 + 2.5 \$ 5 m2 8

COMPUTE f, (B+1")

COMPUTE f, (8) + f, (8+1°) TO DETERMINE IF A ROOT LIES BETWEEN B AND

(B+10)

IF 1, (B) + 1, (0+1°) = 0, SOLVE 1, (B) FOR B USING NEWTON'S METHOD (SEE THE SOLUTION TO PROBLEM 11.C4) FRINT THE VALUE OF BROOT AND WHETHER FMAX-F AT B IS 2 OR 50

CONSIDER CASES 2 AND 3

FOR VALUES OF B FROM O TO 179" IN

INCREMENTS OF 1° COMPUTE f, (0):

f2(0) = - (67.62+13\$\dot^2)\cos\theta+(4.55\dot^2-193.2) SIN & -1.75 & SIN2 8 + 2.5 & SIN 28

COMPUTE &, (0+1")

COMPUTE f2(B) + f2(B+1")

IF f (0) x f (0+1") = 0, SOLVE f (0) FOR B USING NEWTON'S METHOS PRINT THE VALUE OF BROST AND WHETHER FMAX - F AT B IS 2 OR 5 0

PROGRAM OUTPUT

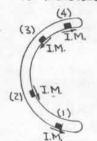
(Q.) Rate of rotation = 2 rad/s At θ = 4°, F(max) - F >= 0 θ (1) = 4.68°

At $\theta = 148^{\circ}$, F(max) - F = < 0 $\theta(3) = 148.57$

(b) Rate of rotation = 14 rad/s At θ = 115°, F(max) - F >= 0 $\theta(4)$ = 115.91

At $\theta = 77^{\circ}$, F(max) - F = < 0 $\theta(2) = 77.63^{\circ}$

NOTE: IN THE ABOVE OUTPUT, THE I IN B(i) DENOTES THE CASE FOR WHICH MOTION IS IMPENDING.



12.C5

GIVEN: TWO POINTS ON THE TRANSECTORY OF A SPACECRAFT: B, AND B, OR TO AND THE RADIAL DISTANCE TO AND THE VELOCITY AT THE APOSEE OR THE PERIGEE

FIND: TIME I FOR THE SPACECRAFT TO TRAVEL BETWEEN THE POINTS (a) B AND C OF PROB. 12.115: NB = 869,4 5 (6) A AND B OF PROB. 12.117;

ANALYSIS

== GM (1+ E cosθ) [Ea. (12,39')] HAVE ...

WHERE H= TARGER NARDER = TREASER NERGER = TO NAP PAROGEE = 180° PREPAGEE = 0 GM = G(MEARTH) MEARTH & (MEARTH) 9 REARTH

(Menon) 9R2 (1+E cos Bus)

THUS THE ECCENTRICITY OF THE TRAJECTORY CAN BE DETERMINED.

FROM PAGE 698 OF THE TEXT HAVE ..

WHERE h is a constant. Then.. $t = \frac{2}{h} \int_{\theta_{c}}^{\theta_{c}} dA$

WHERE dA = 2(r)(rd0)

OR AA = \$ TZAB AND L= EFErzaB

WHERE (IS GIVEN BY EQ. (12.39').

OUTLINE OF PROGRAM

SET VALUE OF AB: AB= 0.05 INPUT UNITS AND CONSTANTS

INPUT WHETHER VALUES ARE KNOWN AT THE

AROGEE OR THE PERIGEE

SET VALUE OF PAP: PAP = O (PERIGEE) BAP = 180 (APOGEE)

TUPUT THE DISTANCE TAP TO AND THE VELOCITY JAP AT THE APOGEE OR THE PERIGEE

INPUT THE VALUE OF B, FOR THE FIRST POINT ON THE TRAJECTORY

INPUT WHETHER THE SECOND POINT ON THE TRAJECTORY IS DEFINED BY THE VALUE OF BZ LCASE I) OR BY THE VALUE OF THE RADIAL DISTANCE TO (CASE 2)

INPUT M/MEARTH

COMPUTE THE ECCENTRICITY E OF THE

TRAJECTORY:

(CONTINUES)

12.C5 continued

```
CASE 11
```

INPUT THE VALUE OF B2 If $\theta_2 < \theta_1$, set $\Delta \theta = -\Delta \theta$ FOR VALUES OF θ FROM θ_1 to $\theta_2 - \Delta \theta$ in INCREMENTS OF AB REMENTS OF A: M.

UPDATE AREA A: M. $A = A + \frac{1401}{2} \left[\frac{(M_{EARTH}) 9R_{EARTH}}{(\Gamma_{AP} I S_{AP})^2} (1+6 \cos \theta) \right]^{-2}$ $A = A + \frac{1401}{2} \left[\frac{(\Gamma_{AP} I S_{AP})^2}{(\Gamma_{AP} I S_{AP})^2} (1+6 \cos \theta) \right]^{-2}$ COMPUTE TIME t: t= CA

PRINT THE VALUES OF TAP, NAP, B. Oz, AND t CASE 2:

INPUT THE VALUE OF 5 SET THE INITIAL VALUE OF B: B=B, WHILE TOTZ IF TIDE OR WHILE TERZ IF COMPUTE T: $\Gamma = \left[\frac{(\frac{M}{M_{\text{ENDTM}}})gR_{\text{ENDTM}}^2}{(\Gamma_{\text{AD}} \cup \Gamma_{\text{AD}})^2}(1+\varepsilon \cos \theta)\right]$ 5,452

UPDATE AREA A: A= A+21200 UPDATE 8: 8 = 8+08 COMPUTE TIME t: t = ZA PRINT THE VALUES OF TAP, NAP, B., TZ, AND t

PROGRAM OUTPUT

(0)

The radial distance to and the velocity at the apogee are, respectively, 3600 km and .8694 km/s $\theta 1 = 180^{\circ}$ $\theta 2 = 290^{\circ}$ Time t = 1 h 18 min 29 s

The radial distance to and the velocity at the apogee are, respectively, 4310 mi and 24371 ft/s $\theta 1 = 180^{\circ}$ r2 = 4035 mi Time t = 0 h 33 min 30 s

HOOM THE MOON 13.1 GIVEN : MASS OF SATELLITE, m= 1500 kg continued 13.4 SPEEL OF SATELLITE, U= 22.9 x 10 2 km/h HTSAS BHTUO (0) PIND: KINETIC ENERGY, T T= 1 m N2 = 1 (4kg)(25m/s)= 1250 N·m T= 1250 J N= 22.9 × 10 Rm/4 = 6.36 × 10 m/s W= mg = (4 kg) (9.81 m/52) = 39.240 N T=1 m 02 = 1 (1500 kg) (6.36×103 m/s)2 Ti+U1-2= T2 T,=0 U1-2=Wh T2= 39,240 N T= 30.337 x 109 N.M h= T2 = (1250 N.M) = 31.855 m NOTE: ACCELERATION OF GRAVITY (39.240N) HAS NO EFFECT ON THE MASS T= 30.3 GJ h= 31.9 M OF THE SATELLITE. (b) ON THE MOON MASS IS UNCHANGED, M= 4 kg 13.2 GIVEN: WEIGHT OF SATELLITE, W= \$70 16 SPEED OF SATELLITE & = 12,500 mi/h THUS T IS UNCHANGED FIND: KINGTIC ENERGY, T WEIGHT ON THE HOON IS, Wm= mgm=(4kg)(1.62m/62) Wm = 6.48 N N= (1,500 mi/h) (h/3600 5) (5280 ft/mi) hm= I = (1250 N:M)= 192,9 M (6'08 M) N= 18,333 ft/s MASS OF SATELLITE = (87016)/32.2 ft/s2) h= 192,9m m = 27.019 16.52/f+ T= = 1 mv= = (27,09)(18,333)2 13.5 GIVEN: DISTANCE d= 120 m T = 4.5405 X109 16.ft 45= 0.75 , NO SLIPPING NOTE: ACCELERATION OF GRAVITY HAS 60 % OF WEIGHT ON FRONT WHEELS NO EFFECT ON THE HASS OF THE 40 % OF WEIGHT ON REAR WHEELS T= 4.54 × 10 16.ft SATELLITE FIND: MAXIMUM THEORETICAL SPEED AT 120 M STARTING FROM REST GIVEN: WEIGHT OF STONE W= 516 (a) FOR FRONT WHEEL DRIVE 13.3 VELOCITY OF STONE V= 80 ft/s (b) FOR REAR WHEEL DRIVE ACCELERATION OF GRAVITY ON THE (a) FRONT WHEEL DRIVE SINCE GO % OF WEIGHT IS DISTRIBUTED ON FRONT MOON, 8m= 5.31 ft/52 FIND: (a) KINETIC BUERGY,T WHEELS, THE HAXIMUM FORCE TO HOVE THE CAR F = 45 N = (0.75) (0.6W) = 0.450 mg HEIGHT h, FROM WHICH STONE FOR 120 m U1-2= (0.450 mg)(120 m)= 54 mg WAS DROPPED (b) T AND h ON THE HOON T = 0 T, + U1-2=T2 (a) ON THE EARTH 0+54mg= 1 m 12 T= 12 m v= 1 (516/1/s2) (80 ft/s) N3= (2)(549)=(108)(9.81 W/52) T= 496.89 lbift N3= 1059.5 T= 497 16.9+ N2= 32.55 W/S T,+U,==T2 T,=0, U,-2=Wh=(516)(h), T2=49716.A N= 117.2 km/h Wh=T2 h= 316 = 99.4 ft (b) REAR WHEEL BRIVE in= 99.4ft USE SAME SOLUTION AS FOR (A) EXCEPT THAT (b) ON THE HOOD 40 % WEIGHT IS DISTRIBUTED ON REAR WHEELS F= 45 N= (0.75)(0.40W)= 0.3 mg MASS IS UNCHANGEL THUS T IS UNCHANGED FOR 120 m U1-2= (0,3 mg) (120m)= 36 mg T= 497 16. 44 WEIGHT ON THE HOON IS: Wm = mgm = (516) (5.31 41/52) T,= 0 T1+41-2= T2 Wm = 0.8245 16 0+36mg=1 mV22 hm= T3 = (497164+) = 6027 ft N2= (2)(36)(9)=(72)(9.81 m/sz) = 706.32 (0,824516) N= 26.58 W/s h= 603ft 13= 95.7 km/h SIVEN: MASS OF STONE, M= 4 kg NOTE: THE CAR IS TREATED AS A PARTICLE IN THIS 13.4 VELOCITY OF STONE, N= 25 M/S PROBLEM, THE WEIGHT DISTRIBUTION IS ASSUMED ACCELERATION OF GEAUITY TO BE THE SAME FOR STATIC AND BYNAMIC ON THE HOOW, 9 m= 1.62 m/52 CONDITIONS, COMPARE WITH SAMPLE PROBLEM FIND: 16.1 WHERE THE VEHICLE IS TREATED A & (a) KINETIC ENERGY T A RIGIO BODY. HEIGHT H, FROM WHICH THE STONE

WAS DROPPED

13.6



GIVEN: 1320 H DEAG PACE TRACK, CAR STARTS FROM REST CARS FRONT WHEELS OFFTHE GROUND FOR FIRST GOH WHEEL'S ROLL WITHOUT SLIPPING FOR REMAINING 1260 ft WITH 60 % OF WEIGHT ON REAR WHEELS 4= 0.60 , 45= 0.85, NO MIR OR ROWNE RESISTANCE FIND: (a) SPEED OF THE CAR AT END OF FIRST GOFF (b) MAXIMUM THEORETICAL SPEED AT FINISH LINE

(a) FIRST 60 ft: REAR WHEELS SKID TO GENERATE THE HAXIMUM FORCE. SINCE ALL THE WEIGHT IS ON THE REAR WHEELS THIS FORCE IS: F=4kN=(0.60)(W)

TI=0 T2= 1 8 020

U1-2=(F)(60ft)=(0.6/W)(60)=36W T1+ U1-2= T2 36W= = = \ W 000 NEO= 2318.4 NEO= 48.15 ft/s

1540=32.8 mi/h

(b) FOR 1320 ft REAR WHEELS SKID FOR FIRST GOFT AND ROLL WITH SCIDING IMPENDING FOR REHAINING 1260 ft WITH 60 % OF THE WEIGHT ON THE REAR (DEIVE) WHEELS. THE HAXIMUM FORCE GENERATED IS:

FIRST GOFT F. = (.6 XW) AS IN (a) REMAINING 1260ft F2 = 43 N = (0.85)(0.60)(W) = 0.510 W T=0 T2== \$ 401320

U1-2= (.6)(W)(60)+(510)(W)(260 = (36+642.6)W = 678.6W 0+678.6W=+ \$ N1520 N1320=43702 V1320 = 209.05 ft/s

N1325 142.5 milh

SEE NOTE FOR PROB. 13.5 FOR DISCUSSION OF WEIGHT DISTRIBUTION

13.7



GIVEN: 1320 FT DRAG RACE TRACK, CAR STARTS FROM REST. CARS' FRONT WHEELS OFF THE GROUND AND REAR WHEELS SKID FOR FIRST 60 ft SPEED AT END OF FIRST GOFT IS 36 milh. WHEELS ROLL WITH SLIPPING IMPENBING FOR REHAINING 1260 ft WITH 75% OF THE WEIGHT ON REAR (DRIVE) WHEELS. 46=0.8045 NO AIR OR ROLLING RESISTANCE

13.7 continued

FIND: SPEED OF CAR AT GNO OF RACE

FIRST 60 ft: SINCE ALL THE CARS WEIGHT IS ON THE REAR WHEELS WHICH SEID, THE FORCE HOUING THE CAR 15

F=4kN=(4k)(W) No= (36 milh) (88 ft/s) (60 milh) 150= 52.8 ft/s T=0 T2=1 moto = 1(12)(52.8 ft/s)2=(1393.9)(12)

> U1-2=(F)(60ft)=(44)(W)(60ft) TI + U1-2= T2 0+ 60 yew= (1393.9)(W) $4\mu = \frac{(1393.9)}{(60)(32.2)} = 0.72149$

FOR 1320 ft FORCE HOUING THE CAR IS FOR FIRST GOFT, F. = (4) (W) = (0.72149) W FOR REHAINING 1260 ST, WITH 75 % OF WEIGHT ON SEYS (DSIDE) THERE'S YND INDENDING 20017 F2 = (45) (0.75) W 45 = 46 (0.80) = (0.72149) 10.80) F2= (0,90186)(75)W=0.6764 45= 0.90186 TI= 0 T2= 3(4) (VIOZO)2

U1-2= (F.)(60ft)+F2(1260ft) = (0.72149)(W)(60ft)+(0.6764)(W)(1260ft) = 43.29 W + 852.3 W = 895.55 W Ti+ U1-2=T2 0+895.55W== (W) (Nissa) Niz= (29)(895.55)=(2)(322 ft/52)(895.55) N1320 = 57,673 V1320 = 240,2 ft/s SEE NOTE FOR PROB 13.5 N1020= 163.7 milh

13.8



GIVEN: 400 M DRAG RACE TRACK, CAR STARTS FROM REST FRONT WHEELS OFF THE GROUND AND REAR WHEELS SEID FOR FIRST 20 M. WHEELS ROLL WITH SLIPPING IMPENDING FOR REMAINING 380 M, WITH 80 % OF THE WEIGHT ON THE REAR DRIVE WHEELS PEAK SPEED AT END OF THE RACE = 270 km/h 4x= 0.7545

FIND:

- (a) COEFFICIENT OF STATIC FRICTION, US (b) SPEED ATTHE END OF THE FIRST 20 m

(a) FORCE MOVING THE CAR FOR THE FIRST 20 M, WITH ALL OF THE WEIGHT ON THE REAR DRIVE WHEELS AND THE WHEELS SKIDDING,

FI=4kN=4kW=(0.75)45)mg 4 p= 0.75 45

FORCE HOUING THE CAR FOR REHAINING 380 M WITH 80% DE THE WEIGHT ON THE REAL (DRIVE) WHEELS AND SLIPPING IMPENDING (CONTINUED)

13.8 continued

F2 = 45 (0.80)(W) = 45 (0.80) (W) = 45 (.80) mg

U400 = (270 km) (1000 m) / (3600 5) T2= 1 m v20= 1 m (75)= 28125 m

U1-2 = F. (20m) + F2 (380m) U1-1=(45)675,1Mg (20m)+(45)680)mg (380) U1-2= 15 45 mg + 30445 mg = 31945 mg T1 + U1-2 = T2 0+31945 Mg= 2812.5M 43= (2812.5)/(319)(9.81)= 0.8987

(b) FOR FIRST 20 m

MA = (0.75)(45)= 0.6741 FI= 42N= (0.6741) (Mg)

T,= 0 T2= 1 m N20

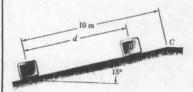
U1-2= (0.6741) (mg) (20m) = 13.481 mg 0 + (13.481) (mg) = 1 m 020 N2 = (2)(13.481)(4.81) = 264.5 N20= 16.26 M/5

SEE NOTE FOR PIB.5

N20= 58.6 lem/h =

45= 0.899

13.9



GIVEN: U ATC = 0 4k= 0.12

FIND: (a) INITIAL V AT A (b) V AS PACKAGE RETURNS TO A

(0)

UP THE PLANE, FROM A TOC. UE= 0 TA= 1 MNA, Tc= 0 UA-c=(-WOINIS"- F)(10 m) N-WC05150 = 0 N = WCosiso

F=4kN= 0.12 Wcos 150 UA-c = - W (sm 150+0,12 cos 150)(10 m) = = W (SIN 15"+ 0.12 cos 15") (10m) TA+UA-c=Tc

> Na2=(2)(9.81)(SINIS*+0.12COSIS*)(10 m) NA = 73.5 NA= 8.57 m/5/

(b) DOWN THE PLANE FROM CTO A

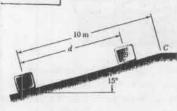
Tc= 0 TA = 1 m NA Uc-a = (W sin 15- F) 10 (FREVERSES DIRECTION) Tetuca=Ta 0+ w (sin 150-0,12 cos 150 X10m)=1 m Va

U2=(2)(9.B1)(SIN 150-0.12 cos 150)(10 m)

 $N_A^2 = 28.039$

N= 5.30 m/5 /

13.10



GIVEN: WAT A= 8m/s/ 4k=0.12

FIND! (a) DISTANCE d PACKAGE HOUES UP THE PLANE (b) VELOCITY UA, AS PACKAGE RETURNS

(A) UP THE PLANE FROM A TO B

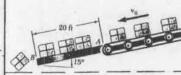
TA= 1 MU2= 18 (8 m/s)= 32 4 UA-B=(-WSINISO-F)d F=MKH=0.1ZN \$ EF= 0 N-WCOS 150= 0 N=WCOS 150 Un-a = - W (51115+ 0.12 cos 150)d = - wd (0.3747) TA +UA-B=TB 32 8 - wd (0,3743) = 0 d= (32)/(9.81)(,3747)

d= 8.70 m (b) DOWN THE PLANE FROM B TO A (F REVERSES DIRECTION) TA = 1 8 UA TB = 0 d=8.72 M/s UB-A = (WSIN 150- F) d= W(SIN 150-0, 12 COS 150) (8.70 M) UB-A = 1.245W

0+1.245W= 3 4 02 TB+UB-A=TA NA2= (21(9.81)(1.245) = 253.9 NA= 4.94 W/S

NA=4,94 M/s &

13.11



GIVEN; AT A, U=No

FOR AB, 44=0,40 AT B, W= 8 ft/s

FIND; No

TB= 1 MUB= 1 M (846) TB=32m

Un-a= (WSINISO- HEN) (20ft) EF= 0 N- WCOSIS" = 0 N= W co5150

UA-13 W(SIN 150- 0,40 cos 150) (20ft) UA-B = - (2.551) (W) = -2.551 Mg

TA+UA-B=TB

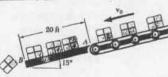
1 mv2-2,551 mq= 32 M

Vo = (2)(32+(2.551)(32.2 ft/s2))

v2= 228.29

U= 15.11ft/5

13.12



GIVEN: AT A, U=V0

AT B, U=0

FOR AB, U=040

FIND: V0

TA= \(\text{M} \text{NO}^{\cup} \) \(\text{TB} = 0 \)

\(\text{UA-Q} = \(\text{W} \sin \text{IN} \cop \text{UR} \text{N} \) \(\text{20 ft} \)

\(\text{EF=0} \)

\(\text{N} = \text{W} \cos \text{IS}^{\cip} = 0 \)

\(\text{N} = \text{W} \cos \text{IS}^{\cip} = 0 \)

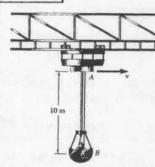
UA-8= W(SINIS*-0.4000515°)(20ft)
UA-8= - (2,551)(W)=-2.551 Mg

TA+UA-6=TB 1 m No2-2.551 mg = 0

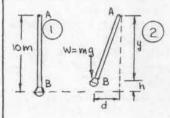
 $N_0^2 (2)(2.551)(32.2 \text{ f+/5}^3)$ $N_0^2 = 164.28$

No= 12.82 ft/s

13.13



GIVEN: CRANEMOUS AT
VELOCITY, V AND
STOPS SUBSERVLY
BUCKET IS TO
SWING NO HORE
THAN 4 M
HORIZONTALLY
FIND: HAXIMUM ALLOWABLE
VELOCITY AD



 $v_1 = v$ $v_2 = 0$ $v_3 = 0$

T2= 0

U1-2=-mgh d= 4m

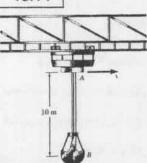
 $\overline{AB}^2 = (10m)^2 = d^2 + y^2 = (4m)^2 + y^2$ $y^2 = 100 - 16 = 84$ $y = \sqrt{84}$ $y = 10 - y = 10 - \sqrt{84} = 0.8349 \text{ m}$ $y = \sqrt{84}$ $y = \sqrt{84}$ $y = \sqrt{84$

1 m v2 - 0.8190 m=0

 $v^2 = (2)(0.8190) = 16.38$

U= 4.05 m/s

13.14



GIVEN: CRANE MOVES AT

VELOCITY U=3M/5

AND STOPS

SUDDENLY

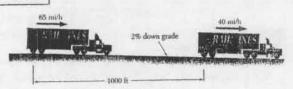
FIND: MAXIMUM HORIZONTAL

DISTANCE 1 MOVED

BY THE BUCKET

REFER TO FREE BODY DIAGRAM IN P.13.13 $U_1 = U = 3 \text{m/s}$ $T_1 = \frac{1}{2} \text{m} U^2 = \frac{1}{2} \text{m} (3 \text{m})^2 = 4.5 \text{m}$ $T_2 = 0$ $U_{1-2} = -\text{mgh}$ $T_1 + U_{1-2} = T_2$ A.5 m - mgh = 0 $A = \frac{4.5}{9.81} = 0.4587$ $AB^2 = (10)^2 = d^2 + y^2 = d^2 + (10 - 0.4587)^2 + (100 - 0.4587$

13.15



SIVEN: CAB WEIGHT, WC= 4000 16
TRAILER WEIGHT, WT= 12,000 16,2% GRADE
70% BRAKING FORCE SUPPLIED BY TRAILER
30% BRAKING FORCE SUPPLIED BY CAB

END:

(4) AVERAGE BRAKING FORCE TO SLOW DOWN FROM 65 MIN TO 40 MIN AS SHOWN

(b) AVERAGE FORCE BETWEEN CAB MIND TRAILER

(a) CAB-TRAILER SYSTEM



v= 65mc/n= 95.33ft/s V= 40 mc/n= 58.67ft/s

FB TSIN= 2/100

 $T_{1} = \frac{1}{2} (M_{T} + m_{c}) \vartheta_{1}^{2} = \frac{1}{2} (M_{T} + m_{c}) (q_{5,33} ft/s)^{2}$ $T_{1} = (q_{1}544) (M_{T} + m_{c}) (\vartheta_{2}^{2}) = \frac{1}{2} (M_{T} + m_{c}) (58,67 ft/s)^{2}$ $T_{2} = \frac{1}{2} (M_{T} + m_{c}) (M_{T} + m_{c})$ $T_{3} = (1721) (M_{T} + m_{c})$

 $T_1 + U_{1-2} = T_2$ $U_{1-2} = -1000 F_B + (W_t + W_c)(20 ff)$ $4544 (M_t + M_c) - 1000 F_B + (W_t + W_c)(20 ff) = 1721 (M_t + M_c)$

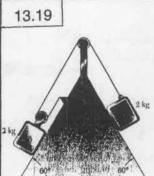
 $F_{8} = \left[(4544 - (121)) \left(\frac{16,000}{(32.2)} + (16,000)(20) \right) \frac{1}{(1000)} = 17227$

FB= 1723 16

(CONTINUED)

13.15 continued (b) TRAILER CONSIDERED SEPARATELY Ti= 12 (m+) (45.33) = 4544 m+ FE T3 = 1 (MT) (58.67) = 1721 MT 0.70FB 4544 MT - 1000 (Fct. 70 FB)+20W=1721MT T1+41-2=T2 FROM (a) FB= 1722.7. 1000 Fc= (4544-1721)(12,000) - (700)(17277)+(201(12,000) Fc= (1052) - 12059+ 240= 86.1 1b SEE NOTE FOR PISS F = 86.1 16(C) 13.16 (a) 65 mi/h 40 mi/h 1000 ft GIVEN: CAB WEIGHT, WC= 4000 16 TRAILER WEIGHT, WT= 12,000 16 2 % UP GRADE FIND (a) AVERAGE FORCE ON THE WHEELS TO SPEED UP F (b) AVERAGE FORCE IN THE COUPLING (b) CAB-TRAILER SYSTEM (0) N= 40 mi/n= 58.67 ft/s 1515 2/100 Uz= 65 mi/h= 95.33 Pt/s T= 1 (m++mc) (0,)= 1 (m++mc) (58.67)= 1721 (m+mc) T2= 1 (m++me)(v2)== (m++me)(45.33)= 4544(m++me) T, + U1-2= T2 U1-2= (1000)(F)-(1000)(2/100)(W+tW) 1721 (m++me) + 1000F-20 (W++We) = 4549 (m++me) 1000 F = (4544-1721) (16,000) +20 (16,000) F= 1403.+ 320 = 1723.1b F=1723 16 (b) TRAILER CONSIDERED SEPARATELY Ti= 2(M+) (58.67)= 1721 M+ T2= 12 (mr) (95.33) = 4544 mr (a) . 1721 mr + 1000 Fc - (1000)(2) WT = 4544 MT 1000 Fc = (4544 - 1721) (17,000) +(20)(12000) Fe= 1052+240= 129216 F== 1292 16(T) SEE NOTE FOR P. 13.5

13.17 GIVEN: 2000-kg CAB 8000-leg TRAILER LEVEL GROUND . TRUCK COMES TO 90 km/h ASTOP IN 1200 M. 60% OF BRAKING FORCE FROMTEAILER 90% OFBRAKING FORCE FROM CAB FIND: (A) AVERAGE BRAKING FORCE WI AUGRAGE FORCE IN THE COUPLING TRAILER AND CAB I WE WIT v= (90 km/h)(1000 km) (1h 3600 s) U1= 25 m/s N= 0 T1+U1-2=T2 T,= 1 (m++ m=) (v,)=(1) (10,000 kg) (25 m/5) T1=3125×10 N-m T2= 0 3125×10-(1200m)(FB)=0 FB= \$125 X103 N·m) = 2604 N·m (1200 m) FB= 2.60 LN CAB CONSIDERED SEPERATELY 1 We -Fc T== 1 me(v,)=(1000)(25m/s) Ti = 625 X103 N.M , T2= 0 0.40 FB = 2604 N.M (FROM (AT) T,+U1-2= T2 625 x103-(0.40)(2604)(1200)+(FC)(1200)=0 Fe=(0,40) (2604)-625=1042-521 Fc= 521 N (c) SEE NOTE FOR PIB. 5 13.18 GIVEN: 2000 Lg CAB, 8000 kg TRAILER AVERAGE BRAKING FORCE BOOON 90 km/h LEVEL GROUND FIND: (a) DISTANCE K. TO COME TO A STOP (b) FORCE IN COUPLING FC TRAILER AND CAB WT U,= 25 m/s T1= 2 (m1+mc) (25)=3125x103 J T2=0 U1-2= FBX 3125×103-(3000) X = 0 X= 1042 M (b) TRAILER CONSIDERED SEPARATELY T= 1 m, (25)2=(4000)(625) Ti= 2500 x103 J T2 = 0 Ti+U1-2= T2 2500×103-(Fc)(x)=0 Feom (a) X= 1092 m 2500 x 103 - Fc (1042) = 0 FC= 2500 x 103 = 2399.2 N 1042 Fc=2,40kN(c) 4 SEE NOTE FOR PI3.5

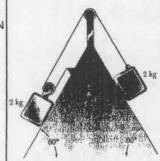


BLOCKS RELEASED FROM REST; NO FRICTION

(a) VELOCITY OF BLOCK B AFTER IT HAS MOUED

(b) TENGION IN THE CABLE.

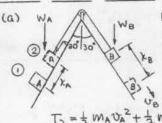
13.20



GIVEN:

BLOCKS RELEASED FROM REST; FRICTION 45= 0,30,46=0,20 FIND:

(a) VELOCITY OF BLOCK B AFTER IT HAS MOVED ZM. (b) TENSION IN THE CABLE.



KINGHATICS X8=ZXA NB= 2NA

A AND B ASSUME B MOVES DOWN U,=0 T,=0

T2 = 1 MAVA2 + 1 MBVB= 1 (2kg) (NB+NB) T3= \$ V2

U1-2=-mag(co=30)(x1)+mgg(cos30) 26 4B=2M 2A= 1m U1-2=(2)(981)(5)[-1+2]

U1-2 = 16.99 J SINCE WORK IS POSITIVE BLOCK B DOES MOVE DOWN

T1+U1-2=T2 0+16.99= = UB

NR = 13.59

UB= 3.69 m/s DOWN TO THE BIGHT

(b)

B ALONE

U,= 0 T,= 0 U2 = 3,69 m/s (FROM (a))

T2 = 1 MB U2 = 1 (2) (3.69) = 13.59 J

U1-2=(mgg)(cos 30°)(x8)-(T)(x8)

U1-2=[(2kg)(9.81m/52)(13)-(T)](2m)

U1-2= 33.98-ZT

TI+U1-2=T2 0+33.98-2T=13.59 2T= 33.98-13.59= 20.39

T= 10.19 N

2 BLOCK

CHECK AT O TO SEE IF BLOCKS MOVE, WITH HOTION IMPENDING AT B DOWNWARD DETERHINE REQUIRED FRICTION FORCE AT A FOR UB BRUILIBRIUM

= E F = No - (MB9) (SIN 30°)=0 NB=(29)(1)= 9 ΣF= T-(MB9) (CUS30°)+(FB)+=0 (FB)+= 45 NB=(0,30)(g) T= (2 g)(13/2)-(0.30)9 T= (13-0.30)(9)

BLOCK A < ZF= NA-(mag)(SIN 30°)=0 NA= (29)(2)=9 1 ZF= 2T-(mag) (cos 30) - (FA) = 0

SUBSTITUTE T FROM D INTO D (2) (FA)= (2)(13-0.30)(9)-139 (Fa)= (13-0.60) g = 1:132 g REQ. FOR EQUIL HAX FRICTION THAT CAN BE DEVELOPED AT A= 45 NA = 0,39

SINCE 0,39 < 1,1329; BLOCKS MOVE (a) A AND B

(FA)+=4kNB=(0.20)g (FA)+4kNA=(0.20)g KINEHATICS X8=2XA UB= 2VA

T2= \$ 082 MAUA+ 1 MB UB= (1) (Rkg) (4) U1-2=-MAQ(cos30)(2A)+MBQ(cos30)XB - (Fal (XA) - (Fo) + (Xa) KB=2M, KA=1M

U1-2=E(zkg)(13/2)(1m)+(zkg)(13/2)(2m) - (0,20)(1m)-(0,20)(2m)][9,81m/52]

U1-2=[(1732)-(0.6)][9.81] = 11.105 J

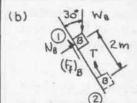
T,+U1-2=T2 0+11,105=1,25Vg

(CONTINUED)

Na= 8.88

UB= 2-98 m/s DOWN TO THE RIGHT

13.20 continued



B ALONE

 $N_1 = 0$ $T_1 = 0$ $V_2 = 2.98 \text{ m/s (FROM (a))}$ $T_2 = \frac{1}{2} M_B V_B^2 = (\frac{1}{2})(2)(2.48)^2$

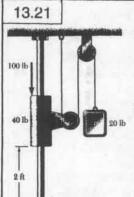
 $N_0 = M_8 g \sin 30^\circ = g N$ $T_2 = 8.88 J$ $U_{1-2} = M_8 g(\cos 30^\circ)(2) - (T1(2) - (F_8)_{\sharp}(2)$

U1-2=(2kg)(9.81 m/5²)(\$\frac{15}{2})(2m)-2T-(0.2)(9 K)(2M)

U1-2 = 2139-2T-0.69

 $T_1 + U_{1-2} = T_2$ 0 +259 -2T-0.49 = 8.88 2T= (213-0.4)(9)-8.88 = 21.179

T= 10.59 N



GIVEN:

SYSTEM AT REST WHEN 100 Ib FORCE 13 APPLIED TO, NO FRICTION, IGNORE PULLEYS HASS

FIND:

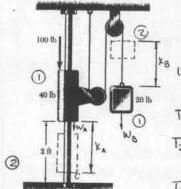
(A) VELOCITY, UA OF A

JUST BEFORE IT HITS C

(B) UA IF COUNTERWEIGHT

B IS REPLACED BY A

20-16 DOWNWARD FORCE



KINEMATICS XB=2XA

UB = 2 NA

(Q) BLOCKS A AND B

T2= 1 MOV8+ 1 MAUA

 $T_2 = \frac{1}{2} (20 \text{ lb}/32.2 \text{ ft/s})(24) (b) \frac{8 \text{ LOCK A}}{1 = 0}$ + $\frac{1}{2} (40 \text{ lb}/32.2 \text{ ft/s}^2)(24) T_1 = 0 T$

T= (60/32,2)(VA)2

U1-2=(100)(xA)+(WA)(xA)-(WB)(XB)

U1-2=(1001b)(2ft)+(401b)(2ft)-(201b)(4ft)

U1-2= 200+80-80= 200 16.ft

(CONTINUED)

13.21 continued

 $T_1 + U_{1-2} = T_2$ $0 + 200 = (60/32.2) V_A^2$ $V_A^2 = (07.33)$

UA=10.36ft/5 €

(b) SINCE THE 2016 WEIGHT AT B IS REPLACED

BY A 2016 FORCE THE KINETIC ENERGY

AT(2) 15 T2=1 MAVA2=1 (40/3)VA2 T1=0

THE WORK DONE IS THE

SAME AS IN PART (Q)

U1-2= 200 lb.ft

TitU1-2= T2

0+200=(20/g)UA2

NA=17.94 ft/s

13.22



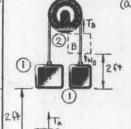
GIVEN:

MA= 11 kg MB= 5 kg
h= 2m
SYSTEM RELEASED FROM
REST
VA= 3 m/S TUST BEFORE
HITTING THE GROUND

FIND:

(a) ENERGY, Ep, DISSIPATED
IN FRICTION

(b) Tension in Each Poetion of coed



21 WA

(a) $U_1 = 0$ $T_1 = 0$ ENERGY $U_2 = U_A = 3 \text{ m/s} = U_B$ $U_2 = \frac{1}{2} (\text{ma+mg}) U_2^2$

 $T_2 = (\frac{16}{2} \text{ kg}) (3 \text{ m/s}) = 72 \text{ J}$

U1-2=Mxg(2)-meg(2)-Ep

U1-2=(6kg)(9.81m/52)(2m)-Ep

U-2= 117.72 - EP

 $T_1 + U_{1-2} = T_2$ 0 + 117.72 - Ep = 72

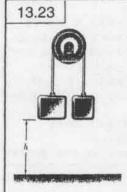
Ep=117.72-72= 45.7 J

b) <u>BLOCK A</u> $T_1 = 0$ $T_2 = \frac{1}{2} m_A U_2^2 = (\frac{11}{2} kg)(3 m/s)^2 + 49.5 J$ $U_{1-2} = (m_A g - T_A)(2) = [(11kg)(9.81 m/s^2) - T_A][2m]$ $U_{1-2} = 215.82 - 2T_A$

 $T_1 + U_{1-2} = T_2$ 0+215.82-2 $T_4 = 49.5$ $T_6 = 83.2 \text{ N}$

 $\frac{B LOCK B}{T_1=0} = \frac{1}{2} m_B V_2^2 = (\frac{5}{4} \frac{kg}{3})(3m/5)^2 = 22.5 J$ $U_{1-2} = m_B g(2) + T_B(2) = -(5 \frac{kg}{3})(9.81 \frac{m}{5}^2)(2m) + 2T_B$ $U_{1-2} = T_2 \qquad \qquad U_{1-2} = -98.1 + 2T_B$ $0 - 98.1 + 2T_B = 22.5$

TB= 60.3 N



GIVEN: WA= 2016; WB= 816 h= 1.5 ft

SYSTEM RELEASED
FROM REST
BLOCK A HITS THE
GROUND WITHOUT

REBOUND BLOCK B REACHES A HEIGHT OF 3.5ft

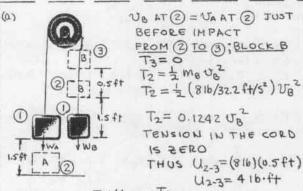
FIND:

(a) Up JUST BEFORE

BLOCK A HITS THE GROUND

(b) ENERGY, Ep, DISSIPATED

BYTHE PULLEY IN FRICTION



 $T_2 + U_{2-3} = T_3$ 0.1242 $V_B^2 = 4$ $V_B^2 = 32.2 = V_A^2$

UA=5.68 ft/5

(b) FROM () TO (2)
BLOCKS A AND B

 $T_1 = 0$ $T_2 = \frac{1}{2} (m_A + m_B) U_2^2$ JUST BEFORE IMPACT, $U_2 = U_B = U_A = 5.68 \text{ ft/s}$ $T_2 = \frac{1}{2} (28 \text{ lb/32.2 ft/s2}) (5.68)^2$

T2= 14 16.f+

(Ep = ENERGY DISSIPATED BY PULLEY)

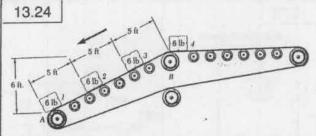
U1-2= (1216)(1.5ft) - Ep = 18-Ep

T1+41-2 = T2

0 + 18-Ep= 14

-Ep= 14-18

Ep=4.00 ft.16 ◀



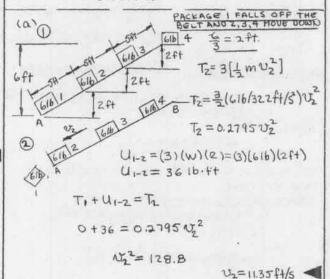
GIVEN:

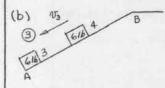
CONVEYOR IS DISENGAGED, PACKAGES
HELD BY FRICTION AND SYSTEM IS
BELEASED FROM REST. NEGLECT MASS
OF BELT AND COLLERS PACKAGE I
LEAUES THE BELT AS PACKAGE & CONES
ONTO THE BELT.

FIND:

(a) VELOCITY OF PACKAGE 2 AS IT LEAVES THE BELT AT A

(b) VELOCITY OF PACKAGE 3 AS IT LEAVES THE BELT AT A.



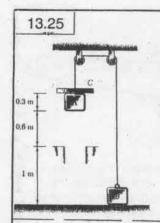


PACKAGE 2 FALLS
OFF THE BELT AND
ITS ENERGY IS LOST
TO THE SYSTEM
AN 3 AND 4 HOUE
DOWN 2 H.

 $T_{2}' = (2)[\frac{1}{2}m\sqrt{3}^{2}]$ $T_{3}' = (61b/32.2 ft/5^{2})(128.8)$ $T_{3} = (2)[\frac{1}{2}m\sqrt{3}^{2}]$ $T_{3} = (61b/32.2 ft/5^{2})(\sqrt{3}^{2})$ $T_{3} = 0.18634\sqrt{3}$ $U_{2-3} = (2)(w)(2) = (2)(61b)(2ft)$ $U_{2-3} = 24 \cdot 16 \cdot ft$ $T_{2} + U_{2-3} = T_{3}$

 $24+24=0.18634N_3^2$ $N_3^2=257.6$

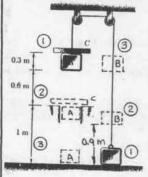
U3=16.05 ft/s ◀



GIVEN: ma= 4 kg mg=5kg mc= 3 leg SYSTEM RELEASED FROM REST

FIND:

UM, JUST BEFORE IT STRIKES THE GROUND



POSITION 1 TO POSITION 2 U,=0 T=0 AT 2 BEFORE C IS REMOVED FROM THE SYSTEM T2=1 (MA+MB+Mc) V2 T2= = (12kg) V2=6V2 U1-2=(MA+Mc-MB)q(9m)

(1-2= (4+3-5)(9)(9m)=(2kg)(9.81m/5)(Am) U1-2= 17.658 J

T1+41-2=T2

0+17.658=602

U2= 2.943

AT POSITION 2, COLLAR C IS REHOUED FROM THE SYSTEM

POSITION 2 TO POSITION 3 T2'= = (MA+MB) (2=(9 kg)(2.943) Ta'= 13.244T

T3= = (MA+ MB) (V3)= 9 032

U2-3= (ma-mo)(g)(0.7m) = (-1 kg)(9.81m/s2)(0.7m)

Usi-3= -6.867 J

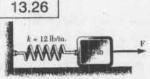
T2 + U2-3=T3

13.244-6.867= 4.5 02

U3= 1.417

VA = U3 = 1.190 W/S

UA= 1.190 m/s

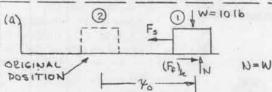


GIVEN: 45= 0.60, 46= 0.40 FORCE F IS SLOWLY APPLIED UNTIL THE TENSION IN THE SPRING IS 20 16

AND THEN DELEASED

FIND:

(a) VELOCITY OF BLOCK AS IT RETURNS TO ITS ORIGINAL POSITION (b) THE MAXIMUM VELOCITY OF THE BLOCK



FIND INITIAL POSITION XO, OF THE BLOCK AT() b= 12 16/m = 144 16/ft

AT 1, FS = 2016 Fs=12x0 2016=(14416/ft) X8 70= 20/144=0.1389 ft

 $T_1 = 0$, $T_2 = \frac{1}{2} \left(\frac{1}{8} \right) V_2^2 = \left(\frac{1}{2} \right) (10 \text{ lb/32.2ft/s}^2) V_2^2$ $T_2 = 0.1553 V_2^2$

U1-2= [-Fsdx+(F+), (+x0); F5=lex=144x U1-2= [-144x2]+(Ff) (Ff) (=(0.4)(10) U1-2= (72/b/f+) (0.1389f+)2+ (416)(-0.1389ft)

U1-2= 1.389 - 0.5556= 0.8335 16.ft

TI+ U1-2=T2 0+0,8335= 0,1553 172 U2= 5.367 102= 2.32 ft/s

AT ORIGINAL POSITION, U=2.32ft/s

(b) FOR ANY POSITION (2) AT A DISTANCE SUMMED TAMES OF THE ORIGINAL POSMOUS

U1-2'= f-Fsdx + J(Fx)adx Ko= 0.1389

(1-2'= [-144x2] + (Ff) (x-x0) (F) = 416

TI+U1-21=T2 0+(7216/4+)(0.1389)2- x2]+(416)(x-0.1389) = 0.1553 U2

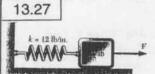
MAX V WHED - 144744= 0

MAX U. WHEN X = 0.027778 M

0.1553 Umax = (72) ((0.1389) - (0.02778)]+(4) (.02778-01389)

0.1553 Umax = 1.3336 - 0.4445 = 0.8891 W2max = 5.725

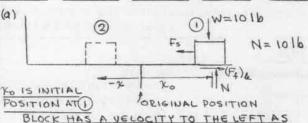
Umax= 2.39 ft/s -



45=0.60, 4k=0.40
FORCE F IS SLOWLY
APPLIED UNTIL THE
TENSION IN THE
SPRING IS 2016
AND THEN RELEASED

FIND:

(a) DISTANCE THE BLOCK MOVES TO THE LEFT BEFORE COMING TO A STOP
(b) WHETHER THE BLOCK THEN MOVES BACK TO THE RIGHT.



BLOCK HAS A VELOCITY TO THE LEFT AS

(SEE P 13.26)

$$K = 12 lb/(n) = 144 lb/ft$$

$$T_1 = 0 \quad T_2 = 0 \qquad F_5 = 144 \times 0$$

$$U_{1-2} = \int_{-F_5}^{-F_5} dx + \int_{-X}^{-F_6} (F_f)_b dx \qquad (F_f)_b = 0.4)(10)$$

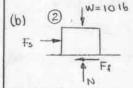
$$V_0 = -144 v^2 + (F_6)_6 (-X - X_0)$$

 $T_1 + U_{1-x} = T_2$ 0-72 (x-x₀)(x+x₀)-4(x+x₀)=0 (b)

 $-72(x-x_0)-4=0$ AT ① $F_S=201b$ $-72x=4-72x_0$ $F_S=kx_0=144x_0$ $x_0=\frac{20}{104}=0.0833ft$

TOTAL DISTANCE MOVED TO THE LEFT = 10+12

 $V_0+x=0.1389+0.0833$ $V_0+x=0.2225t$

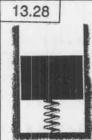


IF FS AT 2 IS LARGER
THAN THE MAXIMUM
POSSIBLE STATIC
FRICTION FORCE, THEN
BLOCK WILL MOVE TO
THE RIGHT

FROM (a) WITH X= 0.0833 ft

 $F_S = (144)(0.0833) = 12 \text{ lb}$ $(F_f)_S = 4/5 N = (0.60)(10) = 6 \text{ lb}$

SINCE F5>(F4)5
BLOCK MOVES TO THE RIGHT

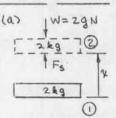


GIVEN:

3kg BLOCK RESTS ON 2kg BLOCK WHICH IS NOT ATTACHED TO A SPRING OF CONSTANT 40 W/M UPPER BLOCK IS EUROBENLY REHOUED

FIND:

(a) Nomax OF 2 kg BLOCK
(b) MAXIMUM HEIGHT, h.
REACHED BY THE 2 kg BLOCK



AT THE INITIAL POSITION ()
THE PORCE IN THE SPRING
EQUALS THE WEIGHT OF
BOTH BLOCKS, i.e. 59 N
THUS AT A DISTANCE X
THE FORCE IN THE SPRING
IS FS= IG-LX

FS= 59-40X

MAX VELOCITY OF THE 2 kg BLOCK OCCURS

WHILE THE SPRING IS IS STILL IN CONTACT

WITH THE BLOCK.

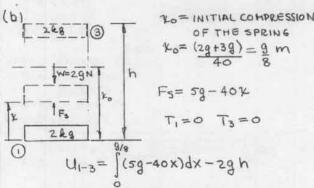
T=0 T3= \(\frac{1}{2} \) (2 kg) (\(\mathcal{U}^2 \))=\(\mathcal{U}^2 \)

 $T_1 = 0$ $xT_2 = \frac{1}{2} m v^2 = (\frac{1}{2})(2kg)(v^2) = v^2$ $U_1 - 2 = \int (5g - 40x) dx - 2gx = 3gx - 20x^2$

 $T_1 + U_{1-2} = T_2$ 0 + 392-202 = $U^2 = U^2$ (1) MAX U WHEN $\frac{dU}{dx} = 0 = 39 - 46$ $\frac{39}{4}$ m

SUBSTITUTE IN (1) 7. (MAX 0) = 0,7358 M VHAX = (3)(9.81)(0.7358)-(20)(0.7358)2 = 21.65-10.83=10.83

Umax= 3.29 m/s ◀



$$U_{1-3} = \frac{59^2 - 209^2 - 29h}{64} - 29h$$

$$T_1 + U_{1-3} = T_3 \qquad 0 + \frac{209^2 - 29h}{64} = 0$$

$$h = \frac{109}{64} = \frac{(10)(9.81)}{64}$$

h= 1.533 m

13.29

GIVEN:

3 kg BLOCK RESTS ON A Z kg BLOCK WHICH IS ATTACHED TO A SPRING OF 40 N/M WHEN UPPER BLOCK IS SUDDENLY REMOUED

EIND:

(a) Nmax OF 2 kg BLOCK (b) MAXIMUM HEIGHT h REACHED BY 2kg BLOCK

(a) SEE SOLUTION TO (a) OF P13.28

Umax=3.29 m/s

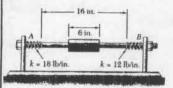
(b) REFER TO FIGURE IN (b) OF PI3.28 T1=0 T3=0 U1-3=) (5q-40x)dx-2gh

SINCE THE SPRING REMAINS ATTACHED TO THE 2 kg BLOCK THE INTEGRATION HUST BE CARRIED OUT THROUGHOUT THE TOTAL DISTANCE h.

T+U+3=T2 O+5gh-20h-2gh=0 $h = \frac{39}{20} = (3)(9.81)$

h=1.472 m

13.30



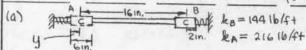
GIVEN:

WE= BID COLLARC COMPRESSES SPRING AT B ZIM. AND. IS DELEASED

FIND:

(a) DISTANCE TRAVELED BY COLLAR WITH NO FRICTION.

(b) SAME AS (Q) WITH FRICTION, 4=0.35



SINCE COLLAR C LEAVES THE SPRING AT B AND THERE IS NO FRICTION IT MUST ENGAGE THE SPRING AT A

TA+UA-B=TB 0+2-1084=0

y= 0.1361 ft=1.633 in. TOTAL DISTANCE = 2+16-(6-1.633)=13,63 M.

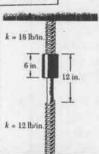
continued 13.30

(b) ASSUME THAT C DOES NOT REACH THE SPRING AT B BECAUSE OF FRICTION

0+2-2804=0 TA+UA-0=TD y= 0.7149+= 8.57 in.

THE COLLAR MUST TRAVEL 16-6+2= 12 In. BEFORE IT ENGAGES THE SPRING AT B. SINCE Y = 8.57 IN.
IT STOPS BEFORE ENGAGING THE SPRING AT B TOTAL DISTANCE = 8.57 IN.

13.31



GIVEN:

Wc=61b. UPPER SPRING IS COMPRESSED ZIN AND COLLAR C IS RELEASED

FINO:

(a) Im, THE HAXIMUM DERECTION OF THE LOWER SPRING (b) Um, THE MAXIMUM VELOCITY OF THE COLLAR

18 1b/in. C 121b/in. (Fe)2

SPRING CONSTANTS 1816/in = 216 16/ft 12 lb/in = 144 lb/ft

MAXIMUM DEFLECTION AT (2) WHEN VELOCITY OF COLLAR C 15 ZERO 102=0 T2=0 N=0 T=0

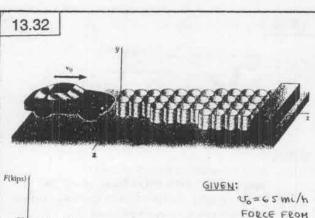
U1-2=Ue+Ug= (Fe),dx- (Fe)2dx+Wc(1+4)

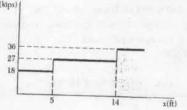
U1-2=(216/6+)(1/6+t)-(144/6/ft)(ym)+616(1+4) U1-2= 3-72 42+6+64=-72 42+64+6

Ti+41-2=T2 0 - 724m+64m+6=0

ym= + ft = 4.00 in.

(b) MAXIMUM VELOCITY OCCURS AS THE LOWER SPEING IS COMPRESSED A DISTANCE 41 THU1-2=T2 10-7242+64+6=(0.09317) 12 -144 y+ 6=0; y'= 0.041667ft SUBSTITUTE 41= 0.041667ft - 0.125 + 0.250+6=0.093170m; Um= 811ft/s=973m/s"

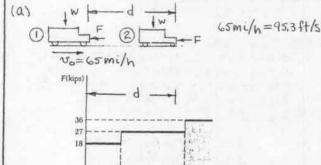




US=65 MI/h
FORCE FROM
CUSHION AS
SHOWN
NEGUECT
FRICTION
W=2250 Ib

EIND:
(a) DISTANCE d, FOR AUTOHOBILE TO COME
TO REST

(b) HAXIMUM DECELERATION, OD



ASSUME AUTO STOPS IN SEDELA ft

U1 = 95.33 ft/s T1 = 2 M U1 = 1 (2250 16) (95346)

Ti= 317,530 lb.ft=317.53k.ft

U2=0 T2=0

 $U_{1-2} = (18k)(5ft) + (27k)(d-5)$ = 90 +27d-135 = 27d-45 k.ft $T_1 + U_{1-2} = T_2$

317.53=270-45

d= 13.43ft

ASSUMPTION THAT d < 14 ft 15 OF (b) MAXIMUM DECELERATION OCCURS WHEN F IS LARGEST. FOR d = 13.43 ft, F= 27k.

THUS $F = MQ_D$ $(27,000 lb) = \left(\frac{2250 lb}{32.2 Pt/s^2}\right) \left(Q_D\right)$

ap = 386 ft/52

13.33

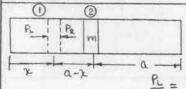
GIVEN:

PISTON AREA A
PISTON MASS M
INITIAL PRESSURE P
PRESSURE VARIES
INVERSELY WITH
VOLUME. PISTON

MOVED 42 AND RELEASED

EIND:

VELOCITY OF THE PISTON AS IT RETURNS



PRESSURES VARY INVERSELY AS THE VOLUME

THITIALLY AT () $\frac{P_R = A \alpha}{P} = \frac{P \alpha}{A(\overline{z}\alpha - x)} P_R = \frac{P \alpha}{(z\alpha - x)}$ $N=0 \quad x = \frac{\alpha}{2}$

T=0 AT ②, $x=\alpha$, $T_2=\frac{1}{2}mv^2$ $U_{1-2}=\int_{\alpha/2}(p_1-p_2)Adx=\int_{\alpha/2}p_0A\left[\frac{1}{x}-\frac{1}{2\alpha-x}\right]dx$

 $U_{1-2} = paA \left[ln x + ln(2a-x) \right]_{a/2}^{a}$

 $U_{1-2} = paA \left[ln a + ln a - ln(a/z) - ln(3a/z) \right]$

U1-2 = paA[ln a²-ln 3a²/4] = paAln(4/3)

 $T_1+U_{1-2}=T_2$ $O+paAln(4/3)=\frac{1}{2}mv^2$ $v^2=\frac{2paAln(4/3)}{m}=0.5754paA$

U=0.759 V-Pan

13.34 F h

GIVEN:

ACCELERATION OF GRAVITY 90 AT EARTHS SURFACE

F(UD:

ACCECTERATION OF GRAVITY

9h AT HEIGHT h ABOUF

THE EARTHS' SURFACE

INTERMS OF G. P.

AND FREDR IN WEIGHT AT IN IF WEIGHT AT FARTHS' SURFACE IS USED FOR (a) h=1 km

 $E = \frac{(\nu + E)_5}{(\nu + E)_5} = \frac{(\nu + E)_5}{(\nu + E)_5} = md\nu$

ATEARTHS' SURFACE (h=0) GHEM/R=mgo

GMe/ $2^2 = 90$ $90 = \frac{GMe/R^2}{(\frac{h}{R}+1)^2}$ THUS $90 = \frac{90}{(\frac{h}{R}+1)^2}$

(CONTINUED)

13.34 continued | R = 6370 km

AT ALTITUDE h "TRUE" WEIGHT F= M 9h=WT ASSUMED WEIGHT WO = mgo - mgh = go-go
ERROR = E = Wo-WT = mgo - mgh = go
Wo $g_h = \frac{g_0}{(\frac{h}{E} + 1)^2}$ $E = g_0 - \frac{g_0}{(1 + \frac{h}{E})^2} = \left[1 - \frac{1}{(1 + \frac{h}{E})^2}\right]$

(b)
$$h = 1000 \text{ km } P = 100 \left[1 - \frac{1}{\left(1 + \frac{1000}{6370}\right)^2}\right]$$

P= 25.3%

13.35

GIVEN:

VELOCITY AT MOONS SURFACE = U. VGLOCITY AT HEIGHT h= U RADIUS OF THE MOON, PM ACCELERATION OF GRAVITY ON THE MOONS' SURFACE, gm

FIND:

FORMULA FOR hu/hu. WHERE HA IS FOUND USING NEWTONS LAW OF GRAVITATION AND HU IS FOUND USING A UNIFORH GRAVITATIONAL FIELD

NEWTONS LAW OF GRAVITATION

$$T_{1} = \frac{1}{2} m v^{2}, T_{2} = \frac{1}{2} m v^{2}$$

$$U_{1-2} = \int_{C} -F_{1} dr \qquad F_{1} = \frac{m g m P^{2} m}{r^{2}}$$

$$P_{1} \qquad P_{1} \qquad P_{2} \qquad P_{2} \qquad P_{3} \qquad P_{4} \qquad P_{4} \qquad P_{4} \qquad P_{4} \qquad P_{5} \qquad P_{$$

U1-2=m3m 2m (Pm - 1 Rm+h) 1 mvo2+mgm(Pm-Pm)=1mv2

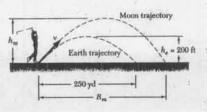
$$h_{n} = \underbrace{(v_{o}^{2} - v^{2})}_{2gm} \left[\frac{P_{m}}{P_{m} - (v_{o}^{2} - v^{2})} \right]$$
 (1)

UNIFORM GRAVITATIONAL FIELD

T= 1 m No2 T2= m N2 (1-2=) (-Fudr=-mgm(Rm+ho-Pm)=-mghu

12 mv3-mghu=12 mv2 hu=(v32-v2)/2gm (2)

DIVIDE (1) BY (2) hn = 1= (Vo2-v2)/2gmPm) 13.36

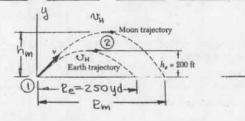


GIVEN:

EARTH TRAJECTORY AS SHOWN MAGNITUDE AND DIRECTION OF WON THE EARTH IS THE SAME ON THE HOOM TRAJECTURY IS A PARABOLA gm= 0.165 ge

FIND:

RANGE PM OF THE BALL ON THE MOON



SOLVE FOR hm

AT HAXINUM HEIGHT THE TOTAL DELOCITY IS THE HORIZONTAL COMDONENT OF THE VELOCITY WHICH IS CONSTANT AND THE SAME IN BOTH CASES T1= 2 m 0 2 T2= 2 m 0 H2

U1-2 =-mgehe EARTH U1-2 =-mgmhm MOON

EARTH & MUZ-mgehe= & MUHZ

MOON 5 MO2- mgmm=5 mJz.

SUBTRACTING -gehetgmhm=0 hm = ge gm hm=(200ft) (ge 0.165ge)= 1212ft

(y-he)=-Ce(x-Pe)2 EARTH

(y-hm) =- Cm (x-2m) 2 HOON

AT X=0, U IS THE SAME THUS dy IS THE SAME

dy = Ce Re = Cm Rm

dx = 0

Ce = Rm

Cm Rm

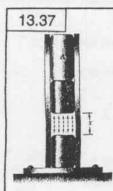
AT X=0, y=0 he = Ce Re hm=Cm Rm

T

he = Cm Em2 = Em Re

hm = ge = Rm Rm = (ge/10.165 ge)(250yd)

2m= 1515 yd 4



ma= 300-9 (NON HAGUETIC) MB= 200-9 (MAGNETIC) K= 4 mm, INITIALLY REPELLING FORCE BETWEEN BAND C 15 F = K/22 BLOCK A IS SUDDENLY REMOVED . NO AIR RESISTANCE

FIND:

(A) HAXIMUM VELOCITY UmoFB (b) MAXIMUM ACCELERATION am of B

13.38

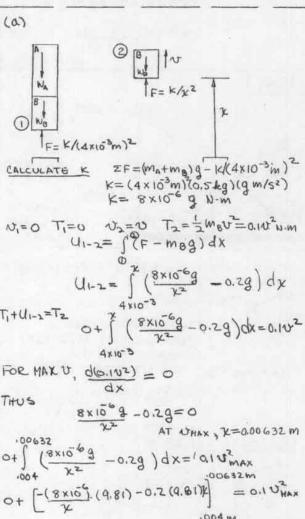
GIVEN:

WB= 0.4-16 (MAGNETIC) WA = 0.6-16 (NON-MAGNETIC) X=0.15 IN. INITIALLY REPOLLING FORCE BETWEEN BAND C 15 F= K/X2 NO AIR RESIST NEE BLOCK A IS PLACED ON BLOCK B AND RELEASED

FIND:

(a) MAXIMUM UELOCITY OF AANDB

(b) MAXIMUM DEFLECTION OF AANDB



M 400.0 = X TA NOTTAFFEE ANNIXAM (d)

WHEN EF ARE THE GREATEST

(8×106)(9.81)/(0.004)2-(0.2)(9.81)=(0.2)QM

ZF=K/x2-WB=MBQ

(a) AT Non ? 0.007906

CALCULATEK EQUILIBRIUM AT (T) X = K/X2-W8=0 4=0.15in = 0.0125ft K= (.0125ft) (0.41b) K=0.0000625 ft3.16 1=0 T=0 UZ=V TZ=I (MA+MB) V T2=2 (116 1/52) 12 T2= 0.01553 12 [F-(WA+WB)]dx 1.0000 625 - 1 dx = 0.01553 0 FOR MAX U d (.001553 U2) 0.0000625-1=0 X=0.007906ft 0.0000625 -17dx= 0.01553Vm Um = 0.10876 Um= 0.3298 ft/s Nm= 3,96 m/s (b) MAXIMUM DEFLECTION WHEN U=0 $T_1=0$ $T_2=0$ $X_1=0$ $Y_2=0$ $Y_2=0$

 $-0.0000625 \left[\frac{1}{x} - \frac{1}{0.0125} \right] - x + 0.0125 = 0$

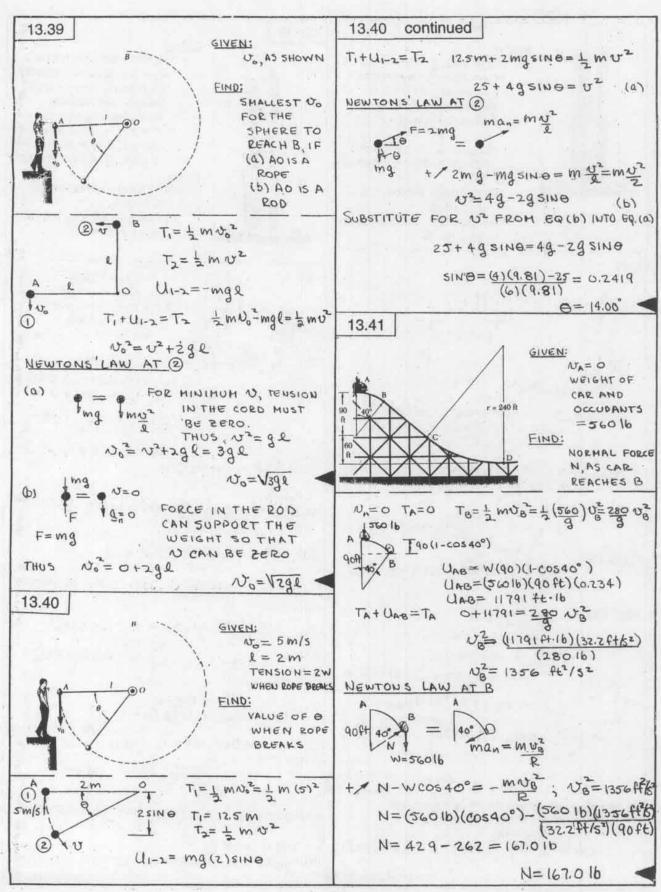
MAXIMUM DEFLECTION = 0.0125-0.005=0.0075 Pt

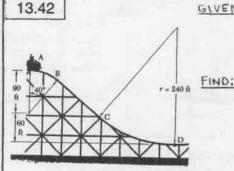
= 0.090 in.

K=0.005ft

NHAX = 0.1628 M/S NH = 162.8 mm/s

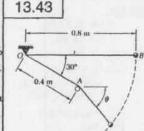
am= 14.72 m/s2





UA= O, CAR AND O CCU PANTS WEIGH 560 16

MAXIMUM NHAM MUMININ DIA NMIN NORMAL FORCE ON THECAR AS IT GOES FROM A TO D



GIVEN:

SPHERE RELEASED FROM REST AT B. (UR=0)

FIND: CORD,

TENSION IN THE (a) JUST BEFORE IT COMES IN CONTACT WITH THE PEG (b) JUST AFTER CONTACT WITH PEG

NORMAL FORCE AT B

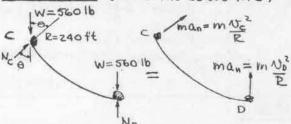
SEE SOLUTION TO PROB. 13.41, Na=167.016

NEWTONS LAW

FROM B TO C (CAR HOVES IN A STRAIGHT LIVE) + NB-WC0540=0 ma No=(560) cos 40°

NB=42916

AT C AND D (CAR IN THE CURUE ATC)



+ A No-Wcose = W De Nc= 560 (coso+ 122)

AT D

$$+1 N_0 - W = + \frac{W}{3} \frac{V^2}{2}$$
 $N_0 = 560 \left(1 + \frac{V^2}{9R}\right)$

SINCE VO > VC AND COSOCI, NO > NC WORK AND ENERGY FROM ATO D

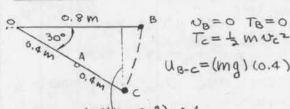
UA-B= W (90+60)=(560 16)(150 ft) UA-B= 84000 lb.ft

0+84000=280 ND2 TA+UA-B=TB N= 300

No= 560 (1+ 302)= 560 (1+300)= 1260 lb

NNIN= NB= 167.010; NHN=N0=126016

VELOCITY OF THE SPHERE AS THE CORD CONTACTS A



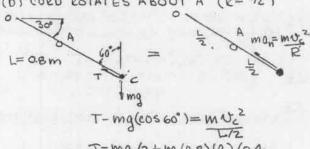
(0.8)(SIN 30°)=0.4 TB+UB-=Tc 0+0.4 mg = 1 muc2

U==(0.8)(9)

NEWTONS LAW CORD ROTATES ABOUT POINT O (a) 0-30 (R=L) L= 0.8m man=mvc + T-malcos60 = MNE

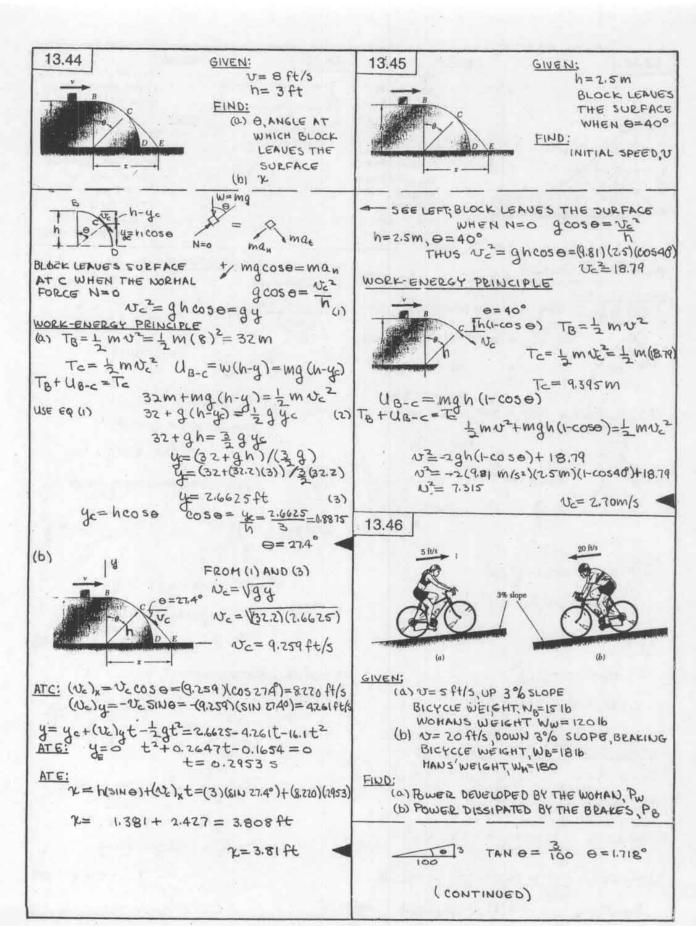
T= mg(cos.60°)+m (0.8) g T= 3 mg T= 1.5 mg

(b) CORD ROTATES ABOUT A (R= 1/2)



T=mg/2+m(0.8)(g)/0.4 T= (+2)mq = = mg

T= 2,5 mg



13.46 continued

W=W8+Ww=15+120 W=13516 Pw=W·Y=(WSINBKU) Pw=(1351(SIN1718°)(5) Pw=20.24 ft:16/5

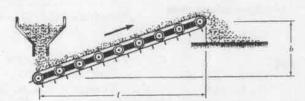
Pw= 20.2 ft.16/5

(b) 10 W W 20 IVs

W= WB+ WM= 18+180
W= 19816

BRAKES HUST
DISSIPATE: THE
POWER GENERATED
BY THE BIKE AND THE
MAN GOING DOWN
THE SLOPE AT
20 ft/s
PB=W.V=(WSINB)(U)
PB=(198)(SINITIB)(20)

13.47



GIVEN:

(a) MASS FLOW RATE, M(KG/H), L(M), b(M) (b) MASS FLOW RATE, W(TONS/H), L(ft), b(ft) MOTOR EFFICIENCY N

FIND:

(a) POWER PINKW (b) POWER IN MP

(a) MATERIAL IS LIFTED TO A HEIGHT B AT

A RATE, (M Kg/H) (g m/s2) = (mg (D/H))

THUS DU = [mg (D/H)] (b(m)) = (mg b) N·M

[1000 N·M/S = 1 kw (3600 S/H) = (3600) N·M

THUS, INCLUDING TOTOR EFFICIENCY of

P(kw) = mg b (N·M/S)

(3600) (1000 N·M/S) (n)

P(kw) = 0.278×10 - 6 mg b

N

(b) Au = [w(tons/H) (2000 lb/Ton) [b(ft)]

3600 S/H

= Wb ft·1b/S; 1 hp = 550 ft·1b/S

WITH M, hP = (W. b) (+16/5))(1 hP (550+16/5))(1) = 1.010×10/4/6

13.48



GIVEN:

2000 ID CAR EGAR WHEEL DRIVE,
SKIDS FOR FIRST 60ft WITH FRONT WHEELS
OFF THE GROUND, ME = 0.60
ROLLS WITH SLIDING IMPENDING FOR
REMAINING 1260 Ft WITH 60% OF
ITS WEIGHT ON REAR WHEELS, 45=0.85

FIND:

(a) HP DEVELOPED AT END OF GOFT
PORTION OF THE RACE
(b) HP DEVELOPED AT THE END OF THE
RACE

(a) FIRST 60 ft (CALCULATE VELOCITY AT 60 Pt)

FORCE GENERATED BY REAR WHEELS =4KW

SINCE CAR SKIDS, THUS F=(0.6)(2000 16)

F= 1200 16

WORK AND ENERGY T=0 T= 1 W V= 1000 V60 T+U1-2=T2 (F)(60ft)=(12001b)(60ft)=12,000 lbft

 $T_1+U_{1-2}=T_2$ $0+72,000=\frac{1000}{9}$ 0_60^2 $0_60^2=(72)(322)=2318.4$ $0_60=48.15$ ft/s

P= (120016) (48.15 ft/s)
P= 57780 ft.16/s

1 hp=550 ft.16/s hp=(57780ft.16/s)=105.1

(b) END OF PACE (CALCULATE VELOCITY AT 1320 ft)

FOR FIRST GOFT, FORCE GENERATED

BY REAR WHEELS - FS = 1200 ib (SEE &))

FOR REMAINING 1260 ft WITH 60%

OF WEIGHT ON REAR WHEELS, THE

FORCE GENERATED AT IMPENDING

SLIDING IS 45 (.60) (W=0.85) (0.60) (2000)

FI = 1020 lb

 $T_1 + U_{1-2} = T_2$ $T_1 = 0$ $T_2 = \frac{1}{2} \frac{1}{3} \frac{1}{320} = \frac{1000}{9} \frac{1}{320} \frac{1}{320}$

U1-2=(F5)(60ft)+(FI)(1260ft) U1-2=(12001b)(60ft)+(10201b)(1260ft) U1-2=1,357,2001b.ft

0+1,357,200=1000 0,320

POWER = FT V1320 = 209 ft/s

P=(1020'16)(209 Pt/s)=218,230 Ft/6

hp = (213,200 ft.1b/s) = 388



1000kg CAR, REAR WHEEL DRIVE

SKIDS FOR FIRST 20 M, WITH FRONT WHEELS

OFF THE GROUND, UK = 0.68

ROLLS WITH SCIDING IMPENDING FOR

REMAINING 380 M WITH 80% OF ITS

WEIGHT ON REAR WHEELS, US = 0.90

FIND:

(a) POWER DEVELOPED AT END OF 20 m (1) kwsho)
(b) Hower Developed at end of the pace (kwsho)

(a) FIRST ZOM (CALCULATE VELOCITY AT ZOM)
FORCE GENERATED BY REAR WHEELS= 4KW

SINCE CAR SKIDS, THUS FS=(0.68)(1000)(9)
FS=(0.68)(1000kg)(981 M/52)=6670.8 M

WORK AND ENERGY TI=0 TZ=1 MN2=500020

TI+UI-Z=TZ

 $U_{1-2} = (20 \text{ m})(F_5) = (20 \text{ m})(6670.8 \text{ H})$ $U_{1-2} = 133420 \text{ T}$

0+133,420 = 500 020 020 = 133,420/500 = 266.83

POWER = (F5) (V20) = (6670.8 N) (16,335 m/s)
POWER = 108,970 J/s=108,97 kJ/s

1 RJ/5=1 kw

1 hp=0.7457 kw Power=109.0 kt/s=109.0 kw Power=109.0 kw = 146.2 hp (0.7457 kw/hp)

(b) END OF RACE (CALCULATE VELOCITY AT 400 M)

FOR REMAINING 380 M, WITH

80% OF WEIGHT ON REAR WHEELS

THE FORCE GENERATED AT IMPEDIALIS

SCIDING IS (45)(0.80)(1000 kg)(9.81 m/s)

FT= 7063.2 N

WORK AND ENERGY, FROM 20 M(2) TO 28 M(3)

U2= 16.335 M/5 (FROM PART (0))

T2=1 (1000 kg)(16.335 M/5)²

T2=133420 T

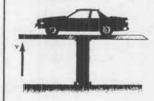
T3=1 MN300= 500 U30

 $U_{2-3} = (F_1)(380 \text{ m}) = (7063.2 \text{ N})(380 \text{ m})$ $U_{2-3} = 2,684,000 \text{ T}$ $T_2 + U_{2-3} = T_3$

 $(133,4201)+(2,684,0001)=5000_{30}^{3}$

POWER = (FI)(V30)=(7063.2N)(75.066 W/S) = 530,200 J

hw POWER = 530,200] = 530 kw hp POWER = 530 kw = 711 hp (0,7457 kw/hp) 13.50



GIVEN:

CAR MASS, Mc= 1200 log LIFTHASS, Mc= 300 log SYSTEM RISES 2.8 M IN 155.

FIND:

(a) AUGRAGE POWER OUT PUT OF PUMP, (PD)A (DIAUTE AGE ELECTRIC POWER, (PE)AWITH 17 = 82%

(a) (Pp) = (F) (Un) = (mc+mc)(g) (Un) Un = 5/t = (28m) /(55)=0.18667m/s) 5 (Pp) = (1760/29+(300/29)](9.81 m/s2)(0.18667m/s) 5

(b)(Pe)=(Pp)/12=675 kw/10.82)

(PE) = 3.35 kw

13.51



GIVEN:

CAR MASS, Mc= 1200 lg

LIFT MASS, Mc= 500 lg

PEAK VELOCITY AT

HID HEIGHT IN

7.5 S INCREASING

UNIFORMLY, VELOCITY

DECREASES UNIFORMLY

TO O, IN ANOTHER

7.55

FIND:

HAXINUM LIFTING

7.05
PEAK DUMP POWER,
P=6 kW, WHEN
VELOCITY IS HAXINUM

NEWTONS LAW

1200 g 12000 Mg = (Mc+Hr) g = (1500+300) g

SINCE MOTION IS UNIFORMLY ACCELERATED

Q= CONSTANT

THUS, FROM (1), F IS CONSTANT
AND PEAK POWER OCCURS
WHEN THE VELOCITY IS A
STANT MAXIMUM AT 7.55.

O CONSTANT MAT

Q = NMAX

P=(6000W)=(F)(UHAX) NHAX=(6000)/F

THUS a = (6000)/(7.5)(F) (2)

SUBSTITUTE (2) INTO (1)
F-1500g = (1500) (6000) /(7.5)(F)
F2-(1500kg) (9.81 m/s) F- (1500kg) (6000 N·m/s) = 0)

F= 14,800 N

F=14.8 &N

13.52 GIVEN:

W= 100 TONS P= 400 hp U= 50 mi/h CONSTANT

FIND:

(a) F, FORCE NEEDED TO QUERCOME AXLE FRICTION, ROLLING RESISTANCE AND AIR RESISTANCE

(b) AP, ADDITIONAL MP TO HAINTAIN THE SAME SPEED UP A 1-PERCENT GRADE

(a) $P = 400 \text{ hp} = (550 \frac{\text{f+.lb}}{5} / \text{hp})(400 \text{ hp}) = 220,000 \frac{\text{f+.lb}}{5}$

N= 50 mi/h= 73.33 ft/s

P= F. V

F= P/v = (220,000 f+1b)/(13.33 f+)

(b) 0 W = 0.333 ft/s $0 = 100 \text{ W} = 0.573^{\circ}$

W=(100 TONS)(2000 16/100)
W=200,000 16

F= 3000 lb .

ΔP=WSINO.V ΔP=(200,000 lb)(SIN.573°)(73.33 ft/s)

AP= 146,667 ft: 16/5 AP= 267 hp

13.53 GIVEN:

W= 600 TONS

UNIFORM ACCELERATION

FROM OTO SOHILL IN 405

CONSTANT SONUL AFTER 405

HORIZONTAL TRACK

FR, FRICTION AND ROLLING

RESISTANCE = 3000 lb

UEGLECT AIR RESISTANCE

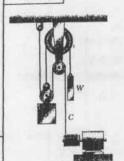
FIND:

P. POWER REQUIRED AS A FUNCTION OF TIME t.

v = 50 mi/h = 73.33 ft/s w = (600 Tous) (2000 lb/Tou) w = 12.00,000 lb $v_0 = 73.33 \text{ ft/s}$ $v_0 = 73.33 \text{ ft/s}$ $v_0 = 73.33 \text{ ft/s}$ $v_0 = 73.33 \text{ ft/s}$ FOR UNIFORM HOTION $v_0 = 73.33 \text{ ft/s}$ v_0

F= 71,311 lb

 $P = F \cdot V = (71,311)(1.833t) = 130,710t \frac{16 \cdot ft}{5}$ P = 130,710t/550 = 238t (hp)FOR t < 40s P = (3000)(73.3) = 400 hp 13.54



GIVEN:

ME= 3000 kg, ELEVATOR HASS MW=1000 kg, COUNTER WEIGHT

FIND:

(a) P(kw) DELIVERED BY
HOTOR WHEN VELOCITY
OF E, VE=3m/S DOWN
AND CONSTANT (QE=0)

(b) P(LW) WHEN $U_E = 3m/s UPWARD$ $Q_E = 0.5 m/s^2 DOWN$

(a) Acceleration = 0

+ TEF=TW-MWg=0 + TEF=ZTC+TW-MEG=0 TW=(1000 kg)(9.81 m/s²) 2TC=(9810N)+(3000 kg)(981m) TW= 9810 N

TC= 9810 N

KINEHATICS

2x=xc 2x=xc Vc=2V==6W/s

P=Tc-Vc=(9810N)(6M/5)=58,860 I/S P(LW)=58.9

(b) ac=0.5 m/s²t Uc=3 m/s \$

COUNTERWEIGHT ELEVATOR

TO THE

W = W | MWQW E = E | MEQE

OUNTER WEIGHT ZF=MQ

EF=TW-NWQ = NWQW

Tw=(1000 kg) ((9.81 m/s2)+(0.5 m/s2)]

TW= 10310 N

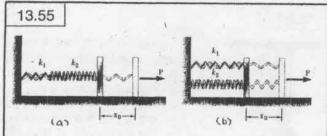
ELEVATOR ZF=Ma +1ZF= ZTC+TW-MFg=-MEQE

2Tc = (3000 leg) [(4.81 m/s2-(0.5 m/s2)]-10310 N

TC= 8810 N N=6m/s (SEE (A))

P= Tc. Uc = (8810N) (6 m/6) P= 52,860 T = 52.86 km

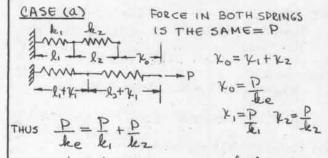
P(kw)= 52.9



P CAUSES DEFLECTION %, IS SLOWLY APPLIED (a) SPRINGS & AND &Z IN SERIES (b) SPRINGS & AND LEZ IN PARALLEL

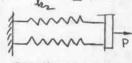
SINGLE EQUIVALENT SPRING Re WHICH CAUSES THE SAME DEFLECTION

SYSTEM IS IN EQUILIBRIUM IN DEFLECTED TO POSITION.



CASE (b)

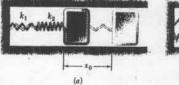
DEFLECTION IN BOTH SPEINGS IS THE SAME = 76

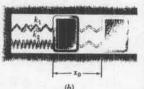


P=k, YotkzXo P=(le1+le2) Xo P= ke Ko

EQUATING THE TWO EXPRESSIONS FOR P = (k,+hz) ko= ke xo Re=ki+kz

13.56





GIVEN:

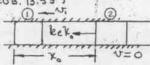
BLOCK OF HASS M BLOCK HOURD TO XO AND RELEASED FROM REST.

FIND:

MAXIMUM VELOCITY, UMAX

continued 13.56

WE WILL USE AN EQUIVALENT SPRING CONSTANT Re (SEE PROB. 13.55)



CHOOSE () AT INITIAL UNDEFLECTED POSITION CHOOSE @ AT X. WHERE U=0

V2=1 ke xo T2=0

Ti+V= T2+V2 0+ 2m v= = 1 ke K2+0 THUS V = UMAX = Yo V ke

(ASE (a) ke= &, k2

(ASE (b) ke=k+k2 VMAX=YoV k+k2

13.57



R= 12RN/m les = 8kN/m m= 16 kg

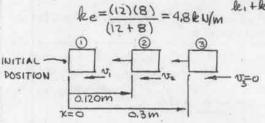
INITIAL POSITION, 300mm TO RIGHT, V= 0

FIND:

(a) HAXINUM VELOCITY, WHAX

(b) VELOCITY 120 mm FROM INITIAL POSITION

USE EQUIVALENT SPRING CONSTANT (SEE P 13.55) FORM SPRINGS IN SERIES , Re= kilz



(a) ATO, SPRING DEFLECTION, O V=0, T== 1 m 0, == 80,2 THUS UI=UHAX

AT (3), $U_3 = 0$ $V_3 = 0$ $V_3 = \frac{1}{2} k_e v_3^2 = \left(\frac{4800}{2}\right) \left(0.3\right)^2 = 216 \text{ N·m}$ T,+V,=T3+V2

80max +0=0+216 U3 = 27

WHAX= 5.20 m/s (b) T2= = MU2=802 V2= 1 Re V2 = (4800)(0.120)2= 34.56 N·M

T2+V2=T3+V3 BU2+34.56=0+216 V2= 22.68 V2= 4.76 M/5 €



W=616 e,= 516/in. 1= 10 lb/in. 123=20 lb/in INITIAL DISPLACEMENT YOF 1.8 IN. TO LEFT FROM UNSTRETCHED POSITION Un= 0

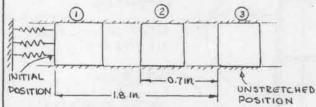
FIND:

(a) MAXIMUM UGLOCITY, UMAX

(b) VELOCITY AT O.7 IN. FROM INITIAL POSTION

EQUIVALENT Re= R, + Rz+ Rz (SEE P13.55 (b))

ke= 5+10+20=351b/in=420 1b/ft



(a) MAXIMUM VELOCITY OCCURS AT (3) WHERE THE SPRINGS ARE UNSTRETCHED T3= 1 MUHAX = 3 UHAX Ti=0 Vi=1 le xo=(47016/ft) (1.811.)

V1= 4.725 1b.ft

 $0+4.725 = \frac{3}{9} V_{HAX}^2 + 0$ $U_{\text{HAX}}^2 = (\frac{32.2 \text{ ft/s}^2}{3 \text{ lb}})(4.725 \text{ lb.ft}) = 50.715$

UHAX=7.12 ft/s --

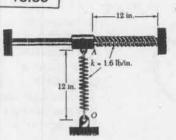
(b) $T_2 = \frac{1}{2} m v_2^2 = \frac{6}{24} v_2^2 = \frac{3}{4} v_2^2$ V2= 1 ke 2= 42016/ft (7m. V2 = 0.7146 16.ft

T,+V, = T2 +V2

$$0+4.725 = \frac{3}{9} U_3^2 + 0.7146$$

 $U_3^2 = (\frac{32.2 \text{ H/s}^2}{(316)})(4.010 \text{ 16 H}) = 43.05$

13.59

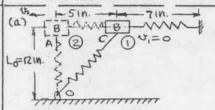


GIVEN:

WB = 10-16 COLLAR B PUSHE TO RIGHT K = SIM. AND RELEASED UNDEFORMED LENGTH OF EACH SPEING, LAKEN R=1.6 16/11. FOR EACH SPRING

FIND:

(a) MAXIMUM VELOCITY, WHAX (b) MAXINUH ACCELERATION, CHAX



MAXIMUM VEWCITY OCCURS AT A WHERE THE COLLAR IS PASSING THROUGH ITS EQUILIBRIUM POSITION

POSITION (R=(1.616/11)/(12111./ft)=19.2 16/ft T1=0 Loc= 152+122= 1311 ALoc= 13in.-12in.=1in.= 12ft. ALAC= SIN = 5 At

V= = = (ALoc) + = le (ALAc) = (19.2 (b) (12+1)+(2+1)

Vi= 1.733 lb.ft $\frac{\text{POSITION (2)}}{\text{T}_2 = \frac{1}{2} \text{ m U}_2^2 = \frac{1}{2} \left(\frac{10}{9}\right) \text{U}_{\text{HAX}}^2 = \frac{5}{9} \text{U}_{\text{MAX}}^2$

V2= 0 (BOTH SPRINGS ARE UNSTRETCHED)

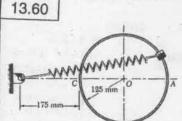
Ti+Vi=T2+V2 0+1.733 = \$ 03 HAX +0 U2 = (1.733 lb.ft)(32.2 f/52)=11.16 ft2
(5 lb)

UNAX = 3,34 ft/s (b) MAXIMUM ACCELERATION OCCURS AT G WHERE THE HORIZONTAL FORCE ON THE COLLAR IS A MUMIXAM

$$\frac{B}{B} = \frac{B}{F} = k \Delta L_{AC} = \frac{B}{m a_{HAX}}$$

ZF= Ma Ficos + F2 = Manax kaloccose+ & Alac = Manax (19.2 16/ft)[(12ft)(5)+(5ft)]=1016 αμαχ 8.615 = 18 a 0=(8.615 1b)(32.2ft/s2) (10 1b)

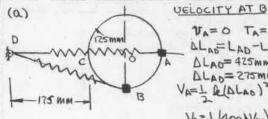
QHAX = 27,7 ft/52



HASS OF COLLAR m= 1.5 kg. R = 400 N/m UNDEFORHED LENGTH OF SPRING, LO= ISOMA COLLAR RELEASED FROH REST AT A

FIND:

(a) VELOCITY OF THE COLLAR AT B, UB (b) VELOCITY OF THE COLLAR AT C. VC



VA= O TA=O ALAS-LAD-LO △LAD= 425mm-150mm DLAD= 275MM=0.275M VA=1 R(DLAO)2

VA= 1 (400 N/m) (0.275m)2

TB= 1 MUB= (15/9)(UB) = (0.75) UB2

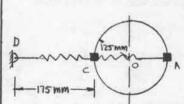
 $L_{BD} = (300^{2} \text{mm} + 125^{2} \text{mm})^{2} = 325 \text{mm}$ $\Delta_{BD} = L_{BD} - L_{0} = (325 \text{mm} - 150 \text{mm}) = 175 \text{mm} = 0.175 \text{m}$

VB=1 &(DBD)=1 (400 N/m)(175m)=6.125 I

TA+VA=TB+VB 0+15.125=0.7502+6.125

 $V_8 = \frac{(15.125 - 6.125)}{(0.75)} = 12.00 \frac{\text{m}^2}{5^2}$

UB= 3.46 M (b) VELOCITY AT C



TA= 0 VA= 15.125 T (SEE (a))

Tc= = mv2= = (1.5kg) v2=0.7502

DLoc= Lo-Loc=(150mm-175m)=-25mm

V== 1 k(ALoc) = 1 (400 U/m) (-0.025 m)2

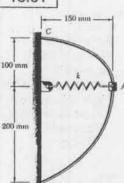
Vc= 0.125 J

TA+VA=Tc+Vc

0+15.125=0.7502+0.125 U2= 15/0.75 = 20

Uc= 4.47 m/s





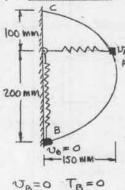
GIVEN:

HORIZONTAL PLANE MASS OF COLLAR m=500-9 UNDEFORMED LENGTH OF SPRING, Lo= 80 mm k= 400 le N/m

FIND:

(a) VELOCITY AT A, VA FOR VELOCITY AT B = O (b) VELOCITY AT C. UC

(a) VELOCITY AT A



TA = 1 M VA = (0.5 kg) VA TA=(0,25) UA

ALA= 0.150 m-0.080 M ALA = 0.070 M

VA= 1 & (ALA)

VA= 1/400×103 N/m)(0.070 m)

VA= 980 J

DLB= 0.200M-0.080M = 0.120M

VB = = 1 & (ALB) = = (400×103 KI/m) (0.120 m)2

NB= 2880 I

TA+VA=TB+VB 0.25UA+980=0+2880

UA= (2880-980) (0.25)

Un = 7600 mysi

UA= 87.2 m/s

(b) VELOCITY AT C



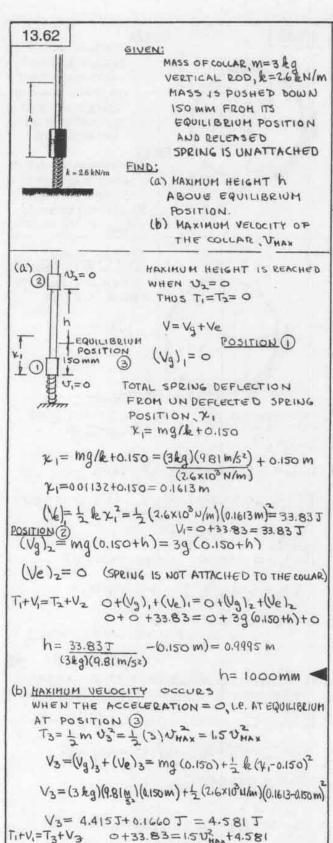
SINCE SLOPE AT B IS POSITIVE THE COMPONENT OF THE SPRING FORCE FP. PARALLEL TO THE ROD, CAUSE'S THE BLOCK TO HOUS BACK TOWARD A TB=0 , VB= 2880 J (FROM PARTIE))

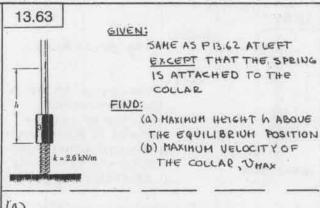
Tc= 1 m v2= (0.549) v2=0.25 v2

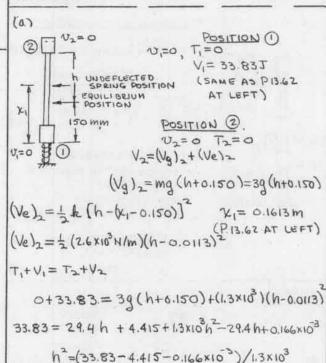
ΔLc= 0.100m-0.080m=0.020m Vc= 1 &(DLc)2= 1 (400×103N/m/0.020m)= 80.0T

TB+VB=Tc+Vc 0+2880=0.2542+80.0 Ve2= 11200 m2/52

UC= 105,8 M/5







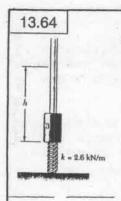
h=(33.83-4.415-0.166×103)/1.3×103 h2= 22.499 h= 0.1500 m h=150 mm

(b) MAXIMUM VELOCITY - SEE (b) AT LEFT

UHAX= AAZMIS

UHAX= (29.249)/1,5=19.50 M/5

UHAX = 4.42 W/5



13.65

GIVEN:

m=3kg le=2.6 kN/m

FIND:

(a) COMPRESSION OF SPRING FROM UNDEFORHED POSITION IF COLLAR COMES TO EQUILIBRIUM

(b) HAXIMUM COMPRESSION IF COLLAR IS SUPPENLY RELEASED

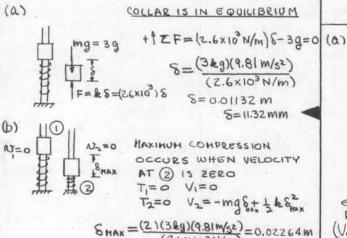
13.66

GIVEN:

VERTICAL PLANE SPRING, &= 316/ft, UNDEFORMED LENGTH A 8,15 UNATTACHED TO COHAR W= 802 0=30° (V=0) NO FRICTION

FINO:

(a) MAXIMUM HEIGHT H ABOUE B REACHED BY THE COLLAR (b) HAXIMUM VELOCITY, THAY OF THE COLLAR



(2.6×103 N/m)

GIVEN:

FIND:

8=22,6 MM

STATIC

TO W.

DEFLECTION. 4st 15 PROPORTIONAL 0=30°= 17/6 RAD R= 12111, = 1 ft

WHEN THE VELOCITY AT E IS ZERO Tc = 0

MAXIMUM HEIGHT ABOUE B IS REACHED

TE= 0 V= Ve + Va

POINTC DLBC=(Ift)(I PAD) DrBC= It t+

(Vc) = 1 le(OLOC) (Vc)e = 1 (316/ft) (If ft) = 0.4112 16.ft

(N°) = MB(1-co20)=(BOS) (14)(1-co2300)

(Ve) q = 0.06699 lb.ft

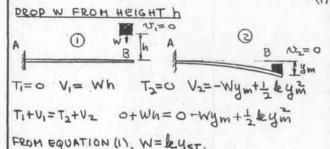
(VE) = 0 (SPRING IS UNATTACHED)

 $(V_E)_g = WH = \frac{8}{16}(H) = \frac{H}{2}(16.ft)$

Tc + Ve = TE + VE

0+0.411Z+0.06699=0+0+H

YM, WHEN W 'S DEOPPED FROM h DENOTE BY & AN EQUIVALENT SPRING CONSTANT STATIC DEFLECTION OF BEAM IS THEN YST (1)



FROM EQUATION (1), W= leyst. Leyst (h+ym) = 1 leyst.

ym -2ystym-2ysth=0 ym=yst(1+12n yst)

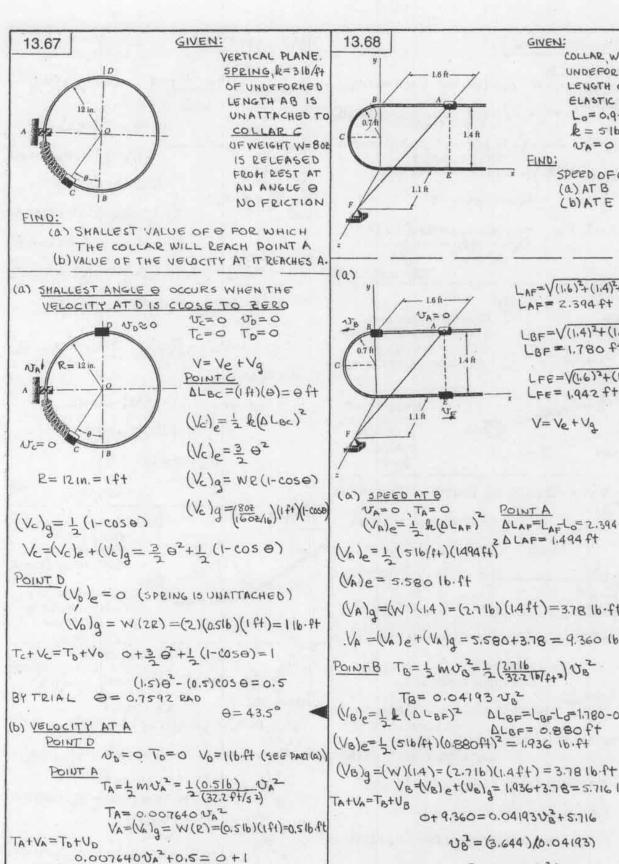
H= 0.956 ft (b) THE MAXIMUM VELOCITY IS AT B WHERE THE POTENTIAL ENERGY IS ZERO, UB=UHAX

Vc= 0.4112+.06699 = 0.4782 1b.ft

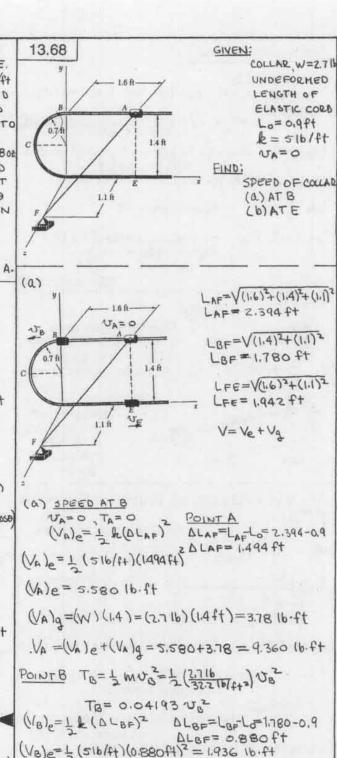
TB= 12 MUB= 12 (116/32.2ft/s2) UHAX TB= 0.07640 V2

VB=0 0 + 0.4782=(0.0764) V HAX Tc+Vc=TB+VB NAX= 61.59 ft2/52

UMAX= 7.85 ft/s



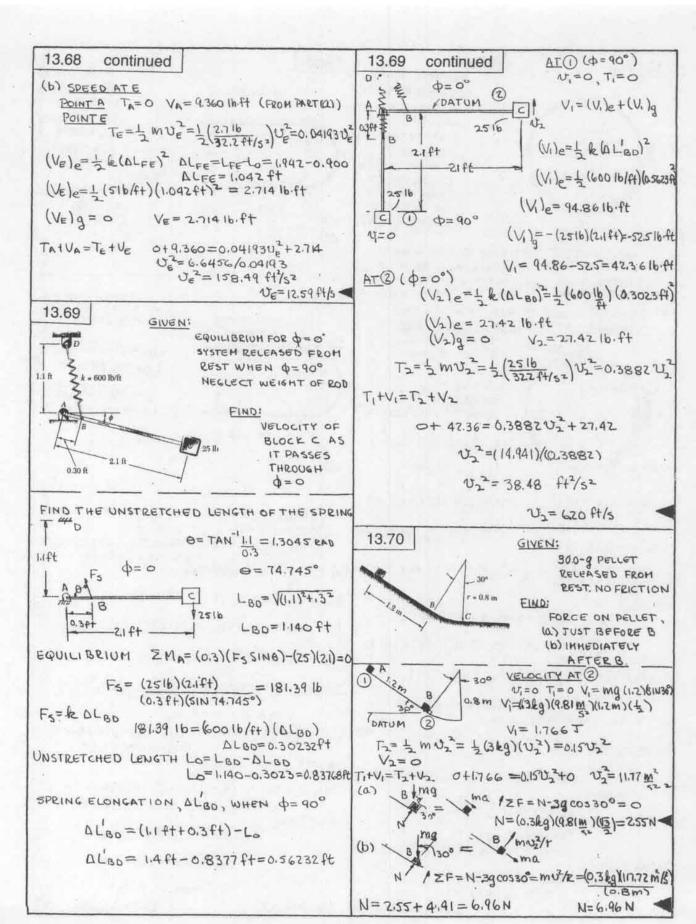
V2= 64.4 ft2/52

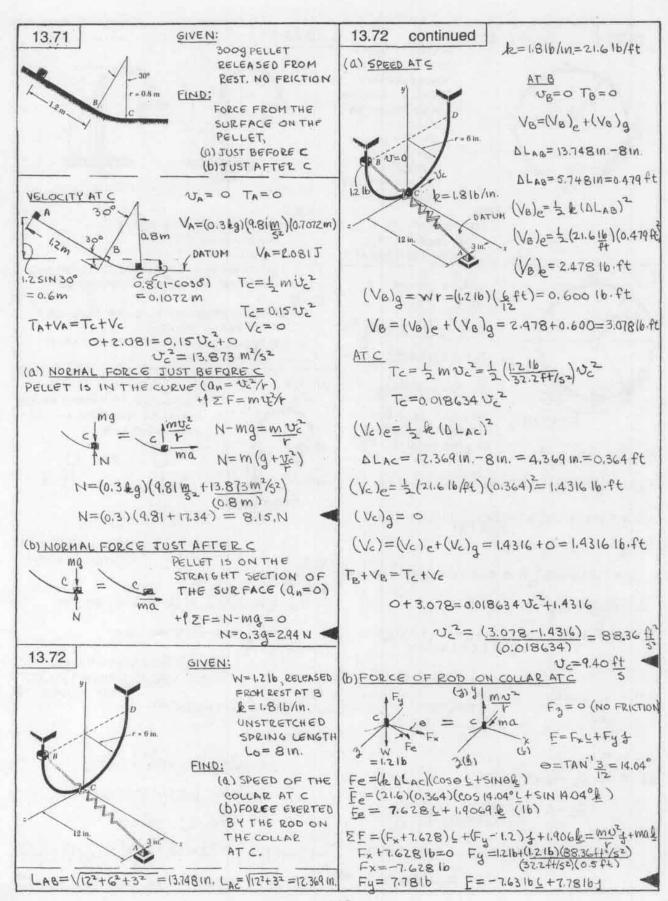


UB= (3.644) (0.04193)

 $V_B = (V_B)_e + (V_B)_q = 1.936 + 3.78 = 5.716 \text{ lb.ft}$

O+ 9.360= 0.04193 V2+5.716





GIVEN:

VERTICAL PLANE

SPRING, &= 3 lb/ft

UNDEFORMED

LENGTH = ARC AB

UNATTACHED TO

COLLAR.

COLLAR WEIGHT

W= 8 02.

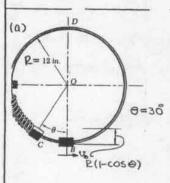
0= 30°.

COLLAR RELEASED

FROM REST AT C.

FIND:

(a) VELOCITY AT B, VC (b) FORCE ON THE COLLAR FROM ROD AT B



 $V_c = 0, T_c = 0$ $T_B = \frac{1}{2} m V_B^2$ $T_c = \frac{1}{2} \frac{(802)}{(1602/1b)(322\frac{11}{52})} V_B^2$ $T_B = 0.07764 V_B^2$

Vc=(Vc)e+(Vc)g ARC BC= AL= RO

ALBC = (1ft)(303(11) ALBC = 0.5236 ft 180

(/c)e=1/2 /e(DLBC)2

 $(V_c)_e = \frac{1}{2} (3 \frac{b}{f+}) (0.5236 ft)^2 = 0.4112 \frac{b}{ft}$

(1/c)g=WR(1-cos 0)=(802) (1602/16)(1ft)(1-cos 30°)

(Vc)g= 0.06699 1b.ft

Vc = (Vc)e+(Vc)g= 0.4112+0.06699=0.47821696

Va = (Va)+(Va)g = 0+0=0

 $T_c + V_c = T_B + V_B$ 0+ 0.4782 = 0.07764 V_8^2 $N_a^2 = 61.59 \text{ ft}^3/\text{s}^2$

(b) W=0.51b mv_B^2/E ma

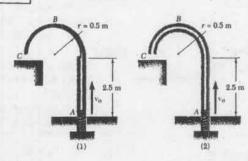
+ = FE-W= mv3/R

 $F_R = 0.5 lb + (0.5 lb) (61.59 ft^2/s^2) (61.59 ft^2/s^2)$

FR = 0.516+ 0.9564 1b= 1.456 1b

FR=1.4561b

13.74



GIVEN:

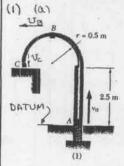
PACKAGE, MASS M= 200-9 INITIAL VELOCITY, UO FRICTION LESS TOBE

(1) TUBE IS OPEN ALONG CIRCULAR ARC

FIND:

(a) SHALLEST VELOCITY V. FOR PACKAGE TO REACH POINT C

(b) FORCE EXERTED BY THE PACKAGE ON THE TUBE.



THE SMALLEST VELOCITY AT B WILL OCCUR WHEN THE FORCE EXECTED BYTHE TUBE ON THE PACKAGE IS ZERO.

++ EF= 0+ mg = m vB

 $V_{B} = g r = (9.81 \text{ m/s}^2)(0.5 \text{ m})$ $V_{B}^2 = 4.90 \text{ 5} \text{ m}^2/\text{s}^2$

TA= 1 MNo VA=0

 $T_B = \frac{1}{2} \text{ m } V_B^2 = \frac{1}{2} \text{ m } (4.905) = 2.453 \text{ m}$

VB = mg (2.5 + 0.5) = 3 mg

 $\frac{1}{2}$ $m\sqrt{6}^2 + 0 = 2.453M + 3Mg$ $\sqrt{6}^2 = 2[(2453) + 3(9.81)] = 63.77$

 $T_c = \frac{1}{2} m U_c^2 V_c = mg (25 m)$ $V_0 = 7.99 m/5$

TA+VA=Tc+Vc

1 mv2+0= 1 mv2+2.5mg

UZ= [63.77- (5.0)(9.81)]

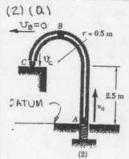
Uc2= 14.72 m2/52

(b)

TEF = Nc = MVc2 = 0.2kg)(14.72 m/s)

TO MORE TO PACKAGE ON TUBE! Nc = 5.89 N=

13.74 continued



THE VELOCITY AT B CAN BE NEARLY EQUAL TO ZERO SINCE THE WEIGHT OF THE PACKAGE IS SUPPORTED BY THE TUBE. THUS, Ug=0 TB=0 VB=mg (2.5m+0.5m) VB=3mg

TA=1 MUO VA= 0

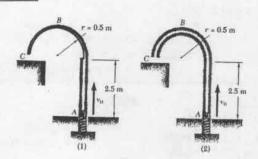
TB+VB=TA+VA

U0=69 (6) Tc= 1 muc Vc= mg (2.5m)

TA+VA=Tc+Vc 1 mvo2+0=1 mvc2+z.smg Vc2= 6g-5g=9.81 m2/s2

muc2 = ZF= Nc= MUc/r Nc=(0.2 kg)(9.8 (m/s2)/0.5m) PACKAGE ONTUBE, N= 3,92N-

13.75



GIVEN:

VEYOCITY AT C (< 3.5 m/s (REQUIRED)

FIND:

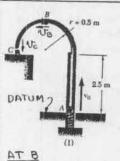
(a) LOOP (2) BUT NOT LOOP (1) CAN SATISFY REQUIREMENT THAT U; C3.5 m/s (D) LARGEST ALLOWABLE VELOCITY UD WHEN LOOP (2) IS USED AND UCC 3.5 m/s.

(a) LOOP (1), THE SMALLEST ALLOWABLE YELOCITY AT B WILL OCCUR WHEN THE FORCE EXERTED BY THE TUBE ON THE PACKAGE IS

+ | EF= 0+mg = mv3/r

UB= gr=(9.81 m/s2)(0.5m)= 4.905 m2/s2 UB= 2.215 W/S

13.75 continued



THE VELOCITY AT B CANNOT B LESS THAN 2.215M/S IF THE PACKAGE IS TO MAINTAIN CONTACT WITH THE

FOR VETO BE AS SHALL AS POSSIBLE, VB MUST BE AS SHALL AS POSSIBLE: THAT IS UB= 2,215 M/S TB= 1 m NB=1 m (2.215)2

TB=2453 M

VB= mg(2.5+0.5)= 3mg

Tc= 1 mv2

Vc= 2.5 mg

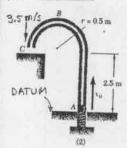
TB+VB=Tc+Vc

2.453 m+3mq = 1 mV2+2.5mg V2= 2 (2.453+0.5(9.81m/s2)]

152=14.72 m3/52

Uc= 3.836 m/s >3.5 m/s THUS, LOOP (1) CANNOT HEET THE REQUIREMENT

(b) LOOP (2)



TA= 1 muo

VA= 0

U= 3.5 m/s TC=1 m (3.5)2 Tc= 6.125 m

Vc= 2.5mg

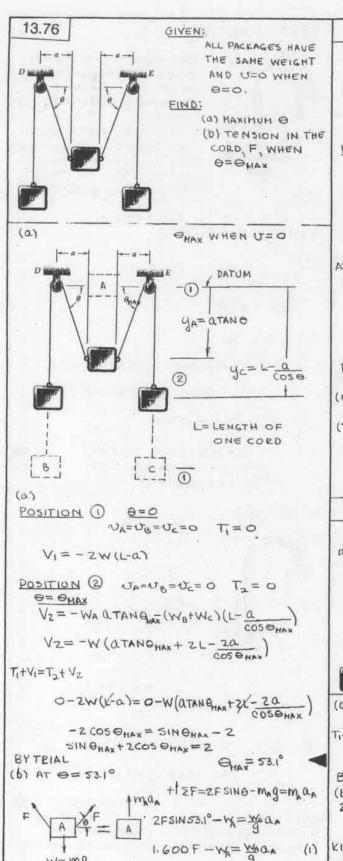
TA+YA=Tc+Vc

+ mvo2 + 0 = 6.125m +2.5mg

Vo2= 2 (6.125+2.59) = 61.3 m2/52

Un= 7.83 m/s

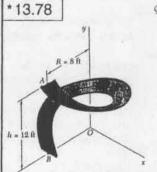
A LARGER VELOCITY AT A WOULD RESULT IN A VELOCITY AT C, GREATER THAN 3.5 W/S



13.76 continued +1 ZF=F-meq=-meac (2) meac mag KINEMATICS an=4A YA=QTANO LET f (0)=-a sec20 ye= atano 6 YA= + (0) 0+ fA(0) 0 LET fo= aTANO 9= fc 0+fc(0)0 THUS YA = fa (53.1°) = - a sec2(53.1°) 4- f. (53.1°) a TAN (53.1°) (0553.1° a= 1,2500c QA = 1.250 REPLACE QAIN (1) BY 1.2500c FROM (3) WA=WB=W=W 1.000 E-M= A (1.52005) F-W=- & ac 1.600 F-W =-1.250 (F-W) 2.850 F= 2.250W F=0.789W 13.77 GIVEN: WA= 216 WB=WC=316 N=0, WHEN 0=0 FIND: O HUMIXAM (D) (b) TENSION F AT OMAX REFER TO FIGURE IN P13.76 (a) AT LEFT (a) 0=0 Ti=0 Vi=-(WB+WC)(L-a)=-6(L-a) 9=0MAX T2=0 V2=-ZQTANOMAX-6 (L-Q COS OMAX) TI+VI=TI+VZ 0-6 (K-a)=0-ZOTANQ - 6(K- a COS QNA -6005 0 HAX = 251N 0 HAX -6 BYTRIAL (b) REFER TO (b) PROB 13.76 2F SIN 36,90-WA= WA QA 1.201F-Z=3 ax F-3=-3 ac (1) F-Wc=-Wcac (2) KINEMATICS QA = 40 = 50236.9 = 1.665

SOLVE (17, (2) AND (3) FOR F (3)

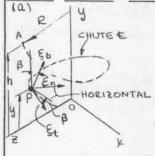
F= 2.31 16



PACKAGES RELEASED FROM REST AT A CHUTE IS BANKED SO THAT PACKAGES DO NOT TOUCH ITS EDGES. NO FRICTION PACKAGE WEIGHT, W=2010. CHUTE IS A HELIX WITH PRINCIPAL HORNAL HORIZONTAL AND DIRECTED TOWARD Y AXIS.

FIND:

(a) ANGLE & FORMED BY THE NORMAL TO THE SURFACE OF THE CHUTE AN THE PRINCIPAL NORHAL (b) MAGNITUDE AN DIRECTION OF THE CHUTE ON THE PACKAGE AT B



AT POINT A UA= O TA= O

VA= man

AT ANY POINT P Tp=1 m U2 Vp=Wy=mgy

En, ALONG PRINCIPAL AND DIRECTED

TA+VA=Tp+Vp NORMAL, HORIZONTAL O+mgh=1mv2+mgy v= 29(h-4)

TOWARD 4 axis EE , TANGENT TO CENTERLINE OF THE CHUTE

ED, ALONG BINORMAL

B=TAN-1 h =TAN-1 (12 ft) B= 13.427°

mab=0 SINCE QUE O

NOTE : FRICTION IS ZERO

ZFt= mat masin B= mat at= gsin B

ZFb= mah Nb-WCOSB=0 NB=WCOSB

ZFn=man Nn=mo=mzg(n-y)=zw(n-y)

THE TOTAL NORMAL FORCE IS THE RESULTANT OF No AND NM, LIES IN THE D-M PLANE AND FORMS ANGLE & WITH m AXIS.

* 13.78 continued

B PED PO tand= Nb/Nn tand = wcos B/2(wh-y)

tand= (e/2(h-y)cosB

GIVEN: (= R[1+(1/2)]=R(1+tanp)=R

THUS $tan \phi = \frac{e}{2(h-y)} cos \beta = \frac{R}{2(h-y) cos \beta}$

tan = 8 ft = 4.113

OR COT \$ = 0,243(12-4) ◀

(b) At POINT B y= O FOR X, Y, Z AXES WE WRITE, WITH W= 2016 Nx= Nb 51NB= WCO5B5INB=(2016)COS 14.327°SINA.30 Nx= 4.517 lb

My = No cosp = Wcos2 = (2016) cos2 14.327° Ny=18,922 16

N=-NN=-SM W-A=-S M W-A

NZ= 2 (2016) (12ft-0) COS 14.327 NZ= -56.76516

N=V(4.517)2+(18.922)2+(-56.765)2 $\cos \Theta_{x} = \frac{Nx}{N} = \frac{4.517}{60}$

N=60.016 0x=85.7°

cos ey = Ny = 18.922

Oy=71.6

 $\cos \Theta_2 = \frac{N_2}{N} = -\frac{56.742}{60}$

02=161.1°

* 13.79 GIVEN:

> F(x,4,2) IS CONSERVATIVE SHOW THAT:

> > $\frac{\partial F_y}{\partial x} =$ $\frac{\partial F_x}{\partial F_y} = \frac{\partial F_y}{\partial F_y}$

FOR A CONSERVATIVE FORCE, EQ (13.22) HUST

BE SATISFIED $F_x = -\frac{\partial V}{\partial x}$ $F_y = -\frac{\partial V}{\partial y}$ $F_z = -\frac{\partial V}{\partial z}$

WE NOW WRITE TE = - DN WON 3W

FINCE DXDy = DYDX:

シテメーシティ シェーシズ

WE OBTAIN IN A SIMILAR WAY

 $\frac{\partial F_{4}}{\partial z} = \frac{\partial F_{2}}{\partial y}$

*13.80

GIVEN:

F=(y==+2x++xy+)/xy=

SHOW:

(Q) F IS a CONSERVATIVE FORCE

FIND:

(b) THE POTENTIAL FUNCTION ASSOCIATED WITH F

(a) Fx = y 2/kyz Fy=ZX/ZyZ $\frac{\partial F_y}{\partial x} = \frac{\partial (/y)}{\partial x} = 0$ $\frac{\partial F_{x}}{\partial y} = \frac{\partial (1/x)}{\partial y} = 0$ THUS $\frac{\partial F_X}{\partial y} = \frac{\partial F_Y}{\partial x}$

THE OTHER TWO EQUATIONS DERIVED IN PROB. 13.80 ARE CHECKED IN A SIMILAR

(b) RECALL THAT Fx=-DV, Fy=-DV, F=-DV V = -ln x + f(y, z)(1)

$$Fy = \frac{1}{y} = -\frac{2y}{2y} \quad V = -\ln y + g(z_1x) \quad (2)$$

$$F_2 = \frac{1}{2} = -\frac{2V}{2}$$
 $V = -\ln 2 + h(x,y)$ (3)

EQUATING () AND (2)

-lnx+f(y,z) = -lny+g(z,x)
Thus
$$f(y,z) = -lny + k(z)$$
 (4)
 $g(z,x) = -lnx + k(z)$ (5)

(5)

EQUATING (2) AND (3)

$$-\ln z + h(x,y) = -\ln y + g(z,x)$$

FROM (5) g(z,x) = - lnz+ l(x) g(z,x)=-lnx+le(2)

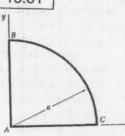
THUS

FROM (4)

SUBSTITUTE FOR f(4,2) IN (1)

V=-lnxyz+C

* 13.81



GIVEN

PARTICLE P(X,U) ACTED UPON BY FORCE F

FIND:

WHETHER F IS A CONSERVATIVE FORCE, AND COMPUTE THE WORK OF F WHEN P(X,Y) DESCRIBES A PATH ABCA CLOCKWISE

(a) F=RyL (b) F=k(yc+x+)

Fy=0 OFx = R OFy = 0 E IS NOT CONSERVATIVE [ky L. dy + hy C. (dx tdy) UABCA = | F.dr = ky L-dx4

S = O F IS PERPENDICULAR TO THE PATH

[kyc.(dxc+dyf)=]kydx

FROM BTOC THE PATH IS A QUARTER CIRCLE WITH ORIGIN AT A.

THUS X2+42=02

4 = VQ2-X2 Jaydx= 1 & Vai-X2 dx ALONG BC

= Thaz

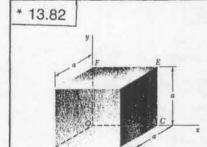
Sky L. dx = 0 (y=0 ON CA)

 $\int + \int + \int = 0 + \frac{\pi k a^2}{4} + 0$ UABCA Tha?

Fy=kx OFx=k, OFy=k OF = OFY, F IS CONSERVATIVE

SINCE ABOA IS A CLOSED LOOP AND F IS CONSERVATIVE,

UABCA = 0



POTENTIAL
FUNCTION
V(x,4,2)=-(x²+4²+2²)/2
ASSOCIATED
WITH FORCE P.

FIND:

(a) ky, 2 Components OFP

WORK DONE

BY P FROM OTO D BY
INTEGRATING ALONG
THE PATH OABD, UGABD,
SHOW THAT UGABO = AV

(a)
$$P_x = -\frac{\partial V}{\partial y} = -\frac{\partial (x^2 + y^2 + z^2)^{1/2}}{\partial x} = \frac{\partial (x^2 + y^2 + z^2)^{1/2}}{\partial y} = \frac{\partial (x^2 + y^2 + z^2)^{1/2}}{\partial y}$$
(b) $U_{ABB} = U_{ABB} + U_{ABB} + U_{BBB}$

O-A Py AND Px ARE PERPENDICULAR TO O-A
AND DO NO WORK
LISO, ON O-A X=y=O AND Pz=1
THUS UO-A= Pzdz= dz= a

A-B PZ AND PY ARE DERDENDICULAR TO A-B
AND DO NO WORK
ALSO ON A-B y=0, Z=Q AND
PX= X/(x²+q²)1/2

THUS
$$V_{A-6} = \int_{0}^{a} \frac{x \, dx}{(x^2 + a^2)^{1/2}} = a(\sqrt{2}-1)$$

B-D PX AND PZ ARE PERPENDICULAR TO

S-D AND DO NO WORK

ON B-D X=a, Z=a Pg=y/(y²+za²)/2

THUS $U_{80} = \int_{0}^{2} \frac{y}{(y^2+za^2)} v_z dy = (y^2+za^2)^{1/2} |^{a}$ $U_{BD} = (a^2+za^2)^{1/2} - (za^2)^{1/2} a (13-12)$

UONBC=JO-A+ WAR+ UBD= a+a(12-1)+a(13-12)

UDABO = a 13

 $\Delta \vee_{oo} = \vee (a, a, a) - \vee (o, o, o) = -(\alpha^2 + \alpha^2 + \alpha^2)^{1/2} \circ$

AV00 = - 953

THUS UDABO = - AVOD

13.83

REFER TO FIG. PIB. BZ ON THE LEFT

GIVEN:

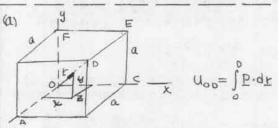
FROM SOLUTION TO (a) OF PROB. 13.82 $P = \frac{x_1 + y_2 + z_2}{(x^2 + y^2 + z^2)} V_2$

FIND:

(4) WORK DONEBY P ALONG THE DIAGONAL

VERIFY:

(b) THAT WORK DONE AROUND THE CLOSED PATH OABDO IS ZERO.



r= x1+y3+2k dr=dx1+dy3+d2k P= x1+y3+2k (x2+y2+22)/2

ALONG THE DIAGONAL X=4=2

THUS
$$P: dr = \frac{3x}{(3x^2)^{1/2}} = \sqrt{3}$$

 $U_{0:0} = \int \sqrt{3} dx = \sqrt{3} a$

U00= 13a

(6)

UMB 00 = UDABO+ 400

FROM PEOB 13.82

UDABO = V3 Q AT LEFT

THE WORK DONE FROM DTO & ALONG THE DIAGONAL IS THE NEGATIVE OF THE WORK DONE FROM OTOD

THUS

40ABDO = 13a - 13a = 0

* 13.84

GIVEN: F= (x+4++2k)/(x2+42+22)3/2

PROVE:

(a) F IS CONSERVATIVE

FIND:

(DITHE POTENTIAL FUNCTION V(X,Y,Z) ASSOCIATED WITH F

(a)
$$F_x = \frac{y}{(x^2+y^2+z^2)^{3/2}}$$
 $F_y = \frac{y}{(x^2+y^2+z^2)^{1/2}}$

$$\frac{\partial F_{x}}{\partial y} = \frac{(x^{2}+y^{2}+z^{2})}{(x^{2}+y^{2}+z^{2})} = \frac{\partial F_{y}}{\partial x} = \frac{(x^{2}+y^{2}+z^{2})}{(x^{2}+y^{2}+z^{2})} = \frac{\partial F_{y}}{\partial x}$$

THUS
$$\frac{\partial F_{x}}{\partial y} = \frac{\partial F_{y}}{\partial x}$$

THE OTHER TWO EQUATIONS DERIVED IN PROB. 13.79 ARE CHECKED IN A SIMILAR FASHION

(b) Recalling that
$$F_x = -\frac{\partial V}{\partial x}$$
, $F_y = -\frac{\partial V}{\partial y}$, $F_z = -\frac{\partial V}{\partial z}$

V= (x2+y2+22) = + f(y,2)
SIMILARLY INTEGRATING DY AND DY SHOWS THAT

THE UNKNOWN FUNCTION F(X,Y) IS A CONSTANT

13.85 GIVEN:

3600-RY LAUNCHED FROM A CIRCULAR ORBIT AT BOOK M ABOVE THE EARTH. ALTITUDE OF GEOSYNCHRONOUS (CIRCULAR) ORBIT = 35770 km

FIND:

- (a) ENERGY NEEDED TO PLACE THE SATELLITE INTO GEOSYNCHRONOUS ORBIT - FROM 300 Rm
- (b) ENERGY NEEDED TO PLACE THE SATELLITE INTO A GEOSYNCHRONOUS DEBIT FROM THE EARTH (EXCLUDE AIR RESISTANCE

GEOSYNCHRONOUS OPBIT

r2 = 6370 km + 35770 km = 42.140×10 m

ORBIT AT 300 km r= 6370km+300km= 6.67x106m

Pc=6370 Rm

FOR ANY CIRCULAR ORBIT OF RADIUS & THE TOTAL ENERGY E = T+V = 1 MU2- GMM

M=MASS OF THE EARTH M= 3600 leg = SATELLITE HASS

13.85 continued

NEWTON'S SECOND LAW F=man: GMm = mu2 T= 1 m v2 mGM E=T+V= 1 GMM - GMM =- 1 GMM GN=gRe E=-1 9Rem E=- 1 (9.81 m/52) (6370 × 10 m) (3600 kg) E= - 716.15 x10 (N.m)

FOR A GEOSYNCHRONOUS ORBIT (1=42.140 x10 m)

$$E_{GS} = \frac{-716 \times 10^{15}}{42.140 \times 10^6} = +7.003 \times 10^9 = -17.003 GJ$$

(a) AT 300 km (r= 6.67 x10 m)

$$E_{300} = -\frac{716 \times 10^{15}}{6.67 \times 10^{6}} = +0.7.42 \times 10^{9} J = -107.42 GJ$$

ADDITIONAL ENERGY AE300 = EGS - E300

ΔE300=90.46J◀ (b) LAUNCH FROM THE EARTH (R=6370 Rm)

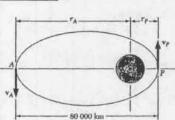
AT LAUNCH PAD E = V = - GMM = - 9 REM E== - (9.81m/s2) (6370×103m) (3600 deg)

EE= - 224.96×109 J = - 224.96 GJ

ADDITIONAL ENERGY AG== EGS- ES

ΔEE= -17.003+224.96=20865€

13.86



GIVEN:

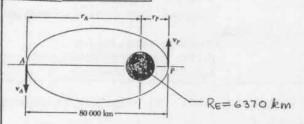
UA/Up=1p/rA ra+rp=80000 lem ELLIPTIC ORBIT

FIND:

ENERGY PER UNIT MASS E/M REQUIRED TO PLACE THE SATELLITE IN ORBIT.

DETERHINE THE TOTAL ENERGY PER UNIT HASS FOR THE ELLIPTIC ORBIT AND SUBTRACT FROM IT THE ENERGY PER UNIT MASS ON THE EARTH TO GET THE ENERGY PER UNIT HASS NEEDED FOR PROPULSION. (EXCLUDING, AIR RESISTANCE, THE WEIGHT OF THE BOOSTER ROCKET AND MANEUVERING!

13.86 continued



TOTAL ENERGY PER UNIT MASS FOR THEORBIT

$$E_0 = T_A + V_A = T_P + V_P$$

$$E_V M = \frac{U_A^2}{2} - \frac{GM}{r_A} = \frac{U_P^2}{2} - \frac{GM}{r_P}$$

$$V_A^2 \left(1 - \frac{V_P^2}{V_A^2}\right) = 2GM \left(\frac{1}{r_A} - \frac{1}{r_P}\right)$$

$$V_A/U_P = \frac{r_P}{r_A} \left(\frac{GIVEN}{r_A}\right)$$

$$2T_A^2 \left(1 - \frac{r_A^2}{r_A^2}\right) = 2GM \left(\frac{1}{r_A} - \frac{r_A}{r_A}\right)$$

$$V_A^2 \left(1 - \frac{r_A^2}{V_P^2} \right) = \mathcal{I}GM \left(\frac{r_P - r_A}{r_A r_P} \right)$$

$$V_A^2 \left(\frac{r_P - r_A}{V_P^2} \right) \left(\frac{r_P - r_A}{V_A V_P} \right) = 2GM \left(\frac{r_P - r_A}{V_A V_P} \right)$$

$$U_A^2 = ZGM \frac{r_p}{r_A} \left(\frac{1}{t_0 + r_A} \right)$$
 (2)

SUBSTITUTING UAIN(2) IN (1)

$$E_{M} = G_{M} \frac{r_{P}}{r_{A}} \left(\frac{1}{r_{P} + r_{A}} \right) - \frac{G_{M}}{r_{A}}$$

$$E_{l}/M = GM \frac{1}{r_{A}} \left[\frac{r_{p} - (r_{p} + r_{A})}{r_{p} + r_{A}} \right] = -\frac{GM}{r_{p} + r_{A}}$$

GM=9Pe2= (9.81 m/52)(6370 x 103 m)

Vp+rA= 80.000 x 103 m (GIVEN)

$$E_0/m = \frac{-(9.81 \text{ m/s}^2)(6370 \times 10^3 \text{ m})^2}{80000 \times 10^3 \text{ m}}$$

Edm = 4.9765 × 106 N-m =-4.9765 MJ

TOTAL ENERGY PER UNIT HASS ON THE EARTH

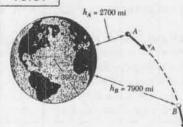
$$E_e = T_e + V_E$$
 $U_e = 0$ $T_e = 0$ $V_e = -\frac{mGM}{R_E}$
 $E_e/m = -\frac{9R_e^2}{R_E} = -\frac{(9.81 \, \text{m/s}^2)(63.70 \times 10^3 \, \text{m})}{R_E}$
 $E_E/m = -62.490 \times 10^6 \, \frac{N-m}{R_E} = -62.49 \, \text{HT/kg}$
ENERGY PER UNIT HASS NEEDED FOR

ENERGY PER UNIT HASS NEEDED FOR PROPULSION, Ep/m = Eo/m - Ee/m

Ep/m = -4.9765 HJ/kg +62.490 HJ/kg

Ep/m = 57.5 MS

13.87



ha AND ha UA=20,2×103 ML

FIND:

ra= ha+R=2700 m1+3960 mL rA= 6660 mL - 18= h8+ R= 7900 MC+3960 M B 1860 mc

AT A, UA = 20.2 × 103 mc = 29627 ft/s TA= 1 m (29627 ft) = 438.87 X10 m VA = - GMM = - 9 Rm

R= 3960 mc = 20,909 x 106 ft VA= - (32.2 ft/52)(20.909 × 106ft)2 m=-400.3 × 106 m

TB= I mUB $V_B = -\frac{GMm}{V_B} = -\frac{gR^2m}{V_B}$

> rB = 11860 mi = 62.621×106ft $V_{B} = -\frac{(32.2 + t/5^{2})(20.909 \times 10^{6})^{2} \text{m}}{(62.621 \times 10^{6} + t)}$

VB=-224.8 X106 M

TA+ VA= TB+VB

438.87×106m-400.3×106m= 1-mUB-224.8×106m Un= 2[43887X106-400.3X106+224.8X106] UB= 526.75 x 106 ft2/52

UB= 22.951×103 ft/5=15.65×103 mi/h

VB= 15.65x103mi

13.88 GIVEN: FIND: V=-GMMM

LUNAR EXCURSION MODULE (LEM)

ENERGY PER POUND NEEDED TO ESCAPE MOON'S GRAVITATIONAL FIELD STARTING FROM (a) MOON'S SURFACE

(b) CIECULAR ORBIT SOME. ABOVE THE MOON'S SURFACE

NOTE: GMHOON = 0.0123 GMEARTH

BY EQ. 12.30 GM MOON = 0.01239 RE

AT @ DISTANCE FROM MOON; 12=0, ASSUME U2=0 E2=T2+V2=0-GMMM =0-0=0

(a) ON SURFACE OF MOON PM=1081mi=5.7077x10ft =0 Re=3960mi=20,909×10ft E=T,+V,=0-0.0123 g Rem U,=0 T,=0

E,=-(0.0123)(32.2ft/s2)(20,909x10ft)m (5.7077×106ft)

WE - WEIGHT OF LEH ON THE EARTH E1=(-30.336 x10 H)m m= W6 E= (-30,336 x 106. ft 3/57) WE

Δ E = E2-E1 = 0+(942.1×10° f+16) WE

ENERGY PER POUND:

ΔE = 942×10 91/16

(b) DOE 6

r= RM+50mi r= (1081 mi+50mi) = 1131mi=5,9717x10 ft

NEWTON'S SECOND LAW: V'= aww L'= Two!= Two Ww

2 GMmm = -1 0.0123 9 RE2 m

E1=- 1 (0.0123) (322 ft/s2) (20.909 x106ft) m

E1 = (14.498×10 ++2/52) WE = 450.2×10 +16 WE

ΔE= E2-E1= 0+450.2×10 ft.16 WE

ENERGY PER POUND

AE = 450×103ft.16

13.89

GIVEN: SATELLITE OF HASS M

CIRCULAR ORBIT OF RADIUS Y ABOUT EACH

FIND:

(a) ITS POTENTIAL ENERGY

(b) IT'S KINETIC ENERGY

(C) ITS TOTAL ENERGY

(a) POTENTIAL ENERGY V= - GMM = - q PM + CONSTANT

CHOOSING THE CONSTANT

(cf Eq 13.17)

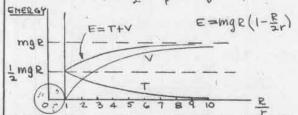
SO THAT V= O FOF= R:

V= mgR(1- R)

(b) KINETIC ENERGY

NEWTON'S SECOND LAW F=man: GMM=mv

(C) TOTAL ENERGY E=T+V = 1 m &R2+mg (1- R)



13.90

GIVEN:

SATELLITE IN A CIRCULAR ORBIT

FIND:

ENERGY REQUIRED TO PLACE IT INTO ORBIT-AT (a) 600 km, (b) 6000 km

BEFORE LAUNCHING: K= R= 6.37x10 M; U,= 0 E = T, +V, = 0 - GHW = - 9R2W = - mgR

IN CIRCULAR ORBIT OF PADIUS 12: [Cf. EQ 12.30]



NEWTON'S SECOND LAW F= Man GHM = M 02 52 U2= GM = 982

E2=T2+V2= 1 m V2 - GHM

E2 = 1 m 9 P2 - 9 P2 = -1 9 P2 M

DE=EZ-E1= - 1 2 RZW - (- mg P) = Rmg(1- RZW) enerch der ke 12

DE/M = Rg (1-R)

(a) 12=6370+600=6970 km ΔE/M=(6.37×106)(9.81)(1-(6370))=33.9 MI

(b) r2 = 6370+6000 = 12370 km $\Delta E/m = (6.37 \times 10^6)(9.81)(1-\frac{6370}{2(12370)}) = 46.4 \frac{N5}{20}$

GIVEN:

EQ (13.17'), Vg= - WRZ DISTANCE ABOVE EARTH'S SURFACE, Y

SHOW!

(a) Vg = Wy (FIRST ORDER APPROXIMATION DERIVE

(b) A SECOND ORDER APPROXHATION

$$A^{2} = -\frac{MS_{5}}{L}$$
 setting $L = S + A : \Lambda^{2} = -\frac{S}{MS_{5}} - \frac{1}{MS_{5}} - \frac{A}{MS_{5}} - \frac{A}{MS_{5}$

WE ADD THE CONSTANT WR, WHICH IS EQUIVALENT TO CHANGING THE DATUM FROM F=00 TO F= R:

(a) FIRST ORDER ARPROXIMATION

(b) SEZOND ORDER APPROXIMATION:

$$\sqrt{d} = MK \left[\frac{K}{R} - \left(\frac{K}{R} \right)_S \right]$$

13.92

GIVEN

CELESTIAL BODY IN CIRCULAR ORBIT. FADIUS F= 60 LIGHT YEARS VELOCITY U=1.2x106mi/h ABOUT A POINT OF HASS, MB

FIND:

BATIO MB/MS, WHERE MS IS THE HASS OF THE SUN

1 0= 1.2 × 10 mi/h= 1.76 × 10 ft/s

Y= 60 LIGHT YEARS

I LIGHT YEAR IS THE DISTANCE TRAVELED BY LIGHT IN ONE YEAR SPEED OF LIGHT = 186,300 mi/s

+= (60 TR)(186,300 Mi) 15280 ft) 365 DAYS) 24/1) 36005 TO r= 1.8612×1018 ft

F = G MB m = M V2

4,645×1021

MB

G MENERY = 9 REMARK = (322 Et) (3960 M EX5280 Et) =4.017 X10

Moun=330,000 ME: CHOUN=330,000 GHEARTH GHSUN=(330,000 X14.077×10'5') = 4.645×1021 ft3/52

G= 4.645×10 /Msun Mg = ru2 = ru2 Mson/4.645X1031 MB/M500 = (1.8612×108)(1.76×106)= 1.241×10

13.93

GIVEN:



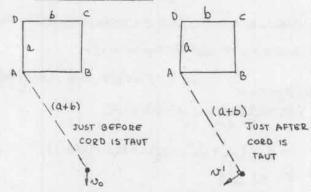
FRICTIONLESS PLATE FIRMLY ATTACHED TO A HORIZONTAL PLANE CORD AGC ATTACHED TO THE PLATE AT A AND TO A SPHERE ATC Un= INITIAL VELOCITY OF SPHERE CAUSES IT TO HAKE A COMPLETE CIRCUIT AND RETURN

FIND:

VELOCITY OF THE SPHERE AS IT STRIKES CIF (a) to 15 PARALLEL TO BC

(b) UD IS PERPENDICULAR TO BC.

(A) No PARALLEL TO BC



ANGULAR MOHENTUM IS CONSERVED ABOUT A

$$b\sigma_o = (a+b)\sigma'$$
 $\sigma' = \frac{b\sigma_o}{(a+b)}$

AS THE SPHERE CONTINUES ITS CIRCUIT TO POINT C ITS VELOCITY IS ALWAYS PERPENDICULAR TO THE CORD AND ENERGY IS CONSERVED THUS V= V'

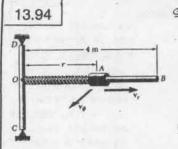
Uc= (a+b)

(b) No PERPENDICULAR TO BC

AS THE SPHERE HAKES A COMPLETE CIRCUIT AROUND THE PLATE ITS YELOCITY IS ALWAYS DERPENDICULAR TO THE COLD AND ENERGY IS CONSERVED

THUS VE=VO

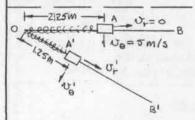
Vc=Vo



k= 750 N/M
UNDEFORMED SPRING
LENGTH, To = 1.5 M
COLLAR HASS, M=2.4 lg
INITIALLY,
r=225 M, Up=5 m/s
Vr=0

FIND:

U' AND U' WHEN



CONSERVATION OF ANGULAR MOMENTUM (ABOUT O)

(2.25m)(m)(5m/s)=(1.25m)(M)(N)

NO FRICTION

CONSERVATION OF ENERGY T+V=T'+V'

T= 30.0 J

V= 210,9 J

V= 9.00 m/s . V;

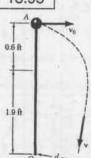
T'= 1.2 U+ 47.2

$$V' = \frac{1}{2} k (Y' - r_0)^2 = \frac{1}{2} (750 \text{ W/m}) (1.25\text{m} - 1.5\text{m})^2$$

V'= 23.44 T

T+V= T'+V'





GIVEN:

ELASTIC CORD FIXED AT O &= 10 lb/ft UNDEFORMED LENGTH, Lo=194 WEIGHT OF BALL, W=1.51b

HORIZONTAL FRICTIONLESS
PLANG
INITIAL VELOCITY U.
PERPENDICULAR T OA

FIND:

(a) SMALLEST ALLOWABLE

Vo IF CORD DOES NOT

BECOME SLACK

(b) CLOSEST DISTANCE & FOR

Vo EQUAL TO HALF VALUE

FOR US FOUND IN (A)

THE CORD WILL NOT GO
SLACK IF N2 15
PERPENDICULAR TO
THE UNDEFORMED CORD
LENGTH Lo, AT @

CONSERVATION OF ANGULAR HOHENTUM

POINT () U1= 00 T1= 1 W U2=0.75 U2

N= = = (1-10)== = (10 lbite) (2.5 ft-1.9ft)=

V1= 1.80016.ft

POINT @ T2= 1 8 02 = 9002

∆L=0 V=0

 $T_1 + V_1 = T_2 + V_2$ $\frac{0.75}{9} v_0^2 + 1.800 = \frac{0.75}{9} v_2^2 + 0$

FROM CONS. OF ANG. HOH. V= 1.315800

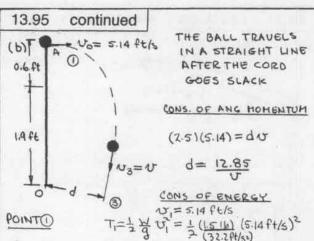
 $0.75 \text{ N}_0^2 ((1.3158)^2 - 1] = 1.800$

 $U_0^2 = (1.8 \text{ lb.ft})(32.2 \text{ ft/s}^2)$

いき 105.67 景

Vo= 10.28 ft

(CONTINUED)



Ti= 0.6154 ft.16

V= 1 k(L-Lo)= 1/2 (1016/ft) (2.5ft-1.9ft)=1.800 16ft

POINT (3)
$$T_3 = \frac{1}{2} \frac{1}{9} v_3^2 = \frac{0.75}{9} v^2$$

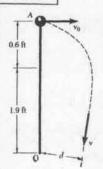
 $V_1 + V_1 = T_3 + V_3 = 0$

0.6154+1.800=0.75v3+0

V= 10.18 ft/s

FROM CONS. OF ANG MOH.

13.96



GIVEN:

ELASTIC CORD FIXED AT O

L= 10 lb/H

UNDEFORMED LENGTH LENGTH

WEIGHT OF BALL, W= LS lb

HORIZONTAL FRICTIONLESS

PLANE

Vo DERDENDICULAR TO OA

d= 0.8ft AFTER CORD

FIND:

(a) INITIAL SPEED NO.

BECOMES SLACK

0.6 ft v₀ 0

CONSERVATION OF ANGULAR MOHENTUM

2.5 00 = 0.8 0 V= 3.125 U5 CONSERVATION OF ENERGY POINT(1)

 $V_i = V_0$ $T_i = \frac{1}{2} \frac{1}{8} v_0^2 = \frac{0.75}{3} v_0^2$ $V_i = \frac{1}{2} \frac{1}{8} (L_i L_0)^2 = \frac{1}{2} (10 \text{ lb/ft}) (2.5 \text{ft-19ft})$

Vi = 1.800 16.ft

13.96 continued

POINT @ N2=N T2= 1 W 0= 0.75 07

 $V_2 = 0$ (CORD IS SLACK) $T_1 + V_1 = T_2 + V_2$

0.75 No +1.800 = 0.75 02 +0

FROM CONS. OF ANG. HOH. , V=3.125 VO

 $\frac{0.75}{9}$ $v_0^2[(3.125)^2-1] = 1.800$

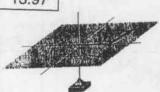
 $V_0^2 = \frac{(1.800 \text{ lb} \cdot \text{ft})(32.2 \text{ ft/s}^2)}{(0.75 \text{ lb})(8.7656)}$

U= 8.816 H2/52

Vo= 2.97 ft

(b) MAXIMUM VELOCITY OCCURS WHEN THE BALLS IS AT ITS HINIMUM DISTANCE FROM O (WHEN d=0.8ft) $\hat{U}_{m} = 3.125 \hat{U}_{0} = (3.125)(2.97) = 9.28 ft/s$ $\hat{U}_{m} = 9.28 ft$

13.97



GIVEN:

SPHERE OF MASS, M=0.64
FORCE BETWEEN
A AND O DIRECTED
TOWARD O OF
MAGNITUDE F=(80/F)N
VA= ZOM/S
HORIZONTAL
FRICTIONLESS PLANE

FIND:

(a) MAXIMUM AND MINIMUM DISTANCES FROM O
(b) CORRESPONDING VALUES OF THE SPEED

90° C r_m r_m 8 r_A 60°

THE FORCE EXERTED ON THE SPHERE PASSES THROUGH O. ANGULAR MONITORM ABOUT O IS CONSERVED

MINIMUM VELOCITY IS AT B WHERE THE DISTANCE FROM O IS MAXIMUM HAXIMUM VELOCITY IS A C WHERE DISTANCE FROM O IS MINIMUM FAMULASINGO-FM MOM

(0.5 M)(0.6kg)(20 M/3)SINGO-FM (6kg)(4)

Um = 8.66/FM

(1)

CONSERVATION OF ENERGY

AT POINT A $T_a = \frac{1}{2} m N_a^2 = \frac{1}{2} (0.6kg) (100 mb)^2 120 J$ $V = \left[F dr = \left[\frac{89}{9} dr = -\frac{89}{7} \right] V_a = -\frac{89}{0.5} = -160 J$

AT DOINT B TO= 1 m vm2 = 1 (0.6 kg) vm = 0.3 vm2

(AND POINTC) $V_{\theta} = \frac{80}{r_{m}}$

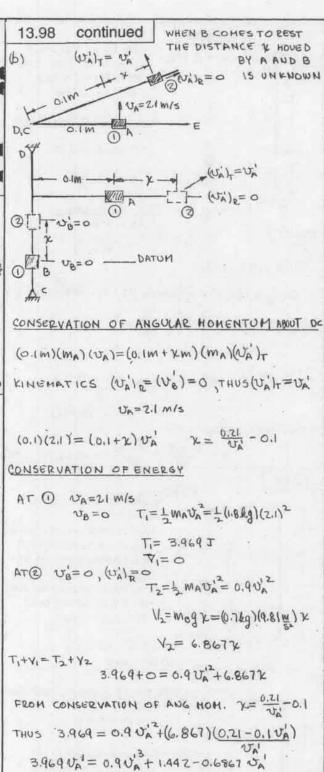
TA+VA=TB+VB

 $120 - 160 = 0.30 \text{ m}^{2} - \frac{80}{\text{Fm}}$ (2) 5085T1TUTE $(3) (NTO (2)) - 40 = (0.3) (8.66)^{2} - 80$ $120 - 160 = 0.30 \text{ m}^{2} - 80$ $120 - 160 = 0.30 \text{ m}^{2} - 80$ $120 - 160 = 0.30 \text{ m}^{2} - 80$

1m -2 m + 0.5625= 0 (CONTINUED)

13.97 continued - tm= 0.339 m AND tm= 1.661m VHAX 1.661 M YHIN =0,339 M (b) SUBSTITUTE I'M AND I'M FROM RESULTS OF PART (A) INTO (I) TO GET CORRESPONDING MAXIMUM AND MINIMUM VALUES OF THE SPEED V' = 8.66 = 25.6 M UNAX 25.6 M Um= 8.66 = 5.21 M UHIN = 5,21 M 13.98 GIVEN: MA= 1.8 Reg MB= 0.7 kg INITIALLY U = 2.1m AND UB= 0 A STOP IS SUDDENLY REMOVE AT B FIND: (a) VA, WHEN MA 15 0,2 m FROM O (b) VA, WHEN Va=0 (0) U= 21 m/s CONSERVATION OF ANGULAR HOHENTUH ABOUT DO (0.1m)(ma)(va)=(0.2m)(ma)(va)_ $(U_A^1)_T = (\frac{0.1}{0.2})(2.1 \text{ m}) = 1.05 \text{ m/s}$ CONSERVATION OF ENERGY 1 UA= 2.1 M/s T=1 (1.8 kg)(2.1 M/s)= 3.969 J VA= 0, CHOOSE DATUM FOR B AT ITS INITIAL POSITION AND NOTE THAT THE POTENTIAL ENERGY OF A DOES NOT CHANGE THUS WE TAKE VIO @ (V'A) = 1.050 m/s (V'A) = Va (KINEMATICS) T= 1 ma [(va) 2+(va) 2] + 2 mg (vg)2 T=== 1(1.8 kg) [(1.050 m/s)+(V'A)2]+= (0.7 kg) (V'A)2 Tz = 0.9923 + 1.25(VA) V= mg (0.1 m) = (0.7kg)(4.81 m/s2)(0.1m)=0.6867 J T+V=T, TY 3.969+0=0.9923+1.25(VA) +0.6867

U'A = V(U'A) + (U'A) = 1.832 52; (U'A) = 1.354 8 U'A = V(U'A) + (U'A) = [(1.05) + (1.354)] = 1.713 8 U'A - 26 0=TAN (U'A) + (U'A) = TAN 1.354 = 37.8°



4.6557 VA= 0.9 VA+1.44Z

UN = 0.316 M/s

5.173 VA = U3 + 1.602

BY TRIAL

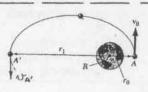
GIVEN:

SATELLITE LAUNCHED



FIND:

HAXIMUM ALTITUDE, USING CONSERVATION OF ENERGY AND CONSERVATION OF MOMENTUM



R=6370 km Vo=500 km+6370 km Vo=6870 km = 6.87 × 106 m Vo=36,900 km/h = 36.9 × 103 s = 10.25 × 103 m/s

CONSERVATION OF ANGULAR MOMENTUM

 $V_{A'} = (\frac{r_0}{r_1}) N_0 = (\frac{6.870 \times 10^6}{r_1}) (10.25 \times 10^3)$ $V_{A'} = \frac{70.418 \times 10^9}{r_1} (1)$

CONSERVATION OF ENERGY

 $V_0 = 10.25 \times 10^3 \frac{M}{5}$ $T_0 = \frac{1}{2} \text{ m V}_0^2 = \frac{1}{2} \text{ m (10.25 \times 10^3)}^2$

 $V_{A} = -\frac{GMm}{r_{o}} \qquad GM = 9R^{2} = (9.81 \text{ m/s}^{2})(6.37 \times 10^{6} \text{ m})^{2}$ $V_{A} = -\frac{(398 \times 10^{12} \text{ m}^{3}/\text{s}^{2}) \text{ m}}{(6.87 \times 10^{6} \text{ m})^{2}} = -57.93 \times 10^{6} \text{ m} (J)$

POINT A' TA' = I MUN'S

 $A^{VI} = -\frac{R}{248 \times 10_{15} M} = -\frac{1}{348 \times 10_{15} M} (1)$

TA+VA = TA+ YA

52.53×106 m-57.93×10 m= 1 m0 -398×10 m

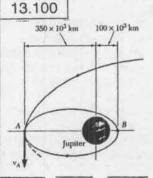
SUBSTITUTING FOR UN FROM LI)

 $-5.402 \times 10^{6} = \frac{(70.418 \times 10^{9})^{2}}{(21)(7)^{2}} - \frac{398 \times 10^{12}}{r_{1}}$ $-5.402 \times 10^{6} = \frac{(2.4793 \times 10^{21})}{r_{1}^{2}} - \frac{398 \times 10^{12}}{r_{1}}$

(5.402x 106) 1,2-(398x1012) 1, + 2.4793x 1021=0

r= 66.7x106m, 6.87x10 m

rmax=66,700 lem



GIVEN:

 $V_{\text{M}}=26.9 \text{ km/s}$ Mass of Jupiter $M_{\text{S}}=319 \text{ Mg}$

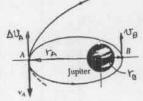
FIND:

AVA, TO BRING THE

SPACE CRAFT TO

WITHIN 100×103 km

CONSERVATION OF ENERGY



POINT A $T_A = \frac{1}{2} m (U_A - \Delta U_A)^2$ $V_A - G M_T M$

 $Q_{E} = 6.37 \times 10^{6} \text{ M}^{2} = 319 \text{ GM}^{2} = 319 \text{ G} \text{ M}^{2} = (319)(9.81 \frac{8}{52})(6.37 \times 10^{6} \text{ M})^{2}$ $GH_{J} = (319)(9.81 \frac{8}{52})(6.37 \times 10^{6} \text{ M})^{2}$ $GH_{J} = 17.6.98 \times 10^{5} \frac{\text{m}^{3}}{52}$

 $V_{A}=-\frac{(126.98\times10^{15}\text{m}^{3}/\text{s}^{2})\text{m}}{(350\times10^{6}\text{m})}$ $V_{A}=-\frac{(362.8\times10^{6})\text{m}}{(362.8\times10^{6})\text{m}}$

 $T_{B} = \frac{1}{2} m U_{B}^{2}$ $V_{B} = -\frac{G N_{5} m}{V_{B}} = -\frac{(126.98 \times 10^{15} 3/5^{2}) m}{(100 \times 10^{6} m)}$ $V_{B} = -(1269.8 \times 10^{6}) m$

TA+VA=TB+VB

 $\frac{1}{2} m (V_A - \Delta V_A)^2 - 362.8 \times 10^6 m = \frac{1}{2} m V_B^2 - 1269.8 \times 10^6 m$ $(V_A - \Delta V_A)^2 - V_B^2 = -1814 \times 10^6$ (1)

CONSERVATION OF ANGULAR MOMENTUM

VA=350X106M VB=100X106M

VAM (VA-AVA)=VBMVB

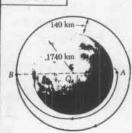
 $v_{B} = \left(\frac{r_{A}}{r_{B}}\right) \left(v_{A} - \Delta v_{A}\right) = \left(\frac{350}{100}\right) \left(v_{A} - \Delta v_{A}\right) \quad (z)$

SUBSTITUTE UB IN (2) INTO (1)

(UA-DOLA) [1-(3.5)] =- 1814×106

(TA-DVA)= 1612.4x106 (VA-DVA)= 712.648x103m (TAKE + ROOT3, - ROOT REVERSES FLIGHT DIRECTION) 5 VA= 269x103 m (GIVEN) DVA= (26.4x103 m - 12.698x103 m)

AVA= 14.20 km/5



GIVEN:

AT ENGINE SHUTOFF AT A

VA= 1740+=1748 & M

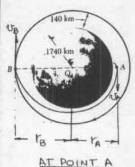
AT B, VB=1740+140=1880 M

COMMAND HODULE IN A

CIECULAR ORBIT

FIND;

- (a) SPEED AT A AT ENGINE SHUTOFF.
- (b) PELATIVE VELOCITY LEM APROACHES COHHAND MODULE AT A



CONSERVATION OF ANG. HOHENTUR

 $W_{B} = \frac{r_{A}}{r_{B}} U_{A} = \frac{1748}{1880} U_{A}$ $U_{B} = \frac{r_{A}}{r_{B}} U_{A} = \frac{1748}{1880} U_{A}$ $U_{B} = 0.9298 U_{A}$ (1)

CONSERVATION OF ENERGY

TA = 1 M VA VA = -GMMOON M

Mnoon = 0.0123 MEARTH

GHMOON=0.0123 GHERRTH = 0.0123 g REPETH

GM MOON = (0.0123)(9.81 M) (6.37×10 m)

G.M. = 4.896×1012 M3 TA= 1748×103 M

 $V_A = \frac{-(4.896 \times 10^{12} \text{m}^3 \text{K}^2) \text{ m}}{(1748 \times 10^3 \text{m})} = -2.801 \times 10^6 \text{ m}$

 $\frac{\text{AT POINT B}}{V_B} = \frac{1}{2} \text{ m V B}^2 \quad V_B = 1880 \times 10^3 \text{ m}$ $V_B = -\frac{GM_{MON M}}{V_B} = -\frac{(4.896 \times 10^{12} \text{ m}^3/\text{s}^2)M}{(1880 \times 10^3 \text{ m})} = -2.604 \times 10^6 \text{ m}$

TA+VA=TB+VB; = MVA-2.801x106m=1 MVB-2104x109M

 $\gamma_{A}^{2} = U_{B}^{2} + 393.3 \times 10^{3} \left(\frac{M^{2}}{5^{2}} \right)$ (2)

(a) SPEED AT A

SUBSTITUTE UB IN (1) INTO (2)

UA2 (1- (0.9298)2)=393.3×103

UA = 2.903×106 UA = 1.704×10 M VA = 1704 M

(b) AT POINT B

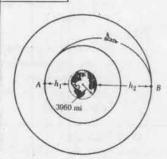
FROM (1) AND RESULT IN (a) UB=(0.9298)(1704)

COMMAND HODULE IS IN CIRCULAR ORBIT, 18=188XID6M

Veirc = V GHHOW = V 4.896x1012 = 1613.8 M

RELATIVE VELOCITY= VCIEC VB=1613.8-1584.0=29.8 5

13.102



GIVEN:

h=200 mi. h=500 mi.

FIND:

FOR A SPACECRAFT TRANSFERRING FROM A CIRCULAR ORBIT TO A CIRCULAR ORBIT AT B

(A) INCREASES IN SPEED AT A AND

(b) TOTAL ENERGY
PER UNIT HASS TO
EXECUTE THE TRANSFER

BETWEEN A AND B



 $W_A = W_B = \frac{23.549}{V_A} V_B = \frac{23.549}{21.965} V_B$

1= 3960 mi+200 m = 4160 mi

The 21.965×106ft UA=1.0721UB (1)

TB = 23.549×106ft R=(3960)(5280)=20.909×106ft

CONSERVATION OF ENERGY GM=gR=(32,2 \$\frac{1}{5})(20.909x100)
GM= 4.0777 x 1015 ft /52

POINT A: $T_A = \frac{1}{2} m V_A^2 V_A = \frac{GHm}{FA} = -\frac{(4.077 \times 10^{15})m}{(21.965 \times 10^6)}$

OINTB: $T_B = \frac{1}{2} m V_B^2 V_B = \frac{-GHM}{r_B} - \frac{(14.077 \times 10^6)M}{(23.549 \times 10^6)} = -59779 \times 10^6 M$

TATVA=TBTVB

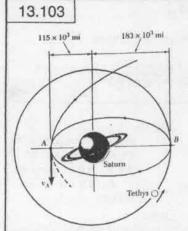
1 m 0 2-640.89 x 106 m= 1 m 0 -597.79 x 106 M 22-02= 86.219 x 106

FROM (1) $V_A = 1.0721 U_B$ $V_B^2 (1.072)^2 1) = 86.219 \times 10^6$ $V_B^2 = 576.98 \times 10^6 \text{ ft}^2/\text{s}^2$ $V_B = 24,020.4 \text{ ft}/\text{s}$ $V_B = 25.66 \text{ ft}/\text{s}$

CIRCULAR ORBIT AT A AND B (EQ. 12.44) $(V_A)_C = \sqrt{\frac{GH}{F_A}} = \sqrt{\frac{14.077 \times 10^{15}}{2.1965 \times 10^6}} = 25316 \text{ ft/s}$ $(V_B)_C = \sqrt{\frac{GH}{F_B}} = \sqrt{\frac{14.077 \times 10^{15}}{23.549 \times 10^6}} = 24450 \text{ ft/s}$

(A) INCREASES IN SPEED AT A AND AT B AUA=UA-(UA)c='25753'-25316 = 437 ft/s AUB=(UB)c-UB)= 24449-24000 = 429 ft/s (b) TOTAL ENERGY PER UNIT MASS E/m = \(\frac{1}{2} \rightarrow \frac{1} \rightarrow \frac{1}{2} \rightarrow \frac{1}{2} \rightar

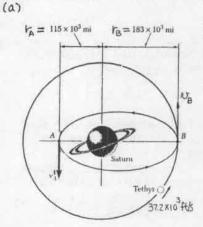
E/m= 216x10 ftb.



UA= 68.8 x103 ft/s UFH5=37.2x10 ft/5

FIND: (a) DECREASE IN SPEED, AVA OF A SPACECRAFT AT A TO ACHIEVE AN

IN CIRCULAR ORBIT ELLIPTICAL ORBIT THROUGH A AND B (b) THE SPEED UP OF THE SPACE CRAFT AS IT REACHES B



1= 607.2x10 ft ra=966.2×106ft

JA = SPEED OF CPACECRAFT IN THE ELLIPTICAL ORBIT AFTER ITS SPEED HAS BEEN DECREASED

ELLIPTICAL ORBIT BETWEEN AAND B CONSERVATION OF ENERGY

POINT A TA= 1 m 0 VA = GM SATM

MSA = HASS OF SATURN, DETERMINE GMSA FROM THE SPEED OF TETHYS IN IT'S CIRCULAR ORBIT

(EQ 12.44) VCIEC = VGMSAT GMSAT = FUCIRCE GMSAT = (966.2 x10 ft) (37.2 x10 ft/s)= 1.337 x10 8 ft3/52 $V_A = -\frac{(1.337 \times 10^{18} \text{ ft}^3/\text{s}^2) \text{ m}}{(607.2 \times 10^6 \text{ft})} = -2.202 \times 10^7 \text{ m}$

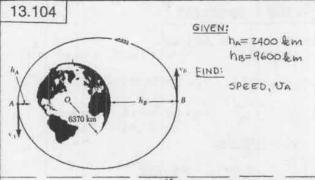
POINT B $T_B = \frac{1}{2} m v_B^2 v_B = \frac{GH_{SAT} m}{r_B} = \frac{(1.337 \times 10^{18} t_1^2 t_2^2) m}{(466.2 \times 10^6 4t)}$

TA+VA=TB+VB; = WU'2-2.202x109M=1MU2-1,384 X109M

CONSERVATION OF ANGULAR HUMEN UM

YA MUA = YB MUB " B= YA VA = 607.2 x10 6 UA = 1.6784 UA UA [1-(0.6284)]=1.636×109 VA= 52005ft/5 DUA= UA-V'= 68800-52005= 16 79594/5

 $U_B = \frac{t_A}{r_B} U_A' = (0.6284)(52005) = 32700 + 1/5$



ra= 6370km+2400km TA= 8770 km ra=6370 km+9600 km=15970 km

CONSERVATION OF MOHENTUM VAMUA= VBMUB

 $U_B^* = \frac{V_A}{V_B} U_A = \frac{8770}{15970} U_A = 0.5492 U_A$ (1)
CONSERVATION OF ENERGY TA = 1 maja VA = -GMM TB 1 MUB VB - GHM GM=9R= (4.91%; (6370×103m)= 398.1×102 m2

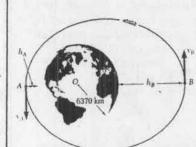
 $V_{A} = \frac{-(398.1 \times 10^{12})}{8770 \times 10^{3}} = 45.39 \times 10^{6} \text{m}$

$$V_B = -\frac{(398.17.10^{12})}{(15970\times10^3)} = -24.93 \text{ m}$$

TA+VA=TB+VO = 1000= 45.39 X:060= 500= 2493Ki0602

SUBSTITUTE FOR UB IN IN FROM (1) UA [1- (0.52427] = 10.92 × 10 6 UA = 58.59×106 52 UA= 27.6×103 km/h -

 $\cdot 13.105$



Un=26.3×10 km/h UB=185×10km/n FIND:

ALTITUDE, ho

NA= 26.3×10° km/h UA = 7.31×103 M/s UR=18.5×10 km/n=5.14×10 m/s

CONSEQUATION OF MOHENTUM KAMUA=KBMUB

$$r_A v_A = r_B v_B$$
 $r_A = \frac{U_B}{V_A} r_B = \frac{18.5}{26.5} r_B$ $r_A = 0.7034 r_B$ (1)

13.105 continued

CONSERVATION OF ENERGY

TA = 1 m UA2 TA = 1 m (7.31x103) = 26.69x10 m

TB= 1 m UB TB= 1 m (5.14×103)= 13.20×106 m.

VA= - GMM GM=gR=(9.81 1) (6370×10)2 GM= 398.1x10 2 m3/52

VA=-398, 1×1012

TATVA=TRTVB

26.69 × 10 6/ - 398.1× 10 1/ = 13.20 × 10 6/ - 398.1× 10 8/ SUBSTITUTE FOR IA FROM (1) $\frac{398.1 \times 10^{12}}{\text{Fe}} \left[\frac{1}{(0.7034)} - 1 \right] = 13.49 \times 10^{6}$

1= 80.37 ×10-9

VB= 12.4 42×106 m=12442 lem

h= r8-R= 12442km-6370km= 6070km

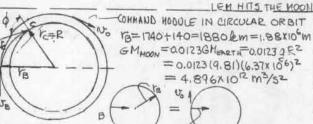
* 13.106

GIVEN:

COMMAND MODULE IN CIRCULAR OFBIT AT AN ALTITUDE OF 140 km ATTACHES LEM CAST ACRIFT AT RELATITE VELOCITY 0F200m/s

FIND:

UE AND & AS THE



R=1740 km

ZF=Wan GMm = myo Vo=VGMm = 14.886×1012 No=1614 m/s VB=1614-200=1414 m/s

CONSERVATION OF ENERGY BETWEEN BAND C 1 mVB-GHMM = 1 mVc2 - GHMM

V2= UB+2GMm (FB-1) Uc2=(1414 m/s)2+2(4.896×1012m2/52)(1.88×106-1)

12=1,999×1040.4191×106= 2.418×106 m2 Vc= 1555 %

* 13.106 continued

CONSERVATION OF ANGULAR HOMENTUM

ramur=RMUc SINO SING = 120 = (1.88 x 106m)(1414 m)

(1.74 ×106m)(1555 (4) D= 79.26 \$=79.3°

13.107

GIVEN:

SATELLITE PROJECTED AT VELOCITY US AT AN ANGLE & WITH ITS INTENDED CIRCULAR

FIND;

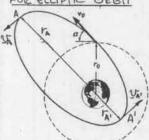
THAX AND THIN



V. FOR CIRCULAR OPBIT OF PADIUS to F=man GHM = m 152

BUT NO FORMS AN AUGLE OF WITH THE INTENDED CIRCULAR PATH

FOR ELLIPTIC OPBIT



CONS OF ANG MOMENTUM tomuo cosa= tamua

VA= (0050) Vo

CONS OF FIFERY

1 mvo- GMm =

U0-VA2 = 2-6M (1-10)

SUBSTITUTE FOR UA FROM (1)

BUT 102= GH, THUS 1-12 305 d= 2 (1-12) cos & (1/2 / 2/ (1/2)+1=0

SOLYING FOR, k

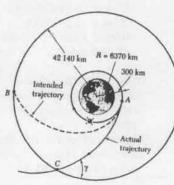
$$\frac{V_0}{V_A} = + 2 + \sqrt{\frac{4 - 4\cos^2\alpha}{4 - 4\cos^2\alpha}} = \frac{1 + \sin\alpha}{1 - \sin^2\alpha}$$

1= (1+sind)(1-sind) 10=(17sind) to P ALSO YALID FOR POINT

THUS

MAX = (I+SINX) TO MIN= (HSINX) NO



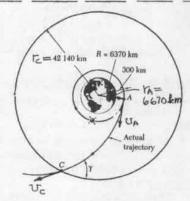


SATELLITE AT A

12 LAUNCHED
WITH A YELOCIN
RELATIVE TO A
SPACE PLATFORM
IN CIRCULAR
ORBIT OF
(VAI_R=3.44 km/s)

EIND:

ANGLE Y AT WHICH THE SATELLITE CROSSES THE CIRCULAR ORBIT AT C.

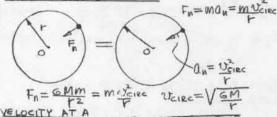


R=6370km YA=6370km+300km YA=6.67.X106M

r=42.14x 106m

GM= g R 2 GM= (9.81 \$)(6.37x10 m) GM= 398.1x10 2 m2/52

FOR ANY CIRCULAR OBBIT



 $(N_A)_{CIEC} = \sqrt{\frac{GM}{Y_A}} = \sqrt{\frac{398.1 \times 10^{12} \text{ m}^3}{(6.67 \times 10^{6} \text{ m})}} = 7.726 \times 10^{3} \text{ m}$

JA=(7)A 161RC+(1)A 1 = 7.726x103+3.44x10=11.165x103M

YELOCITY ATC

CONSERVATION OF ENERGY TATVA=TC+VC

UC = UA + 2 GN (+ - +) = 11.165x10) + 2 (398.1x10) (1 LOUX 10 66710)

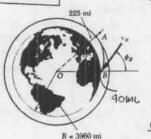
Uc= 124.67x10-100.48x10= 24.19x10 = 24.19x10

CONSERVATION OF ANGOLAR HOHENTOM

FAMULA = rcm Uccos 8 TAVA -16.67x106)(11.165x103)

COS 8 = TAVA -16.67x106)(4.919x103)

cosx=0.35926 X=68.9° 13.109

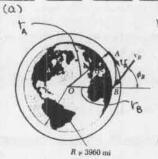


GIVEN:

VEHICLE IN CIRCULAR
ORBIT AT ALTITUDE
OF 225 ML. SPEED
DECREASED AT A
SO THAT IT REACHES
ALTITUDE AT B OF
40 ML AT AN
ANGLE \$\phi_{B} = 60°\$

FIND:

(a) UA, AS VEHICLE LEAVES ITS CIRCUXAR ORBIT (b) UB



7=3960 mc+225 mc=4185mc

18=3960mc+40mi=4000mi 18=4000x5280=21120x103ft

R=3960 m = 20909 x103ft

GM=g E= (322 H) (20909 MOR)

GN= 14.077X1015 ft3/52

CONSERVATION OF ENERGY

TA= 1 M UA VA - GHM = 14.077X10 M -637.1X10 M

TB= 1 MUB VB=-GHM = -14.077x16 M=-6665x10m

TA + VA=TB+YB

1 MUA- 637,1×10°M= 1 MUB-666.5×10°M

UA=UB-58.94×106

CONSERVATION OF ANGULAR HOHENTUH

 $V_{B} = \frac{(r_{A}) V_{A}}{(r_{B})(SIN\Phi_{B})} = \frac{4185}{4000} \left(\frac{1}{SINGOO}\right) V_{A}$ $V_{B} = \frac{(r_{A}) V_{A}}{(r_{B})(SIN\Phi_{B})} = \frac{4185}{4000} \left(\frac{1}{SINGOO}\right) V_{A}$ (2)

SUBSTITUTE UB FROM (2) IN (1)

UA=(1208UA)2-58.94X106

 $V_A^2[(1.208)^2-1] = 58.94 \times 10^6$

UA2= 128,27 × 106 ft/52

(a) $V_A = 11.32 \times 10^3 \text{ ft/s}$

(b) FROM (2)

 $v_B = 1.208 \, v_A = 1.208 \, (11.32 \times 10^6) = 13.68 \times 10^3 \, \text{ft/s}$ $v_B = 13.68 \times 10^3 \, \text{ft/s}$

*13.110

GIVEN:

VEHICLE AT A IN
CIRCULAR ORBIT IS
GIVEN AN INCREMENTAL
UELOCITY AUA TOWARD
O. ALTITUDES AS SHOWN
ENERGY EXPENDITURE
IS 50% OF THAT
USED IN PROB B.109

FIND:

VB AND DB

R=3960 ML

VA 225 ml ΔUA

VA 225 ml ΔUA

VIII

3960 mi

TA=3960 MI+225 MI Valenc VA=4185 MI=22.097 X10 ft TB=3960 MI+40 MI=4000 MI VB=21.120 X10 ft

GM=gR=(37.2 ft) (3960) (5280) Hg

GM=14.077×10 15 ft3/52

VELOCITY IN CIRCULAR GEBIT AT 275 M ATTROS



Man Calcier 122

MENTOUS SECOND LAW

F= man $\frac{GMM}{k_{A}^{2}} = \frac{M(\lambda_{A})^{2}}{k_{A}} = \sqrt{\frac{4.017 \times 10^{15}}{22.097 \times 10^{6}}} = 25.24 \times 10^{3} \text{ ft/s}$

ENERGY EXPENDITURE

FROM PROB. 13.109 NA=11.32×103 ft/s

ENERGY, DE109 = 1 m (VA) circ - 1 m NA?

DE109 = 1 m (VA) circ - 1 m (11.32×10)?

ΔΕ109= 254.46×106 m ft.16 0 ΔΕ10= (0.50)ΔΕ109 = (254.46×106m)/2 ft.16 THUS, ADDITIONAL KINETIC ENERGY AT A 15 ½ m (Δ 0_A)² = ΔΕ110=(254.46×106m)/2 (1)

CONSERVATION OF ENERGY BETWEEN A AND B

 $T_{B} = \frac{1}{2} m V_{B}^{2} \qquad V_{B} = -\frac{GHm}{Y_{A}}$ $T_{A} + V_{A} = T_{B} + V_{B}$ $\frac{1}{2} m (25.24 \times 10^{3})^{2} + 254.46 \times 10^{6} m - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^{15} m}{22.097 \times 10^{6}} = \frac{1}{2} m V_{B}^{2} - \frac{14.077 \times 10^$

 $V_B^2 = 637.4 \times 10^6 + 25496 \times 10^6 - 1274.1 \times 10^6 + 1333 \times 10^6$ $V_B^2 = 950.4 \times 10^3$

COUSE EVATION OF ANGULAR HOMENTUM BETWEEN AMOBE TO MIVA RICE = 18 M UR ALM BE

13.111



GIVEN:

LEM AT AN ALTITUDE OF 140 &M IS SET ADRIFT FROM A CIRCULAR ORBIT AND ITS SPEED IS REDUCED

EIND:

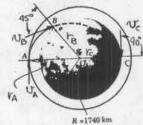
(a) SHALLEST REDUCTION

OF SPEED TO HAKE SURE

THE LEM WILL HIT THE

MOON

(b) THE REDUCTION IN SPEED WHICH WILL CAUSETHE LEM TO HIT THE HOON AT 450



YA=1740 km+140 km=1880 x10 m YB=16= R=1740 km=1740 x10 m

GM_{MOON} = 0.0123 GMg = 0.0123 g P₅² = (0.0123)(981 m/52)(63 m²) GM_{MOON} = 4.896×10¹² m²/5²

VELOCITY IN A CIRCULAR ORBIT AT HOLEM ALTITUDE

UCIEC VEMMOON = V4.896X1012 M3/52 = 1.6138×103 M

(A) AN ELLIPTIC TRAJECTORY BETWEEN A AND C, WHERE THE LEM IS JUST TANGENT TO THE SURFACE OF THE HOON, WILL GIVE THE SHALLEST REDUCTION OF SPEED AT A WHICH WILL CAUSE IMPACT COMPSERVATION OF ENERGY (A AND C)

TA= 1 m NA VA= -GHM = - 4.896X10 M = -2.604X10 M

TC= \$ M V2 VC= -CHMM = 4.896x10 M - 2.814x10 M
TC = 1740x103

TA+VA=Tc+Vc = mv2=2.604x10m== mv2=2.81410x10m

CONSTRUCTION OF ANGULAR MOMENTUM (AANDC)

VC= 1/4 VA = 1/880 VA = 1.0805 VA

REPLACE UZ IN (1) BY (2)

UAZ=(1.0805 VA)Z 419.1X10Z UAZ[(1.0805)Z 1] = 419.1X10Z VAZ= 2.50ZX10G VA=158Z M

AVA=(VALIE-VA=1614-1582=31.5mg

(b) CONSERVATION OF ENERGY (A AND B)
SINCE TB=TC COND OF ENERGY IS THE SAME

AS BETWEEN A AND C.
THUS FROM (1) UA=UB-419.1×103 (1)

CONSERVATION OF ANGULAR HOHENTUM (A AND B)
TAMVA=TBMVASIN \$ \$=450

UB= TAVA = 1880UA = 1.528 UA (3)

REPLACE UB IN (1') BY (3)

 $\Delta \Delta^2 = (1.528 \ \Delta^2) - 419 \times 10^3$ $\Delta \Delta^2 = 313.48 \times 10^3 \quad \Delta_A = 560 \ \text{M}$ $\Delta \Delta_A = (\Delta_A \times 10^3 \ \Delta_A = 1614 - 560 = 1053 \ \text{M}$ *13.112

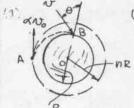
GIVEN:

SPACE PROBE IN CIRCULAR ORBIT OF RADIUS NR. WITH VELOCITY U. ABOUT A PLANET OF RADIUS P.

SHOW THAT:

(a) PROBE WILL HIT THE PLANET AT AN ANGLE O WITH THE VERTICAL, IF ITS VELOCITY IS DEDUCED TO XVO WHERE X = SINO \(\frac{2(n-1)}{n^2 \sin^2 \text{O}}\)

(b) PROBE WILL HISS THE PLANET IT X > \(\frac{2}{1+n}\)



(a) CONSERVATION OF ENERGY AT A TA=1 m(~Vo)?

> VA= - GNM NR

 $T_{D} = \frac{1}{2} M O^{2}$

M=MASS OF PLANET M=NASS OF PROBE VB=-GMM

TA+VA=TB+VB

 $\frac{1}{2}M(AN)^{2}\frac{GMM}{NR} = \frac{1}{2}MN^{2}\frac{GHM}{R}$ (1)

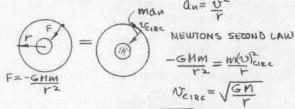
CONSERVATION OF ANGULAR HOMENTUM

NRM XNO = EMNTSING

REPLACE OF IN (1) BY (2)

$$(\omega N_0)^2 \frac{26M}{NR} = \left(\frac{N\omega N_0}{51N0}\right)^2 - \frac{2GM}{R}$$
 (3)

FOR ANY CIRCULAR OPBIT



FOR r= NR Uo=VCIEC = VGM

(b) PROBE WILL JUST HISS THE PLANET IF 0 > 90°,

NOTE: $N^2 - 1 = \frac{1}{(N-1)(N+1)} = \sqrt{\frac{2(N-1)}{N^2 - 51N^2q_06}} = \sqrt{\frac{2}{N+1}}$

13.113



GIVEN:

V_P AND V_A AS SHOWN
SHOW THAT:

U_A = 2 GN r_P
TA+V_P r_A

Up2 = 2GM ra rate ra

CONSCRIATION OF ANGULAR HOHENTUM

 $V_A m U_A = V_P m U_P \qquad \qquad V_A = \frac{V_P}{V_A} U_P \qquad \qquad (1)$

I MUP2-GHM = 1 MUA2-GHM (2)

SUBSTITUTING FOR UA FROM (1) INTO (2) $V_p^2 - \frac{2CM}{T_p} = \left(\frac{T_p}{V_A}\right)^2 V_p^2 - \frac{2CM}{T_A}$

 $\frac{\left(1-\left(\frac{r_p}{r_A}\right)^2\right) \upsilon_p^2 = 7GH\left(\frac{1}{r_p}-\frac{1}{r_A}\right)}{V_p^2 = 2GH\left(\frac{r_A-r_p}{r_A-r_p}\right)}$

WITH ra- ra = (ra-ra) (ra+ra)

 $V_p^2 = \frac{2.6 \, \text{M}}{r_0 + r_0} \frac{r_0}{r_0} \tag{3}$

EXCHANGING SUBSCRIPTS PAND A

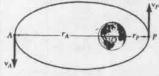
Un = ZEM TO (QED)

13.114 GIVE

EARTH SATELLITE OF HASS M
DESCRIBING AN ELLIPTIC ORBIT
FA IS HAXIMUM AND FOR HINIMUM
DISTANCES TO EARTHS CENTER

SHOW THAT:

TOTAL ENERGY E = - GMM , WHERE M = MASS OF THE EARTH



SEE SOLUTION TO PEOB 13.113 (FBOYE) FOR DERIVATION OF EQUATION (3)

Up= 2GM FA

TOTAL ENERGY AT POINT P 15

$$\begin{split} E &= T_{p} + V_{p} = \frac{1}{2} M \mathcal{O}_{p}^{2} - \frac{GMM}{\Gamma p} \\ &= \frac{1}{2} \left[\frac{F_{G}MM}{\Gamma_{A} + \Gamma_{p}} \frac{r_{A}}{\Gamma_{p}} \right] - \frac{GMM}{\Gamma_{p}} \\ &= GMM \left[\frac{V_{A}}{\Gamma_{p} (\Gamma_{A} + \Gamma_{p})} - \frac{1}{\Gamma_{p}} \right] = GMM \frac{(\Gamma_{A} - \Gamma_{A} - \Gamma_{p})}{\Gamma_{p} (\Gamma_{p} + \Gamma_{p})} \end{split}$$

E= - GHM

NOTE: RECALL THAT GRAVITATIONAL POTENTIAL
OF A SATELLITE IS DEFINED AS BEING
ZERO AT AN INFINITE DISTANCE FROM
THE EARTH

GIVEN:

SPACECRAFT OF HASS M IN CIRCULAR OPBIT OF PADIUS K ABOUT THE EARTH

SHOW THAT:

(A) ADDITIONAL ENERGY DE TO TRANSFER IT TO A CIRCULAR OPBIT OF LARGER RADIUS & AE= SMM(13-17)

(b) AHOUNTS OF ENERGY AT A AND B ARE DEA - KT DE DE DE KIND

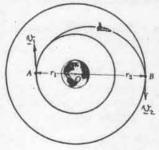


(a) FOR A CIRCULAR ORBIT OF PADIUS + F=man: GHM = MU $U^{2} = \frac{GM}{F}$ $E = T + V = \frac{1}{2} \frac{M}{M} U^{2} - \frac{GHM}{F} = -\frac{1}{2} \frac{GHM}{F}$ (1)

THUS DE REQUIRED TO PASS FROM CIRCULAR ORBIT OF PADIUS FITO CIRCULAR ORBIT OF PADIUS 12 15 DE=ELES=-TEHM +TEHM

(b) FOR AN ELLIPTIC ORBIT WE RECALL EQ (3) DERIVED IN PROBLEM 13.113 (WITH Up= U1)

N1 = 26M 12



AT POINT A: INITIALLY SPACECRAFT IS IN A CIRCULA ORBIT OF RADIUS Y

AFTER THE SPACECRAFT ENGINES ARE FIRED AND IT IS PLACED ON A SEMI-GLLIPTIC PATH AB, WE RECALL

U,= 26M 1/2

TI= 12 MO1= 12 MT(TI+FZ) AT POINT A, THE INCREASE IN ENERGY IS

$$\Delta E_{A} = T_{1} - T_{CIRC} = \frac{1}{2} M \frac{1 GM r_{2}}{r_{1}(r_{1} + r_{2})} - \frac{1}{2} M \frac{GM}{r_{1}}$$

$$\Delta E_{A} = \frac{GM M (2 r_{2} - r_{1} - r_{2})}{2 r_{1}(r_{1} + r_{2})} = \frac{GM M (r_{2} - r_{1})}{2 r_{1}(r_{1} + r_{2})}$$

RECALL EQ (2): DEA = 12 DE (Q.E.D)

A SIMILAR DERIVATION AT POINT B YIELDS, DEBTINEDE (QED)

13,116

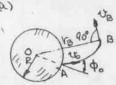
GIVEN:

HISSILE FIRED FROM THE GROUND WITH VELOCITY TO AT AN ANGLE O WITH THE VERTICAL, REACHES A MAXIMUM ALTITUDE & P. WHERE P IS THE PACINS OF THE EARTH

SHOW THAT: (a) SIN \$ = (1+0) 1 - \(\times \) \(\frac{\times \times \) \(\frac{\times \times \}{17} \)

WHERE VESC = ESCAPE VELOCITY

(b) PANGE OF ALLOWABLE VALUES OF TO



RMU. SIND = TE MUB

TO B PROPERTY OF ANG. MOM.

TO B PROPERTY OF ANG. MOM.

TO B PROPERTY OF ANG. MOM. UB= PUOSINDO = NOSINDO

TA+VA=TB+VB 1 m vo- GMM P = 1 m vo- GMM (1+0)R

$$U_0^2 - U_B^2 = \frac{2 \text{GMM}}{R} \left(1 - \frac{1}{1 + \alpha}\right) = \frac{2 \text{GMM}}{R} \left(\frac{\alpha}{1 + \alpha}\right)$$
SUBSTITUTE FOR U_B FROM (1)

$$N_{0}^{0}\left(1-\frac{(1+\alpha)_{2}}{2!N_{3}\Phi^{0}}\right)=\frac{S}{SCHM}\left(\frac{1+\alpha}{q}\right)$$

PROM EQ. (12.43): UESC = 26M

$$\Lambda_{3}^{0}\left[1-\frac{(1+\alpha)_{3}}{2^{1}N_{3}\varphi^{0}}\right]=\Lambda_{5}^{\text{esc}}\left(\frac{1+\alpha}{\alpha}\right)$$

$$\frac{\sin^2 \phi_0}{(1+\alpha)^2} = 1 - \left(\frac{v_{\text{esc}}}{v_0}\right)^2 \frac{\alpha}{1+\alpha} \qquad (2)$$

(b) ALLOWABLE VALUES OF VO (FOR WHICH HAXIMUM ALTITUDE = XP)

FOR SINGO = 0 , FROM (2)

$$0 = 1 - \left(\frac{N_{\text{ESC}}}{N_{\text{c}}}\right)^{\frac{1}{2}} \frac{\lambda}{1+\lambda}$$

FOR SINDO=1, FROM Z

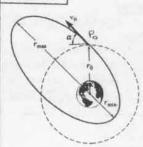
$$\frac{1}{(1+\alpha)^2} = 1 - \frac{\sqrt{\log \alpha}}{\sqrt{\log \alpha}} \frac{1}{1+\alpha}$$

$$\left(\frac{\sqrt{\log \alpha}}{\sqrt{\log \alpha}}\right)^2 = \frac{1}{\alpha} \left(1+\alpha - \frac{1}{1+\alpha}\right) = \frac{1+2\alpha + \alpha^2 - 1}{\alpha \cdot (1+\alpha)} = \frac{2+\alpha}{1+\alpha}$$

$$\sqrt{\log \alpha} = \sqrt{\frac{1+\alpha}{2+\alpha}}$$

$$\sqrt{\log \alpha} = \sqrt{\frac{1+\alpha}{2+\alpha}}$$

*13.117



GIVEN:

FROM PROB. 13.107 MIN = 10 (1-51NX) MAX = 10 (1+51NX)

: TAHT WOHE

ENTENDED

CIRCULAR DEBIT

AND RESULTING

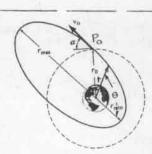
ELLIPTIC ORBIT

THE ENDS OF

THE MINORANS

OF THE ELLIPTIC

ORBIT AT PO



IF THE POINT OF INTERSECTION
PO OF THE CIRCULAR AND
ELLIPTIC OPBITS IS AT AN
END OF THE MINOR AXIS,
THEN UO IS PARALLEL TO
THE MAJOR AXIS. THIS
WILL BE THE CASE ONLY
IF \$\alpha + 90^\circ=\text{ONLY}
IS IF \$\alpha 50.00 \text{ONLY}
US HUST THEREFORE
PROVE THAT

COS 0=-21NK (1)

WE RECALL FROM EQ (1239):

 $\frac{1}{r} = \frac{GM}{N^2} + C \cos \Theta \qquad (2)$

WHEN 6=0, r= rmin AND rmin= ro(1-SINK)

FOR 0=180°, r= rmax= Vo (HSINW)

$$\frac{1}{r_0(1+SININ)} = \frac{G11}{W^2} - C \tag{4}$$

ADDING (3) AND (4) AND DIVIDING BY Z:

SUBTRACTURE (4) FROM (3) AND DIVIDING BYZ:

$$C = \frac{1}{2r_0} \left(\frac{1}{1-51Nd} - \frac{1}{1+51Nd} \right) = \frac{1}{2r_0} \frac{251Nd}{1-51N^2d}$$

C= 21N4

SUBSTITUTE FOR GH AND C INTO EQ (2)

$$\frac{1}{r} = \frac{1}{r_0 \cos^2 \alpha} \left(1 + \sin \alpha \cos \theta \right) \tag{5}$$

LETTING V= 10 AND 0 = 00 IN EQ (5), WE HAVE

COSS X = 1+ SING COSOO

$$cos\theta^o = \frac{cillar}{328_20_{i-1}} = -\frac{2ing}{8in_3q} = -8inq$$

THIS PROVES THE VALIDITY OF EQUIN AND THUS PO IS AN END OF THE MINOR AXIS OF THE ELLIPTIC ORBIT * 13.118

GIVEN:

SPACE YEHICLE UNDER GRAVITATIONAL ATTRACTION OF A PLANET OF HAS H (FIG. 13.15, SHOWN BELOW)

FIND:

(a) TOTAL ENERGY PER UNIT HASS, E/M, IN TERMS
OF FMIN AND UMON AND THE ANGULAR
MOMENTUM PER UNIT MASS, N.

DERIVE:
(b) $\frac{1}{F_{min}} = \frac{GM}{h^2} \left[1 + \sqrt{1 + \frac{2E}{m} \left(\frac{h}{GM}\right)^2}\right]$

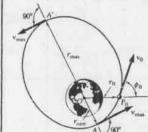
SHOW THAT: (C) ECCENTRICITY E=VI+ZE (h)2

(d) TRAJECTORY IS Q,

HYPERBOLA IF E>O

FULIPSE IF E=O

PARABOLA IF E<O



ANGULAR MOMENTUM PER UNIT HAS

h = Ho = Ymin M Umax

h= rmin Umax(1)

(b) ENERGY PER UNIT MASS

E/M= 1/m (T+V)

(b) FROM EQ. (1): WHAX= h/rmin. SUBSTITUTING INTO (2)

$$E/M = \frac{1}{2} \frac{h^{2}}{r_{min}^{2}} - \frac{GM}{r_{min}}$$

$$\left(\frac{1}{r_{min}}\right)^{2} - \frac{2GM}{h^{2}} \cdot \frac{1}{r_{min}} - \frac{2(E/M)}{h^{2}} = 0$$

SOLVING THE QUADRATIC: The GM + VGM 12 2(E/M)

$$\frac{1}{r_{min}} = \frac{GM}{h^2} \left[1 + \sqrt{1 + \frac{2E}{M} \left(\frac{h}{GM} \right)^2} \right]$$
 (3)

(c) ECCENTRICITY OF THE TRAJECTORY EQ (12.39') $\frac{1}{r} = \frac{GM}{h^2} (1 + \varepsilon \cos \theta)$

WHEN $\Theta=0$, $\cos\theta=1$ AND $r=r_{min}$, THUS $\frac{1}{r_{min}}=\frac{GM}{h^2}(1+\varepsilon)$ (4)

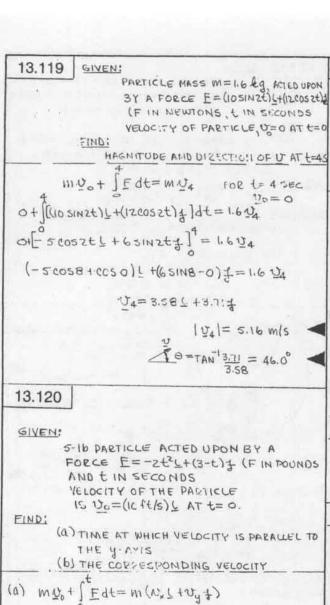
COMPARTIG (3) AND (4) E= VI+ 2E (h)2 (5)

(d', RECALLING DISCUSSION ON PAGES 708,709 AND IN VIEW OF EQ. (5)

1. HYPERBOLA IF E>1, THAT IS IF E>0 2. PARADOLA IFE=1, THAT IS IF E=0 3. ELLIPSE IF E<1, THAT IS IF E<0

NOTE: FOR CIRCULAR OFBIT E=0 AND $1+\frac{2\pi}{M}(\frac{\Lambda}{GM})^2=0$ OR $E=-(\frac{GM}{M})^2\frac{M}{2}$

BUT FOR CIRULAR ORBIT UZ GH AND HZ UZ TE GMY
THUS E = - 1 M (GH)Z = - 1 GHM (CHECKS WITH (1)
FOUND IN 13:115)



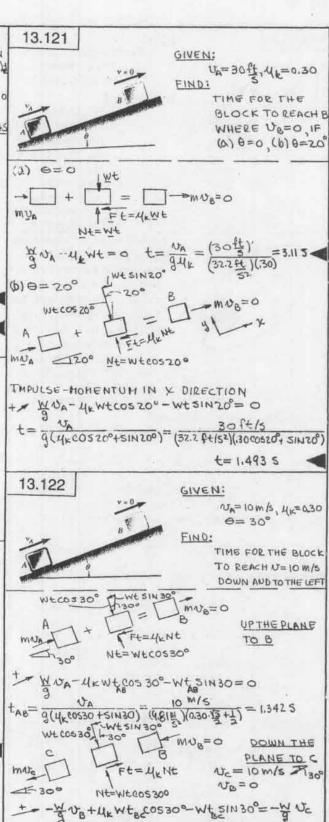
(a) TIME AT WHICH VELDCITY IS PARALLEL TO

BUT
$$0_{x=0}$$
, IF VELOCITY IS PARALLEL TO y -AXIS
$$\frac{g}{g} = \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} = \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} = \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} = \frac{1}{2} \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}$$

STRICE THE X COMPONENT OF THE VELOCITY IS ZERO +3= 2.329

t=1,3265 (b) SUBSTITUTE t=1.326 IN (1)

$$c_{\perp} + \left[3(1326) - \frac{(1326)^{2}}{2}\right]_{\frac{1}{4}} = \frac{5}{32.2} v_{4} + \frac{5}{32.2} v_{4}$$



10 m/s

TBC = 9(51N30°-4KC0530°) (9.81 M2)(1-3(19))

t= tAB+tBC=1.342+4.244= 5.59 5

=4.744 5

GIVEN:

REAR (DRIVE) WHEELS OF A CAR SLIP FOR FIRST GOFT WITH FRONT WHEELS JUST OFF THE GROUND, 4k = 0.60 WHEELS POLL WITHOUT SLIPPING FOR THE REMAINING 1260 FE WITH 60 % OF THE

WEIGHT ON THE REAR WHEELS. 45=0.85 IGNORE AIR AND ROLLING RESISTANCE

FIND:

(a) SHORTEST TIME FOR THE CAR TOTRAVEL THE FIRST GOFT STARTING FROM REST

(b) MINIMUM TIME FOR THE CAR TO RUN THE WHOLE RACE

(a) FIRST 60 ft

VELOCITY AT 60 Ft REAR WHEELS SKID TO GENERATE THE MAXIMUM FORCE RESULTING IN MAXIMUM VELOCITY AND HINIHUM TIME SINCE ALL THE WEIGHT IS ON THE REAR WHEELS THIS FORCE IS F=4KN=0.60W

WORK AND ENERGY: To + Uc-60= TGO To=0 U0-60=(F)(60) T60= = moleo

137 = (2)(0.60)(60 ft) (32.2 ft/s2) N60= 48.15 ft/s

IMPULSE - HOMENTUM

60 + 60 = 60 - molo A Ft=UENT Nt=Wt

+ 0+4kWt -6-60 W 060 060= 48 15 ft/s to-60= 48.15 ft/s (0.60) (32.2ft/s2)

to-60= 2.495

(b) FOR THE WHOLE PACE

THE HAXIMUM FORCE ON THE WHEELS FOR THE FIRST 60 ft 15 F=4 W=0.60W FOR REMAINING 1260 Ft THE HAXIMUM FORCE IF THERE IS NO SLIDING AND 60 % OF THE WEIGHT IS ON THE BEAR (DRIVE) WHEELS IS F=4<(0.60)W= 0.85)(0.60)W=0.510W

VELOCITY AT 1320 ft

WORK AND ENERGY TOTY +U = TI320

To=0 40-60=(0.60 W)(60 ft), 4=(0.510 W)(1260ft) T1320= = W01320

0+36W+(1510)(1260)== 2 \$ 101320 Viszo= zogoft/s

IMPULSE-MOHENTUM FROM GOFT TO 1320 Ft

(00 + 00 = 00 → MV1320 → MV60 1 Ft=45Nt Nt=Wt F=Us N=0.510W V60=48.15 ft/s V1320=209 Pt/s

(W) (48.15)+0.510Wto-1320= \$ (209); t60-1320= 9.7955 to-1320=to-60+t60-1320=2.49+9.80=12.295

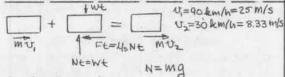
13.124

GIVEN:

TRUCK ON LEVEL ROAD TRAVELING AT 90 km/h BRAKES ARE APPLIED TO SLOW IT TO 30 km/h ANTISKID BRAKING SYSTEM LIMITS BRAKING FORCE SO THAT WHEELS ARE AT IMPENDING SLIDING, 45 = 0.65

FIND:

SHORTEST TIME FOR TRUCK TO SLOW DOWN



mvi-4. Nt=mv2 m(25 M/s)-(0.65)m (9.81 mg)t=m (8.33 m/s)

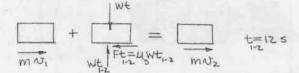
 $t = \frac{25 - 8.33}{(0.65)(9.81)} = 2.615$

13.125 GIVEN:

TRAIN DECREASES SPEED FROM 200 km/h TO 90 km/h AT A CONSTANT PATE IN 12 S.

FIND:

SMALLEST ALLOWABLE COEFFICIENT OF FRICTION IF A TRUNK IS NOT TO SLIDE



1=200kn/n=55,56 m/s N=90km/h=25.0 m/s

(55.56 m/s)-45 (9.81 m/s) (125) = 25 m/s

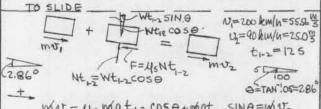
$$\mu_5 = \frac{(55.56 - 25.0)}{(4.81)(12)} = 0.2596$$
 $\mu_5 = 0.260$

13,126 GIVEN:

TRAIN DECREASES SPEED FROM 200 km/h TO GORMIN DOWN A 5% GRADE AT A CONSTANT RATE IN 12 5.

FIND:

SHALLEST COEFFICIENT OF FRICTION IF A TRUNK IS NOT



WUI-US WIG t+2. COSO+Mgt 1-2 SINO=WUZ

(55,56 m)-Us(9.81 3)/125)(005286)+4.81 2)(125)(1125)(11286)=25 m

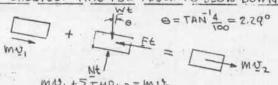
45= 55.56-25.0+(9.81)(12)(6IN 2.86°) = 0.310 (9.81)(12)(co52.86°)

GIVEN:

TRUCK SLOWS FROM 60 milh To zomil DOWN A 4% GRADE WITH ITS WHEELS JUST ABOUT TO SLIDE US=0,60

FIND:

SHORTEST TIME FOR TRUCK TO DLOW DOWN



MU, +ZIMP1-2= MUZ MU, +WtS'NO-Ft=MU2

N= 60 mc/h = 88 ft/s N=WCOSO W=MQ 0= 20 mc/n= 29.33 ft/s F= 45 N=45 WCOS 0

boxx88ft)+1m/322ft)(t)(sin229°)-(0.60)(m/(322ft)(cos2.29°)(t)

13.128



GIVEN:

INITIAL BOAT SPEED= U.= 8 .nu/h. BOAT SPEED ID SEC AFTER CIINNAKER IS WAISED= 15=12 milh. W=980 16

FIND:

NET FORCE PROVIDED EY THE SPINNAKER OVER THE ID SEC. IN TER AL

V. = 8 mc/n = 11.73 ft/s

t=10 SEC

V2= 12 mc/h= 17.60 ft/s

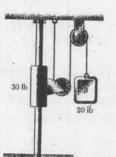
M.U, + IMP = MUZ

m (11.73 ft/s)+ F, (10 s) = m (11.60 ft/s) $F_N = \frac{(18016)(17.60ft/s-11.73ft/s)}{(32.2 ft/s^2)(105)} = 17.8616$

NOTE:

FN IS THE NET FORCE PROVIDED BY THE SAILS. THE FORCE ON THE SAILS IS ACTUALLY GREATER AND INCCUDES THE FORCE NELSED TO OVERCOME THE WATER RESISTANCE ON THE HULL.

13.129



GIVEN:

SYSTEM RELEASED FROM

FIND:

TIME FOR ATO REACH A VELOCITY OF 2 ft/s

KINEMATICS

LENGTH OF CABLE ! ONSTANT

L= =XA+XB dL = 2NA+NB= 0

NB= -2 VA

 $A \left[A \left(W_{A} V_{A} \right)_{z} \left(V_{A} \right)_{z} = 2 \text{ ft/s} \right]$

(2T)t, (MUA), +(2T)(t,2)-Wat,2= M(UA)2 0+(2T-30) $t_{1-2}=\left(\frac{30}{9}\right)(2)$

| multiple (T-15) t1-2= 30 (1)

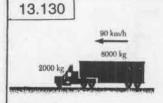
8 $| w_8 \mathcal{N}_8 | = 0$ $| w_8 = \frac{3}{3} = \frac{3}{3}$ (UB)= 2(UA)= 4 ft/5 1

+ (mova) -T(t+2)+WB(t+2)=(MBVB), $B | (M_B N_B)_2 > + (20 - T)(t_{1-2}) = \frac{20}{9} (4)$ (2)

ADD EQ. (1) AND (2) (ELIMINATING T)

(20-15)(t1-2)=(30+80)=110 t1-2= 22 = 0.6835

t=0.683 sec.



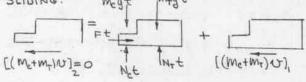
Mc=2000 kg
MT=8000 kg
INITIAL U=90 km/n
FINAL U=0
TRAILER BRAKES FAIL
45=0.65

FIND:

(b) FORCE ON THE COUPLING DURING THIS TIME

U= 90 km/h = 25 m/s

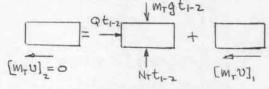
(a) THE SHORTEST TIME FOR THE EIG TO COHE TO A STUP WILL BE WHEN THE FRICTION FORCE ON THE WHEELS IS MAXIMUM. THE DOWNWARD FORCE EXERTED BY THE TRAILER ON THE CAB IS ASSUMED TO BE ZERO. SINCE THE TRAILER BRAKES FAIL ALL OF THE BRAKING FORCE IS SUPPLIED BY THE WHEELS OF THE CAB, WHICH IS MAXIMUM WHEN THE WHEELS OF THE CAB ARE AT IMPENDING SLIDING. W. 14.



$$\frac{1}{(m_c + m_f v)_2} = -Ft + [(m_c + m_f)v],$$

$$0 = -(0.65)(2000 kg)(9.81 m/s^2)(t_{1-2}) = (10000 kg)(25 mg)$$

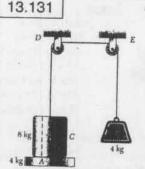
(b) FOR THE TRAILER



$$(m^{\perp}\alpha)^{3} = -6 + 1 - 5 + [m^{\perp}\alpha]^{1}$$

$$0 = -\phi(19.60 \text{ s}) + (8000 \text{kg})(25 \text{m/s})$$

Q= 10204 N



GIVEN:

MA = 4 kg MB = 4 kg MC = 8 kg SYSTEM IS RELEASED FROM REST

EIND:

(a) VELOCITY OF

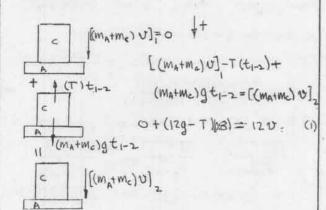
BLOCK B AFTER

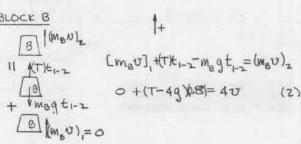
8 SEC.

(b) FORCE EXERTED

BY CON A

(a) BLOCKS A AND C





ADDING (1) AND (2), (ELIMINATING T)

$$(12g-4g)(0.8) = (12+4)U$$

$$U = \frac{(8 \log)(9.81 \text{ m/s}^2)(0.85)}{16 \log } = 3.92 \text{ m}$$

$$U_B = 3.92 \text{ m}$$

(b) COLLAR A ++

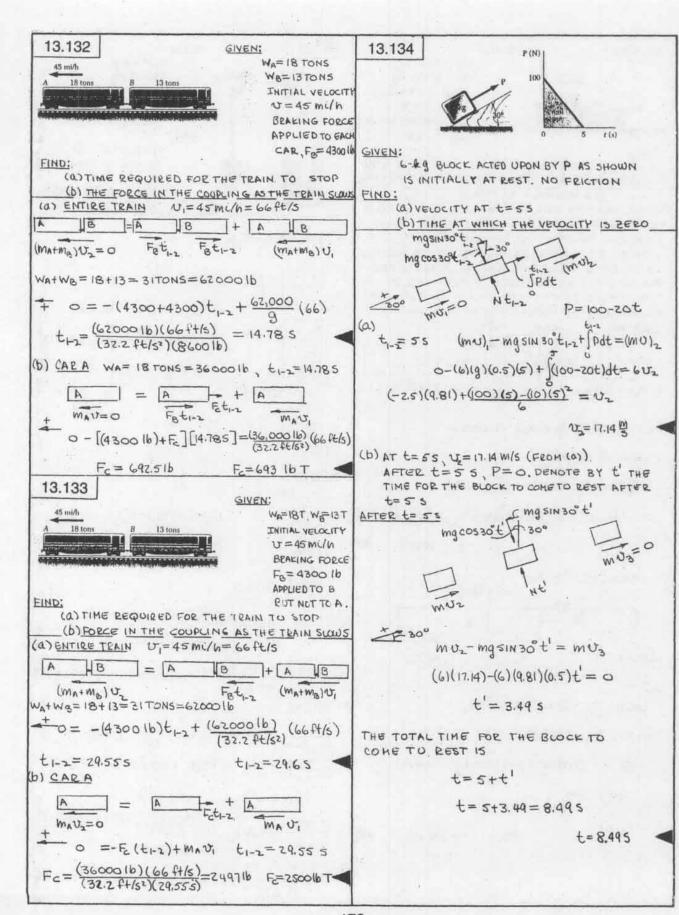
$$t^{(M_AU)} = 0$$
 o $t^{(F_c + M_A g - T)} t_{1-2} = (M_AU)_2(3)$
 $t^{(K_c)} = t^{(K_c)} = 0$
 $t^{(M_AU)} = 0$
 $t^{(M$

T= (4Ag)(3.92 M) +(4 kg)(9.81 M)

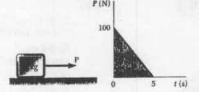
(0.85)

T= 58.84 N

SOLVING FOR FO IN (3) FC=(4 kg)(3.92 kg)-(4 kg)(9.81 kg2)+58.84N=39.2N







A 6-kg BLOCK IS ACTED UPON BY THE FORCE P AS SHOWN AND IS INITIALLY AT EEST. COEFFICIENTS OF FRICTION, US=0.60 Mp=0.45

EIND:

(a) VELOCITY OF THE BLOCK AT t=5 S (b) HAXINUM YELOCITY OF THE BLOCK

CHECK TO SEE IF THE BLOCK MOVES WHEN P

$$F_s = 45N$$
 $P = M_5 M_0$
 $P = M_5 M_0$
 $P = 35.3 N$
 $P = 35.3 N$

SINCE 35.3 N IS LESS THAN THE INITIAL VALUE OF P=100 N, THE BLOCK HOVES.

 $P = 100 - 20t \quad t_{1-2} = 55 \quad F = H_{E} mg = (0.45)(6)(g)$ $m v_{1} = \int_{0}^{t_{1-2}} P dt - F t_{1-2} + m v_{2}$ $0 = \int_{0}^{t} (100 - 20t) dt - (0.45)(6)(9.81)(5) = 6.v_{2}$

0= 500-250-132.4 +6 Uz

(b) DETERMINE TIME AT WHICH THE VELOCITY IS A HAXIMUM, WHICH HUST OCCUP AT tess

t= 3.68 s WHEN I IS MAXIMUM

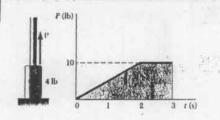
SUBSTITUTE t=3.685 IN EQ.(1)

0=(100)(3.68)-10(3.68)-97.47+6UHAX

 $v = \frac{134.67}{6} = 22.45 \text{ m/s}$

UHAX=22.5 M/S

13.136



GIVEN:

A 4-16 BLOCK IS ACTED UPON BY THE FORCE P AS SHOWN AND IS INITIALLY AT REST NO FRICTION

FINC:

(a) YELOCITY AT t= 25 (b) VELOCITY AT t= 35

THE BLOCK DOES NOT MOVE UNTIL P=41b FROM t=0 TO t= 25 P= 5t THUS, THE BLOCK STARTS TO MOVE WHEN t=4/5=0.85

$$| M v_2 \qquad (a) \text{ FOR } 0 < t < 2 \text{ S}$$

$$P = \text{St}$$

$$| \int P dt \qquad | + \quad t_1 = 0.85 \quad t_2 = 2 \text{ S}, v_1 = 0$$

$$| \int P dt \qquad | + \quad \int P dt - w(t_2 - t_1) = m v_2$$

$$| \int W (t_2 - t_1) \qquad t_1$$

$$| \int W v_1 \qquad 0 + \int S t dt - 4(2 - 0.8) = \frac{4}{9} v_2$$

$$| \int W v_1 \qquad 0 + \int S t dt - 4(2 - 0.8) = \frac{4}{9} v_2$$

 $\sqrt{2} = \frac{(32.2 + \frac{1}{52})}{4(16)} \left[\left(\frac{5}{2} + \frac{16}{5} \right) \left[(25)^{2} + (0.8 + 5)^{2} \right] - (416)(25 + 0.85) \right]$

U2= 28.98 ft/s

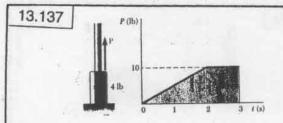
N2=29.0 ft/s

(b) FROM t=25 TO t=35

(418)

 $U_3 = 29.0 + 48.3 = 77.3$ ft/s

U3= 77.3 At/s

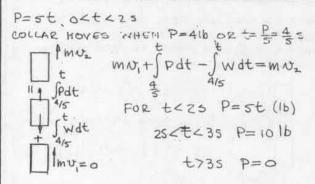


A FORCE P(16) AS SHOWN. NO FRICTION

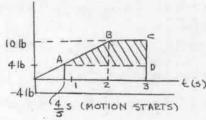
FIND:

(a) THE HAXIMUM VELOCITY OF THE COLLAR, UMAX (b) THE TIME WHEN THE VELOCITY IS ZERO.

(a) DETERMINE TIME AT WHICH COLLAR STARTS TO HOVE



FOR t<35 W=41b
THE HATIMUM VELOCITY OCCURS WHEN THE
TOTAL IMPULSE IS MAXIMUM.



 $AREA_{ABCD} = HAX IMPULSE = \frac{1}{2}(616)(\frac{6}{5}5)+(616)(15)$

AREA_{ASCO} = 9.6 lb.s

$$0 + 9.6 \text{ lb.s} = \frac{(4 \text{ lb})}{(32.2 \text{ ft/s2})} N_{\text{MAX}}$$

(b) VELOCITY IS ZERO WHEN TOTAL IMPULSE
15 ZERO AT THAT

FOR 45<+<35, IMPULSE=9.6(16.5), PART (a)

FOR At BEYOND 35 IMPULSE = - 4 At (16.5)

TOTAL IMPULSE = 0 = 9.6-4 At At= 245

TIME TO ZERO VELOCITY t= 35+2.45=5.45

13.138

p (MPa)
pa

GIYEN:

20-9 BULLET
10 MM DIAMETER RIFLE
BARREL
EXIT VELOCITY OF THE
BULLET = 700 M/S
TIME BULLET TO EXIT
= 1.6 MS
VARIATION OF PRESSURE

AS SHOWN

FIND:

P(MA) - P= C-Czt AT t=0 p=-po=C1-C2(p) C= -DO AT t= 1.6x103s 10=0 -t(ms) 0=C1-C2(1.6×1035) 1.6×1035 C2= P0/(1.6x1035) A Podt m= 20×10 kg 1.6×1035 $A = TT(10^{-3})$ O+A pdt= mu A= 78.54 X10 m2 1.6×10 3 (C,-c2t) dt= 20x103 (78.54×10 6 M2) (C,)(1.6×103)-(12)(1.6×103)]= (20x103kg)(100 m/s) 1.6×103C1-1.280×10 C2= 178.25×10 (1.6×103 m2.5) Po-(1.280×106 m2.52) Po= 178.25×10 kg/m -Po= 222.8 × 106 N/M2

13.139 GIVEN:

25-9 BULLET, IOMM DIA. PIFLE BARREL EXIT VELOCITY = 520 m/s
TIME FOR BULLETTO EXIT= 1.44 ms
PLESSURE HODEL -t (0.16 ms)
p(t)=(950 HPa)(e

Po=223 MPa

"/ ERROR IF GIVEN EQUATION FOR PLE) IS USED TO CALCULATE THE EXIT VELOCITY



INITIAL VELOCITY AT TAKEOFF = 10 m/s. VELOCITY AFTER TAKEOFF = 12 M/S AT 50 IMPACT TIME

FIND:

= 0.185. VERTICAL COMPONENT OF THE AVERAGE IMPULSIVE FORCE ON ATHLETES FOOT FROM THE GROUND. (INTERMS OF HIS WEIGHT W)

Pv = 6.21W

13.141

Landing pit

GIVEN:

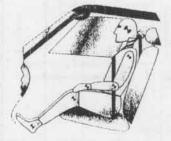
VELOCITY BEFORE LANDING = 30ft AT 350 IMPACT TIME BEFORE COHING TO A STOP = 0.22 5 WEIGHT=18516

FIND:

HORIZONT AL COMPONENT OF THE AVERAGE IMPULS 'VE FORCE ON THE ATHLETES FEET

Pu=64216

13.142



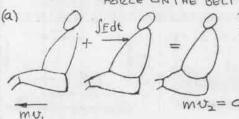
GIVEN:

AUTOMOBILE TRAVELING AT 45 MU/h CONES TO A STOP IN 110 MS. FORCE ACTING ON MAN AS SHOWN MANS WEIGHT = 200 lb

EIND:

(a) AVERAGE IMPULSIVE FORCE EXERTED ON THE BELT AS SHOWN

(b) HAXIMUM FORCE FM EXERTED ON THE BELT FORCE ON THE BELT IS OPPOSITE



DIRECTION SHOWN

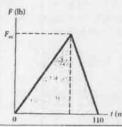
mu= 0

N= 45 mi/h= 66 ft/s W= 200 b

MU, -) Edt = MU2 | Fdt = FAVE At

(2001b) (66A/S) - FAVE (01105)=0 At= 0.1105

FAUE = (200)(66) = 3727 lb FAVE = 3730 lb (b)



IMPULSE = AREA UNDER F-t DIAGRAM= 1 Fm (.110'5)

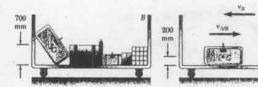
FROM (a), IMPULSE= FAVE Dt= (3727 16) (1105)

+ Fm (.110)=(3727)(.110) Fm= 7450 lb

13.143 GIVEN:

1.602. GOLF BALL HAS A VELOCITY OF 125 ft/s AFTER IMPACT DURATION OF IMPACT = to = 0.5 ms FORCE DURING IMPACT F= F_SIN(11+/4)

FIND: HAXIMUM FORCE FW. ON THE BALL 0=125 ft/s MUZ MU,=0 0+) Fm sin 1 tdt = (1.6/6) (125) MU, + Fdt=MU2 Fm= 122016 0.5%10-3



GIVEN:

15 kg SUITCASE A 40-kg LUGGAGE CARRIER B INITIAL VELOCITY OF CAERIER, UB=0.8 M

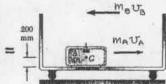
FIND:

(a) VA/B

- (b) UB, AFTER THE SUITCASE HITS THE RIGHT SIDE OF THE CARRIER WITHOUT PCESSUILD
- (C) ENERGY LOST BY THE IMPACT OF THE SUITCASE ON THE FLOOR OF THE CARPIER

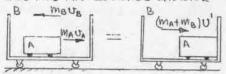
(a) SINCE THERE ARE NO EXTERNAL FORCES ACTING ON THE SYSTEM OF THE SUITCASE A AND THE LUGGAGE CARRIER E. IN THE HORIZONTAL DIRECTION, LINEAR MOMENTUM IS CONSERVED





N = 0 $N_B = -0.8 \, \text{m/s}$ $N_A = V_{A/B} + V_B$ $N_B = 40 \, \text{kg}$ $N_A = 15 \, \text{kg}$ $N_B = 40 \, \text{kg}$ $N_A = 15 \, \text{kg}$ $N_A/B = \frac{(40 \, \text{kg})(0.8 \, \text{m/s})}{(15 \, \text{kg})} + 0.8 \, \text{m/s} = 2.93 \, \text{m/s}$

(b) HOMENTUM IS CONSERVED BEFORE AND AFTER
THE SUITCASE HITS THE LUGGAGE CAPRIER



- MAUA +MBUB = (MA+MB)UI

U = MAUA+MBUB
(MA+MB)

FROM (a)

 $U_A = U_{A/B} + U_B = 2.43 - 0.8 = 2.13 \text{ m/s}$ U' = (15)(2.13) - (40)(0.8) = 0 U' = 0

(C) BEFORE SUITCASE FALLS, E= MAG (.7M)
AFTER SUITCASE HITS THE BOTTOM OF THE
CARRIER E2= 1 MA VA2+ 1 MB VB2+MAG (0.200M)

ENERGY LOST, $\Delta E_{1} = E_{1} - E_{2}$ $E_{1} = 15 g (.7)$ $\Delta E_{1} = (15)(9.81)(0.7) - \frac{1}{2}(15)(2.13)^{2} - \frac{1}{2}(40)(0.8)^{2} - (15)(9.81)(0.7)$ $\Delta E_{1} = 26.7 \text{ J}$ 13.145



GIVEN:

EFFORE COUPLING, ZO-MY CAR IS TRAVELING AT 14 km, AS SHOW!!

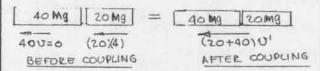
40-MY CAR HAS ITS WHEELS LOCKED

4k=0.30, 10-MY CAR ONLY

FIND:

- (A) VELOCITY OF BOTH CARS IMMEDIATELY
 AFTER COUPLING
 - (b) THE TIME FOR BOTH CARS TO COME

(a) THE MOMENTUM OF THE SYSTEM CONSISTING OF THE TWO CARS IS CONSERVED IMMEDIATELY BEFORE AND AFTER COUPLING.



+ $\sum mv = \sum mv'$ 0 + (20 Mg)(4 km/h) = (20 Mg + 40 Mg)(v') $v = \frac{(20)(4)}{(20)(40)} = 1.333 \text{ km/h}$

(b) AFTER COUPLING

$$\frac{60 \text{ Mg}}{6005=0} = \frac{60 \text{ Mg}}{5 \text{ F}_{\text{t}} \text{dt}} + \frac{60 \text{ Mg}}{600}$$

THE FRICTION FORCE ACTS ONLY ON THE 40 Mg CAR SINCE ITS WHEELS ARE LOCKED THUS,

 $F_f = \mathcal{U}_K N_{40} = (0.30) (40 \times 10^3 \text{kg}) (9.81 \frac{\text{m}}{\text{Sz}})$ $F_c = 117.72 \times 10^3 \text{ N}$

FROM (a) U=v=1.333 km/h=0.3704 m/s

IMPULSE MOHENTUM,

 $2mv_1 + \int_0^\infty F_4 dt = 2mv_2$ $(60\times10^3 kg)(.3704 \text{ m/s}) - \int_0^\infty (17.72\times10^3 \text{ N}) dt = 0$ $t = \frac{(60\times10^3)(.3704)}{(117.72\times10^3)} = 0.18885$





WA= 190 16

WB= 12516

PAFT WR= 300 16

VA/R= 2 FE/S

TOWARD B, AFTER

THE RAFT BREAKS

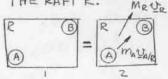
LOOSE FROM ITS

ANCHOR.

FIND:

(a) SPEED OF THE RAFT, UR, IF B DOES NOT HOVE (b) SPEED UB OF B. IF THE RAFT IS NOT TO HOVE

(a) THE SYSTEM CONSISTS OF A AND B AND THE RAFT R. MOTE



MOHENTUM IS CONSERVED

$$\begin{array}{cccc} \dot{\mathcal{V}}_{A} = \dot{\mathcal{V}}_{A/Q} + \dot{\mathcal{V}}_{R} & \dot{\mathcal{V}}_{B} = \dot{\mathcal{V}}_{B/Q} + \dot{\mathcal{V}}_{R} & \dot{\mathcal{V}}_{B/Q} = 0 \\ \dot{\mathcal{V}}_{A} = 2 ft/s + \dot{\mathcal{V}}_{R} & \dot{\mathcal{V}}_{B} = \dot{\mathcal{V}}_{R} \\ & o = m_{A} \left[2 f + \dot{\mathcal{V}}_{R} \right] + m_{B} \dot{\mathcal{V}}_{R} + m_{Q} \dot{\mathcal{V}}_{Q} \\ \dot{\mathcal{V}}_{R} = \frac{-2 m_{A}}{(m_{A} + m_{B} + m_{Q})} = \frac{-(2 ft/s)(|q_{O}|b)}{(|q_{O}|b + |2 s|b + 300|b)} \end{array}$$

UR=0.618 ft/s

(b) FROM EQ (1)

$$\begin{array}{ll}
0 = M_{A}N_{A} + M_{B}U_{B} + 0 & (U_{R} = 0) \\
N_{B} = -\frac{M_{A}N_{A}}{M_{B}} & N_{A} = N_{A/R} + N_{R} = z \text{ ft/s} \\
N_{B} = -\frac{(z \text{ ft/s})(1901b)}{(1251b)} = 3.04 \text{ ft/s}
\end{array}$$

UB= 3.04 ft/s

13.147

GIVEN:

MA=1500 kg
MB=1200 kg
BOTH CARS
TOGETHER, SKID
AT 10° NORTH OF
EAST AFTER
IMPACT

FIND:

(Q) WHO WAS

GOING FASTER

(b) SPEED OF

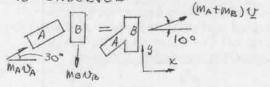
THE FASTER

CARL IF SLOWER

CAR WAS GOING

AT 50 &m/b

(a) TOTAL MOMENTUM OF THE TWO CARS



ZMU, X: MAUACOS 30 = (MA+MA NUTCOS 10 (1)

ZMU, y: MANA SIN 30° - MBUB= (MAHMB) USIN10° (Z)
DIVIDING (1) INTO (Z)

$$\frac{V_B}{V_A} = \frac{(T_A N_3 0^\circ - T_A N_1 0^\circ) (M_A \cos 30^\circ)}{M_B}$$

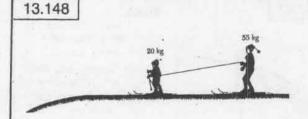
$$\frac{v_B}{v_A} = (0.4010) (1500) (0530^\circ)$$

$$\frac{N_B}{N_A} = 0.434$$
 $N_A = 2.30 \, N_B$

THUS, A WAS GOING FASTER

(b) Since UB was the Scower CAR

UA = (2,30)(50) = 115, 2 km/h



NOTHER AND CHILD TRAVELING AT 7.2 km/h
INITIALLY. Mm=55 kg Mc=20 kg
CHILD'S SPEED DECREASES TO 3.6 km/h
IN 3 5 AS THE MOTHER POLLS ON THE ROPE

FIND:

(a) MOTHERS SPEED AT THE END OF THE . 3 S INTERVAL

(b) Average value of the tension in THE DOPE DURING THE 35 INTERVAL

(a) CONSIDER HOTHER AND CHILD AS A SINGLE SYSTEM. ASSUMING THE FRICTION FORCE ON THE JKIS IS NEGLIGIBLE MOMENTUM IS CONSERVED

$$(m_e v_e) \qquad (m_H v_H) \qquad (m_e v_e') \qquad (m_H v_H')$$

$$m_e v_e + m_H v_H = m_e v_e' + m_H v_H'$$

Vc=10H=7.2km/h Nc=3.6 lem/h

(20)(7.2)+(55)(7.2)=20(3.6)+(55)(U/)

V'n=8.51 km/h

(b) CHILD ALONE

$$C + C = C$$

$$m_c v_c$$

$$m_c v_c$$

$$+=35$$

McVe - Faxt = McVe

N= 7.2 km/h= 2 m/s N= 3.6 km/h=1 m/s

$$(zolg)(zm/s) - F_{AV}(3s) = (zolg)(1m/s)$$

 $F_{AV} = \frac{(zolg)(1m/s)}{(3s)} = 6.67 lg.m/s^2$

FAV = 6.67 N

L V₀

13.149

GIVEN:

A AND B ON A HORIZONTAL FRICTION LESS PLANE ARE ATTACHED BY AN INEXTENSIBLE CORD OF LENGTH L MASS OF A = MASS OF B NB = Vo, VA = O INITIALLY

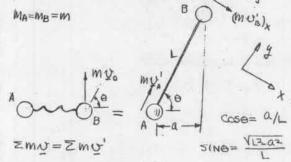
FIND:

(a) Na AND UB AFTER THE CORD BECOMES TAUT (b) THE ENERGY LOST AS

(a) FOR THE SYSTEM CONSISTING OF BOTH

BALLS CONNECTED BY A CORD THE TOTAL

HOMENTUM IS CONSERVED / (MU.)



 $X: -mN_0 \cos \theta = m(v_B^*)_X$ (1)

$$(U_B)_X = - V_0 \cos \theta = -U_0 \frac{\alpha}{L}$$

y: $m v_0 \sin \Theta = m v_A + m(v_8) y$ (2)

THUS FROM (2) UO SINO = 2VA

FROM (3)
$$(v_{B}^{i})_{y} = v_{A}^{i} = (N_{0}/2L)\sqrt{L^{2} - \Omega^{2}}$$

$$(v_{B}^{i})_{y} = v_{A}^{i} = (N_{0}/2L)\sqrt{L^{2} - \Omega^{2}}$$

$$v_{B}^{i} = \sqrt{(v_{B}^{i})_{X}^{2} + (v_{B}^{i})_{Y}^{2}} = v_{e}\sqrt{\frac{n^{2}}{L^{2}} + \frac{(L^{2} - \Omega^{2})}{4L^{2}}}$$

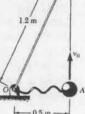
$$v_{B}^{i} = (v_{0}/2L)\sqrt{L^{2} + 3\Omega^{2}}$$

(b) INITIAL $T = \frac{1}{2} m \alpha J_0^2$ $T' = \frac{1}{2} m (U_A)^2 + \frac{1}{2} m (U_B)^2 = \frac{1}{2} m (U_0 / 2L)^2 [(L^2 a^2) + (L^2 3d)]$ $T' = \frac{1}{2} (m U_0^2 / 4L^2) (2L^2 + 2a^2) = (m U_0^2 / 4L^2) (L^2 + a^2)$ $\Delta T = T - T' = \frac{1}{2} m U_0^2 - (m U_0^2 / 4L^2) (L^2 + a^2)$

ΔT=(mvo/412)(L2-Q2)



2-kg SPHERE CONNECTED BY AN INEXTENSIBLE CORD OF LENGTH 1.2M TO POINT O ON A HORIZONTAL FRICTIONLESS PLANE INITIAL VELOCITY Vo PEPPENDICULAR TO OA



FIND:

MAXIMUM ALLOWABLE Vo IF IMPULSE OF THE FORCE EXERTED ON THE CORD IS NOT TO EXCEED 3N.S.

FOR THE SPHERE AT A IMMEDIATELY BEFORE 1 :: L AFTER THE CORD BECOMES TAUT

MNO + FAt = MYA

MUSTING-FLE=0 FAt= 3 N.S W.= Z- ka 3 N.S (2ka XSIN 65.38°)

Un= 1.650 m/s

13.151

GIVEN:

MASSES ALL INITIAL VELOCITY OF THE BALL AS SHOWN, NO ENERGY LOST IN THE IMPACT



EIND:

(a) VELOCITY OF THE BALL IMMEDIATELY AFTER IMPACT (b) IMPULSE OF THE FORCE EXECTED BY THE PLATE ON THE BALL

(a) FOR THE SYSTEM WHICH IS THE EALL AND THE PLATE MOMENTUM IS CONSERVED

ALL FORCES ARE NON-IMPULSIVE EXCEPT THE EQUAL AND OPPOSITE FORCES BETWEEN THE PLATE AND THE BALL

+ (mu) = - (wu') + (mu') p

(1.125kg)(3m/s)=(-0.125kg)(vo)+(0.250kg)vp

$$\nabla \rho' = 0.5 \, \mathcal{N}_{0}^{1} + 1.5$$
(1)

SINCE THERE IS NO ENERGY LOST THE KINETIC ENERGY OF THE SYSTEM IS CONERVED

(CONTINUED)

13.151 continued

BEFORE IMPACT, T= 2 MBUB= 12 (0.125 Rg) (3 M/s)=0.5637

AFTER IMPACT T= 1 MB(UB) + 1 Mp(Up)2 SUBSTITUTE FOR US FROM (1)

T=2(0.125kg)(v;)2+(2)(0.250kg)[0.5v;+1.5]

T= 0.09375(00)2+0.1875UB+ 0.2813

T=T' 0.563=0.09375(v')2+0.1675 vg+ 0.2813

$$U_{B}^{12} + ZU_{B}^{1} - 3 = 0$$
 $U_{B}^{1} = -2 + \sqrt{4+12} = -1 + 2 = -3 + 11$

(NB=-3 M/S BEFORE IMPACT) NB=1 M/S

(b) BALL ALONE

+ (0.125kg)(-3m/s)+FAt = (0.125kg)(1 m/s)

FAt=0.5 N.51

13.152

MAAA

HAAAF

MAM!

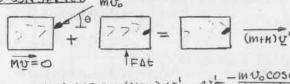
GIVEN:

BULLET FIRED INTO THE BLOCK AS 5HOWN.

FIND:

ANA JATHOSISOH VERTICAL COMPONENTS OF THE IMPULSE ON THE BULLET.

FOR THE SYSTEM WHICH IS THE BULLET AND THE BLOCK, MOHENTUM IN THE HODIZONTAL DIRECTION IS CONSERVED



U=-MU.cose -mvocose=(H+m)U'

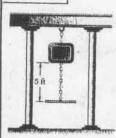
BULLET ALONE

> - muocoso + Px at = mu Px At = mNocose[1-Mm]

+1-mvosino+PyAt= 0

 $P_x \Delta t = \frac{m M}{m+M} N_0 \cos \theta$

Py At = MUSINO



GIVEN:

RIGID BEAM WEIGHS ZAO ID
BLOCK WEIGHS GO ID
INITIAL VELOCITY OF THE
BLOCK = O AND IT IS
DROPPED FROM 5 ft.

FIND:

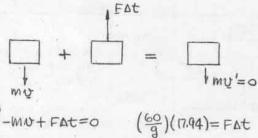
INITIAL IMPULSE EXERTED ON THE CHAIN AND THE ENERGY ABSOLBED BY THE CHAIN IF THE SUPPORTING COLUMNS ARE,

(a) PIGID, (b) EQUIVALE NT TO TWO ELASTIC SPRINGS

VELOCITY OF THE BLOCK JUST BEFORE IN PACT $T_1=0$ $V_1=Wh=(601b)(5ft)=3001b.ft$ $T_2=\frac{1}{2}mv^2$ $V_2=0$ $T_1+V_1=T_2+V_2$ $0+300=\frac{1}{2}(\frac{60}{9})v^2$

 $U = \sqrt{(600)(32.2)/60} = 17.94 \text{ ft/s}$

(a) PIGID COLUMNS



FAT= 33.43 lb.5 \uparrow ON THE BLOCK

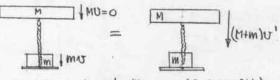
ALL OF THE KINETIC ENERGY OF THE BLOCK IS ABSORBED BY THE CHAIN

 $T = \frac{1}{2} \left(\frac{60}{9} \right) (17.94)^2 = 300 \text{ ft.lb}$

E=300 ft.16

(b) ELASTIC COLUMNS

MOMENTUM OF SYSTEM OF BLOCK AND BEAM IS CONSERVED



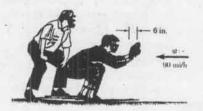
mv = (H+m)v' $v' = \frac{m}{(M+M)}v = \frac{60}{300}(17.94 \text{ ft/s})$ v' = 3.59 ft/s

REFERRING TO FIGURE IN PART (a)

-mu+Fat=-mu!Fat=m(u-u!)=(601g)(17.94-3.59)=26716.5

E= 1 mu2-1 mv2-1 mv2 60 [(17.44)2-(3.59)]-240 (3.59)2 E=240 [416

13.154



GIVEN:

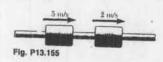
WB=5-02
INITIAL SPEED
OF THE BALL
= 90 Mi/h
AYERAGE
SPEED OF
THE GLOUE
DURING IMPACT
= 30 FE/S
OUER A 6 IN.
DISTANCE

FIND:

AVERAGE IMPULSIVE FORCE EXERTED ON THE PLAYERS HAND

 $F_{AV} = \frac{mv}{t} = \frac{(5/16 \text{ lb})(132 \text{ ft/s})}{(32.2 \text{ ft/s}^2)(1/60 \text{ s})} = 76.9 \text{ lb}$

13.155



GIVEN:

IDENTICAL COLLARS
WITH VELOCITIES
AS SHOWN.
C=0.65, M=1.2 kg
NO FRICTION

FIND:

(D) UA' AND U'B
AFTER IMPACT
(D) ENERGY LOST
DURING IMPACT

(a) TOTAL MOMENTUM IS CONSERVED

+ m ua + m ub = m ua+ m ub | + (+5 m/s)+(+ zm/s) = ua+ ub |

TM/S = V'A+V'B (1)
ES ALONG LINE OF IMPACT

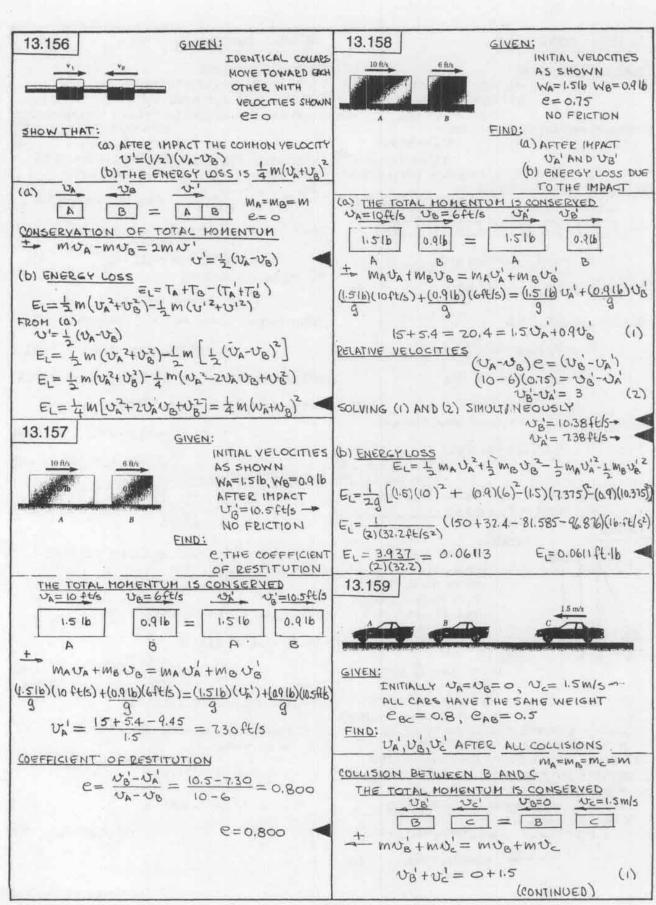
RELATIVE VELOCITIES ALONG LINE OF IMPACT $U_B' - U_A' = e(U_A - U_B)$ e = 0.65 $U_B' - U_A' = (0.65)(5 m/s - 2 m/s) = 1.95 m/s (2)$

ADDING (1) AND (2) 200 = 895 0 = 4.48 =

Va = 7m/s-4.48 m/s = 2.53 m/s-

(b) ENERGY LOST DURING IMPACT

 $E_L = T_A + T_B - T_A' - T_B'$ $E_L = \frac{1}{2} (1.2 \text{ kg}) [5^2 + 2^2 - (4.475)^2 - (2.525)^2]$ $E_L = 1.559 \text{ N/m}$



13.159 continued

RELATIVE VELOCITIES

$$\frac{(U_{B}-U_{c})(e_{Bc})=(U_{c}^{\prime}-U_{b}^{\prime})}{(-1.5)(0.8)=(U_{c}^{\prime}-U_{b}^{\prime})}$$

$$-1.2=U_{c}^{\prime}-U_{B}^{\prime}$$
(2)

SOLVING (1) AND (2) SINULTANEOUSLY

U=0.15 M/S+ SINCE UD > UC, CAR B COLLIDES WITH CAR A COLLISION BETWEEN A AND B

mus + mus = mus + mus

$$V_A^1 + V_B^{11} = 0 + 1.35 \tag{3}$$

RELATIVE YELDCITIES

$$U_{A}' - U_{B}'' = 0.675$$
 (4)

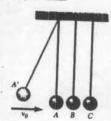
SOLVING (3) AND (4) SIMULTANEOUSLY

Un'= 1.013 m/s -UB'= 0.338 m/5

SINCE NEL NB C DA THERE ARE NO

FURTHER COLLISIONS

13.160



GIVEN:

SPHERES A, B, C OF EQUAL WEIGHT INITIAL VELOCITY OF A IS UO AND BANDC ARE AT REST. E IS THE SAME FOR ALL SPHERES

FIND:

(a) U' AND U'B AFTER THE FIRST COLLISION (b) U" AND U' AFTER

THE SECOND COLLISION

(C) FOR IN SPHERES, THE VELOCITY V' AFTER IT IS HIT FOR THE FIRST TIME

(d) USING THE RESULT FROM PART (C) THE

VELOCITY OF THE LAST SPHERE FOR 11=6 8=045

(a) FIRST COLLISION (BETWEEN A AND B) THE TOTAL HOMENTUM IS CONSERVED UN=UD UD=O

MUA+MUB=MUA+MUR

ひっ= ぴっ'+ひら

13.160 continued

RELATIVE VELOCITIES

Uge = UB-UA

SOLVING EQUATIONS (1) AND (2) SINULTANEOUSLY U/= U0(1-6)/5

UB=Vo(1+e)/2

(b) SECOND COLLISION (BETWEEN BAND C) THE TOTAL MOHENTUM IS CONSERVED NB' Uc=0 UB" Uc!

(B) (C) = (B) (C) MUB + MUC = MUB" + MUC

USING THE RESULT FROM (a) FOR UP

PELATIVE YELOCITIES

SUBSTITUTING AGAIN FOR NA FROM (a)

$$v_0(\underline{He})(e) = v_c' - v_B^{\parallel}$$
 (4)

SOLVING EQUATIONS (3) AND (4) SIMULTANEOUSLY

$$v_c' = v_o(1+e)^2/4$$

(C) FOR N SPHERES

N-1 TH COLLISION

0 0 = 0 0 h-1 N N-1 N

WE NOTE FROM THE ANSWER TO PART

(b), WITH
$$n=3$$

$$v'_{n} = v'_{3} = v'_{c} = v_{0}(1+e)^{2}/4$$
or $v'_{3} = v_{0}(1+e)^{(3-1)}/(2^{(3-1)})$

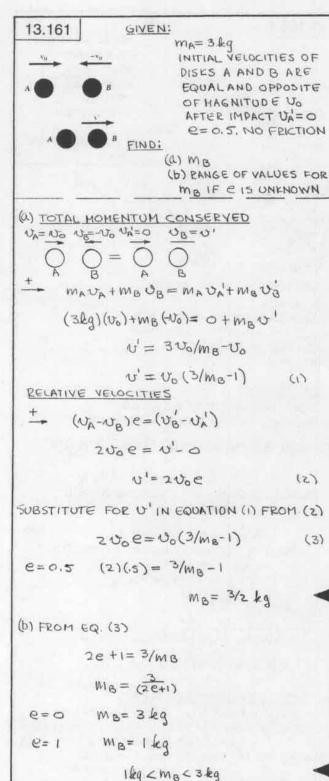
THUS FOR N BALLS

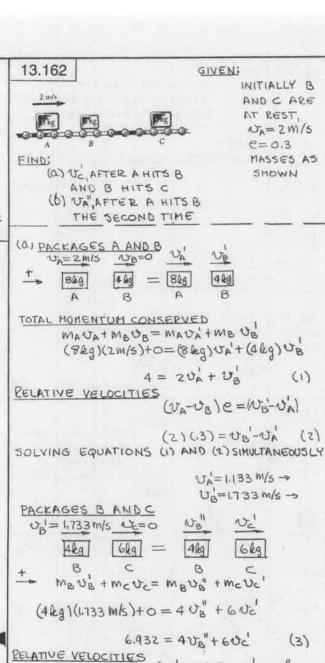
(d) FOR N= 6 AND @= 0.95

FROM THE ANSWER TO PART (C) WITH N=6

N'6= 0.881 Vo

N,=0.881Vo





10TAL HOHENTUH CONSERVED

(8)(1.133)+(4)(0.381)=804"+408"

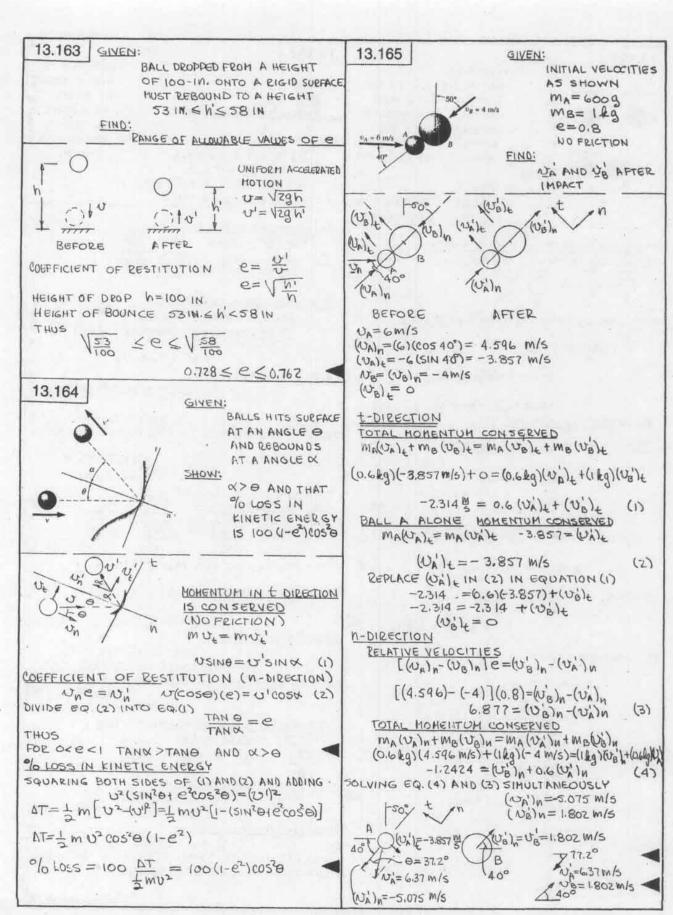
10.588=804"+408"

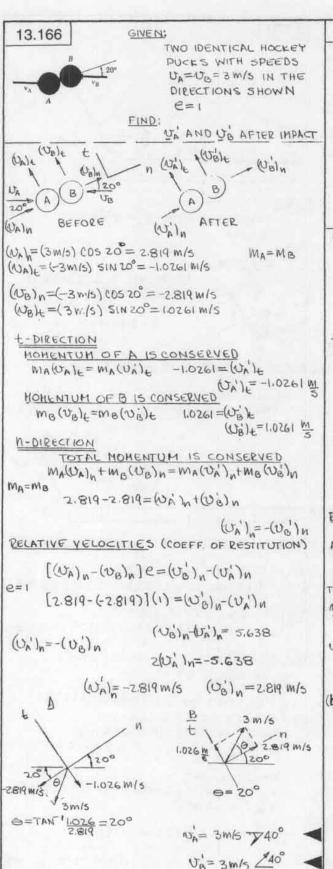
(5)

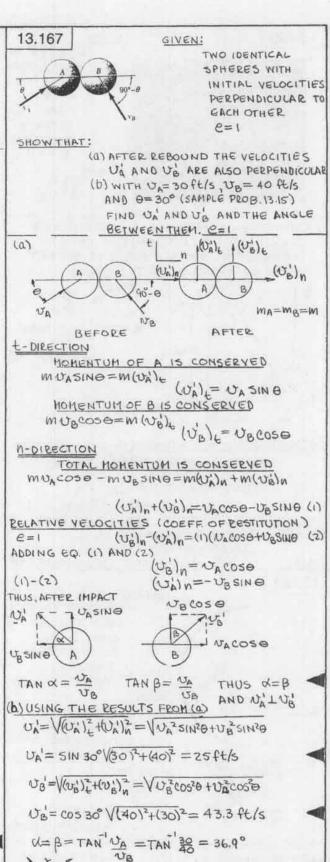
RELATIVE VELOCITIES

(UA'-UB") = UB"-UA" (1.133-0.381)(0.3) = 0.22 56 = UB"-UA" (6) SOLVING (5) AND (6) SIMULTANEDOSLY,

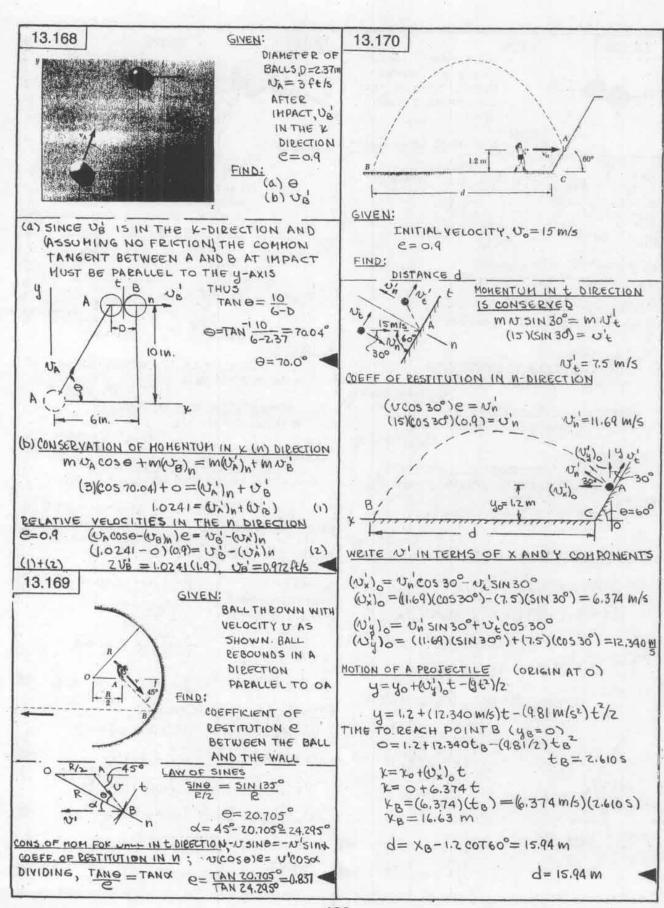
UA"=0.807W/5-

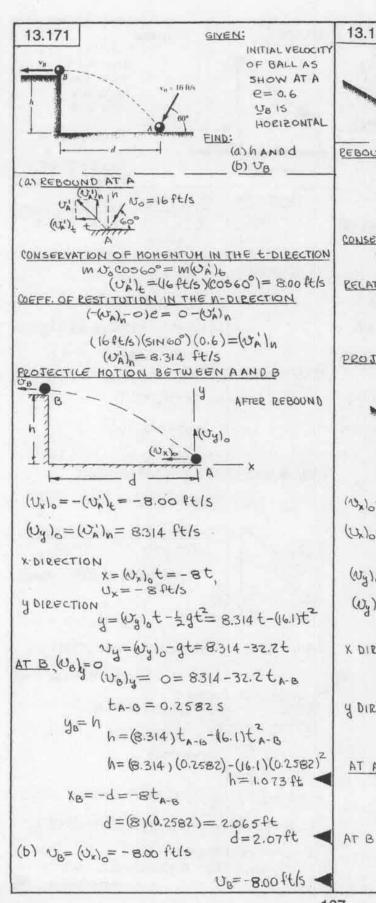


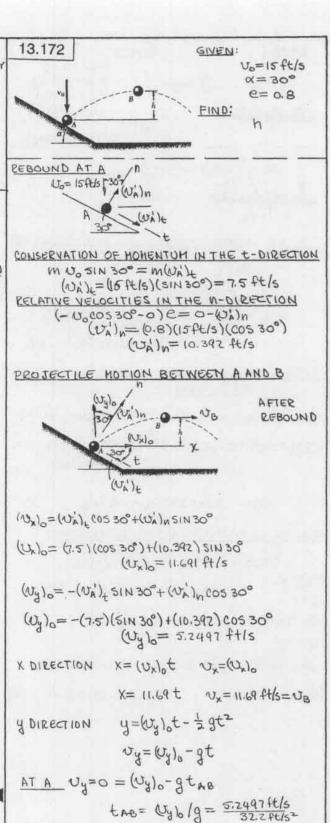




×90-β ×= 180-(x+90-β)] = 90°



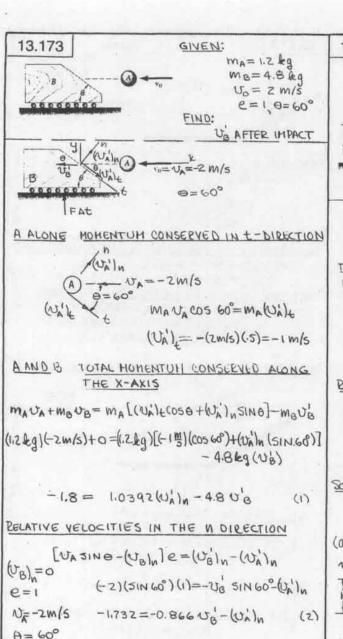




ta-B = 0.1630 5

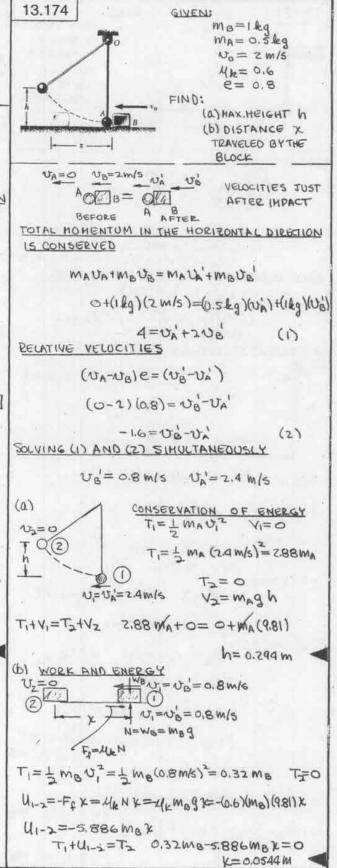
h=(5.2497)(.1630)-(16.1)(.1630)=0.428ft.

y= h=(vy) ot - 8- 9th-8



SOLVING (1) AND (2) SIMULTANEOUS LY

(U'A) n= 1.184 m/s UB=0.632 m/s-





ma=ma=1.5kg INITIALLY. U=5 m/s, U=0 NO FRICTION (1) e=1 (2) 0=0

FIND:

(a) MAXIMUM DEFLECTION OF THE SPRING

(b) FINAL VELOCITY OF BLOCK A UN= 5M/S UB UB UA

PHASE I IMPACT

CONSERVATION OF TOTAL MOMENTUM MAUA+MBUB=MANA'+MBUB MA=MB 5+0= VA+VA (1)

RELATIVE VELOCITIES (UA-UB) = (VG-VA)

> (5-0) e= UB-UAL (2)

(S) DING (I) AND (2)

5 (1+e) = Vo

SUBTRACTING (Z) FROM (1) 5(1-e) = NA

e=1 UB=5 m/s UA=0 e=0 UB=2.5 m/s UA=2.5 m/s

(a) CONSERVATION OF ENERGY PHASE II

CI

B

B

B

A

k=80 le= 80 H/m

TI= 1 MB(VB)2= 1 (1.5kg) (5m/s)= 18.75 J

V1=0 AT X= XMAX, T2=0; V2=== &(XMAX)= (40)(XMAX)2 T1+1=T2+12

18.75+0= 0+40 kur

XHAX = 0.685 M 0=1

E=2 BOTH A AND B HAVE THE SAME VELOCITY INITIALLY AT $0.05 \times 2.5 \text{ m/s}$ THUS $T_1 = 1 \times 10^{10} \text{ (Ma+Mb)} (0.3 \times 10^{10} \text{ (3.5 m/s)}^2 \text{ (3.5 m/s)}^2 \text{ (3.5 m/s)}^2$

T= 9.375 J V= 0 AT X=XHAX T2=0 V2=1 & XMAX = 40 KHAX

Tity = Taty2 9.375+0= 40x2 MAX

KHAX=0.484 M e=0

13,175 continued

(b) e=1, BLOCK B IS RETURNED TO POSITION (1) WITH A VELOCITY OF 5W/S - SINCE ENERGY IS CONSERVED, AND IMPACTS BLOCK A WHICH IS AT REST, IN THE IMPACT TOTAL HOHENTUH IS CONSERVED AND PHASE I IS REPEATED WITH THE VELOCITIES OF A AND B INTERCHANCED THUS NA"= 5M/S- AND NO "=0. SINCE THERE IS NO FRICTION THESE VELOCITIES ARE THE FINAL VELOCITIES OF A AND B

e=1 Na"=5m/5-

6=0 BLOCKS A AND B ARE RETURNED TO POSITION () WITH THE SAME YELOCITY OF 25 M/S - SINCE ENERGY IS CONSERVED. THERE IS NO ADDITIONAL IMPACT AND THE SPEING SLOWS BLOCK B DOWN AND A AND B SEPARATE WITH A CONTINUING WITH A VELOCITY OF 2.5 M/S TO THE RIGHT

UA 11-2.5W/-0=0

13.176



ma=mB=1.5 12 INITIALLY UA=5m/s, UB=0 Mb=0,3,45=0.5

FINAL POSITION OF (a) BLOCK A (b) BLOCK B

IMPACT SEE PHASE I OF SPOB 13,175, R=1 AFTER IMPACT N'= 0 N'= 5 m/s

PHASE II No=5m/s N=0 (2) F=4KN

HAXIMUM DEFLECTION XHAX OF THE SPEING

WORK AND ENEDGY T== mo(v'e)== 1 (1.5kg)(5m/s)= 18.75] T2=0 U1= 1-2xdx-J4Kmogdx U1-2=-1 (80) (XHAY)- (03) (1.5 kg) (9.81 M) XHAY

T1+U1-2=T2 18.75-40 KHAX -4.4145 KHAX = 0

XHAX 0.632 M C

PHASE III PETURN OF B TO POSITION (1) BEFORE IMPACT WITH A

T2=0, U2-1=+ 1 &(XMAX)2-4KMB9 KHAX

 $T_1 = \frac{1}{2} m_B O_B^{1/2} = (0.75) O_B^{1/2} \qquad (I_{2-1} = (40)(0.632)^2 - (4.4145)(0.632)^2$ U2-1= 15.977- 2.790= 13.173 0+13.173=(0.75) UB UB= 4.191 m/s-

AFTER IMPACT WITH A AT POSITION () RELATIVE VELOCITIES

(N"-U')(e) = (N-1); (4.191-0)(1)= UA"-UE

(1) (CONTINUED)

13.176 continued

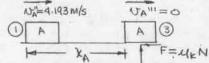
CONSERVATION OF TOTAL MOMENTUM AT (1) $+ + m_{A}U_{A}^{+} + m_{B}U_{B}^{+} = m_{A}U_{A}^{+} + m_{B}U_{B}^{+} \qquad m_{A} = m_{B}$ $0 + 4.191 = U_{A}^{+} + U_{B}^{+} \qquad (3)$

ADDING EQUATIONS (1) AND (2) 2(4.191) = 2UA"

Un = 4.191 m/s

FROM 50 (2) NB 4.191 - 4.191 = 0

(a) PHASE IX (VELOCITY OF B = O AT ())

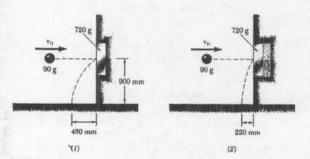


T=1 ma(v) = (0.75 kg) (4.191 m/s) N=MAG

FINAL POSITION OF A x=2.98→m

(b) NB = O AT IMPACT POINT AND THE SPRING IS UNDEFLECTED AT THIS POINT.

13.177

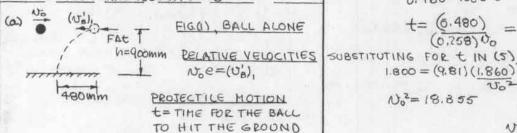


GIVEN:

BALL REBOUNDS AS SHOWN IN FIGURES (1) AND (2), FOAM RUBBER BEHIND PLATE IN (2)

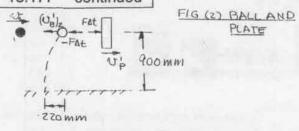
FIND:

(a) COEFFICIENT OF RESTITUTION e BETWEEN THE BALL AND THE PLATE (b) THE INITIAL VELOCITY US



0.480 m = voet





RELATIVE VELOCITIES

$$\begin{array}{c} + \\ & (U_B - U_P) e = U_P + (U_B)_2 \\ & U_B = U_0 \quad U_P = 0 \\ & U_0 e = U_P + (U_B)_2 \end{array}$$
 (2)

CONSERVATION OF HOHENTUM

$$\begin{array}{ll}
+ & m_B v_B + m_P v_P = m_B (-v_B')_z + m_P (v_P') \\
(0.09 \text{ kg}) (v_0) + o = (0.09 \text{ kg}) (-v_B')_z \cdot (0.720 \text{ kg}) v_0^2 \\
v_0 = (-v_B')_z + 8 v_P' \\
\end{array} (3)$$

SOLVING (2) AND (3) SINDLTAMEOUSLY FOR

$$(U_B)_2 = U_{\underline{(8C-1)}}$$

PROJECTI LE MOTION

$$0.220 \,\mathrm{m} = N_0 \left(\frac{8e-1}{4} \right) t$$
 (4)

DIVIDING EQ. (4) BY EQ. (3)

$$\frac{0.220}{0.480} = \frac{8e^{-1}}{9e}$$

$$4.125e = 8e-1$$

e=0.258

(b) FROM FIG (I)

PROJECTILE HOTION k= T a+5

0.900= (1.81) t 1.80=9.81t (5)

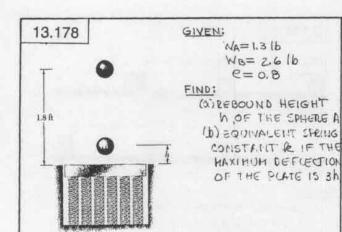
EQUATION (1)

$$t = \frac{(6.480)}{(0.758)} = \frac{1.860}{0.00}$$

1.800 = (9.81)(1.860)

No= 18.855

Vn= 4,34 m/s



(a) VELOCITY OF A AND B AFTER IMPACT

INITIAL VELOCITY OF A (BEFORE IMPACT) T,=0 V = (MAQ) (1.8 ft) (A) 0=U T2= 1 MAUA V2 V2= 0

T1+V1=T2+V2 ACAN == PAWB! +C 1) = (2) (1,8+t) (32.2 ft/s2) = (15.9 2 ft/s2 15A=10,77 ft./5 1

VELOCITIES AFTER IMPACT

CONSERVATION OF MOMENTUM MA (-UA) + MB (UB) = MNA + MB (-UB) 1.3 (-10,77)+0=1.3 U/ + 2.6 (-U6)

-10,77 = Un' - 7 Un KHATIVE VELOCITIES

-NA-NB (= (-NB 1)) (15.77) (0.8) = 12e+12h (2) CT. . IS C' AND CO', SIMULANEOUSLY

Nr. = 2.15 fils No = 6.46 ft/s

PERSOND HEIGHT OF A. (CONSERVATION OF ENERGY) NB = MB H T3 = 0 $T_2+V_2=T_3+V_3$ 5 MAUX 2+0=0+M3 h (Z) 1 (A) I

h== (2.15 ft/s) /32.2 ft/s2=0.0720 ft (b) | B -- (2) T2=1 MB 1/8 V2=0 NB=646 Als 3h=0.216ft T4=0 V4= V4= 1 k (3h)2

1 - 1 T2+12=T41/4 = (0,216) &=72.216/ft

13.179 GIVEN:

FIGURE AS SHOW IN 13.178 (LEFT) d1 8.5 = 8W, d1 E.1 = AW EQUIV SINGLE SPRING R=514

FIND:

(a) VALUE OF E FOR WHICH IS A MAXIMUM (b) CORRESPONDING VALUE OF h

(C) CORRESPONDING HAXIMUM DEPLECTION OF B (a) IMITIAL VELOCITY OF A (BEFORE IMPACT) FROM SOLUTION TO PROB. 13,178, Un= 10,77 ft/s V

UA=10.77 At/6 1 WA' B = 100=0

CONSERVATION OF HOMENTUM

+ MA (-UA) + MB UB = MA UA' + MB (-UB)

13 (-10.77) +0 = 13 0x +26 (-06) - 10,77 = UA - ZUa (1)

RELATIVE VELOCITIES

(-NA-UB) @= (-NB-UA) (2) 10,77 e= UB+UA

SOLVING (1) AND (2) SIMULTANEOUS LY FOR UA

3 NA=(10,77)(2e-1)

IN IS MAXIMUM WHEN UN' IS HAXIMUM, THAT IS WHEN C=1

(b) FOR C=1 Un= 10.77/3 = 3.59 ft/s

FOR A ALONE UN=0 CONSERVATION OF ENERGY

T2=0 V2=WAh U'= 3.59 Pt/s $T_1+Y_1=T_2+V_2$

> 1 Wa (U') +0 = 0+ WA h $h = \frac{1}{2} \frac{(3.59 \text{ ft/s})^2}{(32.2 \text{ ft/s}^2)} = 0.200 \text{ ft}$

(C) FOR B ALONE

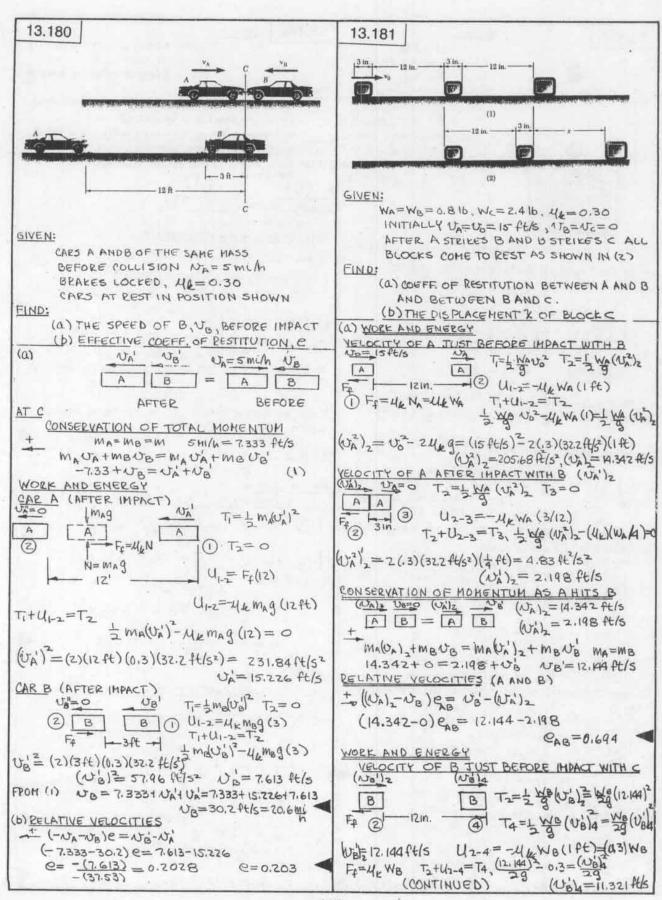
CONSERVATION OF ENERGY UB

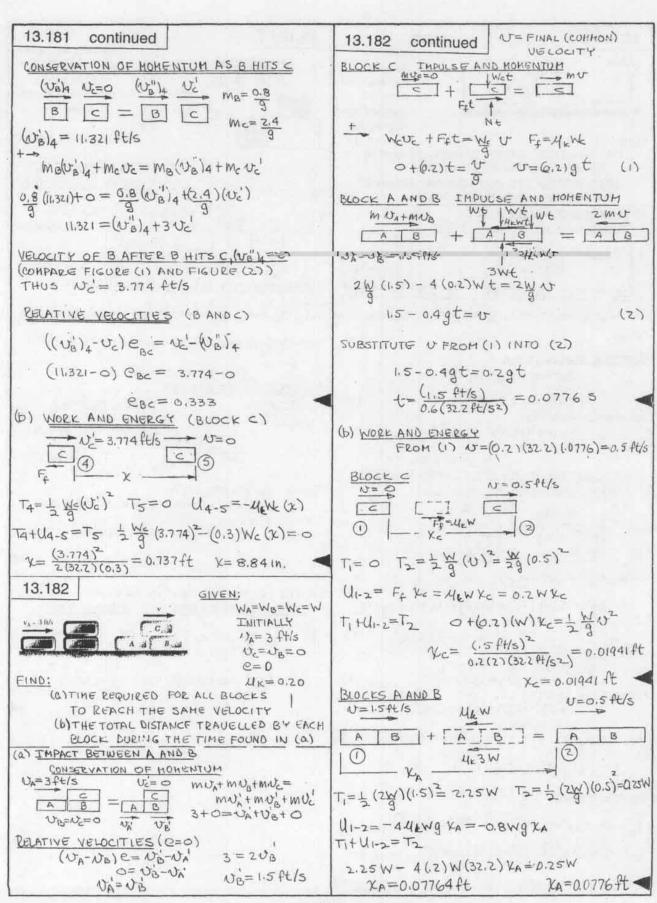
FROM (a), EQ (Z) UB= 10.77 8-UA' NB = (10.77)(1)-3.59 NB = 7.18 ft/s T= 1 WB (UB)= 1 (2.616) (7.18 Hb)

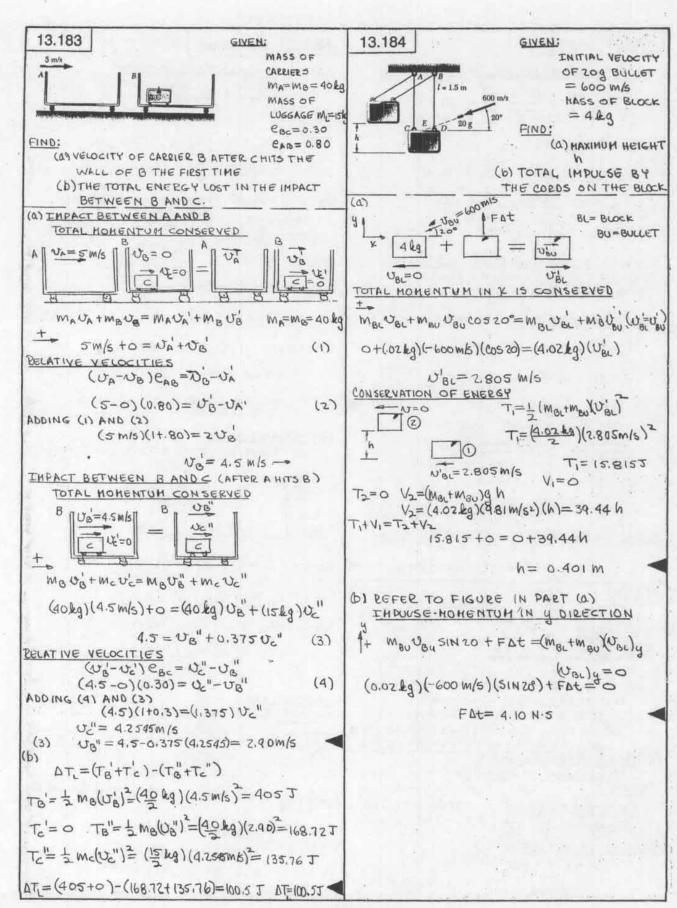
Ti= 2.08 ft.16 T3=0 V3=1 &x2 A=51b/in=601b/ft

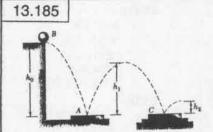
V3.= = (60)(x)2= 30x2 T, + Y = T2+ V3 $2.09 + 0 = 0 + 30x^2$ x = 0.263 ft

(1)









BALL AND THE PLATES

GIVEN:

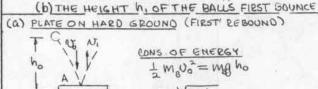
MASS OF BALL MB=709 ho=210 g BALL DROPS FREE LY FROM B h_=0.25 M POIS=M=AM FOAH RUBBER

13.186

GIVEN:

MB= 7009, MA=3509 A STRIKES B WITH VELOCITY Uo AT 60° AS SHOWN CORD BC ATTACHED TO B IS INEXTENSIBLE NO FRICTION

VELOCITY OF EACH BALL AFTER IMPACT, CHECK THAT NO ENERGY IS LOST IN THE



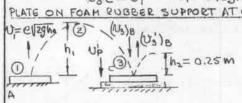
FIND:

CONS OF ENERGY I mavo= mg ho 50=129 ho=

RELATIVE YELOCITIES

U= e Vzgho U00=U1 PLATE ON FOAH RUBBER SUPPORT AT C

(a) COEFFICIENT OF RESTITUTION BETWEEN THE



CONSERVATION OF ENERGY

POINTS () AND (3) V1=V3=0

1 MBU = 1 MB(N3)B

(U3) 3= e 129ho

CONSERVATION OF HOHENTUM

1+ AT 3 MB (-U3) B+ MPUP = MB (U3) B-MPUP $\frac{Mp}{Mp} = \frac{210}{70} = 3$ -e/2gho =(v3) 2-3 Up (1)

PERATIVE YELOCITIES

[(-N2)2-(02)]C =-U2-(U3)B

E (29ho+0 = Up+ (03)B (2) 4(U) 1 = 129 ho (3e2 e)

CONSERVATION OF ENERGY AT 3, (U's) = /29 h2 THUS 4 VZghz = VZgho (3e2-e)

36-6-1.633=0

e= 0.923

b) FROM (a), N = e Vzghp POINTS () AND @ CONS. OF ENERGY & MBU, = MBY h, = e22gho=ghi

N= (0.923)2(1.5)=1.278m.

UB IS IN THE XDIRECTION

BALL A ALONE

MOMENTUM IN & DIRECTION IS CONSERVED

THUS (Uà) = 0 MA(UA) = MA (Uà) t, (UA) = 0

BALLS A AND B TOTAL MOMENTUM IN X DIRECTION CONSERVED

MAUSINGO= MA(UA) NSINGO+MBUB (0.350) 1 Un = (0.350) 1 Un +0.700 UB No= NA+ 2,309 Na

RELATIVE VELOCITIES (N-DIRECTION)

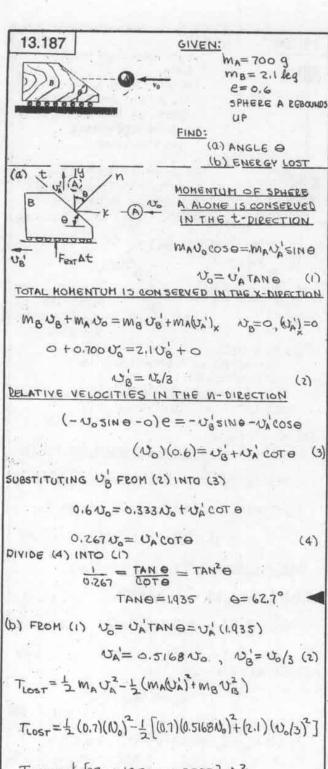
[(UA)n-(VA)n]e=(U6)n-(VA)n (No-0](1) = Un SINGO-NA

Uo= 0.866 NB - VA (2) (S) ONA (1) DNIOOA

> 200= (2.309 +0.866) UB NB= 0.6300

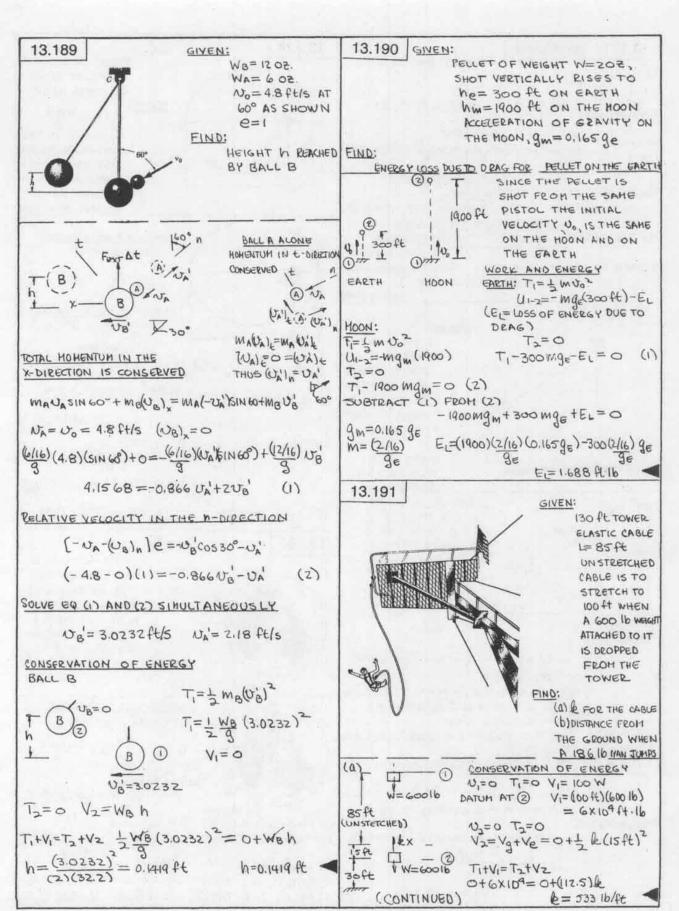
FROM (1) NA= No-(2.309)(.630 No) = -0.455 No UA'= 0 A55NO &

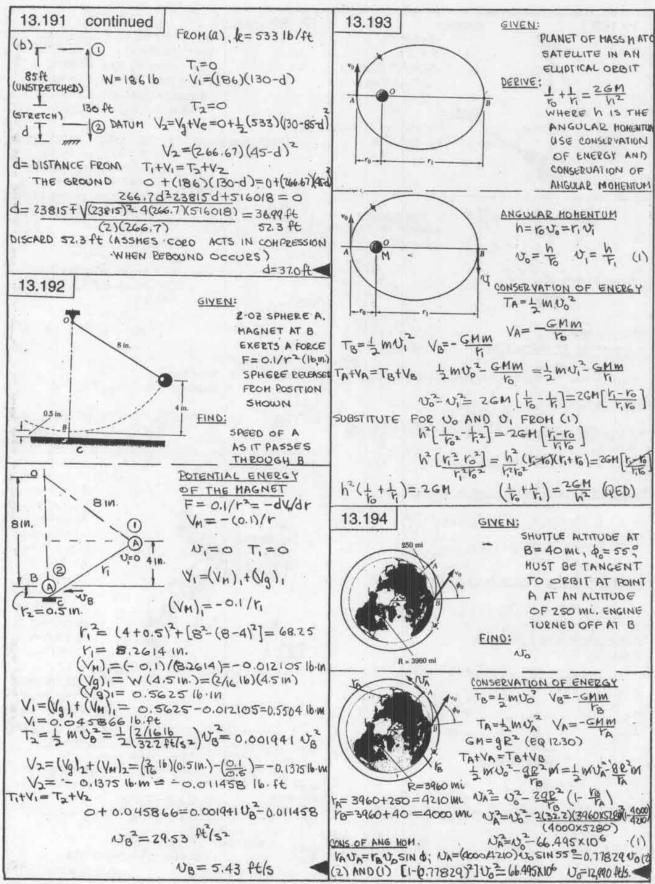
ENERGY AT = 1 MAU2 - 1 MAU12 - 1 MAU2 $\Delta T = \frac{1}{2} \left[(0.350) \left(v_0^2 - (0.455 \, v_0)^2 \right) - (0.700) \left(0.6304 \right) \right]$ AT=1 [0,350(1-0.2065)-.700 (0.3969)]V2 AT= 1 [0.278 - 0.278] 03=0 (CHECK)

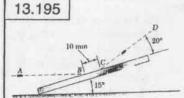


GIVEN: WA=316 12 ft/s WB= 9 16 e=0.50 SPHERE A REBOUNDS 0= 40° NA= 12ft/s FIND: (a) VELOCITIES AFTER IMPACT. UA AND UB (b) ENERGY LOST 4 MONTH. HOMENTUM OF THE SPHERE A ALONE IS CONSERVED IN THE t-DIRECTION MA(VA) += MA(VA) + (VA) = 0 (NA')+=0 (UA)=NA' 500 (No)(0.6)= va+va cote (3) TOTAL MOMENTUM IS CONSERVED IN THE X-DIRECTION MANA COS 50°+MONG=MA (-UA) COS 50+MBNB UB=0 UA= 12 ft/s (3/(12)(0550°)+0=3/(Un')(0550°)+(1)UB 23.140 = -1.9284 NA + 9 UB (NA-UB) e=(v1 cos 500+VA) (12-0)(0.5)= 0.6428 No+ NA (b) TLOST = 1 MA WA - 1 [MAUA 12+ MB(Wa)2] TLOST = \$ [0.7-0.1870 - 0.2333] 02 TLOST = 0.1400 No T TLOST = 2 [12.064 - 3.212] = 4.42 ft.16 TLOST 0.140002

13.188







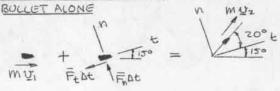
GIVEN:

25-9 BULLET
FNITIAL YELOCITY
N= 600 m/s,
HOLIZONTAL
RICOCHET
VELOCITY Uz
= 400 m/s
AT 20°

BULLET LEAVES A 10-MM SCRATCH ON THE PLATE AT AN AVERAGE SPEED OF 500 M/S FIND:

THE MAGNITUDE AND DIRECTION OF THE AVERAGE IMPULSIVE FORCE EXERTED BY THE BULLET ON THE PLATE

EMPOLSE AND MOMENTUM



MUI + FAt = MUZ

<u>t ειρεστίου</u> μυζεος 15°- FeAt = m. υ2 cos 20°

FeAt = (0.025 kg) [600 m/s cos 15°- 400 m/s cos 20°]

Fe At = 5.092 kg-m/s

 $\Delta t = \frac{5_{BC}}{V_{AV}} = \frac{0.010 \text{ m}}{500 \text{ m/s}} = \frac{20 \times 10^{-6} \text{ s}}{5}$ $F_{t} = (5.092 \text{ kg} \cdot \text{m/s}) / (20 \times 10^{-6} \text{ s}) = 254.6 \times 10^{8} \text{ kg} \cdot \text{m}$

FE= 254.6 RN

N DIRECTION

/+.-mu, siniso+FnAt = muz sin 200

Fn At= (0.0252g) [600 m/s SIN15°+400 m/s SIN28]

Finat = 7.3025 Ag.m/s At=20x10"5

Fn=(13025 : 3 m/s)/30x10 = 365.1 x10 kg m

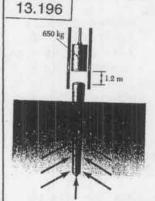
Fn= 365.1 RN

 $F = \sqrt{254.6}^{2} + (365.1)^{2} = 445.1 \text{ kn}$ $F = \sqrt{150} + (365.1)^{2} = 445.1 \text{ kn}$ $F = \sqrt{150} + (365.1)^{2} = 445.1 \text{ kn}$ $\Theta = TAN^{-1} = TAN^{$

x=34.9+15°=49.9°

FORCE OF THE BULLET F= 445 km

499



GIVEN:

DROPS IZM AND
DRIVES A 140 kg
PILE 110 MM INTO
THE GROUND
C=0

FIND:

AYERAGE RESISTANCE OF THE GROUND TO PENETRATION

VELOCITY OF THE HAMMER AT IMPACT

CONSERVATION OF ENERGY

Tit V1 = T2 + V2

 $0+7.652=0.3250^{2}$ $0^{2}=23.54 \text{ m/s}^{2}$ 0=4.852 m/s

YELOCITY OF PILE AFTER IMPACT

SINCE THE IMPACT IS PLASTIC (C=0), THE VELOCITY OF THE PILE AND HAHHER ARE THE SAME AFTER IMPACT

CONSERVATION OF HOMENTUM

 $O_{H}=4.852 \text{ M/S}$ H P $V'=O_{H}'=O_{H}$

THE GROUND REACTION AND THE WEIGHTS ARE

THUS $M_H U_H = (M_H + M_P) U'$ $N' = \frac{M_H U_H}{(M_H + M_P)} = \frac{(650)}{(650 + 140)} (4.852 M/s) = 3.992 M/s$

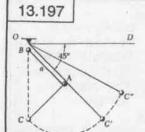
WORK AND ENERGY d=0.110 M

 $T_2 + U_{2-3} = T_3$ $T_2 = \frac{1}{2} \left(\frac{1}{12} \left(\frac{1}{12} \right)^2 \right)^2$

 $T_3 = 0$ $T_2 = \frac{1}{2}(650 + 140)(3.992)^2$ $T_3 = 6.295 \times 10^3 \text{ J}$

U2-3= (MH+MP) 9d-FAVd=(650+140)(9.81)(110)-FAV(110)
U2-3= 852.49 - (0.110) FAV

T2+U2-3=T3 6.295×103+852.49-(0.110)FAV= 0 FAV= (7147.5)/(0.110)=64.98×103N FAV=65&N



GIVEN:

SPHERE RELEASED FROM REST AT B. CORD OF LENGTH ZQ BECOHES TAUT AT C

FIND:

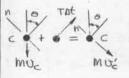
VERTICAL DISTANCE FROM OD TO THE HIGHEST POINT C" REACHED BY THE SPHERE

VELOCITY AT POINT & (BEFORE THE CORD IS TAUT) CONSERVATION OF ENERGY FROM B TO C.



TB= 0 VB = mg (2) (12) a = mga [2 Tc= 1 MIC2 Vc=0 TR+VB = Tc + Ve

YELOCITY AT C (AFTER THE CORD BECCHES TAUT) LINEAR MOHENTUN PERPENDICULAR TO THE CORD IS CONSERVED.



0=45° + - MUZ SINO = MUZ NE=(12/2)(12)/9a

NOTE: THE WEIGHT OF THE SPHERE IS A

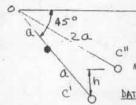
Nc=1/2 ga = 24/ga

NON-IMPULSIVE FORCE

CTOC

VELOCITY AT C' (CONSERVATION OF ENERGY) Tc = 1 m(v'c) Vc=0 Tc1 = 1 m(Nc1)2 Vc1=0

NC=NoI C'TO C" (CONSERVATION OF ENERGY)



 $L^{c_1} = \frac{3}{7} M(\Omega_{c_1}^{c_1})_3$

TC1=1m (24/ga)2

To= 12 mga

Ne= 249a

V== 0 Tc = 0

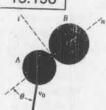
To+Vo=To+Vo 1 m ga+0=0+mgh

You = mah

h= Ea

h= 0.707a

13.198



GIVEN:

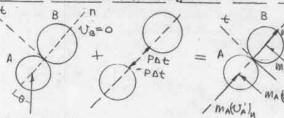
MA AND ME SLIDING ON A FRICTIONLESS SURFACE INITIALLY, UB=0 NA = NO AT ANGLE & COEFFICIENT OF RESTITUTION, e

SHOW:

THAT IN COMPONENT OF THE VELOCITY OF A AFTER IMPACT IS.

(a) POSITIVE IF MAZEMB (b) NEGATIVE IF MA < EMB

(c) ZERO IF MA= EMB



MAUA=MAUO

DISKS A AND B (FOTAL MOHENTUH CONSERVED)

MAUA + MBUB = MAUA + MBUB

NORMAL DIRECTION: MANO COSO + O = MA (U) ,+ MB (B) , ()

PELATIVE VEWSITIES

$$[V_{A}\cos\theta - (V_{B})_{n}]e = (V_{B})_{n} - (V_{A})_{n}$$

$$V_{0}(\cos\theta)e = (V_{B})_{n} - (V_{A})_{n} \qquad (2)$$

MULTIRY (2) BY MB AND SUBTRACT IT FROM (1)

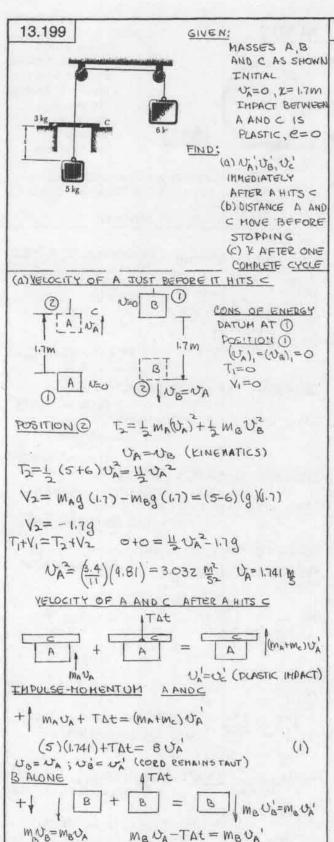
$$(V_A')_N = (V_0 \cos \Theta) \left(\frac{M_A - e M_B}{(M_A + M_B)} \right) \tag{3}$$

FROM EQUATION (3)

(UA) N POSITIVE (a) MAZEMB

(b) MA < emb (Va) N NEGATIVE

(c) ma= emo (va) n = 0



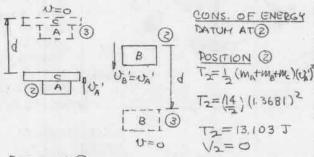
13.199 continued

AND C AS SHOWN ADDING GOVATIONS (1) AND (2), 11(1.741) = 1404

TNITIAL

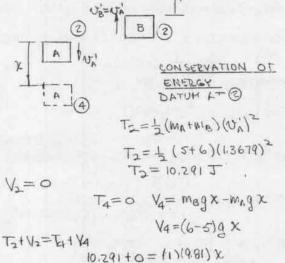
AT = 1.3679 m/s

UA=0, 1= 1.7M UA'= UB'= UC'= 1.368 M IMPACT BETWEEN DISTANCE A AND C MOVE REFORE STOPPING



POSITION (3) $T_3 = 0$ $V_3 = (m_A + m_c)gd - m_B gd$ $V_3 = (8 - 6)gd = 2gd$ $V_3 = (8 - 6)gd = 2gd$ $V_3 = (8 - 6)gd = 2gd$

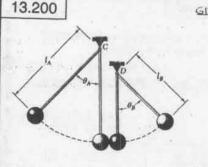
(b) AS THE SYSTEM RETURNS TO DOSITION (2), AFTER STOPPING IN POSITION (3), ENERGY IS CONSERVED AND THE VELOCITIES OF A B AN C BEFORE THE COLLAR AT C. IS REMOVED, ARE THE SAME AS THEY WERE IN (A) ABOVE WITH THE DIRECTIONS REVERSED. THUS, NA = U' = U' = 1.3679 M/S. AFTER THE COLLAR C IS REMOVED THE VELOCITIES OF A AND B REMAIN THE SAME SINCE THERE IS NO IMPULSIVE FORCE ACTING ON EITHER.



1=1.049

(5)

(6)(1.741)-TAt= 6NA

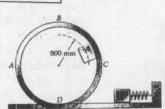


GIVEN:

SPHERE A IS EFLEASED FROM REST AT AN ANGLE OA. SPHERE B IS AT REST, IS HIT BY A, AND RISES TO A **HONLXAH** ANGLE 8 = 9

EIND:

OB IN TERMS OF RB/LA AND C.



G"EN:

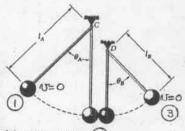
300-9 BLOCK SPRING OF CONSTANT R= 600 N/M 13 INITIALLY COMPRESSED 60 MM WHEN THE BLOCK IS RELEASE D. NO FRICTION

FIND:

13.201

FORCE EXERTED BY THE LOOP ABOD ON THE BLOCK AS IT PASSES THPCUGH. (a) POINT A

(b) POINT B, (c) POINT C



CONSERVATION OF ENERGY (1) + (2) DATUM AT (2) SPHERE A POSITION (1)

UA=0 T=0 VI= Mgla(1-cose)

MA=MB=M OA-OB Ti+Vi=T2+ V2

POSITION (2) T2=1 MUA2

0+mg.la (1-cose)=== m v2+0 152= 292A (1-COS OA) (1)

CONSERVATION OF MOHENTUM AT (2)

MUA + MUB= MUA+ MUB (A)(B) = (A)(B)MUA UBO MUA MUB NA+O= UA+UB (2)

BELATIVE VELOCITIES AT(2)

(UA-NB) C= UB-UA NAE = UB-UA

ADDING EQUATIONS (2) AND (3) AND SOLVING UB = = (Ite)UA

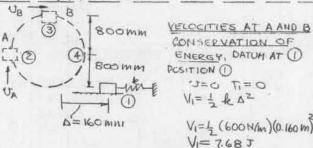
CONSERVATION OF ENERGY @-3 SPHERE B

POSITION @ T2=1 m(vg) V2=0

V3=mg(B(1-cos 08) 1 m(vb) +0=0+mg(g(1-coseb)

SUBSTITUTE UB FROM EQ. (4) INTO EQ. (5) 1 (He) 2 NA = 2 gl B (1-COS DA) (6)

DIVIDE (1) INTO (6) AND SET ON OB $\frac{U_A^2}{2928(1-\cos\theta_B)} = \frac{2928(1-\cos\theta_B)}{2928(1-\cos\theta_B)}$ 28/la=(1+e)2/4



POSITION (2) T2= 1 M NA= 1 (0.3) UA = 0.15 UA V2= mg (0 800m) = (0.3kg)(g)(0.8m)=0,24g T1+V1 = T2+V2 0+ 7.68 = 0.15 UA +0.249

UZ= 7.68-(0.24)(9.81)

UAZ 35.50 m3/52 POSITION(3) T3=1 m 08=1 (0.3) U8=0.15 UB2 V3= mg(1.6m)=(0,3)(9)(1.6)=0.48q 0+7:68 = 0.1502 +0.489 T1+V1=T2+V2

UB= 7.68-(0.48 (9.81)-19.81 m2

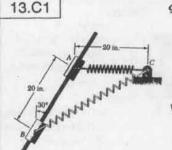
POSITION (4) SINCE Y4=V2 THE VELOCITY UA= Ne U= 35.50 m2/52 NEWTONS SECOND LAW

EF= NA= man an= NA2 = (35.50 M/52) (0.8 M) NA= (0,3 kg) (35,50 m2/52)

(b) NA= 13.31 N-AT B Mae ZFn= NB+mq = man an= NB2 = (19.81 m2/52) [] = (UB)= 29(B(1-COSOB) (5) NB=(0.3kg)(19.81 m3/52)-(0.3kg)(9.81 m3) NB= 4.49N 1 Mat I ZFn=Nc=man an=ve2

Nc=(0.3 lg)(35,50 m2/52)/(0.8 m) Nc= 13.31 N

(4)



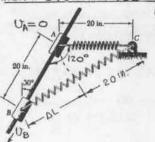
GIVEN

COLLAR WA= 12 16
SPEING IS
UNSTRETCHED WHEN
COLLAR IS AT A.
COLLAR RELEASED
FROM REST AT A

FIND:

YELOCITY AT B
FOR &= 0.1 lb/in
TO 2.0 lb/in IN
0.1 lb/in INCLEMENTS

WRITE EQUATION FOR UB IN TERMS OF A



212714NA 20120(02)5-05+05-2(20)05(20) 2005(20)4008=(1405) 2105(12)11 64 (14)

CONSERVATION OF ENERGY $V_A = 0$ $T_A = \frac{1}{2} m V_A^2 = 0$ $T_B = \frac{1}{2} m V_B^2 = \frac{6}{9} V_B^2$ $V_A = 0$ (DATUM AT A)

VB= -(121b)(20 ft)(cos 30)+1 (fe(16/10)(1210/ft)(1.220ft)(2)

VB=(-17.32 + 8.932 le)(16.ft), (INPUT le IN 16/IN.)

TA+VA=TB+VB 0+0=(616)UB-17.32+8826

 $U_8 = [92.95 - 47.933 \text{ k.}]^{1/2} (ft/s) (1)$

OUTLINE OF PROGRAM

INPUT & IN (1) IN 16/4N IN 0.1 16/11 INCREMENTS
AND STOP WHEN &= 2.0 16/11

PRINT VALUES OF UB (IN ft/s)
NOTE: COLLAR MEYER REACHES B FOR

k>(92,95)/47,933)= 1,939 1b/in.

PROGRAM OUTPUT

13.C1

K (LB/IN)	VELOCITY	(FT/S)	
0.10	9.39		
0.20	9.13		
0.30	8.86		
0.40	8.59		
0.50	8.31		
0.60	8.01		
0.70	7.71		
0.80	7.39		
0.90	7.06		
1.00	6.71		
1.10	6.34		
1.20	5.95		
1.30	5.54		
1.40	5.08		
1.50	4.59		
1.60	4.03		
1.70	3.39		
1.80	2.58		
1.90	1.37		

13.C2 | GIVEN:

CAR WEIGHT, W= 2000 lb

FOR FIRST GOFT ALL WEIGHT IS ON

THE REAR WHEELS WHICH ARE SUPPING

FOR REHAINING 1260 FT, 60 % OF THE

VIEIGHT IS ON THE REAR WHEELS

WITH JUI APING IMPENDING.

US=0.60 UR=0.85

AERODYNAMIC DRAG FT=0.0098 UZ

VITH U'IN FT/S AND FT IN 16.

FIND:

VELOCITY AND ELAPSED THE WITHAND WITHOUT DRAG.
EVERY 5 ft FOR THE FIRST GO ft AND
EVERY 90ft FOR THE REMAINING 1260 ft.

ANALYSIS USE WORK AND ENERGY IN INCREMENTS OF DX = 0.1 Pt BETWEEN CTH AND CHITH INTERVAL

+ = = (U=0 FOR (=0)

 $T_{i} + U_{(c) \rightarrow (i+1)} = T_{i+1} + \frac{1}{2} m v_{i}^{2} - (F_{0} + F_{i}) \Delta x_{i} = \frac{1}{2} m v_{c+1}^{2}$ $V_{i+1} = \left[v_{i}^{2} + \frac{29}{W} (F_{0} - F_{0}) \Delta x_{i} \right]^{\frac{1}{2}} + F_{0} = 0.0098 v_{i,0}^{2}$

Atc = ZAXL FIRST (

FIRST GOFT FF = 4KW=(0.85)W FOR REMAINING 1200 FE

4xc=0.1 ft g=3z.2 ft/s W=2000 lb F_= (0.60)45W= 0.36W

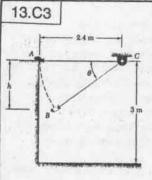
CUTLINE OF PROGRAM

IDENTIFY N, AND VID AS THE UELOCITIES IN THE CTH INTERVAL WITHOUT AND WITH DRAG, WITH UC==0 AND Ft=0.85W USE A LOOP TO SOLVE FOR UL+1 AND TO SOLVE FOR ti, SUM AXI TO FIND XI AND SUM At TO FIND ti, PRINT UL, ti, XI at 5ft INTERVALS REPEAT FOR REHAINING 1260 ft WITH Ft=0.36W.

PRINT YL, UL, ti AT 90 th INTERVALS

13.C2

DISTANCE (FT)	V(FT/S)	T(S)	V(FT/B)	T(S)	
DISTRICE (FA)		DRAG	DRAG	12.27.1	
5.		0.719	13.89	0.720	
10.		1.017	19.64	1.018	
15.	24.07		24.05	1.247	
20.		1.439	27.76	1.440	
25.	31.08				
30.	34.05			1.764	
35.	36.78		36.67	1.905	
40.		2.035		2.037	
45.	41.70	2.158	41.55	2,161	
50.	43 95	2 275	43.78	2.278	
55.	46.10	2.386	45.90	2.390	
60.	48.15	2.492	47.92	2.496	
60 FT. TO			INTERVALS		
150.	72.65		71.76	4.000	
240.			89.00	5.119	
330.	105.78			6.057	
420.	118.93	6.803	115.02	6.882	
510.	130.77	7.524	125.60	7,630	
600.	141.62	8.184	135.09	8.320	
690.	151.70	8.798	143.71	8.966	
780.	161.15	9.373	151.63	9.575	
870.	170.08	9.917	156.95	10.155	
960.	178.56	10.433		10.709	
1050.	186.66	10.926			
1140.	194.42		178.06		
1230.	201.88	11.852		12.253	
1320.	209.07	12.290	188.96	12.736	



GIVEN: 5- kg BAG ROPE = 2.4 M LONG INITIAL VELOCITY ZERO

SIND:

FOR VALUES OF MAXIMUM TENSION FM FROM 40 TO 140 M IN S-N INCREMENTS, THE (a) DISTANCE IN (b) DISTANCE OF FROM

(b) DISTANCE & FROM
THE WALL TO THE
DOINT WHERE THE
BAG HITS THE

NALYSIS l=24m c

BAG HOVES ALONG A
CIRCULAR ARC AB UNTIL
THE LOPE BLEAKS (RABIUS, L)
NEWTONS LAW

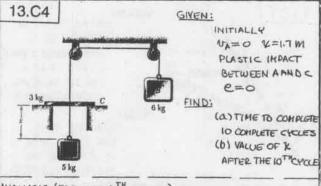
$$\begin{array}{ll} sin\theta = h & conservation of energy \\ v = vzgh & vzgh \\ sin\theta = \frac{h}{2} & Fm = 2mg h + mgh = 3mg h \\ \theta = sin^{-1} \frac{h}{2} & (z) & h = \frac{Fm k}{3mg} & (3) \end{array}$$

PEON B TOD (PROJECTIVE TRAJECTORY) $N_H = U \leq IN\Theta \Rightarrow d = (L - l\cos\Theta) + U_H + D$ (4) $U_{AF} = U \cos\Theta + (3-h) = U_U + g + \frac{1}{2} \frac{1}{2}$ (6) $t_D = -\frac{U_U}{2} + \sqrt{\frac{1}{2} \frac{1}{2} + \frac{1}{2} \frac{1}{2} - h}$ (5)

WITH $\ell=2.4m$, m=5kq, $q=q.81m/s^2$ in Equation (3), and for Fm in 5n increments from 40to 140 N, solve for h. For each h, solve for υ (Eq.1), and υ (Eq.2). Solve for υ And υ (Eq.6) and with υ and h. solve for υ (Eq.5) and with υ h, to solve for υ in (Eq.4). Print h and d for each Fm.

PROGRAM OUTPUT

FORCE (NEWTONS)	H (METERS)	d (METERS)
40	0.652	0.503
45	0.734	0.585
50	0.815	0.668
55	0.897	0.752
60	0.979	0.839
65	1.060	0.927
70	1.142	1.017
75	1.223	1.109
80	1.305	1.203
85	1.386	1.300
90	1.468	1.401
95	1.549	1.505
100	1.631	1.615
105	1.713	1.731
110	1.794	1.854
115	1.876	1.989
120	1.957	2.137
125	2.039	2.306
130	2.120	2.504
135	2.202	2.753
140	2.283	3.101



ANALYSIS (FOR THE LTH CYCLE)

REFER TO FIGURES IN THE SOLUTION TO PROB. 13.199

FROM TO (2) CONSERVATION OF ENERGY (REFORE IMPACT) $T_i = 0$, $V_i = 0$, $T_2 = \frac{1}{2}(S+6)(V_A) = \frac{11}{2}(V_A)(V_A)$ $V_2 = (S-6)g(Y_1) = -g\chi_1$ $T_1 + V_1 = T_2 + V_2$ $O + 0 = \frac{11}{2}(V_A) = g\chi_1$, $(V_A) = \sqrt{\frac{1}{11}}g\chi_1$ (1)

TIME ($t_1 = 3$)

ACCELERATION FROM () TO (2) IS CONSTANT,
THUS AVERAGE YELDCITY IS (VI) = O+(VI);

AND
$$(\overline{U}_{A})_{c} = \frac{\chi_{c}}{(\pm_{1-2})_{c}} (\pm_{1-2})_{c} = \frac{2\chi_{c}}{(U_{A})_{c}} = \frac{2\chi_{c}}{\sqrt{\frac{2}{11}}9\chi_{c}}$$

$$(\pm_{1-2})_{c} = \frac{1498\sqrt{\chi_{c}}}{(2)}$$

(AFTER IMPACT) AT @

TMPXSE-HOHENTUH FOR A AND C

5(UA); + TAL=(8)(UA);

IMPULSE-HOHENTOH FOR B

 $6(v_A)_C$ Tate $6(v_A)_C$.

ADDING $1|(v_A)_C = 14(v_A)_C$ $(v_A)_C = \frac{1}{14}(v_A)_C = \sqrt{\frac{119}{98}}(3)$ FROM (2) TO (4), (SEE (b) IN SOLUTION TO PROB. 13.199)

CONSERVATION OF ENERGY DATUM AT (2) $T_2 = \frac{1}{2}(5+6)(N_A^{-1})^2 = \frac{1}{2}\frac{19}{298}(3+2) = \frac{1}{19}(4+2)$

T4=0 V4 = MBg Ri+1-MAg Ri+1=6-5)g Ri+1=g Ri+1
T2+V2=T4+V4
1219Ri+0=0+g Ri+1
Yi+1=121 Ri (4)

TIME @ TO @ (+24) [(2)(121) (1/6) [196 VI) (5) (+24) [(196 VI) (

THE FROM (2 TO 3) AND FROM (3) 10(2) $T_2 = \frac{1}{2} (m_A + m_B + m_C) (V_A^*)_i^2 = 7 (\frac{11974}{98}) = \frac{779}{98} \text{ k}_i$ $T_3 = 0 \quad \forall_3 = (m_A + m_C) \text{g di} - m_B \text{g di} = \text{(8-6) g di} = \text{2g di}$ $T_2 + V_2 = T_3 + V_3 \quad \frac{779}{98} \text{ k}_i + 0 = 0 + 29 \text{ di}$ $(t_{2-3})_i = \frac{2 \text{ di}}{(V_A^*)_i}$ $(t_{2-3})_i = \frac{2}{2} (\frac{77}{196} \text{ k}_i) / \frac{11976}{196} = 0.7488 \text{ k}_i$ (6)

 $(t_{2-3})_i = 2(\frac{77}{196}k_i) / \frac{119}{98} = 0.7488 / \frac{1}{19}$ (6) $(t_{3-2})_i = (t_{3-4})_c = 0.7488 / \frac{1}{19}$ (7)

(CONTINUED)

13.C4 continued

TOTAL TIME TO COMPLETE THE LTH CYCLE EQS, (2)+(6)+(7)+(5) ti=(t12)+(t2-3)(+(t3-2)i+(t2-4)i

ti=(1.498 +0.7488+0.7488 +1.1766) \Ti

ti= 4.172 \xi (8) OUTLINE OF PROGRAM

SET X = 1.7 M ((=1)

(A) CALCULATE YELL FROM EQUATION (4) FOR L= 1 TO L= 10. FOR EACH VALUE OF X USE EQUATION (B) TO DETERHING & FOR THE L' CYCLE, SUM t'S TO OBTAIN THE TOTAL TIME THROUGH THE 10TH CYCLE.

(b) FOR C=10 OBTAIN & FOR THE TENTH

PRINT TOTALTIME AND & FOR THE 10TH CYCLE. PROGRAM OUTPUT

TOTAL TIME=23.1 SECONDS

X FOR THE TENTH CYCLE=0.01367 METERS

13.C5

GIVEN

MB= 700 g, MA= 350 9 No=6 m/s NB=0 0 = 20° TO 150° IN 10° INCREMENTS

FIND:

UA AND UB AFTER IMPACT AND ENERGY LOST FOR,

(a) e=1

(b) e=0.75

(c) e=0

ANALYSIS:

DEVELOP FORHULAS FOR U'S AND U'S IN TERMS OF & AND &

CONSERVED IN THE t. DIRECTION

MA(UA)+=M WALE (UA) = O THUS WA' = O AND WA' IS ALONG THE N AXIS

KINEMATICS

(Wa) = UB

continued 13.C5

CONSERVATION OF MOHENTUM IN THE & DIRECTION FOR A AND B TOGETHER

MAUASINGO = MAUASINGO + MBUB MA=0.350 kg MB=0.700 kg UA=Nb=6 M/S 6 SINDO = NASINDO+ ZUB

PELATIVE VELOCITIES IN THE IN DIRECTION (NA-0) @=NBSINGO-NA NA=00=6 M/S

6e = NBSINE O-NA (2)

MULTIPLY (2) BY SINGO AND ADD TO (1) TO GET US UB' = 651NO0 (1+e) (2+51N2 Q)

SUBSTITUTE (3) IN (2) FOR UA (4) Un'= 651N20-12E (2+51N200)

FOR 00 390°, TOL= O, AND BALL A AT A VELOCITY OF 6 MIS HITS BALL B WHICH IS AT O VEWCITY AND IS NOT CONSTRAINED BY THE CORD. THUS IF ONLY HAGNITUDES ARE CONSIDERED WA AND UB HAVE VALUES FOR 110°< 9 > 90° WHICH ARE THE AS FOR 0= 90° ENERGY LOST

DE= = MUA- - (MAUA -+ MBUB) AF = 1 (0.350)[VA-VA]-1 (0.700) V6 (5)

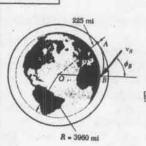
OUTLINE OF PROGRAM

INPUT O, INTO EQUATIONS (3) AND (4) FROM 20° TO 90° IN INCREHENTS OF 5° FOR E=1, C=0.75 AND C=0 TO OBTAIN U' AND U'S, SUBSTITUTE NA AND US IN (5) TO OBTAIN DE. PRINT C. O. U. U. DE, DE PROGRAM OUTPUT

13.C5

e	THETA (DEG)	VEL A (M\S)	VEL B (M\S)	* E LOST
1.00	20.	-5.337	1.939	0.0
1.00	30.	-4.667	2.667	0.0
1.00	40.	-3.945	3.196	0.0
1.00	50.	-3.278	3.554	0.0
1.00	60.	-2.727	3.779	0.0
1.00	70.	-2.325	3.911	0.0
1.00	80.	-2.081	3.979	0.0
1.00	90.	-2,000	4.000	0.0
0.75	20.	-3.920	1,696	41.3
0.75	30.	-3,333	2,333	38.9
0.75	40.	-2.702	2,797	36.3
0.75	50.	-2.118	3.109	33.8
0.75	60.	-1.636	3.307	31.8
0.75	70.	-1.284	3.422	30.4
0.75	80.	-1.071	3.482	29.5
0.75	90.	-1.000	3.500	29.2
0.00	20.	0.332	0.969	94.5
0.00	30.	0.667	1.333	88.9
0.00	40.	1.027	1.598	82.9
0.00	50.	1.361	1.777	77.3
0.00	60.	1.636	1.890	72.7
0.00	70.	1.838	1,956	69.4
0.00	80.	1.959	1.990	67.3
0,00	90.	2,000	2.000	66.7
VALUE	S FOR ANGLES OSE FOR 90 DE		DEGREES ARE 7	THE SAME

GIVEN:



INITIAL CIPCULAR OPBIT OF 225 ML ABOVE THE SURFACE OF THE EARTH INCREMENTAL VELOCITY DUA TOWARD THE CENTER OF THE BARTH

NO AND PO AT AN AUTHUR OF 40 MI FOR ENERGY EXPENDITURE OF 5 TO 100 % DF THAT USED IN PROB. 13.109 IN 5% INCREMENTS

13.C6 continued

2GM (to ta)= 58.93×106 ft3/52

EQUATION FOR COR

CONSERVATION OF ANGULAR HOHENTUM

TACUA LIET TO UB SIN 4B

OB = SIN [raWalciec / raWal OUTLINE OF PROGRAM

INPUT CONSTANTS INTO EQUATION (4) AND SOLVE FOR UB FOR VALUES OF K OF 5% TO 100 % AT INTERVALS OF 5% FOR EACH VALUE OF UB AND USING THE GIVEN CONSTANT VALUES OF (VALUEC, YA AND YB, USE EQUATION (5) TO SOLVE FOR DB. PRINT K, UB AND Da.

ANALYSIS



CONSERVATION OF ENERGY TA = 1 M(VA) CIEC + (AY)

VA=-GMM

AT POINT B TB=1 MUB2

VB= - GMM ta= 3960 + 225 = 4185 mi VB= - 1 rB= 3960 + 40 = 4000 mi TA+VA=TB+VB (UA)218c = 922

= m[(UA kiec+QU)]-GMM = 12mUB-GHM UB= (NA) CIEC HAUD+ ZGM [to - ta] (1)

ENERGY EXPENDITURE IN PROB. 13.109 LET NA = UELOCITY AT A IN PROB. 13.109 TO BRING THE VEHICLE TO B AT \$ = 600 FROM 13.109, UA = 11.32 × 103 ft/s ENERGY EXPENDITURE, E= 1 M [(V) Like (VA)]

ENERGY EXPENDITURE IN THIS PROBLEM KE=1 M (AUA)2, WHERE K IS THE % GNERGY

USED IN PROB 13.109. SOLVING FOR (AUX) AND REPLACING E BY EQUATION (2)

$$\left(\Delta U_{A}\right)^{2} = \frac{K}{100} \left[\left(U_{A}\right)_{CIRC}^{2} - \left(U_{A}\right)^{2} \right]$$
 (3)

EQUATION FOR UB (SUBSTITUTE (3) INTO (1) NB= {(VA) = (VA) = (VA) = (VA) = + ZGM [+ - 1] = 2

 $\frac{\text{CONSTANTS:}}{\left(V_{A}\right)_{\text{CIRC}}^{2}} = \frac{9 R^{2}}{r_{A}} = \frac{32.7)\left(3960)(5280)\right]^{2}}{(4185)(5280)}$

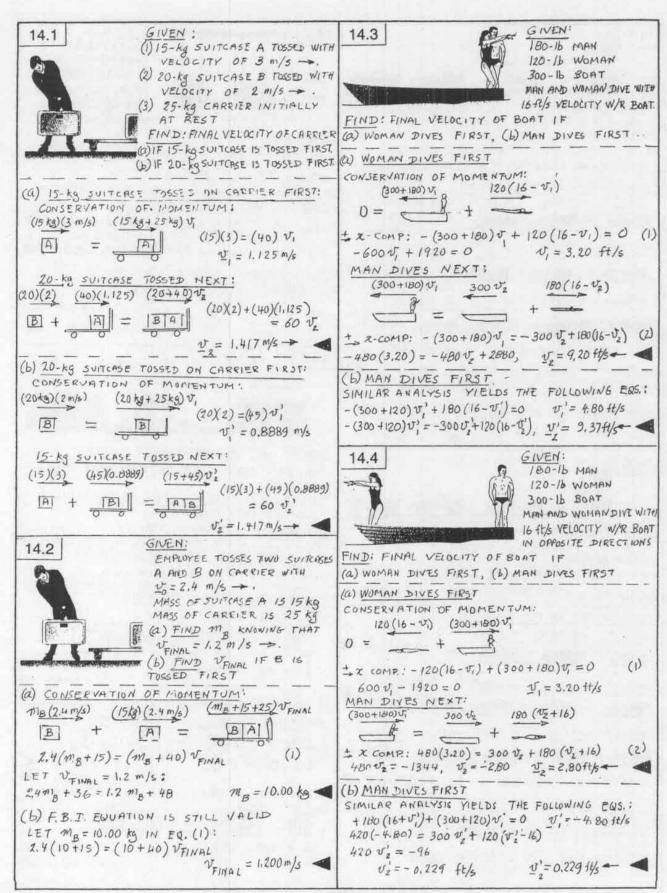
 $(N_A)^2_{\text{CIRC}} = 637.07 \times 10^6 \text{ ft}^2/\text{s}^2$ FROM 13.109, $(N_A)^2 = (11.32 \times 10^3)^2 = 128.14 \times 10^6 \text{ ft}^2/\text{s}^2$

(CONTINUED)

PROGRAM OUTPUT

13.C6

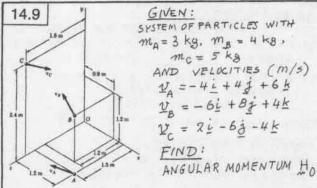
K (%)	VB (FT/S)	PHI (DEGREES)
5.	26860.	79.5
10.	27329.	75.1
15.	27791.	71.8
20.	28245.	69.2
25.	28692.	67.0
30.	29132.	65.0
35.	29566.	63.3
40.	29993.	61.7
45.	30414.	60.3
50.	30830.	58.9
55.	31240.	57.7
60.	31644.	56.6
65.	32044.	55.5
70.	32438.	54.5
75.	32828.	53.6
80.	33214.	52.7
85.	33595.	51.8
90.	33971.	51.0
95.	34344.	50.3
100.	34712.	49.5





40 tons

ひ=0



SYSTEM OF PARTICLES WITH ma= 3 kg, mB = 4 kg, mc = 5 kg AND VELOCITIES (M/s) 1 = - 4 i + 4 j + 6 k UB = - 61 + 8j + 4k Vc = 21-63-4k

Ho = 1 x x m V + 2 x m V + 2 x m Cho USING DETERMINANT FORM FOR VECTOR PRODUCTS AND FACTURING MASSES! 14 2 K | 16 2 K H = (3k8) 1.2m 0 1.5m +4 0.9 1.2 1.2 +5 0 2.4 1.8 1-4 m/s 4 m/s 6 m/s -6 8 4 2 -6 -4 =-18 1-39.6j+14.4k-19.2i-43.2j+57.6k+6i+18j-24k H = - (31.2 kg·m²/s)i - (64.8 kg·m²/s)j + (48.0 kg·m²/s)k

GIJEN: 14.10 SYSTEM OF PARTICLES OF PRUB, 14,9

FIND: (a) POSITION VECTOR & OF MASS CENTER G.

(b) LINEAR MOMENTUM OF SYSTEM.

(c) ANGULAR MOMENTUM HG OF SYSTEM. ALSO: VERIFY THAT ANSWERS TO PROBS. 149 AWD 14.10 SATISFY EQUATION

Ho = = Xn1 V + HG

(a) EQ. (14,12): mE= Zmiz;

(3+4+5) = 3 (12 i+15k)+4(0.9i+1.2j+1.2k)+ 5(2.41+1.8k)

12 2 = 7,2 6 + 16,8 j + 18,3 K

E = (0,600 m) i + (1,400 m) j+ (1,525 m) k ◀

(b) L = Zm, 2; = 3(-4i+4j+6k)+4/-6i+8j+4k)+ 5 (21-6j-4K)

L=(-26.0 kg.m/s)i+(14.00 kg.m/s)j+(14.00 kg.m/s)k

(c) HG = ENG X MA TA + EBBXMB TB + EC/6 X MC CC WHERE ENG = 10 - 2 = 1,21 +1,5k - (0.61 +1.4 + 1.525 k) = 0.6 i - 1.4 j - 0,025 k

EB/6 = = = = 0.3i - 0.2j - 0.325 k 20/6 = 2 - = = -0.6i + à + 0.275 x

H=(3kg)(0.6m-1.4m-Q025m+40.3-02-0,25+5-0.6 1 0.275 -4n/s 4m/s 6 m/s | -6 8 4 | 2 -6 -4 =-24.9i-10.5j-9.6k+7.2i+3j+4.8k-11.75i-9.25j+8t

HG = - (29.45 kg. m/k) i - (16.75 kg.m/k) i + (3.70 kg.ni/k) k (CONTINUED)

continued 14.10

WE COMPUTE EXMT:

Exmit = TexL = (0.6i+1.4j+1.525k)x(-26i+14j+14k) = a6 7.4 1.525 = -1.75 i -48.05 j +44.8 t

THUS: Exm T+ Hg=-1,75i-48.05j+44.8k-29.45i-16,75j+320k =-31,26-64.8j+48.0k

WHICH IS THE EXPRESSION OBTAINED FOR HOIN PRUB. 14.9.

14.11

GIVEN: SYSTEM OF PARTICLES WITH m=3 kg, m=4kg, mc=5kg AND VELOCITIES (M/S) V=-41+43+6k V= Vx 1 + vg j + 4 k V = 21-6j-4k FIND: (a) V, AND V, FOR WHICH

HO IS PARALLEL TO E AXIS

(b) CORRESPONDING Ho.

H= Exxmy + Exmy + Exm v + Exm v C $= (3 kg) \begin{vmatrix} \frac{1}{4} & \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1}{4} \\ -4 \frac{1}{4} & \frac{1$

=-18 = -39.6j+14.4k+(19.2-48 Vy) =+(48 Vg-14.4)j+ (3.6 Vy-4.8 V,) k+6i+18j-24k

H = (7.2-4.8 of) i +(-36+4.8 of) + (-9.6+3.6 of-48 of) (a) FOR H TO BE / ZAXIS:

Hy = -36 +4,8 -5 = 0 H=7,2-4,8 Vy =0 Ux = 7.50 m/s, Ux = 1.500 m/s

(b) Ho = H2 K = (-9.6+3.6×1,500-4.8×7.50) K H = - (40.2 kg·m/s) k

14.12 GIVEN: SAME SYSTEM OF PARTICLES WITH SAME VELOCITY DATA AS IN PROB. 14.11 FIND:

(a) UX AND UY FOR WHICH HO IS PARALLEL TO YANIS. (b) CORRESPONDING HO.

SEE SOLUTION OF PROB. 14.11 FOR DERIVATION OF EQ. (1):

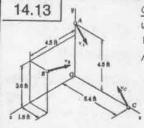
H = (7.2-4.8 M) = +(-36+4.8 V2) =+(-9.6+3.6 V, -4.8 V) =

(a) FOR HO TO EE // TY AXIS:

He = -9.6 +3.6 Vy -4.8 V=0 H = 7.2 - 4.8 Vy = 0 U = 1,500 m/s -9.6+3.6 (1,500)-4.8 v2 = 0 V=-0,875 m/s v = -0.875 m/s, v = 1,500 m/s

(b) H = Hy j = [-36+4.8x(-0.875]]j

H = - (40.2 kg·m²/s)j



GIVEN: SYSTEM OF PARTICLES

WITH $W_A = 9.66 \, lb$, $W_B = 6.44 \, lb$, $W_c = 12.88 \, lb$ AND VELOCITIES (ft/s) $V_A = 4\dot{i} + 2\dot{j} + 2\dot{k}$ $V_B = 4\dot{i} + 3\dot{j}$ $V_C = -2\dot{i} + 4\dot{j} + 2\dot{k}$

FIND: ANG. MOMENTUM H.

 $\begin{array}{l} H_{b} = \frac{\pi}{4} \times m_{A} \underbrace{v}_{A} + \frac{\pi}{6} \times m_{B} \underbrace{v}_{B} + \frac{4}{5} \times m_{C} \underbrace{v}_{C} \\ USING DETERMINANT FORM FOR VECTOR PRODUCTS AND FACTORING MASSES! \\ H = \frac{9.66}{32.2} \begin{vmatrix} i & j & k \\ 0 & 4.5 & 0 \\ 4 & 2 & 2 \end{vmatrix} + \frac{6.44}{32.2} \begin{vmatrix} i & j & k \\ 4 & 3 & 0 \end{vmatrix} + \frac{12.88}{32.2} \begin{vmatrix} i & j & k \\ 5.4 & 0 & 0 \\ 4 & 3 & 0 \end{vmatrix} + \frac{12.88}{32.2} \begin{vmatrix} i & j & k \\ 2 & 2 & 4 & 2 \end{vmatrix} \\ = 0.3(9i - 18k) + 0.2(-13.5i + 18j - 9k) + 0.4(-10.8j + 21.6k) \\ H = -(0.720 \ ft \cdot |b \cdot 5)j + (1.440 \ ft \cdot |b \cdot 5)k \end{array}$

14.14 GIVEN: SYSTEM OF PARTICLES OF PROB. 14.13

FIND: (a) POBITION VECTOR & OF MASS CENTER G.
(b) LINEAR MOMENTUM OF SYSTEM.

(C) ANGULAR MOMENTUM H OF SYSTEM.

ALSO: VERIFY THAT HANGERS TO PROBS. 14.13 AND

14.14 SATISFY, EQUATION

HO= Exm V+ HG

(a) EQ.(14.12): $m\bar{\xi} = \sum m_i \bar{\xi}_i$ WHERE $m_A = 0.3$, $m_B = 0.2$, $m_C = 0.4$, m = 0.9 $0.9\bar{\xi} = 0.3(4.5j) + 0.2(1.8i + 3.6j + 4.5k) + 0.4(5.4i)$ $\bar{\xi} = (2.60 + 1)i + (2.30 + 1)j + (1.00 + 1)k$

(b) $L = Zm_1 \underline{v}_1 = 0.3(4\underline{i} + 2\underline{j} + 2\underline{k}) + 0.2(4\underline{i} + 3\underline{j}) + 0.4(-2\underline{i} + 4\underline{j} + 2\underline{k})$ $L = (1.200 \text{ lb·s})\underline{i} + (2.80 \text{ lb·s})\underline{j} + (1.400 \text{ lb·s})\underline{k}$

(c) $H_G = \frac{v}{A/G} \times m_A \frac{v}{A} + \frac{v}{B/G} \times m_B \frac{v}{B} + \frac{v}{C/G} \times m_C \frac{v}{C}$ WHERE $\frac{v}{A/G} = \frac{v}{A} - \frac{v}{2} = \frac{v}{A} \cdot 5 \cdot j - (2.8 \cdot i + 2.3 \cdot j + \frac{k}{A}) = -2.8 \cdot i + 2.2 \cdot j - \frac{k}{A}$ $\frac{v}{B/G} = \frac{v}{B} - \frac{v}{2} = \frac{1.8 \cdot i + 3.6 \cdot j + 4.5 \cdot k - (2.8 \cdot i + 2.3 \cdot j + \frac{k}{A}) = -1.4 \cdot 1.3 \cdot j + 3.5 \cdot k}{\frac{v}{C/G} = \frac{v}{2C} - \frac{v}{2} = 5.4 \cdot i - (2.8 \cdot i + 2.3 \cdot j + \frac{k}{A}) = 2.6 \cdot i - 2.3 \cdot j - \frac{k}{A}$ $H_G = 0.3 - \frac{v}{A} \cdot \frac{v}{A} \cdot \frac{v}{A} \cdot \frac{v}{A} + 0.2 - \frac{v}{A} \cdot \frac{v}{A}$

H=-(0,420 ft.lb.s) i+(2,00 ft.lb.s) j-(3,64 ft.lb.s) k COMPUTE = xm =: =xm = = x L = (2,8 i+2,3 j+k) x (1,2 i+2,8 j+1,4 k)

 $= \begin{vmatrix} 1 & 1 & K \\ 2.8 & 2.3 & 1 \\ 1.2 & 2.8 & 1.4 \end{vmatrix} = 0.420 i - 2.72 j + 5.08 k$

THUS: \(\begin{align*} \times \times

WHICH IS THE EXPRESSION OBTAINED FOR HO IN PROB. 14.13.

14.15 | GIVEN :

AS IT PASSES THROUGH O AT t=0, IT EXPLOYES INTO A (450 b), B (300 b), C (150 b)

AT t = 45, POSITIONS OF A AND B ARE A(3840ft, -960ft, -1920ft) B(6480ft, 1200ft, 2640ft)

FIND: POSITION OF C AT THAT TIME

MOTION OF MASS CENTER:

SINCE THERE IS NO EXTERNAL FURCE, = = Vot = (1200 ft/s) i (45) = (4000 ft) i

EQUATION (14.12)

 $m\bar{z} = \sum m_i \underline{z}_i$: $(9.00/g)(4800i) = (450/g)(3840i - 960j - 1920k) + (300/g)(6480i + 1200j + 2640k) + (150/g) \underline{z}_c$

 $150 \underline{t}_{c} = (900 \times 4800 - 450 \times 3840 - 300 \times 6480) \underline{i} + (450 \times 960 - 300 \times 1200) \underline{a} + (450 \times 1920 - 300 \times 2640) \underline{k}$ $= 648,000 \underline{i} + 72,000 \underline{j} + 72,000 \underline{k}$

== (4320 ft) i + (480 ft) j + (480 ft) k

14.16 GIVEN:

30-16 PASSES THROUGH O WITH VELUCITY U = (120 ft/s) WHEN IT EXPLUDES INTO FRAGMENTS A (1216) AND B (1816).

AT t = 3 5, POSITION OF A 15 A (300 ft, 24 ft, -48 ft).

FIND: POSITION OF B AT THAT TIME

ASSUME: ay = -g = - 32.2 ft/s2

MOTION OF MASS CENTER:

IT MOVES AS IF PROJECTILE HAD NOT EXPLODED.

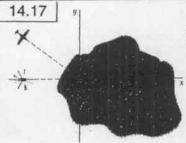
 $\overline{Z} = \sqrt{t} i - \frac{1}{2}g t^{2} j$ $= (120 \text{ ft/s})(3s)i - \frac{1}{2}(32.2 \text{ ft/s}^{2})(3s)^{2} j$ = (360 ft)i - (144.9 ft)j

EQUATION (14.12):

m== Zn; =:

 $m_{\underline{i}} = m_{\underline{A}} \stackrel{*}{=}_{\underline{A}} + m_{\underline{B}} \stackrel{*}{=}_{\underline{B}}$ $\frac{30}{9} (360 \stackrel{!}{=} -144, 9 \stackrel{!}{\underline{j}}) = \frac{12}{9} (300 \stackrel{!}{=} + 24 \stackrel{!}{\underline{j}} - 48 \stackrel{!}{\underline{k}}) + \frac{18}{9} \stackrel{*}{=}_{\underline{B}}$

2 = (400 ft) i - (258ft) j + (32,0H) K



GIVEN:
AIRPLANE: MA = 1500kg
HELICOPTER: MH = 3000 kg
COLLIDE AT 1200 M
ABOVE U.

4 MIN. BEFORE:
HELICOPTER WAS
B.4 KM WEST OF O;
PLANE WAS 16 km WEST

AND 12 km NORTH OF O.

AFTER COLLISION, HELICOPTER BREAKS INTO

H, (1000 kg) AND H2 (2000 kg)

FIND: POINT A WHERE WRECKAGE OF PLANE WILL BE FOUND, KNOWING THAT FRAGMENTS OF HELL. COPTER WERE AT H (500m, - 100m) AND H (600m, - 500m).

MOTION OF MASS CENTER G.

AT COLLISION: $\underline{v}_{H} = (\underline{6400m}) : = (35,00 \text{ m/s}) : \underline{i}$ $\underline{v}_{A} = ... \frac{(16000 \text{ m}) : - (12000 \text{ m}) : }{4(603)} = (66,67 \text{ m/s}) : - (50 \text{ m/s}) : \underline{i}$ VELOCITY OF HASS CENTER: $(m_{H} + m_{A}) : \underline{v}_{H} = m_{H} : \underline{v}_{H} + m_{A} : \underline{v}_{A}$ 4500 $: \underline{v}_{H} = 3000 (35,000 :) + 1500 (66,67 : -50 :)$ $: \underline{v}_{H} = (45,556 : - (16,667 : -50 :) : - (16,667 : -50 :)$

VERTICAL MOTION OF G: $k = \frac{1}{2}gt^2$ $t = \sqrt{\frac{2R}{g}} = \sqrt{\frac{2(1200m)}{9.81 \text{ m/s}^2}} = 15.6415$ POSITION OF G AT TIME OF GROUND IMPACT: $\overline{Z} = \overline{V}t = (45.556i - 16.667j)(15.641)$ $\overline{Z} = (717.55m)i - (260.69m)j$ (1)

FROM EQ. (14.12):

 $(m_H + m_A) \bar{\epsilon} = m_{H_1} \underline{\epsilon}_{H_1} + m_{H_2} \underline{\epsilon}_{H_2} + m_A \underline{\epsilon}_A$ (2)

4500 (712,55 = -260,69)=

 $|000(500\dot{i}-100\dot{j})+2000(600\dot{i}-500\dot{j})+1500\dot{z}_{A}|$ $|1.5\dot{z}_{A}=(4.5\times712.75-500-2\times600)\dot{i}+(-4.5\times260.69+100+2\times500)\dot{j}$ $\dot{z}_{A}=(1004m)\dot{i}-(48.7m)\dot{j}$

14.18 GIVEN: SAME AS FOR PROB. 14.17.

FIND: POINT WHERE FRAGMENT HZ WILL BE FOUND, KNOWING THAT WRECKAGE OF PLANE WAS FOUND AT A (1200 in, 80 m) AND FRAGMENT. H, AT H, (400 in, -200 m).

SEE SOLUTION OF PROB. 14.17 FOR DERIVATION OF & = (712.55 m) i - (260,69 m) j (1)

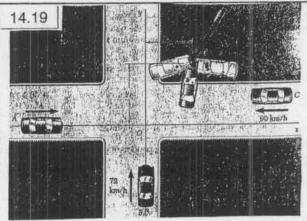
(mH+mA) = mH1 + mH2 = H2+ MH2 (2)

SUBSTITUTING DATA:

4500(712.55 i -260,69 i) = 1000(400 i -200 j)+2000 2H2+1500(1200 i +80 j)

22H2 = (4,5 × 712,55 - 400 - 1,5 × 1200) i+ (-4,5 × 260.69 + 200 - 1,5 × 80) j

1/4= (503 m) = - (547 m) j



GIVEN: CARS A (1500kg), B (1300kg), AND C (1200kg). WERE TRAVELING AS SHOWN WHEN CAR A HITS CARB. AT THAT INSTANT CAR C IS AT $x_C = 10 \, \text{m}$, $y_C = 3 \, \text{m}$. CAR C HITS A AND B, AND ALL CARS HIT $P(z_p, y_p)$. FIND: (a) TIME & FROM FIRST COLLISION TO STOPAT P. (b) SPEED V_A OF CAR A

KNOWING THAT Zp = 18 m, yp = 13.9 m

MOTION OF MASS CENTER

FINAL POSITION OF MASS CENTER OF SYSTEM IS THE SAME AS IF THE CARS HAD NOT COLLIDED AND HAD KEPT MOVING WITH THEIR ORIGINAL VELOCITIES,

 $(m_A + m_B + m_C) \stackrel{!}{=} p = m_A (v_A t) \stackrel{!}{i} + m_B (v_B t) \stackrel{!}{j} + m_C (z_C \stackrel{!}{i} + y_C \stackrel{!}{j} - v_C t \stackrel{!}{i})$

WHERE $U_B = 72 \text{ km/h} = 20 \text{ m/s}$, $U_C = 90 \text{ km/h} = 25 \text{ m/s}$ $4000 U_p = 1500 U_A t i + 1300(20 t) j + 1200(10 i + 3 j - 25 t i)$

 $z_p = (0.375 \, v_A - 7.5) \, t \, \underline{i} + 3 \, \underline{i} + 6.5 \, t \, \underline{j} + 0.9 \, \underline{j}$ THUS: $x_p = (0.375 \, v_A - 7.5) \, t + 3 \, \underline{j} + 6.5 \, t + 0.9$ (1)
(2) MAKINE $u_1 = 13.9 \, m : 13.9 = 6.5 \, t + 0.9$

(a) MAKING yp = 13,9 m: 13.9 = 6.5 t + 0.9 t = 2.00 s

(b) MAKING $x_p = 18 \text{ m}$ AND t = 25: $18 = (0.375 V_A - 7.5) 2 + 3$ $V_A = 40 \text{ m/s} = 144 \text{ km/h}$

14.20 GIVEN: SAME AS FOR PROB. 14.19.

FIND: COORDINATES OF POLE P, KNOWING THAT

VA = 129.6 km/h AND THAT TIME FROM FIRST

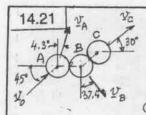
COLLISION TO STOP AT P 13 t= 2.45.

SEE SOLUTION OF PROB. 14.19 FOR DERIVATION OF $x_p = (0.375\sqrt[3]{4} - 7.5)t + 3$, $y_p = 6.5t + 0.9$ (1)

MAKING $V_A = 129.6 \text{ km/h} = 36 \text{ m/s}$ AND E = 2.4 s IN Eqs. (1):

 $x_p = (0.375 \times 36 - 7.5)(2.4) + 3 = 17.40m$ $y_p = 6.5(2.4) + 0.9 = 16.50 m$

2p=17.40m, yp=16.50m



GIVEN: 3 BALLS OF SAME MASS,
BALL A STRIFES B AND C
WHICH ARE AT REST.
BEFORE IMPACT, No= 12 ft/s
AFTER IMPACT, NC= 6.29 ft/s
EIND:
(a) VA, (b) VR AFTER IMPACT

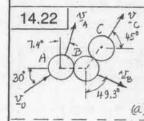
CONSERVATION OF LINEAR MOMENTUM

IN X DIRECTION: $m(12 ft/3)cos45 = mV_{A}sin43 + mV_{B}sin37,4 + m(6.29)cos30^{\circ}$ $0.07498 V_{A} + 0.60738 V_{B} = 3.0380$ (1)

IN 3 DIRECTION: $m(12 ft/s)sin45 = mV_{A}cos4.3 - mV_{B}cos37.4 + m(6.29) + 1n30^{\circ}$ $0.99719 V_{A} - 0.79441V_{B} = 5.3403$ (2)

(2)

(3) MULTIPLY (1) BY 0.79441, (2) BY 0.60738, AND ADD: $0.6652 + V_{B} = 5.6570$ $V_{A} = 8.50 ft/s$ (b) MULTIPLY (1) BY 0.99719, (2) BY -0.07498, AND ADD: $0.6652 + V_{B} = 2.6290$ $V_{B} = 3.95 ft/s$



EVEN 3 BALLS OF SAME MASS

BALL A STRIKES B AND C

WHICH ARE ATREST.

BEFORE IMPACT, Vo = 12 ft/s

AFTER IMPACT, Vc = 6.29 ft/s.

FIND:

(a) VA, (b) VB AFTER IMPACT.

CONSERVATION OF LINEFE MOMENTUM

IN X DIRECTION:

 $m(12 \text{ ft/s}]\cos 30 = mV_A \sin 7.4 + mV_B \sin 49.3 + m(6.29)\cos 45$ $0.12880 V_A + 0.75813 V_B = 5.9446$ (1)

IN 4 DIRECTION:

 $m(12 \text{ ft/s})\sin 30^\circ = m \, V_A \cos 7.4^\circ - m \, V_B \cos 49.3^\circ + m \, (6.29)\sin 45^\circ$ $0.99167 \, V_A - 0.652 10 \, V_B = 1.5523$ (2)

(a) MULTIPLY (1) BY 0.65210, (2) BY 0.75813, NMD ADD: $0.83581 V_A = 5.0533$ $V_A = 6.05 ft/s$

(b) MULTIPLY (1) BY 0. 99167, (2) BY - 0, 12880, AND ADD: 0.83581 √8 = 5.6951 · √8 = 6.81 ft/s ◆

GROUND WITH \$\frac{1}{2} = (10 \text{ m/s}) \frac{1}{2} IS HIT BY 50-9 ARROW WITH \$\frac{1}{2} = (60 \text{ m/s}) \frac{1}{2} + (80 \text{ m/s}) \frac{1}{2}.

FIND: DISTANCE FROM D UNDER POINT OF IMPACT TO P WHERE BIRD HITS THE GROUND.

CONSERVATION OF MUNIENTUM!

(3000g)(10ms) i + (50g)(60j + 80K) = (3050g) V

VELOCITY OF BIRD AND ALITY AFTER IMPACT:

V = (9.8361 m/s) i + (0.98361 m/s) j + (1.8115 m/s) L

VERTICAL MOTION:

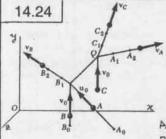
y=y+1/2t-1/2gt MAKE y=0: 0=15m+(0.9836111/2)t-1/2(9.81 11/2)t

(CONTINUED)

14.23 continued

t' - 0.20053t - 3.0581 = 0 $t = \frac{0.20053 + \sqrt{(0.20053)^2 + 4(3.0581)}}{2} = 1.8519s$ $+ \sqrt{(0.20053)^2 + 4(3.0581)} = 1.8519s$ $+ \sqrt{(0.20053)^2 + 4(3.0581)} = 1.8519s$

 $\frac{z}{p} = (\sqrt{2}, t) \cdot \dot{t} + (\sqrt{2}, t) \cdot \dot{k}$ $= (9.836)(1.8519) \cdot \dot{t} + (1.3115)(1.8519) \cdot \dot{k}$ $= \frac{z}{p} = (18.22 \text{ m}) \cdot \dot{t} + (2.43 \text{ m}) \dot{k}$



BEFORE COLLISIONS ALPHA
PARTICLE A MOVED WITH

L=-(480 M/s)+6003-640 E,

NUCLEI B AND C MOVED

WITH \$20 = (480 m/s) 3.

AFTER COLLISIONS, THEY

MOVED ALONG PATHS WHERE

A, (240, 220, 160), A, (320, 300, 200)

A, (240,220,160), A, (320, 300, 200)
B, (107,200,170), B, (74,270, 160)
C1(20,212,130), C, (200,260,115)

FIND: SPEED OF EACH PARTICLE AFTER COLLISIONS.

MASS OF OXYGEN NUCLEUS = M, HAS OF & PARTICE = 1/2 M
BEFORE COLLISIONS .

OX PARTICE: 41 = -480 i +600 j -640 k
NUCLEI B AND C: 1/2 = 480 j

AFTER COLLISIONS:

(DIMENSIONS IN MM)

 $\underline{V}_{A} = V_{A} \frac{A_{1}A_{2}}{A_{1}A_{2}} = \frac{90i+80j+40K}{120} V_{A} = (0.6667i+0.6667j+0.533k) V_{A}$ $\underline{V}_{B} = V_{B} \frac{B_{1}B_{2}}{B_{2}B_{3}} = \frac{-33i+70j-10k}{76.03} V_{B} = (-0.4229i+0.8971j-0.12816k) V_{B}$ $\underline{V}_{C} = V_{C} \frac{G_{1}C_{1}}{C_{1}C_{2}} = \frac{48j-15k}{50.29} V_{C} = (0.9545j-0.2983k)$

CONSERVATION OF MOMENTUM:

-120 L + 150 j - 160k + 960 j = (0.16 67 i + 0.1667 j + 0.08333 K) VA (-0.4229 i + 0.8971 j -0.12016 k) vz + (0.9545 j -0.2983 K) Vc

EQUATING THE COEFFICIENTS OF THE UNIT VECTORS:

0.1667 VA - 0.4229 VB = -120 (1)

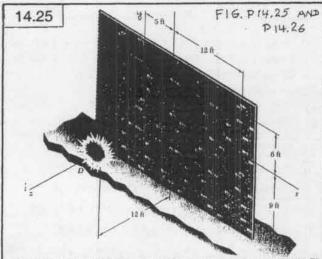
0. $1667 V_A + 0.8971 V_B + 0.9545 V_C = 1110$ (2) 0.08333 $V_A - 0.12816 V_B - 0.2983 V_C = -160$ (3)

MULTIPLY (2) BY 0.2983, (3) BY 0.9545 AND # DD : 0.12926 $V_{\overline{p}}$ + 0.14528 $V_{\overline{b}}$ = 178.39 (4)

MULTIPLY (1) BY 0.14528, (4) BY 0.4229 AND ADD. 0. $7888 V_A = .58.01$ $V_A = .735$, 4 m/s $V_A = .735 \text{ m/s}$

FROM (1): $0.1667(735,4) - 0.4229V_B = -120$ $V_B = 573,6 \text{ m/s}$ $V_B = 574 \text{ m/s}$

From (3): $0.08333(735.4) - 0.12816(573.6) - 0.2983 \sqrt{c} = -160$ 17 = 495.4m/s $\sqrt{c} = 495.m/s$



GIVEN: 12-16 SHELL EXPLODES ATD INTO FRAGMENTS, A(5/b), B(416), AND C(3 16), WHICH HIT WALL AS SHOWN. VELOCITY OF SHELL WAS Yo = (40 +45) i-(30 +1/5)j-(1200 +4/5) 1. FIND: SPEED OF EACH FRAGMENT.

CONSERVATION OF MOMENTUM!

(12/g) = (5/g) 1 + (4/g) 1 + (3/g) 10 12 (40 i-30) - 1200 k) = 5 (- 5 i - 12 t) VA + 4(31+101-31)のよう(-31-31-51)か

EQUATE COEFFICIENTS OF UNIT VECTORS! (1) (1) -25 VA + 3 VB

 $\frac{4}{3}V_B - \frac{9}{5}V_C = -360$ (8) B-63 VA-3 VB-13 VE = - 14,400 (3)

SOLUING THESE EQUATIONS SIMULTANEOUSLY: 5 = 1677.64, V = 1389,84, V = 1229,51

V = 1678 ft/s; V = 1390 ft/s; V = 1230 ft/s

SEE FIGURE AT TOP OF PAGE 14.26

GIVEN: 12-16 SHELL EXPLOYES AT DINTO FRAGIE A (416), B(316), AND C(516), WHICH HIT WALL HS CHOWN. VELOCITY OF SHELL WAS OF- (40ft/s) i- (30ft/s) j-(1200ft/s) k

FIND: SPEED OF EACH FRAGMENT

CONSERVATION OF MOMENTUAL! (12/g) 10 = (4/g) 1/A + (3/g) 1/8 + (5/g) 1/2 12 (40 i - 30 j - 1200k) = 4 (- 5 i - 12 k) VA + 3(31+31-31) 16+5(-31-5)

EQUATE COEFFICIENTS OF UNIT VECTORS:

(1) - 20 JA + 2VB

(2) VB -3 VC = -360 (F)

(E) - 48 VA - 2 VB - 4 VE = - 14, 400 (3)

SOLVING THESE EQUATIONS SIMULTANEOUSLY: Va= 2097.05, VB=1853.11, VC= 737.705

V = 2097 ft/s; V = 1853 ft/s; V = 738 ft/s

DERIVE H = E x m + H, WHERE 14.27 H = Z (zixmi Vi) (14.24) HG = Z(zi x mi vi) AND M = TOTAL MASS OF SYSTEM E = POSITION VECTUR OF G; T = VELUCITY OF G. MAKING \$ = 3+ 2; IN = Q. (14.7); H = Z (= + =;) x m; Y; = Ex Zmi Vi + Zz' xmi Vi BUT Im V = my

AND, BY (14.24): \(\Si\) \(\times\) = \(\text{H}_6\)

THEREFORE: $H_0 = E \times m v + H_0 \quad (Q.E.D.)$

DERIVE EM = HB (14,23) 14.28 (14,11) DIRECTLY FROM E W = H0 BY USING EQUATION DERIVED IN PROB. 14,27.

WE REDUCE THE FORCES TO THE VECTORS SHOWN. IT FOLLOWS THAT (1) IM = 2 X ZF+ ZM

FROM PROB 14,27: H = 5 × m + HG H = Exmy+Exmu+ HG = UX mV + 2 X + 10 + HG

BUT TXMV= O AND Ma= EF. THUS $H_g = E \times \Sigma F + H_G$ (2)

SUBSTITUTE FUR EMO FROM (1) AND H FROM (2) INTO (14.11):

TOX EF+ IMG = EXEF+ HG (Q.E.D.) ZMG = HG OR

14.29

GIVEN: NEVITATION PRAME OXYZ AND FRAME AZ'y'z' IN TRANSLATION W/R TO OXYZ. LET H' = [X m: T' (1) AND HA = \ 2 2 xm 1 1/2 (2) WHERE I' AND I DENVIT VELOCITIES WIR AZ "+

AND DAYS, RESPECTIVELY SHOW THAT HA = HA AT GIVEN INSTANT IF, AND ONLY IF, ONE OF THE FOLLOWING CONDITIONS IS SATISFIED AT THAT INSTANTS

(c) VA = O WITH RESPECT TO OZHZ,

(b) A COINCIDES WITH MASS CENTER G OF SYSTEM OF PARTICLES.

(c) In IS DIFFECTED ALONG AG.

(CONTILIED)

14.29 continued

WE RECALL:

 $\frac{H'_{A}}{H} = \sum \underline{v}_{i}^{i} \times m_{i} \underline{v}_{i}^{i} \qquad (1)$ $\frac{H}{A} = \sum \underline{v}_{i}^{i} \times m_{i} \underline{v}_{i}^{i} \qquad (2)$

LET U = " + T' IN EQ. (2):

 $H_{A} = \sum \underline{\xi}_{i}^{2} \times m_{i} (\underline{v}_{A} + \underline{v}_{i}^{2}) = (\sum m_{i} \underline{\xi}_{i}^{2}) \times \underline{v}_{A}^{2} + \sum \underline{\xi}_{i}^{2} \times m_{i} \underline{v}_{i}^{2}$ $BUT, BY (14.12): \sum m_{i} \underline{\xi}_{i}^{2} = m \underline{\xi}^{2} = m AG$ RECALLING EQ.(1), WE WRITE

Ha = mAGX YA + HA

THIS EQUATION REDUCES TO HA = H'A IF
(a) VA=0, (b) A=G, (c) VA // AG (Q.E.D.)

14.30 GIVEN:

FRAME AZYZ' IN TRANSLATION WITH RESPECT TO NEWTONIAN FRAME OZYZ.

LET HA = \(\frac{1}{2}\) xm\(\frac{1}{2}\)

WHERE \(\frac{1}{2}\) AND \(\frac{1}{2}\) ARE DEPINED WR FRANE AZY\(\frac{1}{2}\)

AND LET \(\frac{1}{2}\) BE THE SUM OF THE MOMENTS OF THE EXTERNAL FORCES ABOUT A.

SHOW THAT THE RELATION EM, = HA
IS VALID IF, AND ONLY IF, ONE OF THE FOLLOWING
CONDITIONS IS SATISFIED:

(b) A CONCIDES WITH MASS CENTER G OF SYSTEM OF PARTICLES,

(c) a 15 DIRECTED ALONG AG

DIFFERENTIATE EQ.(1): $H_A' = \sum_i \times m_i Y_i' + \sum_i \times m_i y_i$ $= \sum_i \times m_i Y_i' + \sum_i \times m_i a_i'$ BUT $Y_i^1 \times Y_i' = 0$ AND $a_i' = a_i - a_A$ THUS: $H_A' = \sum_i (k_i^1 \times m_i a_i) - (\sum_i m_i x_i^2) \times a_A$ BUT, BY (14. 12): $\sum_i m_i x_i' = m_i x_i' = m_i AG$ AND, SINCE a_i IS ACCELERATION WIR NEWTONIAN

PRAME, WE HAVE, BY EQ. (14.5), $\sum_i (x_i^1 \times m_i a_i) = \sum_i (x_i^1 \times F_i) = \sum_i M_A$ THEREFORE' $H_A = \sum_i M_A - m_i AG \times a_A$

THIS EQUATION REDUCES TO HA = EM IF

- (a) a= OSFRAME Ax'y'z' IS IN UNIPORM TRANSCHION WR NEWTONIAN FRAME Oxy 2 AND IS ITSELF A NEWTONIAN FRAME,
- (b) AG=0; A COINCIDES WITH G,
- (c) $\overrightarrow{AG} \times a = 0$; a_A is directed along AG

 (Q.E.D)

14.31 GIVEN: REFERRING TO PROB. 14,1,
ASSUNIE THAT

(1) 15-kg SUITCASE FIRST TOSSED WITH \(\frac{1}{2}\) = 3 m/s \(\rightarrow\)
(2) 20-kg SUITCASE THEN TOSSED WITH \(\frac{1}{2}\) = 2 m/s \(\rightarrow\)

(3) 25-kg CARRIER INITIALLY HT REST.

FIND: ENERGY LOST AS

(a) FIRST SUITCHSE HITS CARLIER

(b) SECOND SUITCASE HITS CARRIER

(a) BEFORE FIRST SUITCASE HITS CARKIER! $T_0 = \frac{1}{2} m V_0^2 = \frac{1}{2} (15 \text{ kg}) (3 \text{ m/s})^2 = 67,50 \text{ J}$

FIRST IMPACT: CONSERVATION OF MORE 117-111 (15 kg)(3 m/s) = (25+15) V_1 $V_2 = 1.125$ m/s $V_1 = \frac{1}{2}(25 \text{kg} + 15 \text{kg})(1.125 \text{m/s})^2 = 25.313 \text{J}$ $V_2 = 1.125$ $V_3 = 1.125$ $V_4 = 1.125$ $V_5 = 1.125$ $V_6 = 1.125$ $V_7 = 1$

(b) JUST BEFORE SECOND SUITCASE HITS: T' = T, + \frac{1}{2}(20 kg)(2 m/s) = 25.313 J + 40 J = 65,313 J

5ECOND IMPACT: CONSERVATION OF MENENTUMY (25kg+15kg)(1,125m/s)+(20kg)(2m/s)=(60kg) V2 V=1,4167m/s

 $T_2 = \frac{1}{2} (60 \text{ kg}) (1.4167 \text{ m/s})^2 = 60.208 \text{ J}$ ENLOST = $T_1^2 - T_2^2 = 65.313 \text{ J} - 60.208 \text{ J} = 5.10 \text{ J}$

14.32 GIVEN; COLLISIONS DESCRIPED
IN PROB. 14.5. WE RECAL! THAT
INITIAL VELOCITY OF CAR A WAS $\sqrt{n} = 1.920$ m/s
AFTER A HITS B: $(V_B)_1 = 1.680$ m/s
AFTER B HITS C! $(V_B)_2 = 0.210$ m/s
AFTER A AGAIN HITS B: $(V_B)_3 = 0.23625$ m/s
HASS OF EACH CAR = 1500 kg

FIND! ENERGY LOST HETER ALL COLLISIONS HAVE TAKEN PLACE.

PROM SOLUTION OF PRIR. 14.5 WE HAVE THE FOLLOWING FINAL VELOCITIES! $V_A = 0.21375 \text{ m/s}$, $V_B = 0.23625 \text{ m/s}$, $V_C = 1.470 \text{ m/s}$

INITIAL ENERGY:

To = 1/2 m1 V02 = 1/2 (1500kg)(1,920 m/s)=2764.8J

FINAL ENERGY: $T_{4} = \frac{1}{2}mV_{A}^{2} + \frac{1}{2}mv_{B}^{2} + \frac{1}{2}mv_{C}^{2} = \frac{1}{2}m(V_{A}^{2} + V_{B}^{2} + V_{C}^{2})$ $= \frac{1}{2}(1500 \text{ kg})[(0.21375 \text{ m/s})^{2} + (0.23625 \text{ N/s})^{2} + (1.470 \text{ m/s})^{2}]$ = 1696.8 J

 $\frac{ENERGY\ LOST}{=7,-7,=2764.8\ J-1696.EJ}=1068\ J$

14.33 | GIVEN:

14.3 JUMP FROM SAME END OF 300-16 BOAT WITH VELOCITY OF 16 H/s WITH RESPECT TO BOAT.

WORK DONE BY WOMAN AND BY MAN IF WOMAN DIVES FIRST

TOTAL KE, AFTER WOMAN DIVES FROM PART a OF SOLUTION OF PROB. 14.3: VEL. OF BOAT = $(\vec{V_B})_1 = 3.20 + t/s$ THUS, VEL. OF WOMAN = $(\vec{V_W})_1 = 16 - 3.20 = 12.8 + t/s$ K.E. = $T_1 = \frac{1}{2} m_W^2 (\vec{V_W})_1^2 + \frac{1}{2} (m_B + m_M) (\vec{V_B})_1^2$

 $T_1 = \frac{1}{2} \frac{(R0)}{322} (12.8)^2 + \frac{1}{2} \frac{480}{32.2} (3.20)^2 = 381.61 + 1.16$ WORK OF WOMAN = $T_1 = 902 + 1.16$

TOTAL K.E. AFTER MAN DIVE

FROM ANSWER TO PART & OF PROR. 14. 3:

VEL. OF BOAT = $(V_B)_L = 9.20 \text{ ft/s}$ THUS, VEL. OF MAN = $(V_M)_2 = 16 - 9.20 = 6.80 \text{ ft/s}$ K.E. = $T_z = \frac{1}{2} m_W (V_W)_1^2 + \frac{1}{6} m_M (V_W)_2^2 + \frac{1}{7} m_B (V_B)_2^2$ = $\frac{1}{2} \frac{120}{32.1} (17.8)^2 + \frac{1}{2} \frac{180}{32.2} (6.80)^2 + \frac{1}{2} \frac{300}{32.2} (9.20)^2$ $T_z = 828.82 \text{ ft.} B$

WORK OF MAN = T-T = 828.82 - 381.61 = 447 ft. 16 ◀

14.34 GIVEN:

BULLET OF PROB. 14,7 FIRED WITH $V_0 = 1500$ H/s THROUGH 6-16 BLOCK A BECOMES EMPEDDED IN 495-16 BLOCK B. BLOCKS MOVE WITH $V_A = 5$ ft/s and $V_B = 9$ ft/s. T

ENERGY LOST AS BULLET

(a) PASSES THROUGH PLOCK A

(b) BECOMES EMBEDDED IN BLOCK B

FROM ANSWERS TO FROB. 14.7: WEIGHT OF EULLET = W = 0,800 of = 0,0500 14 VEL. OF BULLET BETWEEN BLOCKS = V = 900 H/s

(a) ENERGY LOST AS BULLET PASSES THROUGH A
INITIAL K.E. = To = \frac{1}{2} \frac{w}{3} \tau_0^2 = \frac{1}{2} \frac{0.10500 \text{ b}}{32.2 \text{ ft/s}^2} (1500 \text{ ft/s})^2

T = 1746.89 \text{ ft. /b}

K.E. OF SYSTEM AFTER BULLET PASSES THROUGH H $= T_1 = \frac{1}{2} \frac{W}{3!} \cdot U_1^2 + \frac{1}{2} \frac{W}{3!} \cdot U_4^2 = \frac{1}{2} \frac{0.0500}{32.2} (900)^2 + \frac{1}{2} \frac{6}{32.2} (5)^2$ $T_1 = 628.88 + 2.33 = 631.21 + 1.16$

EN. LOST = 75-7, = 1746,89-631,21 = 1116 ft.16 ◀

(b) ENTRGY LOST AS FULLET LEWITES EMBEDDED IN B FINAL KE. = $T_z = \frac{1}{2} \frac{WA}{g} v_\mu^2 + \frac{1}{2} \frac{(W_B + v_\tau)}{g} v_B^2$ $T_z = \frac{1}{2} \frac{6}{322} (5)^2 + \frac{1}{2} \frac{11.95 + 0.05}{32.2} (9)^2 = 8.616 \text{ ft/b}$

EN. LOST = 7- T = 631,21-8.618 = 623 ft. 16

THE ASSUME PLASTIC IMPACT AND THAT ENERGY ABSORBED BY EACH AUTOMOBILE EQUALS ITS K.E. WITH RESPECT TO HOVING FRAME ATTACHED TO MASS CENTER OF SYSTEM



(a) SHOW THAT EA/EB = m_B/m_B , WHERE (EA AND EB ARE ENERGIES ABSORBED BY A AND B. (b) FIND EA AND EB IF m_A = 1600 kg, m_B = 900 kg, V_A = 90 km/h, V_B = 60 km/h.

BEFORE COLLISION; VEUXITY TO OF MASS CENTER G:

(MA+MB) T = MAVA - MBVB

T = MAVA - MBVB

MA+MB

MOTION OF AUTOS RELATIVE TO G: $V_{A/G} = V_A - \bar{v} = V_A - \frac{m_B V_B - m_B V_B}{m_{A+m_B}} = \frac{m_A V_A - m_B V_B}{m_{A+m_B}} = \frac{1}{Z} \frac{m_A v_B}{(m_A + m_B)^2} (V_A + V_B)^2$ (1)

 $T_{B/6} = \frac{1}{2} m_B V_{B/6}^2 = \frac{1}{2} \frac{m_A^2 m_B}{(m_A + m_B)^2} (V_A + V_B)^2$ (2)

AFTER COLLISION.

SINCE THERE IS NO EXTERNAL PORCE, G KEEPS
MOVING WITH VELOCITY TO.

SINCE IMPACT IS PLASTIC! TO = T = T

AND TAGE = Yag = 0. THUS: TAGE TO BE

IT POLLOWS THAT F. = TAGE AND EB = TBGE

IT POLLOWS THAT EA = TA/6 AND EB = TB/Q
(a) DIVIDING (1) BY (2):

 $\frac{E_0}{T_0} = \frac{T_{A/b}}{T_{B/6}} = \frac{m_B}{m_A} \qquad (O. E. D)$

(b) SUBSTITUTING IN (1) AND (2) THE GIVEN DATA, $M_A = 1600 \text{kg}$, $M_B = 900 \text{kg}$, $V_A = 90 \text{km/h} = 25 \text{m/s}$, $V_B = 60 \text{km/h} = 16.61 \text{m/s}$ WE FIND $E_A = 180.0 \text{kJ}$, $E_B = 320 \text{kJ}$

14.36 GIVEN: CAR COLLISION OF PROB. 14.35 DEFINE: SEVER MY OF A COLLISION = E/ED

WHERE E = ENERGY ABSORBED BY CAR IN EULLISION,

AND E = EN. ABSORBED BY SAME CAR IN A TEST

WHERE IT HITS AN IMMOVABLE WALL WITH VELOC. VO

SHOW THAT COLLISION OF PROB. 14.35 IS (MA/MB)

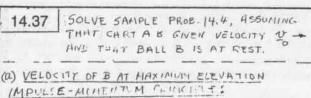
TIMES MORE SEVERE FOR CAR B THAN FOR CAR A.

ENERGIES ABSORBED IN TESTS OF A AND B

 $(E_A)_0 = \frac{1}{2} m_A V_0^2$ $(E_B)_0 = \frac{1}{2} m_B V_0^2$ (3 SEVERITY OF COLLISION FOR CAR A = $E_A/(E_A)_0$ SEVERITY OF COLLISION FOR CAR B = $E_B/(E_B)_0$

RECALLING EQS. (3) AND FROM PROB. 14:35 THAT EA/IB = MB/M, WE HAVE

SEVERITY OF COLL. FOR B = $\frac{E_B(E_A)_0}{E_A(E_B)_0}$ = $\frac{m_A}{m_B} \frac{\frac{1}{2}m_A v_0^2}{\frac{1}{2}m_B v_0^2} = \left(\frac{m_A}{m_B}\right)^2$ (Q.E.D.)



WE NOTE THAP WHEN B REACHES HAS MUM HEIGHT $(\mathcal{V}_{\mathcal{B}})_2 = (\mathcal{T}_{\mathcal{A}})_2$

Emy + EExt Imping = Emy IN X COMP : 11/A V = (MA + ME)(VA) $(V_B)_z = (V_A)_z = \frac{m_A}{m_A + m_B} V_0 \rightarrow (1)$

(b) MAXIMUM HEIGHT REACHED BY B CONSERVATION AFENELGY

A TOTAL VI = MAGE TI = IMAV Vz = mag t. + magh T = 1 (MA + 1/E) (VA) 2 T,+V,=T2+V2:

1 mar + mage = 1 (mu+ n B) (4) 2+ mage + megh SUBSTITUTING FOR $(V_A)_Z$ FROM (1): $\frac{1}{Z} m_A V_0^{-2} = \frac{1}{Z} \frac{m_A}{m_A + m_B} V_0^{-2} + m_B g R$ $R = \frac{V_0}{J} \frac{2 m_A + m_B m_B - y A^2}{(m_A + m_B) m_B}, \quad R = \frac{m_A}{m_A + m_B} \frac{V_0^2}{2g}$ (SAME ANSWER AS FOR PART & OF SP14.4)

14.38

GIVEN: BALL A HITS WITH 15 BALLS B AND C WHICH ARE AT REST. ASSUME CONSERVATION OF ENERGY.

(CONTINUED)

FIND: FINAL VELOCITY OF EACH BALL, IF (a) A STRIKES B AND C SIMULTANEOUSLY, (b) H HITS & BEFORE !T HITS C

(a) A STRIKES B AND C SIMULTANETUELY

myB CONSERVATION T 0=30° OF MOMENTUM + 9 y COMP: 0 = 111's sin 30" $\sin \theta = \frac{2}{2^{\pm}} = 0.5$ $\theta = 30$ - mive sinso で= 10

1 Z COMP: MUD= MUA +2mu Cos 30, 15-15=V 13

CONSTRUATION OF ENERGY: Jo- JA = 22E 1 m V = 1 m 1 2 + 1 (2 m 1 B) 1 Vo + VA = 2 VB DIVIDE (2) BY (1):

ADD (1) AND (3): $2V_0 = \frac{2+3}{5}$ V_0 FROM (1): $V_A = V_0 - \frac{6}{5} V_0 - \frac{3}{5} V_0/5$ ひまるはるい VA = 0,200 Vo +; VB = 0.693 Vo = 30°; 1-0.693 Vo 30°

(b) A HITS B BEFORE C 14.38 continued $m \underline{V}_{\rm p}$

CONS. OF MOMENTUM FROM 1 TO'R: (UA)= 0- Ugas30 to 26 COMP.: MV = m (VA) + mVg cos30 (VA) = VE sin30" 4 g core: 0 = moraly +m upsinsu" SQUALE BOTH MEMBERS OF (4) AND (5) AND ADD: (6)

CONS. OF ENERGY FRUM I TO 2! VH= V0-VE 1 1115 = 1 migrat + 1 mit CARRYING IN TO (6) AND SOLVING FOR UB: UB = V COS 30 (7)

CONS. OF NOMENTUM FROM 1 703: \$ 2 (01) F: m = m(TA) + m vg cos 30"+m vccos 30" +Ay cours: 0 = m(vA) + m = sin30 - mvcsin30" (VA) = TO SIN 30 - V COSSO (B) (VA)y = - Vo sin 30 cos 30 + VC sin 30 SWUARING AND ADDINE: 12 = 0.25 15-0.866 16 16 + 16 (10)

CONS. OF ENERGY FLON 1 TO 3: されできまかいよ+まかできまかいと Va=tra+VB+Vc SUBSTITUTE FOR TA AND US FROM (10) AND (7) INTO (1): 30 = 0.35 Vo - 0.866 Vo Vc + Vc + 0.75 Vo + Vc , V= 0.433 Vo CARRYING 1273 (8) AND (7): (U,1) = 0.2510 - 0.433 (0530° = - 0.1250 Vã (+ A) = -1.433 70 + 0,435 % sin 30 = -0.2165 Vo

THUS: 1-0,250 N, T 60; UB= 0,866 V, 230; U=0,433 V6 300



GIVEN: A HITS B WITH V = 15ft/s, ASSUME CONS. OF ENERGY FIND: MAGNITUDES OF TA, UB, AND UC.

CONS. OF MEMENTUM! to a comp. m Vo cos45"- m VB singuit m Vcas 30" (1) + [y aup: 111 , 5 , 1145" = m VA - mVB cos30 + mVES (1130" . (2) MULTIPLY (1) BY SIN 30°, (2) BY COS 30°, SUBTRACT, AND

SOLVE FOR UB: UB = 0.8660 VA - 0.2588 VO (3) CARRY INTO (1) AND SOLVE FOR VC

UC = 0,8165 U - 0.57735(0.8660 VA - 0,2588 VO) VC=-0.5VA+0.9659 V (4) CONS. OF ENERGY!

1 - n vo = 1 m va+ 1 m va + 1 m vc, vo=va+va+vc SUBSTITUTE FOR UB AND VC FROM (3) AND (4). To = UA + (0.8660 VA - 0.25 88 VD) + (-0,5 VA +0.1659 VD) 2VA - 1,4141 VOVA = 0 VA = 0.7071 VO TRUM (3) ANA(4): UB= 0,3536 VO, V= =0.6124 VO GIVEN DATA: 1/0 = 15 +t/s. THEREFURE;

VA = 10.61 ft/s; VB = 5,30 ft/s; VC = 9.19 ft/s



GIVEN:

A HITS B WITH U = 15ft/s ASSUME CONSERVATION DF ENERGY. .

FIND: MAGNITUDES OF VA, VB, AND VC.

CONS, OF MOMENTUM!

#2 com P: m v = cos 30" = m v = sin45" + m v = cos 45" (1)

+Py com P: m v = sin30" = m v = - 111 v = cos 45" + m v = sin45" (2)

SUBTRACT (2) FROM (1, AND DIVIDE BY MI

1/8 = 0.707/1/4+0.25881 (3) 0.3660 Vo = - VA +1.4142 VB ADD (1) AND (2) AND DNIDE BY MI

1,3660 Vo = VA + 1.4142 2 (Vc = -0.7071 VA + 0.9659 V (4)

CONS. OF ENERGY:

15 = UA+ UB+ VC = 1 11 V2 = 1 m VA + 1 n1 V2+ 2 m V SUBSTITUTE FOR UB HAD VE FROM (3) AND (4): Vo = VA+(0.7071VA+0.258815)+(-0.7071VA+0.9659V5)

2252 - VO VA =0 Un = 0.516 FECH(2) AND (4): 1 = 0.6124 Jo, Vc=0.6124 Vo

GIVEN DATA: 10 = 15 H/s. THEREFORE: U_=7.50 ft/s; VB= 9.19 ft/s; VC = 9.19 ft/s

14.41

GIVEN: Vo = 8 m/s. POTENTIAL ENERGY OF SPRING = 120J. CORD CUT WHEN B = 30.

FIND: MA AHL MAFTER CORD IS CUT.

CONS, OF LINEAR MOM. WR FRAME GZ'Y'E



CONS. OF ENERGY W/R FRAME GZ'TZ'; 120J= 120J= 1 (2.5) VA + 1 (1.5) Vp2

SUBSTITUTE FOR (B FRUM (1) INTO (2): 5 VA + 3 (5 VA) = 480 VA=61/520 FROM (1): UB'= 5 (6 m/s) T' = 10 11/3 50

WITH RESPECT TO FIXED FRAME OXYES V = V, + VA = 8 M/. → + 6 m/s 20 (3)

 $\underline{V}_B = \underline{v}_b + \underline{v}_B' = B m/s \rightarrow +10 m/s \sqrt{5} b$ (4)

FOR 0-30: Eu.(3):

\$ 7 COMP: (TA) = 8 - 6 COS30 = 2.804 m/s

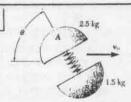
+ y con: (1"A) = 6 sin 30" = 3 11/6

V= 4.11 m/3 & 46.90

+ x comp; (CB) = 8+10 cos 30° = 16,660 m/s 15 y comp: (Te)y = - 10511130 = - 5 m/s

V = 17,39 m/s \$ 16.7°

14.42



GIVEN: Up = 8 m/s. POTENTIAL ENERGY OF SPRING = 120J. CORD CUT WHEN 0= 120.

UA AND UB AFTER THE CORD IS CUT.

(3")

(41)

SEE SOLUTION OF PROB. 14,41 FOR DERIVATION OF ENC. (3) AND (4). WITH 0 = 120", WE HAVE

UA= U+ VA = 8 m/s + 6 m/s & 60

UB = U + UB = 8 m/s + 10 m/s 7 60°

EQ:(3'): X COMP: (VA) = 8+6 COS 60° = 8+3= 11 m/s + 14 COMP: (VA) = 65in 60° = 5,196 m/s

Va= 12,17m/s & 25,3° EQ. (4'): 1, X COMP; (VB) = 8-10 COS 60° = 8-5 = 3 m/s + 9 y comp: (VB) = -10 sin 60 = - 8.660 m/s

V = 9.17 m/s \$ 70.9°

14.43

GIVEN:

THREE SPHERES EACH OF MASS M. A AND B ARE CONNECTED BY TAUT, INEXTENSIBLE CORD. C STRIKES B AS SHOWN. ASSUME CONS. OF ENERGY. FIND:

VELOCITY OF EACH SPHERE AFTER IMPACT.

EFFECT ON CONSTRAINTS ON FINAL VELOCITIES

U = VA 260

BECAUSE CORD AB IS INEXTENSIBLE COMPONENT OF 1 A LONG AB MUST BE EWUAL TO YA.

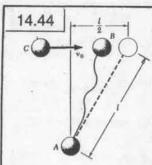
U = VA 160+ VB/A530 (1) CONS. OF MOMENTUM FOR SYSTEM;

my=my+2my+myBA H y COMP; D = 2mvAsin60-mVBAsin30 VB/A = 2 13 VA \$ 2 COMP: mv = mv + 2 mv cos 60 + mv 8/4 cos 30

DIVIDING BY M AND SUBSTITUTING FOR VB/4 FROM (2): No= VC + VA + V3 (2 V3 VA) 1 = V - 4 VA

CONS. OF ENERGY: $\frac{1}{2}mV_0^2 = \frac{1}{2}mV_A^2 + \frac{1}{2}mV_B^2 + \frac{1}{2}mV_C^2,$ 0=20A+VB/A+Vc (4) SUBSTITUTE FOR UB/A AND UC FROM (?) AND (3) INTO (4): X=2V2+12V4+ 00-8 VOVA+16VA UA = 15 V & 60° -c = 15 Vo

FROM (3): U= U - 16 V0 = -15 VO FROM(1) AND(2): \$2 = 40 \$60 + 813 \$0 \$30, \$2 = 0.96 10 \$13.90



GIVEN:

THREE SPHERES, EACH OF MASS m. A AND B ARE CONNECTED BY INEXTENSIBLE CORD WHICH IS SLACK, C STRIKES B AS SHOWN WITH PERFECTLY PLASTIC IMPACT. FIND!

(a) VELOCITY OF EACH SPHERE AFTER CORD BECOMES TAUT. (b) PRACTION OF INITIAL K.E. LOST WHEN CORD BECOHES TAUT.

(a) DETERMINATION OF VELOCITIES

IMPACT OF C AND B かか=かか+かり, サーザーショ

CONS. OF ENERGY (PERFECTLY ELASTIC IMPACT):

1 mv== 1 mv=+ 1 mv= SQUARE (1): V2+2V6 V1+V1= V0

SUBTRACT (2): 2 1/2 1/1 = 0

V = U CORRESPONDS TO INITIAL CONDITIONS AND SHOULD BE ELIMINATED. THEREFORE

FROM (1): U, = V

CORD AB BECOMES TAUT

BECAUSE CORD IS TNEXTENSIBLE, COMPONENT OF UB ALONG AB MUST BE EQUAL TO VA. CONS, OF MOMENTUM!

mv = 2mv + m VB/A ty comp; 0= 2m va sin60 - mvB/A 51n 30 (3) VEVA = 2 V3 VA

=> I COMP: MV = 2 M VA COS 60° + m VB/A COB 30° DIVIDING BY M AND SUBSTITUTING FOR UB/A FROM (3):

Up = 2 UA (0,5)+(2V3 VA)(V3/2) 10 = 4 VA VA = 0,250 VO

V = 0,250 V \$60 -

CARRYING INTO (3): VB/A = 2 J3 (0,250 V6) = 0.866 V6 1 2 A = 0.250 V 30 THOS: UB = VA + VB/A

= 0.250 V \$66+0,866 V 30 128/A=0.866 % V= (0.250 V, car60" + 0,866 V, coo 30") i + (0,250 vo sin60°- 0.866 tosin30") j

V= 0.875 0, 1-0.2165 } VB=0,901 5 13.9 1 B = 0, 4012915 1 13.90

(b) FRACTION OF K.E. LOST

ての=台州では TFINAL = 1 M VA2 + 2m VB2 + 2 m VC = $\frac{1}{2} m (0.250 \sqrt{0})^2 + \frac{1}{2} (0.90139 \sqrt{0})^2 + \frac{1}{2} m (0)$ = = = 10 (0.875) 15 K.E. LOST = To-TFINAL = 2 m (1-0.875) 5 = 1 1 my FRACTION OF K.E. LOST = A

GIVEN: 14.45

360-kg SPACE VEHICLE WITH 1 = (450 Mak, AS IT PHSSES THROUGH D, EXPLOSIVE CHARGES SEPARATE IT INTO 3 PARTS: A(60kg), B(120kg), AND C(180kg). SHORTLY INTTER, THE POSITIONS OF THE 3 PARTS ARE A (72m,72m,648m), B(180m,396m,972m), C(-144m,-288m,576m). VELOCITY OF B IS UB=(150 m/s) i+(330 m/s) j+(660 m/s) k. X - COMP OF VELOCITY OF C 15 (VC) = - 120 m/s. FIND: VELOCITY OF A.

CONSERVATION OF AIVGULAR MOMENTUM ABOUT O SINCE VEHICLE PASSES THROUGH O, HO = 0, OR

Ho= 1/4 × m, VA + 28 × m, VB + 2, × m c Vc = 0 USING DETERMINANT FORM:

1 = 1 = 1 H = 60 72 72 648 + 120 180 396 912 + 180 - 144 - 288 576 -0 (24) (VA) (VA) -120 (VZ), (VZ) 150 330 660

EQUATING TO ZERO THE COEFF. OF i, j, k, AND DIVIDING BY 60:

(1) 72(VA) = 648(VA)y - 1188×103-864(VC) - 1728(VC)4 = 0

(1) 648(VA)2-72(VA)2+54.0×103-207.36×103+432 (VE)2=0

(E) 72 (VA)y-72(VA)2 +0-452 (VC)y-103.68×103=0 OR, AFTER REDUCTIONS:

(VA) = 9(VA) = 12(VC) = 24(VC) = 1650 (1)

 $-(V_A)_2 + 9(V_A)_2 + 6(V_C)_2 = 2130$ (2)

(VA)y - (VA)2 - 6 (VC)4 = 1440 (3)

CONSERVATION OF LINEAR MOMENTUM

my = may + may + mc Ve

360(450 k)=60[(0A) i+(VA) j+(VA) j+120[170i+330j+660k]+ 180[-120i+(vc), 1+(vc), b]

EWVATING THE COFFE OF THE UNIT VECTURS AND DIVIDINGBY 60: (UA) = 60 m/s $(4)(V_{H})_{x} + 300 - 360 = 0$

 $(V_A)_g = -660 - 3(V_A)_g$ (5) (1) (VA) +660 +3 (VC) =0

(b) (VA)2+1320+3(VC)2=2700 (VA)=1380-3(VC)2

SUBSTITUTING FROM (4), (5), (6) IN TO (2) AND (3): - 1380 +3 (Vc) +9(60)+6(Vc) = 2130 (UC) = 930 m/s

- 660-3 (Vc) + -60-6 (Vc) +=1440 (Vc) =-240 m/s SUBSTITUTING FOR (VC), AND (VC), INTO (5) AND (6):

(VA) = -660-3(-240) = 60 m/s (VA)2= 1380 -3 (330) = 390 m/s

RECALLING FROM (4) THAT (VA) = . 60 m/s, WE HAVE

= (60.0 m/s) i + (60.0 m/s) j+(390 m/s) k

CHECK

SINCE EQ. (1) WAS NOT USED IN OUR SOLUTION, WE CAN USE IT TO CHECK THE ANSWER. SUBSTITUTING THE VALUES OBTAINED FOR (VA)2, (VA)y, (VC)2, AND (NO)4 INTO THE LEFT. HAND MEMBER OF EQ.(1), WE OBTAIN 390-9(60)-12(330)-24(-240)=

390-540-3960+5760=1650

GIVEN: 14.46

IN SCATTERING EXPERIMENT OF PROB. 14.24 IT IS KNOWN THAT PARTICE A IS PROJECTED PROM An (260, -20,340) AND COLLIDES WITH C AT Q (200,180, 140)

COURDINATES OF BO WHERE PATH OF BINTERSECTS 27 PLANE

CONS. OF ANGULAR MOMENTUM ABOUT Q: SINCE PATHS OF A HETER COLLISIONS AND OF C BEFORE AND AFTER COLLISION PASS THROUGH Q THE CORRESPONDING ANG, MOMENTA ARE ZERO (FIG. P14.24). CONS. OF ANGULAR MOMENTUM OF ALL PARTICLES ABOUT W. IS EXPRESSED AS

$$Q A_0 \times \frac{m}{4} \frac{a_0}{b} + Q B_0 \times m v_0 = Q B_1 \times m v_B$$
 (1)

WHERE QA0= 1 - 16 = (260 i-20 j+340 k)-(200 i+180 j+140 k) =60i-200j+200k

QB=(A2)1+(Ay); +(AZ)x

QB,= 2B,-20=(1071+200+1701)-(2001+180+1401) =-93i+20j+30k

40 = -480i+600g-640k/ AND, FROM SOLUTION OF PROB. 14.24i

VB=VB 28= 573.6(-0.4229i+0.8971j-0.12816k) = -242,62 + 514.6 } -73.51 K

SUBSTITUTING INTO (1) AND USING DETERMINANTS!

1 2 3 1 2 3 K1 m 60-200 200 + m Dz By DZ = m -93 20 -480 600 -640 0 480 0 -2426 514.6 -73.51

EQUATING THE COEFFICIENTS OF & AND KS (E) 1/4 (8000) - 480 07 = -16908 12=3939mm

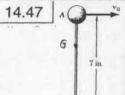
1 (-60000) + 480 AX = - 43006 AX=-58,35mn

x 8= x a+ bx = 200-58,4

23, = 141,6 mm

ZB0 = Z0+DZ=140+39,4

28, = 179,4 mm



GIVEN: 5-16 SPHERE A AND 2-16 SPHERE B CONNECTED BY RIGID ROD REST ON HORIZON TAL, FRICTION (ESS SUPTACE WHEN A IS GIVEN VELOCITY 15 = (10.5 ft/s) i.

(a) LINEAR MOM. AND ANG. MCM. HG (b) VA AND VE AFTER 180 ROTATION.

POSITION OF MASS CENTER AG+ BG = 7 in. AG (516) = BG(216), BG = 2.5 AG 3,5A6=7, A6=2 in.

(a) LINEAR AND A NG. MOMENTUM.

L=may = 516 (10,5 ft/s) i=(1.6304 16.5)i L=(1.630 16.5) 4

H = GA x m N = (2 in.) 1 x (1.6304 /b.5)1 =-(3.2669 in. 16.5) k =-(0.27174 ft.16.5) K H = - (0.272 ft . 16.5) k

(CONTINUED)

14.47 continued

(b) VELOCITIES OF A AND BAFTER 180 ROTATION

CONS. OF LINEAR MOMENTUM: MAV = MAVA+MBVB. (5/9)(10,5) = (5/9) V + (2/9) V = 50A+20 = 52,5 CONS. OF ANG. MOM, ABOUT G' to tAMAU = - EAMAU+ EBMBUR (2 in. 15/g)(10.5)=-(Zin.)(新)v/+(5in)(2/g) v/+

MULTIPLY BY & AND DIVIDE BY 2: -50 +50B = 52.5 ADD (1) AND (2): 70 = 105 DB=+15,00 ft/s. FROM (1): 5VA+2(15)=52.5 VA = + 4,50 ft/s

U'_A = (4.50 ft/s) 1 ; U' = (15.00 ft/s) 1

14.48

GIVEN: 5-16 SPHERE A AND 2-16 SPHERE B CONNECTED BY RIGID ROD REST ON HORIZONTAL, FRICTIONLES SURFACE WHEN B IS GIVEN VELOCITY U = (10.5 H/s) i (a) LINEAR MOM, AND ANG, MOM, H. (b) VAND VB AFTER 180° ROTATION.

POSITION OF HASS CENTER A6 + BG = 7 in. AG (516) = 3G(26), B6 = 2.5AG, 3.5AG=7, AG=2 in BG-5 in

(a) LINEAR AND ANG. MOMENTUM

 $L = \mathcal{W}_B \stackrel{V}{\sim} = \frac{2/b}{32.5 \text{ ft/s}^2} (10.5 \text{ ft/s})^2 = (0.6522 / b.s)^2$ L=(0,65216.5)i

H=GBXMB V=- (51 n.) j x (0.6522 16.5) i =+(3.261 in. 16.5) K

H =+ (0.272 ft.165)k

(6) VELOCITIES OF A AND B AFTER 180° ROTATION.

CONS. OF LINEAR MOMENTUM Mav mov = mava+mevB 4 (2/g)(10.5)=(5/g) VA + (2/g) OB TG 七日十 50 +2VB = 21 CONS. OF ANG. MOM, ABOUT G: MAVA+9 EBMBO = EMAVA-EBMBOB

(5 in.)(2/g)(10.5) = (21n.)(5/g) VA - (5 in.)(2/g) VB

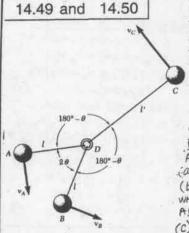
MULTIPLY BY & AND DIVIDE BY 2:

5 VA - 5 VB = 52,5 (2)

SUBTRACT (2) FROM (1): 1 =- 4.50 ft/c 70% = -31,5

FRON(1): 5VA +2 (-4.50) = 21 Va = + 6.00 ft/s

1 = (6.00 ft/s) 1; v'= - (4.50 ft/s) i



GIVEN:

THREE IDENTICAL SPHERS
CONNECTED TO RING D AT
THEIR MASS CENTER
SLIDE ON HORIZONTAL,
PRICTIONLESS SURFACE
(0'= 2 0 cos 0)

VA = NB = ND WHEN
CORD CD BREAKS.
FIND APTER CORDS
AD AND BD BECOME TAUT

(a) SPEED OF RING D

(b) RELATIVE SPEED AT

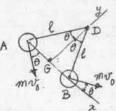
WHICH A AND B ROTATE

ABOUT D

(C) PERCENT ENERGY OF

PROB. 14.49: ASSUME 0 = 30°.
PROB. 14.50; ASSUME 0 = 45°.

WE CONSIDER THE FOLLOWING TWO POSTICUS OF THE SPHERES A AND B AND THE RING D. POSITION 1: IMMEDIATELY AFTER CORD CD BREAKS



LINEAR MOMENTOM: $L = (2m v_0 \cos \theta) \dot{L}$ (1)

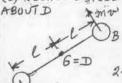
ANGULAR MOMENTUN ABOUT G: HG = 2 ((sin t)) (m t) sint) (2)

POSITION 2: AFTER COLDS AD AND BD BECOME TAUT:

RECALLING (1): $L = (2m)\overline{V} - (2mV_{i}\cos\theta)\underline{L} \qquad \overline{V} = (V_{i}\cos\theta)\underline{L}$

 $V_D = \overline{V} = V_0 \cos \theta$ (3)

(b) RECATIVE SPEED WAT WHICH A AND B ROTATE



ANG. MOMENTUH ABOUTG: Ha=(2 m v))

RECALLING (2):

2mvl-2lm v5 sin 0 v= v0 sin 0 (4)

C) ENERGY LOST:

CONSIDERING SYSTEM OF 3 SPHERES!

INITIALLY', Vc = (e'/e) vA = (20050) V5. THEREFORE

To = 1 m VA+ 1 m VB2+ 2 m Vc = m Vo2 (1+2cos20)

 $T_S = \frac{1}{2} (2m) U_D^2 + 2 (\frac{1}{2} m U^2) + \frac{1}{2} m U_C^2$

= m [v2 cos20 + m (v sin20)2+ z v6 co2 6]

 $= m v_0^2 (3 \cos \theta + m (v_0 \sin \theta)^2 + 2 v_0 \cos \theta)$ $= m v_0^2 (3 \cos \theta + \sin \theta)$

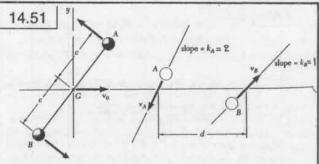
 $7.\cos s = 100 \frac{T_0 - T_0}{T_0} = 100 \frac{1 + 2\cos^2\theta - 3\cos\theta - \sin^4\theta}{1 + 2\cos^2\theta} = 100 \frac{\sin^2\theta \cos^2\theta}{1 + 2\cos^2\theta}$ (5)

PROB 14.49: MAKING 8=30' IN EUS. (3), (4), AND (5)

(a) 0.866 vo. (b) 0.250 vo. (c) 7.50 %

PROB. 14.50: MAKING 0 = 45° IN EQS. (3),(4),4ND(5);

(a) 0.707 vo. (b) 0.500 vo. (c) 12.50%



GIVEN: TWO SMALL IDENTICAL SPHERES A AND B, CONNECTED BY A CORD SLIDE ON A HORIZONTAL, FRICTION-LESS SURFACE. INITIALLY THEY ROTATE WITH DEBrod/S ABOUT G, AND G HAS VELOCITY TO = Vol. AFTER CORD BREAKS, SPHERES MOVE ALONG PATHS WITH KA = 2, KB = 1, MND d = 625 mm.
FIND:

(a) SPEEDS No, VA. AND VB, (b) length 20 of word

BEFORE BREAK: Lo = (2m) D L = 2m Vo

AFTER BREAK: $d = mv_B + mv_B = m(-v_5 v_A + v_b v_B)i$ $+ m(-v_5 v_A + v_b v_B)i$

SETTING L = LO AND EQUATING COEFF. OF UNIT VECTORS!

SUBTRACTING (2) FROM (1): \$\frac{1}{\sqrt{5}} \textsup A = 2\frac{1}{\sqrt{5}} \textsup A = 2\frac{

CONSERVATION OF ANGULAR MOMENTUM

BEFORE BREAK: (HG) = 2 m c + = 2mc (8004/s) = 16 mc

ATTER BREAK: HG=HA=M(VB), d=m (4V2V) (0,625m) = 2,5m Vb

SETTING HG=(HG); 2.5 m Vo = 16 m c2 Vo = 6.40 c (5)

CONSERVATION OF ENERGY

BEFORE BREAK: $T_0 = \frac{1}{Z}(2\pi)V_0^L + \frac{1}{Z}(2\pi)(c\theta)^2 = m\left(V_0^L + c^2\theta^2\right)$

LETTING B=Brods AND ASING(5): To= m(40,966464 c2)
AFTER BREAK: T= 1 m Vp2 + 1 m VB2

RECALLING (3), (4), AND (5)

 $T = \frac{1}{2}m(20v_0^2 + 32v_0^2) = 26mv_0^2 = 1064.96mc^4$

SETTING T=T: 1064, 96 C4 = 40.96 C4 + 64 C 1024 C4 = 64 C4 = 0.0625 C= 0.250 M

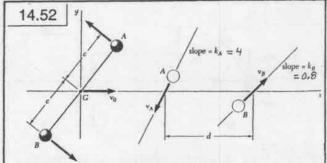
FRUM (5): Vo = 6.40(0.0625) = 0.400 m/s

FROM (3): $V_A = 2\sqrt{5}(0.4) = 1.789 \, \text{m/s}$ FROM (4): $V_B = 4\sqrt{2}(0.4) = 2.26 \, \text{m/s}$

ANSWERS:

(a) Vo = 0.400 m/s; VA = 1,789 m/s; VB = 2.26 m/s

(b) LENGTH OF CORD = 2C = 500 mm



GIVEN: TWO SMALL IDENTICAL SPHERES A AND B. CONNECTED BY A CORD OF LENGTH 2C = 600 mm SLIDE ON A HORIZONTAL, FRICTIONLESS SURFACE, INITIALLY THEY ROTATE WITH &= 12 rad/s ABOUT G, AND G MOVES WITH U = VOL . AFTER CORD BREAKS, SPHERES MOVE ALONG PATHS WITH KD = 4 AND KB = 0.8.

FIND: SPEEDS NO, VA, AND NB, (b) DISTANCE d

CONSTRUATION OF LINEAR HOMENTUM BEFORE BREAK: Lo=(2m) To



L=mv+mv=m(-100++510)+m(-100+10)

SETTING L= LO AND EQUATING COEFT OF UNIT VECTORS (1) (1) -1 VA+ 5 VB = 2 VO

D-特本+端切=0,

SUBSTITUTE FOR UD INTO (1): VA = 117 V (3) - 15 VA + 17 VA = 2 VS

PROM (2): 1/3 = 1/17 17 18 VB = 141 V (4)

CONSERVATION OF ANGULAR MOMENTUM

BEFORE BREAK! (Hg) = 2mc + = 2m (0,3 m) (12 rad/s) = m(2.16) AFTER BREAK: HG=HA=m(VB), d=md # To=Ind V6 SETTING HG=(HG); 2 and vo=m(2.16) vod=1.08 (5)

CONSERVATION OF ENERGY BEFORE PREAK: $\frac{1}{6} = \frac{1}{2} (2m) v_0^2 + \frac{1}{2} (2m) (c \dot{\theta})^2$ = mvo2+m(0.3×12)= m(vo2+12,96)

AFTER BREAK: T= 1 m VA+ 1 m VB RECALLING (3) AND (4): T= 1/2 mv (17/4 + 41) = 7.25 m v 0 SETTING 7= To: 7.25 0 = 0+12.96 2 =1.440 0/5 FROM (3): UA = 117 (1.440) = 2.969 m/s FROM (4): 0 = 141 (1.440) = 4.610 m/s FROM (5): d= 1.00 - 0.750 m

ANSWERS! (a) vo= 1,440 m/s; VA= 2,97 m/s; VB = 4,61 m/s

(b) DISTANCE d = 0.750 m = 750 mm

14.53 D

GIVEN: BALL A HITS BALL B WITH U = (12 ft/s)c, THEN C, THEN SIDE OF TABLE AT A' (WHERE a = 66 in.) WITH 15 =- (5,76 ft/s) j (a) VELOCITIES OF BAND C (b) DISTANCE C WHERE BALL CHITS SIDE

(3)

4

4

(3)

CONSERVATION OF LINEAR MOMENTUM mo i = - mvaj + m(ve), i + m(ve), j + mvci

EQUATING COEFF. OF UNIT VECTORS! (NB) + V = V = 12 ft/s (1) (1) MNg = M(NB)2+MNC

(VA) = VA = 5.76 #/5 (2) 0 = - m vA + (VB), CONSERVATION OF ANG. MOMENTUM ABOUT CORNERD

5 (301n.) Vo = - (66in.) VA + (72.5in.) (VB) + C VC 30(12) = -66(5.76)+(72.5)(5.76) + CV

CUC = 322.56 CONSERVATION OF ENERGY 1 mV02 = 1 m VA + 1 m [(VB) + (VB) + 1 m Vc

DIVIDING BY M, MULTIPLYING BY 2, AND SUBSTITUTING FOR TO, VA, (VB), THEIR VALUES AND (VE) = 12-12 FROM (1): (12) = (5.76) + (12-Vc) + (5.76) + Vc

DIVIDING BY 2: VC = 12 VC + (5,76) = 0, V = 6 + 1,68 WITH VC = 6-1.68 = 4.32, EQ. (3) YIELDS C = 74.7 (IMPOSSIBE) 1 = 7,68 Als - 1 THEREFORE: V= 6+1,68 = 7,68

TROH (1): (4B) = 12-7.68 = 4.32

No = (4.321/2) + (5.76 ft) OR 1/3 = 7,20 ft/s 253,10 FROM (3): C(7.68) = 322,56.

(SEE FIGURE OF PROB. 14,53) GIVEN: BALL A HITS B WITH T=(15-FL/5) 65 THEN C; BALL C HITS SIDE AT C=48 in, WITH U= (9.6+1/5) L FIND: (a) UA AND UB, (b) DISTANCE Q,

CONSERVATION OF LINEAR MOMENTUM

mv i = - m vaj + m (vB), i + m (vB), j + m v i (NB) = 15,-9,6=5.40 ft/s (1) (1) mvo = m(vB) +mvc

(UB) = UA DO = - MVA HUBY

CONSERVATION OF ANGULAR MOMENTUM ABOUT CORNER D + 5 (30in.) vo=- a va + (72.5in.) (v) + c vc

USING EQ. (2): SUBSTITUTING GIVEN DATA AND 30 (15 ft/s) = - a VA + 72.5 VA + 4B/9.6 ft/s)

(a-72,5)VA= 10,8

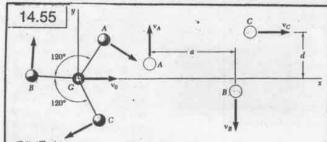
CONSERVATION OF ENERGY 1 m Vo = 1 m VA2+1 m [(VB) x+(VB))]+1 m VE DIVIDING BY M, MULTIPLYING BY 2 MID SUBSTITUTING I (15)= UA+(5,40)2+UA2+(9,6)2

U=7.20 14/5 VA= 7.20 ft/s UA = 51,84

PROH (1) AND (2): 1 = (V8) 1 + (V8) 4 = (5.40 ft/s) i + (7.20 ft/s) j VB = 9,00 H/s 253,10

PROM(3): (a-72,5)7,20 = 10,8

a = 74.0 in. a=72,5+45



GIVEN: THREE SMALL IDENTICAL SPHERES CONNECTED BY 200-AM-LONG STRINGS TO RING & SLIDE ON A HORIZONTAL, FRICTIONLESS SURFACE.

INITIALLY, SPHERES ROTHTE ABOUT & WITH O,8 m/s RELATIVE VELOCITY AND RING MOVES WITH 5=(0.4m/s) SUDDENLY RING BREAKS AND SPHERES MOVE FREELY HS S'HOWN WITH a = 346 mm.

FIND:

(a) VELOCITY OF EACH 5 PHERE, (b) DISTANCEd.

CONSERVATION OF LINEAR MOMENTUM

BEFORE BREAK: L = (3m) = 3m (0,4i)=m(1.2ms)i APTER BREAK: L= mvaj - mvBj + mvc i

[= L]: mv[+m(va-vB)j = m(1,2 m/s)i THEREFORE:

> 14= 1,200 m/s -> (2) 10, =1,200 m/s

CONSERVATION OF ANGULAR MUMENTUM

BEFORE BREAK: +) (Ho) = 3 m (v = 3 m (0,2m) (0,8m/s) = 0.480 m

+2 Ho=-m25AXA AFTER BREAK! mvc MUA 4 0.346M + mva (3/2+0,346) + mvcd Ho = (Ho) : 0,346 m vA + m vcd = 0,480 m FROM(1): OB RECALLING (2): 0,346 VA+ 1,200 d = 0,480 (3) d=0,400-0,28833 Va

CONSTRUATION OF ENERGY

BEFORE BREAK!

 $T_0 = \frac{1}{2} (3m) \overline{v}^2 + 3 (\frac{1}{2} m v)^2)$ $= \frac{3}{2} m (v_0^2 + v)^2) = \frac{3}{2} [(0.4)^2 + (0.8)^2] m = 1,200 m$

AFTER BREAK:

T= 1 m VA+ 1 m VB2+ 1 m Vc

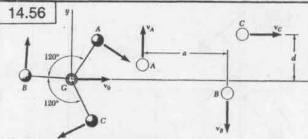
SUBSTITUTING FOR UB FROM (1) AND V FROM (2): [UA2+ VA2+ (1,200)2] = 1,200 VA2 = 0.480 VA= VB = 0,69282 m/s

(a) VELOCITIES!

v = 0.693 m/s ↑; v = 0.693 m/s ↓; v = 1.200 m/s →

(b) DISTANCE d:

PRON (3): d=0.400-0.28833(0.69282) = 0.20024 m d=200 mm



GIVEN:

THREE SMALL IDENTICAL SPHERES CONNECTED BY STRINGS OF LENGTH & TO RING G SLIDE ON A HORIZONTAL, FRICTIONLESS SURFACE. INITIALLY, SPHERES KOTATE ABOUT G AND KINI.

MOVES ASSHOWN. SUDDENLY RING BREAKS AND SPHERE MOVE PREELY IN 24 PLANE. WE KNOW THAT で = (1.039 m/s)主, ひ= (1.000 m/s)上,

a = 416 mm, d = 240 mm.

FIND:

(1)

(a) VEL. V OF RING, (b) LENGTH & OF STRINGS (C) RATE IN rad/s AT WHICH SPHERES WERE ROTATING

CONSERVATION OF LINEAR MOMENTUM

(3m) v = m v + m v + m v

3 m Vo L = m (1,039 m/s) j - 11 VB j + m (1,800 m/s) i

EQUATING COEPF. OF UNIT VECTORS! (1) 3 Vo = 1, BOO m/s

(a) v = 0.600 m/s

1 0=1.039m/s-VB

(1) · V= 1.039 m/s

CONSERVATION OF ANGULAR MOMENTUM BEPORE BREAK: +) (Ho) = 3 on 62 à

AFTER BREAK: +2 H = - m VAZA mvc mv4 0.416m +m VA (VA + 0, 416) 0,240m. + m V (0,240) = m (1.039 X 0.416) +m (1.800)(0.240) = m(0,864224)

 $(H_0) = H_0: \Im m \ell^2 \dot{\theta} = m (0.864224)$ $\ell^2 \dot{\theta} = 0.28807$ (2)

CONSERVATION OF ENERGY

BEFORE BREAKS

ての= 美(3m) で+3(まかび)=3mが+3m(l) =3 m(0,600) +3 m 202

AFTER BREAK:

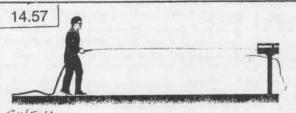
T= = m NA + = more + = m Nc $= \frac{1}{2} m \left[(1.039)^2 + (1.039)^2 + (1.800)^4 = \frac{1}{2} m (5.399)^4 \right]$

T=To: 1 m (5,399) = 3 m (0,600) 2+3 ml2 +3 (3)

l'0 = 1.4397 DIVIDING (3) BY (2): 0 = 64397 = 4,9976

(b) FROM(2): L2 = 0.28807 £=0,2401 m l= 240 mm

(c) RATE OF RUTATION = 0 = 5,00 rad/s



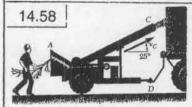
GIVEN:

VEL. OF STREAM = 25 m/s A = 300 mm

FIND: FORCE EXERTED BY STREAM ON MAILBOX.

 $\begin{array}{c|ccccc} (Dm) \frac{V_0}{V_0} & P\Delta t & \pm_0 \dot{Z} & coMP \\ \hline \rightarrow \mathbb{Z} & + & = 0 & (\Delta m) V_0 & - P\Delta t = 0 \\ P = \underbrace{Am}_{\Delta t} V_0 & = (PA V_0) V_0 & = (PA V_0^t) \\ P & = (1000 k_S / n_1^3)(300 \times 10^6 m_1^t)(25 m/s)^t \\ P & = 187.5 N \end{array}$

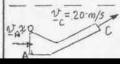
NOTE: FORCE P SHOWN ON SKETCH IS FORCE APPLIED BY MAILBOX ON STREAM, FURCE EXERTED BY STREAM ON MAILBOX IS 187.5 N ->



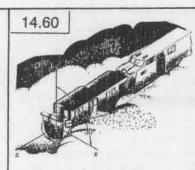
GIVEN:

TREE LIMBS ARE FED INTO SHREDDER AT RATE OF 5 kg/s AND CHIPS HRE SPEWED WITH VC= 20 m/s.

FIND: HORIZ. COMP. OF FORCE EXERTED ON HITCH AT D.



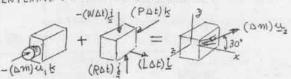
EQ. (14.38)! $(\Delta m) v_A + \Sigma Fot = (\Delta m) v_C$ $\Sigma F = \Delta m v_C = (5 kg/s)(20 m/s = 25^\circ)$



GIVEN:
ENGINE PROPELS FLOW
AT SPEED OF 12 mi/h.
PLOW PRUJECTS 180 TONS
OF SMOW PEK MINETE
WITH VELOCITY OF
40 ft/s W/R TO CAR.
FIND:

(a) FORCE EXERTED BY ENGINE ON CAR (b) LATERAL FORCE EXERTED BY TRACK.

WE MEASURE ALL VELOCITIES WE PLOVE CHE AND APPLY THE IMPULSE-HOMENTUM PRINCIPLE TO THE PLOW CAR, THE SNOW IT CONTAINS. AND THE SNOW ENTERING IN THE TIME INTERVAL Dt.



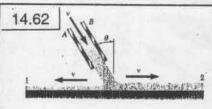
-(Om) u, k+(PDt) k+(RDt) j-(WDt) j= (Am) uz (cos 30° i + 5in 30° j)

EQUATING THE COEFF, OF THE UNIT VECTORS!

(a) $-(\Delta m)u_1 + P\Delta t = 0$ (b) $L\Delta t = (\Delta m)u_2 \cos 30^{\circ}$ $L = \frac{\Delta m}{\Delta t} u_2 \cos 30^{\circ}$ (c)

WITH GIVEN DATA: $\omega_1 = 12 \text{ mi/h} = (7.60 \text{ ft/s}, \quad \omega_2 = 40 \text{ ft/s}$ $\frac{\Delta m}{\Delta t} = (180 \text{ fons/min}) (\frac{1 \text{ min}}{603}) (\frac{2000 \text{ lb}}{1 \text{ ton}}) (\frac{32.2 \text{ ft/s}}{32.2 \text{ ft/s}}) = 186.34 \text{ lb·s/ft}$ (a) EQ.(1): P = (186.34 lb·s/ft) (17.60 ft/s) P = (3280 lb) k

E) FTO (2): 1 = (186.34 | b. s/f+)(40 ft/s) cos 30° 1 = (6 1100 12).



GIVEN:

U = 40 m/s, 0 = 30

TOTAL PORCE
EXERTED BY STREAM
ON PLATE = 500N &.

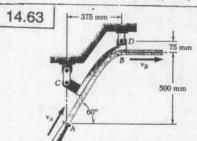
FIND: Q, AND Q.

OF RESULTING
STREAMS.

WE NOTE THAT $Q = Q_1 + Q_2$ (1) SEE SOLUTION OF PROELITION OF QSIND = $Q_2 - Q_1$ (2) $P = Q \cup COSD$ (3) FROM (3): $Q = \frac{P}{Q \cup COSD} = \frac{500 \text{ N}}{(1 \text{ kg/L})(40 \text{ m/s}) \cos 30} = 14.43 + L/s$ = 866.03 L/min

ADDING (1) AND (2): $Q(1+\sin\theta) = 2Q_2$ $Q_z = \frac{1+\sin\theta}{2}Q = \frac{1+\sin30^{\circ}}{2}(866,03 \text{ L/min}) = 649.52 \text{ L/min}$

FROM (1): $Q_1 = Q - Q_2 = 866.03 - 649.52 = 216.51 L/min$ $Q_1 = 217 L/min$; $Q_2 = 650 L/min$



GIVEN!

WATER DISCHARGED

AT RATE Q=1.2 m/ain

WITH VA=VB=25 m/s

FIND:

COMPONENTS OF

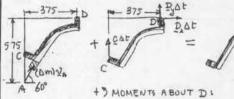
REACTIONS AT C AND D.

(NEGLECT WEIGHT

(Dm) VB

OF VANED.

WE APPLY THE IMPULSE- MOMENTUM PRINCIPLE TO THE BLADE, WATER IN CONTACT WITH THE ELASE, AND WATER STEINING THE BLADE IN INTERVAL DT.



575(0m) $V_A \cos 60^\circ - 375(\Delta m)V_B \sin 60^\circ - 375 C \Delta t = 75(\Delta m)V_B$ 375 C = $\frac{\Delta m}{\Delta t}$ (25 m/s)(575 cos 60°-375 sin 60°- 75).

C = -7.484 $\frac{\Delta m}{\Delta t}$

BUT $\Delta m = (Q = (1000 \text{kg/m}^3)) (1.2 \frac{m^3}{605}) = 20 \text{kg/s}$ THU3: C = -7.484(20) = -149.68 M $C_2 = 0, C_3 = 149.7 \text{ M} = -149.7 \text{ M}$

 $\pm 2 \cos(P_1: (\Delta m)) v_A \cos 60^\circ + D_2 \Delta t = (\Delta m) v_B$ $D_2 = \frac{\Delta m}{\Delta t} (25 \text{ m/s}) (1 - \cos 60^\circ) = (20 \text{ kg/s}) (25 \text{ m/s}) (1 - \cos 60^\circ)$ $D_1 = 250 \text{ N} \rightarrow$

+4 y comp.; (Am) $^{\circ}H$ sin $^{\circ}H$ + CAL+ $^{\circ}D_{g}$ $^{\circ}D_{g} = -\frac{\Delta m}{\Delta L}(25 \text{ m/s}) \sin 60^{\circ} - (-149.7 \text{ N})$ $= -(20 \text{ kg/s})(25 \text{ m/s}) \sin 60^{\circ} + 149.7 \text{ N}$ = -433,0 N + 149.7 N = -283.3 N $D = 283 \text{ N} + \frac{1}{2}$

14.64 ASSUME THAT BLADE ABOF SAMPLE PROBLET.

IS IN THE SHAPE OF AN HIC OFCICLE.

SHOW THAT RESULTANT FORCE F EXERTED BY THEBLADE
ON THE STREAM IS APPLIED AT MIDPOINT C OF ARC AB.

WE APPLY THE IMPULSE-HUMENTUM PRINCIPLE TO THE PORTION OF STREAM IN CONTACT WITH THE BLADE AND ENTERING IN CONTACT IN INTERVAL AT.

WE RECALL THAT $u_A = u_B = M$ + 3 MOMENTS ABOUT O! $R(\Delta m)u + MOM.OFF = R(\Delta m)u$ THUS: MOM.OF F ABOUT O = 0 ; LINE OF ACTION OF F PASSE
THROUGH O.

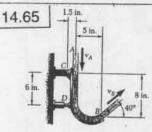
 $\pm 2 \operatorname{COMP}$: $(\Delta m) u - (F\Delta t) \sin \alpha = (\Delta m) u \cos \theta$ $\otimes v.: F(\Delta t) \sin \alpha = (\Delta m) u (1-\cos \theta)$ (1) $\pm 4 \cos \alpha P.: O + F(\Delta t) \cos \alpha = (\Delta m) u \sin \theta$ (2)

DIVIDE (1) BY (2): $\tan \alpha = \frac{1 - \cos \theta}{\sinh \theta} = \frac{2 \sin \frac{\pi}{2}}{2 \sin \frac{\pi}{2} \cos \frac{\pi}{2}} = \tan \frac{\pi}{2} \qquad \alpha = \frac{\theta}{2}$

THUS: LINE OF ACTION OF E BISSECTS & AOB,

E IS APPLIED AT MIDPOINT C OF ARC AB.

(Q.E.D.)



GIVEN:

STREAM OF WATER WITH

Q = 150 gal/min

VA = VB = 60 ft/s

REMOTION AT D HORIZONTAL-

COMPONENTS OF REACTIONS AT C AND D (NEGLECT WEIGHT OF VANE)

WE APPLY THE IMPULSE-PICHENTUM PRINCIPLE TO THE VAME, THE WATER IN CONTACT WITH IT, AND THE MISS AM OF WHICK ENTERING AND LEAVING THE SYSTEM IN THE INTERVAL A t. WE NOTE THAT $\Delta m = QQ \Delta t = \frac{62.4}{32.2 ft/s} (\frac{150 gal}{60 s}) \frac{15t^3}{7.48 gal} \Delta t = (0.6477 lb/s) \Delta t$



DHom. about C: - (Δm) v_g(1.5in.) + D Δt(6in.) = ... = (Δm) v_g cos 40°(8in.) + (Δm) v_g sin 40 (6.5in)

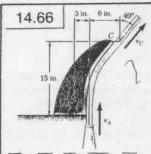
 $DDt(6in.) = (0.6477 \frac{16.5}{5t}) Dt(60 ft/s) (11.806 in.)$ $D = 76.47 \frac{16}{5t}$

5,2 COMP.: C, Dt + D Dt = (Dm) \$ COS 40" C= (0.6477 16.5) (60 ft/s) COS 40"-76.4716 = -4671.

+1 y comp: $-(\Delta m)V_A + C_3 \Delta t = (\Delta m)V_B \sin 90^\circ$

Cy = (0.6477 16.5)(60 ft/s)(5in 40°+1) = 63.8 16

 $C_{x} = 46.7 \text{ lb} \leftrightarrow$, $C_{y} = 63.8 \text{ lb}^{1}$ $D_{x} = 76.5 \text{ lb} \rightarrow$, $D_{y} = 0$



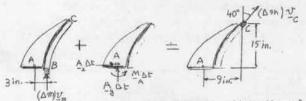
GIVEN: STREAM OF WATER WITH Q = 200 gal/min AND UR = V = 100 ft/s

FIND:

FORCE-COUPLE SYSTEM APPLIED TO VANE AT A. (NEGLECT WEIGHT OF VANE)

WE APPLY THE IMPIRSE-AND ISLITURE INFRIENCE TO THE VAR THE WATER INCOMPA" . 191711 AND THE MASS AM OF WATER SHILL IL - - - 1 ENVIRING THE SYSTEM IN At. WE NOTE THAT

DM = 0 Q D = 62.4 16/4 20080) 7.4880 Dt = (0.8636 16.5) Dt



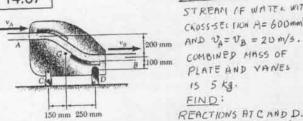
± x comp: A Δt = (Dm) Vc sin 40"=(0.8636 Δt)(100 ft/s) sin 40" A=55.5/b-A = 55.5116

+ 9 y COMP. : (Am) UB + Ay At = (Am) VE COS 40'

Ay = (0.8636)(100 ft/s)(ws40-1) = - 20.2 lb, Ay=20.2 lb

+3 MOMENTS HEOUT HI (Dm) 15 (3 in) + MADE = - (Dm) 1/2 sin 40 (15 in) + (Dm) Uz cos 40 (9 in) MA = (0.8636 16.5/ft)(100ftk)[-(15in)sin+0+(9in)cos+0-3in] M = 496/6-in) = -496.3 lb.in.

14.67

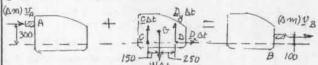


150 mm 250 mm

GIVEN:

STREAM OF WATER WITH CROSS-SEL TION A= 600mm AND VA = VB = 20 m/s. COMBINED MASS OF PLATE AND VANES 15 5 54. FIND:

WE APPLY IMPULSE-MOMENTUM PRINCIPLE TO PLATE, VANES WATER IN CONTACT WITH PLATE, AND MASS ON OF WATER ENTERING AND LEAVING SYSTEM IN INTERVAL IST.

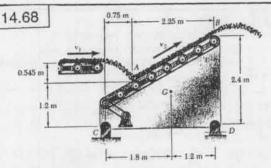


An = (Qat= PAUAt = (1000 +87m3)(600x10 m3)(20 M6) At = (12 kg/s) At 1,2 comp: (Am) th + D2 at = (Am) UB, D= (1,200)(20-20) = 0 4) MOM. ABOUTD: (AM) of (500) + C At (400) - WAT (250) = (AM) of (100) 400 C = (12 kg/s)(20 m/s)(100-300) + (5 x 9.81N)(250) = -35,738 C=89.3 No

C = -89.344 N Py COMP: (-89,344 - 5 x 9,81 + Dy) OF = 0

 $D_{\rm g}=+138.39$ RECALLING THAT $D_{\rm g}=0$.

D=138.4N

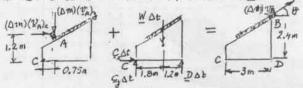


GIVEN! COAL DISCHARGED FROM FIRST TO SECOND CONVEYOR BELT AT RATE OF 120 kg/s WITH U = 3 m/s AND V = 4,25 m/s. MASS OF SECOND BELT ASSEMBLY AND COAL IT SUPPORTS IS 472 kg. FIND: COMPONENTS OF REACTIONS AT CAND D.

MASS OF COAL ENTERING AND LEAVING SYSTEM IN AT: Dm = (120 kg/5) DE VELOCITY IF WITH WHICH COAL U,=3 m/5 HITS SECUND BELT : (14) = 1 = 3 m/s -(2) (2/A)2 (VA) = 129h = 12 (9.81) (0.545)

(IA) = 3,27 m/s + (3)

WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE TO THE SECOND BELT ASSEMBLY, THE COAL IT SUPPORTS, A 113 THE MAKS LAT OF COAL INTTING IT IN INTERCAL ACT.



WE NOTE THAT can 0 = 1.25m

t) MOH. ABOUT C: (ATI)(24)2 (1.2m) + (DM)(1/1)4 (0,75m) + (WDt)(1.8m) - (DDt)(3m)

= (Dm)(VB cos A)(2, 4m) - (Bm)(VB sin 4)(3m) D(3 m) = (472 kg)(9,81 m/s*)(1,8 m) + (120 kg/s) [(3 m/s)(1.2m) + (3,27m/s)(0.75 m)]

-(120 kg/s)(4.25mg) [2.4m) cos 28.07-(3 m)sin 28.07")] D(3m) = 8334.6 N.m+726.30 N.m-360.08 N.m = 8700 N.m

D=0, D= 2900N1 D = 2900 N

to 2 COMP: (DM)(VA), + CZ St = (DM) VB COS D

C = (120 kg/s) (4,25 m/s) cos 28,07 - (120 kg/s) (3 m/s) C=90.0N-> = 90,0 N

+94 COMP: - (Ban)(VA)y + Cy St+DOL-WAt = (Om) VE SIND Cy=-2900 N+ (472kg)(4.81m/s)+(120kg/s)(3.27+4.25 sin 28.07) m/s = 2362,7 N C = 2360N4



WHEN BELT IS AT REST:) EH, = 01 D(3m) - W(1.8m) =0 3D-(472×9.81)(1.8)=0 D = 2778 N D = 2730 N)

C = 471 × 9.81 - 2778 C= 1851 NT

GIVEN! PLANE CRUISES AT 900 km/11.

SCOOPS AIR AT RATE OF 90 kg/s AND DISCHARGES IT AT 660 M/S RELATIVE TO PLANE,

FIND; TOTAL DRAG DUE TO AIR FRICTION

WE APPLY EQ. (14.39): EF = dm (v - v4)

WITH RESPECT TO PLANE. WE HAVE: ZF = D = TOTAL DRAG,

 $v_B = 660 \text{ m/s}, \quad v_A = 900 \text{ km/h} = 900 \frac{1000 \text{ m}}{3000 \text{ s}} = 250 \text{ m/s}$

EQ. (14.39): D = (90+8/s) (660 m/s - 250 m/s)

D=36.9 KN

14.70

GIVEN:

PLANE IN LEVEL FLIGHT AT STOMI/h. DRAG DUE TO AIR FRICTION = 7500 16 EXHAUST VEL . = 1800 ft/s RELATIVE TO PLANE

FIND: RATE IN 16/5 AT WHICH AIR PASSES THRU ENGINE

WE APPLY EU. (14,39): EF = dm (v - v) WITH RESPECT TO PLANE.

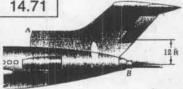
WE HAVE EF = DRAG = 7500 16

U = 1800 ft/s, U = 570 mi/h = 836 ft/s

EU. (14.39): 7500 /b = dm (1800 ft/s -836 ft/s)

dm = (7,780-16.5)(32,2 ft/3) = 251 16/5

14.71



GIVEN:

ENGINE SCOOPS IN AIK AT A AT RATE OF 200 1b/s AND DISCHARGES IT AT B AT 2000 ft/s W/R PLANE FIND:

THRUST OF ENGINE WHEN ATRPLANE SPEED IS (a) 300 mi/h, (b) 600 mi/h.

WE APPLY IMP. - MOM. PRINCIPLE USING VELOC. W/R PLANE



\$2 COMP; (Am) uA + FAt = (Am) UB

 $F = \frac{\Delta m}{\Delta t} (u_B - u_A) = \frac{200 \text{ b/s}}{32.2 \text{ ft/s}^2} (2000 \text{ ft/s} - v)$ +) MOM, ABOUT B: $-(\Delta m) u_A (125t) + (F \Delta v) d = 0$ (1)

Fd = Am (12ft) 11 = 200 6/3 (12 ft) v

(a) v = 300 mi/h = 440 ft/s

EQ(1): $F = \frac{200}{32.2} (2000 - 440) = 9,689 /b$

Pa. (2): Fd = 200 (12)(440) = 32,795 16.56

DIVIDE (2) BY (1): d = 3.38ft

MNSWER: 9690 16, 3,38 ft BELOW B

(b) v = 600 mi/n = 880 ft/s

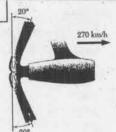
 $EQ.(1): F = \frac{200}{32.2}(2000 - 880) = 6,956 / 6$

EQ.(2): Fd = 200 (12)(880) = 65,590 /b.ft

DIVIDE (2) BY (1): d = 9.43 ft

ANSWER: 696016, 9.43 ft BELOW B

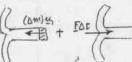
14.72



GIVEN:

IN REVERSE THRUCT. ENGINE SCOOPS AIR AT RATE OF 120 kg/s AND DISCHARGES IT AS SHOWN WITH VELOCITY OF 600 m/s RELATIVE TO ENGINE. FINI: REVERSE THRUST WHEN PLANE SPEED IS 270km/h.

WE APPLY IMPULSE-FIGHENTIM PRINCIPLE WIR PLANE

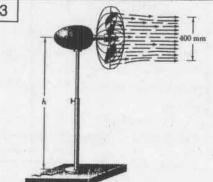


towa.

4 = V= 270 km/h =75 m/s 112 = 600 m/s

(F IS OPPOSITE TO REVERSE THRUST OF ENGINE) + x comp: - (am) w + Fat = 2 [1/2 (am) w sin 20"] F= Om (11 + 12 sin 20") = (120 kg/s)(75 + 600 sin 20") m/s F= 33.6 KN

14.73



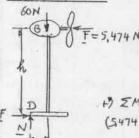
GIVEN: FLOOR FAN DELIVERS AIR WITH SPEED OF 6 m/s. IT IS SUPPORTED BY A 200-MM - DIAMETER CIRCULAK BASE AND ITS TOTAL WEIGHT IS 60 N.

FIND: MAX, HEIGHT & IF FAN IS NOT TO TIP OVER. (USE @ = 1.21 kg/m3 FOR HIR AND ASSUME VA = 0.)

THRUST:

FROM EQ. (14.39): F = dm (VB-VA) = (Q(V-V) = (1.39) $F = (1.21 \text{ kg/m}^3) \frac{97}{4} (0.400 \text{ m})^2 (6 \text{ m/s})^2 = 5,474 \text{ N}$

FREE BODY: FAN



FORCE EXERTED ON : . / BY HIR STREAM IS EWUHL AND OPPOSITE TO THRUST.

WHEN FAN IS ABOUT TO TIP OVER, NORMAL FORCE N IS APPLIED AT D.

+) ZMD = 0:

(5474 N) & - (60 N)(0,1 m)=0

h= 1.096 m

GIVEN:

MAX, DOWNWARD AIR SPEED PRODUCED BY HELICOPTER 15 BO Ft/s. WEIGHT OF HELICOPTER AND CREW 15 3500 1b. FIND!

MAX. LOAD THAT HELICOPTER CAN LIFT WHILE HOVERING, (ASSUME 8 = 0.076 16/ft3 FOR AIR.)

WE USE EQ. (14,39) TO DETERMINE THE THINKTF: F = dm (VB-VA) = QQ(V-0) = QA 52 = 3 AU2 F = 0.076 16/52 \$\frac{T}{1}(304t)^2(80+t/s)^2 = 10,678 16

THE LIFT PROVIDED BY THE BLADE IS EQUALAND CIPPOSITE, THAT IS 10,678 16 T. WE WRITE +1 ZF4 = 0: 10,678 16 - W - 3500 16 = 0 W = 717816 W=718016

14.75



GIVEN;

AIRLINER CRUISES AT 600 mi/h WITH EACH OF ITS THREE ENGINES DISCHARGING AIR AT 2000 11/5 RELATIVE TO PLANE.

SPEED OF PLANE AFTER IT HAS LUST THE USE OF (a) ONE ENGINE (b) TWO ENGINES (ASSUME THAT DRAG IS PROPORTIONAL TO UZ.)

WE USE EQ. (14.39) TO DETERMINE THE TOTALTHRUST OF THE ENGINES:

 $F = \frac{dm}{dt}(v_B - v_A)$ WHERE $v_B = 2000 \text{ ft/s}$ V = SPEED OF PLANE 7HUS: F = din (2000 - V)

THE DRAG IS D= AU

EQUATING THRUST AND DRAG: dm (2000 - v) = &v WITH THREE ENGINES V = 600 mi/h = 880 H/s

SUBSTITUTING IN EU. (1):

 $\left(\frac{dm}{dt}\right)_{3}(2000-880) = 4(880)^{2}$ (dm) = 691.43 4

(a) WITH TWO ENGINES:

(dm) = = 2 (dm) = = (691,43 k) = 460,95 R

SVESTITUTING III EU. (1):

460.95年 (2000-サ)= ねか 12+ 460.95 v -921.9 × 103= 0

V = -460.95+V(460.95) + 4(421.4 × 103) = 756.96 ft/s v=516 mi/h

(b) WITH ONE ENGINE:

 $\left(\frac{dm}{dt}\right)_1 = \frac{1}{3}\left(\frac{dm}{dt}\right)_3 = \frac{1}{3}(691.43 \text{ k}) = 230.48 \text{ k}$

SUBSTITUTING IN EQ. (1): 230,484 (2000-0)= 400=

12+230.48 7 - 460.95 × 105 = 0

15 = - 230,48+ (230.48)2+ 4(460.95×103) = 573,41 ft/s v=391 mi/h ✓



GIVEN:

16-Mg PLANE MAINTAINS U= 774 km/h WITH X= 18. IT SCOUPS AIR AT RATE OF 300 kg/s AND DISCHARGES IT AT 665 m/s RELATIVE TO PLANE

FIND: (a) INITIAL ACCELERATION IF PILOT CHANGES TO HORIZONTAL FLIGHT WITH SAME ENGINE SETTING (b) MAX. HORIZONTAL SPEED THAT WILL BE ATTAINED. (ASSUME THAT DRAG IS PROPOLITIONAL TO J.)

DETERMINATION OF THRUST

SINCE AIRPLANE IS ACCELERATED IN HURIZONTAL FLIGHT WE USE A REFERENCE FRAME AT REST WITH RESPECT TO THE ATMOSPHERE WHEN USING EQ. (14:34) TO DETERMINE THE THRUST F (CF. FOUTNOTE, PAGE 860). F = dm (vB - VA)

WHERE VA = 0, VB = VDISCH! PLANE = 665 m/s - 174 km/h (1000m) = 665 m/s - 215 m/s = 450 m/s

F=(300kg/s)(450 m/s-0) = 135,0 kN AIRPLANE CLIMBING (NO ACCELERATION)



IFX 18 = 0 1350KN-D-Wsin 18=0

D = 135,0 kN - (16Mg)(9,81 mg) sin 18 = 135.0 kN - 48,50 kN = 86,50 kN

(a) AT STIRT OF HORIZONTAL FLIGHT THRUST AND DRAG ARE STILL THE SAME

IF=ma F-D=ma (135,0-86.5) ×103N=(16×103kg)a a = 3.03 m/5"

(6) AT MAX, SPEED IN HURICONTAL FLIGHT

WE HAVE a=0 (1) $F_m - D_m = 0$

Fm = dm (11-Vn1) = (300 kg/s) (665 m/s - Vm) (2)

ON THE OTHER HAND

$$D_m = k V_m^2 \tag{3}$$

BUT, INITIA LLY, WE HAD D= 86.50 KN AND U= 774 km/h = 215 m/s AND, THEREFORE

$$D = \Re v^{2}$$

$$86.50 \times 10^{5} N = \Re (215m/s)^{2}$$
(4)

DIVIDING (3) AND (4) MEMBER BY MEMBER:

86.50×103 = (215)1 Dm = 1.8713 Vm

SUBSTITUTING FOR FM FROM (2) AND FUR D. FROM (5) INTO (1):

300 (665 - Vm) - 1.8713 Vm = 0

Vm + 160.32 Vm - 106.61 x10=0 $V_m = \frac{-160.32 + \sqrt{(160.32)^2 + 4(106.61 \times 10^3)}}{2.56.05} = 256.05 \text{ m/s}$

= (256,05 m/s) 3600 s -= : 921.78 km/h

15 = 922 km/h

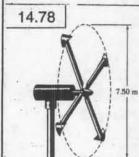
GIVEN:

WIND TURBINE-GENERATOR'S OUTPUT-POWER RATING IS 5KW FOR 30 km/h WIND SPEED.

FIND FOR THAT WIND SPEED (a) KINETIC ENERGY OF AIR PARTICLES ENTERING CIRCLE PER SECOND.

(b) EFFICIENCY OF THIS ENERGY-CONVERSION SYSTEM. (ASSUME C=1.21 bg/m3 FOR AIR.)

(a) KINETIC ENERGY PER SECOND $= \frac{1}{2} \frac{\Delta m}{\Delta t} v^{2} = \frac{1}{2} (Qv^{2} = \frac{1}{2}(QAv)v^{2} = \frac{1}{2$



GIVEN:

WIND TURBINE GENERATOR PRODUCES 28 KW OF ELECTRIC POWER WITH AN EFFICIENCY OF 0.35 AS AN ENERGY -CONVERSION SYSTEM

FIND:

(a) KINETIC ENERGY OF AIR

PARTICLES ENTERING CIRCLE

PER SECOND

(b) WIND SPEED

(ASSUME P=1.21 kg/m² POR AIR.)

(a) KINETIC ENERGY PPR SECOND = 28 kW = 80 kJ/s = EFFICIENCY = 0.35

(6) WIND SPEED

K.E. PER SECON = $\frac{1}{2} \frac{\Delta m}{\Delta E} v^2 = \frac{1}{2} ? Q v^2 = \frac{1}{2} ? (Av) v^2 = \frac{1}{2} ? Av^3$ THEREFORE: $80 \text{ kJ/s} = \frac{1}{2} (1.21 \text{ kg/m}^3) \frac{\pi}{4} (7.50 \text{ m})^2 v^3$ $v^3 = 2793.1 \quad v = 14.411 \text{ m/s} \quad v = 51.9 \text{ km/h}$

14.79 GIVEN:

PLANE CRUISING IN LEVEL FLIGHT AT 600mi/b SCOOPS IN AIR AT RATE OF 200 lb/s AND DISCHARGES IT AT 2200 ft/s RELATIVE TO PLANE.

FIND: (A) POWER USED TO PROPEL PLANE,
(b) TOTAL ENGINE POWER (C) EFFICIENCY OF PLANE

(a) PROM EQ.(14.39); THRUST = $F = \frac{dm}{dt}(v_B^- v_A)$, WHERE $v_B = 2200 \text{ ff/s}$. $v_A = 600 \text{ mi/h} = 880 \text{ ft/s}$ $F = \frac{200 \text{ lb/s}}{32.2 \text{ ft/s}^2}(2200 - 880) \text{ ft/s} = 8,198,8 \text{ lb}$

PROPULSIVE POWER = FV = (8,198,816)(880 ft/s) = 7,2149 X106 ft-16/s = 13,120 hp

(b) POWER LOST IN EXHAUST = 1 am VEX = 1 200 (2200 - 880) = 5.4112×106 ft.1b/s = 9.838 MP

TOTAL POWER = 13,120 hp+ 9,838 hp = 22,960 hp

(c.) EFFICIENCY = 13,120 hp = 0,571

14.80 GIVEN:

PROPELLER OF SMALL PLANE HAS 6-FLDIAMETER SLIPSTREAM AND PRODUCES 800-ID THRUST
WHEN PLANE IS AT REST ON GROUND.

FIND: (a) SPEED OF THE AIR IN THE SLIPSTREAM,
(b) VOLUME OF AIR PASSING THROUGH TROPELLER PERSECOND.
(C) KINETIC ENERGY IMPARTED TO THE AIR PER SECOND.

(a) SPEED V OF AIR

APPLY EQ. (14.39), ASSUMING AIRENTERS SLIPSTREAM WITH ZERO VELOCITY:

THRUST = $F = \frac{dm}{dt}v = PQv = \frac{1}{8}(Av)v = \frac{7}{4}Av^2$

800 1b = 0.076 1b/ft \$ \$ (6ft) \$ 0 L

(ASSUME & = 0.076 16/ft) FOR AIR.)

80016 = $(0.066734 | b \cdot s^2/ft^2) v_3^2$ $v^2 = 11.988 ft^2/s^2$ v = 109.49 ft/s v = 109.5 ft/s

(C) KINETIC ENERGY IMPARTED TO AIR PER SECOND $\frac{\Delta T}{\Delta t} = \frac{1}{2} \frac{dm}{dt} v^2 = \frac{1}{2} ((Av) v^2 = \frac{1}{2} (\frac{1}{3} Av^2) v = \frac{1}{2} Fo$ $= \frac{1}{2} (800 \text{ Ib}) (109.49 \text{ ft/s}) = 43.800 \text{ ft/ls/s}$



GIVEN: PELTON-WHEEL TURBINE,
RATE AT WHICH WATER IS DEFLECTED BY BLADES EQUALS RATE
AT WHICH WATER ISSUES FROM
NOZZLE: AM/At = PAVA.
PIND: (a) VELOCITY V OF
BLADES FOR MAXIMUM POWER,
(b) MAXIMUM POWER,

(C) MECHANICAL EFFICIENCY.
(USE NOTATION OF SP 14,7)

IMPULSE-HOMEN TUM PRINCIPLE
AS IN SAMPLE PROB. 14.7:

\$\frac{1}{2} \comp: (\Deltam) \times - \bar{F} \Delta = (\Deltam) \times \cos \theta \\

BUT IVOW \Deltam = A \bar{V} \Delta \Delta = \times \Delta - \times \\

THUS: \bar{F}_{\tau} = A \bar{V} \Delta \Big(\varV_{\tau} - \varV \Big) (1 - \cos \theta) \\

OUTPUT POWER = \bar{F}_{\tau} \varV = A \bar{V} \Delta \Big(\varV_{\tau} - \varV \Big) (1 - \cos \theta) \\

OR: OUTPUT POWER = A \bar{V} \Delta \Big(\varV_{\tau} - \varV \Big) (1 - \cos \theta) \\

OR: OUTPUT POWER = A \bar{V} \Delta \Big(\varV_{\tau} - \varV^2 \Big) (1 - \cos \theta) \\

OR: OUTPUT POWER = A \bar{V} \Delta \Big(\varV_{\tau} - \varV^2 \Big) (1 - \cos \theta) \\

OR: OUTPUT POWER = A \bar{V} \Delta \Big(\varV_{\tau} - \varV^2 \Big) (1 - \cos \theta) \\

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OR: OUTPUT POWER = A \bar{V} \Delta \Big(\varV_{\tau} - \varV^2 \Big) (1 - \cos \theta) \\

OR: OUTPUT POWER = A \bar{V} \Delta \Big(\varV_{\tau} - \varV^2 \Big) \Big(\varV_{\tau} - \varV^2 \Big) \Big(\varV_{\tau} - \var

(a) FOR MAX, POWER: d(POWER)/dV = 0: $APV_A(V_A - 2V)(1-\cos\theta) = 0$ $V = \frac{1}{2}V_A$

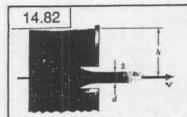
(b) MAX, POWER: MAKE V = 1 VAIN EQ (1): MAX, POWER = APVA(VA-1VA)(1-650) VA = 14AP(1-650) VA

(C) EFFICIENCY INPUT POWER = $\frac{1}{Z} \frac{\Delta m}{\Delta t} U_A^2 = \frac{1}{Z} (\Theta 0) U_A^2 = \frac{1}{Z} (\Theta A U_A) U_A^2$ = $\frac{1}{Z} A \Theta U_A^3$ (2)

DINIDE (1) BY (2): $\mathcal{D} = \frac{\delta U T P U V T P U W T R}{I N P U T P U W T R} = \frac{A P V_A (V_A - V)(1 - \cos \theta) V}{\frac{1}{2} A P V_A^3}$ $\mathcal{D} = 2 \frac{V}{V_A} (1 - \frac{V}{V_A})(1 - \cos \theta)$

NOTE: HAXIMUM EFFICIENCY IS OBTAINED WHEN V= 1/2/2

DAMAX = 2 (1/2)(2) = 1



GIVEN:

CIRCULAR REENTRANT

ORIFICE (BORDA'S

MOUTHPIECE)

V=0, V2=V=V2gh

SHOW THAT:

d = D/NZ

WE APPLY IMPULSE-MOMENTUM PRINCIPLE TO SECTION OF WATER INDICATED BY DASHED LINE AND TO MASS OF WATER DM ENTERING AND LEAVING IN DE.



 $\pm_{x} \times comp$: $0 + P\Delta t = (\Delta m) v = (P \Delta \Delta t) v = (P A_{x} v \Delta t) v$ THUS: $P = (P A_{x} v^{2} = (P A_{x} v^{2} v^{2} + (P A_{x} v^{$

BUT, RECALLING THAT THE PRESSURE AT A DEPTH h 15 p= @gh, WE HAVE

SUBSTITUTING THIS EXPRESSION IN (1) ATIE THE GAPRESSION GIVEN FOR V!

ession given For
$$V$$
:
$$egh_{\frac{\pi}{4}}D^{z} = e_{\frac{\pi}{4}}d^{z}(2gh)$$

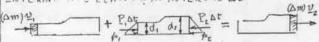
$$D^{z} = 2d^{z}$$

$$d = \frac{D}{\sqrt{z}}$$
(Q.E.D.)



GIVEN:
HYDRAULIC JUMP.
CHANNEL WIDTH = b.
EXPRESS RATE OF FLOW
Q IN TERMS OF b,d,d,

WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE TO THE WATER SECTION SHOWN AND TO THE MASS OF WATER AND ENTERING ALL LEAVING IN INTERVAL AL.



 $t_{3} \propto conp.: (\Delta m) V_{1} + P_{1} \Delta t - P_{2} \Delta t = (\Delta m) V_{2}$ $(P Q \Delta t) V_{1} + P_{1} \Delta t - P_{2} \Delta t = (P Q \Delta t) V_{3}$ $(P Q (V_{1} - V_{2}) = P_{2} - P_{1}$ (1)

BUT
$$Q = A_i v_i = b d_i v_i$$
 $v_i = Q/b d_i$ (2)

AND
$$Q = A_2 v_2 = b d_2 v_2$$
 $v_2 = Q'/b d_2$ (3)

ALSO:
$$P_i = \frac{1}{2}p_i A_i = \frac{1}{2}(8d_i)(6d_i) = \frac{1}{2}86d_i^2$$
 (4)

SIMILARLY:
$$P_{z} = \frac{1}{2}\delta b d_{z}^{2}$$
 (5)

SUBSTITUTE FROM (2), (3), (4), (5) INTO (1):
$$\left({^{C}Q} \left(\frac{Q}{bd_1} - \frac{Q}{bd_2} \right) = \frac{1}{2} \chi_b \left(d_2^2 - d_1^2 \right) \right)$$

$$Q^2 \frac{d_2 - d_1}{b d_1 d_2} = \frac{1}{2} \delta b \left(d_2 + d_1 \right) \left(d_2 - d_1 \right)$$

DIVIDING THROUGH BY dz-d, AND RECALLING THAT 8=Pg:

$$\frac{Q}{Ed_1d_2} = \frac{1}{2}g b (d_1 + d_2)$$

$$Q = b \sqrt{\frac{1}{2}} g d_1d_2 (d_1 + d_2)$$

* 14.84

GIVEN: FOR CHANNEL OF PRUB. 14.83: b=12 ft, d=4ft, d=5-ft FIND: RATE OF FLOW.

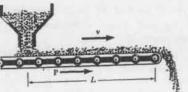
SEE SOLUTION OF PRUB. 14.83 FOR DERIVATION OF

Q = b \(\frac{1}{2} \) g d de (d+d2)

SUBSTITUTING THE GIVEN DATA :

Q = (12 ft) $\sqrt{\frac{1}{2}}$ (3 2,2 ft/s=)(4 ft)(5 ft)(9 ft) Q = 646 ft/s

14.85



GRAVEL FALLS ON CONVEYOR BELT WITH NO VELOCITY AND AT THE CONSTANT RATE q = dm/dt

(a) FIND MAGNITUDE OF PORCE P REQUIRED TO MAINTAIN A CONSTANT BELT SPEED.

(b) SHOW THAT K, E, REQUIRED BY GRAVEL IN GIVEN TIME INTERVAL IS HALF THE WORK DONE BY P. WHAT HAPPENS TO THE OTHER HALFOF WORK OF P?

(a) WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE TO THE GRAVEL ON THE BELT AND TO THE MASS DAY OF GRAVEL HITTING AND LEAVING BELT IN INTERVAL DE

 $P = \frac{\Delta m}{\Delta t} v = qv \qquad P = qv$

(b) KINETIC FIVERBY ACQUIRED PER UNIT TIME:

$$\frac{\Delta T}{\Delta t} = \frac{1}{2} \frac{\Delta m}{\Delta t} v^2 = \frac{1}{2} q v^2 \tag{1}$$

WORK DONE PER UNITTIME:

$$\frac{\Delta U}{\Delta t} = \frac{P\Delta z}{\Delta t} = PU$$

RECALLING THE RESULT OF PART a:

$$\Delta U = (q v) v = q v^2$$
 (2)

COMPARING ERS. (1) AND (2), WE CONCLUDE THAT

$$\frac{\Delta T}{\Delta L} = \frac{1}{Z} \frac{\Delta U}{\Delta L} \qquad (Q, E, D.)$$

THE OTHER HALF OF THE WORK OF P IS
DISSIPATED INTO HEAT BY PRICTION AS
THE GRAVEL SLIPS ON THE BELT BEFORE
REACHING THE SPEED W.



GIVEN: CHAIN OF LENGTH & AND MASS M FALLS THROUGH SMALL HOLE IN PLATE. CHAIN IS AT REST WHEN Y IS VERY SMALL.

FIND IN EACH CASE SHOWN'S

(A) ACCELERATION OF FIRST LINK A MS FUNCTION OF Y.

(b) VELOCITY OF CHAIN AS LAST LINK PASSES THRU HOLE.

IN: CASE 1; ASSUME THAT EACH LINK IS AT REST UNTIL

IT FALLS THRUHOLE

IN CASE 2, ASSUME THAT ALL LINKS HAVE THE SAME SPEED AT ANY GIVEN INSTANT

CASE 1: WE APOLY THE IMPULSE-MOVIENTUM PRISCIPLE TO THE PORTION OF CHAIN WHICH HAS ALREADY PASSED TROUGH THE HOLE AT TIME & AND TO THE PORTION WHICH WILL PASS IN INTERVAL AT.

$$\frac{\Delta m_Z (\text{ND HOM})}{1 \text{ gran}} + \frac{1 \text{ gran}}{1 \text{ gran}} \frac{1 \text{$$

DIVIDE BY ST AND LET ST +0; gy = y dy + v dy at = d(yv)

MULTIPLY BOTH SIDES BY yodt AND NOTE THAT odt = dy:

SET gr= u AND INTEGRATE!

$$\int_{0}^{3} g \, y^{2} dy = \int_{0}^{3} v \, du$$

$$\frac{1}{3} g \, y^{3} = \frac{1}{2} \left(\frac{3}{3} v \right)^{2} \qquad v^{2} = \frac{2}{3} g \, y \qquad (1)$$

(a) DIFFERENTIATE (I) WITH RESPECT TO t:

 $2v\frac{dv}{dt} = \frac{9}{3}g\frac{dy}{dt}$ or $2va = \frac{2}{3}gv$ $a = \frac{1}{3}g$

(b) AS LAST LINK PASSES THROUGH HOLE, $g = \ell$ AND EQ () YIELDS $v^2 = \frac{2}{3}g\ell$ $v = \sqrt{\frac{2}{3}g\ell}$

CASE 2. (a) AT time t, THE PORCE CAUSING THE ACCEL-BRATION OF THE ENTIRE CHAIN IS THE WEIGHT OF THE LENGTH & OF CHAIN WHICH HAS PASSED THEOUGH

$$\overline{J}$$
 \overline{J} \overline{J}

INTEGRATING IN & FROM O TO IT AND IN & FROM O TO E

\(\frac{1}{2} \nu = \frac{1}{2} \frac{1}{2} \frac{1}{2} \]
\(\nu = \sqrt{ge} \)

14.87 A

GIVEN:

CHAIN OF LENGTH & AND MASS M 15 LYING IN A PILE ON FLOOR, IT IS RAISED AT A CONSTANT V. FIND FOR ANY Y;

(a) MAGHITUDE OF FURCE P.

(a) WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE TO THE LENGTH & OF CHAIN WHICH IS OFF THE FLOOR AND TO THE LENGTH AY WHICH WILL BESET IN MOTION DURING THE TIME INTERVAL AT.

$$\frac{\partial}{\partial t} \int_{0}^{\infty} \frac{dt}{dt} dt = \int_{0}^{\infty} \frac{dt}{dt} dt = \int_{0}^{\infty} \frac{dt}{dt} dt$$
(course of

+14 COMP: my U+Pat-my = at = m. 3+04 v

PAt = # (9406-40+34+ vay)

DIVIDING BY Δt : $P = \frac{m}{e} (gy + v \frac{\Delta t}{\Delta t})$

NOTING THAT $\Delta z/\Delta t = v'$, $P = \frac{m}{\ell}(g\dot{z} + v^2)$

(b) THE REACTION OF THE FLUOR IS EQUAL TO THE WEIGHT OF CHAIN STILL ON THE FLUOR:

 $R = mg - mg\frac{H}{\ell}$ $R = mg\left(1 - \frac{H}{\ell}\right)$

14.88

GIVEN:
CHAIN OF LENGTH & AND MASS
M IS LOWERED INTO A PILE ON
THE FLOOR AT CONSTANT V
FIND FOR ANY Y:

(a) MAGNITUDE OF FORCE P. (b) REACTION OF THE FLOOR.

a) P IS EQUAL TO THE WEIGHT OF CHAIN STILL OFF THE FLOOR: P= mgy/l

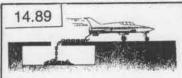
(DWE APPLY THE IMPULSE MOMENTUM PRINCIPLE TO THE LENGTH &- Y OF CHAIN ON THE FLOOR AND TO THE LENGTH DY WHICH INTO THE FLOOR IN DE:

+A 1 COMP: - m A U - 2 (l-y + Ay) At + R At = 0 SOLVING FOR R:

R= #[9(6-4+03)+0 A4]

BUT AS = T AND AY -O WHEN At -O.

THEREFORE: $R = \frac{om}{\ell} [g(\ell-g) + \upsilon^2]$



GIVEN:

AS PLANE OF MASS M LANDS WITH VOON CARRIER, ITS TAIL HOOKS INTO END OF CHAIN OFLENGTH &.

FIND: (a) MASS OF CHAIN REQUIRED TO REDUCE PLANE SPEED TO BY (WHERE B < 1), (b) MAX, FORCE EXERTED BY CHAIN ON PLANE.

LET M' = MASS OF CHHIN PER UNITLENETH X = DISTANCE TRAVELED AT TIME t

CONSERVATION OF LINEAR MOMENTUM (m+m2)0

to x cuma: MV = (m+m'x)v (a) WE WANT V = BYO FOR X = P. SUBSTITUTE: mvp = (m+m'E)Bvo

m v (1-B) = m'lB vo (2) MASS OF CHAIN = m'l = 1-13 m

(6) SOLVE EQ. (1) FOR V: (3)

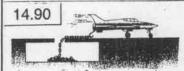
 $a = \frac{dv}{dt} = -\frac{mv_0}{(m+m'x)^2}m'\frac{dx}{dt} = -\frac{mm'v_0v}{(m+m'x)^2}$

OR, AECALLING(3): $\alpha = -\frac{m^2m^3v_0^2}{(m+m^2x)^3}$ (4)

DECELERATION IS MAXIMUM FOR 2C=0. WE HAVE $(-a)_{max} = \frac{m^2 m^3 v_0^2}{m^3} = \frac{m^3 v_0^2}{m}$

WRITING | F | max = m | a | max AND RECALLING (2):

IFImax = m1 v 2



GIVEN:

AS 6000- hg PLANE LANDS AT IBOKM/h ON CARRIER, ITS TAIL HOOKS INTO END OF

80-m-LONG CHHIN OF MASS OF 50 Kg/m.

EIND: (a) PINX. DECELERATION OF FLANE, (6) VELOCITY WHEN ENTIRE CHAIN IS PULLED DUT

SEE SULV TION OF PRUB, 14.89 FOR DERIVATION OF EQS. (3) AND (5).

(a) FROM EQ. (5): MAX. DECEL. = $(-a)_{max} = \frac{m!}{m!} v^2 = \frac{50 \text{ kg/m}}{6000 \text{ kg}} \left(\frac{180}{3.6} \text{ m/s}\right)^2$ MAX. DECEL. = 20.8 m/52

(b) FROM EQ. (3), FOR x = l = 80 m: $\overline{v_{\text{max}}} = \frac{m v_0}{(m+m^2 \ell)} = \frac{(6000 \, \text{kg})(180 \, \text{km/h})}{6000 \, \text{kg} + (50 \, \text{kg/m})(80 \, \text{m})}$ Vmax = 108.0 Km/h



GIVEN:

EACH OF THE THREE ENGINES OF SPACE SHUTTLE BURNS PROPELLANT AT RATE OF 340 Kg/S AND EJECTS IT WITH A RELATIVE VELOCITY OF 3750 m/s

FIND: TOTAL THRUST PROVIDED BY THE THREE ENGINES

FROM EQ. (14.44) FOR EACH ENGINE P= dm u = (340 kg/s)(3750 m/s) = 1,275 x 106 N FOR THE 3 ENGINES : TOTAL THRUST = 3(1.275×106N) = 3.83MN

14.92

GIVEN:

THE THREE ENGINES OF SPACE SHUTTLE PROVIDE . A TOTAL THRUST OF 6 MN. PROPELLANT IS EJECTED WITH A RELATIVE VEL. OF 3750 m/s.

FIND: RATE AT WHICH PROPELLANT IS BURNED BY EACH OF THE THREE PNGINES.

THRUST OF EACH ENGINE: P= 1 (6MH)= 2×106 N EQ. (14.44): P= # 4 2 × 106 N = dm (3750 m/s) dm - 2×10°N = 533 kg/s

GIVEN: 14.93

ROCKET FIRED VERTICALLY FROM GROUND WEIGHT OF ROCKET (INCLUDING FUEL) = 2400 / WEIGHT OF FUEL = 2000 16

FUEL EJECTED AT RATE OF 25 16/5 WITH RELATIVE VELOCITY OF 12,000 fts.

FIND; ACCELERATION OF ROCKET (a) AS IT IS FIRED,

(6) AS LAST PARTICLE OF FUEL IS BEING CONSUMED

EQ. (14. 44): P= dm u= 25 16/5 (12,000 ft/s)= 300 x 103 +12F=ma: $a = \frac{P}{m} - \frac{w}{m} = \frac{(300 \times 10^3)/9}{w/9} - g$ = 1 ma $a = \frac{300 \times 10^3}{W} - g$

(a) AS ROCKET IS FIRE D:

 $a = \frac{360 \times 10^{9}}{2100} - 32,2 = 125.0 - 32,2 = 92,8$ W= 2 400 16 PROM(1): a = 92,8 ft/s2 4

(b) AS LAST PARTICLE OF FUEL IS BEING CONSUMED: W= 2400-2000 = 40016

PROM(1): $\alpha = \frac{300 \times 10^{3}}{400} - 32.2 = 750 - 32.2 = 717.8$ a = 718 ft/s=1

GIVEN: 14.94

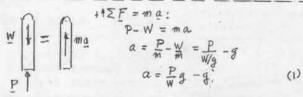
ROCKET FIRED VERTICALLY PROM GROUND. WEIGHT OF ROCKET (INCLUDING PUEL) = 3000 16

WEIGHT OF FUEL = 2500 16.

FUEL CONSUMES AT RATE OF 30 16/s. ACCELERATION IN CREASES BY 750 ft/s" FROM TIME ROCKET IS FIRED TO TIME WHEIL LAST PARTICLE OF FUEL IS CONSUMED.

FIND:

RELATIVE VELOCITY WITH WHICH IS EJECTED



AS ROCKET IS FIRED EQ. (1) YIELD.

$$a_b = \frac{P_F^2}{3000/b} - g \tag{2}$$

WHEN LAST PARTICLE IS FIRED!

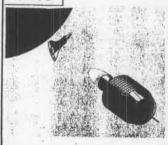
$$a_0 + 750 \, tt/s^2 = \frac{Pg}{50016} - g$$
 (3)

SUBTRACT (2) FROM (3): 750 = Pg $(\frac{1}{500} - \frac{1}{3000})$ 750 = (1.6667×10-3) Pg P = 450×103

BUT, PROM EQ. (14.44):

$$P = \frac{dm}{dt} u : \frac{450 \times 10^3}{g} = \frac{30 \text{ lb/s}}{g} u = 15,000 \text{ ft/s}$$

14.95



GIVEN:

SATELLITE IS FIRED TO INCREAGE ITS VELOCITY BY 8000 ft/s. WEIGHT OF SATELLITE (INCLUDING FUEL) = 10,000 lb. FUEL CJECTED WITH RELATIVE VEL. OF 13,750 HS FIND!

ENGINE OF COMMUNICATION

WEIGHT OF FUEL CONSUMED

WE APPLY IMPULSE-MOMENTUM PRINCIPLE TO ATELLITE AND FUEL EXPELLED IN INTERVAL AL

+> mv = (m-Am)(v+1v)- Am(u-v-Av) mos = priv - tram + man - uam + vam + second-order terms mov=uom

BUT Dom = q Dt AND M= m - qt THUS (Mo-9t) DV = ug Dt

$$t \to 0: \frac{dJ}{dt} = \frac{uq}{m_0 - qt}$$

$$v = \int_0^t \frac{uq}{m_0 - qt} dt = -u \left[\ln \left(m_b - qt \right]_0^t \right]$$

$$V = u \ln \frac{m_0}{m_b - qt}$$

(1)

(CONTINUED)

14.95 continued

EXPRESSING EQ. (1) IN EXPONENTAL FORM:

$$\frac{m_0}{n_0 - qt} = e^{v/u} \tag{2}$$

SETTING Mo=(10,000 1b)/g, U= 8000 ft/s, 4= 13,750 ft/s, AND EXPRESSING Q IN 16/s, WE HAVE

$$\frac{10,000/g}{(10,000-9c)/g} = e^{\frac{90000}{13,750}} = e^{\frac{0.52182}{1,7893}} = 1,7893$$

$$10,000-9t=\frac{10,000}{1,7893}=5,588.8$$
 9t = 4,411.2 16

WEIGHT OF PUEL EXPANDED = 9t = 4410 16

14.96 GIVEN: COMMUNICATION SATELLITE OF PROB 14.95 FIND: INCREASE IN VELOCITY AFTER 2500 16 HAS BEEN CONSUMED.

SEE SOLUTION OF PROB 14,95 FOR DERIVATION OF ED. (1). $v = u \ln \frac{m_0}{m_0 - qt}$

FROM DATA OF PROBS, 14, 95 AND 14, 96: m = 13,750 ft/s, m = (10,000.16)/g, 9t = (2,500 16)/g SUBSTITUTE IN (1):

 $v = (13,750 \text{ ft/s}) \ln \frac{10,000/g}{7,500/g} = (13,750 \text{ ft/s}) \ln (1.3333)$ v = 3960 ft/s

14.97

GIVEN:

A 540-Kg SPACECRHET IS MOUNTED ON TOP OF ROCKET OF MASS OF 19 Mg, INCLUDING 17.8 Mg OF FUEL. FUEL IS CONSUMED AT THE RATE OF 225 kg/s AND EJECTED WITH A RELATIVE VELOCITY OF 3600 m/s.

FIND:

MAXIMUM SPEED OF SPACECRAFT IF ROCKET IS FIRED VERTICALLY FROM THE GROUND.

SEE SAMPLE PROB. 14.8 FOR DERIVATION OF (1) v=uln mo -gt

DATA:

u = 3600 m/s, q = 225 kg/s, moduel = 17 800 kg m = 19000 kg + 540 kg = 19540 kg

WE HAVE MSWI = 9t, 17800 kg = (225 kg/s) E $t = \frac{17800 \text{ kg}}{225 \text{ kg/s}} = 79.111 \text{ s}$

MAX. VEWCITY IS REACHED WHEN ALL FUEL HAS BEEN CONSUME, THAT IS, WHEN 9t = MSud . EO. ()YITLDS

$$v_m = u \ln \frac{m_0}{m_0 - m_{fuel}} - gt$$

= (3600 m/s) $\ln \frac{19540}{19540 - 17600} - (9.81 m/s^2)(79.1115)$
= (3600 m/s) $\ln 11.230 - 776.1 m/s = 7930.8 m/s$
 $v_m = 7930 m/s$

GIVEN:

A 540-kg SPACECRAFT 15 MOUNTED ON A TWO-STAGE ROCKETS EACH STAGE ITAS A MASS OF 9,5 Mg, INCLUDING B.9 Mg OF FUEL. FUEL IS CONSUMED AT A RATE OF 215 kg/s AND EJECTED WITH A RELATIVE VELOCITY OF 3600 m/5. AS STAGE A EXPELS ITS LAST PARTICLE OF FUEL ITS CASING IS JETTISONED. FIND: (a) SPEED OF RUCKET AT THAT INSTANT,

(b) MAXIMUM SPEED OF SPACECRAFT

SEE SAMPLE PRUB. 14.8 POR DERIVATION OF v=uln mo-gt-gt (1)

(a) FIRST STAGE u = 3600 m/s , 9 = 225 kg/s , MASS OF FUEL = mg = 8900 kg mo = 2 (9500 kg) + 540 kg = 19540 kg WE HAVE $m_s = qt_1$ $t_1 = \frac{m_s}{q} = \frac{8900 k_0}{225 km}$ t,= 39,556 s SUBSTITUTE INTO (1): 19540 v = (3600 m/s) la 19540 -(9.81 m/s²) (39.5560) = (3600 m/s) ln 1.8365 - 388.04 m/s = 1800.3 m/s N = 1800 m/s

(b) SECOND STAGE u = 3600 m/s, 9 = 225 kg/s, MASS OF PUZL= ing = 8900 kg m, = 9500 kg + 540 kg = 1040 kg t, = 39,556 = REPLACING V BY V2-V, AND MOBY M, IN EQ. (1): V2-V, = 4 ln m1 - 9t2 = (3600 m/s) ln 10040 - (9.81 m/s) (39.5565) = (3600 m/s) En 8,8070 - 388.04 m/s = 7444 m/s v= = v1 +7444 = 1800+7444 = 9244, v= 9240 m/s

GIVEN: SPACECRAFT OF PROB. 14,97. 14.99 FIND: ALTITUDE REACHED WHEN ALL THE FUEL OF THE LAUNCHING ROCKET IS CONSUMED.

WE RECALL DATA FROM PROB. 14.97 AND EQ(1). SETTING V = dy/dt, WE HAVE

dy = (ln mo - gt) dt fay = 50 (men aloge - gt) de = u 5 en mogt dt - 1/2 gt SETTING Ma-qt = , WE FIND THAT u fln mo at at = u fl - ln z) (-modz) = u mo [z ln z - z] THUS: h= shdy = um (2 ln z-2-0+1)- 1 gt2 $h = u \left[E + \frac{m_0 - 9E}{9} \ln \frac{m_0 - 9E}{m_0} \right] - \frac{1}{2} g t^2$

GIVEN DATA: 41 = 3600 m/s, 9 = 225 kg/s, m = 19540 kg. t = 79.111s, $qt = m_s = 17800 kg$, $m_s - qt = 19540 - 17800 = 1740 kg$ $h = 3600 \left[79.111 + \frac{1740}{225} ln \frac{1740}{19540} \right] - \frac{1}{2} (9.81) (79.111)^3$ $= 3600 \left(79.111 - 18.704 \right) - 30698 = 186770 m$ h=186.8km ◀ [NOTE THAT & WAS ASSUMED CONSTANT]

GIVEN: SPACECRAFT AND TWO-STAGE 14,100 LAUNCHING ROCKET OF PROB. 14.98.

FIND ALTITUDE AT WHICH (U) STAGE A IS RELEASED.

(b) FUEL OF BOTH STAGES HAS BEEN CONSUMED.

SEE SOLUTIONS OF SAMPLE PROB. 14:8 AND PROB. 14.99 FOR DERIVATION OF EQ. (2): $h = u \left[t + \frac{m_0 - qt}{q} \ln \frac{m_0 - qt}{m_0} \right] - \frac{1}{2} \xi^{t}$

(a) FIRST STAGE

PRUM PROB 14,98 WE HAVE tl=3600 m/s, 9=225 kg/s, mo=19 540 kg, t,=39,556 s 9t, = mg = 8900 kg, mo - 9t, = 19540 - 8900 = 10640 kg EQ.(2) YIELDS

 $h_1 = (3600)[39.556 + \frac{10640}{225} lm \frac{10640}{19540}] - \frac{1}{2}(9.8)[39.556]^2$ = (3600)(39.551-28.744) - 76747 = 31248 m h, = 31.2 km

(b) SECOND STAGE USING AGAIN PU. (2) AND ADDING h, AND UT to IT, $h_{z} = h_{1} + v_{1}t_{2} + u\left[t_{2} + \frac{m_{1} - qt_{2}}{q} \ln \frac{m_{1} - qt_{2}}{m_{1}}\right] - \frac{1}{2}g t_{2}^{2}$

FROM PROB. 14.90, WE HAVE \$\psi_1 = 1800.3 m/s, \$\pi = 3600 m/s, \$q = \$25 kg/s, \$\pi = 39.556 s\$ m1 = 10 040 kg, 9t = m5 = 8900 kg, m, -9t = 1140 kg

h2 = 31248 + (1800.3)(39,556) + 3600 [39,556 + 1140 Bn 1140] -1 (9.81)(39.556)2-

h2 = 31248+71213+3600 (39.556-11.023)-7675 h= 197,5 km = 197 500 m

GIVEN: 14.101

COMMUNICATION SATELLITE OF PROB. 14,95 FUEL CONSUMED AT RATE OF 37,5 16/5.

FIND: DISTANCE FROM SATELLITE TO SHUTTLE AT t = 605.

SEE SOLUTION OF PROB 14.95 FOR DERIVATION OF (1)

v=u ln mo-95 SETTING U = dx/dt, WE HAVE

dz = (u ln mo-gt) dt (1) x=5 (uln mo-9t) dt = - ust n mo-9t dt

SETTING MO-9t = 2 WE HAVE dt = - Mode AND x = mou seln zdz = mou [z ln z - z] = mou (z ln z - z+1)

 $= \frac{m_1 u}{q} \left(\frac{m_0 - qt \ln \frac{m_0 - qt}{m_0} - 1 + \frac{qt}{m_0} + 1}{x = u \left(t + \frac{m_0 - qt}{q} \ln \frac{m_0 - qt}{m_0} \right)} \right)$ (2).

GIVEN DATA: 9=(37.5 1b/s)/9, t=605,

AND PROM PROB. 14,95: n= 13,750 +t/s, m= (10,0001b)/8 THUS: mo-qt = (10,000/g) - (37.5/g)(60) = (7750 16)/g

AFTER SUBSTITUTION, EQ. (2) YIELDS 2=(13,750 ft/s)(60 + 7750 fn 7750)s =(13,750)(60-52,678) = 100,680 ft =(100,680 ft) $\frac{1 \text{ ni}}{5280 \text{ ft}} = 19.068 \text{ mi}$

X=19,07 mi

233

GIVEN:

ROCKET OF PROB. 14.93.

FIND: (a) ALTITUDE AT WHICH ALL FUEL IS CONSUMED, (6) VELOCITY OF ROCKET AT THAT TIME.

SEE SAMPLE PROB. 14.8 FOR DERIVATION OF v=uln mo - gt

AND SOLUTION OF PROB. 14. 99 POR DERIVATION OF $h = u \left[t + \frac{m_0 - qt}{q} \ln \frac{m_0 - qt}{m_0} \right] - \frac{1}{2} gt^2$

PROM STATEMENT OF PROB. 14. 93, WE RECALL u = 12,000 ft/s, mo = (24001b)/g, qt = mg = (2000 1b)/g t= mi = 2000 15 = 805

(a) ALTITUDE AT WHICH ALL FUEL IS CONSUMED SUBSTITUTING DATA IN EQ (2):

h = (12,000 ft/s)[80 + 2400 - 2000 en 3400 - 2000]s. - + (32.2 ft/s1)(BOS)2

h = (12,000)(80-28,668) - 103,040 = 512,944 ft h = 512,944 = 97.148 mi h = 97.1 mi

(b) VELOCITY OF RUCKET AT THAT TIME

SUBSTITUTING DATA IN EQ. (1):

2400 V = (12,000 ft/s) en 2,000 - (32,2 ft/s*)(805) = 12,000 ln 6 - 2576 = 18,925 ft/s

v = 18, 930 ft/s

14.103

GIVENI JET AIRPLANE WITH "V = SPEED OF AIRPLANE

IL = RELATIVE SPEED OF EXPELLED GASES SHOW THAT MECHANICAL EFFICIENCY 15 9 = 20

EXPLAIN WHY 7=1 WHEN U= U.

THRUST P IS OBTAINED FROM EN. (14.39):

WHERE WA = W = AIRPLANE SPEED B= 46 = EXHAUST VEL, REL, TO PLANT

THUS! F= #(4-v)

USEPUL POWER = FV = dm (u-v)v

WASTED POWER = K.E. IMPARTED PER SECOND TO EXHAUST GASES WHOSE ABSOLUTE VEL, 19 41-V. = 1 dm (u-v)2

TUTAL POWER = USEFUL POWER + WASTED POWER = dm [(u-v)v+ 1/2(u-v)2] = dm (4x5-v2+ 1/2 12+ 1/2 v-4x0)

= = = dm (ut-v2) = = dm (u+v)(u-v)

EPPICIENCY = 9 = USEFUL POWER = (u-v)'V (Q.E.D.)

WHEN MET, THE ABSOLUTE VELOCITY MINT OF THE EXPELLED GASES IS ZERO. THUS, NO FIVERBY IS IMPARTED TO THE EXPELLED GASES AND NO POWER IS WHSTED.

14.104

GIVEN:

ROCKET WITH SPEED 10, EXPELLING FUEL WITH RELATIVE SPEED &.

SHOW THAT MECHANICAL EFFICIENCY 15 3 = 240(1+v). EXPLAIN WHY 9=1 WHEN LI = U.

WE RECALL EN (14.44) FOR THEMST P OF ROCKET! . F = dm en

USEFUL POUTER = PU = WMU

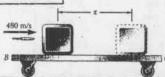
WASTED POWER = K.E. ENERGY IMPORTED PER SECUND TO EXPELLED FUEL WHOSE ABSOLUTE VELOCITY IS LL-U. == = == (4-1)2

TUTAL POWER = US EFUL POWER + WASTED POWER = om uv+ + of (u-v) = 1 dm (24+ 42++2-244) = 1 dt dm (12+15")

EFFICIENCY = TO TALL POWER (Q. E.D.)

WHEN U=V, THE ABSOLUTE VELOCITY 11-V OF THE EXPELLED PUEL IS ZERD . THUS, IND ENTERGY 15 IMPARTED TO THE EXPELLED FUEL AND NO POWER IS WASTED.

14,105



GIVEN:

30-9 BULLET FIRED WITH U = 480m/s INTO 5-kg BLOCK A, WHICH RESTS ON 4- KG CAST C. ML = 0.50 BETWEEN BLUCK A AND CARTC.

FIND (a) FINAL VELOCITY U, OF CARTAND BLUCK, (b) FINAL POSITION OF BLOCK ON CART.

CONSERVATION OF LINEAR MOMENTUM

mo to = (mo + mA) v = (mo + mA + mc) vs

(0,030kg) (480 m/s) = (5,030 kg) v' = (9,030 kg) v+

 $U' = \frac{0.030}{5.030} (480 \, \text{m/s}) = 2.863 \, \text{m/s}$

 $U_f = \frac{0.030}{9.030} (400 \,\text{m/s}) = 1.5947 \,\text{m/s}$ Ng = 1.595m/s (a) ANSWER 15

(6) WORK - ENERGY PRINCIPLE

JUST AFTER IMPACT:

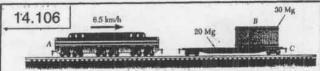
 $T' = \frac{1}{2}(m_0 + m_A) v^2 = \frac{1}{2}(5.030 \text{kg})(2.863 \text{ m/s}) = 20.615 \text{ J}$

FINAL KINETIC ENERGY: $T_{s} = \frac{1}{2} \left(m_{0} + m_{A} + m_{C} \right) v_{s}^{2} = \frac{1}{2} (4,030 \text{kg}) (1.5947 \text{m/s})^{2} = 11.482 \text{ J}$

WORK OF FRICTION FORCE:

F=14N=14(mo+ma)g=0,50(5,030)(9,81)=24,672 N WORK = U=- Fx = -24.6722

x=0.370 m T'+U= 7: 20,615-24,672x=11,482



GIVEN: BO-Mg ENGINE A WITH V= 6.5 km/h STRIKES 20-Mg FLATICAR C WHICH IS AT REST AND CARPLES 30-Mg LOAD B. A AND C ARE COUPLED UPON IMPACT. B CAN SLIDE ON C WITH M4 = 0,25 FIND VELOCITY OF CAR C (a) IMMEDIATELY AFTER IMPACT

(b) AFTER B HAS SLID TO A STUP RELATIVE TO C.

CONSERVATION OF LINEAR MOMENTUM

PIRST NOTE THAT E WILL NOT HOVE DURING COUPLING OF A AND C, SINCE THE FRICTION FORCE EXERTED ON B BY C IS NONIMPULSIVE: FAT= H NOT SO.

MAV = (MA+MC) v' = (MA+MC+NB) V (BOMg)(6.5 km/h) = (100 Mg) v'= (130 Mg) V+

(a)
$$v' = \frac{80}{100} (6.5 \text{ km/h})$$

(b)
$$V_f = \frac{80}{130} (6.5 \, \text{km/h})$$

14.107



GIVEN: THREE IDENTICAL CARS WITH VELOCITIES SHOWN, CAR B IS FIRST HIT BY CHA A. FIND FINAL YELDCITY OF ENCHORS IF (a) ALL CARS GET AUTOMATICALLY COUPLED, (b) A AND B GET COUPLED, BUT B AND C BOUNCE OFF EACH OTHER WITH e =1 (i.e. NO ENERGY LOSS).

(a) ALL CARS AUTOMATICALLY COUPLED CONSERVATION OF LINEAR MOMENTUM: mAVA+MBVB+mcVc=(mA+mB+mc)V5 m (6 mi/h) + 0 -m (4.8 mi/h) = (3 m) vz Vy = 6-4,8 = +0,4 Vy = 0,400 m/h-

(b) CARS A AND B ONLY GET COUPLED CONSERVATION OF LINERE MOMENTUM FOR A AND B:

$$A = A B m(6mi/h) = (2mi) V'$$

$$V' = 3mi/h$$

CAR C HITS AND BOUNCES OFF CHRS A AND B m (4.8 m/h) (2m)U" m vo 2711 (3 mi/h) B CONS. OF LINEAR MOMENTUM:

2 m (3) - m (4.8) = 2m "+ m "c (1) 2 v"+ v' = 1.2 mi/h

(CONTINUED)

continued 14.107

CONSERVATION OF ENERGY (e=1): RELATIVE VELOCITY AFTER AND BEFORE IMPACT ARE N' - U" = (3 + 4, 8) MI/H SUBTRACTING (2) FROM (1) v"= -2,20 mi/h 30"=1.2-7.0 U" = U" = 2, zomi/h ← THUS:

SUBSTITUTING J'= -2.20 mi/h M (1): 2 (-1,20 mi/h) + 0 = 1,2 mi/h v' = + 5.60 mi/h v' = 5,60 mi/h →

14.108 GIVEN:

9000-16 HELICOPTER A 13 TRAVELING DUE EAST AT 75 milh AT ALTITUDE OF 2500 ft WHEN IT IS HIT BY 12,000-16 HELICOPTER B. THEIR ENTANGLED WRECKAGE FALLS TO THE GROUND IN 12 s AT POINT LOCATED 1500 ft EAST AND 384 ft SOUTH OF POINT OF IMPACT

FIND VELOCITY COMPONENTS OF HELICOPTER B JUST BEFORE CULLISION, (NEGLECT AIR RESISTANCE,)

VELOCITY OF WRECKAGE IMMEDIATELY AFTER

COLLISION

UP 18

LAST

BUT:

$$x = U_{2}'t + V_{3}' \frac{1}{2} + U_{2}' \frac{1}{2}$$
 $x = U_{2}'t + U_{3}' \frac{1}{2} + U_{2}' \frac{1}{2}$
 $x = U_{2}'t + U_{3}' \frac{1}{2} + U_{2}' \frac{1}{2} = \frac{1500 \text{ ft}}{12 \text{ s}} = 125 \text{ ft/s}$
 $x = U_{2}'t + U_{2}' \frac{1}{2} \frac{1}{2} = \frac{384 \text{ ft}}{12 \text{ s}} = 32 \text{ ft/s}$
 $-h = U_{3}'t - \frac{1}{2}gt^{2}$
 $U_{3}' = -\frac{h}{t} + \frac{1}{2}gt$
 $u_{3}' = -15.133 \text{ ft/s}$

THUS: U' = (125ft/s) i - (15.133ft/s) j+(32ft/s) k

IMPACT: CONSERVATION OF LINEAR MOMENTUM MA VA + MB VB = (MA+MB) U'

AFTER SUBSTITUTING DATA AND EXPRESSION FOUND FOR T', AND NOTING THAT UA = 75 mi/h = 110 ft/s,

$$\frac{9000 \text{ lb} (110 \text{ ft/s}) \dot{i} + \frac{12,000 \text{ lb}}{9} \dot{y}_{B}}{\frac{21,000 \text{ lb}}{9} [(125 \text{ ft/s}) \dot{i} - (15, 133 \text{ ft/s}) \dot{j} + (32 \text{ ft/s}) \dot{k}}$$

SOLVING FOR 1/21

\$\frac{4}{5} = 1.75 [(125+t/s)i-(15.133+t/s)j+(32 \frac{5}{15})f-(82.5+f/s)i

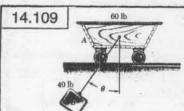
IT FOLLOWS THAT

(NB) = 1.75(125+45)-82,5 +t/s = 136.25+1/s = 92.90 mi/b

(VB) = - 1.75 (15,133 ft/s) = - 26,48 ft/s (VB) = 1,75(32 ft/s) = 56.0 ft/s = 38.18 mi/b

ANSWER:

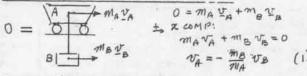
92,9 mi/h PAST, 38,2 mi/h SOUTH, 26,5 ft/s DOWN



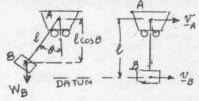
GIVEN:
BIOCK B ISSUSPENDED
FROM 6-FT CORD
ATTACHED TO CART H
SYSTEM IS RELEASED
FROM REST WHEN B = 35°.

FIND: VELOCITIES OF A AND B WHEN B = 0.

CONSERVATION OF LINEAR MOMENTUM



CONSERVATION OF ENERGY



INITIALLY: $T_0 = 0$ $V_0 = W_B \ell(1 - \cos \theta)$ = $m_B g \ell(1 - \cos \theta)$

AS B PASSES UNDERA:

$$T = \frac{1}{2} M_A V_A^2 + \frac{1}{2} m_B V_B^2 \qquad V = 0$$

To+Vo = T+ V:

 $m_B g l(1-\cos\theta) = \frac{1}{2} m_A v_A^2 + \frac{1}{2} m_B v_B^2$ $m_A v_A^2 + m_B v_B^2 = 2 m_B g l(1-\cos\theta)$

SUBSTITUTING FOR VA FROM (1):

$$m_A \left(\frac{m_B}{m_A}\right)^2 v_B^2 + m_B v_B^2 = 2 m_B g \ell (1 - \cos \theta)$$
 $\frac{m_B}{m_A} (m_A + m_B) v_B^2 = 2 m_B g \ell (1 - \cos \theta)$
 $\frac{m_A + m_B}{m_A} v_B^2 = 2 g \ell (1 - \cos \theta)$
 $v_B = \sqrt{\frac{z m_A}{m_A + m_B}} g \ell (1 - \cos \theta)$

GIVEN DATA:

$$m_A = \frac{W_A}{M_A + M_B} = \frac{60 \, lb}{60 \, lb + 40 \, lb} = 0.6$$
 $\ell = 6 \, ft, \quad \theta = 35^{\circ}$

TB = \(\frac{2(0.6)(32,2 ft/s^2)(6 ft)(1-cos35°)}{} = 6.4752 ft/s

CARRYING THIS VALUE INTO (1):

$$V_A = -\frac{M_B}{M_A} V_B = -\frac{W_B}{W_A} V_B = -\frac{40/b}{60/b} (6.4752 ft/s)$$

= -4.3168 ft/s

ANSWER:



GIVEN:

9-kg BLOCK B STARTS FROM REST AND SLIDES DOWN 15-kg IVEDGE A. FIND:

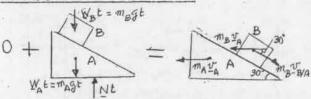
(a) VELOCITY OF B
RELATIVE TO A ATTER
IT HAS SLID 0.6 m

(b) CURRESPONDING VELOCITY OF WEDGE A. (NEGLECT FRICTION.)

WE RESOLVE YB INTO ITS COMPONED TO YAT AND YBA \$30

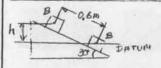


IMPULSE - MOMENTUM PRINCIFLE



 $\Sigma m v_{D} + \Sigma \Gamma t = \Sigma m v_{D}$ $+ \chi \text{ COMP.}; \qquad 0 + D = m_{B} v_{B/A} \cos 30 - m_{A} v_{A} - m_{B} v_{A}$ $1 \tau_{A} = \frac{m_{B} \cos 30}{m_{A} + m_{B}} v_{B/A} = \frac{(9 \text{ kg}) \cos 30}{15 \text{ kg} + 9 \text{ kg}} v_{B/A}$ $v_{A} = 0.32476 v_{B/A} \qquad (1)$

CONSERVATION OF ENERGY



70 = 0 V = 11 g h = (7 kg)(9.81 m/s)(0.60) cm30 = 26.487 J

V=0

REFERRING TO VELOCITY TRIANGLE SHOWN ABOVE HAD USING THE LAW OF CUSINES!

 $T = \frac{1}{2} m_A v_A^2 + \frac{1}{2} m_B \left(v_A^2 + v_{B/A}^4 - 2 v_A^2 v_{B/A} \cos 30^{\circ} \right)$

RELALLING (1) AND SUBSTITUTING THE GIVEN DAIM

T= 1/2 (15)(0,37476) UB/A +

 $+\frac{1}{2}(9)[(0.32476)^{2}+1-2(0.32476)\cos 30]U_{BM}^{2}$ $=0.79102U_{BM}^{2}+2.44336U_{BM}^{2}=3.2344U_{BM}^{-2}$

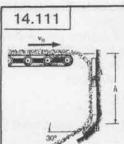
 $T+V = T_0 + V_0$: 3.2344 $V_{8/4} = 26.487 J$ $V_{8/6} = 2.8617 m/3$

NA 1 = 2,86 m/s ₹ 30°

(b) FROM EO. (1): VA = 0, 32476 (2.8617 m/s = 0.92936 m/s

v = 0,929 m/3-

(a)

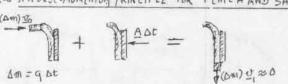


GIVEN:

MASS Q OF SAND DISCHARGED PER UNIT TIME FROM CONVEYER BELT AND DEFLECTED BY PLATE AT A SO THAT IT FALLS IN A VEKTICAL STREAM UNTIL IT 15 DEPLECTED BY PLATE AT B. FIND FORCE REQUIRED TO HOLD (a) PLATE A, (b) PLATE B. NEGLECT FRICTION BETWEEN SAND AND

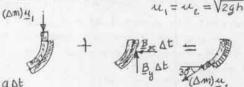
(a) IMPULSE-MUMENTUM PRINCIPLE FOR PLATE A AND SAND

PLATES.)



+2 comp.; (Dm) 10 - A Dt = 0 A = AM No = q No

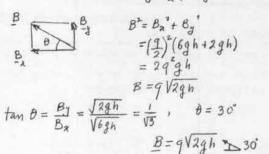
(b) IMPULSE-MOMENTUM PRINCIPLE FOR PLATE BANDSAND

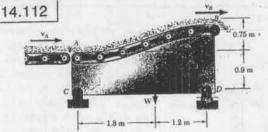


Am = q Dt

\$ x comp: 0 - B2 at = - (am) u2 cos 30" $B_{x} = \frac{\Delta m}{\Delta t} u_{z} \cos 30^{\circ} = 9\sqrt{28 h} \frac{\sqrt{3}}{2}$

+ & y comp: (Am) u, - By At = (Am) uz sin 30° By = Am (u1-u2 sin 30°) = 9 /29h (1-12) By = 29 1/29h





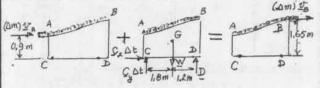
GIVEN:

SAND RECEIVED AT A AND DISCHARGED AT B AT A RATE OF 100 kg/s AND WITH UA = UB = 4.5 m/s. COMBINED WEIGHT OF COMPONENT AND SAND IT SUPPORTS IS W = 4 KN.

FIND:

REACTIONS AT C AND D.

WE APPLY THE IMPULSE-MOMENTUM PRINCIPLE TO THE COMPONENT, THE SAND IT SUPPORTS AND THE SAND IT RECEIVES IN THE INTERVAL AT.



\$2 COMP: (Am) VA + Cx At = (Am) VB $C_{z} = \frac{\Delta m}{\Delta t} (v_{B} - v_{A}) = (100 \text{ kg/s})(4.5 \text{ m/s} - 4.5 \text{ m/s}) = 0$

+ 9 MOMENTS ABOUT C:

 $-(\Delta m)V_A(0.4m)-(W\Delta t)(1.8m)+(D\Delta t)(3m)=-(\Delta m)V_B(1.65m)$ $3D = 1.8W + \frac{\Delta m}{\Delta t}(0.9V_A) - \frac{\Delta m}{\Delta t}(1.65V_B)$ = 1.8 (4000M) + 0.9 (100 kg/s) (4.5 m/s) - 1.65 (100 kg/s) (4.5 m/s) = 6862,5 N D=2,29 KN9 D = 2287.5 N

+ ty comp .: COAt - WAt + DAt = 0

 $C_y = W - D = 4000 N - 2287.5 N = 1712.5 N$

RECALLING THAT C2 = 0: C= 1.712 kN

NOTE. IF COMPONENT WAS STOPPED AND THE SAND WAS NOT MOVING, WE WOULD HAVE

C = 1,600 KN 1, D = 2,40 KN 1

14.113 100 mm 150 mm

GIVEN:

EACH OF THE POUR ROTATING ARMS OF SPRINKLER CONSISTS OF TWO STRAIGHT PORTIONS OF PIPE FORMING 120 ANGLE. EACH ARM DISCHARGES WATER AT THE RATE OF 20 L/min WITH RELATIVE VELOCITY OF 18 m/s. PRICTION IS EQUIVALENT TO COUPLE M = 0.375 Nm. FIND:

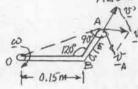
CONSTANT RATE AT WHICH SPRINKLER ROTATES.

WE APPLY THE IMPULSE - MONENTUM PRINCIPLE TO THE SPRINKLER, THE WATER IT CONTAINS, AND THE MASS AM OF WATER ENTERING IN INTERVAL AL.



EQUATING MOMENTS ABOUT AXIS OF RUTATION 4): $0 + MDE = 4 \left[MONENT OF \left(\frac{Dm}{H} \right) \mathcal{V} \right]$

MAT = MOMENTOF (Am) & (1)



THE VELOCITY &

OF THE WATER LEAVING

AN ARM IS THE

RESULTANT OF THE

VELOCITY & RELATIVE

TO THE ARM AND OF THE

VELOCITY & OF NUZZLE:

WHERE U'= 18 m/s AND VA = (OA) W BUT APPLYING THE LAW OF CUSINES TO TELANGLE OAB:

 $(0A)^2 = (0B)^2 + (BA)^2 - 2(0B)(BA) \cos 120^\circ$ = $(0.15m)^2 + (0.10m)^2 - 2(0.15m)(0.10m) \cos 120^\circ$

(OA) - 0.0475 m2

THEREFORE:

+7 MOM, OF V ABOUT D = MOM, OF V'+ MOM, DF VA

=(0.15m) & 'cos 30°- (0A)(OA) W

 $= (0.15 \, \text{m}) (18 \, \text{m/s}) \cos 30^{\circ} - (0 \, \text{A})^{2} \omega$

 $= 2.3383 \text{ m}^{4}/5 - (0.0475 \text{ m}^{2}) \omega$

SUBSTITUTING-INTO ED. (1) AND RECALLING THAT M = 0.375 N.m.

(0.375 N·m) At = (Am) [1,3383075-(0.047501)]

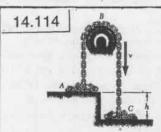
DIVIDING BY St, AND NOTING THAT

 $\frac{\Delta q n}{\Delta E} = QQ = (1 \text{ kg/L})(80 \text{ L/min}) \frac{1 \text{min}}{60 \text{ S}} = \frac{4}{3} \text{ kg/s}$

WE HAVE 0.375 N·m = (4 kg/s)[2.3383 m/s - (0.0475m2) W]

2.3383 m/s-(0.0475m²) \(= 0.20125 m/s

ω = 43.306 rad/s ω = 414 rpm

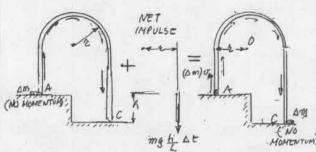


GIVEN:

WHEN GIVEN AN INITIAL
SPEED OF, THE CHAIN
KEEPS MOVING OVER
THE PULLEY.
FIND:
HEIGHT h,

(NEGLECT PRICTION.)

WE APPLY THE MAPILLE-MOMENTUM PRINCIPLE TO THE PORTION OF CHAIN OF MASS OF AND LENGTH L IN MOTION A TIME L AND TO THE ELEMENT OF LENGTH DZ AND HASS AN = M DZ WHICH WILL BE SET IN MOTION IN THE TIME INTERVAL L to



WE NOTE THAT THE ELEMENT AT A ACCURES A LINEAR MOMENTUM (AM) I WHICH IS A DOED TO THE SYSTEM, WHILE THE MOMENTUM OF THE ELEMENT AT C IS LOST TO THE SYSTEM.

EQUATING MONTENTS ABOUT O: +) O + (mg L Dt)t = (Dm) vr = (m Dz) vr

 $h = \frac{\Delta z}{\Delta t} \frac{v}{g} = v \frac{v}{g} \qquad h = \frac{\Delta z}{g} = v \frac{v}{g}$

14.115



GIVEN:

RAILROAD CAR OF MASS

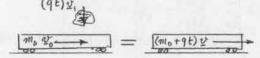
mo AND LENGTH L

APPROACHES CHUTE AT

SPEED VO TO BE WADED WITH SAND AT RATE dm/dt = q .

FIND: (a) MASS OF CAR AND LOAD AFTER CAR HAS PASSED. (b) SPEED OF CAR AT THAT TIME.

CONSEPTATION OF MOMENTURY IN HORIZONTAL DIRECTION WE CONSIDER THE CAR AND THE MASS OF SAND GE WHICH FALLS IN TO THE CHR IN THE TIME E.



 $t_{0} \approx 2 \text{ COMP.}$: $m_{0} v_{0} = (m_{0} + qt) v$ $v = \frac{4m_{0} v_{0}}{m_{0} + qt}$ (1)

LETTING V = dx IN(1):

 $da = \frac{m_0 v_0 dt}{m_0 + qt} \qquad x = m_0 v_0 \int_0^t \frac{dt}{m_0 + qt}$ (CONTINUED)

14.115 continued

$$z = \frac{m_0 v_0}{q} \left[\ln \left(m_0 + q t \right) \right]_0^c = \frac{m_0 v_0}{q} \ln \frac{m_0 + q t}{m_0} \tag{2}$$

USING THE EXPONENTIAL FORM: Mo + qt = mo e qx/movo

WHERE My + 9t REPRESENTS THE MASS AT TIME E, AFTOR THE CAR HAS MOVED THROUGH Z.

(a) MAKING X=L IN (2), WE OBTAIN THE FINAL MASS' ms = mo + qts = mo e 9L/mos

(b) MAKING E = t, IN(1), WE OBTAIN THE FINAL SPEED: v = mo vo = mo v = v e -91/movo

14.116





GIVEN:

SPACE VEHICLE DESCRIBING CIRCULAR ORBIT ABOUT THE EARTH AT SPEED OF 15,000 mi/h RELEASES AT ITS FRONT END A CAPSULE WITH A GROSS WEIGHT OF 1200 1b, INCLUDING BOO 16 OF FUEL , WHICH IS CONSUMED AT THE RATE OF 40 16/5 AND EJECTED WITH RELATIVE VELOCITY OF 9000 ft/s.

FIND :

(a) TANGENTIAL ACCELERATION OF CAPSULE AS IT IS FIRED,

(b) MAX, SPEED ATTAINED BY THE CAPSULE.

PROMER. (14.44):

ROMER. (14.44):
THRUST =
$$P = \frac{dm}{dt} = \frac{40 \text{ W/s}}{32.2 \text{ ft/s}} = (9000 \text{ ft/s})$$

$$= \frac{360 \times 10^3}{32.2}$$

(a)
$$P = m_b a_t$$
: $\frac{360 \times 10^3}{32.2} l_b = \frac{1200 l_b}{32.2 fl_b r} a_t$
 $a_t = \frac{360 \times 10^3}{1.2 \times 10^3} ft/s^2$ $a_t = 300 ft/s^2$

(b) MAX, SPEED OF CAPSULE RELATIVE TO SPACE VEHICLE I SBTAINED PROM EXPRESSION DERIVED IN PROB. 14.95. OR FROM EXPRESSION OBTHINED IN SAMPLE PROB. 14.8 BY MITTING THE TERM DUE TO FRAVITY.

$$V_{c/v} = u \ln \frac{m_0}{m_0 - qt}$$

WHERE AL = (9000 Ft/s) mo = 1200 1b, mo - qt = 1200 1b - 800 1b = 400 1b

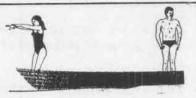
THUS:

VCN = (9000 ft/s) 2113 = (9000 ft/s)(1.0986) = 9887.5 ft/s = 6741 mi/b

Vc = Vx + Vc/v = 15,000 mi/h + 6741 mi/b = 21,741,5 mi/h

v = 21,700 mi/h

14.C1



WOMAN OF WEIGHT WAS STANDS READY TO DIVE NITH VEWCITY THE RELATIVE TO BOAT OF WEIGHT Wh . MAN OF WEIGHT WIN READY TO DIVE FROM OTHER, END OF BOAT WITH RELATIVE VELOCITY No

VELOCITY OF BOAT AFTER BOTH SNIMMERS HAVEDINED IF (a) WOMAN DIVES PIRST, (b) MAN DIVES FIRST USE W = 1201b, Wm = 1801b, W = 3001b, AND (PROB, 14.4): Now = Nom = 16 ft/s

(1) No = 14 ft/s, No = 18 ft/s .

(11) 15w=18ft/s, vm =14ft/s

ANALY515

(a) WOMEN DIVES FIRST &

Th' = VEL. OF BOAT AFTER WUMAN DIVES Nh = VEL OF BOAT AFTER BOTH SWIMMERS HAVE DIVED

CONSERVATION OF MOMENTUM! Wer (Ver - V) (W+ + Wm) V) U's = Who vin (1) 0 = - War (Var - V'b) + (Wb+Wm) VB

 $(W_b + W_m)v_b^* = W_b v_b + W_m (v_b + v_b)$

SUBSTITUTING FOR U_b^1 FROM (1): $\underbrace{+}_{b} U_b = \underbrace{W_{ue} V_w}_{W_w + W_m + W_b} - \underbrace{W_m V_m}_{W_m + W_b}$ (2)

(b) MAN DIVES FIRST:

INTERCHANGE SUB'UT AND SUB MIN (2) AND CHANGE ALL SIGNS

Vb = - Wm Vm Wm+Ww+Wb + War Vor (3) OUTLINE OF PROGRAM

INPUT WWW, Wm, Wb, VW, Vm, AND EQS. (2) AND (3).

PROGRAM OUTPUT

PROB. 14.L (a) Woman dives first Velocity of boat = -2.800 (b) Man dives first Velocity of boat = -0.229

(a) Woman dives first (L) Velocity of boat = -3.950 (b) Man dives first Velocity of boat = -1.400

(ii) (a) Woman dives first Velocity of boat = -1.650 (b) Man dives first Yelocity of boat = 0.943 14.C2 GIVEN:

COORDINATES X1, 31, 4; , WITH VELOCITIES OF COMPONENTS (V2); (Vy); (V2)1.

COMPONENTS OF ANGULAR MOMENTUM OF SYSTEM ABOUT ORIGIN O. USE PROGRAM TO SOLVE PROES, 14.9 AND 14, 13.

ANALY515

$$\underline{H}_{0} = \sum_{i=1}^{m} \underline{t}_{i} \times m_{i} \ \underline{v}_{i} = \sum_{i=1}^{m} m_{i} \left| \begin{array}{c} \underline{i} \\ \underline{z}_{i} \\ \underline{v}_{k} \end{array} \right|_{i} \ \underline{k} \\ \underline{v}_{k} \rangle_{i} \ (\underline{v}_{k})_{i} \ (\underline{v}_{k})_{i} \end{array}$$

$$H_{x} = \sum_{i=1}^{n} m_{i} \left[y_{i}(v_{x})_{i} - z_{i}(v_{y})_{i} \right]$$
 (1)

$$H_{k} = \sum_{i=1}^{m} m_{i} \left[2i \left(\nabla_{\mathbf{x}} \right)_{i} - \chi_{i} \left(\nabla_{\mathbf{y}} \right)_{i} \right] \tag{2}$$

$$H_{z} = \sum_{i=1}^{l} m_{i} \left[z_{i}(v_{y})_{i} - y_{i}(v_{z})_{i} \right]$$

$$(3)$$

OUTLINE OF PROGRAM

ENTER PROBLEM NUMBER AND SYSTEM OF UNITS USED IF SI UNITS, ENTER FOR i=1 TO i=n: $m_i(kg)$; x_i , y_i , z_i (m); $(v_x)_i$, $(v_y)_i$, $(v_y)_i$, $(v_z)_i$ (m/s)

IF U.S. CUSTOMARY: UNIS, ENTER FOR i=1 TO i=n: $W_i(lb)$; x_i , y_i , z_i (ft); $(v_x)_i$, $(v_y)_i$, $(v_y)_i$, $(v_z)_i$ (ft/s)

AND COMPUTE $m_i = W_i/32.2$

COMPUTE THE SUMS (1), (2), AND (3).

PRINT PROBLEM NUMBER

PRINT VALUES OBTAINED FOR H2, Hy, H2.

IF SI UNITS, RESULTS ARE EXPRESSED IN kg/m/s.

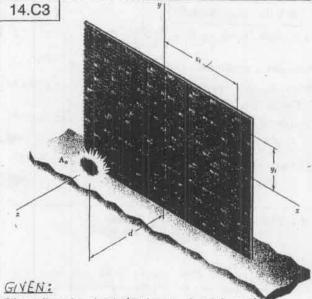
IF U.S. CUSTORIARY UNITS, RESULTS ARE

EXPRESSED IN ft. 16.5.

PROGRAM OUTFUT

Problem 14.09 Hx = -31.2 kg*m^2/s Hy = -64.8 kg*m^2/s Hz = 48.0 kg*m^2/s

Problem 14.13 Hx = 0.000 ft*lb*s Hy = -0.720 ft*lb*s Hz = 1.440 ft*lb*s



SHELL MOVING WITH VELOCITY OF CHAPMENTS

No. No. No explodes in three Framents of WEIGHTS

WI, W. W. AT POINT AD AT DISTANCE of FROM WHILL.

FRAGMENTS HIT THE WALL AT POINTS A: (2=1,2,3)

OF COOKDINATES X: AND Y:

FIND: SPEED OF EACH FRAGMENT AFTEL EXPLOSION
USE PROGRAM TO SOLVE (a) PROB. 14.25 (b) PROB. 14.26 .

ANALYS 15

DETERMINE DIRECTION COCINES OF PATH A A: (i=1,23)
FIRST CONPOTE & = 1/2; + y; + d2 (i)

THEN $(\lambda_z)_i = x_i/\ell_i$, $(\lambda_y)_i = y_i/\ell_i$, $(\lambda_z)_i = -d/\ell_i$ (2)

CONSERVATION OF LINEAR MOMENTUM:

 $\frac{1}{9}(W_1 + W_2 + W_3)(V_2 + V_3 + V_4 + V_4) = \frac{W_1}{9}V_1 + \frac{W_2}{9}V_2 + \frac{W_3}{9}V_3 +$

2. COMP: $W_1(\lambda_2)_1 U_1 + W_2(\lambda_2)_2 U_2 + W_3(\lambda_3)_3 U_3 = (W_1 + W_2 + W_3) U_3$ (3) 4. COMP: $W_1(\lambda_3)_1 U_1 + W_2(\lambda_3)_2 U_2 + W_3(\lambda_3)_3 U_3 = (W_1 + W_2 + W_3) U_3$ (4) 2. COMP: $W_1(\lambda_3)_1 U_1 + W_2(\lambda_3)_2 U_2 + W_3(\lambda_3)_3 U_3 = (W_1 + W_2 + W_3) U_3$ (5) THESE 3 EQS. ARE SOLVED SIMULTANEOUSLY FOR V_1, V_2, V_3

CONTLINE OF PROGRAM
ENTER PROBLEM NUMBER
ENTER VALUES OF V2, V2, AND d
ENTER VALUES OF W1, 21, 41 FOR 1=1,2,3
COMPUTE DIRECTION COSINES FROM EQS. (1) AND (2)
COMPUTE COEFF. IN EQS. (3), (4), (5) AND SOLVE FOR V, V, V3
BY COMPUTING

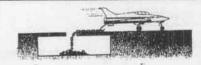
 $\mathcal{D} = \begin{bmatrix} w_1^{\prime}(?_{\lambda_1})_1 & w_2^{\prime}(?_{\lambda_2})_2 & w_3^{\prime}(?_{\lambda_2})_3 \\ w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_2 & w_3^{\prime}(?_{\lambda_3})_3 \\ w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_2 & w_3^{\prime}(?_{\lambda_3})_3 \end{bmatrix}, \quad \mathcal{D}_1 = \begin{bmatrix} (\sum u_1^{\prime})_{x_1^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_1 \\ (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_3 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_3^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_3 \end{bmatrix}, \quad \mathcal{D}_1 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_1 \\ (\sum u_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_3 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_3 \end{bmatrix}, \quad \mathcal{D}_1 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_3 \end{bmatrix}, \quad \mathcal{D}_2 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_3 \end{bmatrix}, \quad \mathcal{D}_2 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_2 \end{bmatrix}, \quad \mathcal{D}_3 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_2 \end{bmatrix}, \quad \mathcal{D}_3 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_2 \end{bmatrix}, \quad \mathcal{D}_3 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_2 \end{bmatrix}, \quad \mathcal{D}_4 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_2 \end{bmatrix}, \quad \mathcal{D}_4 = \begin{bmatrix} (\sum u_1^{\prime})_{x_2^{\prime}} & w_1^{\prime}(?_{\lambda_3})_1 & w_2^{\prime}(?_{\lambda_3})_1 \\ (\sum w_1^{\prime})_{x_2^{\prime}} & w_2^{\prime}(?_{\lambda_3})_1 & w_3^{\prime}(?_{\lambda_3})_2 \end{bmatrix}$

AND $v_1 = D_1/D$, $v_2 = D_2/D$, $v_3 = D_3/D$ PROGRAM OUTPUT

> (O) Problem 14.25 VA = 1678 ft/s VB = 1390 ft/s VC = 1230 ft/s

(b) Problem 14.26 VA = 2097 ft/s VB = 1853 ft/s VC = 738 ft/s

14.C4

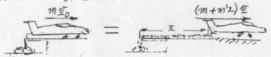


GIVEN: AS GOOD-KY PLANE LANDS ON CARKER AT 180 Km/h, ITS TAIL HUOKS INTO END OF 80-m LONG CHAIN OF MALS PER UNIT LENGTH M' = 50 to/m LYING BELOW DECK.

TAP ... HTE USING 5-M INCREMENTS, THE DISTANCE TRAVELED BY THE PLANE AND THE CORRES PONDING VALUES OF THE TIME, THE VELDELTY, AND THE ACCELERATION OF THE PLANE (ALC HIS AM) OTHER KETALD IS FURLE.)

D+=

CONCERNATION OF LINE BY MORE PROTECTION !



$$\pm m \sqrt{n} = (m + m'z)v \tag{1}$$

LETTING v = dz/dt:

$$m V_0 dt = (m + m'z) dz$$

$$t = \int_0^z \frac{m + m'z}{m V_0} dz = \left[\frac{(m + m'z)^2}{2mm'V_0} \right]_0^z = \frac{(m + m'z)^2 - m'z}{2mm'V_0}$$
(2)

SOLVING (1) FOR
$$V: U = \frac{\pi U V_0}{m + m^2 Z}$$

DIFFERENTIATING (1) WITH RESPECT TO t: 0 = m'dz v + (m+m'z) dv

NOTING THAT da/At =
$$v$$
 AND $dr/At = a$?
$$0 = m^3 v^2 + (m + m^2 z)a \qquad \alpha = -\frac{m^3 n^2}{m + m^2 z} \qquad (4)$$

OUTLINE OF PROGRAM

ENTER 11 = 6000 kg, 111' = 50 kg/m, Up = 180 km/h = 50 m/s FOR X = 0 TO X = 80 A AND USING 5-M INCKEMENTS CALCULATE t. V, AND a FROM tas. (2), (3). (4) AND TABULATE

PROGRAM OUTPUT

Distance	Time	Velocity	Acceleration
(m)	(s)	(km/h)	(m/s^2)
0.000	0.000	180.000	-20.833
5.000	0.102	172.800	-18.432
10.000	0.208	166.154	-16.386
15.000	0.319	160,000	-14.632
20,000	0.433	154.286	-13.120
25.000	0.552	148.966	-11.809
30.000	0.675	144.000	-10.667
35.000		139.355	
40.000	0.933	135.000	-8.789
45.000	1.069	130,909	-8.014
50.000	1.208	127.059	-7.327
55.000	1.352	123.429	-6.717
60.000	1.500	120.000	-6.173
65.000	1.652	116.757	-5.686
70.000		113.684	-5.249
75.000		110.769	-4.855
80,000	2.133	108.000	-4.500

14.C5



GIVEN:

A 16-Mg PLANE MAINTAINS A CONSTANT SPEED OF 774 km/h WHILE CLIMBING AT AN ANGLE & = 18. PLANE SCOOPS IN AIR AT RATE OF 300 kg/s AND DISCHARGES IT AT A RELATIVE SPEED OF 665 m/s. PILOT THEN CHANGES ANGLE OF CLIMB & WHILE MAINTAINING THE SAME ENGINE SETTING. FIND FOR VALUES OF & FROM O TO 20 USING 1" INCREMENTS!

(a) INITIAL ACCELERATION OF PLANE, (b) MAXIMUM SPEED THAT WILL BE ATTAINED. (ASSUME DRAG TO BE PROPURTIONAL TO U".)

ANALYSIS

FROM EQ. (14.39): THRUST = P = dm (u-v) DENOTING RATE dm/at by R:

P= R(u-v) (1) WHILE CLIMBING AT US AND ON: (2) Po = R(u-vo) SINCE PLANE IS IN EQUILIBRIUM: D = P - mg sin &

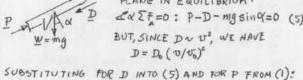


(a) PLANE CLIMBING AT ANGLE & AND SPEED V:

Po-Do-mysina=ma

(b) MAX, SPEED WHILE CLIMBING AT ANGLEK:

ACCEL, WILL THEN BE ZERU AND PLANE IN EQUILIBRIUM . BUT, SINCE DN UZ, WE HAVE



R(u-v)-(Do/v2) v2-mg sin x = 0 v + Rv v + (v, /D,) (mg sing - Ru) =0 C = (vo/Do)(mg sing-Ru) B = R 15/Do.

MAX. SPEED: $v = \frac{1}{2} \left(-B + \sqrt{B^2 + C} \right)$

OUTLINE OF PROGRAM

ENTER m=16×10 kg, 5=774ka/h = 215 m/s, 0 = 18° R = 300 kg/s, u = 665 m/s, g = 9.81 m/st. USE EQS. (2) AND (3) TO CALCULATE PO AND DO. CALCULATE B = R Vo / Do.

FOR & FROM O TO 20, WITH 1° INCREMENTS, (a) USE EQ.(4) TO CALCULATE a

(b) USE EQS. (6) AND (7) TO CALCULATE Umay

(CONTINUED)

14.C5 continued

PROGRAM O'ITPUT

celeration	on max v
m/s^2	km/h
3.031	921.796
2.860	913.933
2.689	906.020
2.518	898.060
2.347	890.053
	873.907
1.666	857.594
1.497	849.378
	841.126
1.160	832.839
	824.518
	816.166
210173022344	807.785
	799.375
	790.940
	782.481
	774.000
	765.499
	756.981
	3.031 2.860 2.689 2.518 2.347 2.176 2.006 1.836

14.C6 GIVEN;

ROCKET OF WEIGHT 2400 16, INCLUDING 2000 16 OF FUEL, IS FIRED VERTICALLY FROM GROWN IT CONSUMES PUEL AT RATE OF 25 16/5 AND EJECTS IT WITH RELATIVE VELOCITY OF 12,000 ft/s.

FIND FROM TIME OF IGNITION TO TIME WHEN LAST PARTICLE OF FUEL IS CONSUMED, AND AT 4-5 TIME INTERVALS:

- (A) ACCELERATION IL OF ROCKET IN ft/52,
- (6) ITS VELOCITY WIN ft/s,
- (c) ITS ELEVATION H ABOVE GROUND IN MILES.

ANALYSIS

WE RECALL TROM SANFLE PROB. 14.8 THAT

$$v = u \ln \frac{m_0}{m_0 - qt} - gt \tag{1}$$

WHELE IT - VELOCITY OF ROCKET

MO = INITIAL WEIGHT OF ROCKET AND FUEL

Q = RATE AT WHICH FUEL IS CONSUMED

W = RELATIVE VELOCITY AT WHICH FUEL IS EJECTED

LEITING t = dy/dt AND INTEGINING y FROM 0 TO h: $h = \int_{0}^{h} dy = u \int_{0}^{t} ln \frac{m_{o}}{m_{o} - qt} dt - \frac{1}{2}gt^{*}$ (2)

TO CALCULATE THE INTEGRAL, WE SET MO-96 = 2

AND SETAIN $dt = -\frac{m_0}{q} d\pm$, THEREFORE: $\int_{0}^{t} \frac{m_0}{m_0 - qt} dt = \int_{1}^{t} -\ln z \left(-\frac{m_0}{q} dz \right).$ $= \frac{m_0}{q} \int_{1}^{t} \ln z dz - \frac{m_0}{q} \int_{1}^{t} \ln z dz - \frac{m_0}{q} dz = \frac{m_0}{q} \int_{1}^{t} \ln z dz - \frac{$

= mo [thr zdz - mo [zlnz-z] = mo (zlnz-z+1)

THUS, EQ. (2) YIELDS

 $h = \frac{m_0 u}{q} \left(\frac{m_0 - qt}{m_0} \ln \frac{m_0 - qt}{m_0} - 1 + \frac{qt}{m_0} + 1 \right) - \frac{1}{2} gt^2$ $h = u \left(t + \frac{m_0 - qt}{q} \ln \frac{m_0 - qt}{m_0} \right) - \frac{1}{2} gt^2 \qquad (3)$

REWRITING EQ. (1) AS V=u ln 1110 - u ln (mo-9t) - gt

AND DIFFERENTIAT NE WITH RESPECT TO t.

$$a = \frac{dv}{dt} = -u = \frac{-q}{m_0 - qt} - g$$
 $a = \frac{uq}{m_0 - qt} - g$ (4)

(CONTINUED)

14.C6 continued

OUTLINE OF PROGRAM
ENTER q = 31.2 ft/s, mo = 2400/g, mg = 2000/g.

q = 25/g, u = 12,000 ft/s

COMPUTE FINAL TIME = tg = mg/q = 2000/25 = 80 s

POR t PROM O TO 80 s AT 4-5 INTERVALS,

COMPUTE

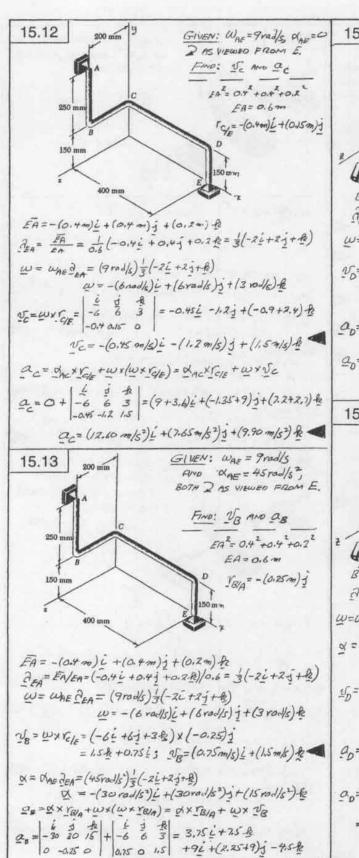
- (a) ACCELERATION a FROM EQ. (4)
- (b) VELOCITY & FROM ER. (1)
- (c) ELEVATION A FROM EQ. (3), DIVIDING RESULT BY 5280 TO BETAIN A IN MILES.

PROGRAM OUTPUT

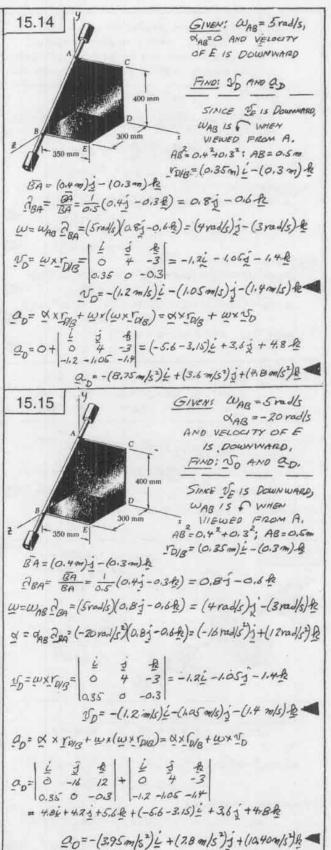
t	a	v	h
8	ft/s^2	10~3 ft/s	mi
0.000	92.800	0.000	0.000
4.000	98.235	0.382	0.143
8.000	104.164	0.787	0.584
12,000	110.657	1.216	1.341
16.000	117.800	1.673	2.434
20.000	125.695	2.159	3.883
24.000	134.467	2.679	5.714
28.000	144.271	3.236	7.952
32,000	155.300	3.835	10.628
36.000	167.800	4.481	13.775
40.000	182.086		17.431
44.000	198.569	5.940	21.639
48.000	217.800		26.449
52.000	240.527	7.688	31.921
56.000	267,800	8.702	38.122
60.000	301.133	9.838	45.137
64.000	342.800		53.066
68.000	396.371	12.596	62.037
72.000	467,800		72.213
76.000	567.800		83.814
80.000	717.800		97.148

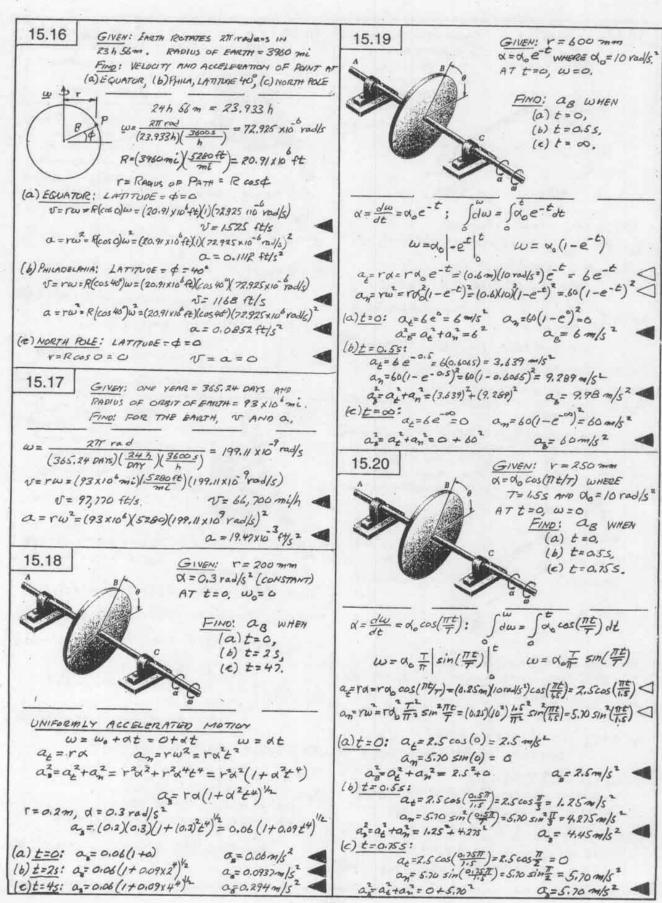
GIVEN: 0=1.5t3-4.5t2+10 GIVEN: 0= 0, SIN(T) - 0,50, SIN(27) 15.1 15.4 WHERE Go= 6 rad, T=45. FIND: Q, W, AND X WHEN (a) t=0, (b) t=45. Fino: Θ , ω , area when (a) t = 0, (b) t = 2s. w= de = 0, T cos(T) -0.50 2T cos(27) w= d9/dt = 4.5t2-9t d = dw/dt = 9t - 9 $Q = \frac{d\omega}{dt} = -\Theta_0\left(\frac{\pi}{T}\right) \sin\left(\frac{\pi t}{T}\right) + 0.5\Theta_0\left(\frac{2\pi}{T}\right) \sin\left(\frac{2\pi t}{T}\right)$ @= 10 rad (a) t=0: w=0 $\omega = 6 \frac{\pi}{4} - 0.5(6) \frac{2\pi}{4}$ $\alpha = 0$ a = - 9 rad/52 (a) t=0: 0=0 (b) t=45: B=1.5(4)3-4.5(4)+10 6=34 nad W= 4.5(4)2-9(4) 0=65in(21)-0.5(6)sin(41)=6-0, 0=6rd W= 36 nad/5 x = 27 rad/52 a = 9(4)-9; W= 6(#) cas(2/1) - as(6) 2/1 cas(4/1) GIVEN: 0=1.5t3-4.5t2+10 15.2 = 6 7 (0) -0.5(6) 27 (-1) = 67 FIND: t, O, AND O WHEN W=0 w=de/dt = 45t 2 -9t a = dw/dt = 9t -9 $=-6(\frac{\pi}{4})^2(1)+3(\frac{2\pi}{4})^2(0)=-\frac{3}{6}\pi^2$ FOR W=0: 4.5t2-9t=0 a = -3.70 nad/s t = O AND t=Z. GIVEN: 0=0, sin (Tt)-0.50, sin (ZTt) 15.5 t=0: 0 = 10 rad, x = -9 rad/s2 WHERE O = 6 red, T=45 t= 25: 0=1,5(2)-4,5(2)+10, FIND: 0, W, AND X WHEN t= 15 d= grad/s2 X = 9(2)-9, w=de=0, Fcos(平)-0.50, 27 cos(27t) GIVEN: 0=00(1-e-14) WITH 6=0.40 rad 15.3 x = div = -0 (\$\frac{T}{2}\sin(\frac{T}{T})\ +0.50 (\frac{277}{T}\sin(\frac{277}{T}\) FIND: 6, W, AND OX INHEN (a) t=0, (b) t=35, (c) t=00 t=15: ==6sin(#)-0.5(6) sin(2#) 0=0.40 (1-e-+4) =6 \frac{\sqrt{2}}{2} -0.5(b)(1), \Quad \Quad =1.243 nod $\omega = \frac{d\Theta}{dt} = \frac{1}{4}(0.40)e^{-\frac{t}{4}} = 0.10 e^{-\frac{t}{4}}$ (w= 6(7) cos(7) - 0.5(6)(21) cas(27) a = dw = - 4/0.10)e-t/4 =-0.025e-t/4 = $6\left(\frac{\pi}{4}\right)\frac{\sqrt{2}}{2} - 0.5(6)\left(\frac{\pi}{2}\right)(0)$, co = 3.33 now/s Q=-6(#) sn(#)+0.5(6)(27) sin(27) (a) t=0: 0=0.40(1-e) 0=0 = $-6\left(\frac{77}{4}\right)^2\frac{\sqrt{2}}{2} + 0.5(6)\left(\frac{277}{4}\right)^2(1)$, x = 4.79 rad/s² w= 0,10 e0 W=0.1 ned/s d = -0.025e0 d= - 0.025 nad/s -15.6 GIVEN: t=0, W=0 0=0.40(1-e t=65, w = 3300 gm = 11017 nad (b) t=35: =0.40(1-0.4724), B=0.211 nal THEN COASTS TO IZEST IN 805. W=0.10 e-3/4 FIND: NUMBER OF REVOLUTIONS W=0.0472 nod/s = 0.10(0.4724) $0.4 = -0.025 e^{-3/4}$ (a) TO REACH SPEED OF 3800 April. (b) To const TO REST. = -0.025 (0.4724) d=-0.01/81 nod/52 UNIFORMLY ACCELERATED MOTTON: WO = 0, £ = 65. (e) t= D: 0=0.40(1-e-00) (a) w= w0+at; 1107 = 0+a(6), x= 1107 rad/s2 = 0,40 (1-0) 0=0.4 rod 0= Wot+ 2 at = 0+ 1 (10 T) (65) = 330 TT rad W=0,10e-00 0 = (33011 nod) 1 rev 0 = 165 nev W=0 d=-0.025e-0 (b) W,= 110 Trad/s, W= 0 WHEN E=805 d = 0 w2= w,+ xt; 0=1101+ d(80s), x=-110 1 rod/s2 0= w,t+ 1 dt= (11077)(80) - 1 (110 11)(80s)2 = 8800 TT - 4400TT = 4400TT nad 0 = (4400 TT) 1 rev 0 = 2200 rev €

15.7 GIVEN: ROTOR CONSTS TO REST IN 4 mm. 15.10 GIVEN: WAB - 7.5 nods FROM RATED SPEED OF WE = 6900 ym. DAS VIEWED FROM B. FOR UNIFORMLY ACCELERATED MOTION, dag=0. FIND: (a) AHG. ACCEL, &. (b) NUMBER OF REVOLUTIONS FIND: Wo= 6900 apm (27) = 722.57 nad/s, t=4mm=2405 NJ AND a (a) W= Wo+ at; 0=722.57+a(240) a=-3,0107 nod/s = -3.01 nod/s2 AB = 20+9+122 AB= 25 in. (b) == wot + = at = (722.57)(240) + = (-3.0107)(240) 0=173,416-86,708 = 86,708 nad 0=86,708 and (100), 0=13,800 nay € Am= AB = 1 (201-95+12 R) 15.8 GIVEN: X=-RO. W= WAB = (7.5 mod/s) 1/25 (20i -9j+12A) FIND: (a) VALUE OF & FOR WHICH W= 8 rad/s INHER Q=O AND Q= 4 mad/s WHEN CU=0. W= (bnad/s)i- (2.7 nad/s)j+ (3.6 nad/s)-A (6) ANGULAR VELOCITY INHEN @= 3 rad. -FE/B = - (12 im.) & $\omega \frac{d\omega}{d\theta} = -R\theta$ x=-10 V= WXTEB= (61-2.7)+3,64) x(-124) [wdw = -] + Odo ; | 102 = - 1/2 &62 = 72 + 32.4 L -1=(32.4 m/s) i+(72 m/s) j $\frac{1}{2}(0-8^2) = -\frac{1}{2}k(4^2-0)$ $k = 45^{-2}$ af = xxE/B+wx(wxE/B)=xxE/B+wxVE $\left| \frac{1}{2} \omega^2 \right| = - \left| \frac{1}{2} (4s^{-1}) \Theta^2 \right|$ a=0+6-2.7 3.6 =-259.26+116.6j+(432+87.4)-k [wdw=-]Redo; Brodle a=-(259,2 m/s2) 1+(116.6 m/s2) j+(519 m/s2) & 1(w2-82)=- 1(4)(32-0) ω2-64=-36; ω=64-36=28; ω=5.29 nd 15.11 GIVEN: WAB = 7.5 red/s GIVEN: X = -0.25 W; WHEN t=0 W=20 mays 15.9 DAS VIEWED FROM B. FIND: (a) REVOLUTIONS BEFORE W= 0. D/AB = - 30 nonly 2 (W) TIME WHEN W=0. (R) TIME WHEN WE O.O. WO. FINO: V. AND a d=-0.25w; w dw =-0.25w; dw=-0.25dB AB = 20+92+122 [dw = -0.25 | d6 ; (0-20) = -0.250 AB = 25 in. 0= 80 mad B=(Bonad) rev @= 12.73 rey 14B - (20 m.) L 7A8 AB = 15(201-95+12-B) 1 d=-0.25w; dw =- 0.25w; W = WAS TAB = (7.5 rad/s) 1/25 (201-99+12A) $\int_{\omega} \frac{d\omega}{\omega} = -0.25 \int_{\omega}^{t} dt$ w= (6 nads)i-(22 nad/s) + (3.6 rad/s) & enw =-0,25t M = dAB PAB = (-30rad/s2) = (20i-9j +12A) t=-1/25 (lnw-ln 20) = 4(ln 20-lnw) d=-(24rad/s) + (10.8 rad/s2) j-(4.4 rad/s2) & Vc = ax rela = (6i - 2.75+3.6A) x(-20i) (1) t=4 ln 20 = -54-R-725 NE=-(72m/s)j-(54m./s)k t=4 ln 30=4 ln 00 FOR W=0 ac= xxelg+wx(wxxelg) = xxxelg+wxxe (c) FOR W = 0.01 W0 = 0.01(20) = 0.2 rad ac= -24 10.8 -144 + 6 -2.7 3.6 USE EQ(1): t=4ln(20)=4ln 100=4(4.605) -20 0 0 a= 2885 +216 \$ + (157.2+259.2) = +324 5 -432 & t= 18,42 S ac= (405 m, 152) i + (612 m, 152) j-(216 m, 152) &



a = (12.75 m/s2) i + (11.25 m/s2) j + (3 m/s2) &





15.21 (b) an=37.51n/s2-15.22



FIND: (a) W AND of OF PULLET. (b) aB

a=9 in.152

v=rw 1511/5 = (6in) w ; W= w = 25 nad/s)

a=rd; 911.15 = (6in) X d=1.5 nad/52]

= a= 91m/52 an=rw=(6in. X2. Snad/s)2 an= 37.5 m/s2 + Cy=9in/s2

a= 38.6 in/s2 5765 la,



GIVEN: W= 4 rad/s 2 FIND: & FOR WHICH a= 120 in./52

= ax=ra = (6in) X an= rw= (bin / 4rad/s) = 96in./52 6im. a = a + a = (120 in/s") = (60) 2+ (96 in./s") x = 12 mad/s 2) Ø = ±12,

15.23

GIVEN: d = 120 rays2) WHEN t=0, W=0

FIND; ac WHEN (a) t=0.55, (b) t = 25.

d=120 rod/52 at=rd=(0.025m)(120rad/s) a= 3 m/s2+

(a) t=0.55: w= xt = (120 rad/s2)(0.55) = 60 rad/s a = rw = (0,025 m)(60 rad/s)2 an= 90 m/s2-

a==a=+a==3+902 aB=90,05m/s2

(b) t = 25: W = at = (120 rad/52)(25) = 240 rad/5 an= rw= (0.025m)(240 rad/s)2 an= 1440 m/s2 ag=a2ta,=3+14402 ag=1440 m/s2

15.24

GIVEN: RATED SPERD OF DRUMS 15 2400 rpm SANDER CONSTS TO REST IN 105. FIND: VE AND OC (a) BEFORE POWER

IS CUT OFF, (b) 95 LATER.

r=01025m wo = 2400 rpm = 251,3 nalls

(a) N= rw= (0,025m)(251,3 rods); N=6.28 m/s a = rw=(0.025 m)(251.3 nod/s)25 a=1579 m/52

(b) WHEN ==105, W=0. w= wo + at i 0=251,3 rads + a(106); x=-25.13 rads

INHEN £ = 95: w = Wo + at; w = 251,3 rad/s - (25,13 rad/s)(95)= 25,13 10/s N= ru= (0.025m) 25.13 rad/s); nf =0.628 m/s (ac) = rd=(0.025m) -25.13 ral/s); (ac) = 0.628m/s? (ac) = rug= (0.025m)(25.13 rad/s)2; (ac) = 15.79 m/s

a= 15.80 m/s2 a= (a)+(a) = (0.628 m/s)+(15.79 m/s)

15.25

GIVEN: WB = (30 ralls) j

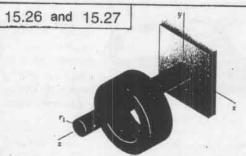
IF NO SLIPPING OCCURS FIND: (a) WA (b) ACCELERATIONS OF POINTS IN CONTACT.

WB = 30 rodf (a) VELOCITIES! FOR NO SLIPPING. rawa = raws (6in.) WA = (2.4/n.) 30 rad/s) WA = 12 rad/s 8=2.4in. WA = - (12 rad/s) } ra = 6 in, -

(b) ACCELEMATIONS ! Lug=30 rad/s wa= 12 rods an ra= 6 in. -

an = 1/4 ma = (6in x12 nod/s) = 864 in./s2 = 72 ft/s2 a=-(72 ft/s) i

ag=rgwg=(2.4in × 30 ralls)= 2160 in 152= 180 f4/52 aB= (180 ft/52) i



FIND: (a) ANG. VELOCITY OF SHAFT: WA WA

(b) ACCHERATIONS OF POINTS SHAFT AND RING WHICH ARE IN CONTACT.

PROB. 15.26: IN TERMS OF WA, T, T2, AND Y3.
PROB. 15.27: WHEN WA = 25 rodb, Y, = 12 mm,
T2= 30 mm, AIID Y3 = 40 mm

ALSO, FIND ACCEL. OF POINT ON OUTSIDE OF B.



(a) AT POINT OF CONTACT $\Gamma_1 W_0 = V_2 W_0 \qquad W_0 = V_1$

(b) ACCEL OF POINTS OF CONTACT

SHAFT A: Q = 1, WA

RING B: $\alpha_{B} = r_{2} \omega_{B}^{2} = r_{2} \left(\frac{r_{1}}{r_{2}} \omega_{p} \right)^{2}$

an = 1/2 wa

ACCEL OF POINT D ON OUTSIDE OF PLINE

 $a_D = r_3 \omega_B^2 = r_3 \left(\frac{r_1}{r_2} \omega_A \right)^2; \quad a_D = r_3 \left(\frac{r_1}{r_2} \right)^2 \omega_A^2$

PROB. 15.2) WA= 25 rad/s, V,=12 mm Va= 30 mm, V3=40 mm

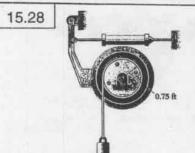
(a) $W_B = \frac{Y_1}{Y_2} W_A = \frac{12 \, \text{mm}}{30 \, \text{mm}} (25 \, \text{rad/s}); \ W_B = 10 \, \text{rad/s})$

(b) $a_A = v_i w_A^2 = (12 mm)(25 rad/s)^2 = 7.5 × 10^3 mm/s^2$ $a_A = 7.5 m/s^2 \downarrow$

 $a_{g} = \frac{r_{1}^{2}}{r_{2}} w_{A}^{2} = \frac{(12 \text{ m/m})^{2}}{(30 \text{ m/m})^{2}} (25 \text{ rad/s})^{2} = 3 \times 10^{3} \text{ m/m/s}^{2}$ $a_{R} = 3 \text{ m/s}^{2}$

 $a_D = r_3 \left(\frac{r_1}{r_2}\right)^2 u_A^2 = (40 \text{ mm}) \left(\frac{12 \text{ mm}}{30 \text{ mm}}\right)^2 (25 \text{ rad/s})^L$ $a_D = 4 \times 10^3 \text{ mm/s}^2$

a0=4m/s2



GIVEN:

WHEN \$ = 9 \(\tilde{A} = 9 \\ \tilde{B} \)

BRAKE IS APPLIES

AND BLOCK COMES

TO REST AFTER

MOVING 18 FET,

ASSUMING UNIFORM

MOTION, FIND:

(a) \$\tilde{A}\$ OF DRUM

(b) TIME TO

COME TO REST

BLOCK A:

 $15^2 - 15^2 = 2a.5$ $0 - (9ft/s)^2 = 2a(18ft)$

a = - 2.25 PWs = a = 2.25 fys = 1

DRUM:



 $V_A = r\omega_0$ $9 \text{ ft/s} = (0.75 \text{ ft}) \omega$ $\omega_0 = 12 \text{ rad/s}$

a = rx -(2.25ft/s²) = (0.75ft) x

X=-3rad/s2; X=3rad/s2)

UNIFORM MOTIONI W=0 WHEN t=t, $w=\omega_0+\alpha t$: $0=(12 \text{ rad/s})-(3 \text{ rad/s}^2)t$, t=45

15.29

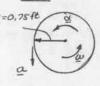


GIVEN:
WHEN t=0, V=0.
WHEN t=5, BLOCK
HAS MOVED 16ft ASSUMING UNIFORM
MOTION, FIND:
(a) & OF DRUM
(b) W OF DRUM
WHEN t=45

BLOCK A: $S = \sqrt[3]{6}t + \frac{1}{2}at^2$ $16ft = 0 + \frac{1}{2}a(5s)^2$

a=+1.28 ft/s2 a=1.28 ft/s2

DRUM:



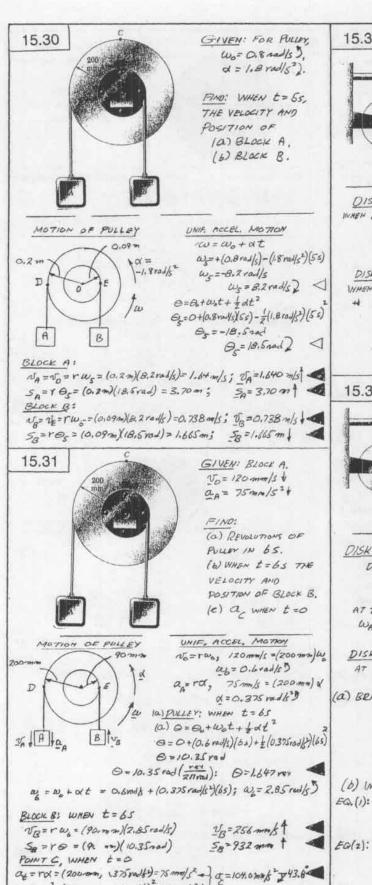
a = r d $(1.28 + t/s^2) = (0.75 + t) d$ $a = 1.707 rad/6^2$ $d = 1.707 rad/6^2$

UNIFORM MOTION W= 0 WHEN t=0

W=Wotat

WHEN t = 45: $\omega = 0 + (1.707 \text{ rad/s}^2)(45)$ $\omega = 6.83 \text{ rad/s}$

W=6,83 rad/s)



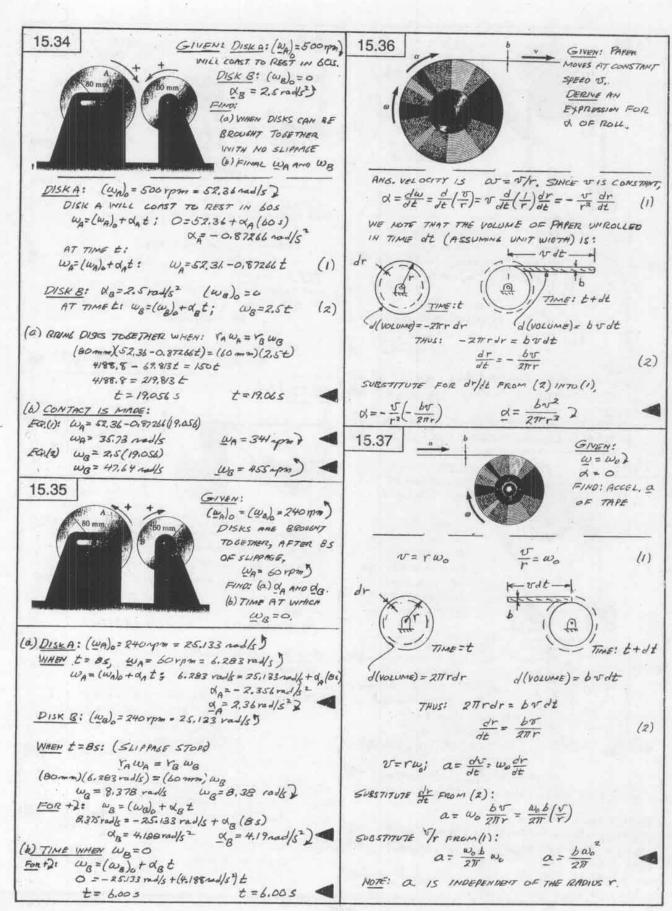
02= YW= (200 mmy 3.6 rad/s) = 72 mm/s=4)

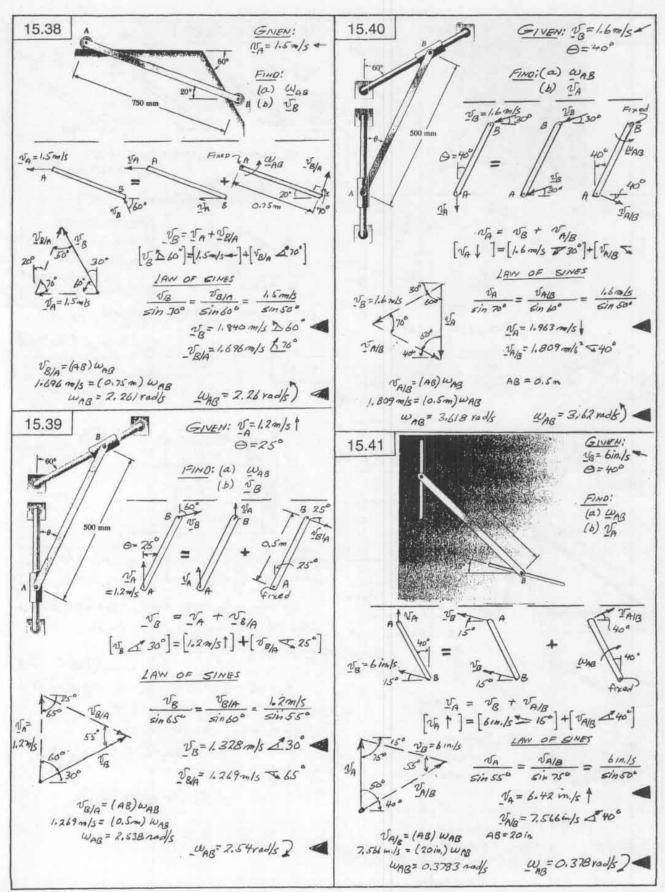
15.32 GIVEH: WHEN \$ =0, (Wa) = 450 rpm 2 (WE) = D AFTER SLIPPAGE, WHEN t=65 WA= 140 ypm 2 FIND: DURING SLIPPROS DIA AND ONB DISKA: (W) = 450 rpm = 47.124 ned/s 7 INNEN 1=6: WA = 140 rpm = 14.661 nads) WA = (WA) + XA I 14.661 nod/5 = 47.124 rod/s + 0/4 (65) 0 = 5.41 madk) 01 == -5.41 rad/s DISK B: Wo = 0 WHEN #=65: (END OF SLIPPACE) + rawa = rawa! (3in) (14.661 rad/s) = (5in) (WB) WB= 8.796 nad/s) wg=(wg)+agt 8.796 nad/s = 0+00 (65) OR= 1.466 rad/s) Olg=1,466 mad/52 15.33 GIVEN: DISKA: (WA)=500 Tpm] WILL COAST TO REST IN 60s DISKB: (Wa) =0 XB= 2.5 rad/52) FIND: (a) WHEN DISKS CAN BE BROUGHT TOGETHER WITH NO SLIMME (b) FIMAL WA AND WR. DISK A: (WA) = 500 rpm = 52,36 rad/s 2 DISK A WILL COAST TO REST IN GOS WA = (WA) + OA t; 0=52.36 mod/s + dA (600) d= -0.87266 rad/s AT TIME t: Wa=52,36 - 0,87266 t (1) WA = (WA) + dA 2 3 DISK B: dg= 2.5 nod/s" (w3)0=0 (2) AT TIME t: Wa= (Wa) + dati WB= 2.52 (a) BRING DISKS TOGETHER WHEN: YAWA = YBWB (3 in. 152.36-0,87261t)=(510.1)(2.5t) 157.08 - 2.618t = 12.5t 157,08 = 15.118t t=10.395 (b) WHEN CONTACT IS MADE (t = 10.393) EQ.(1): WA= 52.36 -0.87266(10.39) wa = 4/3 rpm 2 Wa = 43,29 mad/5

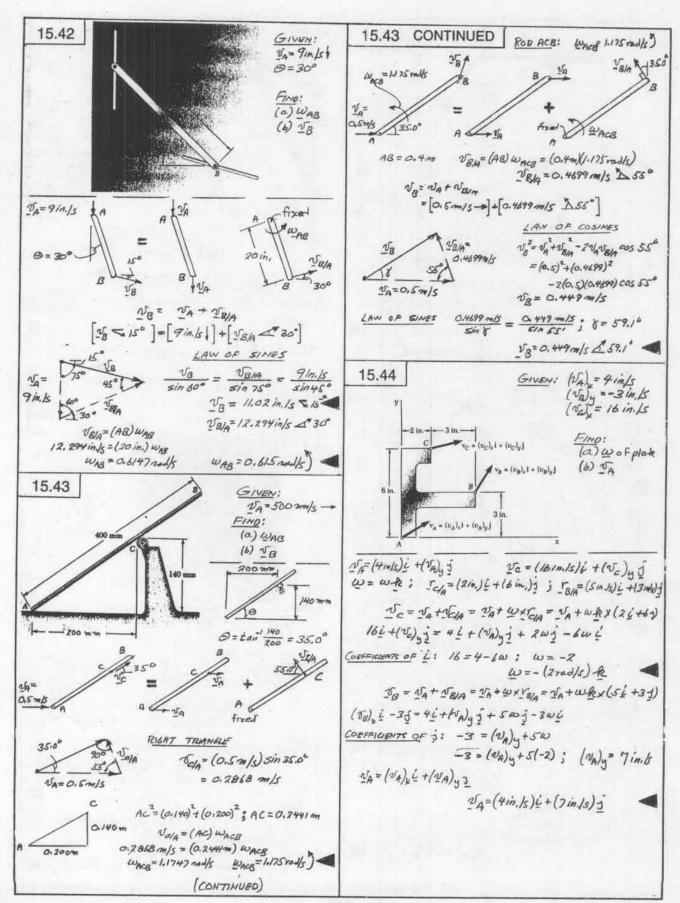
WB=2.5(10.39)

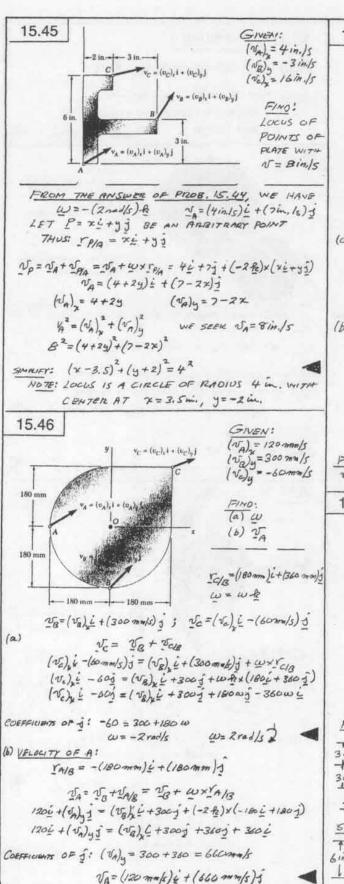
WR= 25,975 rad/s

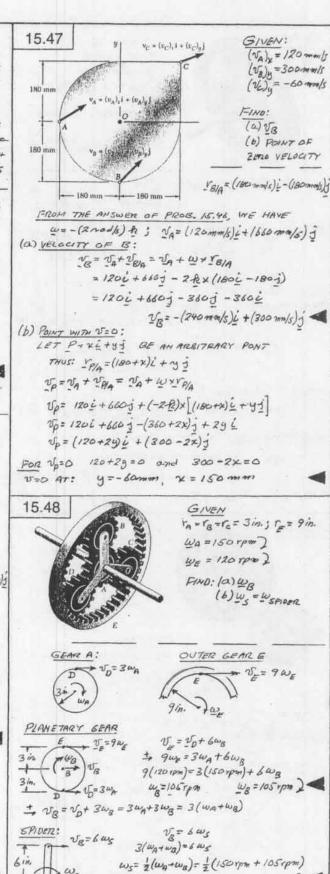
WB=24Bym)





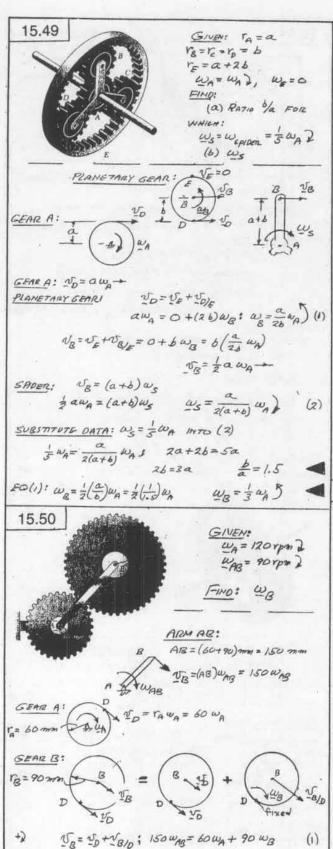






Ws=127,5 pm

w=127.5 rpm)



15.50 CONTINUED

EQ(1): +2 150 WAB = 60 WA + 90 WB

DATA1 WA = 120 rpm 2 WAB = 90 rpm 2 150(90 rpm) = 60(120rpm) +90 wg UB=+707pm WB= 70 rpm)



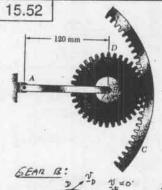
GIVEN: WAB = 42 rpm 2 FIND! (a) WA FOR WHICH Wals 20 mm) (b) WA FOR WHICH WB=0 (CURVILINEAR TRANSLATION

SEE FIRST PART OF SOLUTION OF PROB 15.50 FOR DERIVATION OF

(a) For wa= 20 ypm), mB=-20 ypm EQ(1): +) 150 (42 rpm) = 6000 + 90 (-20 rpm) WA= 135 pm] WA= + 135 mpm

(b) FOR WB=0:

EQ(1): + 2 150 (42 rpm) = 60 WA +0 WA=+105 ypm WA= 105 rpm)



GIVEN: WAB = 20 rad/s) FINO:

> (120mm/20%) = 2,4m/st

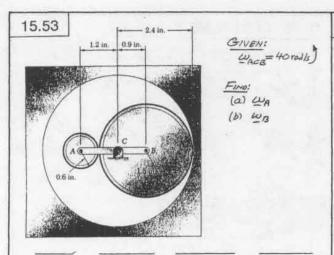
(a) WB (b) VD ARM AB 4 VB=

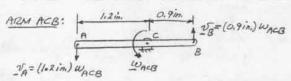
A TOIE UB=2.4 mils

V8= NE+NB/E = 0+(BE) WB (a) BE = 0.05m: 2.4 m/st = 0 + (0.05 m) wg 1 WB = 48 rad/s WB=48 rad/s)

(b) DE = (0.05 VI): TD = VE + VD/E = O + (DE) WB V= 0+ (0.05 V2)(48) TD= 3,39 mls Up = 3,39 m/s 245° €

(CONTINUED)





$$\frac{DISK B:}{\sqrt{2}} = \int_{A \times B} \int_{B \times B} \int_{B$$

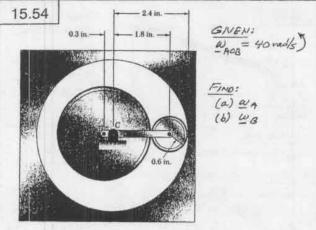
DISK ROLLS ON D:
$$V_D = V_B + V_{DIB} = V_B + (BD) w_B$$

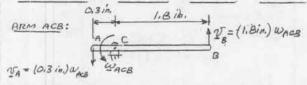
+1 $O = (0.9 \text{ in.}) w_{ACB} - (I.5 \text{ in.}) w_B$
 $w_B = 0.6 w_{ACB} = 0.6 (40 \text{ rad/k})$
 $w_B = 24 \text{ rad/s}$

POINT OF CONTACT E OF THE DISIS: $\Sigma_E = \Sigma_B + \Sigma_{E/B} = \Sigma_B + (EB) \omega_B$ $+ \uparrow \quad \Sigma_E = (0.9 \text{ in}) \omega_{ACB} + (1.5 \text{ in}.)(0.6 \omega_{ACB})$ $V_E = (0.9 \text{ in} + 0.9 \text{in}.) \omega_{ACB} = (1.8 \text{ in}.) \omega_{ACB}$

DISK A

$$U_{E} = V_{A} + V_{E/A} = V_{A} + (AE)W_{A}$$
 $U_{E} = V_{A} + V_{E/A} = V_{A} + (AE)W_{A}$
 $U_{A} = \frac{1.8 + 1.2}{0.6} \omega_{ACB} = 5 \omega_{ACB}$
 $\omega_{A} = 5(40 \text{ rad/s}) = 200 \text{ rad/s}$



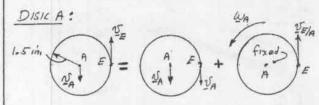


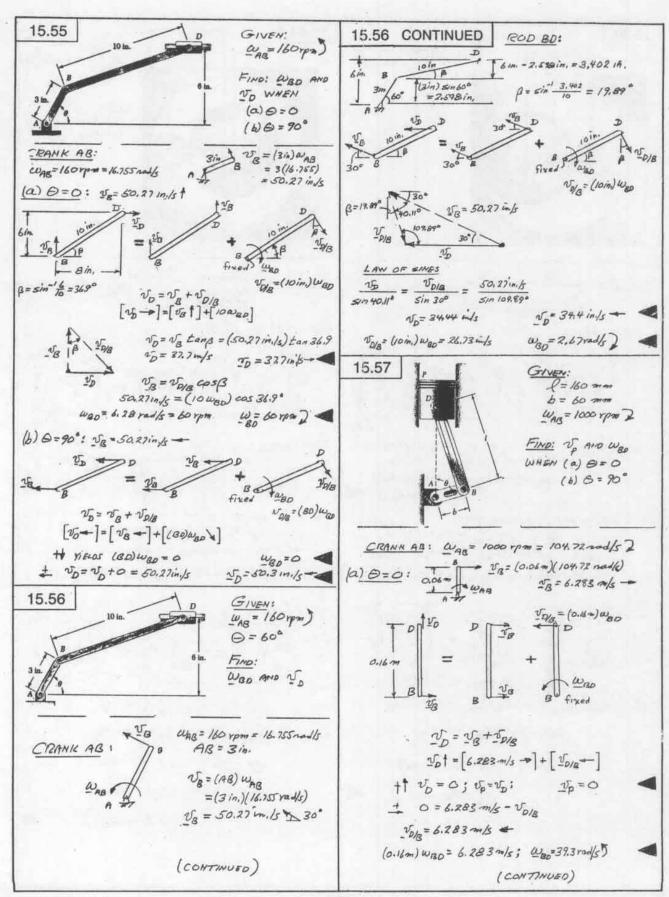
$$\frac{DISUB:}{V_E} = \sum_{B} \sum_{D=B} \sum_{B=B} \sum_{B=B} \sum_{D=B} \sum_{D=$$

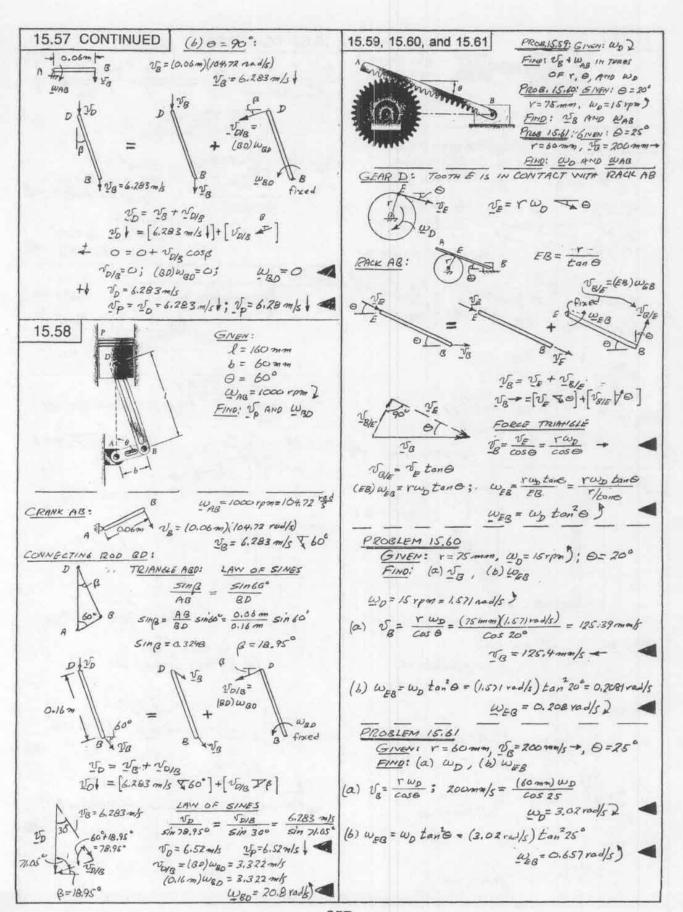
DISK ROLLS ON D:
$$\nabla_D = V_B + \nabla_{D/B} = V_g + (BD) \omega_B$$

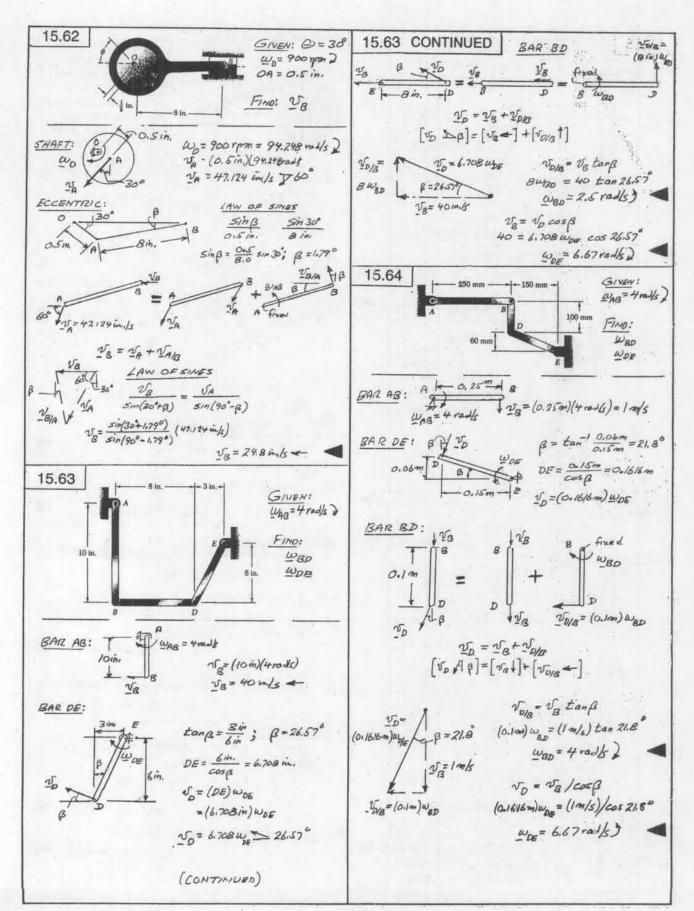
+1 $O = (1.8 \text{ in}) \omega_{ACB} - (0.6 \text{ in}.) \omega_B$
 $\omega_B = 3 \omega_{ACB} = 3 (40 \text{ rad/s})$
 $\omega_B = 120 \text{ rad/s}$

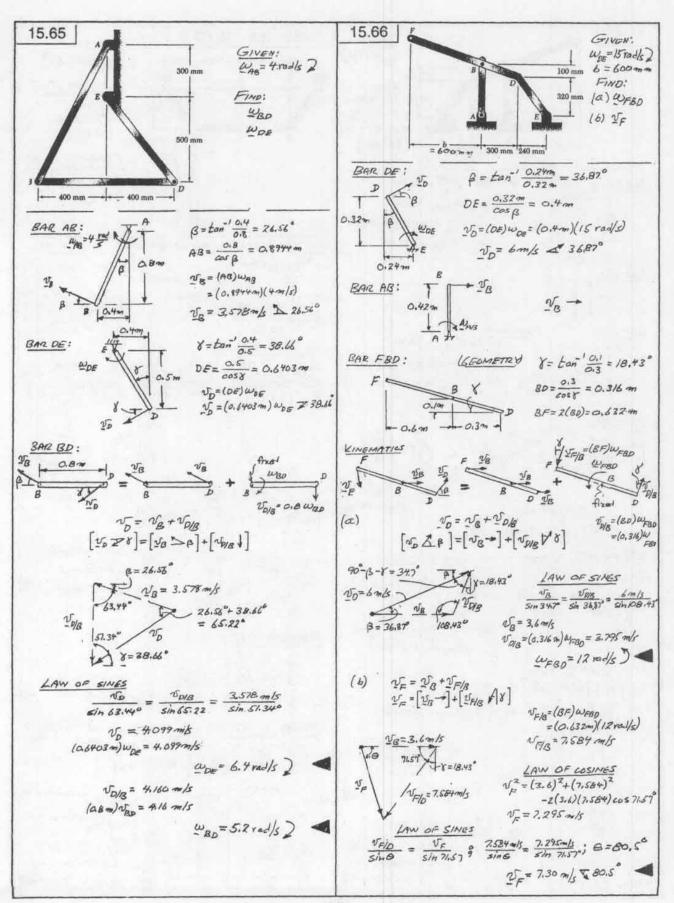
POINT OF CONTACT E OF THE DISIG: $V_E = V_B + V_{E/B} = V_B + (EB) \omega_B$ $+ V_E = (1.8 in.) \omega_{ACB} + (0.6 in.)(3 \omega_{ACB})$ $V_E = (1.8 in. + 1.8 in.) \omega_{ACB} = (3.6 in.) \omega_{ACB}$

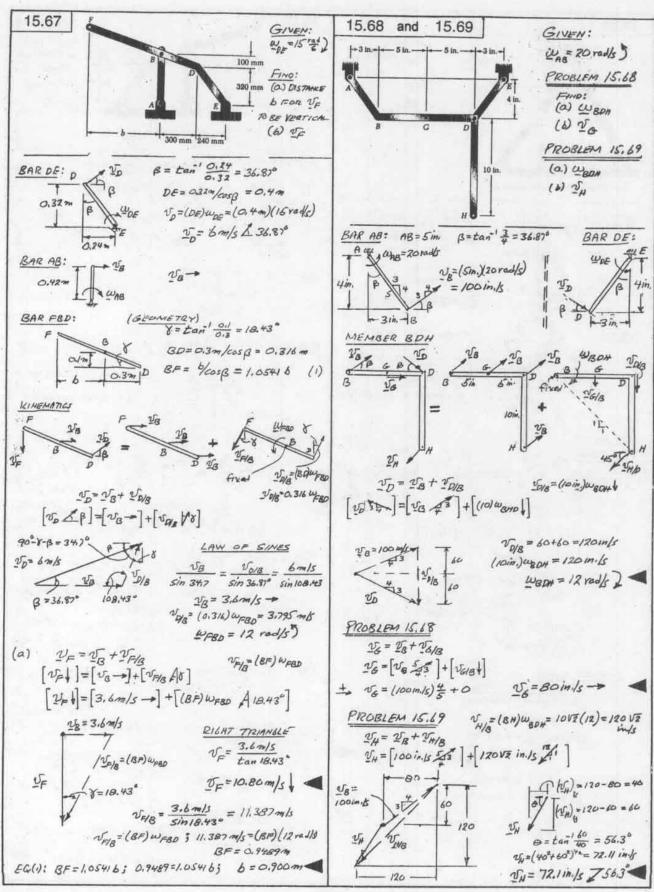


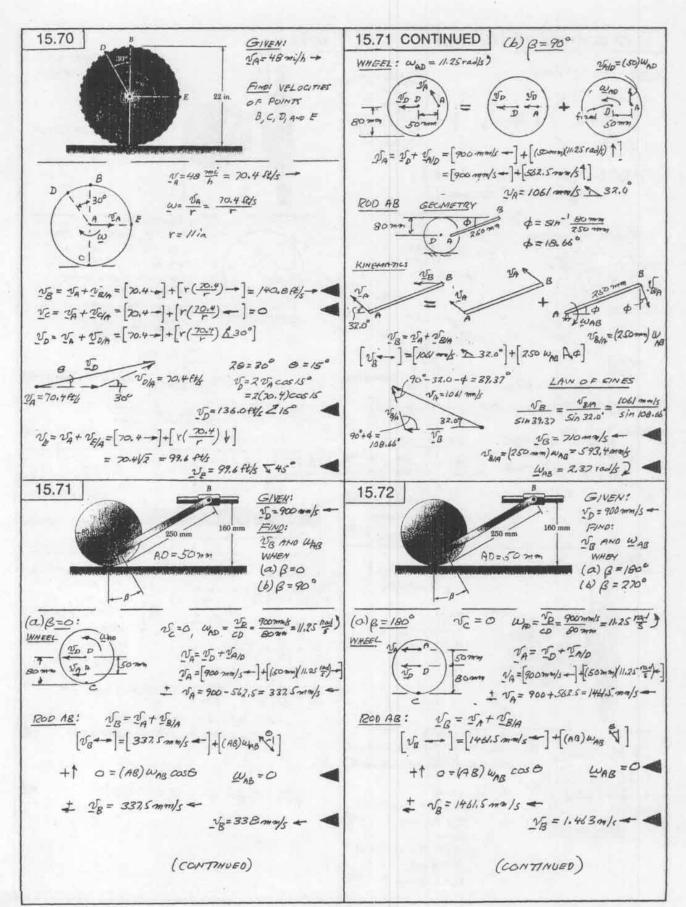


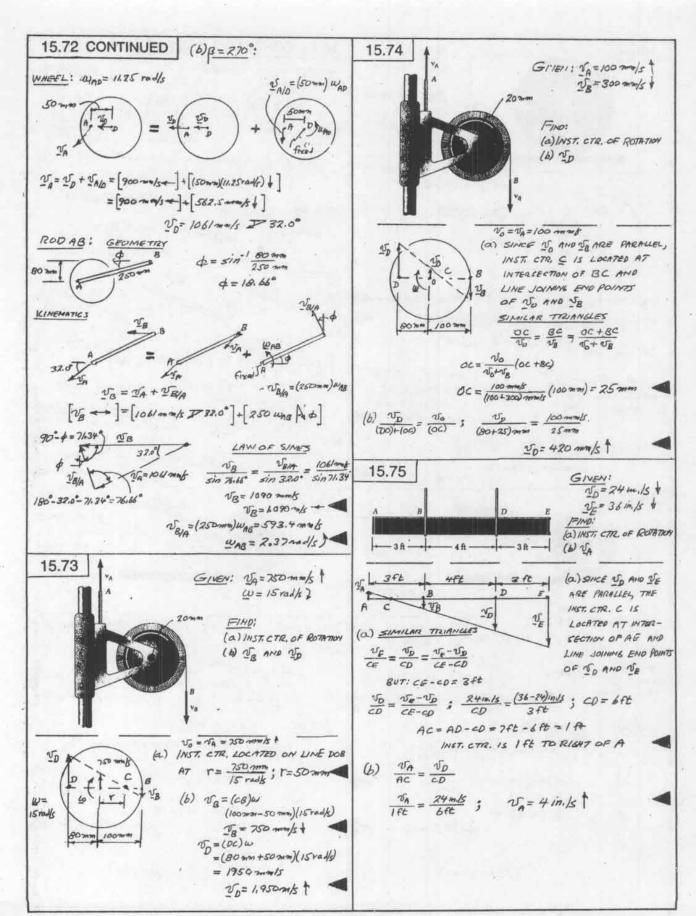


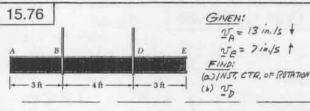


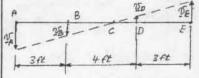












(a) SINCE NA AND NE ARE
PARALLEL, THE INST.

CTR. C IS LOCATED

AT INTERSECTION OF

AE AND LINE JOINING

END POINTS OF NA AND NE

(a) SIMILAR TRIANGLES

$$\frac{AC}{\sqrt[3]{A}} = \frac{CE}{\sqrt[3]{E}} = \frac{AC + CE}{\sqrt[3]{A} + \sqrt[3]{E}}$$

$$AC = \frac{v_A}{v_A + v_E} \left(Ac + cE \right) = \frac{13 in.ls}{(13 + 2) in.ls} \left(10 ft \right) = 6.5 ft$$

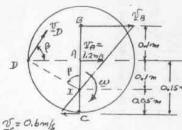
CD = AD - AC = 7Ft - 6.5Ft = 0.5At

INST. CTR. IS 0.5 Ft TO LEFT OF D

(b)
$$\frac{V_D}{CD} = \frac{V_A}{AC}$$
; $\frac{V_D}{0.562} = \frac{13 \text{ in./s}}{6.562}$; $V_D = \int \ln /s$

15.77 $v_A = 1.2 \text{ m/s}$ $r_1 = 150 \text{ mm}$

VA=1.2m/s - VELOCITY OF LOWER RACK IS V=0.6m/s - FIND: (x) W



SINCE NA AND NE
ARE PARALLEL THE
INST. CTR. OF ROTATION
IS AT THE INTERSECTION OF BC AND
THE LINE JOINNA THE
END POINTS OF
NA AND NE

(a) ANGULAR VELOCITY $v_A^-=(AI)\omega$ $1.2 \, rm/s = (0.15 \, m) \, \omega$ $\omega = 12 \, vod f \, 2$

(b) UPPER RACK $V_R = V_g = (RI)\omega$ $V_R = (0.2m)(12 \text{ rad/s})$ $V = 2.4m/s \rightarrow$

VELOUTY OF POINT D: $\beta = \tan^{-1} \frac{0.15 \, m}{0.1 \, m} = 56.3^{\circ}$ $DI = \frac{DA}{\cos \theta} = \frac{0.15 \, m}{\cos 56.3^{\circ}} = 0.1803 \, m$

VD= (DI)W VD= (0.1803m)(12 radk) VD= 2.16 m/s \$ 56.3° 15.78 200 mm/s

GIVEN: IMMER RABUS = 30 mm

OUTER PARIUS = 60 mm

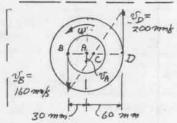
FINO: (a) INST. CTR. OF ROTATION

(b) I BLOCK = VA

(c) LENGTH OF CORE

WRAPPER OR UNWRAPPED PER SECOND

ON EACH PULLEY.



(a) SINCE I'S MO I'S ARE PARALLEL, INST. CTR. IS LOCATED AT THE INTERSECTION OF BD AND LINE JOINING END POINTS OF LE YOU

8C = CD = BC+CD & BUT BC+CD= 90 mm

BC = 90mm; BC=40mm; AC=BC-AB=40mm-30mm=10mm 160 = 360; BC=40mm; AC=BC-AB=40mm-30mm=10mm

b) $\frac{V_{8LOCK} = V_A}{\omega} = \frac{V_B}{\omega} = \frac{V_B}{400mm} = 4 \text{ mod } 5$

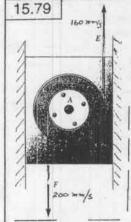
NA=(AC) W=(10mm)(4rad/s)=40mm/s & Succe 40mm/s &

SINCE OF AND VAL, CORD IS UNWRAFFED AT RATE (VA+VO)/IS

VA+VD=40+200=240 ANN/S; ZHO MIND, UNWRAFFED S

[INVERT PULLEY: VB+>VA+, CORD IS UNWRAFFED AT RATE (Vg-VA)/S

VB-VA=160-40=120 MIN/S; 120 MIN, UNIWRAFFED S



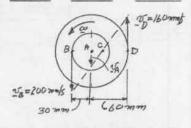
GIVEN: INNER PULLEY = 30 mm OUTER PULLEY = 60 mm FIND: (a) INST, CTR, OF ROTATION

-IND: (a) INST. CIR. OF 120 TATION

(b) IBLOCK = VA

(<) LENGTH OF CORD WIRTHED OR

UNWEAPPED PER BECOMD ON GACH PULLET



BC = 50mm; RC = BC - AB = 50mm = 30mm = 20mm

Not. CTR. C IS 20mm TO RIGHT OF A

(b) \$\int_{\text{BLOGGE}} = V_A: \omega = \frac{\pi_6}{8} \text{BC} = \left(\text{300 mon} \text{ is \omega \text{mon}} \right) \frac{\pi_6}{8} \text{complete} \text{ in \text{1.8}}

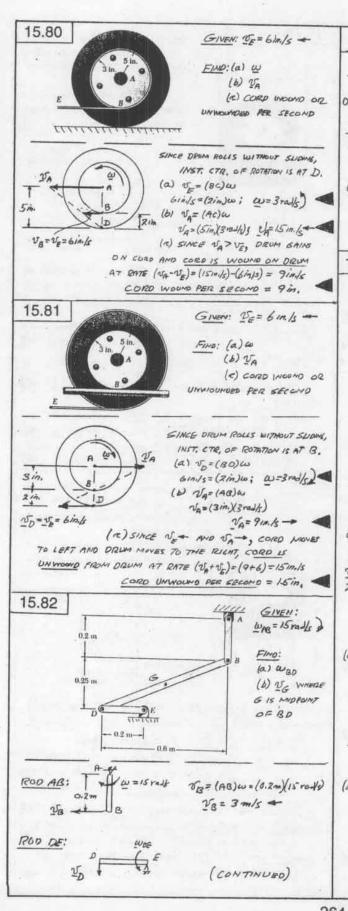
(A) OUTER PULLEY: Upt AMONG to come is UNWRAPPER AT ("STA) S

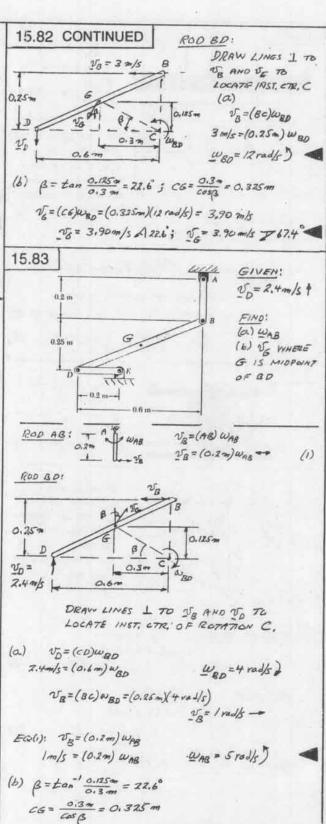
NOTHING TO THE PULLEY: Upt AMONG to come is UNWRAPPER AT ("STA) S

NOTHING PULLEY: Not > Not come is UNWRAPPER AT ("STA) S

INHER PULLEY: Not > Not come is UNWRAPPER AT ("S-"IN) S

Vg-Vg= 200-80= 120 mm/; 120 mm, UNWRAPPED/S

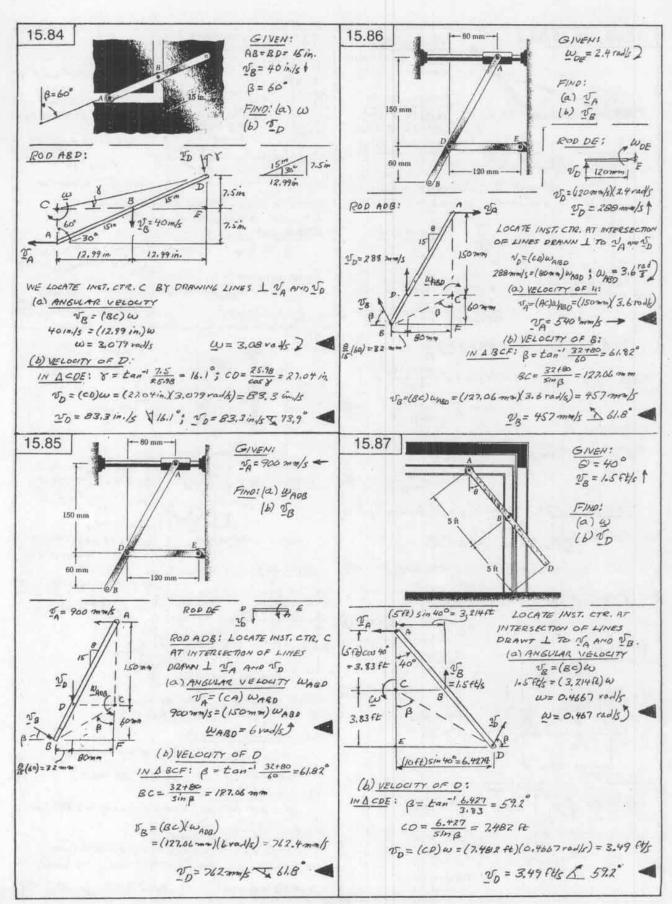


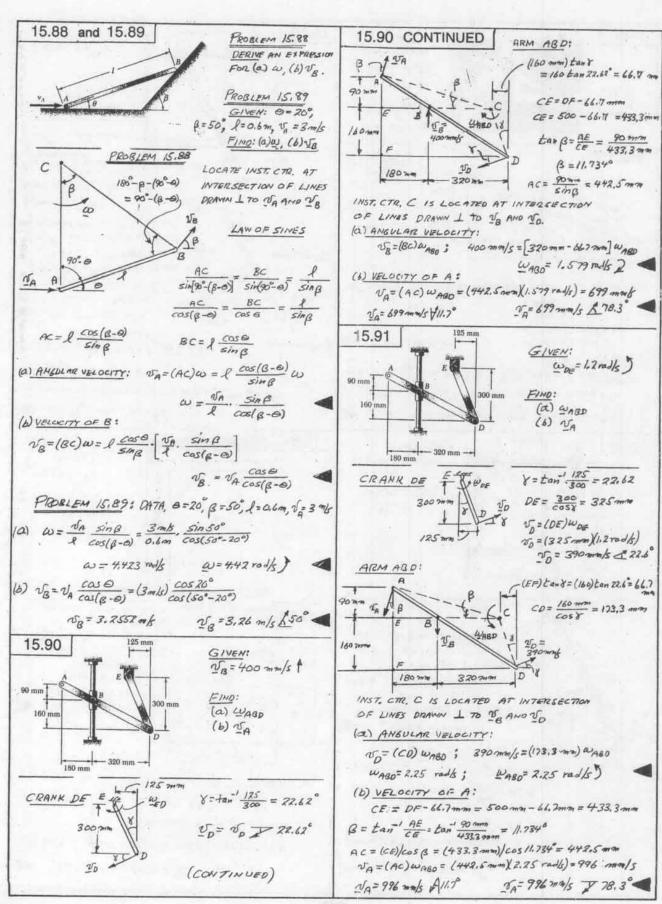


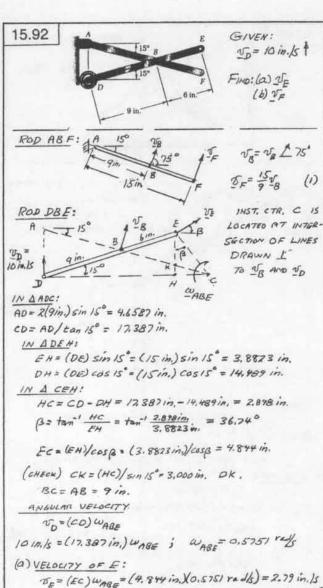
V6=(C6) WBO = (0.325m) 4rad/s)=1.300 mf

N=1.300 m/s 167.4° €

26=1.300 m/s \$22.60



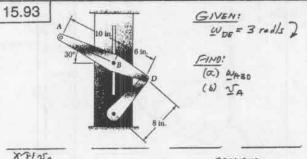


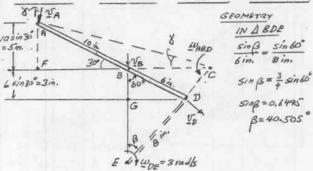


TE = (EC) WABE = (4.844 in. XO.5751 rad/s) = 2.79 in./s VE= 2.79 10./5 136.7°

(6) VELOCITY OF F;

TB = (BC) WARE = (9 in, XO, 575) rad/s) = 5.176 in. 15 Eali): v= 15 18 = 15 (5.176 in. k) = 8.63 in. 15 V= 8.63 in./s \$ 75°





FG = (0E) cosp = (8in) cosp = 6.083 in. IN A GOE: IN A BCE: BC = (BE) tang = [B6+EG] tang = (3 in.+6.083 in) tan B = 7.759 in EC = (BE)/cos B = (3in. +6083in.)/cas B = 11.948in. FB = (AB) cos 30 = (10 ini) cos 30 = 8.660 in FC = FB +BC = 8,660 in .+ 7,759 in = 16.419 in.

IN A AFC: 8 = tan AF = tan 15in. = 16.937° AC = FC = 16.419in = 17.163 in.

ARM DE: VD = (DE) WDE = (Bin.)(3 rads)

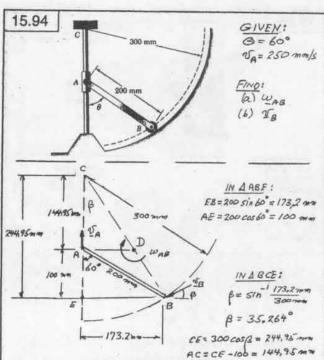
VD = 24 m/s VB MEMBER ABD: THE INST. CTR. C IS LOCATED AT INTERESCTION OF LINES DRAWN I NE AND VO.

CD=EC-ED= 11.946 in. -8in. = 3.946 in.

(a) ANGULAR VELOCITY WARD :

VD=(CD) WARD 24 in./s = (3.946 in.) WABO WARD = 6.08 rad/s) WABO 6.082 rad/s

(b) VECOCITY OF A: NA= (AC) WABO = (17.163 in.) (6.082 rad/c) 24 = 104.4 in.15 Va= 104,4 in./5 \$ 8 = 104,4 in./5 \$ 16,90 VA = 104.4 in./s 7 73.1°



$$\frac{SMILATE TREAMBLES: ACAD AND CEB}{CD} = \frac{AD}{CB} = \frac{CA}{CE}$$

$$\frac{CD}{300 mm} = \frac{AD}{173.2 mm} = \frac{/44.95 mm}{244.95 mm}$$

$$CD = 177.53 mm \qquad AD = 102.49 mm$$

$$BD = CB = CD = 300 mm = 177.53 mm = 122.47 mm$$

THE INST. CTR. IS LOCATED AT POINT D

WHICH IS THE POINT OF INTERSECTION OF LINES DRAWN I TO DA ATTO UB

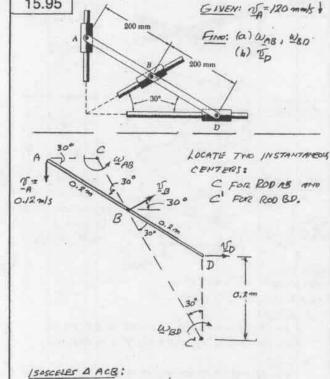
(a) ANGULAR VELUCITY WAS:

$$\nabla_{A} = (AD) \, \omega_{AB}$$

250mm/s = (102,49 mm) ω_{AB}
 $\omega_{AB} = 2.439 \, \text{rad/s}$
 $\omega_{AB} = 2.447 \, \text{rad/s}$

(b) VELOGIY OF B;

$$V_B = (BD) \omega_{AB} = (122.47 mm) (2.439 rad/s)$$
 $V_B = 298.7 mm/s$
 $V_B = 299 mm/s = 735.3$



15.95

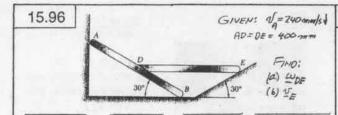
ISOSCELES A BC'D BD = DC'=0,2m BC'= 2(BD) cos 30°=2(0,2m) cos 30°= 0,3464 m a) ROD ABL VA=(Ac)WAB 0.120 m/5 = (0.11547m) WAB WAB = 1.0392 ned/s WAS= 1.039 mays)

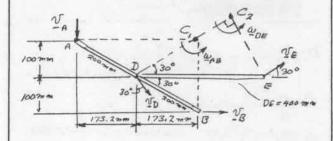
AC=BC=(0.1 m)/cos30 = 0.11547m

25 = (BC) WAB SINCE AC = BC, VB = VA 15 = 0, 12 m/s d

ROD BD: NB = (BC) WBD 0.12 m/s = (0.3464 m) WBD WBD = 0, 3464 rad/s W = 0.346 valls 7

(b) VELOCITY OF D: No= (DC') WBD = (0.2 = (0.3464 radk) Vp = 0.06928 masks V5 = 69.3 mals -





INE LOCATE TWO INST. CTRS. AT INTERSECTIONS OF LINES DRAWN AS FOLLOWS:

C: FOR ROD AB, DRAW LINES I TO TA+ YB

GEONETRY: $AC_1 = (400 \text{ mm})\cos 30^2 = 346.4 \text{ mm}$ $BC_1 = (400 \text{ mm})\sin 30^2 = 200 \text{ mm}$ $DC_1 = AD = 200 \text{ mm}$ $DC_2 = (DE)\cos 30^2 = (400 \text{ mm})\cos 30^2 = 346.4 \text{ mm}$ $EC_2 = (0E)\sin 30^2 = (400 \text{ mm})\sin 30^2 = 200 \text{ mm}$

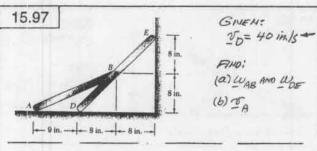
ROD AB: VA = (AC,) WAB; 240 mm/s = (346.4 mm) WAB WAB = 0.69284 rad/s)

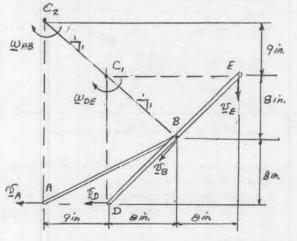
> V5=(DC,) WAB = (200 mm)(0.69284 rad/s) V5=138,57 rad/s A. 30°

ROD DE: $v_D = (DC_2) w_{DE}$ 138.57 mm/s = (346.4 mm) woe

(a) WD= 0.400 red/s WD= 0,4 rad/s)

(b) $V_{\pm} = (EC_2) \omega_{0E} = (200 \text{ mm}) (0.400 \text{ rad/s})$ $V_{\pm} = 80 \text{ mm/s} \qquad V_{\pm} = 80 \text{ mm/s} 130^{\circ}$





INE LOCATE TWO INST. CTRS. AT INTERSECTIONS OF LINES DRAWN AS FOLLOWS:

C; FOR ROD DE, DRAW LINES I TO Up AND YE CZ: FOR ROD AB, DRAW LINES I TO UT AND YB

GEOMETRY! $BC_{j} = (8in.)\sqrt{2} = 8\sqrt{2}$ in. $0C_{j} = 16in.$

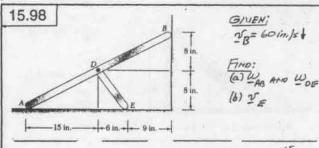
 $BC_2 = (9in + Bin)VI = 17VI in.$ $AC_2 = 25in.$

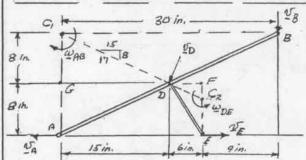
 $\omega_{DE} = 0.4 \text{ rad/s}$ $(a) \underline{ROD DE} : N_D = (DC_i) \omega_{DE}$ $40 \text{ in/s} = (16 \text{ in.}) \omega_{DE}$ $\omega_{DE} = 2.5 \text{ rad/s}$ $\omega_{DE} = 2.5 \text{ rad/s}$

 $T_{B} = (BC_{i}) \omega_{0E}$ $= (BV_{2} in_{i})(2.5 \text{ rad/s})$ $T_{B} = 20 V_{2} in_{i}/s V_{45}$

 $\frac{1200 \text{ AB:}}{20 \text{ V2 in/s}} = (8c_2) \omega_{AB} \\
20 \text{ V2 in/s} = (17 \text{ V2 in}) \omega_{AB} \\
\omega_{AB} = \frac{20}{17} \text{ rad/s} = 1.1765 \text{ rad/s} \\
\omega_{AB} = 1.176 \text{ rad/s}$

(b) $V_A = (AC_2) \omega_{AB}$ = (25 in Xi186 radk) $V_A = 29.41$ in.ls $V_A = 29.4$ m/s





WE LOCATE TWO INST. CTRS. AT INTERSECTIONS OF LINES DRAWN AS FOLLOWS! C, : FOR 1200 AB DRAW LINES LTO VARIOUS CO ! FOR ROD DE DRAW LINES ITE TO AND VE

GEOMETRY: OC, = (82+152) = 17 in. SINCE A GOG AND A DEC, ARE SIMILAR, $\frac{C_2F}{8 in_i} = \frac{C_2D}{17 in_i} = \frac{6 in_i}{15 ln_i}$

C2F = 3,2 in. C2 D= 6.8 in.

EC=8in,-CeF=8-3.2=4.8in.

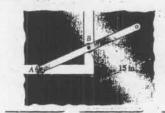
(a) RODAB: NB=(BC,)WAR 60 in./s = (30 in.) WAR WAB= 2 rad/s) WAR = 2 rad/s

> VD=(DC)WAR Vp = (17 in.)(2 rad/s) = 34in./s

1200 DE: ND = (04) WDE 34ini/s = (6.8in.) WE

WOE = 5 rad/s WDE = 5 rad/s)

(6) T=(EC2) WDE V== (4,8in,)(5 rad/s) v= 24 in./s V=24 in./s - 15.99



GNEN AB = BD = 15 in.

DESCRIBE THE SPACE CENTROOF AND BOOY CENTROOF OF ROD ABD.

LET: AB = 1 = 15 in.

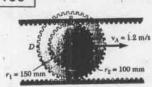
SPACE CENTRODE: COORDINATES OF INST. CTR. $\chi^{2} + y^{2} = \ell^{2}(\cos^{2}6 + \sin^{2}6)^{2} = \ell \sin 6$

x3+42= -12 SPACE CENTRODE IS A QUARTER CIRCLE OF L= 15 in, PADIUS CENTERED AT INTERSECTION OF TRACES IN WHICH INHEELS A AND B MOVE

BODY CENTRODE: DRAW LINE CE WHICH CONNECTS INST. CTR. C AND POINT E LOCATED MIDWAY BETWEEN A AND B.

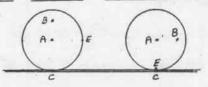
SINCE CE = AE = 1 P = 7. Sin., WE NOTE THAT BODY CENTRODE IS A SEMICIRCLE OF 7.5- in, RADIUS CENTERED AT E.

15.100



GIVEN: GEAR ROLLS ON STATIONARY LOWER IZACK.

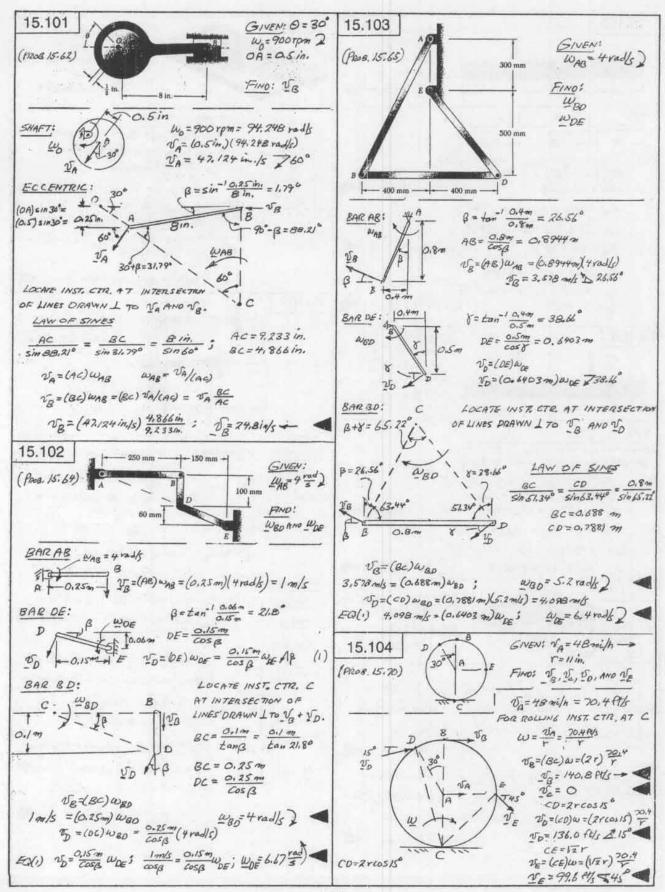
DESCRIBE THE SPACE CENTROOF AND BODY CENTRODE OF THE GEAR.

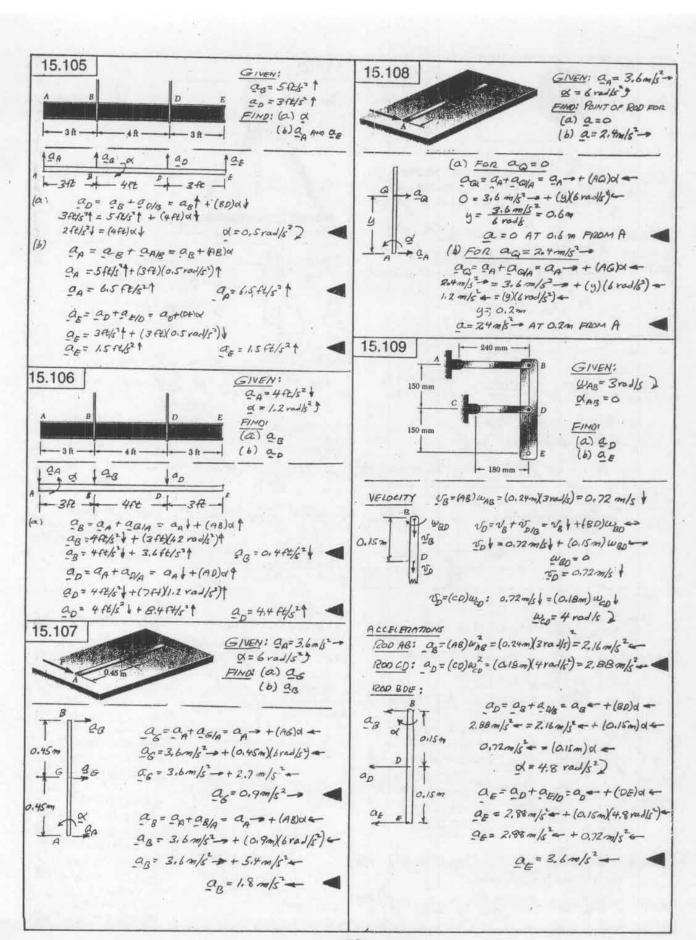


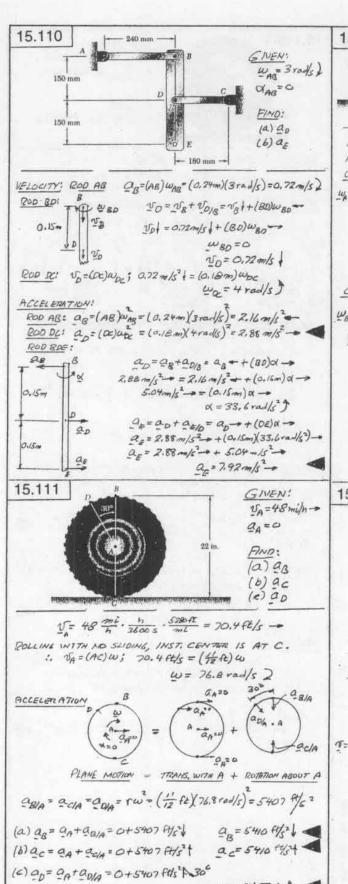
SINCE BEAR ROLLS ON LOWER RACK, THE INST. CTR. IS ALWAYS AT POINT OF CONTACT BETWEEN GEAR AND LOWER RACK.

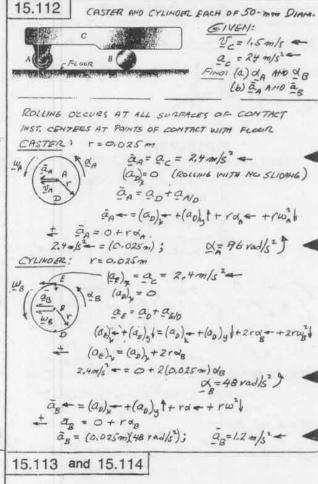
SPACE CONTRODE: LOWER RACK

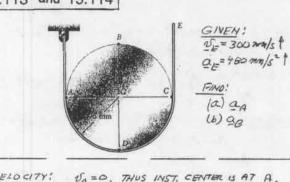
BOOK CENTRODE: CIRCUMFERENCE OF GEAR

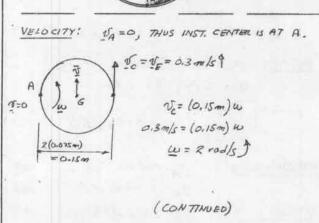




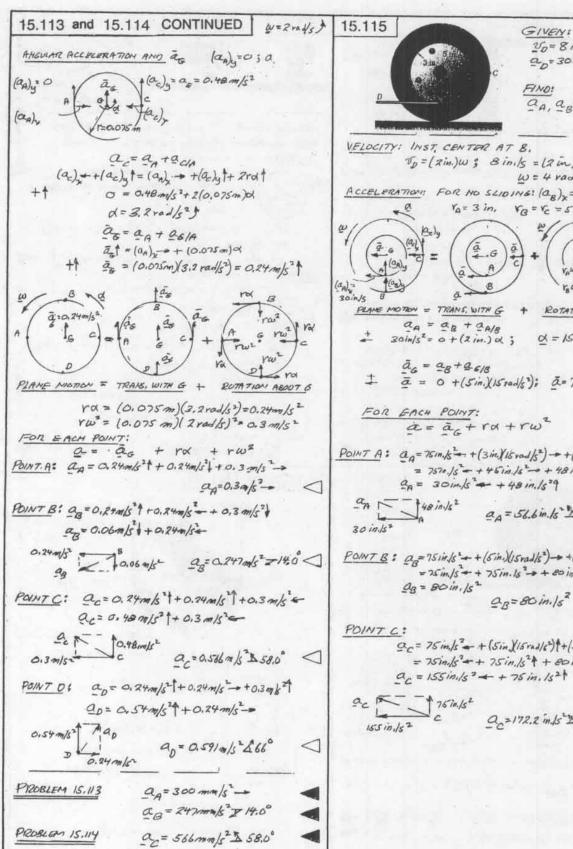


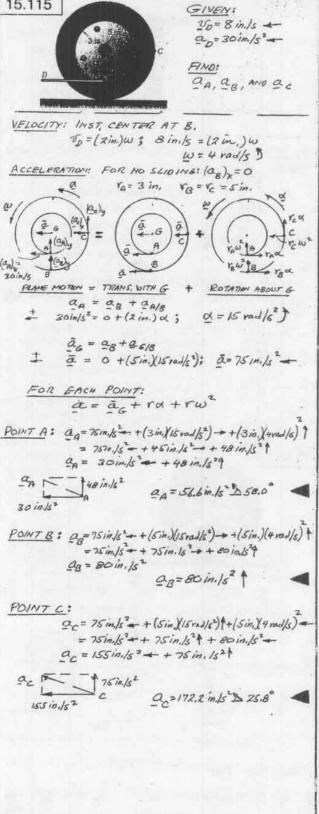




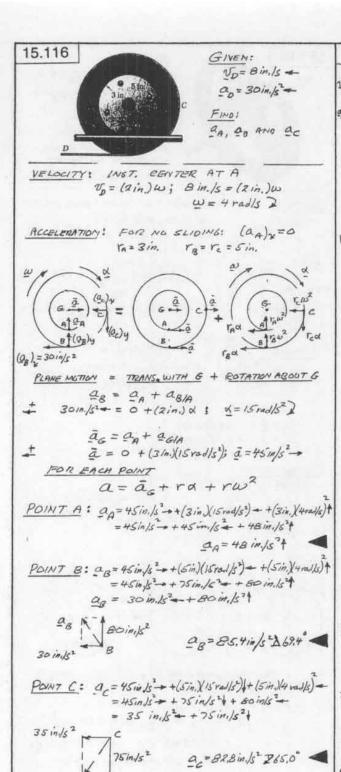


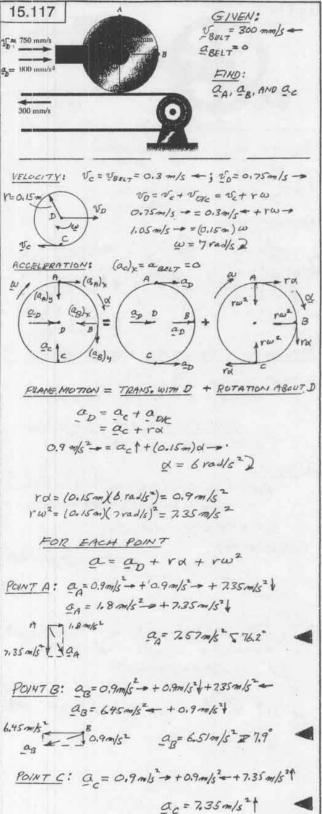
QD=540 P/5 \$ 60°



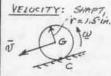


ap= 59/mm/5 2 66.0°









ROLLING, THO SLIDING

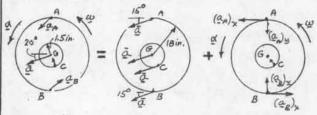
INST. CENTER AT C

\$\times = \text{VW}

U = 0.8 rad/s

\$\text{VW}

ACCELERATION:

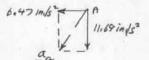


PLANE MOTION = TRANS. WITH 6 + ROTATION ABOUT 6

FOR EACH POINT range 18in.

(a) POINT A: ap=(0.51n,b2) \$\text{\$Z0}^0 + (18in) \(\frac{1}{2} \text{\$Y\$ and } \frac{1}{2} \) \(+ \frac{1}{2} \text{\$I\$ in \(\frac{1}{2} \) \(\frac{1}

an = 6.47 in/s2 + 11.69 in/s2+



a_A=/3.36 in 1/2 761.0° ◀

(b) POINT B:

a_B = (0.5 in./s²) P20° + (18in.) /3 rad/s²) -+ (18in.)(0.8 rad/s²) +

= 0.470 in./s² -+ 0.171 in./s² + 6 in./s² -+ 11.52 in./s²

a_B = 5.53 in./s² -+ 11.35 in./s² +



a= 12.62 in/s \$ 64.0

15.119

GIVEN:

TA = TB = TE = 3 in.; TE = 9 in.

WA = 150 rpm), WA = 0

WE = 0

FINO: MACHITUDE OF

ACCELERATION OF TOOTH

OF GEAR D IN CONTACT

WITH (A) GEAR A, (b) GEAR E.

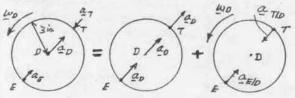
 $\begin{array}{c} V_{\overline{E}LOCITY}: \quad T=7007H \ of \ GEARD \ D \ IN CONTRCT WITH GEARA\\ \hline & \mathcal{G}_{\overline{F}ARS} \\ \hline & \mathcal{T}_{\overline{F}}=\Gamma \omega_{A}=(3 \ \ in.) \ \omega_{A}\\ \hline & \mathcal{S}INCE \ \mathcal{V}_{\underline{F}}=\mathcal{O}, \ E \ IS \ INST. \ CENTEL \\ OF \ GEARD \\ & \mathcal{T}_{\overline{F}}=2r \ \omega_{D}\\ & (3 \ \ in.) \omega_{A}=2(3 \ \ in.) \omega_{D}\\ & \omega_{D}=\frac{1}{2} \ \omega_{A}\\ \hline & \mathcal{T}_{\overline{D}}=r \omega_{D}=(3 \ \ ih.) \frac{1}{2} \ \omega_{\overline{P}}=(h.5 \ \ in.) \omega_{A}\\ \hline & \mathcal{T}_{\overline{D}}=r \omega_{D}=(3 \ \ ih.) \frac{1}{2} \ \omega_{\overline{P}}=(h.5 \ \ in.) \omega_{A}\\ \hline \end{array}$

SPIDED $U_D = (6in.) \omega_S$ $U_D = (6in.) \omega_S$ $U_D = (6in.) \omega_S$ $U_D = \frac{1}{4} \omega_A$

ω= 150 rpm = 15.708 rad/s 2 ω= 2 ω= 7.854 rad/s) ω= 4 ω= 3.927 rad/s 2

ACLELENATION $SPIDSE: M_{S} = 3.927 \text{ rad/s}$ $a_{D} = (40) w_{S}^{2} = (6 \text{ in}) (3.927 \text{ rad/s})$ $a_{D} = 92.53 \text{ in} / s^{2} / s$

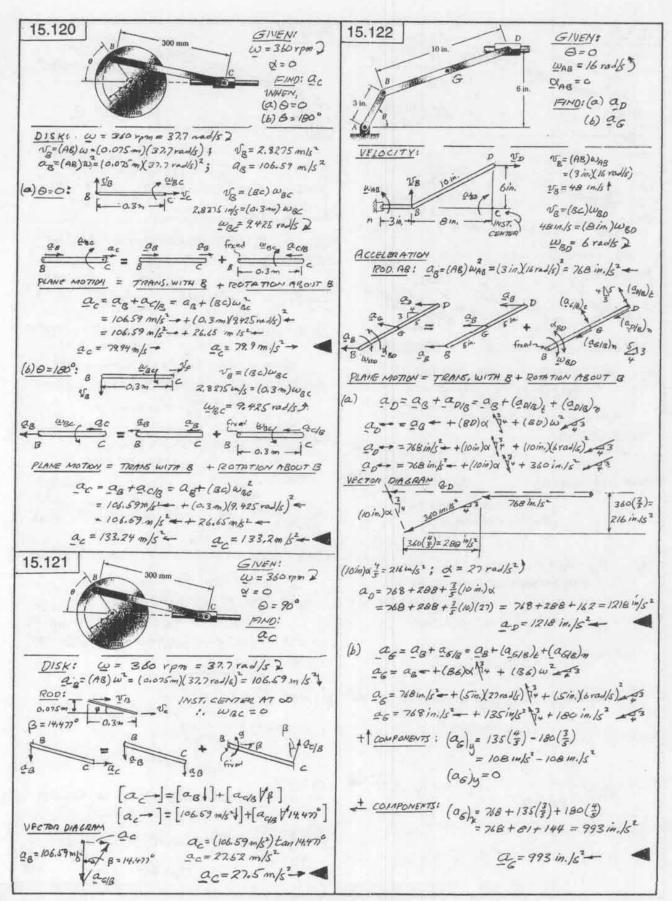
GEAR D:

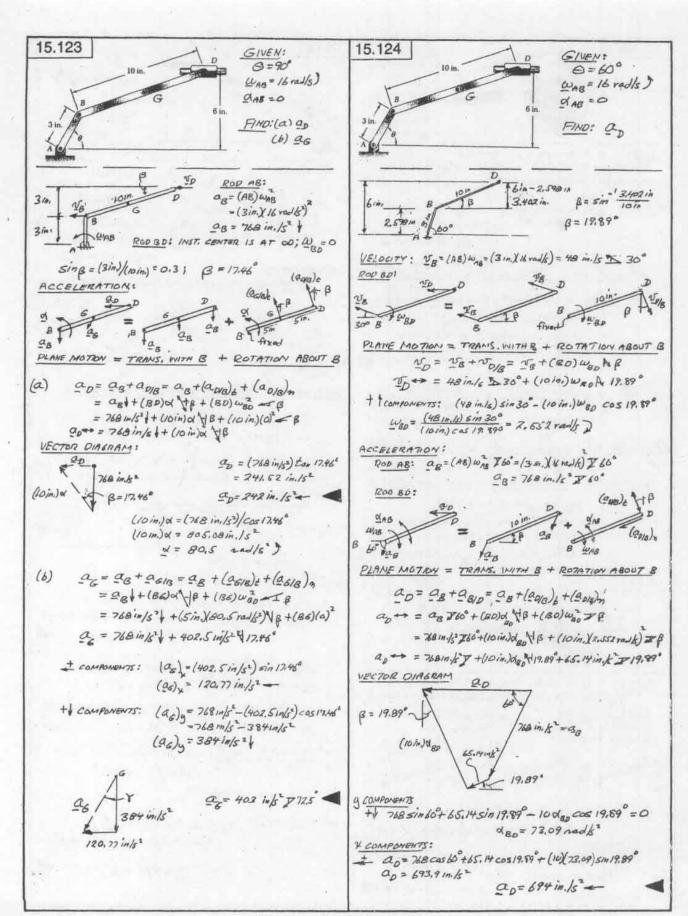


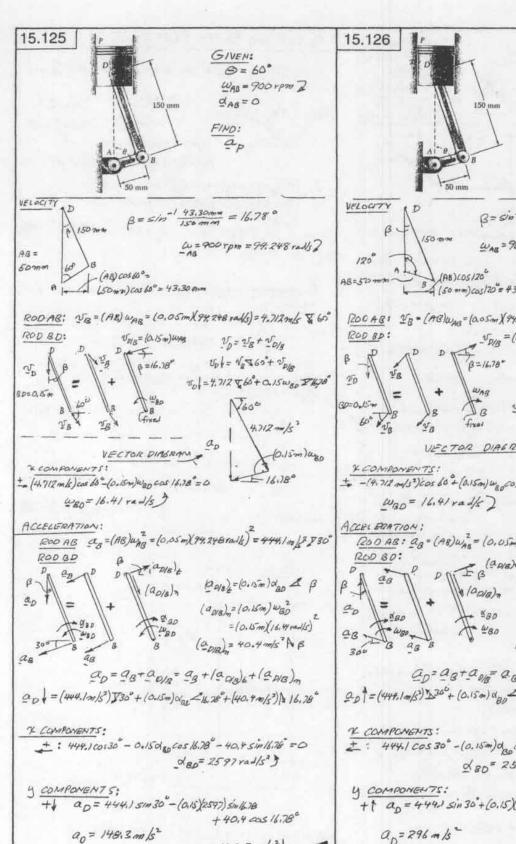
PLANE MOTION = TRANS, WITH D + ROTATION ABOUT D

(a) TOOTH T IN CONTACT WITH SEAR A $\alpha_{T} = \alpha_{0} + \alpha_{T}|_{D} = \alpha_{0} + (07)\omega_{0}^{2}$ $= 92.53 \text{ in/s}^{2} 7 + (3 \text{ in}) 2.854 \text{ rad/s})^{2}$ $= 92.53 \text{ in/s}^{2} /_{D} + 185.06 \text{ in/s}^{2} /_{D}$ $\alpha_{T} = 92.53 \text{ in/s}^{2} /_{D}$ $\alpha_{T} = 92.55 \text{ in/s}^{2} /_{D}$

(b) TOOTH E IN CONTACT WITH SEAR 5 $\alpha_{b} = \alpha_{b} + \alpha_{b} + \alpha_{b} = \alpha_{b} + (E0)\omega_{b}^{\dagger}$ $= 92.53 \text{ in } k^{2} + (3 \text{ in.}) (7854 \text{ rad/s})^{2}$ $= 92.53 \text{ in.} k^{2} + 185.06 \text{ in/s}^{2}$ $\alpha_{E} = 277.6 \text{ in/s}^{2}$ $\alpha_{E} = 278 \text{ in/s}^{2}$

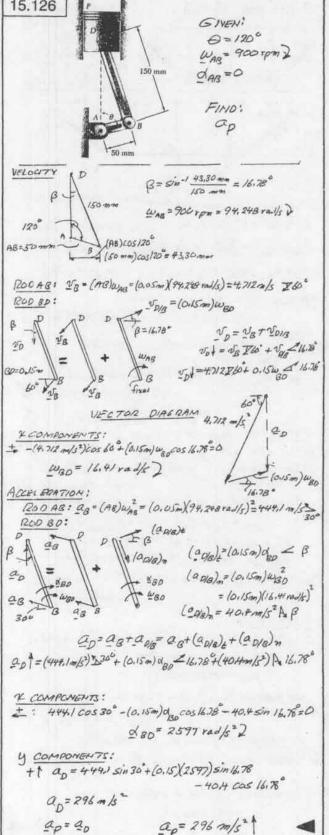


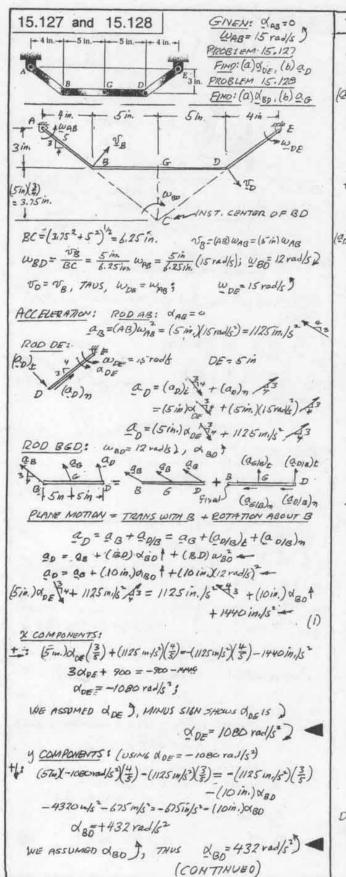




ap=148.3m/s2

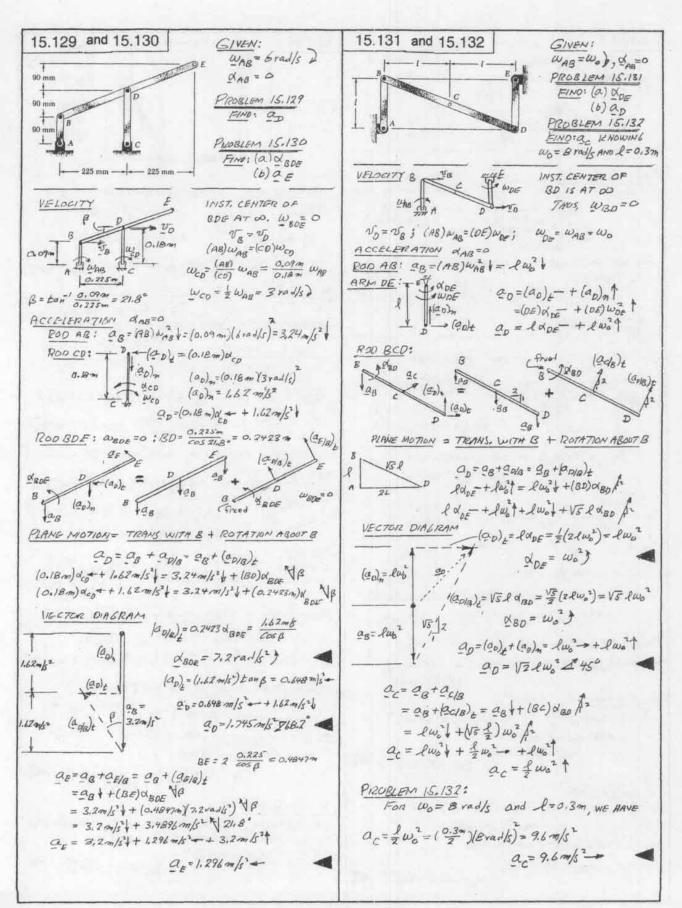
ap=ap

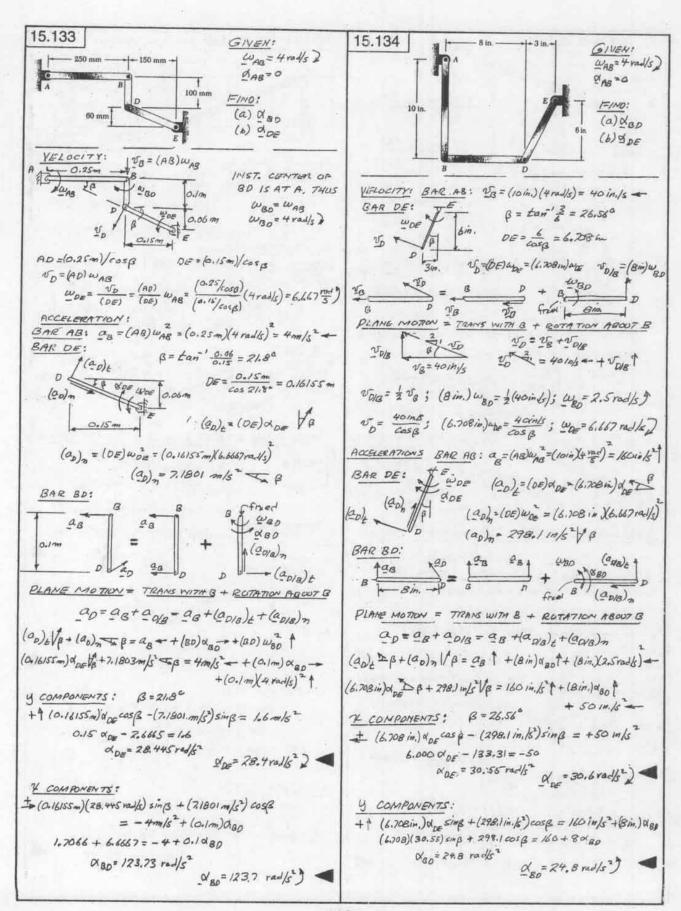


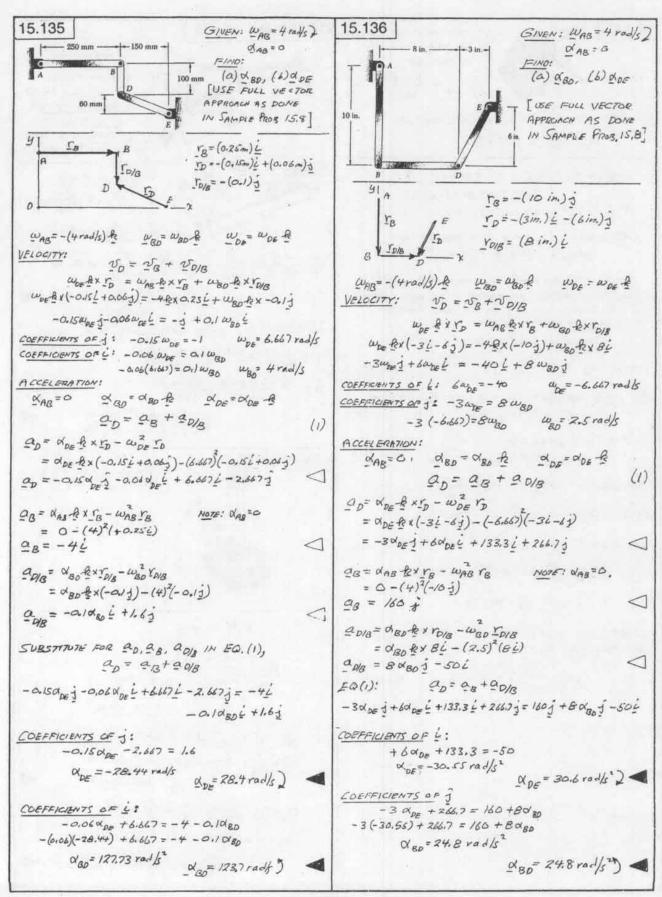


15.127 and 15.128 CONTINUED ACCELERATION OF D: WE KNOW & = 1080 rad/s 2 AND WOE = 15 rad/st ap=(ap) 30 + (as) , 53 Jim DE WOF = (DE) NOE + (DE) WZ = 7 96= (5×1080) + (5×15) Map) a = 5400 m/s2 4 + 1125 m/s2 = 3 $\frac{+}{-}(a_0)_{\chi} = 5400(\frac{3}{5}) - 1/25(\frac{4}{5}) = +2340 \text{ m/s}^2$ (00) y = 5400(4) + 1125(3) = + 4995 m/s2 PD 1 120 y= 4995 ap=5520 in/52 \$64.9° (20) = 2340 = D ACCELERATION OF 6: WE AGAIN USE THE FREE-BODY EQUATION. a= a + a 6/8 = a + (a 6/2) + (a 6/8) m a= a3+ (B6)dsof+ (BD) WBD + RECALL INE FOUND OF BD = +432 rad/s AND USE THIS VALUE HERE TOWNHER WITH UBO = 12 rads AND USE

OB = 1125 in/s = 12 rads a= 1125 m/s243+ 15in. 1+432 rad/s) ++ (5in)(12101/s) -= [900 in. /5 + +625 in /5 2] + 2160 in/s + 720 in/s = a= 2835 in/s + 1620 in/s2= a 2835 in/s2 Q6=3270 m/s2 1 60,3° € 1620 In/52 -VECTOR DIAGRAM OF EG(1) 1=1NIS# 4320 (aD)+ lade) + 675 START 1440 DIMBUSIONS IN: in./52







15.137



INSTANTANEOUS CENTER OF ROTATION AT C (a) SHOW THAT

(b) SHOW THAT ac=0, IF, AND OWLY IF, $Q_{\mu} = \frac{x}{w} \tilde{y}_{\mu} + w x \tilde{y}_{\mu}$

$$N_A = N_c + N_{A/c} = N_c + \omega \times (r_{A/c})$$

$$N_A = N_c + \omega \times (r_A - r_c)$$

$$N_A = \omega \times (r_A - r_c)$$

$$N_A = \omega \times (r_A - r_c)$$

CROSS MULTIPLY EACH MEMBER BY W WX NA = WX [WX (YA - Ye)]

SINCE WI TO PLANE CONTAINING (YA-YO), CROSS MULTIRYING TWICE BY W IS EQUIVALENT TO MULTIRYING (YA-YO) BY WE AND ROTATING IT THROUGH 1800. THUS,

SOLVING FOR VE: YE YA + WX VA (Q.E.D.)

(b) STACE WE WANT Q = 0, WE SHALL WRITE

Q= 2 + 0 C/A = 0 (1)

FROM PART a: VC/A = Vc - YA = WXJA

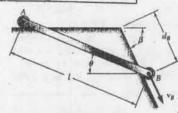
EG(2): $Q_{C/A} = X \times \frac{\omega \times V_A}{\omega^2} - (\omega \times v_A)$

BUT DE ON WEWE, AND SINCE PELVA

aga = \(\frac{\pi}{\omega} \Bullet \frac{\pi}{\pi} \Bullet \times \left(\frac{\pi}{\pi} \times \frac{\pi}{\pi} \right) - \omega \times \frac{\pi}{\pi} \left.

SUBSTITUTING INTO (1) AND SOLVING FOR a A) WE HAVE FOR a=0

*15.138 and 15.139



FROBLEM 15.138

EXPRESS WOF

ROD IN TERMS

OF VB, Q. LAMB

PROBLEM 15.129

EXPRESS & OF ROD

IN TERMS OF VB, Q. L. +B

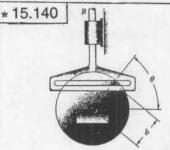
FROSLEM 15.138

 $\frac{d_{\theta}}{\sin \theta} = \frac{R}{\sin \beta}$ $d_{\theta} = \frac{1}{\sin \beta}$ $d_{\theta} = \frac{1}{\sin \beta} \sin \beta$

NB= d(da)= l case de = l cose w

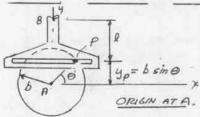
PROBLEM 15.139 NOTE THAT OF STEED.

 $\alpha = \frac{d\omega}{dt} = \frac{N_B \sin \beta}{\ell} \cdot \frac{\sin \Theta}{\cos^2 \Theta} \cdot \frac{d\Omega}{dt}$ $\alpha = \frac{N_B \sin \beta \sin \Theta}{\ell \cos^2 \Theta} \cdot \frac{N_B \sin \beta}{\ell \cos \Theta}$



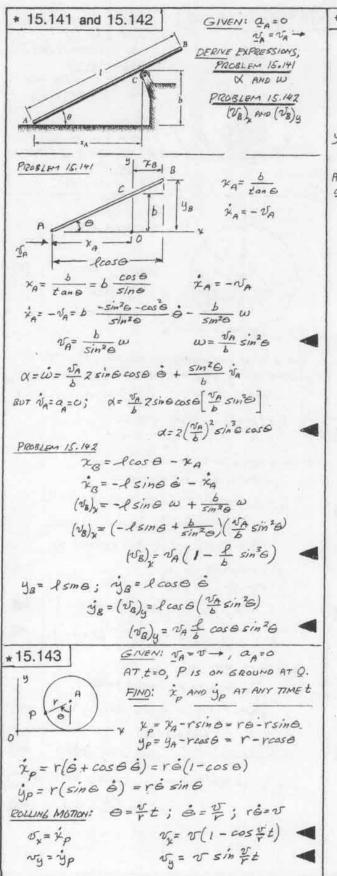
GIVEN: FOR DISK,

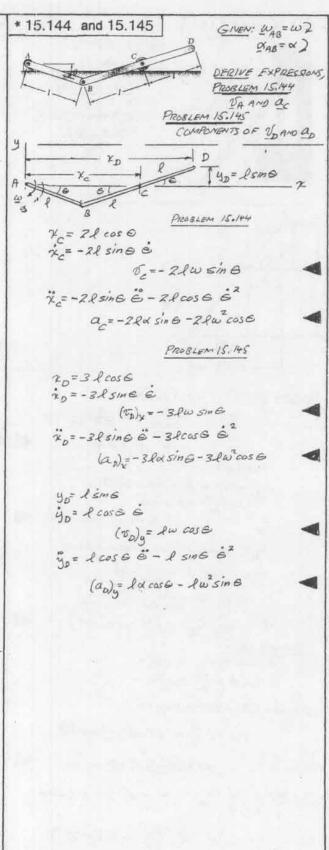
FOR VB AND AB



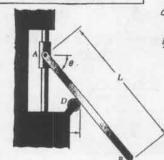
 $y_B = l + y_p = l + b \sin \theta$ $V_B = \dot{y}_B = b \cos \theta \dot{\theta} = b \cos \theta \omega$ $V_B = b \omega \cos \theta$ $Q_B = \dot{y}_B = \frac{d}{d \pm} V_B = \frac{d}{d \pm} (b \cos \theta \dot{\theta})$ $Q_B = -b \sin \theta \dot{\theta}^2 + b \cos \theta \dot{\theta}$

ag = bx cos 0 - b w sin 0





* 15.146 and 15.147



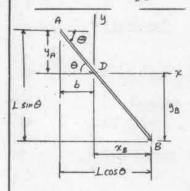
GIVEN: $N_A = V_A$ $Q_A = O$ DERIVE EXPRESSIONS,

PROBLEM IS. 144

(a) was

(b) COMPONENTS OF V_B PROBLEM IS. 147

AB



POSITIVE 615 2

Problem 15.146 $y_A = b tanG$ $v_A' = \dot{y}_A = b \frac{1}{\cos^2 \Theta} \dot{\Theta} = \frac{b\omega}{\cos^2 \Theta}$ $\omega = \frac{v_A}{b} \cos^2 \Theta$

 $\begin{aligned}
\gamma_g &= L\cos \Theta - b \\
\dot{\gamma}_g &= -L\sin \Theta \dot{\Theta} = -L\omega\sin \Theta \\
&= -L\left(\frac{\sqrt{b}}{b}\cos^2\Theta\right)\sin \Theta \\
\dot{\gamma}_g &= \dot{\gamma}_g = -\sqrt{b}\sin \Theta\cos^2\Theta
\end{aligned}$

 $y_{B} = L \sin \theta - y_{A} = L \sin \theta - b \tan \theta$ $\dot{y}_{B} = L \cos \theta \dot{\theta} - b \frac{1}{\cos^{2} \theta} \dot{\theta}$ $= \left(L \cos \theta - \frac{b}{\cos^{2} \theta}\right) \left(\frac{\sqrt{A} \cos^{2} \theta}{b}\right)$

+† $(\mathcal{D}_{\mathcal{B}})_{g} = \mathring{\mathcal{J}}_{\mathcal{B}} = \mathcal{D}_{\mathcal{A}} \left(\frac{L}{b} \cos^{3} \Theta - 1 \right)$

PROBLEM 15.147

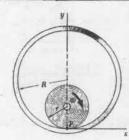
RECALL THAT $\alpha_A = \hat{V}_A = 0$ and $\omega = \hat{\Theta} = \frac{\hat{V}_A}{b} \cos^2 \Theta$

 $\alpha = \frac{\sqrt{a}}{b} \left(-2\cos\theta \sin\theta \right) \dot{\theta}$ $\alpha = -2 \frac{\sqrt{a}}{b} \cos\theta \sin\theta \left(\frac{\sqrt{a}}{b} \cos^2\theta \right)$ $\alpha = -2 \left(\frac{\sqrt{a}}{b} \right)^2 \sin\theta \cos^3\theta$

NOTES SINCE POSITIVE @ IS 2, THE DIRECTION OR & IS).

 $\alpha = 2\left(\frac{\pi_A}{b}\right)^2 \sin 6 \cos^3 0$

*15.148 and 15.149

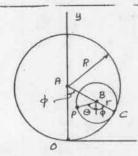


GIVEN: POSITION SHOWN

WEEN ESO W = CONSTANT (x=d) PROBLEM 15-148

DERIVE EXPRESSIONS FOR $(v_p)_\chi$ AND $(v_p)_y$

PROBLEM 15.149
WHEN Y=R/2 SHOW
THAT PATH OF P IS & AXB
AND DERVIE EXPRESSIONS
FOR Vp AND ap



Φ = LOAB G = ANGLE BP FORMS WITH THE VERTICAL

 $6=\omega t$; $\dot{8}=\omega$ (1) $v_{B}=(AB)\dot{\phi}$ $v_{B}=(R-r)\dot{\phi}$

SINCE C IS INSTANTANEOUS CENTER, $V_B = r\omega$ EQUATING THE TWO EXPRESSIONS OBTAINED FOR V_B $(R-r)\dot{\Phi} = r\omega$ $\dot{\Phi} = \frac{r}{P-r}\omega$ (2)

 $\chi_p = (R-r) \sin \phi - r \sin \Theta$ $y_p = R_1 - (R-r) \cos \phi - r \cos \Theta$

DIFFERENTATING AND USING (1) AND (2): $\dot{\chi}_p = (R-r)\cos\phi \dot{\phi} - r\cos\theta \dot{\theta}$ $\dot{y}_p = (R-r)\sin\phi \dot{\phi} + r\sin\theta \dot{\theta}$ $\dot{\chi}_p = (R-r)\cos\phi \left(\frac{r}{R-r}\right)\omega - r\cos\theta \omega$ $\dot{y}_p = (R-r)\sin\phi \left(\frac{r}{R-r}\right)\omega + r\sin\theta \omega$

2p= rω(cos φ -cos ε) yp= rω(sinφ + sin ε)

 $(v_p)_y = \dot{\chi}_p = r\omega \left[\cos \frac{r\omega t}{R-r} - \cos \omega t\right]$ $(v_p)_y = \dot{y}_p = r\omega \left[\sin \frac{r\omega t}{R-r} + \sin \omega t\right]$

PROBLEM 15.149 FOR r = R/2

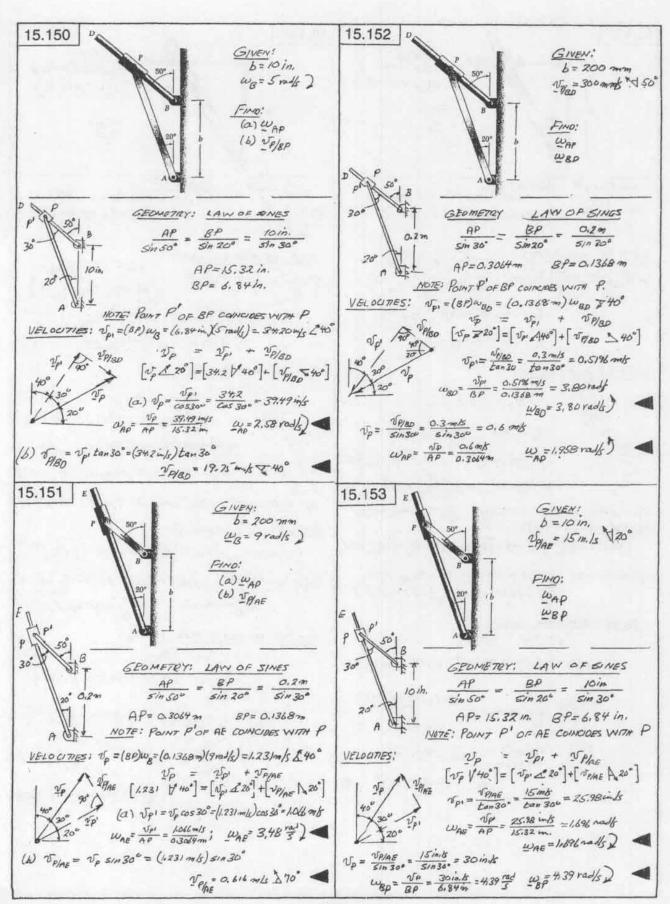
'* = rw(cos wt - cos wt) = 0

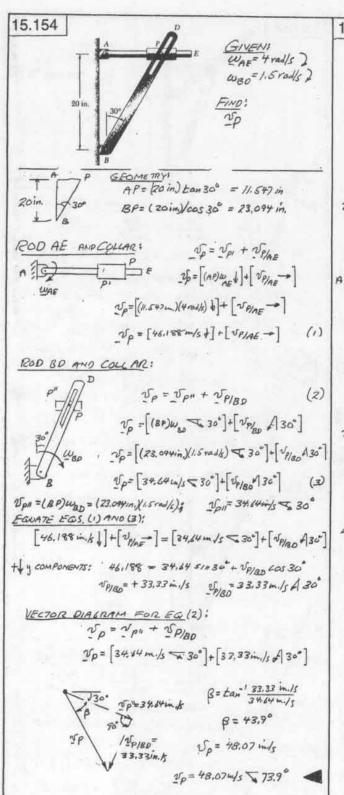
THUS P MOVES ALONG THE Y AXIS

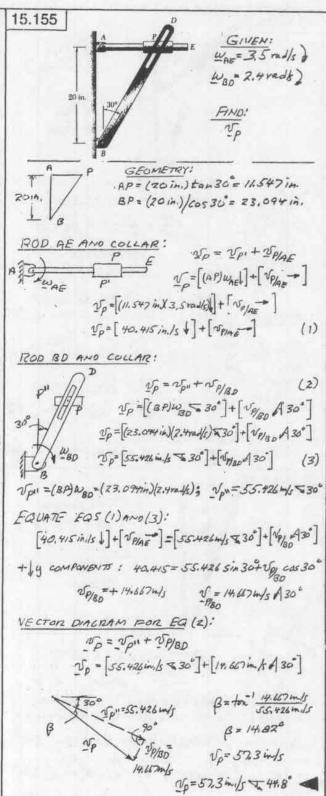
V= yp= γω (sin wt+sin wt) V= 2 rwsinut V=(Rwsin wt) j

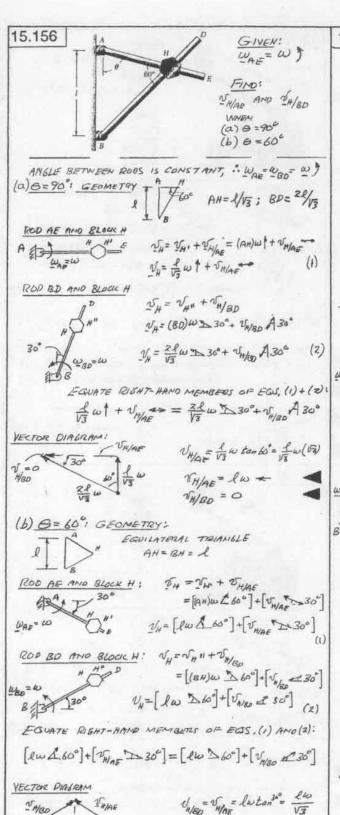
 $a = \frac{dv}{dt} = 2rw(w\cos \omega t)$ [RECALL W=CONSTAINT]

a = (Rw cos wt) j





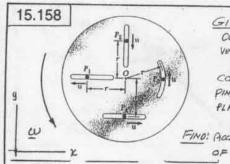




15.157 WAE = W) FIND: TH/AE 24/00 ANGLE BETWEEN ROOS IS CONSTANT, . WAF WED = W) LAW OF SINES GEOMETRY: AH= 1.115 & BH= 0.8/65 R ROD AF AND BLOCK H: VH = VH. + WHIAE = [(AH)W 1450] + [VHAE 1450] E V=[1.115 Pw & 45°]+[VHAF 1 45"] ROD BD AND BLOCK H: VH = VH" + VHIBO =[(BH)W \$750]+[UHBO=150] V- [0.81658md 75°]+[VWBD # 15°] (2) EQUATE RIGHT-HAND MEMBERS OF EQS, (1) AND (2): [1.115 RW & 45°] + [ina= 45°] = [0.8165 Rw \$75] + [ina= 5] VECTOR DIAGRAM EQUATE COMPONENTS IN DIRECTION PARALLEL TO VI 11 145° U=0.8165 Ru EQUATE COMPONENTS IN DIRECTION PARALLE TO UHI ALE (0.8165 lw) cas 60°+ N/180 cas 30°=1.115 lw VH/B0 =+ 0.816 lu 25H/80 0.816 Ru 15° EQUATE COMPONENTS IN DIRECTION PERFENDICULAR TO VIN \$ 450 (0.8165 PW) sin 600- 1/1/00 sm 300 = NH/AE (0.8165 Pa) sings - (0.816 Pa) sin 300 = VA/AE 1/HAF 0.299 fu \$ 45° ◀ NH/AE + 0, 299 Pm

JH/AF 130°

1/10 1 1 30°



GIVEN: CONSTANT ANGLAR VELOGITY = W)

CONSTANT SPEED OF PINS RELATIVE TO PLATE = UL

FIND! ACCELERATION OF EACH PIN.

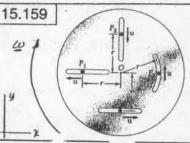
ap = api + app + ac FOR EACH PIN: ACCELERATION OF CONCIDING POINT P': FOR EACH PINI

SPI = YOU TOWARD CENTER D ACCELERATION OF PIN WITH RESPECT TO PLATES

FOR P, PZ, AND P4: apper 0

FOR P4: ap/g= U/4 TOWARD CENTER O

CORIOLIS ACCELERATION FOR EACH PIN Q= 24 W, WITH OR IN A DIRECTION OBTAINED BY ROTATING IL THROUGH 90° IN THE SENSE OF W.



GIVEN: CONSTANT ANGULAR VELOCITY = W]

CONSTANT SPEED OF PINS RELATINE TO PLATE = U

FOR EACH PIN: ap = ap+ + ap/g+ ac ACCELERATION OF COINCIDING PONT P'!

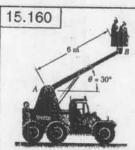
FOR EACH AN: api = rw2 Toward CENTER O ACCRETATION OF PIN WITH RESPECT TO PLATE!

FOR P, P2, AND P4: ap/g =0

FOR P4: app = 42/1 TOWARD CENTER O.

COMOUS ACCELENATION; FOR EACH PIN, Q = 2000, WITH ar IN A DIRECTION OBTAINED BY ROTATING U THROUGH 90° IN THE SENS OF W.

$$a_3 = [r\omega^2] + [\frac{u^2}{r}] + [2u\omega^2]; \quad a_3 = -(r\omega^2 + \frac{u^2}{r} - 2u\omega)L$$



GIVEN! WAB = 0.08 rads 2 BABEC VS/A = 0.2 m/s 7 30 08/A = 0

FINDS (a) NB (6) aB

(0) VELOCITY'S B

VE/3 = VOIA = 0.2 m/s 7366 TB = VB, +V8/9

VB=[(AB)w \$600]+[0.2m/s #300] B 1000 =[(6m)(0.08mk) \$ 66°]+[0.2m/5 7 30°] 0.2915 18 = 0.48m/s \$ 60 + 0.2 m/s \$ 300 10.48 2/5 VE = 0.52m/s \ 82.60 B=22.6°

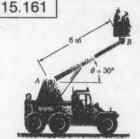
25 (b) ACCEL ERATION:

28' = (AB)W=(6m)(0,08 vad/s)=0.0384 m/s=38.4 m/k 7 30" ac= 2 uw = 2(0.2 m/s)(0.08 rad/s) = 0.032 m/s= 82 ma/s 1/2 600 a=[38.4mm/s 730]+0+[32mm/s 1 66]



B = 39.8° ag=50.0 mm/s

an = 50.0 ma/s \$ 9.8°



GIVEN: WAB = 0,08 rad/s 2 MAB = 0 TBIA = 0,2 m/s 2 30° ABIA = 0 FIND: (a) VB

(C) VELOCITY

0,2m/s

VB/8= VB/A = 0.2 m/s 7 300 VB = VB, + VB/8=

(b) aB

VB=[(ABW \$ 60°]+[VB/8-2 30°]

=[600)(0.08 m/s) \$ 60°]+[0.2 m/s 230°] NB = 0.48m/5 5 60° + 0.2m/5 2 30° B=22.6

8=90-30-22.6=37.4°; N=0.52m/s \$ 37.4° an = an + any +ac; any =0 (b) ACCELBRATION

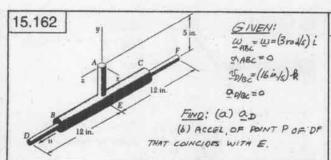
ar = (AB) = (6m)(0.08 rads) = 0.038+m/5= 38/+mm/5 230° a = Zuw = 2(0,2m/s)(0,08 rods) = 0,032m/s= 32mm/s \$ 60° 08=[38,4mm/s 730]+0+[32777/s = 666]



aB

B=39.8° ag=50,0mm/5

aB=500ma/52 7 69.80



(a) POINT D: \(\psi_{\mu_g} = \psi_{\mu_g} = (16 \text{ in } \sigma_s) \text{.} \\ \frac{A}{a} = -(5 \text{ in}) \frac{1}{2} + (12 \text{ in}) \text{.} \\
\[\alpha_0 = -(5 \text{ in}) \frac{1}{2} + (12 \text{ in}) \text{.} \\
\[\alpha_0 = -(5 \text{ in}) \frac{1}{2} + (12 \text{ in}) \frac{1}{2} \\
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\[\alpha_0 = -(5 \text{ in}) \frac{1}{2} + (12 \text{ in}) \frac{1}{2} \\
\]
\[\alpha_0 = -(5 \text{ in}) \frac{1}{2} + (12 \text{ in}) \frac{1}{2} \\
\alpha_0 = -(5 \text

 $= [(45in/s)\frac{1}{2} + (108in/s)\frac{1}{4}] + 0 + [-(96in/s)\frac{1}{3}]$ $a_D = -(51in/s)\frac{1}{2} + (108in/s)\frac{1}{4}$

(b) POINT P OF DF THAT CONCIDES WITH E

APPS = DIGC = (16 in.15) - 12; apps = 0

AE = -(sin) - 13

AP = WXWY AE = -WAE = -(3rolls) AE = (45 in.15) - 1

AR = 2WXVOIG = 2[(3rolls) - 12] (16 in.15) - 1

AP = Ap + Apps + AR

= [(45 in.15^2) - 13] + O+[-(76 in.16^2) - 13]

Ap = -(5/ in.15^2) - 1

15.163

5 in.

Consen:

Conse

(b) POINT P OF DF THAT COINCIDES VAITH E

**Spig = **Spig = (16 in./s) &; ap/g=0

AE = -(5 in.)s; No! = wx AE *(2 rad/s) &x (-5 in./s) s=0

ap = wx NE! = 0

a= = 2 w x NE! = 2 (3 rad/s) &x (16 in./s) &= (96 in./s) i

ap = ap + ap/g + ac

ap = 0 + 0 + (96 m/s) i

a= (96 in./s) i

15.164

GIVEN: ELEVATOR MOVES DOWNWARD AT 40 4%

Fino: CORDULIS ACCELETATION OF ELEVATOR IF

17 IS LOCATED AT: (a) EQUATOR, (b) 40° NORTH, (r) 40° SOUTH.

EARTH MAKES ONE REVOLUTION IN 23 h 5600 = 23,933 h

W = 211 rad

(73,932h/3600 5/h) = 72,92 × 10° 6 yad/s *

W = (72,92 × 10° fad/s) \(\frac{1}{2} \)

W = (72,92 × 10° fad/s) \(\frac{1}{2} \)

U = (72,92 × 10° fad/s) \(\frac{1}{2} \)

Q = 2 \(\text{DX X U} \)

= 2(77,91 × 10 \text{Vad/s}) \(\frac{1}{2} \)

Q = (5.83 × 10° ft/s²) \(\frac{1}{2} \)

(b) AT 40° NORTH:

(0) AT 40 NORTH:

4 = 40 Fds (-cos 40" L - sin 40 j)

Q = 2 Q x 4 = 2(72,92 x 0 fradls) j x (40 Fds) (-cos 40" L - 51 n 40 j)

Q = (4.47 Fd/s²) & Q = 4.47 Fd/s² WET

(c) AT 40 5 SOUTH;

(e) 47. 40 5007#: U = 40 Fth (-cas 40° i + sin 40° j) ac= 2w1 u = 2(72.92×10° radk) j x(40 fb/s)-cos 40° i + sin 40° j) ac= (447×10°) ft/s*

ac= (447×10°) ft/s*

Ac= (447×10°) ft/s*

Ac= (447×10°) ft/s*

* NOTE: EARTH ROTATES COUNTER CLOCKWISE WHEN OBSERVED FROM ABOVE THE NORTH POLE,

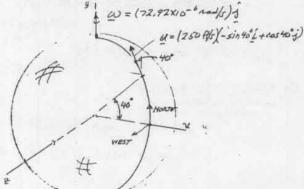
15.165

GIVEN: TEST SLED MOVING DUE NORTH
AT 900 PM/h, AT 40° NORTH LATITUDE,
FIND: CORYOUS ACCELERATION OF

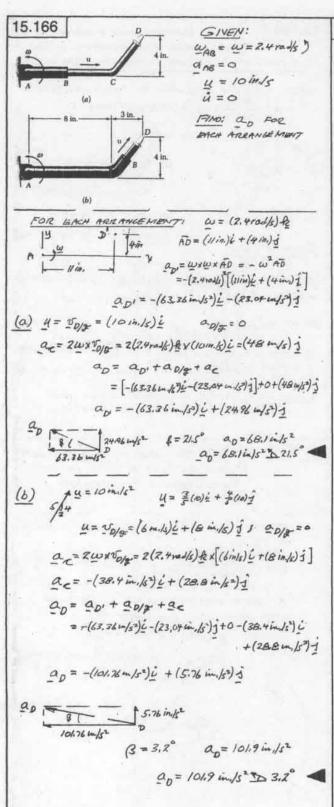
FARTH MAKES ONE REVOLUTION IN 234 56 m

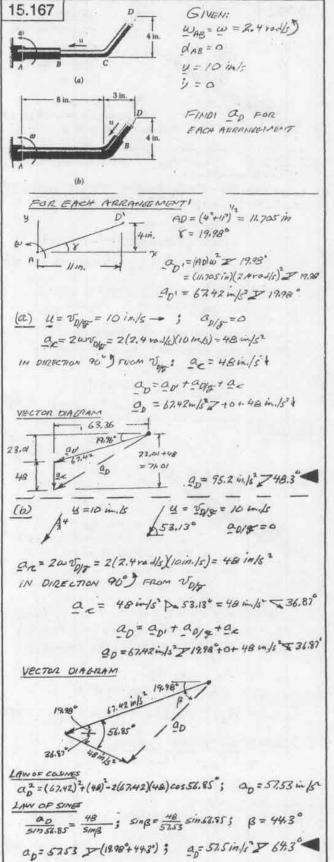
w= \(\left(20,72d)\frac{1}{2}\) = \((72.92\)\frac{1}{2}\)\frac{1}{2}\\
\(\mu = 900\)\kappa_n\|_n = 250\\mathred{n}\|_5\)

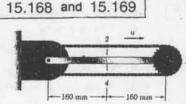
NOTE: BARTH ROTATES COUNTERCLOCKWISE WHEN WIEWED FROM ABOVE THE NORTH POLE



ac = 2wxu = 2(72.92x10 rad/s) & x(250 m/s) - sin40 i + cos40 i)
ac = (23.4x16 3) m/s 4 = (23.4x10 3) m/s west
ac = (23.4x16 3) f/s 2 70 LEFT OF SED





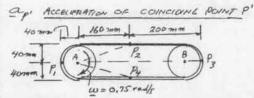


 $\frac{G_{IVEN}}{\omega_{HS}} = 0.75 \text{ rod/s}$ $\frac{\omega_{AS}}{\omega_{AS}} = 0$ $\omega = 80 \text{ somm/s}$ $\hat{\omega} = 0$

FIND; ACCEL, OF LINKS PROB. 15.168: LINKS | + 2. PROB. 15.169: LINKS 3 + 4

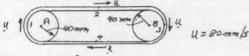
 $a_p = a_{pi} + a_{pi} + a_{k} \qquad (1)$

EACH TERM IS COMEUTED SEPARATELY FOR EACH LINK



 $Q = (AP_1)W^2 = 40 \times 0.75^2 = 22.5 \text{ mm/s}^2$ $Q = (AP_2)W^2 = |60 \times 0.75^2 + 40 \times 0.75^2 + 22.5 \text{ mm/s}^2$ $Q = 90 \text{ mm/s}^2 + 22.5 \text{ mm/s}^2$ $Q = (AP_3)W^2 = (160 + 200)0.75^2 = 202.5 \text{ mm/s}^2$ $Q = (AP_4)W^2 = |160 \times 0.75^2 + 40 \times 0.75^2 + 22.5 \text{ mm/s}^2$ $Q = 90 \text{ mm/s}^2 + 22.5 \text{ mm/s}^2$

OP/S ACCELERATION OF PRELATIVE TO ROTATING FRAME



app = u2/r = (80)/40 = 160 mm/s2 ->
app = app = 0
app = u2/r = (80)2/40 = 160 mm/s2 ->

a CORIOUS ACCELERATION

MAGNITUDE FOR ALL LINKS

Q= 2 Wu = 2(0.75 rod/s /80 mm/s) = 120 mm/s

DIRECTION ROTATE U THROUGH 90° 2

LINK 2: ax = 120 mm/s = +

LINK 3 1 ac = 120 mm/s2 +

UNK 4: ac=120 mm/5 1

ap = ap + app + ax

PROBLEM 15.168:

LINK 1: , ap = (22.5 mm/s -) + (160 mm/s -) + (120 mm/s -)

ap = 302.5 mm/s ->

LINK 2: a= (90mm/s=+22.50mm/s=+)+0+(120 mm/s=+)

ag = 90 mays a + 142.5 mm/s = 168.5 mm/s 767.7

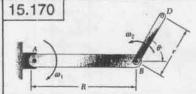
PROBLEM 15.169:

LINK 3: ap3 = (202,5 mm/s =)+(160 mm/s =)+(120 mm/s =)

ap3 = 482.5 mm/s =

LINK 4: apy = (90 mm/s = +22.5 mm/s +) +0+(120 mm/s +)

apy = 90 mm/s = +142.5 mm/s + = 168.5 mm/s 2 5270



GIVEN:

With THAT QLD

PASSES THROUGH A

AND THAT THE RESULT
IS INDEPENDENT OF R, F, B

 $a_{D} = a_{D} + a_{D}/\varphi + a_{X}$ $w_{x} = 2\omega, \quad D$ $BB = r\cos\theta i + r\sin\theta j$ $A = \frac{\omega_{x}}{R} = \frac{\omega_{x}}{R} + \frac{\omega_{y}}{R} = \frac{\omega_{z}}{R} = 2\omega, \quad R$

AD= (R+rcose) + rsm6 j

ap = - w, (AD) = - (R+rcose) w, L - rw, sine j

 $\mathcal{D}_{P/g} = \omega_2 \times (80) = 2\omega_1 + x(r\cos L + r\sin 6 \frac{1}{2})$ $\mathcal{D}_{P/g} = -2\omega_1 r \sin 6 \frac{1}{2} + 2\omega_1 r \cos 6 \frac{1}{2}$

20/5= + W2 (80) = - (201) (rease i + rsine j)

20/5= -40/2 rease i - 40/2 rsine j

 $a_{\kappa} = 2\omega_{\kappa} \times \mathbb{Z}p_{\beta} = 2(-\omega_{\kappa} \frac{1}{2}) \times (-2\omega_{\kappa} \sin \omega_{\kappa} + 2\omega_{\kappa} \cos \omega_{\kappa} \frac{1}{2})$ $a_{\kappa} = +4\omega_{\kappa}^{2} \cos \omega_{\kappa} + 4\omega_{\kappa}^{2} \sin \omega_{\kappa} \frac{1}{2}$

 $Q_0 = a_{D^1} + a_{D/Q} + a_R$ $= -(R + r \cos 6) \omega_1^2 i - r \omega_1^2 \sin 6 j - 4 \omega_1^2 r \cos 6 i - 4 \omega_1^2 r \sin 6 j + 4 \omega_1^2 r \sin 6 j$ $-4 \omega_1^2 r \sin 6 j + 4 \omega_1^2 r \cos 6 i + 4 \omega_1^2 r \sin 6 j$ $a_0 = -\omega_1^2 \left[(R + r \cos 6) i - r \sin 6 j \right].$

QD = - WI (AD) QED

ALTERNATIVE SOLUTION

AT ANY TIME t:

A | 0 = w,t | 0 (20+6) 4 | = 4+6

FOR POINT D:

 $\chi = R \cos \phi + r \cos(\phi + 6) = R \cos \omega_t + r \cos(\omega_t + 6)$ $y = -R \sin \phi + r \sin(\phi + 6) = -R \sin \omega_t + r \sin(\omega_t + 6)$

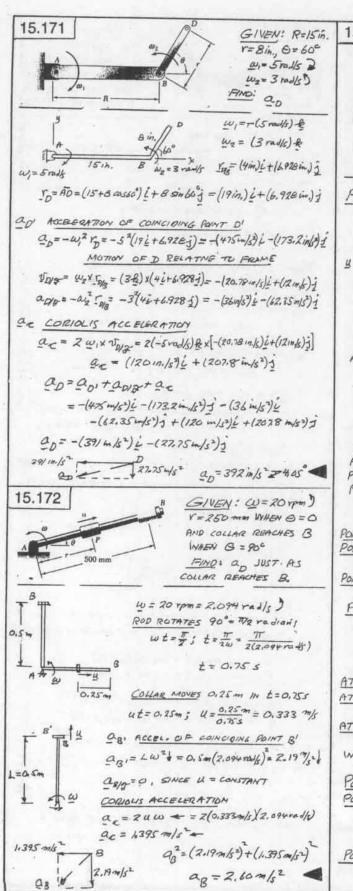
 $\dot{x} = -R\omega$, $\sin \omega_t t - r\omega$, $\sin(\omega_t t + \theta)$ $\dot{y} = -R\omega$, $\cos \omega_t t + r\omega$, $\cos(\omega_t t + \theta)$

 $\ddot{\chi} = -Ru_1^2 \cos u_1 t - ru_1^2 \cos(u_1 t + \Theta)$ $\ddot{u} = +Ru_1^2 \sin u_1 t - ru_1^2 \sin(u_1 t + \Theta)$

 $\ddot{x} = -\omega_1^2 (12\cos\omega_1 t + r\cos(\omega_1 t + 6)) = -\omega_1^2 x$ $\ddot{y} = -\omega_1^2 (-R\sin\omega_1 t + r\sin(\omega_1 t + 6)) = -\omega_1^2 y$

:. ap=-w,2(AD)

WHEN W2 = 2W, OD PASSES THROUGH POINT A DURING ENTIRE MOTION



15.173 and 15.174

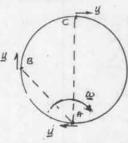


GIVEN: W= 3rad/s)

PROBLEM 15.173: FOR & =0 Find: Qp WHEN PIN IS AT (a) POINT A, (b) POINT B, (c) POINT C.

PROBLEM 15.174: SOLVE SAME PROBLEM IF Q = SVAJ/52) AS PIN IS AT POINTS A, B, + C.

PROBLEM 15.173: AB = 0.1m1+0.1m=; AC=0.2m1



ACCELERATIONS OF

COINCIONS POINTS $\alpha_{A1} = 0$ $\alpha_{B1} = -\omega^{2}(AB) = -(3)(AB)$ $= 0.9 \text{ m/s}^{2} + 0.9 \text{ m/s}^{3} \downarrow$ $\alpha_{C1} = -\omega^{2}(AC) = -(3)^{2}(AC)$ $= 1.8 \text{ m/s}^{3} \downarrow$

ROCELERATIONS OF PIN RELATINE TO THE

ROTATING FRAME = U2/r = (0.09 m/s)2/(0.1 m)= 0.0819/2

WE HAVE:

QA/g= 0.081 m/s2 +

QA/g= 0.081 m/s2 +

ac/= 0.08/m/s 4

CORIOLIS ACCELERATIONS
POINT A: & = 24W = 2(0.09 m/s \ 3 rad/s) = 0.54 m/s = 1
POINT B: SAME MAGNITUDE ac = 0.54 m/s = 1
POINT C: "

ap = ap + ap/g + ac

POINT A: Q= 0 + 0.081 m/s2 + 0.54 m/s2 + 0.621 m/s2 1 POINT B: Q= 0.9 m/s2 + 0.9 m/s2 + 0.081 m/s2 + 0.54 m/s2 = 1.767 m/s2 30.6 POINT C: Q= 1.8 m/s2 + 0.081 m/s2 + 0.57 m/s2 = 2.42 m/s2 +

PROBLEM 15.174 WE NOW ALSO HAVE & = 5 rad/s")
THIS ADDITION CHANGES ONLY THE
ACCELERATIONS OF THE COINCIDING POINT BY
HODING THE TERM &XE

AT POINT A: r=0 and dxr=0

AT POINT B: dr=d(AB)=(5rads2)AB

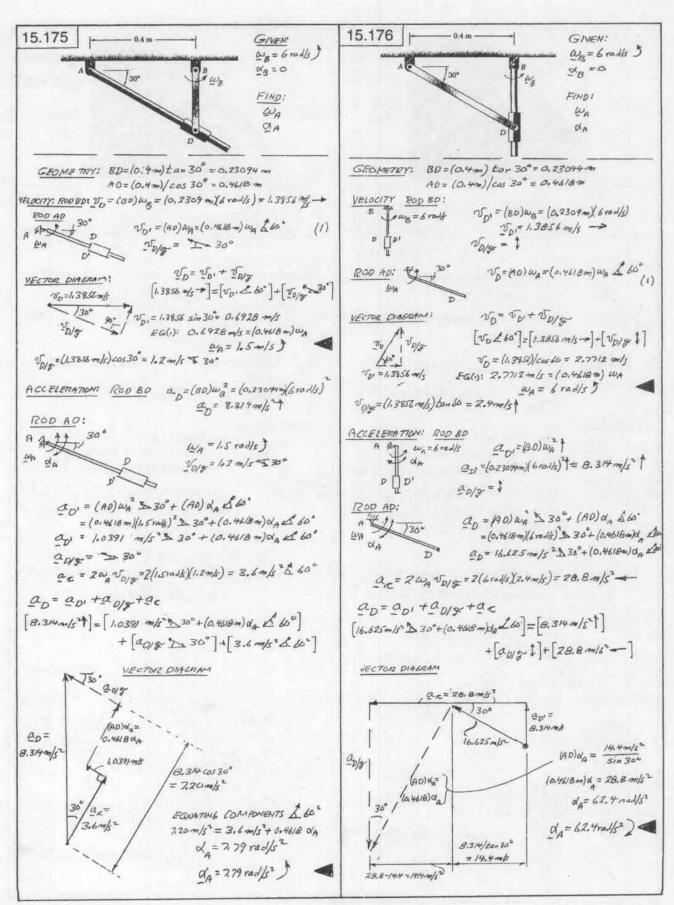
= 0.5 m/s2 + 0.5 m/s24

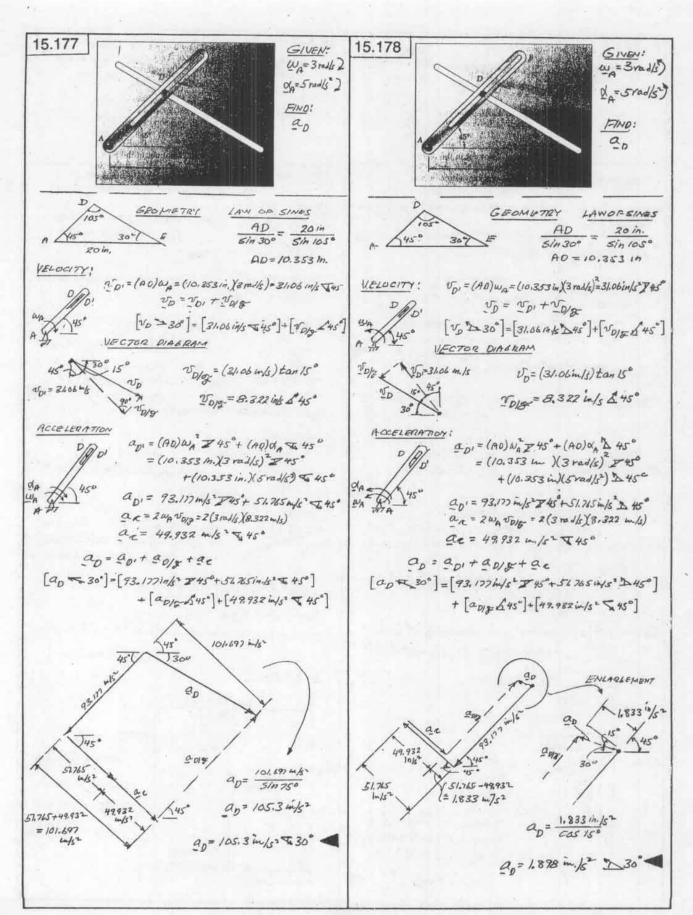
AT POINT C: dr=d(Ac)=(5rads2)(0.2n)=1 m/s2 +

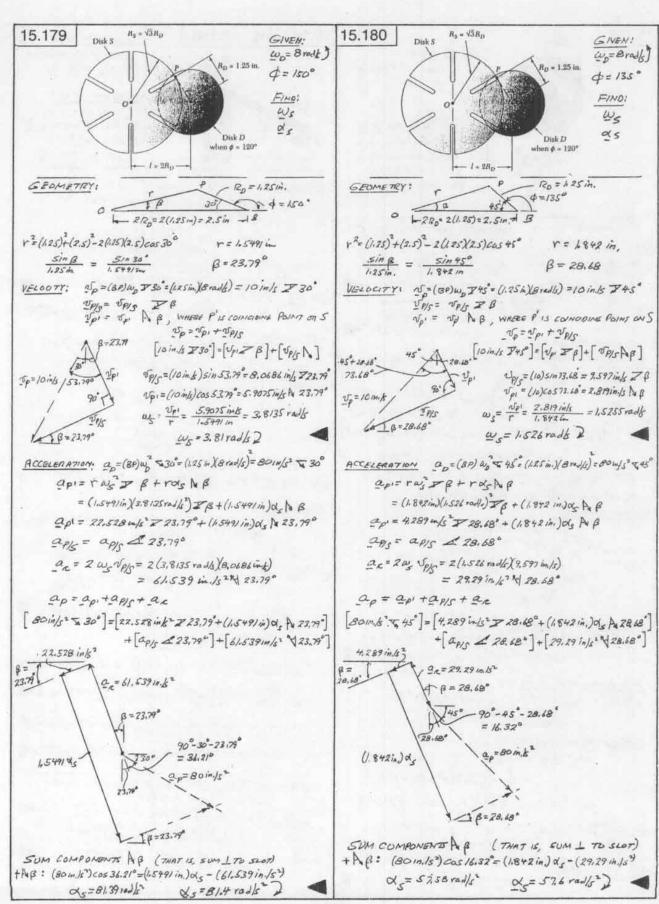
WE NOW ADD OF TO RESULTS OF PILOS, 15,173

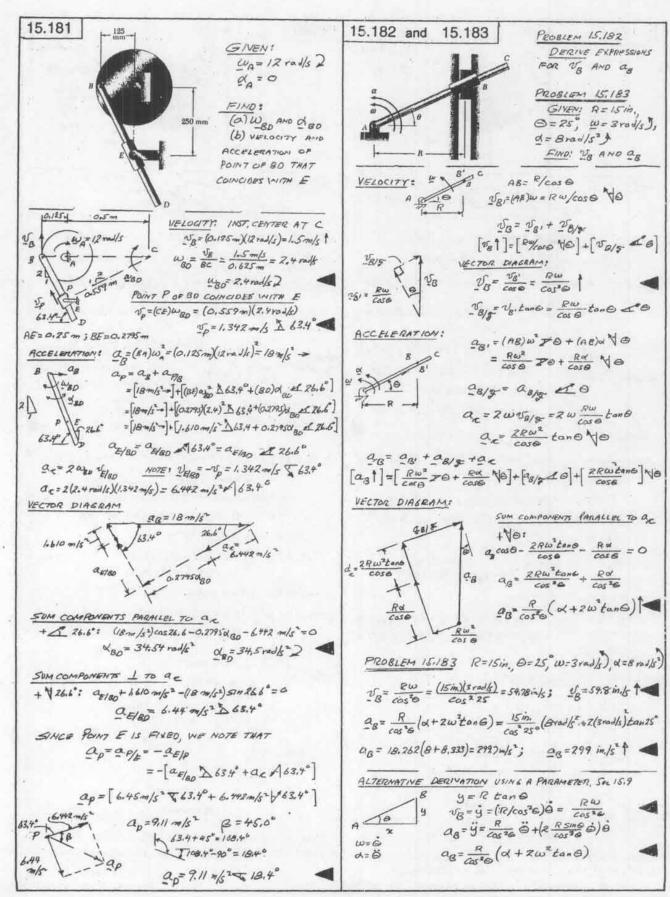
POINT A: a=0+0.621 m/s + 0.5 m/s + 1.521 m/s - +0.9 m/s + 1.521 m/s - +0.9 m/s + 1.4 m/s - +0.33 m/s = 1.021 m/s - +1.4 m/s - +0.33 m/s = 1.033 m/s = 1.733 m/s = 1.733 m/s

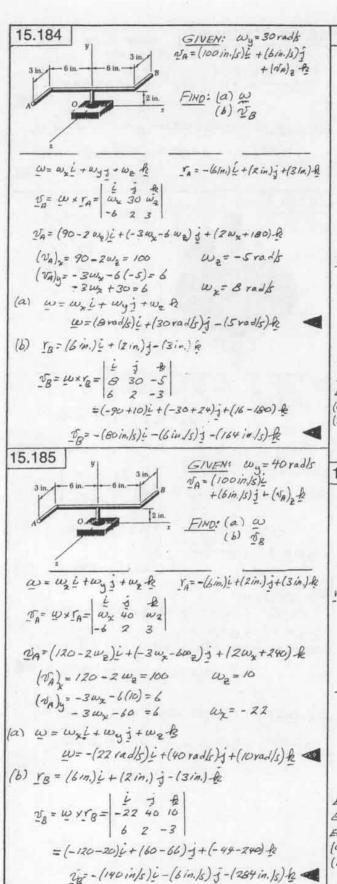
POINT C: Qc= 1 m/6 + 2.42 m/s \ Qc= 2.62 m/s Z 67.6° ◀

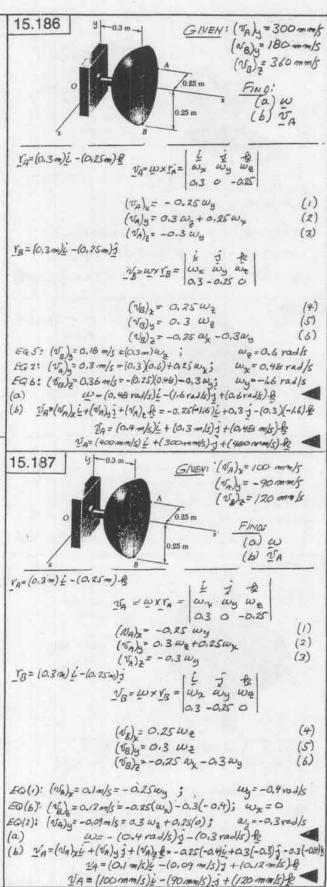


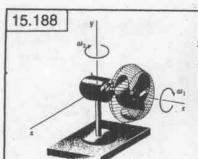












ω=-(260 rpm) i=-(1211 rad/s) i ω=-(2,5 rpm) i=-(17/12 rad/s) j ω=-(2,5 rpm) i=-(17/12 rad/s) j ω=-(2,5 rpm) i=-(17/12 rad/s) j κ=(ω+ω2)=(ω+ω2)οχη+ω2 ×(ω+ω2) κ=(ω+ω2)=(ω+ω2)οχη+ω2 ×(ω+ω2) κ=ω2×ω,=(-17/12 rad/s) i κ=-(9.8696 rad/s²) k κ=-(9.87. rad/s²) k

15.189

GIVEN: W = 1800 rpm

01 = 0

W2 = 67pm

02 = 0

FIND: FOR ROTOR OF MOTOR, &

ω= (1800 γρη) &= (60 π mad/s) &

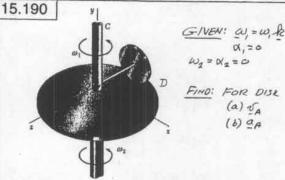
ω= (6 γρη) j = (π/s rad/s) j

ω= 120 ππτον 0= 1=12 ππε 0 χy 2

α= (ω, + ω)= (ω, +ω)σχy + ω2 χ(ω, +ω2)

α= ω2 χω= (π/s rad/s) χ (60 π rad/s) &

α= (118, 44 rad/s) ω α= (118, 4 rad/s²) ω



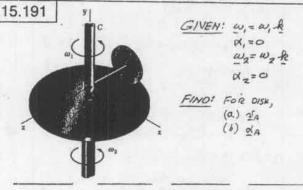
DISK A: (IN ROTATION ABOUT D)

SINCE Wy=W, WA= Wx L+W, J+ Wz-k

POINT D IS POINT OF CONTACT OF WHEEL + DISIG YOB = -r3-RA Yo = WAXYOE | i j te | Yo = WAXYOE | w, w = | O -r -R | Yo = (-Rw, + rw =) i + Rw, j - rw & R (CONTINUED)

15.190 CONTINUED SINCE $\omega_z = 0$, $\sigma_D = 0$ EACH COMPONENT OF T_D is zero $(\sigma_D)_2 = r\omega_2 = 0$; $\omega_z = 0$ $(\sigma_D)_2 = -R\omega_1 + r\omega_2 = 0$; $\omega_z = (R/r)\omega_1$ (a) $\omega_A = \omega_1 + (R/r)\omega_1 = 0$

(b) DISK A: ROTATES ABOUT & AXIS AT RATE W, $\alpha = \frac{d\omega_A}{dE} = \omega_y \times \omega = \omega_r \times (\omega_r + \frac{R}{r}\omega_r + \frac{R}{s})$ $\alpha = \frac{R}{r}\omega_r^2 \cdot \omega = \omega_r^2 \times (\omega_r + \frac{R}{r}\omega_r + \frac{R}{s})$



DISK A: (IN ROTATION ABOUT O)

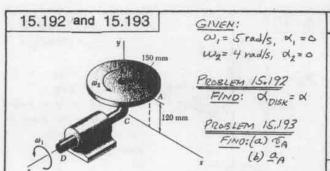
STANCE Wy = W, S + W

 $\mathcal{N}_{D} = (-R\omega_{i} + r\omega_{i})\dot{L} + R\omega_{i}\dot{I} - r\omega_{i}\dot{R} \qquad (1)$

 $\frac{DISK B: \omega_B = \omega_2 \cdot j}{\Sigma_D = \omega_B \times \Gamma_{D/D} = \omega_2 \cdot j \times (-r_2 - R_2) = -R\omega_2 \cdot i}$ (2)

FROM EQS 1 A~02: $v_0 = v_0$: $(-Rw_1 + rw_2)i + Rw_1 - rw_2 k = -Rw_2 i$ cons. of k: $-rw_2 = 0$; $w_1 = 0$; $w_2 = \frac{R}{r}(w_1 - w_2)$ cons. of i: $(-Rw_1 + rw_2) = -Rw_2$; $w_2 = \frac{R}{r}(w_1 - w_2)$ $w_1 = w_1 + \frac{R}{r}(w_1 - w_2) k$

(b) DISKA ROTATES ABOUT Y AXIS AT RATE ω , $\alpha_A = \frac{d\omega_A}{dt} = \omega_S \times \omega_A = \omega_S \int_{\mathbb{R}} \times \left[\omega_S + \frac{R}{r} (\omega_S - \omega_A) \frac{R}{R} \right]$ $\Delta_A = \frac{R}{r} \omega_S (\omega_S - \omega_A) \frac{1}{L}$



DISK: W = W, & + W, & = (4 rad/s) & + (5 rad/s) & RPROBLEM 15.192:

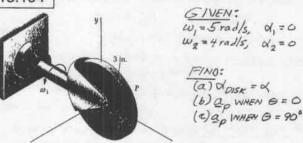
PROBLEM 15.193:

15A = - (0.6 m/s) i+(0.75 m/s) j-(0.6 m/s) A

$$Q_{A} = \times x_{A} + \omega \times y_{A}$$

$$= -20i \times (0.15i + 0.12\frac{1}{2}) + \begin{vmatrix} i & j & -4 \\ 0 & 4 & 5 \\ -0.6 & 0.75 & -0.6 \end{vmatrix}$$

15.194



 $\omega = \omega, \dot{\ell} + \omega_2 \dot{R} = (5 \text{ rad/s}) \dot{\ell} + (4 \text{ rad/s}) \dot{R}$ $\alpha = \omega, \times \dot{\omega} = (5 \text{ rad/s}) \dot{\ell} \times \left[(5 \text{ rad/s}) \dot{\ell} + (4 \text{ rad/s}) \dot{R} \right]$

υρ=ω×γρ=(52+4+)×3ί; υρ=(12 in./s)

ap= xxrp + wxrp =-20jx3++(5++++)x12j

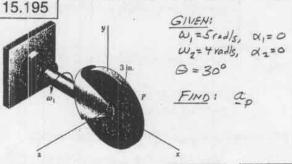
= 60R + 60 R - 48 L = -48 L +120 R

ap = - (48 in./s2) + (120 in./s2) &

(LONTINUED)

15.194 CONTINUED

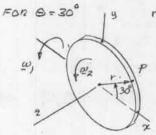
(*)
$$\underline{\Theta} = \underline{90}^{\circ}$$
: $\underline{r}_{p} = (3in)\underline{3}$
 $\underline{r}_{p} = \underline{W} \times \underline{r}_{p} = (5i + 4\underline{R}) \times 3\underline{3}$; $\underline{v}_{p} = -(12ii./s)\underline{i} + (15in./s)\underline{R}$
 $\underline{a}_{p} = \underline{A} \times \underline{r}_{p} + \underline{\omega} \times \underline{v}_{p}$
 $= -20\underline{j} \times 3\underline{j} + (5i + 4\underline{R}) \times (-12\underline{i} + 15\underline{R})$
 $= 0 = 75\underline{j} - 48\underline{j} = -123\underline{j}$
 $\underline{\alpha}_{p} = -(123in./s^{2})\underline{j}$



a= ω, i+ω2 = (Srads)i+ (4rads) &

α= ω, × ω = (5rads)i × [(5rads)i+ (4rads) &

α= -(20 rad/s²)]



r = 3 in, $(rp)_2 = r \cos 30^\circ$ $= (3 in) \cos 30^\circ = 2.598 in$, $(rp)_2 = r \sin 30^\circ$ $= (3 in) \sin 30^\circ = 1.5 in$, rp = (2.598 in) L + (1.5 in) J

= -62 + 10.3925 + 7.5-8

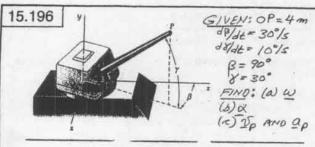
Ep = - (6 in. b) = + (10, 39% in/s) j + (7.5 in. ls) &

ap= &xrp+ wx vp

ap=-20jx(2598i+1.5j)+ 5 0 4 -6 10.892 7.5

= 51,96 & - 41,57 £+(-24-37.5) j +51,96 & ap = -41,57 £ - 61,5 j + 103,92 &

ap=-(41.6 in./52)i-(61.5 in./5) +(103.9 in./52) +



$$P = \frac{d\beta}{dt} d = -(30)/s = -(\frac{\pi}{8} \text{ rad/s}) \frac{d}{dt}$$

$$W = -\frac{d\beta}{dt} \dot{u} = -(10)/s = -(\frac{\pi}{8} \text{ rad/s}) \dot{u}$$

$$V_p = (4m) \sin 30 d + (4m) \cos 30 d R$$

$$V_p = (2m) d + (3.464m) R$$

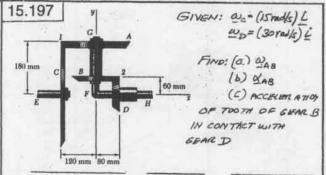
(a)
$$\omega = \omega_1 + \omega_2 = -\frac{77}{18} \dot{\underline{i}} - \frac{77}{8} \dot{\underline{j}} = -0.17453 \dot{\underline{i}} - 0.5236 \dot{\underline{j}}$$

 $\omega = -(0.1745 \text{ rad/s}) \dot{\underline{i}} - (0.574 \text{ rad/s}) \dot{\underline{j}}$

(b) NOTE TURRET ROTATES ABOUT Y AXIS AT RATE W,

$$\frac{1}{\sqrt{p}} = \frac{1}{4} = \frac{1}{\sqrt{p}} = \frac{1}{$$

ap = 0.1828 i + 0.1828 i -0.0609 j+ (-0.1056-0.9498) Az = 0.3656 i -0.0609 j -1.055+ Az



$$\Gamma_{1} = -(0.12 \text{ m}) \stackrel{i}{L} + (0.16 \text{ m}) \stackrel{i}{J}; \quad \Gamma_{2} = (0.08 \text{ m}) \stackrel{i}{L} + (0.06 \text{ m}) \stackrel{j}{J}$$

$$\underline{J}_{1} = \underbrace{W_{C} \times \Gamma_{1}} = \underbrace{15 \stackrel{i}{L} \times (-0.12 \stackrel{i}{L} + 0.16 \stackrel{j}{J})} = (2.7 \text{ m/s}) \stackrel{j}{R}$$

$$\underline{J}_{2} = \underbrace{W_{D} \times \Gamma_{2}} = \underbrace{30 \stackrel{i}{L} \times (0.08 \stackrel{i}{L} + 0.06 \stackrel{j}{J})} = (1.8 \text{ m/s}) \stackrel{j}{R}$$
(1)

T = -0.18 wei -0.12 wej + (0.18 w2 + 0.12 wy) &

From Eq. 1:
$$V_1 = (2.7 \text{ m/s}) \frac{1}{12} \times (0.18 \text{ m}_2 + 0.12 \text{ m}_3) \frac{1$$

FROM EQ 2: $V_2 = (1/8 m/s) R = (0.06 w_2 - 0.08 w_3) R$ COEFFICIENTS of R: $1.8 = 0.06 w_2 - 0.08 w_3$ (4)

SOLVING SIMULTANEOUSLY EQS I AND 2, WE FIND $W_2 = 20$ $w_3 = -7.5$

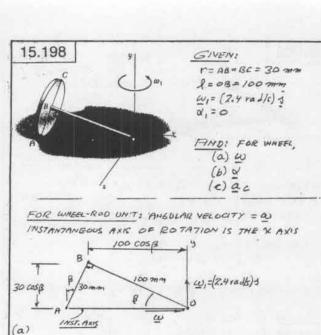
ACCELERATION GEARS AVE ROTATE ABOUT IL MILS
AT RATE WEH = (WAR) = (20 rad/s) L

(E) FOR TOOTH OF GEAR B IN CONTACT WITH GEAR D

Y2 = (0.08m): + (0.06m) = . 2 = (1.8 m/s) &

$$\begin{aligned} & Q_2 = Q_{18} \times Y_2 + w_{18} \times \sqrt{2} \\ & = (-150 \, \text{Å}) \times (0.08 \, \hat{\textbf{L}} + 0.06 \, \hat{\textbf{L}}) + (20 \, \hat{\textbf{L}} - 7.5 \, \hat{\textbf{L}}) \times (18 \, \frac{1}{4}) \\ & = -12 \, \hat{\textbf{J}} + 9 \, \hat{\textbf{L}} - 36 \, \hat{\textbf{J}} - 13.5 \, \hat{\textbf{L}} \\ & Q_2 = -4.5 \, \hat{\textbf{L}} - 48 \, \hat{\textbf{J}} \end{aligned}$$

a=-(4.5 m/s2) =- (48 m/s3) j



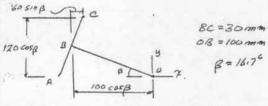
 $t'an \beta = \frac{30}{140}j$ $\beta = 16.7^{\circ}$ CONSIDE MOTION ABOUT YAXIS: $V_{\beta} = (100 \cos \beta)(2.4)$ CONSIDER MOTION ABOUT HAT.AIS: $V_{\beta} = (30\cos \beta)\omega$ $V_{\beta} = V_{\beta};$ $(100 \cos \beta)(2.4) = (30\cos \beta)\omega$ $\omega = \frac{100}{30}(2.4)$ $\omega = (8 \cos \delta/5)\dot{L}$

(b) x= w, xw = (2.4 rad/s) x (8 rad/s) &

Q=-(19.2 rad 62) &

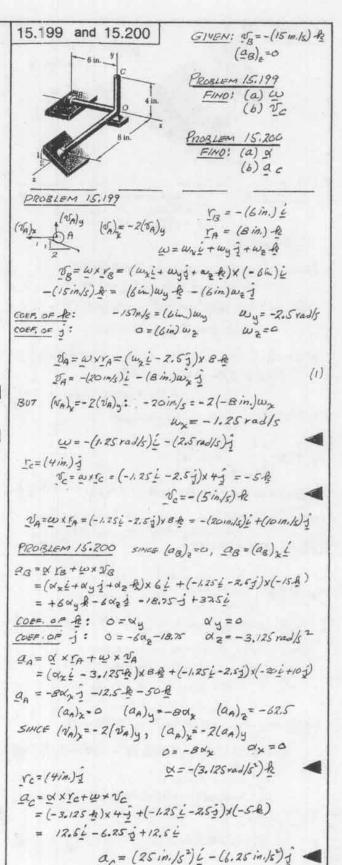
(x) POINT C: CA = 2r = 0.06 m - Sq = (0.06 m) (- sing i + cosp i)

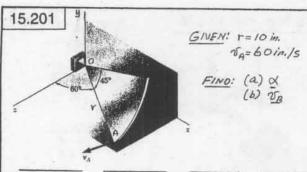
V = W× (CH = (Bradk) i x (0.06m) (-sing i + cosβ j) = 0.48 cosβ & = 0.48 cos 16.7 & V = (0.4598 m/s) &



$$\begin{split} & \Gamma_{C} = -\left(100\cos\beta - 30\sin\beta\right)\dot{i} + (160\cos\beta)\dot{j} \\ & = -\left(95.782 - 8.621\right)\dot{i} + 57.47\dot{j} \\ & \Gamma_{C} = -\left(87.16\,\text{mm}\right)\dot{i} + \left(57.47\,\text{mm}\right)\dot{j} \\ & \Gamma_{C} = \left(0.08716\,\text{m}\right)\dot{i} + \left(0.05742\,\text{m}\right)\dot{j} \end{split}$$

 $\alpha_{c} = \times \times \underline{r}_{c} + \omega \times \underline{r}_{c}$ = -(19.2) + 1(-0.0876 + 0.08747 + 0.4898 + 0





FIND OB: $\overline{OA} = (10in) \sin 60^{\circ} \dot{c} + (10in) \cos 60^{\circ} \dot{c}$ $r_{A} = \overline{OA} = (8.6603 in.) \dot{c} + (5in.) \dot{c}$ $\overline{OB} = (r_{B})_{2} \dot{c} + (r_{B})_{3} \dot{c}$

SCALME PRODUCTS

 $\begin{array}{l}
\overline{OA} \circ \overline{OB} = (OAXOB) \sin 45^{\circ} \\
(B.6603 \underline{i} + 5 - \underline{A}) \circ ((r_{B})_{2} \underline{i} + (r_{B})_{4} \underline{i}) = (\iota o X \iota o) \sin 45^{\circ} \\
B.6603 (r_{B})_{2} = 70.711 \quad (r_{B})_{2} = 8.165^{\circ} \text{ in.} \\
(r_{B})_{3} = \overline{OB}^{2} - (r_{B})_{2}^{2} = 10^{2} - B.165^{\circ}; \quad (r_{B})_{3} = 5.773^{\circ} \text{ in.} \\
\underline{r_{B}} = (B.165^{\circ} \underline{in.}) \underline{i} + (5.773^{\circ} \underline{in.}) \underline{i}
\end{array}$

SINCE Of 1 OA, NA FORMS 30" ANGLE WITH 2 AXIS SA=(60 in/s)(-SIN30" + COS 60")

54= - (30 in/s) + (51.96 in/s) A

PLATE OAR: W= wxi+wyj+wzfe

NA= wxx = wx wy we 8.6603 0 5

-30 i +51.96 1 = 5 wy + (8.6603 wg -5 wg) - 8.6603 wy &

COFF 6: 0 = 8.6603 W2 -5 W2 = -6 rad/s

COFF 6: 0 = 8.6603 W2 -5 W2 = W2 = 0.57735 W2 (

COFF A: 57.96 = -8.6603 WY -> WY = -6 rad/s

UB = -5.773 Wzi + 8.165 Wz = + (5.773 Wz - 8.165 Wy) A

SINCE POINT B MOVES IN TY PLANE $(V_B)_2 = 0 = 5.773 \omega_2 - 8.165 \omega_4$

0 = 5.713 Wx -8.165(-6)

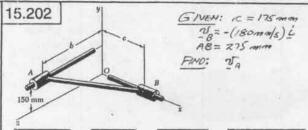
wx = -8.486 rad/s

EG(1): W2 = 0.57735(-8.486) = -4.899 rad/s

w=-(8,49 rad/s)i-(6 rad/s)j-(4,90 rad/s)k

 $(U_B)_{\chi} = -5.773(-4.899) = 28.3 \text{ m/s}$ $(V_B)_{\psi} = 8.165(-4.899) = -40.0 \text{ m/s}$ $(V_B)_{\phi} = 0$

VB= (28.3 in./s) i-(40.0 in./s) j



 $K = 175 \text{ mm}; 225 = 150^{2} + 175^{2} + b^{2}; b = 150 \text{ mm}$ $S_{B} = -(180 \text{ mm/s})^{\frac{1}{2}}; V_{A} = V_{A} + \frac{1}{2}; V_{A} = -175^{\frac{1}{2}} + 150^{\frac{1}{2}} + 150^{\frac{1}{2}} + 150^{\frac{1}{2}} + 150^{\frac{1}{2}} + 150^{\frac{1}{2}}$ $V_{A} = V_{B} + V_{A/B} = v_{B} + \omega \times v_{A/B}$ $V_{A} = -180^{\frac{1}{2}} + v_{A/B} = v_{A} + v_{A/B}$ $V_{A} = -180^{\frac{1}{2}} + v_{A/B} = v_{A/B} + v_{A/B}$

VAR = -1801 + (1500, -1500) + (-1750, -1500) + (1500, +17500) &

COEF OF i: $+180 = +160 w_y - 150 w_z$ (1) COEF. OF j: $0 = -150 w_z$ $-175 w_z$ (2)

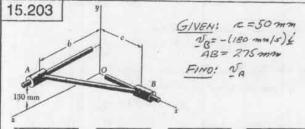
 $\frac{\cos R}{\cos R} = \frac{1}{2} \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2}$ $\frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2}$ $\frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2}$ $\frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2}$ $\frac{\cos R}{2} + \frac{1}{2} \frac{\cos R}{2} + \frac{1}{2} \frac{$

(FO2+FC3) = 6 6 6 1/7 = 0 + 150 my - 150 mg = 150 mg/5 & = 150 mg/5 &

FOR USE IN PEOS IS. 214; WE CALCULATE A POSSIBLE W.

INE SHALL ASSUME Wx = 0. From EQ(2), WE HAVE Wz=0.

EQ(1): 180 = 0 +150 wy w= +(1.2 rad/s)j



C= 50 mm; 2752=1502+502+62; b= 225 mm MB=-(180 mm/s) i; NA=-NA+6; YA/B--501+1505+225-B NA=-NB+-5A/B=-NB+-WX-1/18

NA 1/2 = -1801 + (225 mg -150 mg) (+(-50 mg -275 mg) f +(150 mg + 50 mg) f

 $\frac{\text{COSF OF } i:}{225\omega_{3} - 150\omega_{2}} = \frac{1}{225\omega_{3} - 150\omega_{2}} = \frac{1}{225\omega_{3}} = \frac{1$

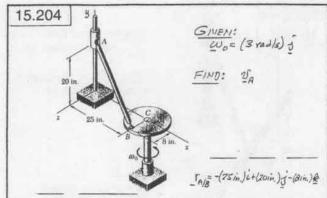
[EQ.1 +3xEQ.2 - 4.5EQ.3]: 180-4550=0 VA = +(40mm/s) k

FOR USE IN PROB. 15.215:

INDETERMINATE, WE CAN ASSUME A VALUE FOR ANY COMPONENT OF US. WE ASSUME WE'D,

EQ(1): +180 = 225 mg

w=(0,8 rad/s)5



DISK: UB = WO Y [B/c = (3ravis)] x (8in.) & = (24 in./s) i NA=NB+ JAB = JB + WX YBIC VAj= (24/n/s) + wx wy we -25 +20 -8

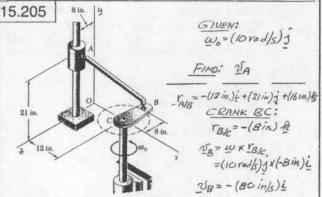
1/45 = 24i+(-8my-20m)i+(-25m2+8m2)j+(20m2+25mg)&

COSE OF 5:
$$-24 = -8\omega_y - 20\omega_z$$
 (1)
 $-25\omega_z$ (2)

COEF OF R:
$$O = 20w_x + 25w_y$$
 (3)

SINCE DETERMINANT OF WX, WY, WZ IS ZERO, W IS INDETERMINATE. WE CAN ASSUME ANY ONE COMPONENT ASSUME Wy=0, EQ. 3 NEW Wy=0.

Fax:
$$V_{A} = 0 - 25(1/2) = -30$$
; $V_{A} = -(30 \text{ m/s}) \hat{S}$



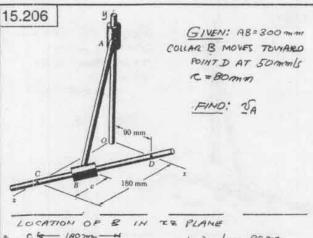
ROD AB: VA = VB + VAIR = VB + W X TELE Voj = - 801 + wx wy we

$$\frac{(OEF, OE \ \ : \ -80 = 16\omega_{1} - 21\omega_{2}}{(OEF, OF \ \ : \)^{2}} = -16\omega_{1} - 12\omega_{2}$$
(1)

0 = 21 Wx + 12 my COST, OF A:

SINCE DETERMINIANT OF WA, WY, WZ IS ZERO, THE ANGULAR VELOCITY IS INDETERMINATE, WE CAN ASSUME VALUE OF ANY ONE COMPONENT.

ASSUME WY = 0. EQ. 2 YIELDS WY = 6 EQ,(1): -80 = 0-2/ wz; wz=-80 rad/s EQ.(2): VA=0-12(-9); 2A = (45.7 in./s) 5 €



(rB)=== R= 80 mm

1200 AB = 300 mm (300mm)2=(100mm)+(40mm)+1/6) ra= (280mm)+ 15 = 50 mm/s 7c= 1-(-2++4)

UB= 208 = 50 (-24+i) = - (44,72 mm/s) & + (22,36 mm/s) L 1/4/3 = - (40mm) + (280mm) 1 - (100mm) h

Vaj = -4472-2+22,36i+(-100 my -280 mg)i +(-40 m2+100 m2) 1+(280 m2+40 my) &

COSE, OF L:
$$-22.36 = -100 \omega_y - 280 \omega_z$$
 (1)
COSE, OF L: $V_A = 100 \omega_z$ $-40 \omega_z$ (2)

$$\frac{\cos F, \ oF \ j}{\cos F \ oF \ p}; \qquad V_A = 100 \ w_k \qquad -40 \ w_e \qquad (2)$$

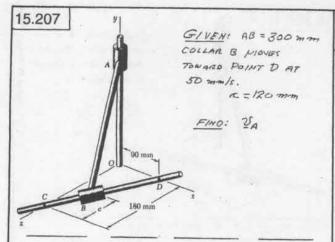
$$\cos F \ oF \ p; \qquad +4472 = 280 \ w_k + 40 \ w_y \qquad (3)$$

SINCE DETERMINANT OF WE WY, WE IS ZEER, THE ANGULAR VELOCITY IS INDETER MINANT, WE CAN ASSUME WALUE OF ANY COMPONENT.

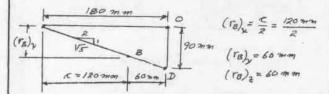
ASSUME W. =0 my=1.118 rad/s EQ. 3: 44,72 = 0 + 40 404; - 22,36 = -100(1,118) - 280 Wa W==-0,3194 rad/s

W= (1.118 rad/s) g-(0.3194 rad/s) &

Eq. 2: Va= 0-40(-0.3194) = 12.777 mm/s Va= (12.78 mm/s) 5



LOCATION OF B IN THE PLANE



ROD AB = 300 mm (300 mm) 2 (60 mm) 2 (60 mm) 2 (7a) 2 $\Gamma_{A} = 28% \times mm$

TA/B = - (60 mm) + (282.75 mm) j - (60 mm) &



VELOCITY OF B: $V_B = SO mm/s$ $A_{cD} = \frac{1}{V_S} (-2 - \frac{1}{K} + \frac{1}{L})$

 $V_B = V_B \frac{\partial}{\partial z} = \frac{50}{V_{\overline{S}}} (\dot{z} - 24) = +(22.36 \text{ mm/s})\dot{z} - (44.72 \text{ mm/s}) \frac{4}{5}$

NAj = 22.36i -44.72 ft + (-60 mg - 207.75 mg) i + (-60 mz + 60 mg) j + (267.75 mg + 60 mg) ft

 $\begin{array}{lll} COEF, OF \dot{L}: & -22.36 = & -60 \, \omega_{y} - 282.75 \, \omega_{z} & [1] \\ COEF, OF \dot{j}: & V_{A} = 60 \, \omega_{z} & -60 \, \omega_{z} & (2) \\ \hline COEF, OF \dot{Q}: & 4472 = 282.75 \, \omega_{z} + 60 \, \omega_{y} & (3) \end{array}$

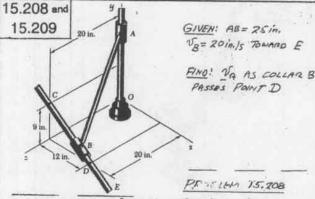
SINCE DETERMINANT OF WILL WE IS ZERO, THE ANGULAR VELOCITY IS INDETERMINANT. WE CAN THUS ASSUME THE VALUE OF ANY COMPONENT.

ASSUME $w_{\chi=0}$: Eq.3: $44.72 = 0 + 60 \omega_{y}$; $w_{\chi} = 0.7463 \text{ rad/s}$ Eq.1: $-22.36 = -60(0.7483) - 287.75 \omega_{2}$

w=-0.0777 rads

ω = (0.7453 rad/s) i - (0.017) rad/s) is EG. 2: NA = O - 60(-0.077) = 4.66

NA= (4,66 mm/s) 5



COLLAR AT D: AB2 = 252 = 122+202+ ra; ra=9in. ra/8 = -(12in) + (9in) - (20in) +2

(OFF. OF i: -16 = -20Wy -9We (1)

 $\cos F$, or \hat{J} : $V_A + 12 = 20 \omega_R$ $-12 \omega_R$ (2) $\cos F$, or \hat{R} : $O = 9 \omega_R + 12 \omega_R$ (3)

SINCE DETERMINANT OF WILLIAMS, WE IS ZERO, THE ANSULAR VELOCITY IS INDETERMINATE, WE CAN THUS ASSUME THE VALUE OF ANY COMPONENT

ASSUME $w_1 = 0$, EQ 3, YIFLOS $w_2 = 0$ EQ. 1: $-16 = 0 - 9w_2$ $w_2 = \frac{16}{9} \text{ rad/s}$ $\frac{60.2}{5}$ $v_A + 12 = 0 - 12(\frac{16}{9})$ $v_A + 12 = -21.33$ $v_A = -(33.3 \text{ in/s})$

PROBLEM 15.209

COLLAR AT C: AB = 25 = 20 + FA; FA = 15in.

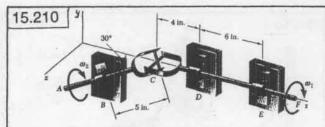
SA/8 = (15in) j - (20 in) - R

VB = (16 in/s) i - (12 in/s) j

VA = VB + VA/0 = VB + WX SAB VA j = 16i-12j + WX WY WA 0 15-20

EQ. 3: $\omega_{x} = 0$ EQ. 2: $v_{A} + 12 = 0$ $v_{A} = -12 \text{ m/s}$ $v_{A} = -(12 \text{ in/s})\sqrt{3}$

MOTE: W IS INDETERMINATE. ANY VALUE CAN BE CHOSEN FOR EITHER WY ORWZ
FOR EXAMPLE, IF WY=0, THEN
EQ. 1: 16=-20WY WY=-0.8 rods



WHEN ARM OF CROSSPIECE ATTACHED TO SHAFT CF IS HORIZONTAL, FIND CU2 OF CHAFT AC.

PLACE ORIGIN AT CENTER OF CROSSPIECE AND DENOTE BY & THE LENGTH OF EACH ARM.

$$r_{H} = -A \sin 30i + l\cos 20i$$

$$w_{1} = -\omega_{1}i$$

$$w_{2} = -\omega_{2}\cos 30i - \omega_{2}\sin 30i$$

SMAFT CF: $v_{h} = \omega_{1} \times I_{h} = -\omega_{1} \cdot I_{h} \times I_{h} = -\omega_{1} \cdot I_{h} = \omega_{2} \times I_{h}$ $= (-\omega_{2} \cos 30^{\circ} i - \omega_{2} \sin 30^{\circ}) \times (-1 \sin 30^{\circ} i + 1 \cos 30^{\circ})$ $v_{h} = -1 \cos 2 \cos^{2} 30^{\circ} + 2 \cos^{2} 30^{\circ} k = -1 \cos^{2} k$ $v_{h} = -1 \cos 2 \cos^{2} 30^{\circ} + \sin 30^{\circ}) \cdot k = -1 \cos^{2} k = -1 \cos$

 $\frac{ceosspiece}{2\delta_{\theta}} = \omega_{x} \underbrace{i + \omega_{y}}_{i} \underbrace{i + \omega_{z}}_{j} \underbrace{k}_{i} \times \ell \underbrace{k}_{i}$ $2\delta_{\theta} = \omega_{x} \underbrace{v}_{\theta} = (a_{y} \underbrace{i + \omega_{y}}_{j} \underbrace{i + \omega_{z}}_{j} \underbrace{k}_{i}) \times \ell \underbrace{k}_{i}$ $\delta_{\theta} = -\ell \omega_{x} \underbrace{j + \ell \omega_{y}}_{i} \underbrace{i}_{i} \qquad (3)$

 $EQ 1 = EQ 3: V_8 = V_8$ $lw_i j = -lw_x j + lw_y i$ $coee. y i: lw_i = -lw_x$ $w_y = 0$ $w_y = 0$ (5)

VH=WXYH= WX WY WE -LSM30 LCOS30 0

> = -lw_cos30° i -lw2sin30° j +(lw2cos30°+lw3sin30°) fr

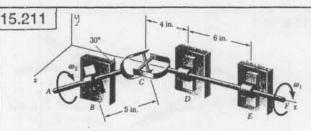
SUBSTITUTE FROM EGS. 4 AMD 5: We AND Wy = 0 TH= - lwa cos 30 i - lwa sin 30 j - lu, cos 30 A

SUBSTITUE FOR VIN FROM EQ. 2.

- lw2 = - lw2cos 36 i - lw2 sin30 1 - lw, cos 36 &

COEF OF B: J=- Lwz sin36 W2=0

ωz=ω, cos30°



VIHEN ARM OF CROSSPIECE ATTACHED TO SHAFT CF

PLACE ORIGIN AT CENTUR OF CROSSPICE AND DENOTE BY & THE LENGTH OF EACH ARM.

SHIRFT AC: $U_H = \omega_2 \times Y_H$ = $(-l \omega_2 \cos 3\omega \dot{t} - l \omega_2 \sin \omega^2) \times -l \cdot \dot{x}$

2/2 - lug cos 3 + lug sm 20 } (5)

 $\frac{Ci.(CSTINE: \quad \omega = \omega_x \dot{u} + \omega_y \dot{j} + \omega_z - \dot{k}}{\sqrt{\varepsilon} = (\omega_x \dot{i} + \omega_y \dot{j} + \omega_z - \dot{k}) \times \dot{k} \dot{j}}$ $V_{\varepsilon} = l \omega_x \dot{k} - l \omega_z \dot{u} \qquad (2)$

 $S_{\mu} = \omega \times Y_{\mu} = (\omega_{\chi} \dot{z} + \omega_{\chi} \dot{j} + \omega_{\chi} \dot{f}) \times (-l \dot{f})$ $S_{\mu} = \ell \omega_{\chi} \dot{j} - \ell \omega_{\chi} \dot{f}$ (4)

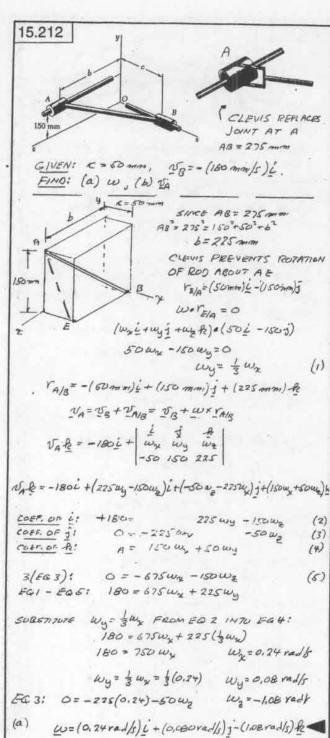
Eq. (1 = F0.3: $\sqrt{8} = \sqrt{8}$ $-lw_1 k = lw_2 k - lw_2 i$ $\sqrt{8} = 2 w + lw_2 = 2 w$. (4)

 $\frac{\cos F, \circ F}{\cos F, \circ F} \stackrel{\text{L}}{=} 1$ $\frac{\cos F, \circ F}{\cos F} \stackrel{\text{L}}{=} 1$ $\frac{\cos F, \circ F}{\cos F} \stackrel{\text{L}}{=} 1$ $\frac{\cos F, \circ F}{\cos F} \stackrel{\text{L}}{=} 1$

FOZ = FO.4: $U_H = U_H$ - $lu_2 \cos 30\frac{1}{2} + lu_2 \sin 30\frac{1}{2} = lu_2\frac{1}{2} - lu_3\frac{1}{2}$

FROM EQ.5: $\omega_{\chi} = -\omega_1$ THUS, $\omega_2 = -\frac{(-\omega_1)}{\cos 3\omega_1}$

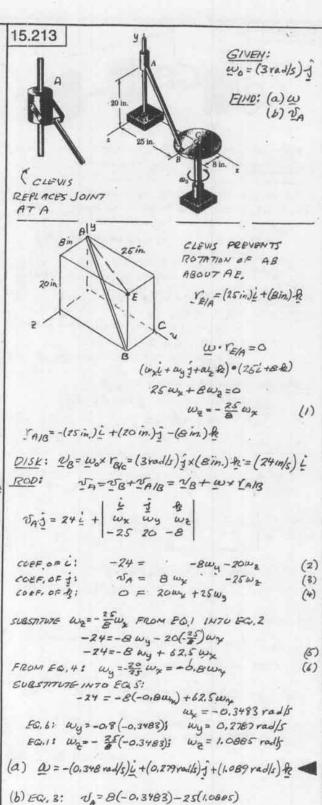
W2= cos 300



(b) EQ 4: VA= 150(0.24) +50(0.08)

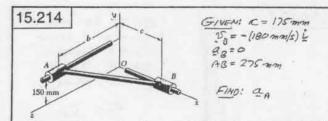
= 36 +4

S= (40 mm/s)-k



=-2.79 -27.21 =-30

VA=- (30 in./s) 5



FROM SOLUTION OF PROB. 15.202 INE RECALL

b = 150 mm; YA/B = - (15 mm) L + (150 mm) f + (150 mm) R

w = + 0.2 rad/s) f

WE NOW CALCULATE!

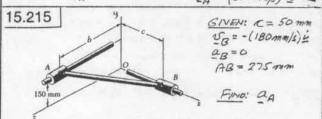
and = (150 ay -150 dz) + (-1750z-150 dz) + (1500x+1750y) & +252 i -216 d

$$\frac{\text{COSE OF LI}}{\text{COSE OF JI}} = -252 = 150 \text{My} - 150 \text{Mz}$$
 (1)
$$0 = -150 \text{Mz} - 175 \text{Mz}$$
 (2)

$$\frac{coef of i}{coef of k} : 0 = -150 \alpha_{\chi} - -175 \alpha_{\chi}$$
 (2)

M IS INDETERMINANT: ASSUME 0,=0, FROM £02, 0,=0
THEN £0.1, YIELDS: -252 = 150 dy; dy =-1.68

£0.3: $a_A + 216 = 0 + 175(-1.68) = -294$ $a_A = -216 - 294 = -510$ $a_A = -(510 \text{ may}) + 2$



FROM SOLUTION OF PROB 15.203, WE RECALL: b = 225 mm; YAIB = - (50 mm) L+ (150 mm) j+(275 mm) & w= (0.8 rad/s) j

VAIB = WXYAIB = O.E. j X (-50 i +150 j + 225 - 16)

QA=QB+QAB=QB+XXXAB+WXVAB

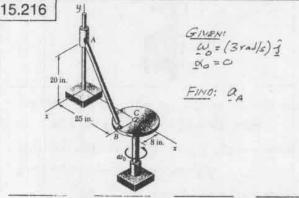
ante = (2250 y-1500) + (-5002-22502) + (15002+5004) + 32i-144 12

(CONTINUED)

15.215 CONTINUED

$$\frac{\text{COFF, OF } \dot{g}: -32 = 225 \text{ My} -150 \text{ M}_{2}}{\text{COFF, OF } \dot{g}: 0 = -225 \text{ My}} -50 \text{ M}_{2}$$
(1)

$$\begin{array}{ccc} Ea.3: & a_A + 144 = 0 + 50(^{-4}/225) \\ a_A = -144 - 7.111 & a_A = -(151.1 \text{ sma}/5^2) & \\ \end{array}$$



FROM PROB 15. 204, WE RECALL: $U_B = (24in, k) \stackrel{!}{\stackrel{!}{\stackrel{!}{=}}} I_{A/B} = -(25in) \stackrel{!}{\stackrel{!}{=}} + (20in) \stackrel{!}{\stackrel{!}{=}} - (8in) \stackrel{!}{\stackrel{!}{=}} : U_B = (127in) \stackrel{!}{\stackrel{!}{=}} : U_B = (127in) \stackrel{!}{\stackrel{!}{=}} : U_B = (127in) \stackrel{!}{\stackrel{!}{=}} : U_B = U_B = U_B = (127in) \stackrel{!}{\stackrel{!}{=}} : U_B = U_B = U_B = U_B = (127in) \stackrel{!}{\stackrel{!}{=}} : U_B = U_B$

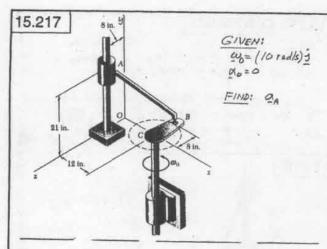
2/2=+72++(-80y-200)+(-2502+80x)+(200x+2504)+ + 366-28,85

$$coep, of P;$$
 $a_A + 28.8 = 8a_x - 25a_z$ (2)
 $coep, of P;$ $72 = 20a_x + 25a_y$ (3)

d 15 INDETERMINATE: ASSUME Xy = 0: Eq. 3: 72 = 2002+0 X2= 3.6

$$PO.2: \quad O_A + 28.8 = 8(3.6) - 25(1.8)$$

$$Q_B = -28.8 + 28.8 - 45$$



FROM SOLUTION OF PROB. 15.205: W= -(80 rad/s) & IA/8 = - (12 in) + (21 in.) + (16 in.) &

WE NOW CALCULATE: U= WOXYBE=10jx-8h=-(80 in./s) L a = wox 1/8 = 10 jx - 80 i = (800 in/5) &

VAL= WXYAL= - = + (-121+213+16-16) UA/8 = (45.714 in/s) - (80 in./s) =

an= aB + XXIAIB + WX SAIB

and = 800 12 + (16dy - 21de) i+ (-12dz - 16dz) 4 +(210,+1204) \$ +174.151+304.765

COEF, OF US -17415= 1604-2102 (1) COEF, OF 1: a-30+16=-160x -12012 (2)

COEF, OF A: -800 = 21 dx +12dy

& IS INDETERMINANT:

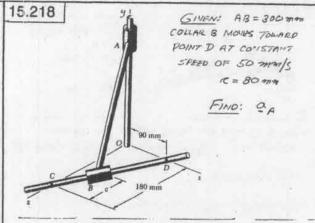
ASSUME dy =0 -174,15 = 0 -21 dz Eq. 1:

d2=8.293

EQ.3: -800 = 210x+0 dx=-38,095

EQ, 2) an - 304.76 = -16(-38,095) -12(8,293) an - 304,76 = 609,53 - 99,50

a= (815 in./52)j



FROM PROB. 15. 206 WE RECALL : 15 = + (22.36 mm/s) i - (44.72 mm/s) fe W = (1.118 rad/s) j - (0.3/94 rad/s) & TAIR = - (40 mm; + (200 mm); - (100 mm; fe

NA/B= WX M/B = 0 1.118 -0.3/7 = (-111.8+89.432) L -40 280 -100 +12.7% + 44.72 &

2/AIR= - (22,37 mm/s) + (12,776 mm/s) + (44,72 mm/s)-R

an = aB + dAB XYAB + WX JAB

anj= 0+ dx dy dz + 0 1.118 -0.3194 -40 280 -100 + 27.37 12.7% 44.72

and = (-100 dy - 280 de) i + (-40 de + 100 dx) j + (2800x+400y) &+ (50.0+4.08) i+7.15 i+25-8

COEF. OF 4: -54,08 = -100dy - 280de (1)

COEF OF 5: 0A - 7.15 = 1000/2 - 40 de (2) coef. of &1 - 25 = 2800 + 4004 (3)

& IS INDETERMINANT; ASSUME QUEO

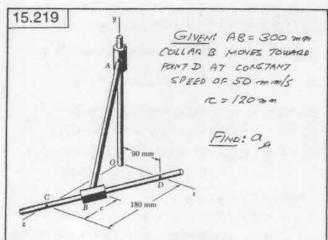
Eq. 1: -54,08 = 0 - 280 ag Q2=0.19314 FQ3! -25= 28002+0 N2 = -0,0893

0 = - (0.0893 rad/52) + (0,19314 rad/52) &

QA - 7.15 = 100(-0.0893) - 40(0.19314) FG, 2: a-7,15 = -8,93 - 7,73

ap= -951

Q= - (9.51ma/52) 1



FROM PROB. 15.207 INE RECALL! 0/ = + (22,36 mm/s) = - (44,72 mm/s) & w = (0.7453 rad/s) j - (0.0777 rad/s) & TAIB= - (60 mm) L+ (287,75 mm, j-(60 mm) R 2/A/B = WXYA/B = 0 0.7453 -0.0777 -60

= (44.718 +22.358) E + 4.662 j + 44.718. R SAIB =- (22,36 mays) & + (4,662 mays) & + (44,718 mays) & a8=0

an = aB + dx raig + wx Jale anj=0+ 4 x

anj = (-600, -287,750,)+(-600, +600,) j+(287,750, +600,) } + (33,33 +0,363) +1,737 + 16.66-

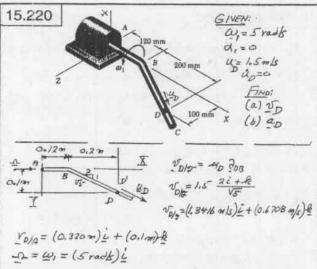
-33,69 = -60dy -287,750/2 COEF, OF L: COEF. OF J; ay-1.737 = 60 0, - 60 X2 (2) -16.66 = 287,750g +600g COGF, OF -RZ:

& IS INDETERMINANT! ASSUME Xy=0 02 = a//7/ EQ.1: -33,69=0-287,75d2: EQ.3: -16.66 = 287.75 xx +0 5 du=-00579

d=-(0.0519 radk) i+(0.1171 rads)-

ay - 1,737 = 60(-0.057) - 60(0.1171) EQ. 2: ay -1.737 = -3.474 - 7.026 ay =-8,76

ay=-(8.76 mm/s2)5



(a) VELOCITY OF D: 1/2=11xrpy= 5ix(0.32i+0.1-12)

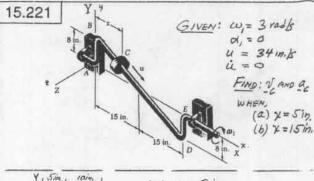
To1 = - (0.5 m/s) 5

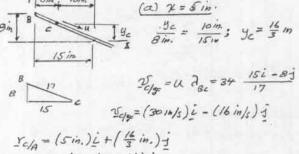
20 = 201+ 2018 = -0.5\$ +1.3416 i +0.628 12 Tp= (1.342 m/s)i-(0.5 m/s)j+(0.67/m/s)&

(b) ACCELLERATION OF D: apr =0; il =0 an=ixro/A+ ixx xxro/A = 0+ -1×00 = 5i × (-0.5j) = - (2.5 m/s2)-R

ar = 2 1 x 50/2 = 2(5 i)x(1.3+16 i+0.6708 1/2)=-(6.7)~1/3) ap=ap+app+ac

=-2.5.4+0-6.715 a=-(6.71m/s)j-(2.5m/s) &

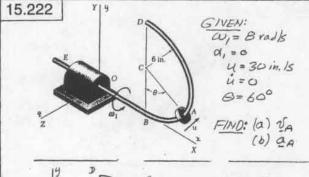


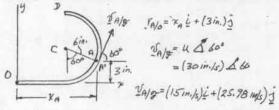


1-= w, i =- (3 rad/s) i

(CONTINUED)

15.221 continued 4=51m. VELOCITY 20, = -1 x re/A = (-3 ro.45) & x [(5 in.) i + (16 in.)] = - (16 in.15) & Ve=Ve+ Ve/g= -(6in/s) &+ (30in/s) =- (16in/s) j Vc= (30 in. ls) i - (16 in/s) = - (16 in/s) & ACCELERATION: 00/8=0; in=0 OC = Ux LOW + Ux Ux LOW = Ux LOW + Ux X CI ac1 = 0 + (-3 rad/s) [x (+16 m./s) & = -(48 m/s2)] ac= 2-0-x 26/5= 2(-3 rad/s) + x [(30 m/s) = -(16 m/s)] a= (96 in./52) & ac= ac+ ac/g+ ac = - (48 in./s2) 1 + 0 + (96 in/s2)-R a=-(48 in 152) j +(96 in/52) A (b) FOR, X = 15 in, (COLLAR C IS IN XE PLANE) VELOCITY: FROM PART a: Voja = (30 m/s) L-(16 m/s) } TOJA (15in.) L; VCI = 1 x fola = - 3 & x 16 £ = 0 T= 20+8=15=0+849= ; V= (30 in. ls)i-(16 in/s) -ACCELERATION: ac/g=0; i=0 QC = 1-x rc/4+ 1-x ve = 0+0; ax IS SAME AS IN PART a: 9x=(96in./s2) & 2c= aci+ac/g+a=0+0+ac; ac=(961n/s2) & 15.222 GIVEN: W,= Bradk 4=30 in. 15 4=0



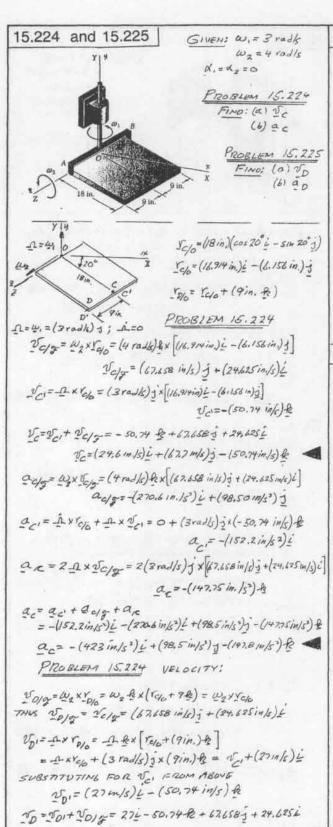


1= w, i = (8 rad/s) i VAI = 1 × (A/O = 8 & × (x = + 3 j) = (24 m./s) & (a) VELOUTY: 2/4 = N/A, + 2/A/5 = 24/2+15/2+25.98] Un= (15in/s) 4+ (26.0 in/s) + (24in/s) +

[CONTINUED)

15.222 continued (b) ACCELETATION: 1 = 0 an/3= + D30= (3011/5) D30= 150 10/5 - D30. Q-A/g==-(129,910/5") + (7510/5") j QAI = 1x TAIO + 1x 1x XAI = 1x (AIO + 12 X JAI QA = 0 + (8 rad/s) i x (24 in 15) & = - (192 m/s2) j a = 2 - a x 1/4/5 = 2 (@rad/s) L x [(1510/s) + (25.9810/s) j a= (415-711/5) } an= an+ an/++ex = - (192 in/s2)-j - (129,911/8) + (7511/8) + (415.7 in/s2) & an = -(129.9 lu/s) L-(117 in/s) ft(416 in/s) & 15.223 GIVEN: W,= Brad/s 04, = 0 u = 30 in.)s 0=120° FIND: (a) VA (b) OA 1 TAIS 19/6 = xA i + (9in. 4) VA/9= 4 7 300 = (30 in/s) \ 300 VA/2=-(15 in/s) + (25.98 in/s) 12= a, i = (8 rad/s) i VAI = CL x rato = 8 Ex (xa E +93) = (72 in/s) & (a) VELOCITY: VA = VAI+ VA/5 = 72-A -15 L +25780 j DA = - (15 in. k) i + (26.0 in./s) j + (72 in./s) & (b) ACCELERATION: _1 = 0 any = 47 7300 = (30 m/s) 7300 = 150 m/s 7 an/2 = - (129.9 in/s2) i - (75 in./52) j QUI = Ux LA/0 + TX Ux LA/0 = - UXX A/0+ TX DA ap=0+(8 rad/s) (x(72 in/s) = -(576 in/s2)5 a=2.1 x V/4/3= 2 (8 rad/s) + x [-(15 in/s) + (25.98 in/s)] a = (415.7 in. 152)-8 an=an+an/2+ac = -(57610/5)j-(129,9 in/5)j-(75 in/5)j+(415.714/5))

QA = - (129.9 in./s) = - (651 in./s) + (416 in./s) /



ND = (57.6 in/s) i + (67.7 in/s) j - (50.710/s) €

(CONTINUED)

15.225 continued

ACCELERATION 1=0

012 = (4 rad/s) & x (67.658 in/s) \$+(24.625 in/s)]

0013 = -(270.63 in/s)] +(98.5 in/s)]

NOTE ! 00/3 = 0 C/3 SIAKE VOIT = VC/2

 $Q_{D} = \frac{1}{12} \times \frac{1}{10} + \frac{1}{12} \times \frac{1}{10}$ $= O + (3 \text{ rad/s}) \frac{1}{12} \times \left[(27 \text{ in./s}) + (50.77 \text{ in/s}) \right]$ $Q_{D} = -(81 \text{ in/s}^2) + (152.2 \text{ in/s}) \frac{1}{12}$

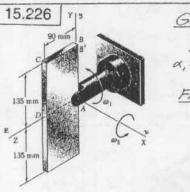
THUS CORIOLIS ACCELERATION OF FOR POINT D

IS SAME AS OF FOR POINT C.

a = -(147.75 in/s') &

ap=ap+ap+ar =-81+-152.21-270.631+98.53-147.75-8

ap=-(423 In/s) + (98.5 in/s) f - (229 in.13) &



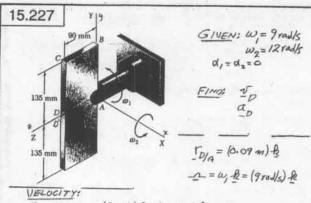
GIVEN: $\omega_1 = 9 \text{ rad/s}$ $\omega_2 = 12 \text{ rad/s}$ $\alpha_1 = \alpha_2 = 0$

FINO: UB

 $Y_{E/A} = 0.135 m j \qquad \underline{C} = \omega_1 R = (9 rad/s) R$ VELOCITY: $\underline{q}_{B} = -0.x \underline{f}_{B/A} = (9 rad/s) R \times (0.135 m j = -le 215 m/s) L$ $\underline{V}_{E/J} = \omega_2 \times \underline{Y}_{B/A} = (12 rad/s) L \times (0.135 m j = -le 62 m/s) L$ $\underline{Y}_{B} = \underline{y}_{B} + \underline{y}_{B/J}$ $\underline{Y}_{B} = -(1.215 m/s) L + (1.12 m/s) R$

ACCELERATION:

 $a_{B} = \underbrace{A \times A \times r_{B|A}}_{E|A} = \underbrace{A \times \delta_{B}}_{B}$ $= (q radk) \underbrace{R}_{E} \times (-1.215 m/k) \underbrace{L}_{E} = -(10.935 m/s^{2}) \underbrace{L}_{E}$ $a_{B/F} = \underbrace{\omega_{2} \times \omega_{2} \times r_{B/A}}_{E} = \underbrace{\omega_{2} \times \delta_{B/F}}_{E}$ $= (12 rad/s) \underbrace{L}_{E} \times (1.62 m/s) \underbrace{R}_{E} = -(12.44 m/s) \underbrace{L}_{E}$ $a_{E} = 2 \underbrace{A \times \delta_{B/F}}_{E} = 2(q rad/s) \underbrace{R}_{E} \times (1.62 m/s) \underbrace{R}_{E} = 0$ $a_{B} = a_{B} + a_{B/F} + a_{E}$ $= -(10.935 m/s^{2}) \underbrace{J}_{E} - (19.44 m/s^{2}) \underbrace{J}_{E}$ $a_{B} = -(30.4 m/s^{2}) \underbrace{J}_{E}$

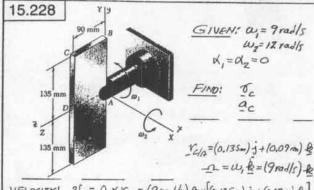


 $V_{D} = I \times I_{D/A} = (9 \text{ rad/s}) f_{0} \times (0.09 \text{ m}) f_{0} = 0$ $V_{D/2} = \omega_{2} \times Y_{D/A} = (12 \text{ rad/s}) i_{0} \times (0.09 \text{ m}) f_{0} = -(1.08 \text{ m/s}) j_{0}$ $V_{D} = V_{D/2} + V_{D/2}$ $V_{D} = -(1.08 \text{ m/s}) j_{0}$

UCCEL FRATION

DI = T X T X LOW = T X TDI = T X D = O

a = a = 1 + 2019 + 9 x = 0 - (12.96 m/s²) + 2 + (19.44 m/s²) i ac = (19.44 m/s²) i - (12.96 m/s²) de



VELOCITY! 2/1=1-xxcA = (9 rod/s) fex (0.135m) 3 + (0.09m) &]
1/1 = - (1.215m/s) L

Ic/g= w2 x rc/a = (12 rad/s) ix [(0.135 m) is +(0.09 m) fz]

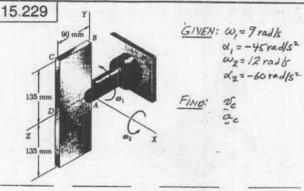
Te/g= (1.62 m/s) ix - (1.08 m/s) ix

No=Ver+No/g: Vc=-(1.215m/s)i-(1.08m/s)j+(1.62m/s)fe ■

ace marion: ac = 1.x1.xrd=1.xnd; = (9radk)&x(-1.215a/s)i=-(10.95a/s)j acg= 42x42xrd= 42x16; = (9radk)&x(-1.215a/s)i=-(10.95a/s)j

an=21×1/2=2(9 rad/s)&x[1.62-1.08]=-(19.44m/s)&-(12.96m/s)& an=21×1/2=2(9 rad/s)&x[1.62m/s)&-(1.08m/s)&]
an=(19.44m/s)&

ac= ac+ ac/s+ ac = -(10.95 m/s) s -(19.44 m/s2) s -(12.96 m/s2) fe +(19.44 m/s2) c ac= (19.44 m/s2) c -(30.4 m/s2) s -(12.96 m/s2) fe



 $\frac{\Lambda}{\omega_{2}} = \omega_{1} \frac{k}{k} = (9 \text{ rad/s}) \frac{R}{k} \qquad \frac{\Lambda}{\omega_{2}} = \omega_{1} \frac{k}{k} = -(45 \text{ rad/s}^{2}) \frac{R}{k}$ $\frac{W_{2}}{\omega_{2}} = \omega_{2} \frac{k}{k} = (12 \text{ rad/s}) \frac{k}{k} \qquad \frac{1}{2} \frac{\omega_{2}}{\omega_{2}} = -(60 \text{ rad/s}^{2}) \frac{k}{k}$ $\frac{Y_{CI}}{\omega_{1}} = (0.135 \text{ m}) \frac{1}{2} + (0.09 \text{ m}) \frac{R}{k}$ $\frac{W_{CI}}{\omega_{1}} = -(1.215 \text{ m/s}) \frac{1}{k} + (0.09 \text{ m}) \frac{R}{k}$ $\frac{W_{CI}}{\omega_{2}} = -(1.215 \text{ m/s}) \frac{1}{k}$

Dely = w2 x 40/4 = (12 radk) i x [(0.135 m) j+(0.09 m) fe]

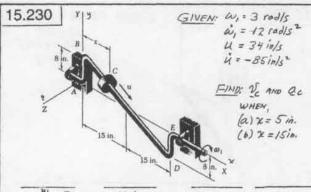
Vc = -(1.215 m/s)i - (1.08 m/s) f +(1.62 m/s) &

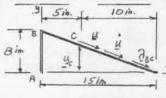
ACCELERATION $Q_{CI} = -\frac{1}{12} \times \frac{1}{6} + \frac{1}{12} \times \frac{1}{2} \times \frac{1}{6} + \frac{1}{12} \times \frac{1}{2} \times \frac{1}{6} + \frac{1}{12} \times \frac{1}{2} \times \frac{1}{6} = \frac{1}{12} \times \frac{1}{2} \times \frac{1}{2$

 $Q_{C/g} = \frac{\omega_2 \times i_{C/g}}{\omega_2 \times i_{C/g}} + \frac{\omega_2 \times \omega_2 + i_{C/g}}{\omega_2 \times i_{C/g}} = \frac{\omega_2 \times i_{C/g}}{(-60 \text{ rad/s}^2) \dot{\Sigma} \times [(0.135 \text{ m}) \dot{\gamma} + (0.09 \text{ m}) \dot{\gamma}_1]} + (1.27 \text{ al/s}) \dot{\Sigma} \times [(1.62 \text{ m/s}) \dot{\gamma}_2 - (1.08 \text{ m/s}) \dot{\gamma}_1]} = -(8.10 \text{ m/s}^2) \frac{\dot{\gamma}_1}{\dot{\gamma}_2} + (5.4 \text{ m/s}^2) \frac{\dot{\gamma}_2}{\dot{\gamma}_2} - (1.94 \text{ m/s}^2) \frac{\dot{\gamma}_1}{\dot{\gamma}_2} - (1.296 \text{ m/s}^2) \frac{\dot{\gamma}_2}{\dot{\gamma}_2} + (1.08 \text{ m/s}^2) \frac{\dot{$

 $a_{c} = a_{c'} + a_{c/g} + a_{k}$ $= (6.075 \, m/s^{2}) \dot{L} - (10.94 \, m/s^{2}) \dot{f}$ $- (14.04 \, m/s^{2}) \dot{g} - (21.06 \, m/s^{2}) de + (19.44 \, m/s^{2}) \dot{k}$

Qc= (25.5 m/s2) L-(25,0 m/s) f-(21.1 m/s2) A





 $\frac{(a)\chi = 5 in.}{3c} = \frac{5 in.}{15 in.}; g = \frac{16}{3} \%$

8 17 ABC 17

20/0= U 2 Be = (34 in/s) 15 1-8 = (30 in/s) 1-(16 in/s)] ac/o= U 2 = (-85 in/s) 15 1-8 = -(75 in/s) + (40 in/s)]

1=101= (3 rad/s); i = wi = (-12 rad/s);

rela = (5 in.) i + (143 in.) i

VELOCITY:

 $\frac{\nabla c_1 = 0. \times c_{da} = (-3 \text{ rad/s}) \dot{c} \times \left[(5 \text{ in.}) \dot{c} + (^{16}/_{3} \text{ in.}) \dot{c} \right] = -(16 \text{ in.}/_{8}) + \frac{1}{6} \\
\mathcal{V}_{c} = \mathcal{V}_{c1} + \mathcal{V}_{c1} = -(16 \text{ in.}/_{8}) + (30 \text{ in.}/_{8}) \dot{c} - (16 \text{ in.}/_{8}) - \frac{1}{3}$

2 = (30 in./s) i - (16 in./s) j - (16 in./s) &

ACCECTATION: ac/2, SEE ABOVE

ac/= in x rep+ in x in x rep= in x rep+ in x v.c.

= (-12 rod/s) i x [(5 in) i + (1/3 in) i] + (3 rad/s) i x (16 in/s) &

ac=-(6+in/82)& - (48 in/82)

an = 21 x Nors=2(3 10 16) x (20 in 15) i - (16 in 15) j] = (96 in/5) & ac= aci + ac/s + ac

=-(6+in/s2) &-(48/m/s2) 3-(75/m/s2) +(40/m/s2) + (96/m/s2) &

Qc = -(75 in/s+) ½ -(8 in/s2) j +(32 in/s2) € (b) <u>y = 15 in</u>. (COLLAR C IS IN X2 PLANE): <u>V</u> = 0

VELOCITY: VC/g = SAME AS IN PART a ABOVE $V = V_{C} + V_{C}/g = O + V_{C}/g$: $V = (30 in/s)\hat{c} - (16 in/s)\hat{g}$

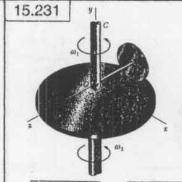
ACCEL GRATIONS

ac/g = SAME AS IN PART & ABOVE ac = 0; SINCE COLLAR LIES ON AXIS OF RUTATION

ac = 2.1 x Tely = SAME AS IN PART a ABOVE

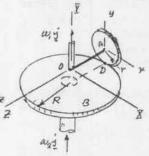
 $a_{c} = a_{c} + a_{c} + a_{r}$ $= 0 - (75 in/s^{2}) + (40 in/s^{2}) + (96 in$

a=-(75111,15") ++(40 in.15") ++(96 in.15") +



GINEN : W, = W, - R W, = 0 W2 = W2 /2 d2 = 0

FIND: FOR DISK,
(a) WA
(b) QA



MOVING FRAME A 242
ROTATES WITH
ANGUM VELOCITY IL = W, J

WOISH/= W, L+W, R

10/4 = - 13 - 15-15

(1) TOTAL ANSWAR VELOCITY OF DISK A: $\omega = \omega_j + \omega_{ask/g} = \omega_2 i + \omega_i j + \omega_2 k$

DENOTE BY D POINT OF CONTACT OF DISIS

Consider ask R: $V_0 = w_2 j \chi(-R k) = -Rw_2 i$ (2)

CONGION SYSTEM OC, OA, AND DISK A.

Voi = -ax rom = wisx(-ris-rig) = - Rwis

 $I_{D} = I_{D}I + I_{D}I_{S} = -R \omega_{1}\dot{L} - r\omega_{2}\dot{L} + R\omega_{2}\dot{L} + r\omega_{2}\dot{L}$ (3) EQUATE $I_{D} = V_{D}$ From Eq. 2 AMP EQ. 3

-Rwz = -Rw, +rwz +Rwz - rwz &

COEFOF j: 0=Rwz -> wx=0

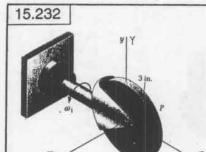
 $\frac{\cos \kappa \omega}{\cos \kappa} \pm 1 - \kappa \omega_{2} = -\kappa \omega_{1} + \kappa \omega_{2}; \quad \omega_{2} = \frac{\kappa}{\kappa} (\omega_{1} - \omega_{2})$

EG3: W=W, j+ 1 (W,-W2) to

(b) DISK A ROTATES ABOUT & AXIS AT RATE W.

 $d = \omega_1 \times \omega = \omega_1 \underbrace{1}_{1} \times \left[\omega_1 \underbrace{1}_{1} + \underbrace{R}_{1} (\omega_1 - \omega_2)\right] \underbrace{1}_{1} \underbrace{1}$

x= w, (w,-w2) R i



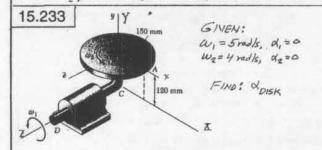
GIVEN: cu, = 5 rad/s, d, = 0 wz= trads, do=0 G=300

FINO! ap

FRAME OXYZ IS FIXED, MOVINE FRAME OXYZ ROTATES WITH ANGUAR VELUCITY _ = W, [= (5 rad/s) [- Plo = (3in.) cos 30 6 + (3in) sin 30 j = (2,598 in.) + (1.5 in.) + work/g= anh = (4 rad/s) &

TP/3 = W DISK/ = X 19/0 = (4 rad/s) & x [2.598 i +15-5] VP/0= (10.392 in/s) j - (6 in/s) = Spi = 1 x / No = (5 malls) & [2.598 + 4.5] No= (2514.15)&

ACCEL 02 ATION: ap= -1 x Up= (Stad/s) ix (2.5 m/s) = -(325 m/s2) j ap/ = " DISHING " TP/g= (4 rad/s) &- [(10.392 in/s)] - (6 m/s) i] appa = - (41.569 m/s') = - (241m./s') j ax = 2.1 4 Very = 2 (5 rad/s) 1 x [(10, 392 in./s) 1 - (6 in./s) 1] ax= (103.92 in./52) & ap=ap+apy+ae=-37.5j-41.569i-24j+103.92 & ap=-(4.6 m/s)i - (61.5 m/s) + (103,91m/s) A



FRAME CXYZ IS FIXED

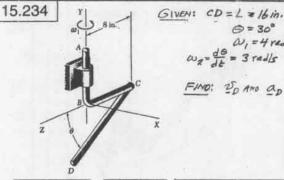
MOVING FRAME BYGE ROTATES INITH ANGULAR VELOGITY _A = W, & = (5 rad/s) & AROUT 2 AXIS

work = w, te + was

X DISK = A x WDISK = W, & x (W, & + W, 1) = -W, W, i

doisk = - (Sradk) 4 radk) i

Disk = - (20 rad/s2) 6



@=30° a, = 4 rad/s wa=do = 3 radis FIND: DO AND ap

1=0,5 FRAME BXYZ IS FIND. MOVING FRAME CEYE ROTATS ABOUT Y ANS WITH IL = WIS 「DIB=-L SIN母 j+(Lcosの-与)を role = - L sine & + L cose &

VELOCITY: VD = 1-x YO/8 = O, jx -LSING; +L(COSO-+)& Tp=Lw(cos6-1)i

VD/7 = w2x rok = w2 is (-15 is + 1 cose &) Volg = - Lwa sing & - Lwa cases

J= Jo: + J0/2 Vp = La, (cos6-1) i - Lw2000 j - Lw251100 &

ACCOLUMNTION: ap = 1 x Vp = w, ix Lw, (coso - 1)i Q01 = - LW, 2(cos & - 1) &

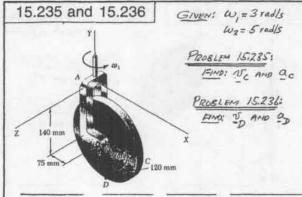
a D/7 = Wex NO/5 = Weix [-LW sino &-Luz coss] 20/7 = +LW2 sin & j - LW2 cos & R ar= 2-1 x voy= 2 w, j x (-Lusin6 & - Las cos6 j) OR=-2LW, Wasing i

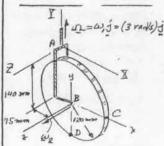
OD = OD 1 TOD/2 + ar =-Lu, (coso-1) &+Lu, sino j-Lu, coso &-2Lb, u, sno L $a_{D} = -2L\omega_{i}u_{1}sin\Theta\hat{l} + L\omega_{2}^{2}sin\Theta\hat{j} + (-L\omega_{1}^{2}(cos\Theta - \frac{1}{2}) - L\omega_{2}^{2}(cos\Theta)\frac{1}{2}$

DATA: 6=30, L=16 in., W,=4 rad/s, W=3 rad/s D= 16(4)(cos 36°-1) L - 16(3)cos 30° j - 16(3) sin 30° 2 250 = (23.4 in/s) i - (41.6 in/s) = - (24 in/s) &

ap= -2(16)(2)(4) sin 30 + 16(3) 310 30 5 +(-16(4) (cos 30 - 1) - 16/3) cos 30)

ap=-(192 in./s2) + (72 in./s2) + - (218 in.153) &





FRAME AXYZ IS FIXED MOVING FRAME BRYZ
ROTHTES ABOUT Y AND
WITH A = (3 rad/s) &

M2 = (5 rod/s) &

PROBLEM 15.235: FOR POINT C - 5/4 = (195mm) - (140mm) + 5: 50/8 = (120mm) - 120mm) + 120mm

$$\begin{split} &\mathcal{T}_{C,l} = \Delta \times f_{cl,k} = (3 \text{ rad/s}) \hat{j} \times (195 \hat{c} - 140 \hat{j}) = -(585 \text{ mm/s}) \hat{k} \\ &\mathcal{V}_{Cl,k} = \omega_{R} \times f_{cl,k} = (5 \text{ rad/s}) \hat{k} \times (120 \hat{c}) = + (600 \text{ mm/s}) \hat{j} \\ &\mathcal{V}_{C} = \mathcal{V}_{Cl} + \mathcal{T}_{Cl,k} = \mathcal{V}_{Cl} = (600 \text{ mm/s}) \hat{j} - (585 \text{ mm/s}) \hat{k} \end{split}$$

ACCEL BE ATIOY

 $Q_{cl} = Q \times V_{cl} = (3 \text{ rad/s}) \dot{g} \times (-58 \dot{s} \text{ mm/s}) \dot{q} = -(1.755 \text{ m/s}^2) \dot{c}$ $Q_{cl} = W_2 \times V_{cl} = (5 \text{ rad/s}) \dot{q} \times (600 \text{ mm/s}) \dot{g} = -(3.00 \text{ m/s}^2) \dot{c}$ $Q_{cl} = 2 \cdot Q \times V_{cl} = 2(3 \text{ rad/s}) \dot{q} \times (600 \text{ mm/s}) \dot{q} = 0$ $Q_{cl} = Q_{cl} + Q_{cl} + Q_{cl} = -(1.755 \text{ m/s}^2) \dot{c} - (3.00 \text{ m/s}^2) \dot{c}$ $Q_{cl} = -(4.76 \text{ m/s}^2) \dot{c}$

PROBLEM 15.236: FOR POINT D

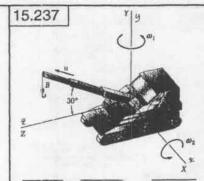
YOLF (75 mm) 1- (260 mm) 1 ; YOLF - (120 mm) 1

VELOCITY

$$\begin{split} \mathcal{D}_{D} &= \Delta \times V_{D/A} = (3 \text{ rad/b} \int_{1}^{1} y(75 - 260 \frac{1}{2}) = -(225 \text{ monk}) \frac{1}{4} \\ \mathcal{D}_{D/Q} &= W_{Q} \times V_{D/Q} = (5 \text{ rad/s}) \frac{1}{4} \times (-120 \frac{1}{2}) = (600 \text{ mon/s}) \frac{1}{6} \\ \mathcal{D}_{C} &= \mathcal{D}_{C} + \mathcal{D}_{C/Q} \\ \mathcal{D}_{C} &= (600 \text{ mon/s}) \frac{1}{6} - (225 \text{ mon/s}) \frac{1}{6} \end{split}$$

ACCELERATION $Q_{D} = (1 \times \sqrt{2}) = (3 \text{ rad/s}) \frac{1}{2} \times (-225 \text{ mm/s}) \frac{1}{2} = -(0.65 \text{ m/s}) \frac{1}{2}$ $Q_{D} = (2 \times \sqrt{2}) \frac{1}{2} = (5 \text{ rad/s}) \frac{1}{2} \times (600 \text{ mm/s}) \frac{1}{2} = +(3.00 \text{ m/s}) \frac{1}{2}$ $Q_{C} = 2(1 \times \sqrt{2}) \frac{1}{2} = 2(3 \text{ rad/s}) \frac{1}{2} \times (600 \text{ mm/s}) \frac{1}{2} = -(3.60 \text{ m/s}) \frac{1}{2}$ $Q_{C} = Q_{C} + Q_{C} \frac{1}{2} + Q_{C}$

ac=-(0.675m/s) + (3,00m/s) j-(3,60m/s)-&



FINO: No AND aB

FRAME ABYZ IS FIXED, MOVING FRAME OXYZ ROTATES

ABOUT Y ANIS WITH - 1 = W, j = (0.25 radk) j

Wz = (0.40 radk) t; AB = Sin30° j + cos30° k

18/A = (AB) 2AB = (10 ft) j + (17.321 AH) - R

W = U 2AB = (1.5 ft/s) 2AB = (0.75 ft/s) j + (1.299 ft/s) R

VELUCTY:

 $\frac{U_{B1}}{U_{B1}} = \int_{\mathbb{R}} X Y_{B1} = (0.25 radk) \frac{1}{2} x \left(\frac{103}{103} + \frac{17.32 \frac{12}{10}}{103} \right) = \left(\frac{4.33}{103} \right) \frac{1}{103}$ $= 0.75 \frac{1}{0} + \frac{17.29}{103} + \frac{17.29}{103} + \frac{17.29}{103} + \frac{17.29}{103} = 0.75 \frac{1}{0} + \frac{17.29}{103} + \frac{17.29}{103} + \frac{17.29}{103} = 0.75 \frac{1}{0} + \frac{17.29}{103} + \frac{17.29}{103} = 0.75 \frac{1}{0} + \frac{17.29}{103} + \frac{17.29}{103} = 0.75 \frac{1}{0} = 0.75 \frac$

 $V_{3/9} = -(6.726 \text{ ft/s}) \cdot \vec{j} + (5.299 \text{ ft/s}) \cdot \frac{1}{12}$ $V = V_{8} + V_{8/9} = (4.33 \text{ ft/s}) \cdot (6.178 \text{ ft/s}) \cdot \frac{1}{2} + (5.299 \text{ ft/s}) \cdot \frac{1}{12}$ $V = (4.33 \text{ ft/s}) \cdot (6.18 \text{ ft/s}) \cdot \frac{1}{2} + (5.30 \text{ ft/s}) \cdot \frac{1}{2}$ ACCELERATION:

COINCIDING POWT AS AS OF 8 MOVING AT U. AS
ROTATES WITH WE ABOUT AS AS ROTATES ABOUT & AXIS
Y AXIS

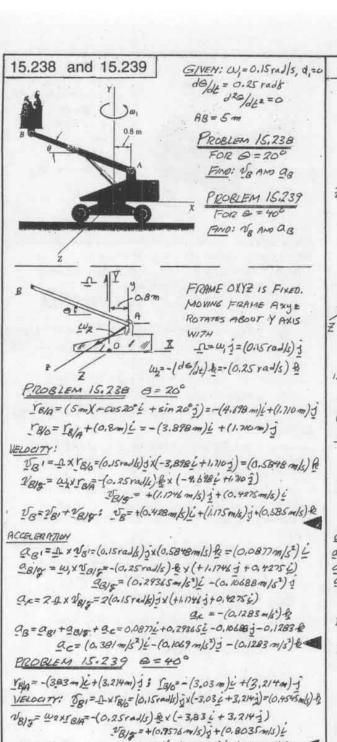
 $2a/g = (0.4 \frac{1}{6}) \times (0.4 \frac{1}{6} \times (10\frac{1}{3} + 17.32 \frac{1}{6})) + 2(0.4 \frac{1}{6}) \times (0.75\frac{1}{3} + 1.299\frac{1}{6})$ $= (0.4 \frac{1}{6}) \times (1.4 \frac{1}{6} - 6.928\frac{1}{3}) + 0.6 \frac{1}{6} - 1.0392\frac{1}{3}$ $= -1.6 \frac{1}{3} - 2.771 \frac{1}{6} + 0.6 \frac{1}{6} - 1.0392\frac{1}{3}$ $= -(2.639465^{2})\frac{1}{3} - (2.171466^{2})\frac{1}{6}$

A TE CORDUS ACCELERATION OF B MOVING WITH
FRAME A LYE 'IN ROTATION AGOUT Y . WITH
ANGULAR VECOCITY _______

2 = 2 1. x 2 818 = 2(0.25 rad/s) & (6.128 AHS) = +(5.299 AHC) &] = (2.650 ft/s) i

 $\alpha_{B} = \alpha_{B1} + \alpha_{B1} + \alpha_{C}$ $= -(1.083 \text{ fd/s}^{2}) - (2.139 \text{ fd/s}^{2}) - (2.171 \text{ fd/s}^{2}) - k$ $+ (2.150 \text{ fd/s}^{2}) - k$

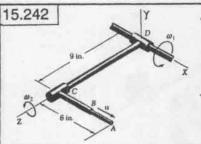
ag = (2.65 A/s) i - (2.44+/s) j - (3.25 A/s2) &



aB = (0.308 m/s2) i - (0,201 m/s2) - (0.241 m/s2) &

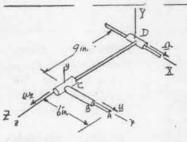
15.240 and 15.241 PROBLEM 15.240 FIND: VA AND CLA PIZOBLEM 15.241 Finn: 1 AND aB 360 r FRAME DXYZ IS FIXED. 180 mm MOVING FRAME Cays D=N,J ROTATES ABOUT D Y AXIS WITH IL=Wj=(Brad/s) 5 w= (12 rad/s) & 360 PROBLEM 15.240: FOR POINT A TOM= (0.15m) + (0.18m) j- (0.36m) A; [A/c=(0.18m)s VELOCITY: VAI = 1 X YOR = (8 rad/s) jx(0.15i+0.18j-0.36-8) DA'= - (1.2 m/s) fe - (2.88 m/s) L 2A/8= Wax Int = (12 rad/s) tox (0.18 m) = - (2.16 m/s) L 29-161+VAIT = - (1.2 m/s) &- (2.88 m/s) i - (2.16 m/s) & V=-(5.0+m/s) 1- (1.2 m/s) 12 2A/g= W2X 5A/g= (12 radis) (x(-2.4 a/s) = - (25.92 a/s)) 5 ar= - (9.6 m/s2)i- (25.9 m/s)j+ (576m/2)A PROBLEM 15,241 FOR POINT B IB/A = (0.15m) & - (0.18m) f - (0.36m) fc; IBIC = - (0.18m) 1 VELOGTY: TB1=1xx8/A=(8ral/s)jx(a,15i-0,18j-0,28) TBI= - (1.2 m/s) & - (2.88 m/s) i VE/5 = Wax TB/E = (12 rad/s) &x (-0.18m) = (2.16m/s) & NB = DB1 + DB/g= - (12m/s)&-(2.88m/s)i+(2.16 m/s)i NB=V8+V8/9 1 1B=+(0.803 m/s) (+(0.958 m/s) +(0.955 m/s) & UB = - (0.72m/s) - - (1.2m/s) & ACCELERATION; ACCECERATION: aB1= 1×08 = (8 rad/s) 5 x (1.2 & -2.80 i) QBI = 1 XVBI = (0.15 rad/s) (x(0.4545 m/s) &= (0.0682 m/s2) = a 31 = - (9.6 m/s2) + (23.04 m/s) & QQq= w, x2(q)==(0,25red(s) &x(+0.9576+0.8035 i) 28/9= (0.2394 m/s2) L-(0.2009 m/s2) 5 Q = 2.1. x 28/5 = 2(0.15 rad/s) fx (+0,75% j +0.80352) ar=-(0,240m/s)/R QB = QBI + QBIS a= aB+ 98/8+a = (0,0682m/s) + (0,7394m/s2)i-(0,2009m/s2)-j-(0,2410m/s2)&

GIVEN: W= 8 rad/s, 0,=0 W= 12 rad/s, 0/2=0 # A= 1x 1/41= (8 ra //s) jx (-1.2-8 -2.88 i) = -(9.6 m/s²) i + (23.04 m/s²) & ar = 2.1 × 5/1/5 = 2(8 rad/s) jx(-2.16 m/s) = (34,58 m/s2) /2 Qq = Qq + 4/6 + 2r = -(96m/s2) = +(230+m/s2) & -(2502m/s2) +(34,58m/s2) & 2B/g= W2X 18/g=(12 rad/s) &x(2.16 m/s) = (25.92 m/s2)1 a = 21 x1/8/9 = 2(8 rad/s) 1x (2.16 m/s) = - (34,56 m/s') & = - (9.6 m/s2) + (23.04 m/s) + (25,92m/s) + - (34.56 m/s) + ag= - (9.6 m/s) i+ (25.9 m/s2) - (1/52 m/s2) &



GIVEN: W = 1.2 rad/s, 0,00 W== 1.5 rad/s, d==0 4= 3 in./s, 4=0

FIND: VA AND QA



FRAME DXYZ IS FIXED. MOVING FRAME Caye ROTATES ABOUT THE YAXE WITH n= wi= (1,2 rad/s) i. W= W2 & (1.5 rod/s) & 4=4 = (3 in./5) =

YAID= (6 in.) + (9 in.) &

VELOGITY TAI= 1. x YAID = (1.2 rad/s) ix (6 in) = + (9 in) to = - (10.8 in to) j VA/g= 02x[A/c+4= (1.5 rad/s) &x (6in.) i+ (3in./s) i TA/2= + (9 in./s) + (3 in./s) L

In = Vai + Valg= - (10.8 in./s) j + (9 in/s) j + (3 in./s) 4 TA= (3/m./s)i-(1.8 in./s)3

ACCEL FRATION

$$a_{Ai} = \Delta \times \Delta \times A_{AD} = \Delta \times A_{A} = (L2 \text{ rod/s}) L \times (-10.8 \text{ ln/s}) \frac{1}{2}$$

$$a_{A} = -(12.96 \text{ in/s}^2) + 2 < 0$$

QA/48 NOTE, SINCE POINT A MOVES IN THE ROTATING FRAME CRYZ THERE IS A CORIOIR ACCELERATION.

aA/2 = W2 X W2 X VA/c + 2 W2 X LL = W2 x (1-5 rads) & x (6 in,) i + 2 (1.5 rads) & x (3 rads) i = (1.5 radle) & x (+911.15) + (9 in/5) +

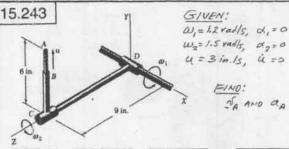
and = - (13.5 in. 152) + (9 in/52) +

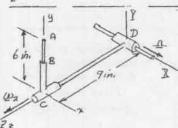
ac: CORIOLIS ACCELERATION DUE TO A MOVING INITH VELOCITY VALT

ar = 21 x NA/2 = Z (1.2 rad/s) Lx (9111.15) + (3111/s)) ar = (216 in./s2) &

a=aAI+aAIx+ax = - (12.96 in. 152) & - (13.5 in. 15) i+ (9 in/52) j+ (71.6 in/52) &

Q= -(13,5 m/s2) i+(9 m/s2) + (8.64 in./s2) +





FRAME DXYZ IS FIXED. MOVING FRAME CX42 ROTATES ABOUT THE PAXIS WITH 1=W, i= (1.2 rod/s) i.

w2 = w2 k=- (1.5 rods) & u=u0=(3 in./s) i

TAID = (6in) + (9in) +

VELOCITY: TAI = 1 x TAID = (12 mails) & A [(61m) 3+(91m) &]: 25A1= (7.214./5) to - (10.814.15) f VAIS = W2 X YA/E = - (1.5 rad/s) Ax x (6 in.) + (3 in.15) 1

VA/g= + (9in./s) + (3in./s) j 1 = JA+ 2 A/3 = (7.2 in.ls) & - (10.8 in.ls) = - (9 in.ls) & + (3 in.ls) } NA = + (9 in./s) i - (7.8 in./s) f + (7.2 in/s) &

ACCELER ATIONS

$$\alpha_{p,1} = \underline{\Lambda} \times \underline{\Lambda} \times \underline{Y}_{A|D} = \underline{\Lambda} \times \underline{N}_{A'} = (1.2 \text{ rad/s}) \underline{\hat{c}} \times \left[(22 \text{ in/s}) \underline{\hat{N}} - (10.8 \text{ in/s}) \underline{\hat{s}} \right] \\
\alpha_{p,1} = \underline{\Lambda} \times \underline{\Lambda} \times \underline{Y}_{A|D} = \underline{\Lambda} \times \underline{N}_{A'} = (1.2 \text{ rad/s}) \underline{\hat{c}} \times \left[(22 \text{ in/s}) \underline{\hat{N}} - (10.8 \text{ in/s}) \underline{\hat{s}} \right]$$

QAID: NOTE SINCE POINT A MOVES IN THE ROTATING FRAME CRYZ THERE IS A CORIDLIS ACCELERATION

aA/2 = W2 XW2 X TAK+ 2 W2 X L = W2 x (1.5 rad/s) &x (610.) 3 + 2 (1.5 rad/s) &x (3 rad/s) 3 = (15 rails) + x (-9 in. 15) = - (9 in. 15) =

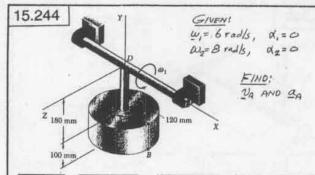
PA/5= - (13,5 in,152) f - (9 in,152) 6

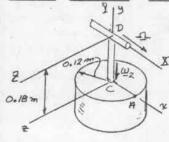
ar: CORIOLIS ACCELERATION DUE TO A MOVING INITH VELOCITY JAIS

ax = 21 x 1/3 = 2(1.2 radk) ix (+19in/s) i+ (3 in/s) j ac= (7.2 in/53) &

an = an + ans +an = -(8.64 in.15) 5 - (12.96 in.152) \$ - (13.5 in.152) \$ + (9 in,/52) ¿ + (22 in/52) te

an=+(9 in.15) 6 - (22.1 in/5) 5 - (5.76 in.15) k





FRAME DXYZ IS FIXED.

MOWING FRAME, Caye,

ROTATES ABOUT THE

I AXIS WITH $\Omega = \omega, \dot{L} = (6 \text{ rad/s}) \dot{L}$ $\omega_z = \omega_z \dot{J} = -(8 \text{ rad/s}) \dot{J}$

VIEL OCITY: Vai = 10 x (A10 = (6 rad/s) & x [+(0.12 m) & -(0.18 m)] TAI = -(1.08 m/6)-R

2/A/5= W2 X YAIC = - (8 rad/s) 1 X (0.12 m) = (0.96 m/s) &

ACCELERATION:

2A15 = 12 × VAIG = -[8 rad/s) 1× (0.96 m/s) & QA15 = - (7.68 m/s) L

ax = 2.0 × 1/4/5 = 2(6 rad/s) (x (0.96 m/s) & a = - (11.52 m/s) -

 $Q_{A} = Q_{A} + Q_{A}/g + Q_{R}$ $= (6.48 \text{ m/s}^{2}) \cdot \hat{s} - (7.68 \text{ m/s}^{2}) \cdot \hat{L} - (41.52 \text{ m/s}^{2}) \cdot \hat{s}$ $Q_{A} = -(7.68 \text{ m/s}^{2}) \cdot \hat{L} - (5.04 \text{ m/s}^{2}) \cdot \hat{s}$

15.245

Y

GIVEN:

ω, = 6 radk, α, = 0

ω_q = 8 radk, α_q = 0

π

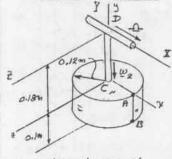
Σ

100 mm

R

120 mm

χ



FRAME DXYZ IS FIXED.

MOVING FRAME, Czys,

ROTATES ABOUT THE

Y AXIS WITH

SL=W, = (6 rad/s) =

W2=W2J=-(8 rad/s) =

+B10 = +(0.12m) = -(0.28m) = TBL= (0.12m) = -(0.1m) =

 $\frac{VELOCITY:}{\sigma_{Bl}^{2} = \Omega_{e} \times r_{Blo} = (6 \text{ rad/s}) \dot{L} \times \left[(0.12 \text{ m}) \dot{L} - (0.28 \text{ m}) \dot{\tilde{J}} \right]}{\mathcal{I}_{Bl}^{2} = - (1.68 \text{ m/s}) \dot{R}}$ $\frac{V_{Blog} = \omega_{g} \times r_{Bl} = - (8 \text{ rad/s}) \dot{\tilde{J}} \times \left[(0.12 \text{ m}) \dot{L} - (0.1 \text{ m}) \dot{\tilde{J}} \right]}{(0.12 \text{ m}) \dot{L} - (0.1 \text{ m}) \dot{\tilde{J}}}$

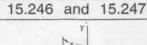
VB=VB+ 18/9= - (1.68 m/s) & + (0.96 m/s) & VB=- (0.72 m/s) &

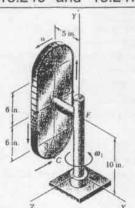
18/7 = (0.96 m/s)-A

ACCELERATION: $a_{B} = \Delta \times \underline{v}_{B} = (6 \text{ rad/s}) \hat{\iota} \times (-1.68 \text{ m/s}) - \underline{\theta}$ $a_{B} = (10.08 \text{ m/s}^{2}) \cdot \hat{g}$ $a_{B} = (10.08 \text{ m/s}^{2}) \cdot \hat{g}$ $a_{B} = -(1.68 \text{ m/s}^{2}) \cdot \hat{g}$ $a_{B} = -(1.68 \text{ m/s}^{2}) \cdot \hat{g}$ $a_{B} = -(1.68 \text{ m/s}^{2}) \cdot \hat{g}$ $a_{B} = -(1.68 \text{ m/s}^{2}) \cdot \hat{g}$

 $a_{B} = a_{B} + a_{B/F} + a_{C}$ $= (10.00 \text{ m/s}^{2}) \dot{g} - (7.60 \text{ m/s}^{2}) \dot{c} - (11.52 \text{ m/s}^{2}) \dot{g}$

a = - (2.68 m/s²) i - (1.44 m/s²) j

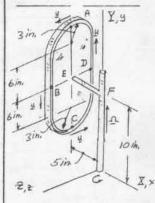




GIVEN: W.=1.6 rad/s, 0,=0 LINK BELT MOVES AROUND PERIMETER AT CONSTANT SPEED 4 = 4.5 in./s.

PROBLEM 15.246 FIND: (a) an (b) aB

PROBLEM 15.247 FIMO: (a) ac (6) 20



FRAME GXYZ IS FIXED. MOVING FRAME, GX42, ROTATES ABOUT THE Y AXIS WITH a= w, j = (1.6 rad/s) j

PROBLEM 15.246: (a) POINT A: 4= (4.5 m./s) & TAG - (5 in.) i+ (19 in.)

VAI = - 1 × SA16 = (1.6 rad/s) j x [-(5 m.) + (19 in)] = (8 in/s) - (8 in/s) - (8 in/s) 2/A/8= 4 = (4,5 in./s) -R

VA = VA, + VA/5 = (8in.15) & + (4.5in.15) & VA= (125 in.15)-R

an = 1 x JA1 = (1.6 radk) + x (8 in./s) & = (12.80 in./s) & 2A/g=- "= - (4.510/s)2 = - (6.75in./s2) =

ax = 21x4 = 2(1.6 rad/s) jx(4.5 in./s) = (14.4 in./s) ; 90=0A1+0A15+0x

a= (12.80 in/s) = - (6.75 in/s2) + (14.4 in. 62) =

an= (27,2in/s2)=-(6,75 in./s2)+

(CONTINUED)

15.246 and 15.247 continued

4=- (4.5 in./s) J PROBLEM 15.246: (6) POINT B: IB/c = - (5in.) + (10in.) + (3in) &

Vg1 = 1 × [816 = (1.611./s) jx -(511.) i+(1011.) j+(311.) 2 Var= (8 in. 15) 12 + (4.8 in.15) 4

VB/8= U= - (4.5 In. 15) j

UB = UB+ 2B/ = (8in/s) + (4.8in/s) i - (4.5in/s) i DB= + (4.8in./s) = (4.5in./s) + (8 m./s) A

ag= 1 x VRI = (1.6 ink) j x (8 in. 4) & + (4.8 in. 6) } ari= (12.8 in.15°) L- (7.68 in.15°) &

a = 2 1. xv Byg = 2(1.6 in./s) j x (-4.5 in./s) j = 0 28 = 281+9818+9c = (12.8 m/s2)i-(268 in/s) &+0+0 an= (12.8 in.15) = - (2.68 in.152) &

PROBLEM 15,247 (a) POINT C: U=-(45 in/s) & Yola = - (57mils) + (1 in.) j 26 = 1 x x = 16 = (1.6 in.16) jx - (5 in.) + (1 in) = (8 in.15) &

2015= 4 = - (4.5in.15) A Je = Ver + Very = (Bin./s) te-(4.5in./s) te Vo=(3.5in./s)-R

ac=-1×150=(1.6 in./s) x (8 in./s) &= (12.80 in.k2) & $a_{c/g} = \frac{u^2}{r}j = \frac{(4.5in./s)^2}{(3in./s)}j = (6.75in./s^2)j$

ax = 2.1 x 2/4 = 2(1.6 in.16) +x (-4.5 in.15) = - (14.40 in/52) [

a= ac+ a4 + + ac

a = (12.80 in.152) £ + (6.75 in.15) = - (14.40 in/52) £ ac = - (16 in.15") + (6,75 in. 15")]

u= (4,511./s)j (b) POINT D: TOIG= - (5in.) + (10in.) + - (3in.) - k

Not = -1 x Tolk = (1.6 in./s) fx - (5in.) + (10in.) f - (3in.) } To 1= (Bin./s) & - (418 in/s) &

TD/3= a= (4.5/n./s) 5

To = 10,+ 10/2 (810/5) & - (4.8 in.15) + (4.5 m/s) }

N=-(4.8 in. 15) + (4.5 in/s) + (8 in./s) &

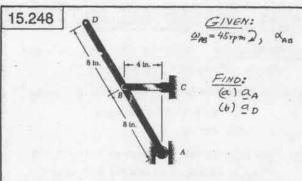
ap= 1 x Vp= (1.6 rad/s) 1x (810.15) A-(4.8 in. k) []

ap= (12.8 in./s2)+ (7.68 in/s2)-1

20/5=0 ax=21×Voy=2(1.6 vol/s) x (4.5 in.6) j=0

ap = ap +ap/s+ax = (12.8 in./s) + + (7.6 & in./s) ++0+0

ap= (12.8 m./52) + (7.68 in./52) &



CRANK BC:
$$\omega_{8c} = (45 \text{ rpm}) \frac{217}{60} = 4.7124 \text{ rad/s}$$
 $\omega_{8c} = (80)\omega_{8c}^2 = (4 \text{ in})(4.7124 \text{ rad/s})^2$
 $\omega_{8c} = 4 \text{ in}$
 $\omega_{8c} = 4 \text{ in}$
 $\omega_{8c} = 4 \text{ in}$

BAR ABD:

VELOCITY:

VELOCITY:

NA THUS, WASO = 0

ACCELERATION

$$D \stackrel{Q_D}{=} Hin,$$

$$Bin. \qquad B = B \qquad B \qquad B \qquad B \qquad B \qquad ABD$$

$$Bin. \qquad Bin. \qquad$$

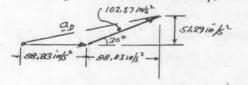
PLANE MOTION = TRANS, WITH B + ROTATION

ABOUT B

app=(80) dABD=(8 in.) dABD } NOTE
app=(80) dABD=(8 in.) dABD } and = app

Pant D: ap = ap + app = ap + app & 300

ap = 88.83 m./s = + 102.57 m/s 2 & 300



177.04 ings = 16.10 Y= 16.10

ap = 184.9 in/s2 2 16.10

15.249 GIVEN: ROTOR IN UNIFORMLY

ACCELERATED MOTION

±=0, W_= 1800 ypm, &=0

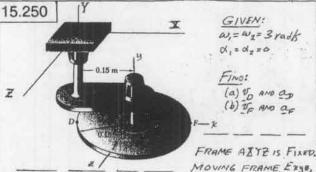
FINO: (a) X, (b) + REQUIRED TO COME TO MEST.

Wo = 1800 rpm (27) = 188.50 rad/s 0 = 1550 rev (27) = 9739 rad

(a) ANGULAR ACCELERATION; (USF LAST OF EQS. 15.16) $W^2 = W_0^2 + 2d(G - G)$ $O = (188,5 \text{ rails})^2 + 2d(9789 \text{ rail} - 0)$ $d = -1.824 \text{ rails}^2$

(b) TIME REQUIRED TO \$708: (USE FIRST OF EQS. 15.16) $W = W_0 + dt$ 0 = 188.5 nad/s - (1.824 rad/s) t

t= 103.3 s



ROTATES ABOUT Y AXIS AT $\underline{\Omega} = \omega_1 j = (3 \operatorname{rad/s}) j$.
(a) POINT D: $\omega_2 = \omega_2 j = (3 \operatorname{rad/s}) j$

 $\mathcal{D} = \mathcal{D}_{1} + \mathcal{D}_{1} = 0$ $\mathcal{D}_{1} = \mathcal{D}_{1} + \mathcal{D}_{1} = 0$ $\mathcal{D}_{2} = \mathcal{D}_{1} + \mathcal{D}_{2} = 0$ $\mathcal{D}_{3} = \mathcal{D}_{3} + \mathcal{D}_{4} = 0$ $\mathcal{D}_{3} = \mathcal{D}_{4} + \mathcal{D}_{5} = 0$

 $Q_{plg} = W_2 \times V_{0/g} = (3 \text{ rad/s}) \hat{g} \times (0.45 \text{ m/s}) \hat{g} = (1.35 \text{ m/s}) \hat{L}$ $Q_L = 2 \cdot \Omega \times V_{plg} = 2(3 \text{ rad/s}) \hat{g} \times (0.45 \text{ m/s}) \hat{g} = (2.20 \text{ m/s}) \hat{L}$ $Q_0 = Q_0 + Q_0 + Q_0 + Q_0 = 0 + (1.35 \text{ m/s}) \hat{L} + (2.20 \text{ m/s}) \hat{L}$ $Q_0 = (4.05 \text{ m/s}^2) \hat{L}$

(b) POINT F: $\omega_2 = \omega_2 \hat{j} = (3 \text{ rad/s}) \hat{j}$ $\Gamma_{F/A} = (0.3 \text{ m}) \hat{\nu} \hat{j}$ $\Gamma_{F/B} = (0.15 \text{ m}) \hat{\nu}$

$$\begin{split} & \mathcal{V}_{F_i} = \underline{\Lambda} \times \mathcal{V}_{F/F} = (3 \text{ vod/s}) \dot{\mathbf{y}} \times (0.3 \text{ m}) \dot{\mathbf{u}} = -(0.9 \text{ m/s}) \dot{\mathbf{g}} \\ & \mathcal{Y}_{F/F} = \mathcal{W}_2 \times \mathcal{V}_{F/F} = (3 \text{ vod/s}) \dot{\mathbf{j}} \times (0.15 \text{ m}) \dot{\mathbf{u}} = -(0.45 \text{ m/s}) \dot{\mathbf{g}} \end{split}$$

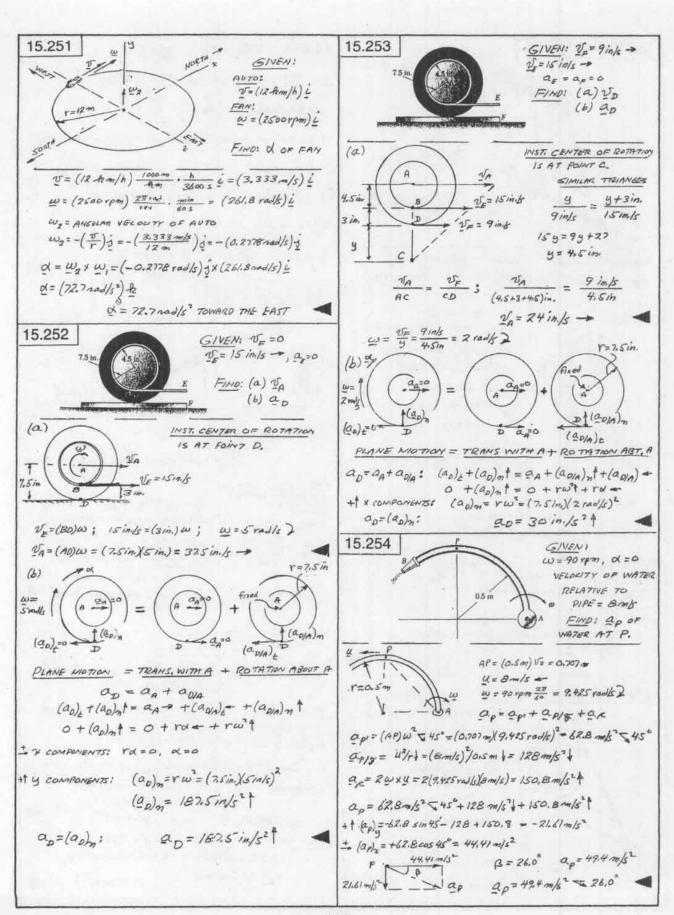
TE= SE+ VE/5= - (0.9 m/s) & - (0.45 m/s) & VF= - (1.35 m/s) &

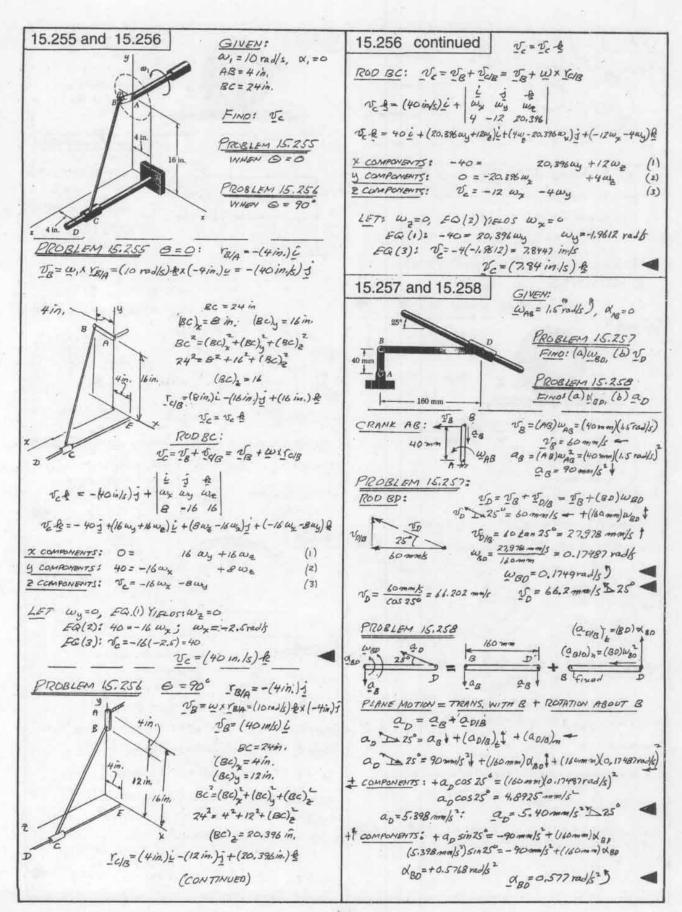
ari=1x Vri= (3 rad/s) jx(-0.9 m/s) &= -(2.7 m/s) &=

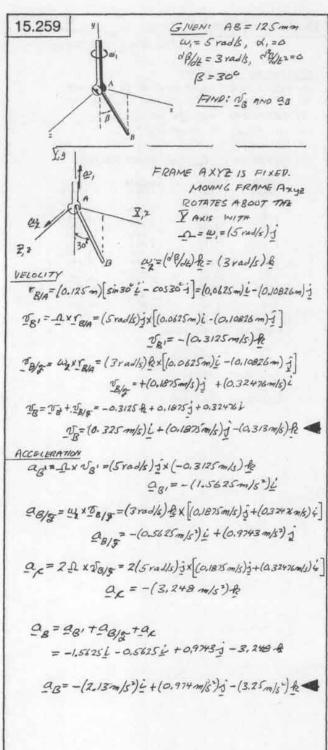
aris= w2 x vri= (3 rad/s) jx (-0.45 m/s) &= -(1.35 m/s) &=

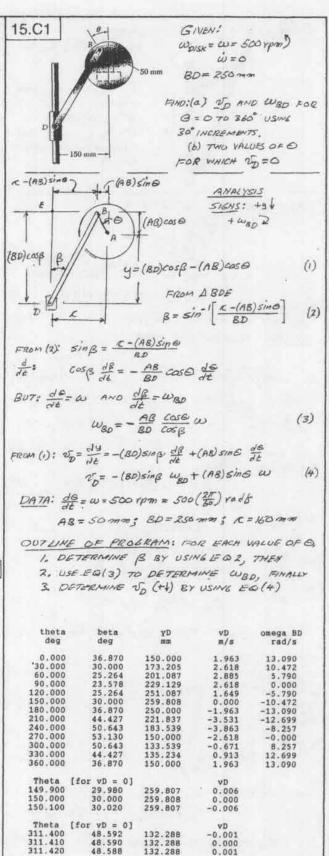
ar= 2-1 x vri= = 2(3 rad/s) jx(-0.45 m/s) &= -(2.7 m/s) &=

 $\alpha_F = \alpha_{F1} + \alpha_{F/pp} + \alpha_{L}$ $= -(2.7 m/s^{*}) \dot{\nu} - (1.35 m/s^{*}) \dot{\nu} - (2.7 m/s^{*}) \dot{\nu}$ $\alpha_F = -(6.75 m/s^{*}) \dot{\nu}$





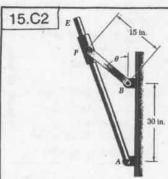




48.588

132,288

0.001



GIVEN: Wap=6 rad/s), Kap=0

FINO:

(1) WAE AND CLAE FOR

© = C TO 180° AT

15° INCREMENTS,

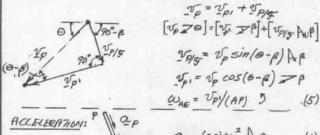
(2) (CLAE) minimum AND

CORRESPONDING VALUE

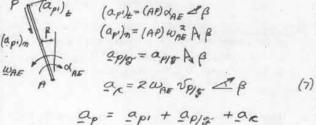
VELOCITY:

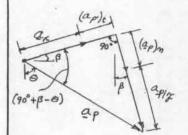
ROD BP: $\sqrt{p} = (BP)W_{BP} \neq \Theta$ (3)

ROD AF: $\sqrt{p} = (AP)W_{AF} \neq B$ (4)



ACLEUENTION: Q_p $Q_p = (g_p) W_{g_p}^2 = (6)$





RIGHT TRIMMICLE: $a_{R} + (a_{P})_{t} = a_{p} \cos(90^{\circ} + \beta - 6)$ $a_{R} + (AP)_{N} = a_{p} \cos(90^{\circ} + \beta - 6)$ $a_{R} = \frac{1}{AP} [a_{p} \cos(90^{\circ} + \beta - 6) - a_{R}]$ (8)

(CONTINUED)

15.C2 continued

DATA: Wap = 6 rad/s BP = 15 in.; A8 = 30 in

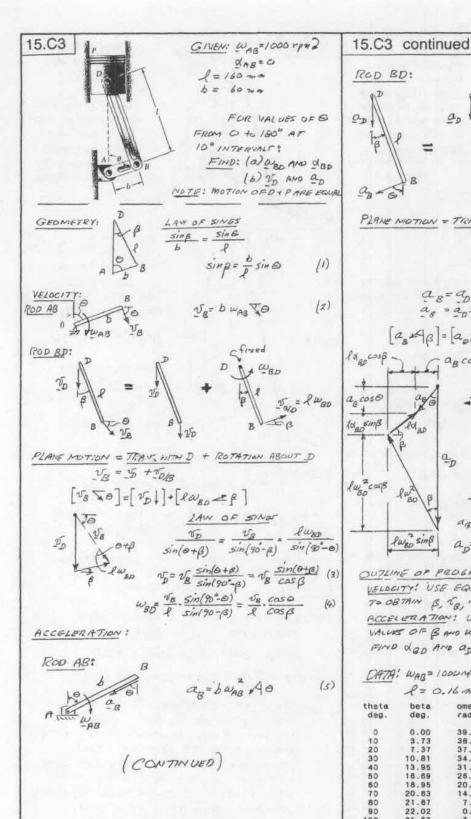
GUTLINE OF PROBRAM:

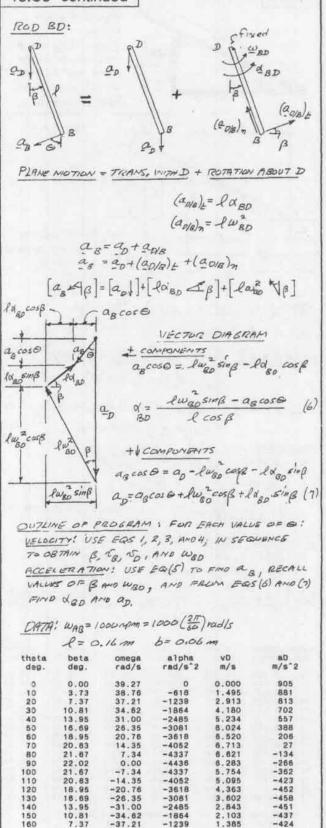
- 1. USE EQS, () AND(2) TO FIND & AND AP.
- 2. USE EQS. (3) AND (4) TO FIND Up AND Up
- 3. DETERMINE WAR BY USING EQ.(5).
- 4. USE EQ.(6) TO FIND OLD
- 5. USE EQ(2) TO FIND ax
- 6. DETERMINE DAE BY USING EQ.(8)

theta	beta	omegaAE	alpha
deg.	deg.	rad/s	rad/s^2
0	0.00	2.000	0.000
15	4.99	1.985	0.712
30	9,90	1.937	1.508
45	14.84	1.850	2,492
60	19.11	1.714	3.818
. 75	23.15	1.509	5.728
90	28.57	1.200	8.640
105	29.02	0.730	13.273
120	30.00	0.000	20.785
135	28.58	-1.144	32,388
150	23.79	-2.860	45.782
165	14.05	-4.920	43.298
180	0.00	-6.000	0.000
The second second second second		Printers II was a printer of the second	

theta for maximum alpha

theta deg.	alpha rad/s^2		
157.0800	48.58693		
157.0900	48.58694		
157.1000	48.58694		
157.1100	48.58693		





0,687

0.000

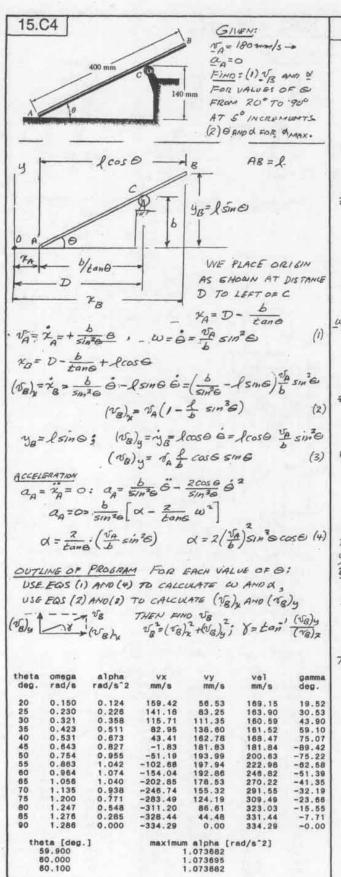
-415

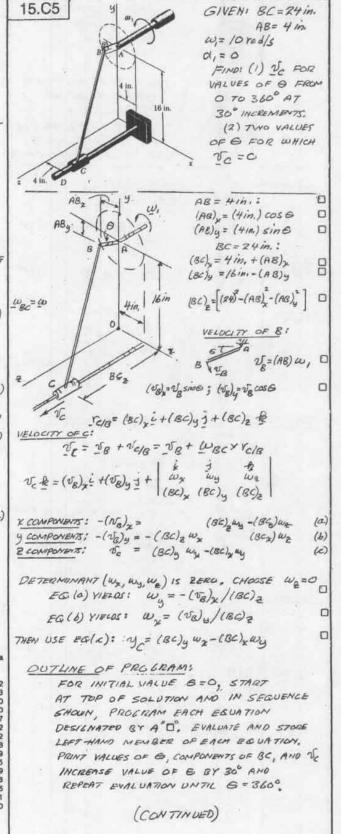
-411

180

0.00

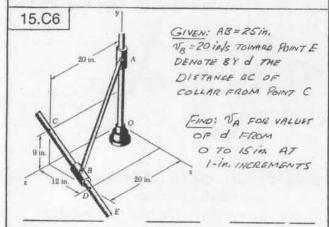
-39.27

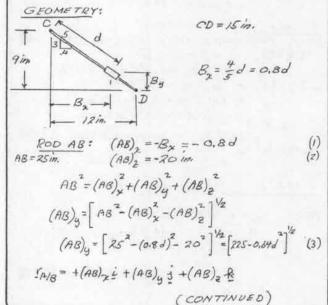


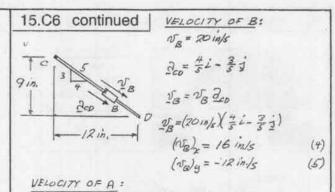


	Compo	nents of	rod BC	Velocity
theta	X	У	Z	of C
deg	in.	in.	in.	in./s
0.000	0.000	-16.000	16,000	40.000
30.000	7.464	-14,000	18.008	35,221
80.000	6.000	-12.536	19.567	23.436
90.000	4.000	-12.000	20.396	7.845
120.000	2,000	-12.536	20.368	-8.908
150.000	0.536	-14.000	19,486	-24.338
180.000	0.000	-18,000	17.889	-35.777
210.000	0.538	-18,000	15.865	-39.977
240.000	2.000	-19.464	13.898	-32.995
270.000	4.000	-20.000	12.649	-12.849
300,000	6.000	-19,464	12,694	14.293
330,000	7.464	-18.000	14.010	33.851
360.000	8.000	-16.000	16.000	40.000

Detern	ination of	values of	theta for v	C = 0
theta	Comp	onents of re	od BC	Velocity of C
104.034	3.030	-12.119	20.492	0.001
104.035	3.030	-12.119	20.492	0.001
104.036	3.030	-12.119	20.492	0.000
104.037	3.030	-12.119	20.492	-0.000
100	Components of rod BC			Velocity
theta	×	У	Z	of C
284.020	4.969	-19.881	12.492	-0.015
284.030	4.970	-19.881	12.492	-0.008
284.040	4.970	-19.881	12,492	0.003







$$V_{A} = V_{B} + V_{A/B} = V_{B} + \omega \times V_{A/B}$$

$$V_{A}j = (V_{B})j + (V_{B})j + \omega_{A} \quad \omega_{B} \quad \omega_{B}$$

$$(AB)_{X} \quad (AB)_{Y} \quad (AB)_{Z}$$

$$\begin{array}{lll} & \mathcal{I} & \textit{COMPONENTS}: & -(\tilde{V_B})_{\chi} = & +(AB)_2 w_3 + (AB)_4 w_4 & (a) \\ & \underline{\mathcal{I}} & \textit{COMPONENTS}: & \hat{V_A} - (\tilde{V_B})_{\chi} = -(AB)_2 w_{\chi} & +(AB)_2 w_{\chi} & (b) \\ & \underline{\mathcal{I}} & \textit{COMPONENTS}: & O = (AB)_{\chi} w_{\chi} - (AB)_{\chi} w_{\chi} & (c) \end{array}$$

DETERMINATE OF $(w_{\chi}, w_{\dot{\chi}}, w_{\dot{\chi}})$ is DERO. CHOOSE $w_{\chi} = 0$ EQ. (a) YIELOS: $w_{\dot{\chi}} = 0$ EQ. (a): $-(v_{\dot{\chi}})_{\chi} = 0 + (AB)_{\dot{\chi}} w_{\dot{\chi}}$ $w_{\dot{\chi}} = -(v_{\dot{\chi}})_{\chi}/(AB)_{\dot{\chi}}$ (6)

EG.(b):
$$N_A - (N_B)_y = O + (AB)_X \omega_Z$$

$$V_A = (N_B)_y + (AB)_X \omega_Z$$

(7)

OUTLINE OF PROGRAM:

FOR INITIAL VALUE d = 0, PROGRAM, IN

SEQUENCES EQUATIONS (1) THROUGH (7)

EVALUATE LEFT-HAND MEMBER OF EACH

EQUATION AND PRINT VALUES OF d, COMPONENTS OF $Y_{A|B}$, AND Y_{C} ,

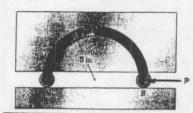
INCREASE VALUE OF d BY f-M. AND

REFEAT PROCESS UNTIL d = CD = 15 in

	Com	Velocity		
d	×	У	Z	VA
in.	in.	in.	in.	in/s
0.000	0.000	15.000	-20.00	-12.000
1.000	-0.800	14.979	-20.00	-12.855
2,000	-1.800	14.914	-20.00	-13.716
3.000	-2.400	14.807	-20.00	-14.593
4.000	-3.200	14,655	-20,00	-15.494
5.000	-4.000	14.457	-20.00	-16.427
6.000	-4.800	14.211	-20.00	-17.404
7.000	-5.600	13.915	-20.00	-18.439
8.000	-8.400	13.568	-20.00	-19.548
9.000	-7.200	13.159	-20.00	-20.754
10.000	-8.000	12.689	-20.00	-22.088
11.000	-8.800	12.147	-20.00	-23,591
12.000	-9.600	11.528	-20.00	-25.327
13.000	-10.400	10.809	-20.00	-27.394
14.000	-11.200	9.978	-20.00	-29.960
15.000	-12.000	9.000	-20.00	-33.333

16.1 and 16.2

GIVEN: W=316



PROBLEM 16.1:

FOR P=516,

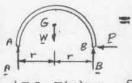
FIND: (a) Q.

(b) REACTIONS,

PROBLEM 16.2:

FOR A=0,

FIND: (a) P, (b) a.



$$\pm \sum F_x = \sum (F_x)_{per} : P = m\bar{a} \qquad (1)$$
+) $\sum M_n = \sum (M_n)_{per} : B(2r) - Wr = m\bar{a} \left(\frac{2r}{n}\right) \qquad (2)$
+† $\sum F_n = \sum (F_n)_{per} : A + B - W = 0 \qquad (3)$

PROBLEM 16.1: P = 51b, W = 31b, m = W/9 $EQ(1)^{1}$ $P = (W/9) \bar{a}$ $a = (P/W)g = \frac{57b}{31b}(32.274/5^{2})$ $\bar{a} = 53.1774/5^{2}$ $\bar{a} = 53.774/5^{2}$

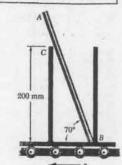
EQ(2): $B(2r) - Wr = \frac{W(\frac{p}{3})(\frac{2r}{\pi})}{3(316) + \frac{516}{\pi}}$ $B = \frac{1}{2}W + \frac{p}{\pi} = \frac{1}{2}(316) + \frac{516}{\pi}$ B = 3.09216 B = 3.0916

EQ(3): A+3.0924-316=0 A=-0.0924 A=0.09216+

PROBLEM 16.2: A=0, W=316, m=4/9

 $E_{G}(z)$: $O-Wr = \frac{W}{g}\bar{a}(\frac{2r}{\pi})$ $\bar{a} = \frac{\pi}{2}g = \frac{\pi}{2}(32.2ft^{2})$ $\bar{a} = 50.58 R/s^{2}$ $\bar{a} = 50.6ft/s^{2}$

EQ(1): $P = \frac{W}{9} \bar{a}$ $P = \frac{W}{9} (\frac{\pi}{2}9) = \frac{\pi}{2} W = 4.71216$ P = 4.7116 - 4.7116 16.3 and 16.4



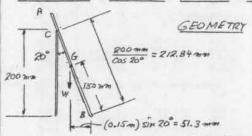
GIVEN:

Roo: m = 2.5 kg, AB = 300 mm

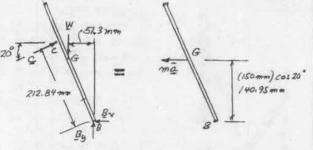
PROBLEM 16.3: FOR a=1.5m/s' = FINO: (a) C, (b) B

PROBLEM 16.4:

FIND: a max FOR ROD
TO REMAIN IN POSITION



W= mg = (2.5 Rg) 9.81 m/s = 24.525 N



+) $\sum M_B = \overline{2}(M_B)_{HF}$; C(212.84 mm) - |M(51.3 mm) = -ma(140.95 mm) C = 0.241 W - 0.6622 ma C = 0.241(24.525 N) - 0.6622(2.52a)(a)C = 5.911 N - 1.656 a (1)

PROBLEM 16.3: a = 1.5 m/s²

EQ(1): C = 5.911 - 1.656 (1.5); C = 3.43 N 2 20°

+1 \(\Sig = \Sig \) \(\text{Pi} \) : By - W + Csin 20° = 0

\(By = 24.525 N - (3.434) \) \(\text{sin 20} \) = 23.35 N \(\text{T} \)

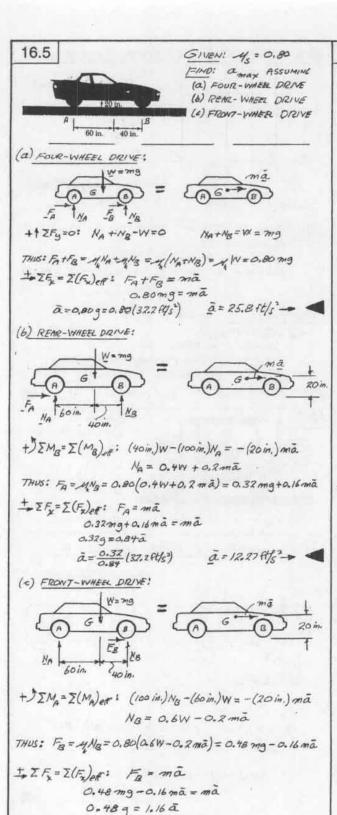
\(\frac{1}{2} \) \(\frac{1} \) \(\frac{1}{2} \) \(\frac{1}{2}

B R 23,35 N B = 24,4 N \$ 73.4°

PROBLEM 16.4: For amay, C=0

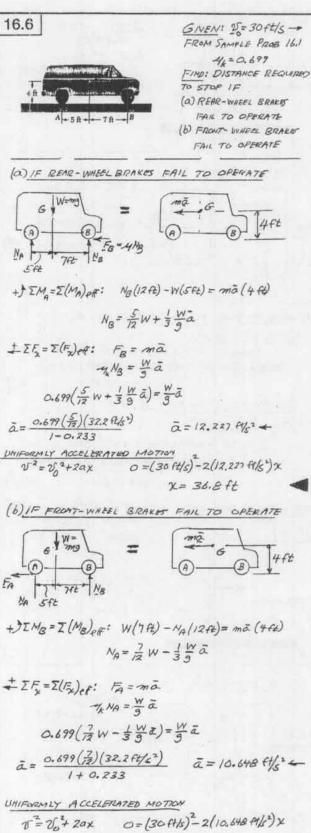
EG(1) $C = 5.911 \, \text{N} - 1.656 \, \text{a}$ $0 = 5.911 \, \text{N} = 1.656 \, \text{a} \, \text{max}$ $a_{\text{max}} = 3.57 \, \text{m/s}^2$

amay 3,57 m/s2

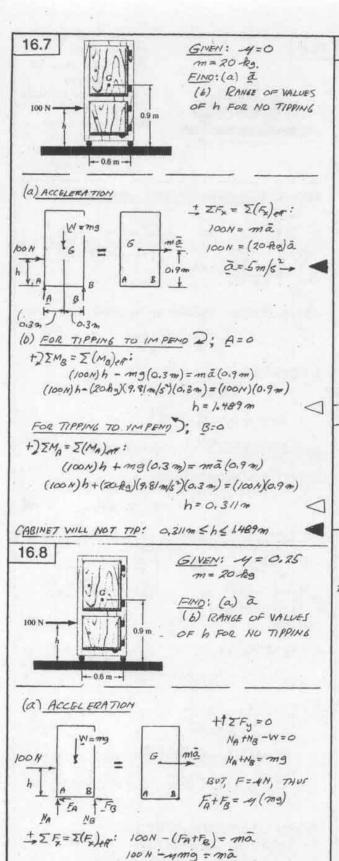


ā = 0.48 (32,2 FY/3)

a= 13.32 ft/s2 -



x= 42.3 ft

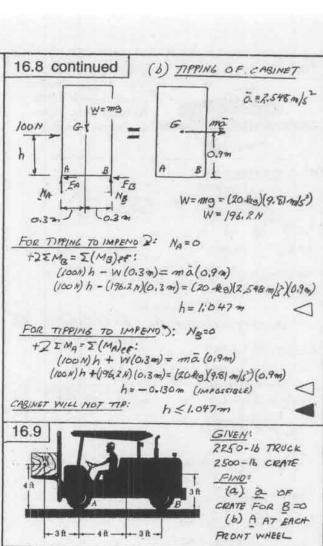


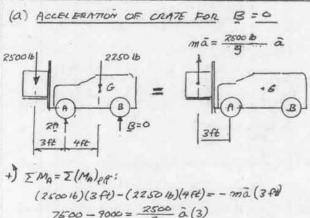
100 N-0.25 (20kg)(9.8/m/s2) = (20kg)a

(CONTINUED)

a= 255 m/s'->

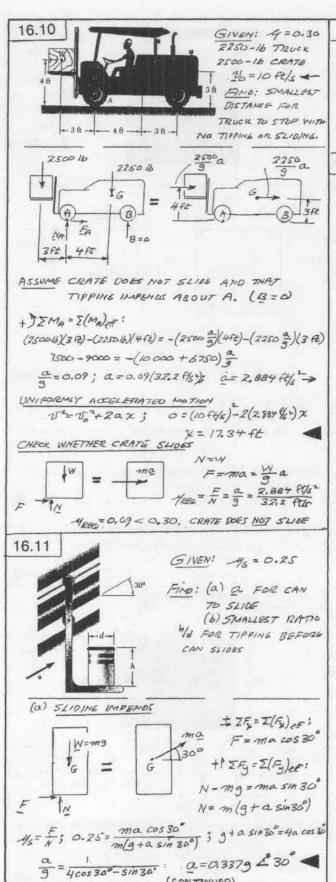
a= 2.548 m/s



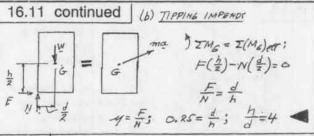


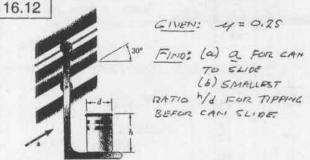
+) $\sum M_A = 2 (M_A)_{eff}$: (250016)(3ft) - (225016)(4ft) = -ma(3ft) $7500 - 9000 = \frac{2500}{9}a(3)$ $a = \frac{1}{5}g = \frac{1}{5}(32.2ft)_{5}^{2}$ a = 6.44ft

(b) REACTION AT A; + | IFy = I/F)eff: 2A - 2500/b - 2250/b = ma 2A - 4750/b = 2500/b (9/5) 2A = 5250/b FOR ONE WHEEL: A = 2625/b

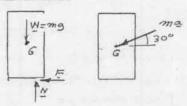


(CONTINUED)





(a) SLIDING IMPENDS:

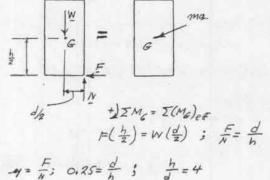


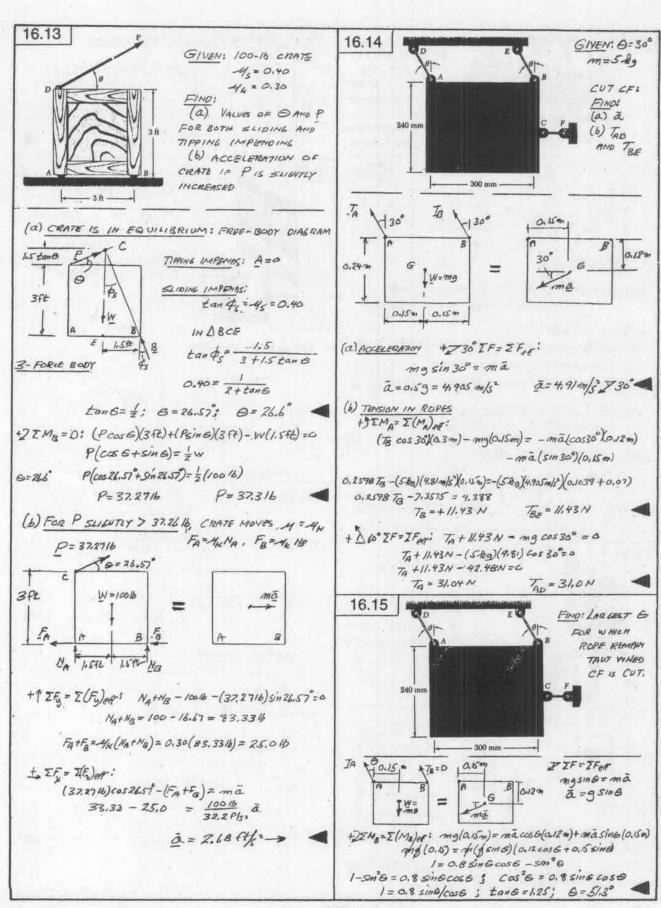
= IFx = I(Fx) = = ma cos 30° + 1 EFy = I (Fy)ex: N-mg = -ma singco N=m(g-asin30°) -45 = # 1 0.25 = macos 300 m(g-a sin 30) 9-a sin 30 = 4 a cos 30°

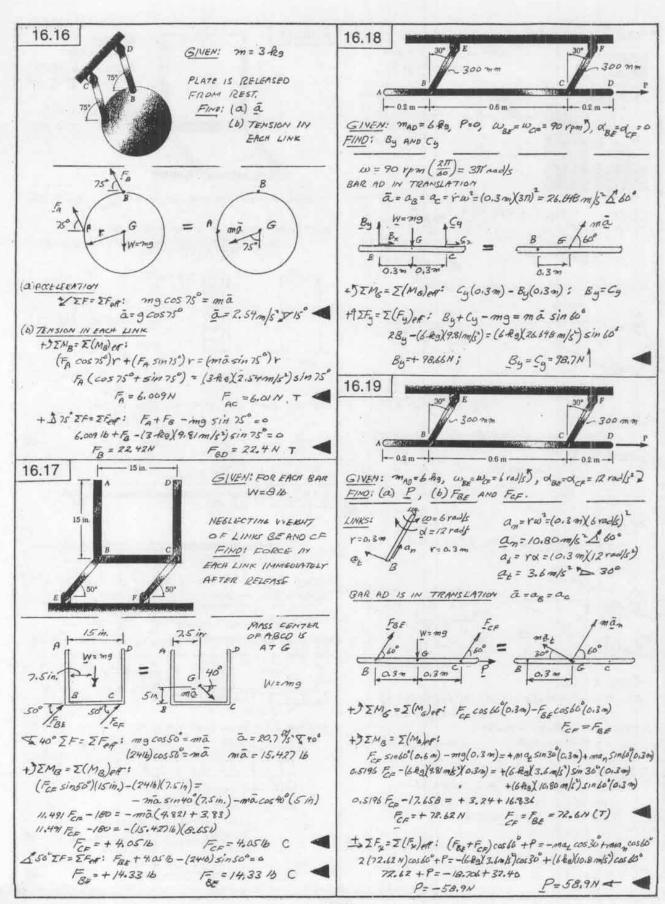
$$\frac{a}{5} = \frac{1}{4\cos 30^{\circ} + \sin 30^{\circ}} = 0.252$$

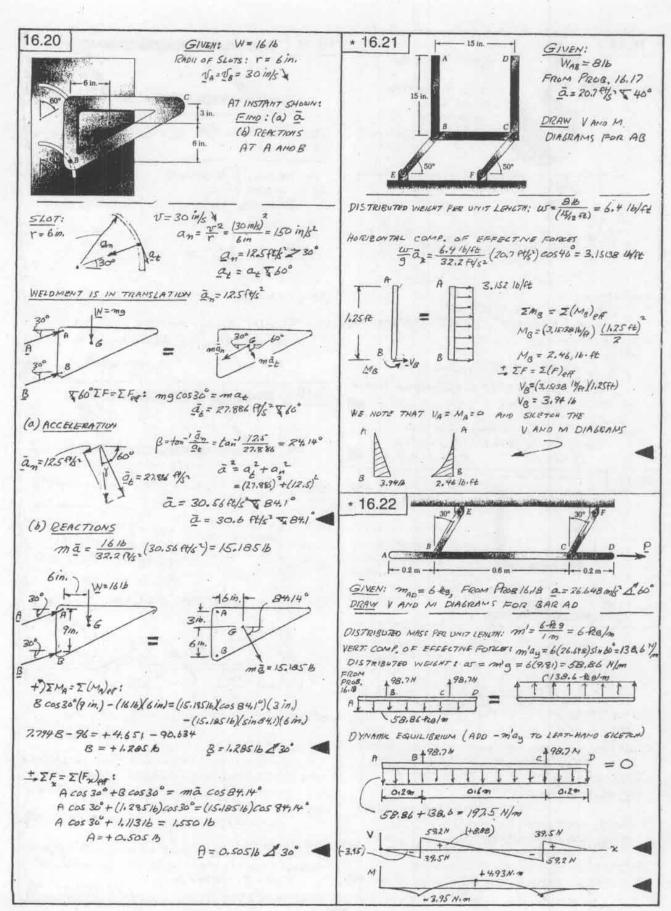
$$a = 0.2529$$

(b) TIPPING IMPENDS;











FOR TRANSLATION

SHOW THAT EFFECTIVE

FORCES ARE (DID) A

ATTACHED TO PARTICLES

AND ARE REDUCE TO

M.A. ATTACHED AT G.

SINCE SLAB IS IN TRANSLATION, EACH PARTICLE HAS SAME ACCEPTATION AS G, NAMELY Q. THE EFFECTIVE FORCES CONSIST OF (DM) à.



THE SUM OF THESE VECTORS IS: $\Sigma(\Delta m_i)\tilde{a} = (\Sigma \Delta m_i)\tilde{a}$ OR SINCE $\Sigma \Delta m_i = m$, $\Sigma(\Delta m_i)\tilde{a} = m\tilde{a}$ THE SUM OF THE MICHIENTS ABOUT 6 IS: $\Sigma \Sigma_i \times (\Delta m_i)\tilde{a} = (\Sigma \Delta m_i \Sigma_i) \times \tilde{a}$ (1)

BUT, I Dome of Action PASSET THROUGH & AND
THAT IT MAY BE ATTACHED AT 6.

16.24



FOR CENTROIDAL ROTATION,

SHOW THAT EFFECTIVE

FOICES CONIST OF VECTOUS

-(DM; |WY; AND (DM;)(XXY!)

ATTACHED TO PARTICLES AND

REDUCE TO A COUPLE IN.

FOR CENTROPAL ROTATIONS a = (91)+ (91) = Xxx. - 102x.

EFFECTIVE FORCES ARE: (Am) a = (Am)(Axx) - (Am) w'x.

 $(\Delta m_i)(o \times v_i)$ $(\Delta m_i)(o \times v_i)$ $= (G \cdot)$ $G \cdot)$

$$\begin{split} & \sum (\Delta m_i) a_i = \sum (\Delta m_i) (\omega \times r_i^*) - \sum (\Delta m_i) \omega^2 r_i^2 \\ & = \alpha \times \sum (\Delta m_i) r_i^2 - \omega^2 \sum (\Delta m_i) r_i^2 \end{split}$$

SINCE G IS THE MASS CENTER, \$ [0mi) Y' = 0 . EFFECTIVE FORCES REDUCE TO A COUPLE,

SUMMINE MONIENTS ABOUT G

$$\begin{split} & \mathbb{E}\left(Y_i' \times \Delta m_i \, a_{i'}\right) = \mathbb{E}\left[Y_i' \times (\Delta m_i)(a \times Y_i')\right] - \mathbb{E}\left[Y_i' \times (\Delta m_i) u^2 Y_i'\right] \\ & \mathcal{B}u7, \ T_i' \times (\Delta m_i) u^2 T_i' = u^2 \left(\Delta m_i\right) \left(Y_i' \times Y_i'\right) = G \\ & \mathcal{A} \times \mathcal{D}, \ Y_i' \times (\Delta m_i)(a \times Y_i') = \left(\Delta m_i\right) Y_i'^2 \times \mathcal{D} \\ & \mathcal{D} \times \mathcal{D} \times \mathbb{E}\left[Y_i' \times \Delta m_i \, a_{i'}\right] = \mathbb{E}\left(\Delta m_i\right) Y_i'^2 = \left[\mathbb{E}\left(\Delta m_i\right) Y_i'^2\right] \\ & \mathcal{D} \times \mathcal{D} \times \mathbb{E}\left[Y_i' \times \Delta m_i \, a_{i'}\right] = \mathbb{E}\left(\Delta m_i\right) Y_i'^2 = \mathbb{E}\left[X_i' \times \mathcal{D} m_i' \, a_{i'}\right] \\ & \mathcal{D} \times \mathcal{D} \times \mathbb{E}\left[X_i' \times \Delta m_i' \, a_{i'}\right] = \mathbb{E}\left(\Delta m_i' \, a_{i'}\right) \\ & \mathcal{D} \times \mathcal{D} \times \mathbb{E}\left[X_i' \times \Delta m_i' \, a_{i'}\right] \\ & \mathcal{D} \times \mathcal{D} \times \mathcal{D} \times \mathcal{D} \times \mathcal{D} \times \mathcal{D} \\ & \mathcal{D} \times \mathcal{D} \times \mathcal{D} \times \mathcal{D} \times \mathcal{D} \times \mathcal{D} \times \mathcal{D} \\ & \mathcal{D} \times \mathcal{D}$$

 $\frac{1}{2} \left(Y_{i, x} \Delta m_{i, \alpha_{i}}^{*} \right) = \frac{1}{2} \left(\Delta m_{i, \beta_{i}}^{*} \right) \left(\Delta m_{i, \beta_{i}}$

THE MOMENT OF THE COUPLE IS IN

16.25 FLYWHEEL: W=6000 16 &=36 in.

AT t=0, W=300 PPM, AT t=10 MIN, W=0

FIND COURLE DUE TO KINETIC FRICTION, (UNIF. ACCEL. MOTIO)

$$\begin{split} \vec{I} &= m \cdot \hat{R}^2 = \left(\frac{6000 \text{ lb}}{32.2 \text{ eV/s}^2}\right) (3 + t) = 1677.0 \text{ lb·fl·s}^2 \\ \omega_0 &= 300 \text{ rpm} \left(\frac{2\pi}{60}\right) = 10 \text{ Trad/s} \\ \omega &= \omega_0 + \text{dt}; \quad 0 = 10 \text{ Trad/s} + \alpha \left(600 \text{ s}\right) \\ \alpha &= -0.05 \text{ 236 rad/s}^2 \end{split}$$

M=Ix=(1677 16.ft.s2) 0.5236 ** (=)=87.81 16.ft
M=67.816.ft

16.26 ROTOR: m=50 Rg, R= 180 mm

FRICTION COUPLE: M = 3.5 N·m

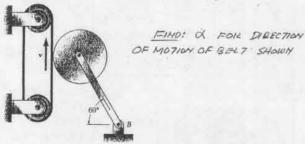
6=0, W6=3600 mm (UNIF. ACCEL. MOTION)

FIND: REVOLUTIONS AS ROTOR COASTS TO REST

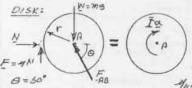
$$\begin{split} \tilde{I} &= m\tilde{R}^2 = (50 \, \text{Rg})(0.180 \, \text{m})^2 = 1.620 \, \text{Rg} \cdot \text{m}^2 \\ M &= I \, \text{M} : \quad 3.5 \, \text{N} \cdot \text{m} = (1.620 \, \text{Rg} \cdot \text{m}^2) \, \text{d} \\ \text{M} &= 2.1605 \, \text{rad/s}^2 \, \left(\text{DECULETIATION} \right) \\ W_0 &= 3600 \, \text{rpm} \left(\frac{2\pi}{60} \right) = 120 \, \text{Trad/s} \\ W^2 &= 4 \, \text{d} + 2 \, \text{d} \, \text{G} : \quad 0 = (120 \, \text{Trad/s})^2 + 2 \left(-2.1605 \, \text{rad/s}^2 \right) \, \text{G} \\ \Theta &= 32.811 \, \text{M} \, \text{rad} \left(\frac{1 \, \text{rev}}{2\pi \, \text{rad}} \right); \quad \Theta &= 5234.8 \, \text{rev} \\ \Theta &= 5730 \, \text{rev} \end{split}$$

16.27

GNEN: 4/4 = 0.40



BELT: UT F N F= MAN

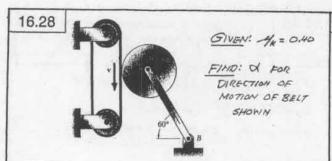


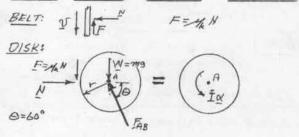
 $\frac{1}{N} \sum_{k=1}^{N} \sum_{k=1}$

MKH + FASSING -mg = 0 FASSING = mg - MKH (2)

 $\frac{EO(2)}{FQ_{i}(1)}: \tan \theta = \frac{mg - 4kN}{N}$ $N = \frac{mg}{\tan \theta + 4k}; N = \frac{mg}{\tan \theta + 4k}; F = 4kN = \frac{mg}{\tan \theta + 4k}$ FABSING = mg - 4kN (2) N = Mg = Mg + 4kN (2) FABSING = mg - 4kN (2) FABSIN

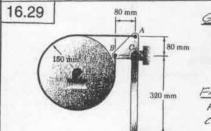
 $d = \frac{r}{I} F = \frac{r}{\frac{1}{2}mr^2}, \frac{mg \, 4\kappa}{\tan \theta + 4\kappa} = \frac{28}{r} \cdot \frac{4\kappa}{\tan \theta + 4\kappa}$





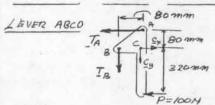
$$+\sum F_{\chi} = \Sigma(F_{\chi})_{eff}: N - F_{AB}\cos G : F_{AB}\cos B = N$$
 (1)
+\ \Sigma F_{\text{g}} = \Sigma(F_{\text{b}})_{eff}: F_{AB}\sigma \sigma \n \text{g} - m_{\text{g}} - m_{\text{g}} \n \text{g} \rightarrow 0

$$\alpha = \frac{Z(9.81 \text{ m/s}^2)}{0.18 \text{ m}} \cdot \frac{0.40}{\tan 60^\circ - 0.40}; \quad \alpha = 32.7 \frac{\text{mod } 5}{5^2}$$



GIVEN: I = 75 Ag. m2
P = 100 N
4k = 0.25
Wo = 240 rpm)

FIND: TIME REQUIRED FOR DISK TO COME TO REST



STATIC EQUILIDIRIUM!

(CONTINUED)

16.29 continued

THUS & WILL BED

(3)

+)
$$IM_{g} = \Sigma(M_{g})_{eff}$$
: $T_{g} r - T_{A} r = \bar{I} \propto$

$$T_{g} - T_{A} = \bar{I} \propto \qquad (z)$$

BELT FRICTION:

$$\beta = 270^{\circ} = \frac{3}{3}\pi$$
 and $\frac{T_{8}}{T_{A}} = e^{4\kappa}\beta = e^{(0.25)\frac{3}{2}\pi}$ | 1/18 | 3.748

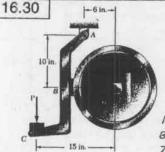
TB = 3.248 TA

$$EG(i)$$
: $T_A + T_B = 400N$; $T_A + 3.248$ $T_A = 400 N$
 $T_A = 94.16N$ $T_B = 3.248(94.16N) = 305.9N$
 $EQ.(2)$: $T_B - T_A = \frac{1}{r}$ d ; $305.9N - 94.16N = \frac{75.80m^2}{0.15m}$ d

UNIF. ACCEL. MOTION

08=at: 877 nad/s = (0.423 Yad/s') t; t=59.45

NOTE: IF & IS REVERSED THEN TA AND TO ARE INTER CHANGED. THIS CAUSEN NO CHANGE IN EG.(1) AND EG(2). THUS FROM EG(3), & IS NOT CHANGED.



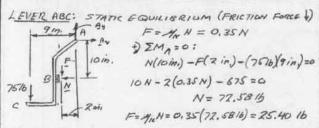
I = 14 W. Ft 52

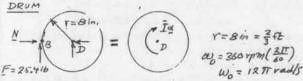
- 1/2 = 0.35

P = 76 16

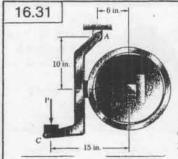
Wo = 360 rpm)

FINO: NUMBER OF
REVOLUTIONS OF DRUM
REFORE IT COMES
TO REST



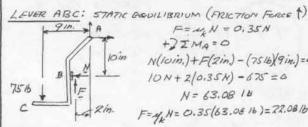


+)
$$\sum M_b = \sum (M_o)_{eff}$$
: $Fr = \sum M_o$
 $(25.416)(\frac{2}{3}R) = (1416.46.5^2)M$
 $M = 1.2097 \text{ rad/s}^2$ (DECELERATION)
 $W^2 = W_o^2 + 2M\Theta$; $O = (1217 \text{ rad/s})^2 + 2(-1.2097 \text{ rad/s}^2)\Theta$
 $G = 587.4 \text{ rad}$
 $G = 587.4 \text{ rad}$
 $G = 587.4 \text{ rad}$



GIVEN: I = 14 16. ft. 52 9 = 0.35 P=7516 Wo = 360 rpm 2

FIND: NUMBER OF REVOLUTIONS OF DRUM BEFORE IT COMES TO REST.



F= MN = 0,35N + IIMA = O N(10in.)+F(2in.)-(7516)(9in.)=0 10N+2/0,35N)-675=0 N= 63.08 14 F=4, H= 0.35(63.08 16)=72.0816

DRUM: F= 22,08

r=811.===# W=360rpm(211) w=12 Tinad/c

+) IMO = EMO) == 0 Fr= IX (22.68 1b)(= ft)= (14.16.ft.5) X $X = 1.5015 \text{ rad/s}^2 (DECELERATION)$ $0 = (1271 \text{ rad/s})^2 + 2(-1.5015 \text{ rad/s}^2) \Theta$ W= W2+200; 6 = 675.8 rad 8 = 675.8 nad (27) = 10256 rev; 0 = 107.6 rev

16.32 r=500 m m=15-hg>

a= (15-85)(9.8/m/s2)

GIVENI FLYWHEEL M= = 120 Rg R= 375 mm V=0 AT 5=0 FIND: (a) a of BLOCK. (b) & AFTER IT HAS MOVED 1.5m.

(CONTINUED)

KINEMATICS KINETICS +) IMB = I (MB) + A: $(mg)r = \widehat{I} \propto + (m_A a)r$ mgr = mg to (a)+mar $m_A + m_E \left(\frac{k}{E}\right)^2$

15-kg+(120-kg)(375mm)2=1,7836 m/s

16.32 continued (a) a=1.7836m/52/ $\alpha = \frac{\alpha}{r} = \frac{1.7836 \text{ m/s}}{0.5 \text{ m}} = 3.567 \text{ rad/s}^2$ a = 3.57 rad/s 2 (b) VA = V + Zas FOR 5=1.5m: V==0+2(1.7836m/5)(1.5m) 25=2.3/m/s + VA = 2.3/3m/s

16.33

GIVEN: SYSTEM RELEASED FROM REST! 1. IF ma = 12 Ag, BLOCK FALLS 3 m IN 4.65 2. IF mp = 24-kg, BLOCK FALLS 3mm IN 3.15 ASSUME CONSTANT ME DUE TO AXLE FRICTION. FIND: I

KINEMATICS or d= a

KINETICS W=mA3 +) EMB = I(MB) eA: (madr-Mr= Ix+(ma)r magr-Mg= I a + mar

CASE 1: 4=3m, == 4.65 y= \frac{1}{2}a +2; 3m= \frac{1}{2}a (4.65)^2; a=0.7836 m/s2 mp= 12-29

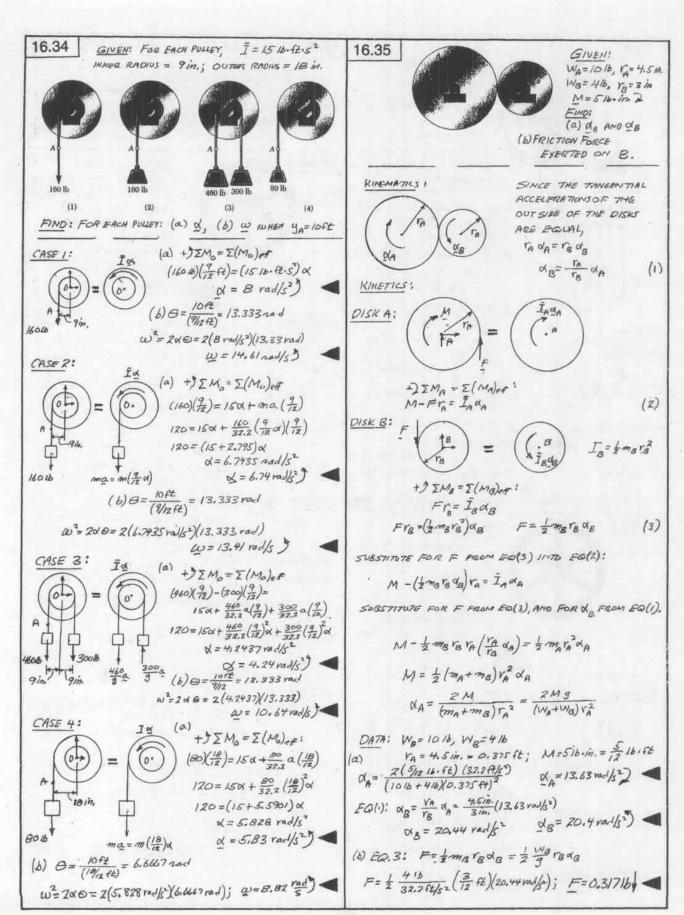
SUBSTITUTE INTO EQ(i) (12-20) (12-20) (0,6m) - Mg = I(0,2836 m/s) + (1260)0,2836 m) (0.60 70.632 - MI = I (0.4727) + 2.0419 (2)

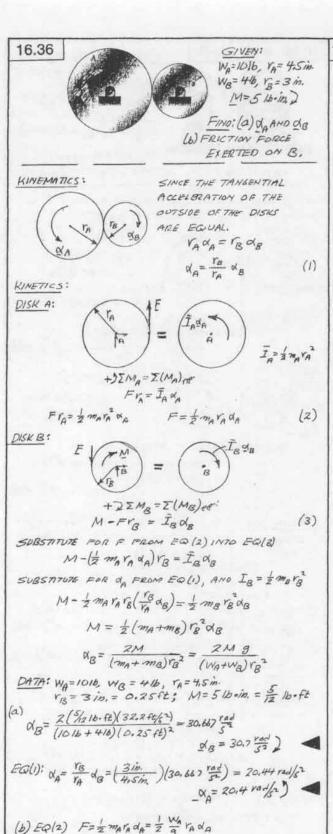
CASE 2: 4=300, 2=3.15 y= \frac{1}{2}at2; 3m = \frac{1}{2}a(3.15)2; a = 0.6243 m/s2 mp = 24-Rg SUBSTITUTE INTO EQ(1):

[24.20](9.8/m/52)(0.6m) - Mg = I (0.6243 m/52) + (24kg)(0.6243 m/52)(0.62) 141.764 - Mr = I (1.0406) + 8.9899 (3)

SUBTRACT EQ(1) FROM EQ(2), TO ELIMINATE M. 70.632 = I(1.0406 - 0.4727) + 6.948 63.684 = I (0.5679) I= 112.14 kg.m

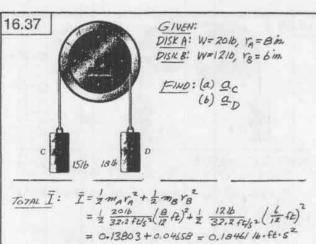
I = 112.1 kg·m

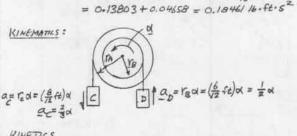


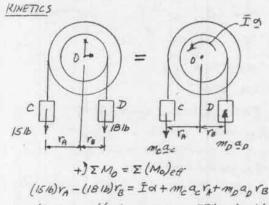


F = 1 (1016) (4.5) (4.5) (20.44 vol/s2) = 1.19016

FRICTION FORCE ON DISILB: F=1.19016.

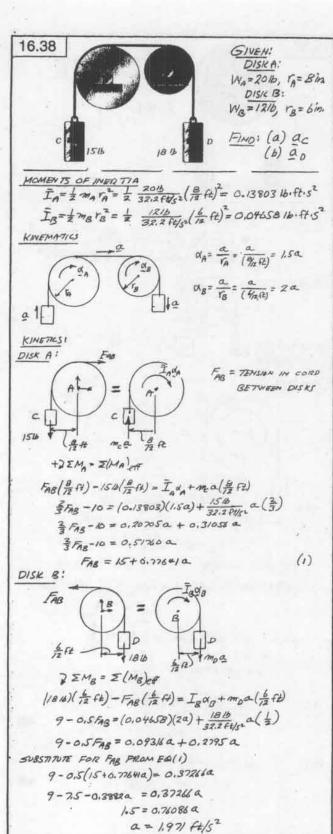






 $\begin{aligned} &(1516)V_{A} - (1816)V_{B} = \bar{1}d + m_{c}a_{c}V_{A} + m_{D}a_{D}V_{B} \\ &(1516)(\frac{9}{72}f_{c}) - (1816)(\frac{6}{12}f_{c}) = 0.18461d + \frac{11516}{32.24f_{c}}(\frac{2}{3}d)(\frac{9}{12}f_{c}) \\ &+ \frac{1816}{32.24f_{c}}(\frac{1}{2}d)(\frac{6}{72}f_{c}) \end{aligned}$

- (a) $a_c = \frac{2}{3}\alpha = \frac{2}{3}(1.8878 \text{ rad/s}^2)$ $a_c = 1.255 \text{ fH/s}^2$
- (b) $a_D = \frac{1}{2}d = \frac{1}{2}(1.8818 \text{ red/s}^2)$ $a_D = 0.941 \text{ ft/s}^4$



BOTH OF AND OF HAVE THE SAME MAGNITUDE

ac=1.971 A/52

ap= 1.971 ft/s=

16.39 and 16.40

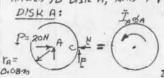


GIVEN: $m_A = 6 + kg$; $m_B = 2 + kg$ N = 70 N, $M_K = 0.15$ $\frac{P_{ROBLEM 16.39}}{(W_A)_0}$; $(W_B)_0 = 0$ $\frac{P_{ROBLEM 16.40}}{(W_{A})_0}$; $(W_B)_0 = 0$ $\frac{P_{ROBLEM 16.40}}{(W_{A})_0}$; $(W_B)_0 = 360 rpm$

FOR EACH PROBLEM!

FIND: (a) AA AMD ag. (b) FINAL VELOCITIES WA AMD WB

WHILE STIPPING OCCURS, A PRICTION FORCE FT IS
APPLIED TO DISK A. AND FT TO DISK B.



 $I_{A} = \frac{1}{2} m_{A} v_{A}^{2}$ $= \frac{1}{2} (6 + 6) (0.08 m)^{2}$ $= 0.0192 + 20 m^{2}$ $\Sigma F_{a}^{a} N = P = 20 N$

+) $\sum M_{A} = \sum (M_{A})_{eff}$: $Fr_{A} = \bar{I}_{A} \alpha_{A}$ $(3N)(0.08m) = (0.0192 \text{ kg} \cdot m^{2}) \alpha_{A}$

0/A= 12.5 rad/s

(XA = 12.5 rad/s)



Isa's

 $\vec{I}_{g} = \frac{1}{2} m_{g} r_{g}^{2}$ $= \frac{1}{2} (3 h_{g}) (0.06 m)^{2}$ $= 0.005 + h_{g} \cdot m^{2}$

+) $\sum M_B = \sum (M_B)_{eq}$: $F_e = \sum_{B} \alpha_B$ $(3N)(0.06m) = (0.0054 - 29.m^2) \alpha_B$ $\alpha_B = 33.33 \text{ rad/s}^2$ $\alpha_B = 33.3 \text{ rad/s}^2$

PROBLEM 16.39: (WA) = 360 rpm (21) = 129 rad (2); (WB) = 0

DISKS WILL STOP SLIDING, WHEN SC = SOI, THAT IS

WHEN WAYA = WAYB

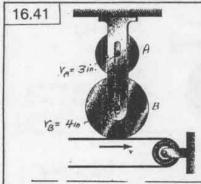
+) $\omega_{A} = (\omega_{A})_{0} - \sigma_{A}t = 12\pi - 12.5(1.0053) = 25.132 \text{ rad/s}$ $\omega_{A} = 25.132 \text{ rad/s} \left(\frac{60}{20}\right) = 240 \text{ yrm} \quad \omega_{A} = 240 \text{ yrm} \quad$

4) $W_B = M_B t = (33.33)(1.00531) = 33.507 \text{ rad/s}$ $W_B = 33.507 \text{ rad/s}(\frac{60}{2\pi}) = 320 \text{ rpm}$ $W_B = 320 \text{ rpm}$

PROBLEM 16.40: $(w_A)_0 = 0$; $(w_B)_0 = 360 \, r_F m (\frac{217}{25}) = 1217 \, \frac{1}{25}$ SLIDNG STOPS WHEN $V_C = V_{C'}$ THAT IS WHEN $w_A Y_A = w_B \, r_B$ $(\alpha_A t) \, Y_A = [(w_B)_0 - \alpha_B t] \, Y_B$ (12.5 t) (0.08) = (12.77 - 33.33 t) (0.06)t = 2.26195 - 2t; t = 0.2539BS

+) $W_A = W_A t - (12.5)(0.75398) = 9.4248 \text{ rad/s}$ $W_A = 9.4248 \text{ rad/s}(\frac{60}{20}) = 90 \text{ rpm}$ $W_A = 90 \text{ rpm}$

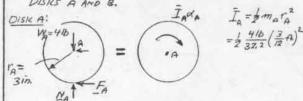
+) $W_8 = (w_8)_0 - \alpha_B t = 1277 - (33,33)(0.75398) = 12.569 mod/s$ $W_8 = 12.569 rad/s (\frac{60}{2\pi}) = 120 \text{ rpm}$ $W_R = 120 \text{ rpm}$



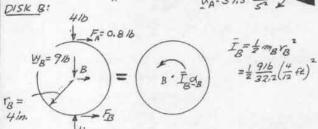
GIVEN Wa= 416 WB = 916 4 = 0.20 AT ALL SURFACES

FIND: INITIAL ANGULAR ACCELERATION OF EACH DISK

ASSUMB THAT SUPPING OCCURS BETWEEN DISKS A AND B.



AZF= ZFAF: N= 416 F=4x N=0,2(416)=0,816 +2 IMA = E (MA) AF: Fra= InaA $(0.8/b)(\frac{3}{12}ft) = \frac{1}{2} \frac{4/b}{32.2}(\frac{3}{12}ft)^2 \times_A$ 0 = 51,52 rad/s2 0A=51.5 mad 2

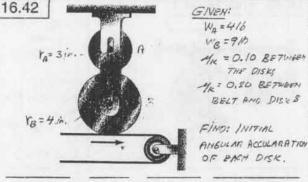


1 IF = IFM: NB = 4+9= 1316 FB = 1/K NB = 0.20 (13 16) = 2.616

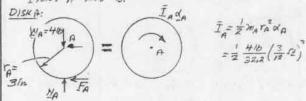
+) IMB = I(MB)AT: (FB-FA) rB = IB XB (2.616-0.816)(4/2 ft)= 1 916 (4/2 ft) ×B a = 38.64 rad/s2 d = 38.6 rad/s2)

KINEMATICS! a=51.52 rad/57 WE CALCULATE THE TANGENTIAL COMPONENTS OF POINTS OF CONTACT (Qc) E (ac)=1AdA= (32 12)(51,52 red/s) = 12.88 A/62 (ac) (Cc) = YBOB = (# 12 12)(38,64 rad/5) = 12.88 AL/52

0 B = 38,64 rad/52 WE FIND THAT SLIPPING DOES NOT OCCUR BETWEEN DISKS. BUT SINCE (Qc)= (Qc) & SLIPANG IMPENOS AND THAT FA=4KH==0.816 PND ABOVE RESULTS AFTE VALID.

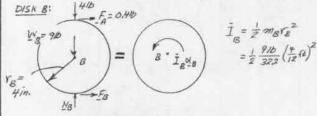


ASSUME THAT SLIPPING OCCURS RETWEEN [ISKS A AND B.



+ \$ SF = S(F) ef : NA = 416 Fa = MN = 0.1 (416) = 0,416 +DEMA=ElMAlagi FAYA = InaA (0.416)(3 st) = 1 416 (3 17) VA

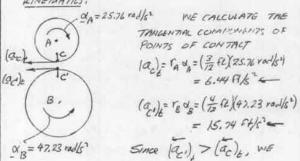
da= 25.76 10.16 NA= 25.8 rad/6-)



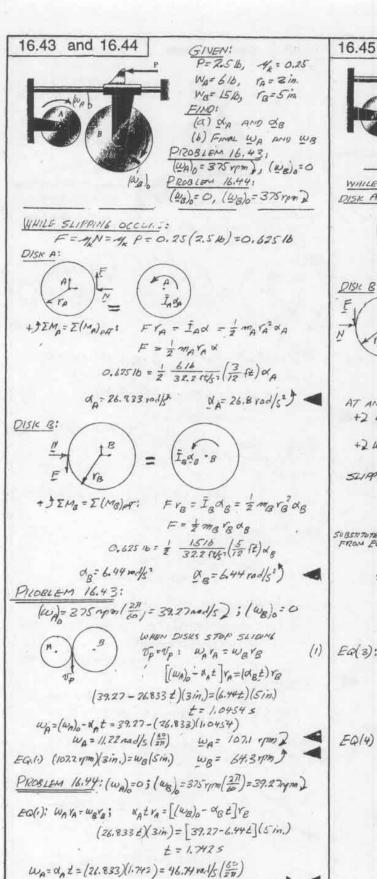
IF= ZFAT: NB=4+9=1316 FB = MR NB = 0.20 (1316) = 2.6 16

+) I MB = I (MB) AT: (FB-FA) YB = IB XB (2.616-0.416)(4 in.) = 1 916 (4 in.) XE Op= 47.23 rad/52

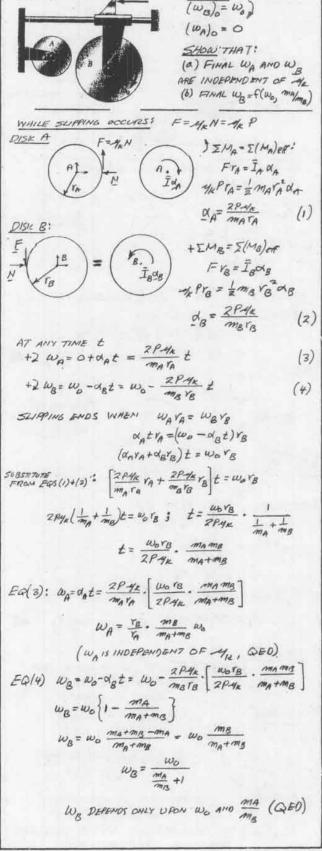
dR=47,2 rad/3) KINEMATICS!



CONFIRM THAT ASSUMPTION OF SLIPPING RETWEEN DISICS IS TRUE



EDI): (446 nom)(3111) = WE(5111)



GIVEN:

WA= 446 YPM)

WB = 268 YPM 2



SHOW THAT

SYSTEM OF

EFFECTIVE FORCES

FOR A SLAB

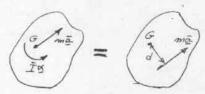
REDUCES TO MO.

AND EXPRESS DISTANCE

FROM ITS LINE OF ACTION

TO G IN TERMS OF RO. AND X.

WE KNOW THAT THE SISTEM OF EFFECTIVE
FORCES CAN BE NEUVOLD TO THE YECTOR AND
AT G AND THE COUPLE IN. WE FURTHER KNOW
FROM CHAFTER 3 OF STATICS THAT A
FORCE-COUPLE SYSTEM IN A PLANE CAN BE
FURTHER LEDUCED TO A SINGLE FORCE.



THE PERPENDICULAR DISTANCE of FROM G TO THE LIME OF ACTION OF THE SINGE VECTOR ME IS EXPRESSED BY WEITING

$$\tilde{J} \propto = (m\tilde{\alpha})d$$

$$d = \frac{\tilde{k}^2}{\tilde{\alpha}}$$

 $d = \frac{I\alpha}{m\bar{\alpha}} = \frac{mk\bar{\alpha}}{m\bar{\alpha}}$ $16.47 \cdot (\Delta m_i)(\alpha \times r_i^i)$



(AMI) SHOW THAT THE SYSTEM OF EFFECTIVE FORCES OF A 121610 SLAB CONSISTS OF THE VECTORS SHOWN ATTACHED TO THE PARTICLES P. OF THE SLAB. FURTHER

SHOW THAT THE EFFECTIVE FORCES REDUCE TO M & ATTACHED AT G AND A COURLE IX.

KINEMATICS



THE ACCELERATION OF P_{i} is $a_{i} = \bar{a} + a_{i}/\epsilon$ $a_{i} = \bar{a} + a_{i}/\epsilon$ NOTE THAT $a_{i} = a_{i}/\epsilon$ $a_{i} = a_{i}/\epsilon$

THUS, THE EFFECTIVE FORCES ARE AS SHOWN IN FIG P 16:47 (also shown above). WE WRITE

 $(\Delta m_i) \underline{a}_i = (\Delta m_i) \underline{a} + (\Delta m_i) (\underline{\alpha} \times \underline{r}_i^*) - (\Delta m_i) \underline{\omega}^2 \underline{r}_i^*$

THE SWI OF THE EFFECTIVE FOR CES IS $\Sigma(\Delta m_i) \underline{a}_i = \Sigma(\Delta m_i) \underline{a} + \Sigma(\Delta m_i) (\underline{d} \times \underline{Y}_i') - \Sigma(\Delta m_i) \underline{w}_i^{\underline{x}_i'}$ $\Sigma(\Delta m_i) \underline{a}_i = \underline{a} \Sigma(\Delta m_i) + \underline{x} \times \Sigma(\Delta m_i) \underline{Y}_i' - \underline{w}^2 \Sigma(\Delta m_i) \underline{Y}_i'$

(CONTINUED)

16.47 continued

WE NOTE THAT

 $\Sigma(\Delta m_i) = m$. AND SINCE G IS THE MASS CENTER $\Sigma(\Delta m_i) v_i' = m_i \bar{v}_i' = 0$ THUS, $\Sigma(\Delta m_i) a_i' = m_i \bar{a}$ (1)

THE SUM OF THE MOMENTS ABOUT & OF THE EFFECTIVE FORCES IS:

 $\Sigma(\underline{r}' \times \Delta m_i \underline{\sigma}_i) = \Sigma \underline{r}' \times \Delta m_i \underline{a} + \Sigma \underline{r}'_i \times (\Delta m_i) (\underline{a} \times \underline{r}'_i)$ $- \Sigma \underline{r}' \times (\Delta m_i) \underline{\omega}' \underline{r}'_i$

I (rixAm; ai)=(EriAmi) =+[rix(xxx))Ami]-w2[(rixri)Ami

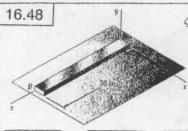
SINCE GISTHE MASS CENTER, IT! AME = 0
ALSO, FOR EACH PARTICLE, Y'XY' = 0

 $\Sigma(\underline{r}_i \times \Delta m_i a_i) = \Sigma[\underline{r}_i \times (\underline{\alpha}_i \times \underline{r}_i) \Delta m_i]$

SINCE $\underline{\vee} \perp \underline{r}_{i}^{*}$, WE HAVE $\underline{r}_{i}^{*} \times (\underline{\vee} \times \underline{r}_{i}^{*}) = \underline{r}_{i}^{2} \times \underline{\wedge} \underline{r}_{i}$ AND $\underline{\Sigma}(\underline{r}_{i}^{*} \times \underline{\wedge} \underline{m}_{i}^{*}) = \underline{\Sigma}(\underline{r}_{i}^{*} \times \underline{\wedge} \underline{m}_{i}^{*}) \times \underline{\wedge} \underline{\wedge} \underline{r}_{i}^{*} = \underline{r}_{i}^{2} \times \underline{\wedge} \underline{m}_{i}^{*} \times \underline{\wedge} \underline{n}_{i}^{*}$

Since $Zr_i^2 \Delta m_i = \bar{I}$ $\Sigma(r_i \times \Delta m_i a_i) = \bar{I} \omega$ (2)

FROM EOS, (1) AND (2) WE CONCLUDE THAT
SYSTEM OF EFFECTIVE FOLCES REDUCE TO
MÃ ATTACHED AT & AND A COUPLE ÍX.



GIVEN: 1.75-16 ROD AB

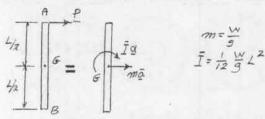
P=0.25/6

L=36/10

FIND: ACCELEMATION

(a) OF A.

(b) DF B.



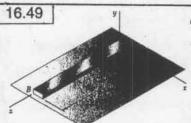
 $\begin{array}{ccc}
\stackrel{+}{\xrightarrow} & \Sigma F_{x} = \Sigma (F_{x})_{x} : & P = m\bar{\alpha} = \frac{W}{g}\bar{\alpha} \\
\bar{\alpha} = \frac{P}{W}g = \frac{0.2546}{1.7546}g = \frac{1}{79} + \frac{1}{12}g = \frac{1}{12}g
\end{array}$

+ $\sum M_g = \sum (M_g)_{eff}$: $P_{\frac{1}{2}} = \sum \alpha = \frac{1}{12} \frac{w}{9} L^2 \alpha$ $\alpha = 6 \frac{P}{W} \frac{3}{2} = 6 \frac{0.25 B}{1.25 B} \frac{3}{4} = \frac{6}{9} 9$

(a) \$\pma_{A} = \bar{a} + \frac{1}{2} \times = \frac{1}{7}g + \frac{1}{2} \cdot \frac{6}{7}g = \frac{4}{7}g = \frac{4}{7}(32.244/67)

\(\alpha_{A} = \bar{1} \times 40 \frac{1}{2} \rightarrow \)

(b) $\pm a_{B} = \bar{a} - \frac{1}{2}\alpha = \frac{1}{7}9 - \frac{1}{2} \cdot \frac{6}{7}9 = -\frac{2}{7}9 = -\frac{2}{7}(32,274/5^{2})$ $a_{B} = 9.2 \text{ At } / 5^{2} + \frac{1}{2} = \frac{1}{7}(32,274/5^{2})$



GINEN: 1.75-16 ROD AB

P=0.2516

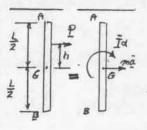
L=3ft

FINDI (a) WHERE P

SHOULD BE APPLIED

FOR aB=0.

(b) CORRESPONDING ACCEL, OF POINT A.



 $\frac{1}{2} I F_{x} = \sum (F_{x})_{AF}$ $P = m \bar{a} = \frac{W}{9} \bar{a}$ $\bar{a} = \frac{P}{W} 9 \rightarrow$ $+2 \sum (F_{x})_{AF} = \sum (M_{x})_{AF} = \frac{1}{12} \frac{W}{3} L^{2} d$ $d = \frac{12Ph}{WL^{2}} 9 2.$

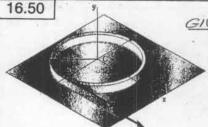
(a)
$$\pm \alpha_{18} = \bar{\alpha} - \frac{L}{2} \times 0$$

 $0 = \frac{P}{W}g - \frac{L}{2} \cdot \frac{12Ph}{WL^{2}} \cdot 9 ; h = \frac{L}{4} = \frac{36im}{6} = 6in.$

THUS, P IS LOCATED 12 in FROM FOOD A.

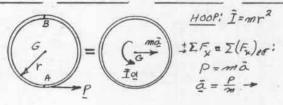
(b)
$$\pm a = \bar{a} + \frac{1}{2}\alpha = \frac{p}{w}g + \frac{1}{2} \cdot 2\frac{p}{w}\frac{g}{2} = 2\frac{p}{w}g$$

a= 2 0.25/b (32.2 ft/s); a=9.29t/s2



GIVEN: P=3N m=2.4-29

> (a) a_A (b) a_B



+) $IM_6 = I(M_6)_{PH}$: $Pr = \bar{I}\alpha = mr^2 \propto \frac{P}{mr}$

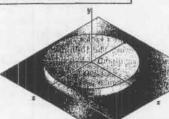
(a)
$$\pm a_A = \bar{a} + r\alpha = \frac{P}{m} + r(\frac{P}{mr}) = 2\frac{P}{m}$$

 $a_A = 2\frac{3N}{2.4Rg} = 2.5m/s^2$
 $a_A = 2.5m/s^2$

(b)
$$\pm q_8 = \bar{a} - r\alpha = \frac{P}{m} - r(\frac{P}{mr}) = 0$$

aB=0

16.51 and 16.52



GIVEN: P=3N m=2.4kg

PROBLEM 16.51

FIND:
(a) QA
(b) QB

PROBLEM 16.52

SHOW THAT FOR 360° ROTATION DISK WILL MOVE DISTANCE TY.

 $\frac{D/SK: \bar{I} = \frac{1}{Z}mr^2}{+\sum_{k} \sum_{k} (F_k) ar}:$ $P = m\bar{a}$ $\bar{a} = \frac{P}{M} \rightarrow$

+) $IM_G = I(Mder)$ $Pr = \overline{I} \propto$ $Pr = \frac{1}{2}mr^2 \propto$ $\alpha = \frac{ZP}{mr}$

PROBLEM 16.51

$$(a) \xrightarrow{+} a_{\beta} = \bar{a} + r\alpha = \frac{\rho}{m} + r \cdot \frac{2P}{mr} = 3\frac{\rho}{m}$$

$$a_{\beta} = 3\frac{3N}{2.489} = 3.75 \text{ m/s}^2$$

a = 3.75 m/s2 ->

(b) $\frac{1}{2} o_B = \hat{a} - \gamma \alpha = \frac{P}{m} - \gamma \cdot \frac{2P}{m\gamma} = -\frac{P}{m}$ $\alpha_B = -\frac{3N}{2.4 \log} = -1.25 \text{ m/s}^2$ $\alpha_B = 1.25 \text{ m/s}^2$

PROBLEM 16.52

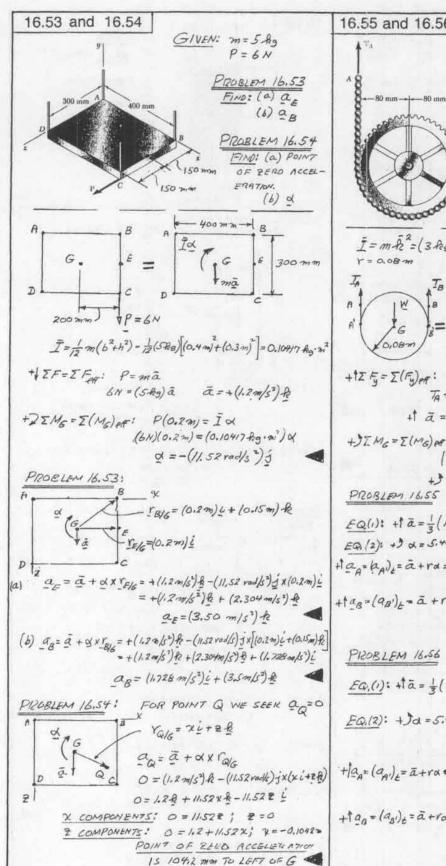
LET $t_i = TIME$ REQUIRED FOR 360 ROTATION $\Theta = \frac{1}{2} \times t_i^2; \quad 2\pi \text{ rod} = \frac{1}{2} \left(\frac{2P}{2\pi r}\right) t_i^2$

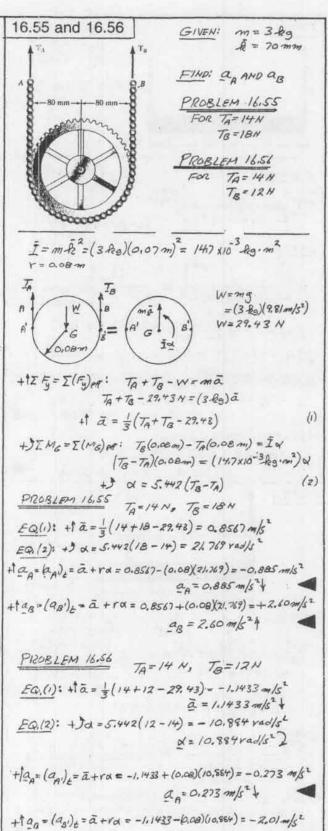
t, 2 = 211 mr

LET 4, = DISTANCE & MOVES
DURING 360° ROTATION

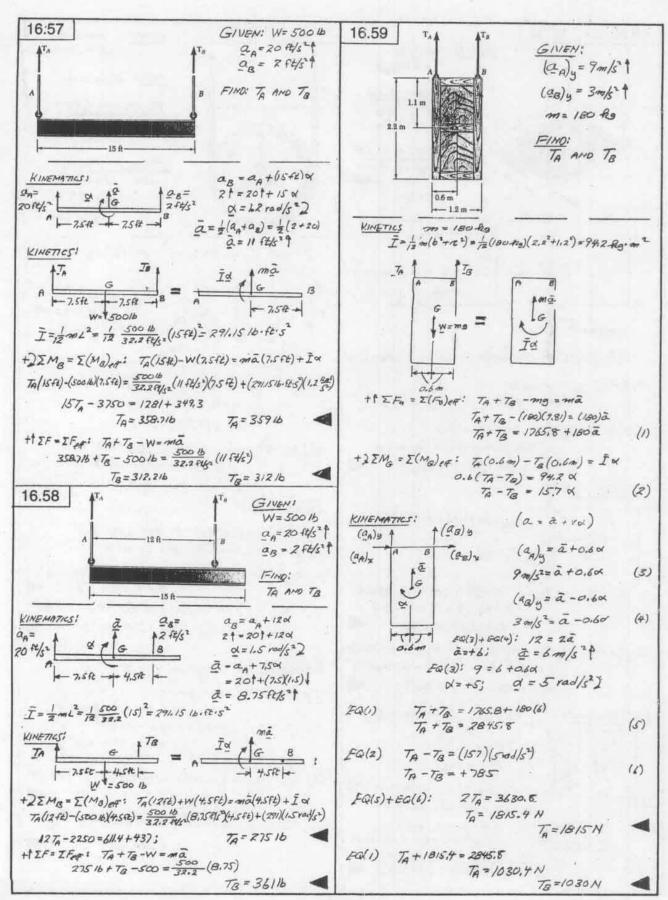
 $\gamma_i = \frac{1}{2}\bar{a}t_i^2 = \frac{1}{2}\frac{P}{m}\left(\frac{2\pi mr}{P}\right)$

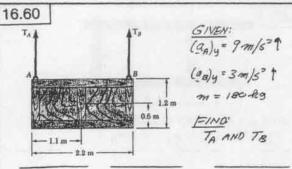
x,= TTr Q.E.O.





a = 2.01 m/s2+





I= 1/2 m(a2+b2)= 1/2 (180-log)(2,22+1,22)= 94,2 login2

KINETICS:

$$+ \uparrow \Sigma F_y = \Sigma (F_y)_{eff}$$
: $T_A + T_B - mg = m\bar{a}$
 $T_A + T_B + (180 kg)(9.81 m/s^2) = (180 kg)\bar{a}$
 $T_A + T_B = 1705.8 + 180.\bar{a}$ (1)

$$\begin{array}{c|c} \underline{(CINEMATICS)} \\ (a_{A})_{\frac{1}{2}} \\ (a_{A})_{\frac{1}{2}}$$

$$EQ(3) + EQ(4)$$
; $12 = 2\bar{a}$
 $\bar{a} = +6\pi/s^2$ $\bar{a} = 6\pi/s^4$

$$FQ(3)-FQ(4)$$
: $6=2.29$
 $\alpha=2.727 \text{ mod/s}^2$ $\alpha=2.73 \text{ rod/s}^2$

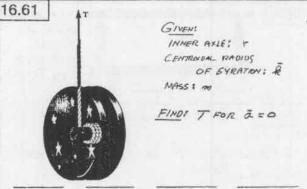
EQ():
$$T_A + T_B = 1765.8 + 180(6)$$

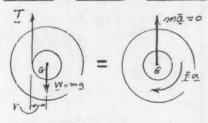
 $T_A + T_B = 2845.8$ (5)

$$FQ.(2)$$
: $T_A - T_B = 85.636(2.727)$
 $T_A - T_B = 233.5$ (6)

$$T_A - T_B = 233.5$$
 (6)

$$EQ(1)-EQ(2)$$
: $2T_g = 26/2.3$
 $T_g = 1306.2 \text{ M}$
 $T_g = 1306 \text{ N}$

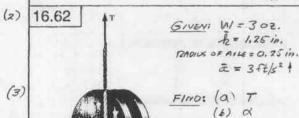


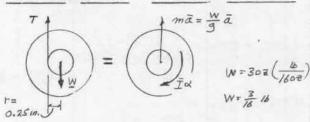


+ + IFy = E(Fy)er: T-mg = 0 ; T=mg

+)
$$IM_6 = I(M_6)_{AF}$$
: $Tr = \bar{I} \times \frac{1}{R^2}$

$$\alpha = \frac{r_8}{\bar{R}^2} \qquad \alpha = \frac{r_8}{\bar{R}^2}$$





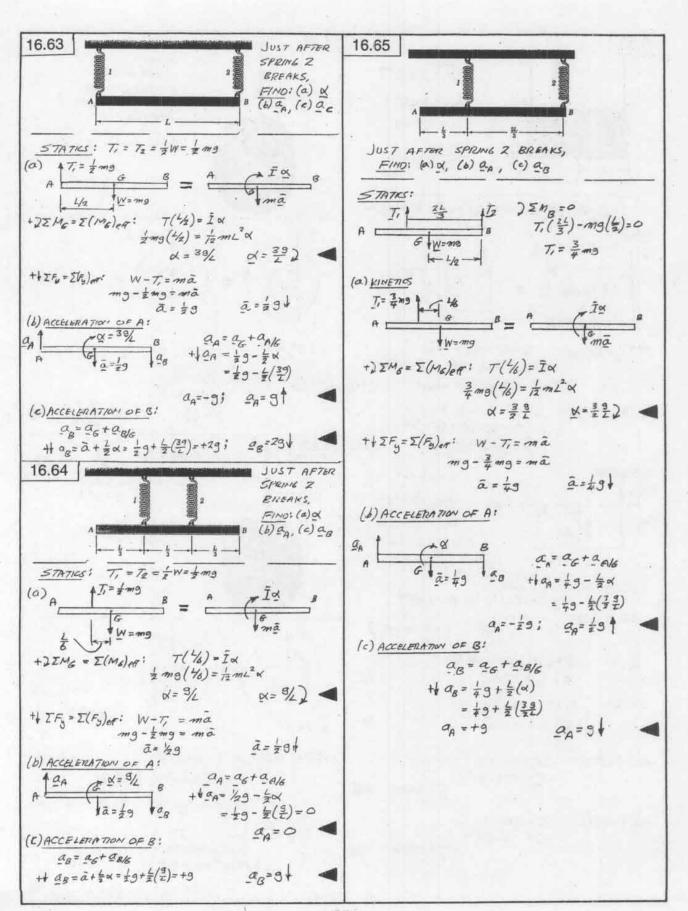
$$+$$
 $\uparrow \Sigma F_y = \Sigma (F_y)_{af}$: $T - W = \frac{W}{g} \bar{a}$

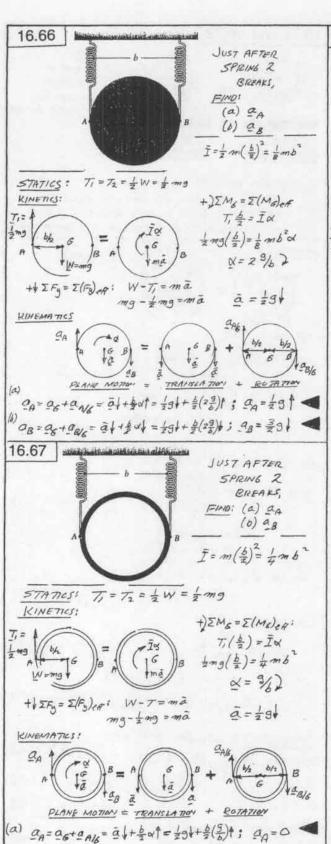
$$T - \frac{3}{16} I_0 = (\frac{3}{16} I_0) \frac{3fl/s^2}{32.2fl/s^2}$$

$$T = 0.205 I_0$$

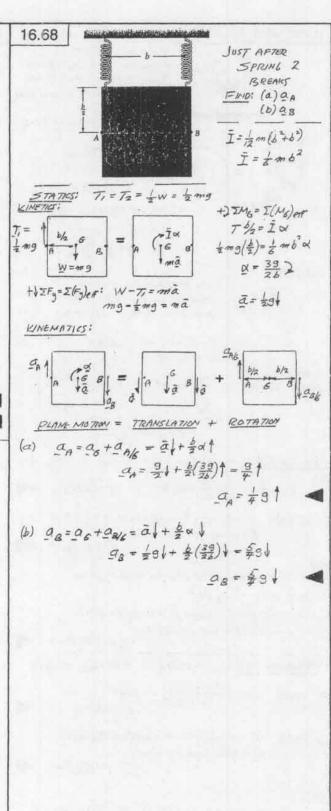
$$T = 0.205 I_0$$

+
$$\int \Sigma M_{6} = \Sigma (M_{6})_{eff}$$
: $Tr = \overline{I} \ d$
 $(6.705 \ b) \left(\frac{0.725}{72} \ fe\right) = m - \overline{F}_{2}^{2} \ d$
 $4.271 \times 10^{3} \ lb \ ft = \frac{3/16 \ lb}{32.27 \ fe}_{5}^{2} \left(\frac{1.725}{12} \ ft\right) \ d$
 $C = 67.6 \ rad/c^{2} \ (2 = 67.6 \ rad/c^{2})$





(b) a3=a6+a8/6= a++ \$ x = = 29++ \$ (96); a8=9 €



16.69 and 16.70 GIVENI Vo= 15 12/5 v=4in, 4/x=0.10. PROBLEM 16.691 Ub = 9 rad/s PROBLEM 16.70: Wo = 18 rads FIND: (a) t, WHEN ROLLING STOP (b) V AT t, (e) DISTANCE TRAVELED AT E, KINETICSE # IFx = 2 (Fx) pg ymg = ma a=1/19-+ DEMS = I (MG) = Fr = I X (1/2 mg)r = 2 mr? d= 5 1/2) KINE MATICS: NUEN SPRENE ROLLS, INSTANT CENTEL OF ROTATION IS AT C AND WHEN tot, To rav v= vo-at = vo-1,9 t a=-wo+dt =-wo+5 1/29t WHEN t=t,1 No-1/18 t, = (- wo + 5 - 1/29 t) + Equi V= rw: To-4,9t, =- wor+ 5 4,9t, Z== 2 (vo+rwo) PROBLEM 16.81 25= 15 ft/s, w= 9 rad/s, r=4in,= 1 st $\dot{\xi}_{1} = \frac{2}{7} \frac{\left(15 + \frac{1}{3}(9)\right)}{0.1(37.2)} = 1.59725 \dot{\xi}_{1} = 1.59725$ (b) FQ.(2): N=No-1/9 ti = 15-0.1 (32.2) (1.5972) V,= 15-5.1479= 2.857 Pt/s J= 9.86 A/5-(c) ā=4/29=0.1(32,29/5°)=3.22 f1/5° + \$ 5, = vot, - \frac{1}{2} at,2 = (15ft/s)(1.5975) - \frac{1}{2}(3.22ft/s2)(1.5975)2 = 23,96 = 4011 = 19.85 ft 5=19.85fb-PROBLEM 16.70: 0,=15 ft/s, wo = 18 radk, r= = ft (a) Ea(3): $t_1 = \frac{2}{7} \frac{(15 + \frac{1}{3}(16))}{0.1(37.2)} = 1.86345$

(b) EG(2): V, = 20-429t=15-0.1(32.2)(1.8634)

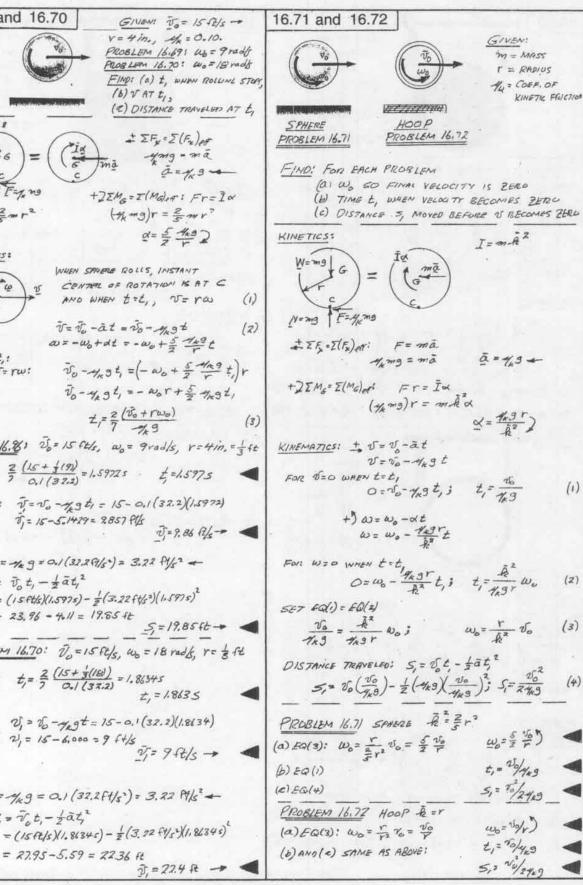
= 27.95-5.59 = 22.36 ft

+ 5, = v, t, - fat,2

(e)

2)= 15-6,000=9 ft/s 2)= 9ft/s ->

a=-4/19 = 0.1 (32,2 Ft/s') = 3,22 Pt/s" -



t. = 1.8635

16.73



GIVEN: SPHERE PLACES,

ON EGET WITH NO VELOUTY.

T = RADIUS,

A/K = COEF. KINETIC PRICTORY

A/K = COEF KINETH PRICTION

FIND: (a) t WHEN SPHERE ROLLS

(b) T AND W WHEN L = t,

+ ZE= Z(F) of: F= ma - 4 mg = ma = 4g = +) ZM6= Z(M6) of: Fr= IX (4, mg) r= = mr2 x

X= 5 4/49 5

KINEMATICS: $\pm \overline{v} = \overline{a}t = \eta_k g t$ (1) $\pm i\omega = \alpha t = \frac{s}{2} \frac{\pi_k g}{r} t$ (2)

C= POINT OF CONTACT WITH BELT $\pm \sqrt{c} = \overline{v} + \omega v = \sqrt{kg} t + \sqrt{\frac{5}{2}} \frac{\sqrt{kg}}{r} t r$ $\sqrt{c} = \frac{7}{2} \sqrt{kg} t$

(a) WHEN SPHENE STARTS ROLLING (t=t,), WE HAVE

V=V; V= 2-4/2 t,

t,= 2 25

(b) VELOGINES WHEN tot,

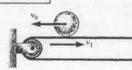
EQ(1): 3=49 (2 1/49)

र्म= = ३ ४, →

EQ(2): W= (5 1/2) (2 1/1)

四三至近り

16.74



GIVEN: SPHERE WITH SO +

AND WO = O PLACED

ON RELT. Y = RADIUS

4/E = COEF, KINETIC FRICTION

FINO: (a) YO SO THAT

SPHERE WILL HAVE NO

LINEAR VELOCITY AFTER IT STARTS ROLLING ON EFLT, (b) t, WHEN SMERE CTANTS ROLLING (c) DISTANCE SPHERE WILL HAVE MOVED LYHEN t=t,

W=mg G F=41,mg N=mg N=mg

 $\pm \sum F_{g} = \sum (F_{g})_{g} : F = ma$ $\Delta = 1/2 = ma$ $\Delta = 1/2 = ma$ $+ \sum F_{g} = \sum (M_{g})_{g} : F_{r} = \sum M_{g} = ma$ $(4_{g} = mg)_{r} = \frac{3}{5} = mr^{2} \times M_{g} = ma$ $\times = \frac{5}{5} = \frac{4/2}{5} = ma$

KINEMATICS:

+ v= vo-at = vo-1/49t (1)

 $+ \int \omega = \alpha t = \frac{5}{2} \frac{4k9}{r} t$ (2)

(CONTINUED)

16.74 continued



BUT, WHEN t=t, $\overline{V}=0$ AND $\overline{V}_{c}=\overline{V}_{1}$ FG(3): $\overline{V}_{1}=\frac{5\pi N^{3}}{2}t$, $\overline{t}_{1}=\frac{2\overline{V}_{1}}{5\pi N^{3}}$

FQ(1): $\vec{v} = \vec{v}_0 - \vec{J}_0 \vec{z}$ WHEN $t \cdot t_1$, $\vec{v} = 0$, $0 = \vec{v}_0 - \vec{J}_0 \vec{z} \vec{v}_1$; $\vec{v}_0 = \frac{2}{5} \vec{v}_1$

DISTANCE WHEN $t = t_i$: $\pm S = v_0 t_i - \frac{1}{2} a t_i^2$ $S = \left(\frac{2}{5}\sqrt{1}\right)\left(\frac{2\sqrt{1}}{5\sqrt{10}}\right) - \frac{1}{2}\left(\frac{4}{10}\right)\left(\frac{2\sqrt{1}}{5\sqrt{10}}\right)^2$ $S = \frac{v_i^2}{4kg}\left(\frac{4}{25} - \frac{2}{25}\right)$; $S = \frac{2}{25}\frac{v_i^2}{4kg}$

16.75



SHOW THAT IN (FOLLIS)
CON BE ELIMINATED BY
ATTACHING

Mat AND mam AT POINT P ON OG WHERE GP = \$2/F

OG= r az= rx

WE FIRST OBSERVE THAT THE SUM OF THE VECTORS IS THE STAME IN BOTH FIGURES TO HAVE THE SAME SUM OF MOMENTS ABOUT G, WE MUST HAVE

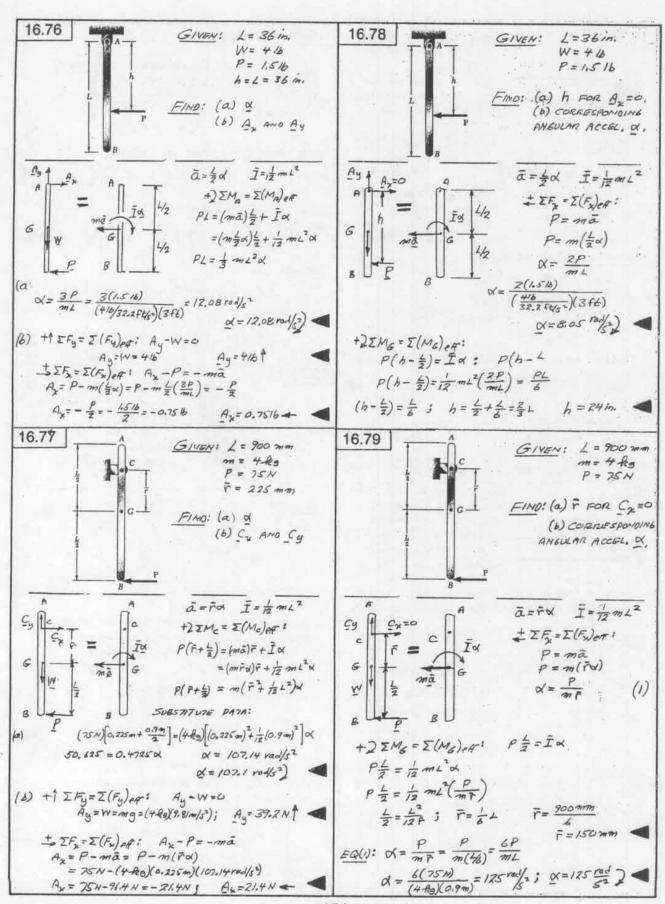
+) $EM_{\varepsilon} = EM_{\varepsilon}$: $\tilde{J} \propto = (m\tilde{\sigma}_{\varepsilon})(GP)$ $m\tilde{h}^{2} \propto = m\tilde{r} \propto (GP)$ $GP = \frac{\tilde{A}^{2}}{\tilde{r}} \qquad (QE, 0.)$

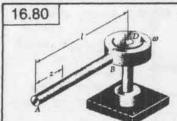
NOTE: THE CENTER OF ROTATION AND THE CENTER OF PERCUSSION ARE INTERCHANCEAGLE, INDEED, SINCE $06=\overline{V}$, WE MAY WRITE $6P=\frac{\overline{h}^2}{60}$ or $60=\frac{\overline{h}^2}{6P}$

THUS, IF POINT P IS SELECTED AS CENTER.

OF KUTATION, THEN POINT O IS THE CENTER.

OF PERCUSSION.





GIVEN: W= 0.25 16/52 1 = 1.2ft a= 150 xpm 2=0,9ft

FIND: TENSION IN ROD (a) IN TERMS OF W, l

2, AND W. (b) FOR GIVEN DATA

ā=rw= (1-1/2)w2 IN HORIZONTAL PLANE:

IFF = EFer: T= mā =(4 2)(1- 3) w2 T= w/(12-=2)w2

SUBSTITUTE DATA; w= 150 rpm (211) = 511 nad/s, 2=0.9ft

T= 0.25 16/ft (1.24)(0.9 ft) - (0.9 ft)2 (5Trad/s)2 T= 1.293 16



GIVEN: FLYWHEEL, CENTER. OF ROTATION AT O, AND MASS CENTER AT G W= 1200 rpm. MAX IMUM FORCE EX FRZED ON SHAFT 15 55 RN+ AND 85 RN+. FIND! (a) MASS OF FLYWHEEL (b) DISTANCE F

W = 1200 rpm (21) = 40 11 rad/s a = rw2

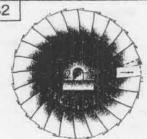
+ TF = ZFeF 85 AN-mg= Man (1) 85-mg=mFw

+ + ZF = ZFOR 55 AN + mg = man 55+mg= mrw2

EQ(2)-EQ(1): 30 km - 2mg =0 30×10 N = 2 m (9.81 m/53) m= 1529-29

m=1529 kg 140 AN = 2m rw EQ(1) + EQ(2): 140 ×103 N = 2(1529 Rg) \$ (407)2 F= 2.90 X10 m

16.82



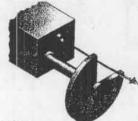
GIVEN! A 45-9 VANE IS THROW OFF FROM BALANCED TURBIN DISK. 00 = 9600 rpm FIND: REACTION AT O

W= 9600 rpm (217) W= 320 11 rol/s CONSIDER VANE BEFORE IT 15 THROWN OFF

+ IF= IF = R = man = m Fw = (45×10-3 Rg)(0,3 m)(320 71)2 R=13.64-RN

BEFORE VANE WAS THROWN OFF DISK INAS BALANCED (R=0). REMOVING VANE AT A ALSO REMOVES ITS REACTION, SO DISK IS UNBALANCED AND REMOTION IS R= 13,64RN-

16.83



GIVEN: 0.125-16 SHUTTER OF RADIUS 0.75 in. W = 24 cycles per seconi

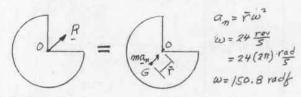
FIND: MAGNITURE OF FORCE EXELTED ON SHAFT BY SHUTTERL

V=0.75in d=3 1/

SEE INSIDE FRONT COVER FOR CENTROID OF A CINCULAR SECTOR

F = 2r sind F= 2(0.7510.) Sin (3/11) 3 (377)

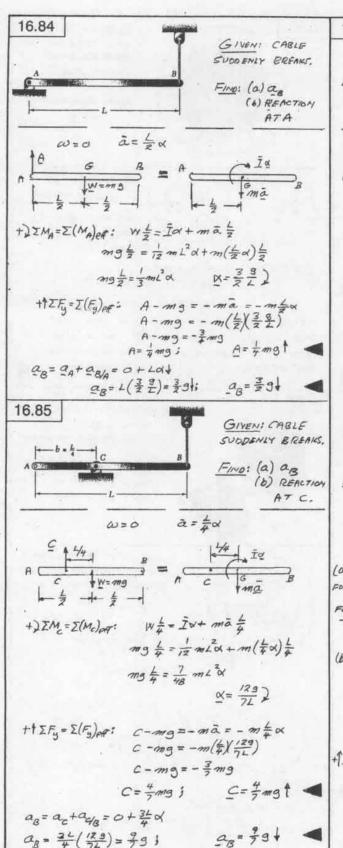
r= 0.15005 in.

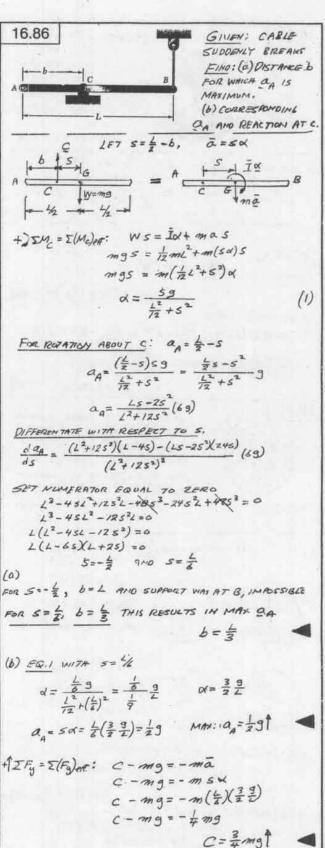


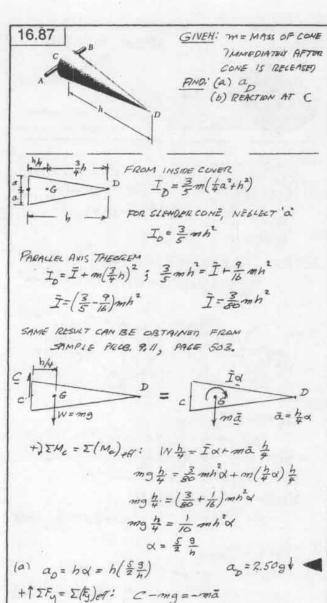
+ 2F = 2FOR R= man=mrw2 = (0.12516) (0.15005 ft) (150.8 rodk) 12= 1.1038 1b 1

FORCE ON SHAFT IS R=1.104162 MAGNITUDE: R=1.104/b

F= 2.90 2m



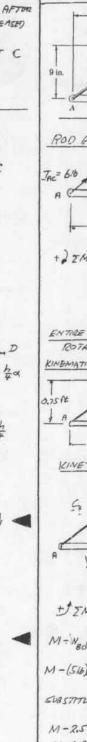


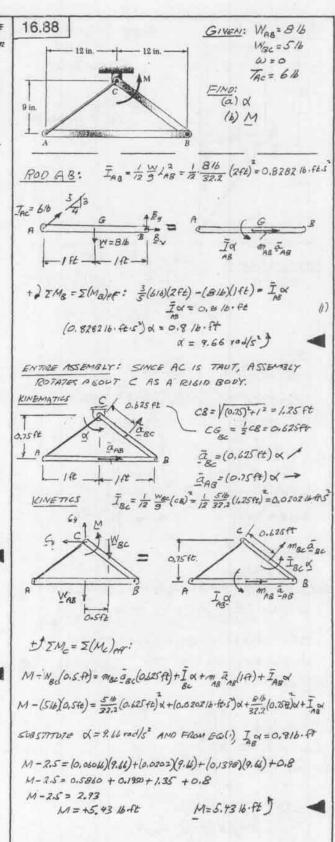


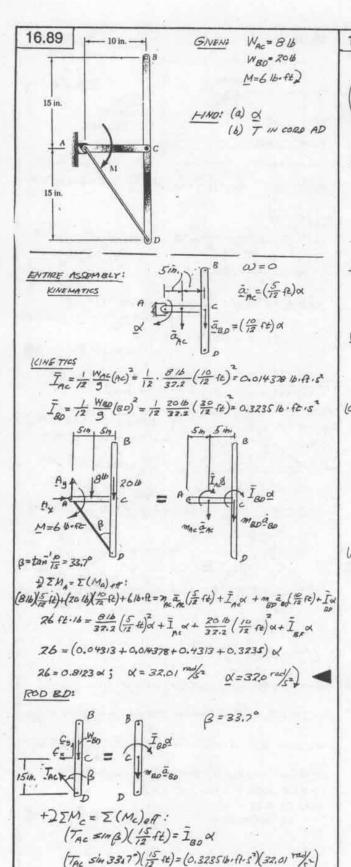
C-mg = -m +x C-mg=-m #(5 g)

C-mg=- 5mg

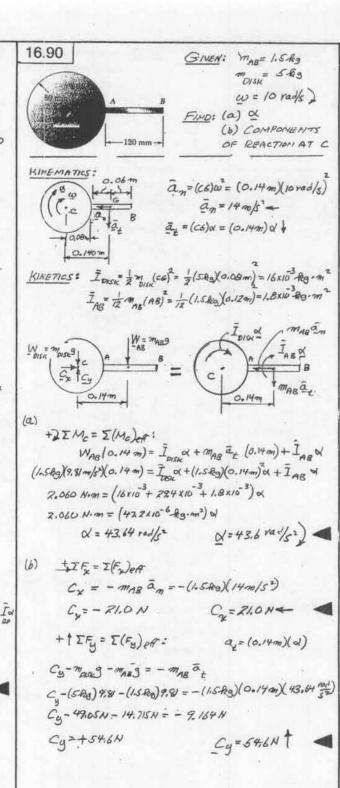
C= 3mg 1



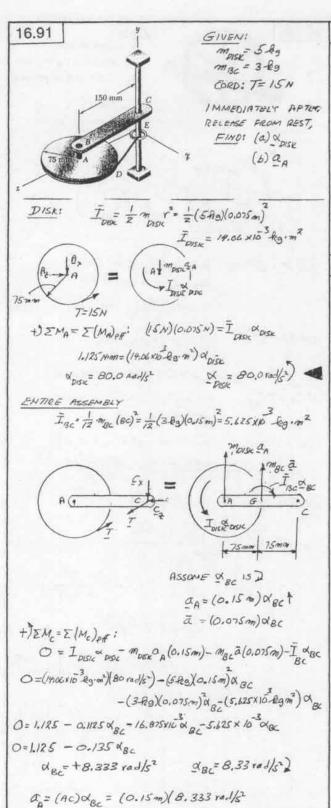




TAC= 14.93 16



TAC= 14.93 16 .



ap=+1,25 m/52

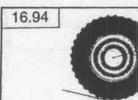
C = (Boradki)j

NOTE: ANSWERS CAN ALSO BE WRITTEN!

16.92 DERIVE IME I OR THE ROLLING DISK OF F16.16.17. W=>ng +) IMc = [(Mc)ef: IMc= (ma)r+IX = (m rx) + I q IM = (mr2+I)d BUT, WE KNOW THAT I = mr + I THUS: IMC = ICX (Q. E. D.) 16.93 FOR AN UNBALANCED DISK SHOW THAT IM = I & IS VALID ONLY WHEN THE MASS CENTER G, THE GEOMETRIC CENTER O, AND THE INSTANTANEOUS CENTER C HAPPEN TO LIE IN A STRAIGHT LINE, KINEMATICS: a= a + a 6/c = ac+ xxxxx + wx(wx roc on, SINCE WI Traje (1) a=ac+ xxr61c- w2r61c KINETICS IMc = I(Mc) ef : IMc = IX + raic x ma RECOLL EQ(1): IMC= IX+16/xm(a+xx/-w2/6/2) ZMc = Ix+rgexmax+mrgex(xxrge)-mwrgexrge BUTIFEKX TEK = O AND & I VEK relexm(dxrele) = mrela THUS: EMc= (I+m rate) & + rate x mac SINCE I = I+m roic IMC = Icx + rexxmac EGI2) REDUCES TO IMC= IN WHEN SIX MAC =0 THAT IS, WHEN YOK AND a ARE COLLINEAR. REFERRING TO THE FIRST DIAGRAM, WE NOTE THAT THIS WILL OCCUR ONLY WHEN POINTS G, Q AND C LIE IN A STRAIGHT LINE.

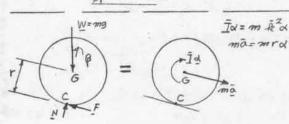
an=1.25 m/s2+

a=-(1.25m/52) L



GIVEN: PROLLING WHEEL

FIND: ā IN TERMS
OF 1, A, B, AND 9.



+) $\Sigma M_e = \Sigma (M_c)_{eff}$: $(W sin \beta) r = (m \bar{\alpha}) r + \bar{I} \propto (mg sin \beta) r = (m r \alpha) r + m \bar{k}^2 \propto rg sin \beta = (r^2 + \bar{k}^2) \propto \propto = \frac{rg sin \beta}{r^2 + \bar{k}^2}$

a=ra=r rgsinB

 $\bar{a} = \frac{r^2}{r^2 + \bar{h}^2} g \sin \beta$



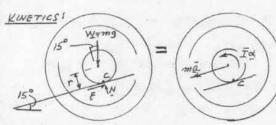


GIVEN: STARTING
FROM REST, FLYWHEEL
MOVES 16 ft IN 405
Y=1.5in.

FINO: I

KINEMATICS; $S = \sqrt[4]{t} + \frac{1}{4} \bar{a} t^2$ $16R = 0 + \frac{1}{2} \bar{a} (40s)^2$ $\bar{a} = 0.02 \text{ ft/s}^2$ Since Y = 1.5 in = 0.125 ft

ā = ra; 0.02 ft/s = (0.125 ft) x x = 0.16 rad/s



+) $\sum M_c = \sum (M_c)_{eff}$: $(mg \sin 15^\circ)r = \overline{L} \times + (m\overline{\alpha})r$ $(mg \sin 15^\circ)r = m\overline{R}^2\alpha + (mr\alpha)r$ $gr \sin 15^\circ = (\overline{R}^2 + r^2)\alpha$

-R2= 6.4953

DATA: r = 0.125 ft, $d = 0.16 \text{ rad/s}^2$ $(32.2 \text{ ft/s}^2)(0.125 \text{ ft}) \sin 15^0 = (\bar{R}^2 + r^2)(0.16 \text{ rad/s}^2)$ $\bar{R}^2 + r^2 = 6.511 \text{ ft}^2$ $\bar{R}^2 + (0.125 \text{ ft})^2 = 6.511 \text{ ft}^2$

-R = 2.55 ft

16.96



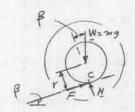
EIVEN:

E = CENTROIOAL

PADIUS OF EYRATION

MS = COEF, STATIC FRICTION

FINDS LARGEST B FOR TROLLING WITHOUT SLIPPING



Ī=mē² m=rx

+) $\Sigma M_C = \Sigma (M_C)_{eR}$: $(mg sing) r = \overline{I} \omega + (ma) r$ $mg sing r = m\overline{h}^2 \omega + mr^2 \omega$ $\omega = \frac{gr}{r^2 + \overline{h}^2} sing \qquad (1)$

 $\pm 3 \ \Sigma F = \Sigma F_{eff}!$ $F - mg \sin \beta = -ma$ $F - mg \sin \beta = -mr \times F - mg \sin \beta = -mr \times$

+ TF= TFor: N-mgcosp=c N=mgcosp

IF ELIPPING IMPENOS $F = M_S N$ OR $M_S = \frac{F}{N}$ $M_S = \frac{F}{N} = \frac{mg \sin \beta - mr \alpha}{mg \cos \beta} = \frac{\sin \beta - \frac{r}{3} \alpha}{\cos \beta}$

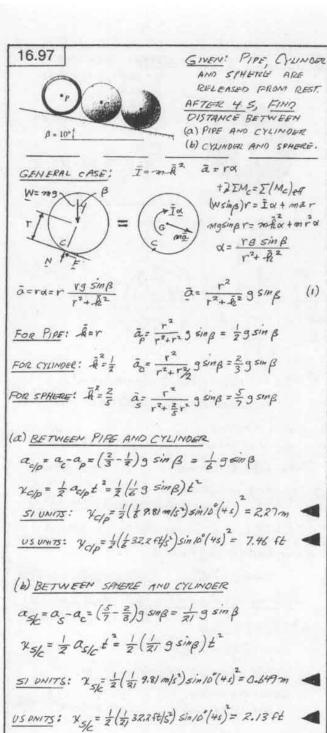
SUBSTITUTE FOR & FROM EQ(1)

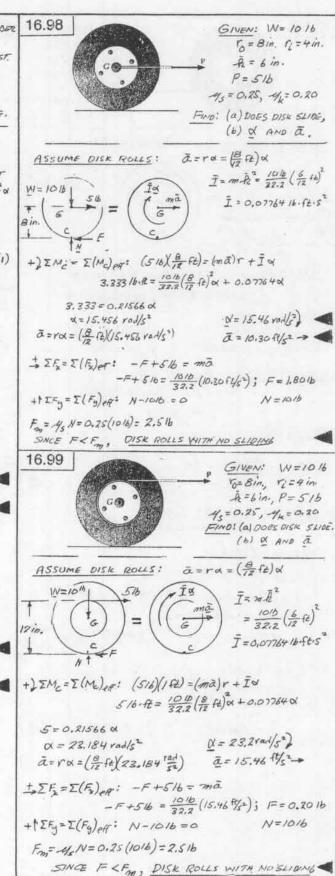
$$-4s = \frac{\sin\beta - \frac{\dot{r}}{g} \cdot \frac{gr}{r^2 + R^2} \sin\beta}{\cos\beta}$$

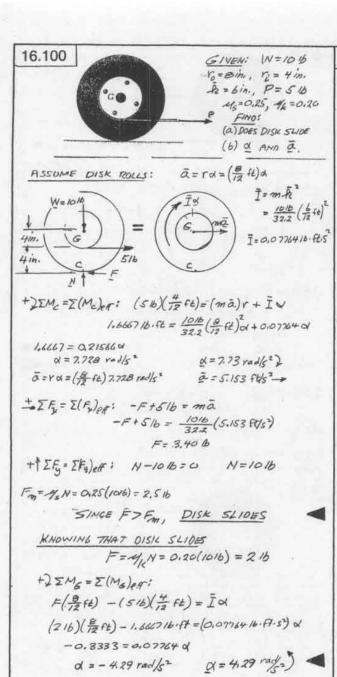
$$y_s = ton \beta \left[1 - \frac{r^2}{r^2 + h^2}\right] = ton \beta \left[\frac{\bar{k}^2}{r^2 + \bar{k}^2}\right]$$

$$ton \beta = y_s \frac{r^2 + \bar{k}^2}{\bar{k}^2}$$

$$ton \beta = y_s \left[1 + \left(\frac{r}{h}\right)^2\right]$$



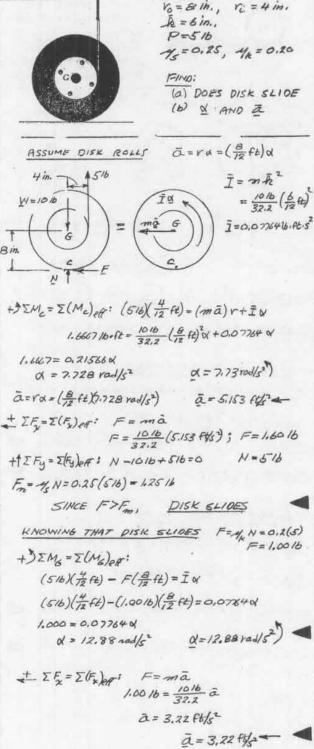




I IF = I (Fx) er:

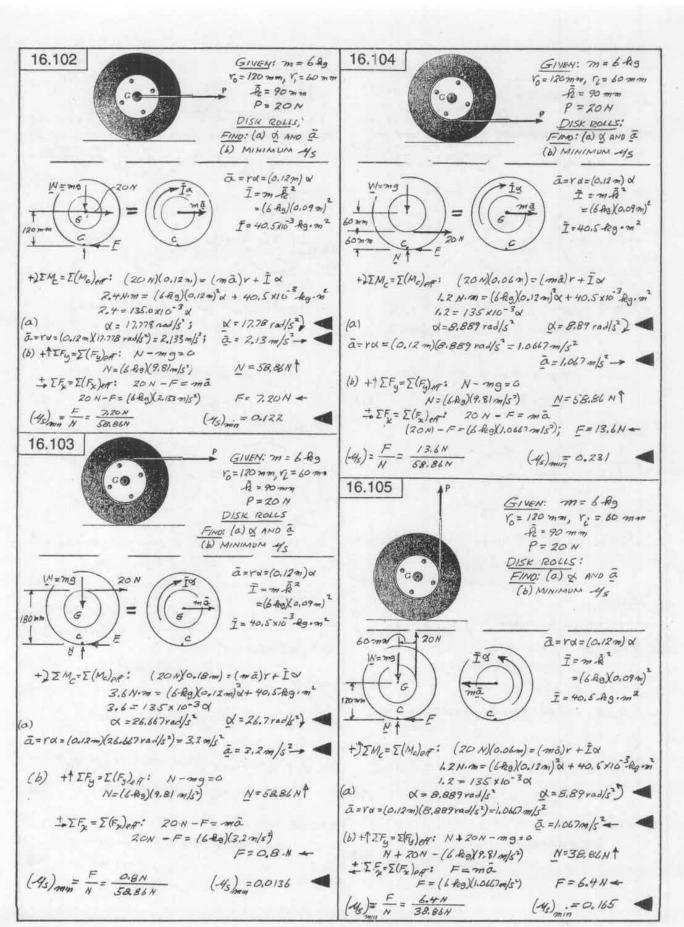
-F+516=ma -216+516= 1016 a

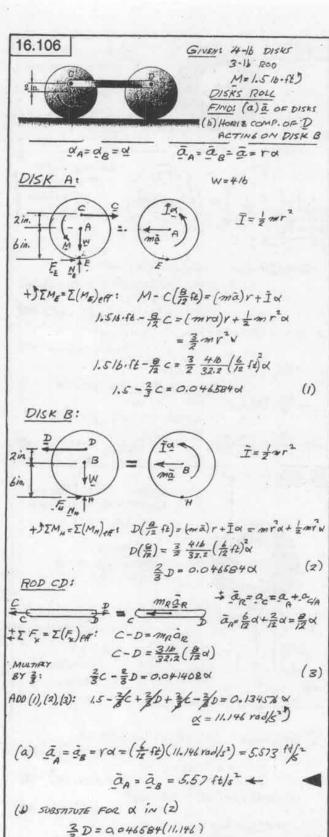
a = 9.66 ft/s2 = 9.66 ft/s2->



GIVEN: W= 1016

16.101

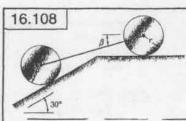




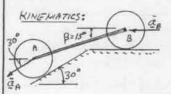
16.107 GIVEN 4-16 DYSKS 3-16 1200 M=1.516. FE DISKS ROLL FIND: (a) & OF DISKS (b) HORIZ COMP. OF D ACTING ON DISKB a = a = a = rx DISK A: W=410 I= = = n - 2 +) IM= = [(M=)+#: M-c(+in)=(ma)+IX 1.516.ft - 12 C = (mra)r + 1 mrd = 3 mr 2 1.5 1b. ft - 4 C = 3 416 (6 ft) & 1.5- + C = 0,046584 X (1) DISK B: Zin. +) EM = E(MN) ago: D(4/2 ft) = (ma) r+ I x = mrx + 1/2 mrx D(4) = 3 416 (6 ft) X \$D=0,0465840 (2) ROD CD = a= a+ ac/A $\hat{a}_{R} = \frac{6}{12} d - \frac{2}{12} d = \frac{4}{12} d$ + IF = I(F)PA: C-D=mROR C-D= 316 (4 x) MULTIPLY 12 C - 1 D = 0.0103520 BY 4: ADD(1),(2),(3): 1.5-\$+\$+\$-\$= 0.10352d x = 14.490 rad/s2) (a) a=a= vx=(= (= ft)(0.10352 ro.1/s)=7.245 ft/s2 an = an = 7.24 ft/s" + (b) SUBSTITUTE FOR & IN (2) D=0,046584 (14.490); D=2.015 16

D= 2.0216 +

D= 0.77916 -

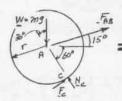


GIVEN: DISKS OF MASS TO AND ROLL ON SURFACES, RELEASE FROM REST WHEN $\beta = 15^{\circ}$, FIND: (a) \bar{a}_A , (b) \bar{a}_B



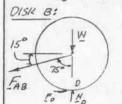
| Sosceles TRIANGE : $\bar{a}_{A} = \bar{a}_{B}$ | DENOTE BY $\bar{a} = \bar{a}_{A} = \bar{a}_{B}$

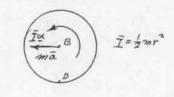
KINETICS: DISK A:





 $t \sum M_c = \sum (M_c)_{eff};$ $(mg \sin 30^\circ) r - (F_{AB} \sin 75^\circ) r = (m\bar{a})r + \bar{1} \propto$ $= (mr\alpha)r + \frac{1}{2}mr^2\alpha$ $mg r \sin 30^\circ - F_{AB}r \sin 75^\circ = \frac{3}{2}mr^2 \qquad (1)$





+) $\Sigma M_D = \Sigma (M_D)_{eff}$: $(F_{AB} \sin 75^\circ)r = (m\bar{a})r + \bar{1}\omega$ $= (mr\alpha)r + \frac{1}{2}mr^2\omega$ $F_{AB}r \sin 75^\circ = \frac{3}{2}mr^2\omega$ (2)

 $EQ(1) + EQ(2); \quad mgr sin30° = 3mr²Q$ $Q = \frac{9}{3r} sin30° = \frac{1}{6} \frac{9}{r}$ $\bar{a} = rQ = r(\frac{1}{6} \frac{9}{r}) = \frac{1}{6} \frac{9}{2}$

PRECALL $\bar{a}_A = \bar{a}_B = a$ $\bar{a}_A = \frac{1}{6}9 \ \mathbb{Z} 30^\circ$

ā = 1/9 ←



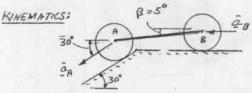
GIVEN: DISKS OF MASS IN

AND ROLL ON SURFACES.

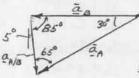
RELEASE FROM DEST

WHEN Q = 5.

FINDS (a) \$\bar{a}_A\$, (b) \$\bar{a}_B\$.



an 730 = an + anis 450



TAW OF SINES TO SINESO THE SINESO THE SINESO THE SINESO

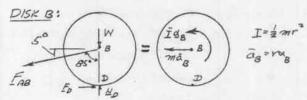
SINTE AB THE AND A TO A 1,

W= mg 5° FAB



 $\tilde{a} = r \alpha_A$ $I = \frac{1}{2}mr^2$

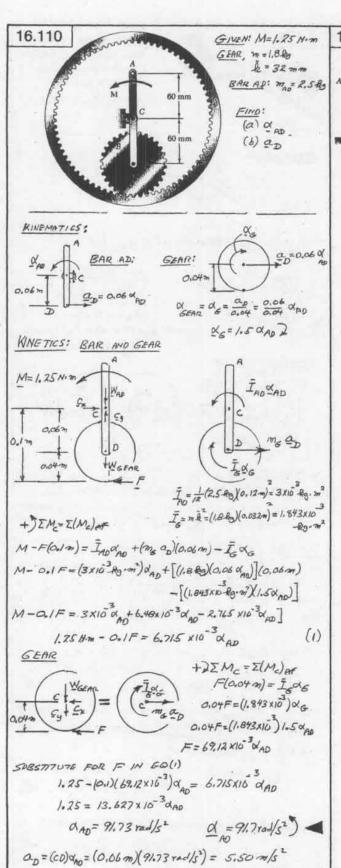
+) $IM_c = I(M_c)eF$: $(mg \le in 30^\circ)r - (F_{AB} \sin bS^\circ)r = (ma)r + IM_A$ $= (mro_h)r + \frac{1}{2}mr^3A$ $mg \sin 30^\circ - F_{AB} \sin bS^\circ = \frac{3}{2}mro_A$ (2)

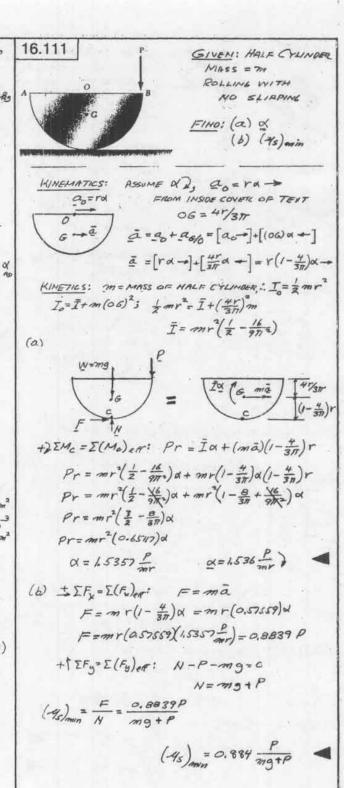


+) $\sum M_D = \sum (M_D)_{a,b} + i \left(F_{AB} \sin 8s^* \right) r = (m \bar{\alpha}_B) r + i \alpha_B$ $= (m r \alpha_B) r + i m r^2 \ell_B$ $F_{AB} = \frac{3}{2} \frac{mr}{\sin 8s^*} \alpha_B \qquad (3)$

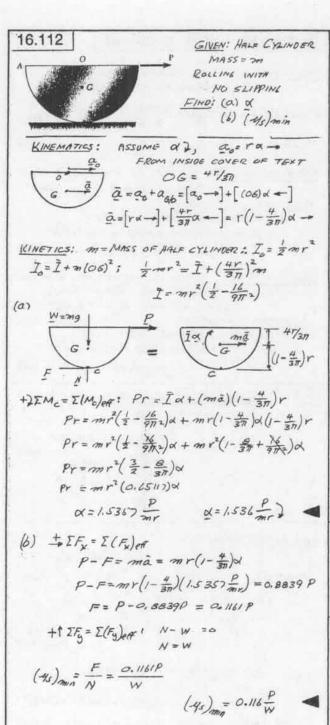
SUBSTITUTE FOR \bar{a}_g FROM EQL) AND F_{AB} FROM EQLS) INTO EQLS) $mg \sin 30^\circ - \frac{3}{2}mr \cdot \frac{\sin 65^\circ}{\sin 85^\circ} (0.90852d_0) = \frac{3}{2}mr d_A$ $0.5 \frac{9}{r} = \frac{3}{2}(0.82657+1)d_A$

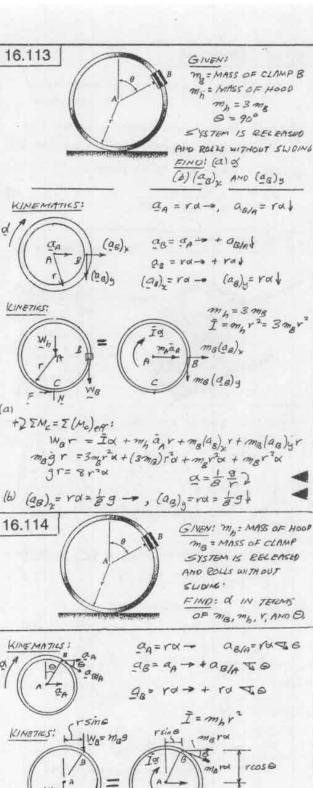
EQ(1) $\bar{a}_{B} = 0.90852 \bar{a}_{A} = (0.90852)(0.18269) = 0.16599$

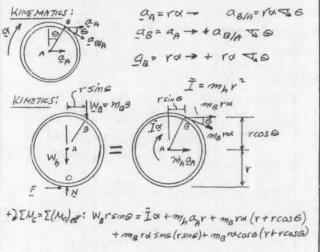




a = 5.50 m/s2 ->







(CONTINUED)

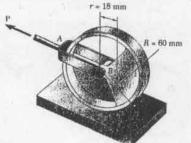
16.114 continued

magraine = mhra+mh(ra)r+mara(1+cose)(r+rcose) + marasine (rsine)

mgrsing = 2mpra + mgra (1+cose) + sin20] = 2m rd + merd [1+2cose + cos =+ sine] mgrsino = r2 [2mh + mg (2+2cos6)]

X= g mB SING mh + mg (1+coso)

16.115 and 16.116

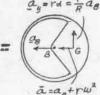


GIVEN: m=1.5 29 A = 44 mm

PROBLEM 16.115: FIND: P WHEN UB= 0.35m/5a= 1.2 m/52 PROBLEM 16.116: FIND: P WHEN VR= 0.35m/5a = 1.2 m/s2-

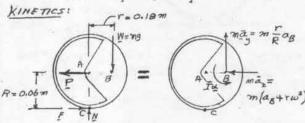
KINEMATICS: CHOOSE POSITIVE VE AND &B TO LEFT au=rd=Fag r=0.018m/

R=0.06%



TRANS, WITH B + ROTATION ABT B = ROLLING MOTION

 $\bar{a} = \left[a_B + r\omega^2\right] + \left[\frac{\Gamma}{R}a_B\right]$



+) IM= I(Mc)eff: PR-Wr=(man)r+(max)R+IX PR-mgr=m(Fag)r+m(ag+rw)R+mk2 ag $= mag\left(\frac{r^2}{R} + R + \frac{\tilde{R}^2}{R}\right) + mr\left(\frac{\tilde{V}_B}{r}\right)^2 R$ $P = mg\left(\frac{r}{R}\right) + mag\left(1 + \frac{r+k}{R^2}\right) + m\frac{r}{R^2}v_B^2$ (1)

(CONTINUED)

16.115 and 16.116 continued

SUBSTITUTE! ON = 1.5 Rg, Y= 0.018M, R= 0.06M, A= 0.044 mm AND g= 9.81 m/s IN EQ(1)

P=1.5(9.81) 0.018 + 1.5(08)(1+ 0.018 +0.04)

P= 4.4145 + 2.4417 aB + 7.5 SR (2)

PROBLEM 16.115: 1/5 = 0.35 m/s = ; No = +0.35 m/s 98= 1.2 m/s = 5 ag=+1.2 m/s2

SUBSTITUTE IN BO(Z) P= 4, 4/45 + 2, 4417 (+1.2) +7.5 (+0.35)

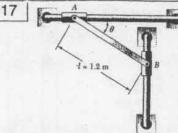
= 4,4145 + 2,9300 + 0,9188 = 18,263 N P= 8.26 N +

PROBLEM 16.116: RECALL WE ASSUMED POSITIVE TO LEFT VB= 0,35m/s -> 1 VB= -0,35m/s a=1.2 m/s -= ; a= -1.2 m/s2 SUBSTITUTE INTO EQ(2):

P = 4.4145 + 2.4412(-1,2) + 7.5 (-0.35) = 4.4145 - 2.9300 + 0.9188 = + 2.403 N

P= 2.40 N -





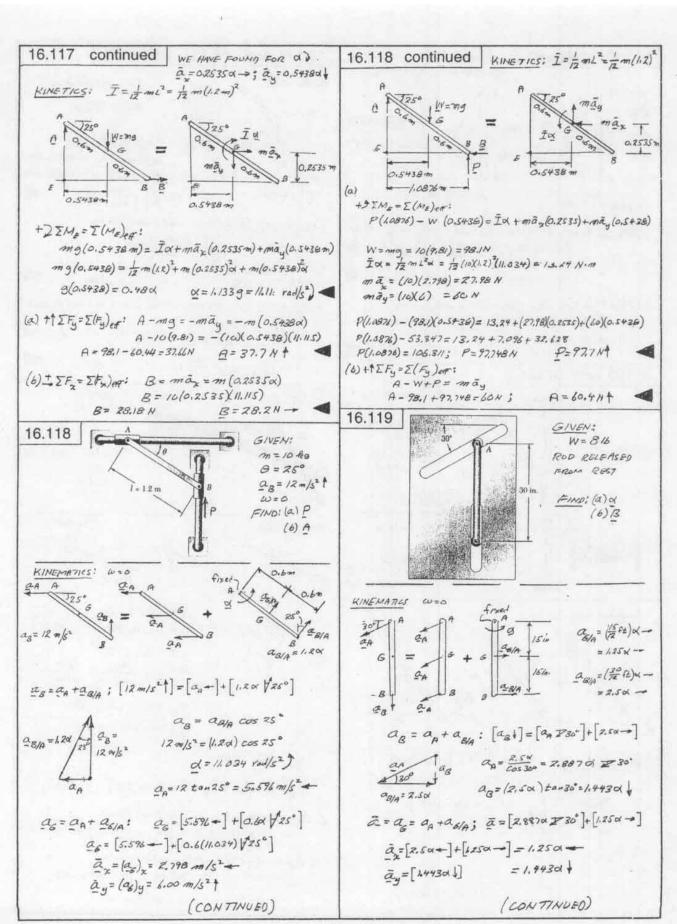
GIVEN; m=10-80 0 = 25° RELEASE AROM REST FIND! (a) A (b) B

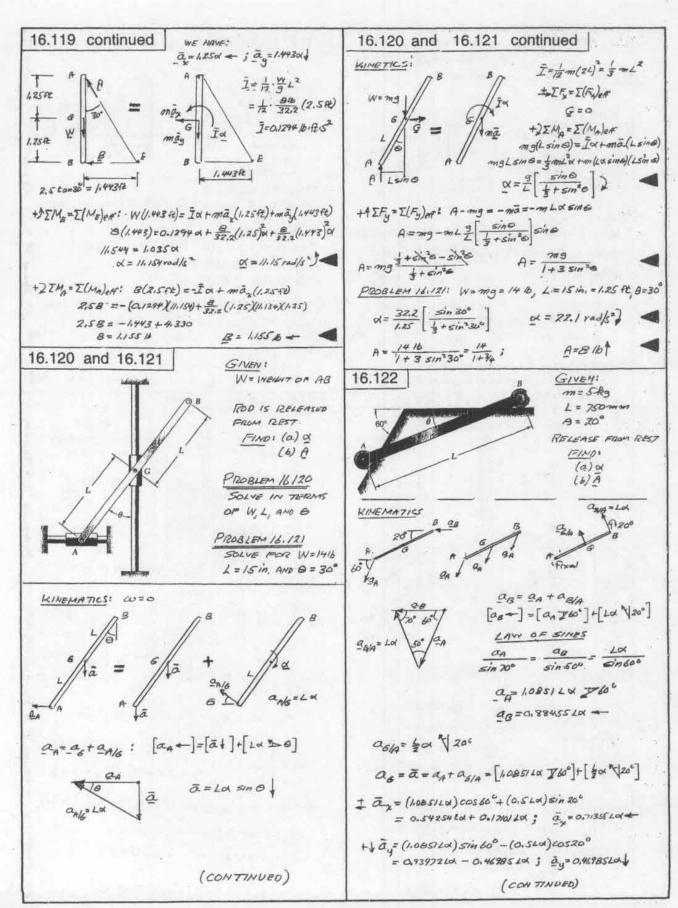
KINEMATICS: ASSUME &) W=0 0.6m a B/A=(1.2)d

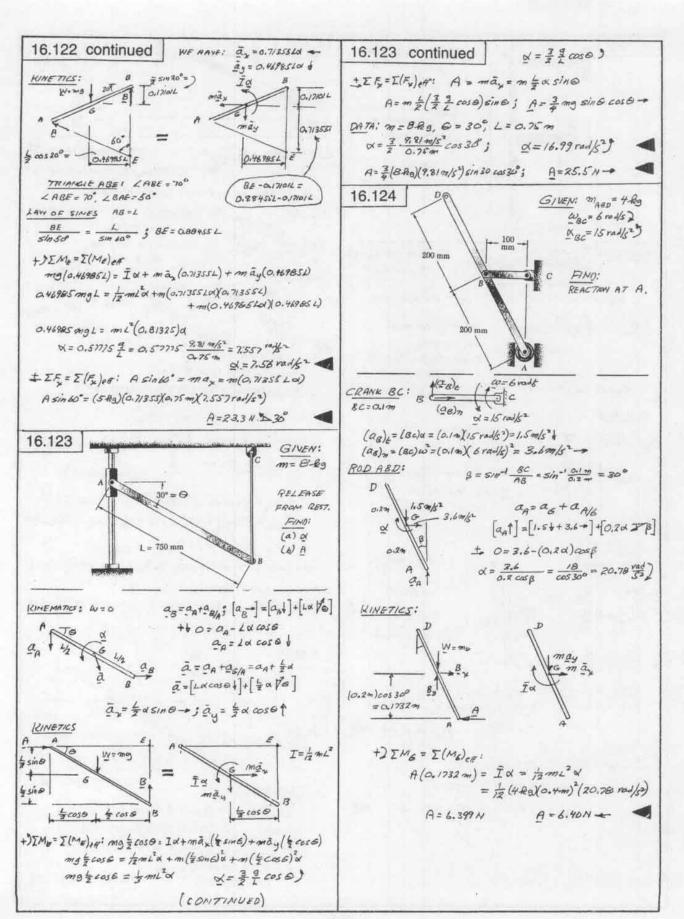
ast = a+ a = [a -]+[1.2 × 1250] an= (1.20) cos 25°=1.08764 aBIA=1,2d a=(1,2d) sin 25°= 0,5071d

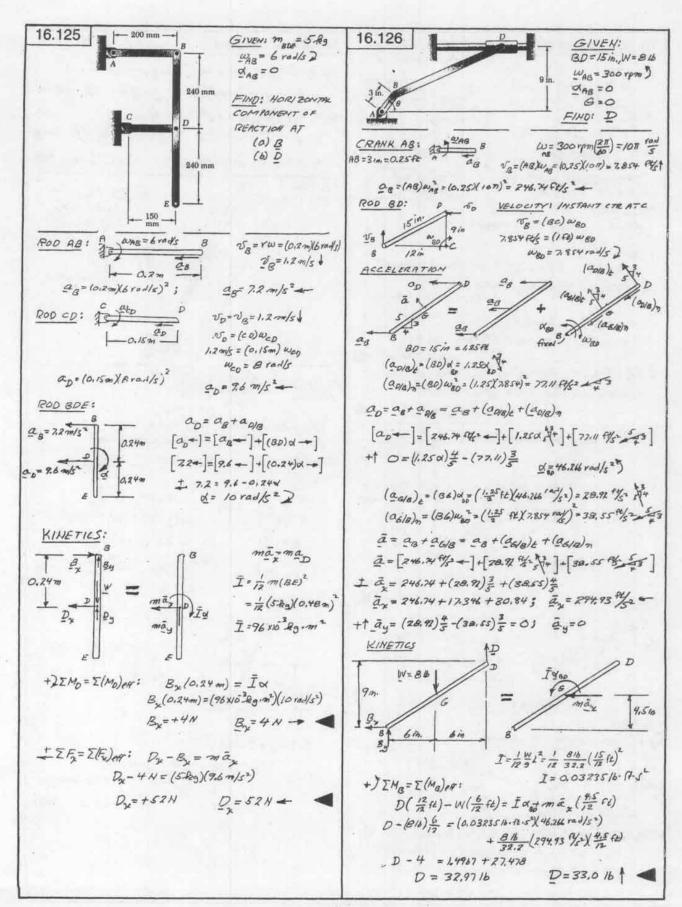
a= a+as/A=[a++]+[0.60 A 250] a= [0.507/d-+ + [0.60 A 25"] a= (a) = [0.5071 x -) +[0.25360 -] az= 0.2535 x -> ay=[0.60 cas250 =-]=0,543801

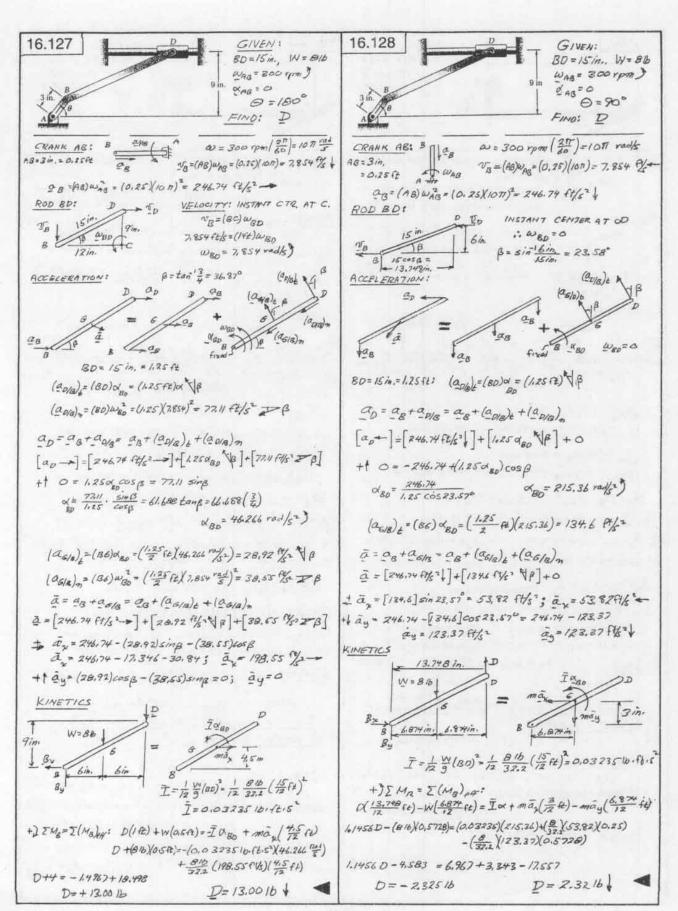
(CON TINUED)

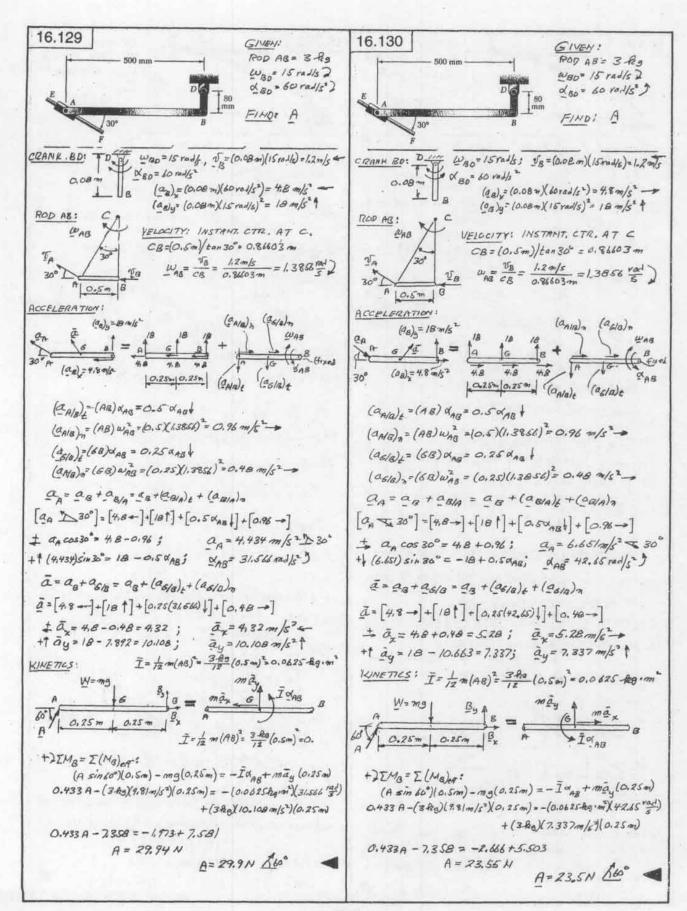


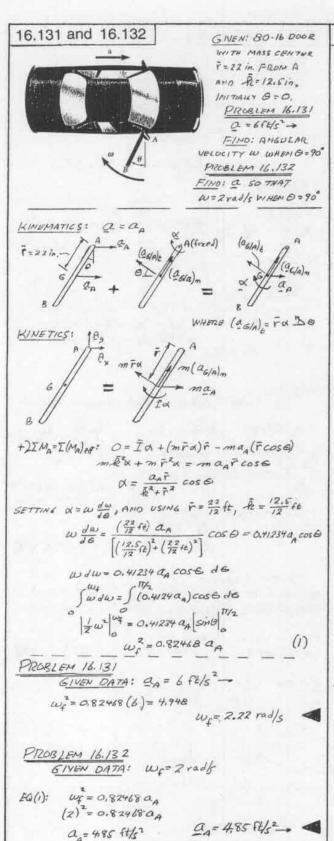


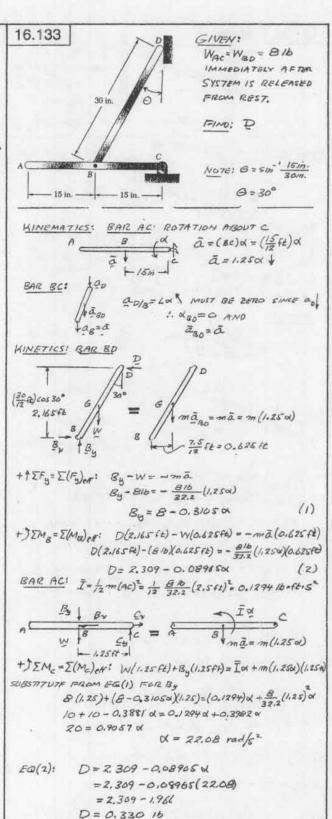




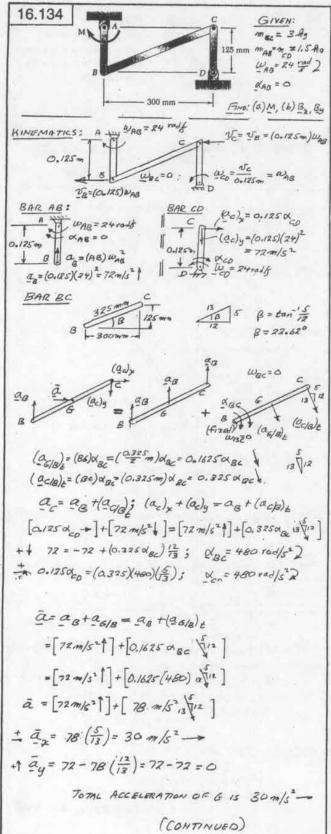


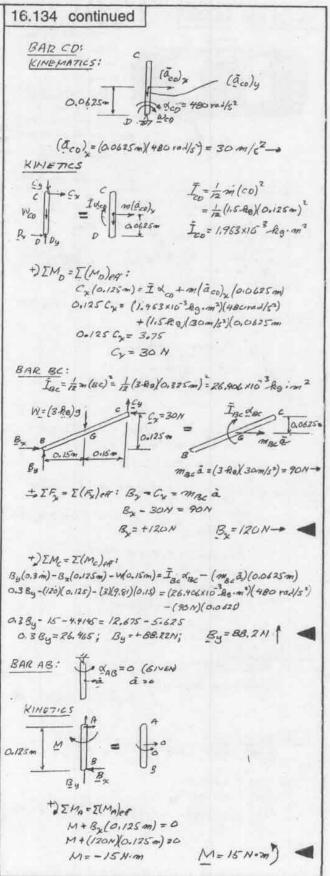


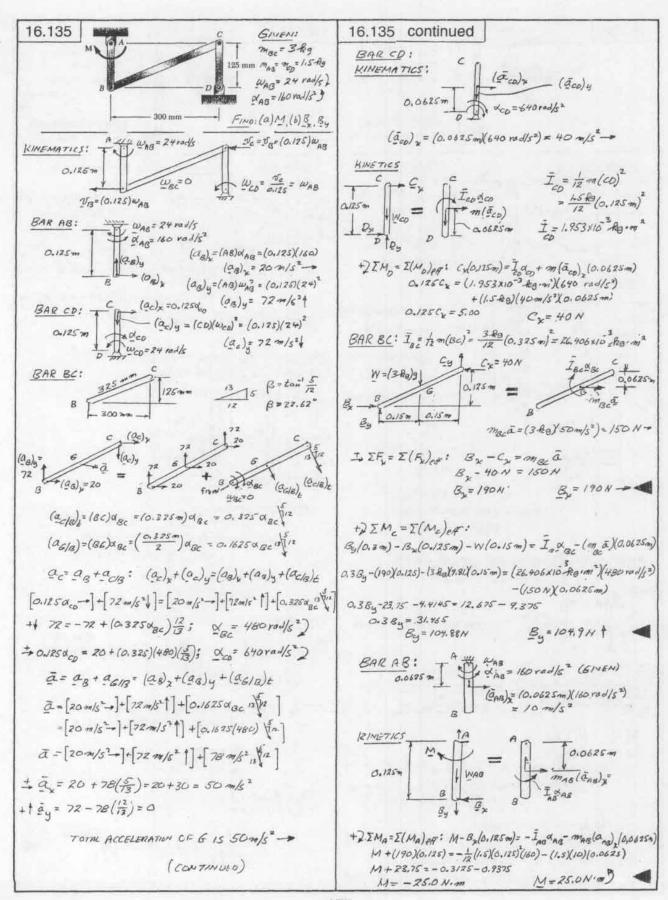


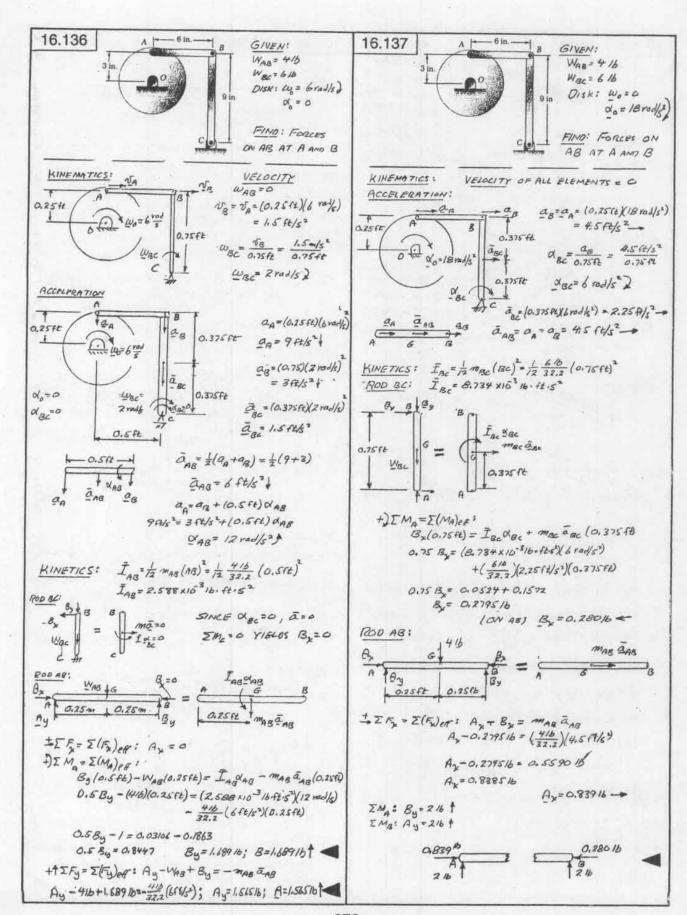


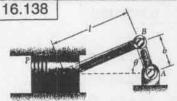
D=0.33016+











GIVEN: WAS = 600 PP]

R= 250 mm, b= 100 mm

MBO = 1,2 AB, mp = 1,8 AB

B= 180°

FINDS FORCES ON

BD AT B AND D

KINEMATICS: CRANK AB: $\omega_{AB} = \omega_{AB} = 600 \text{ rpm} \left(\frac{2\pi}{4\pi}\right) = 62.832 \text{ rad/s}^{\frac{1}{2}}$ $\omega_{AB} = (AB) w_{AB}^{2} = (0.1m)(62.632 \text{ rad/s})^{2}$ $\omega_{B} = 394.78 \text{ m/s}^{2}$

ALSO: NB = (AB) WAB = (0.100)(62, \$32 rad/s) = 6.2832 m/s + CONNECTING ROD BD:

INSTANT. CENTER AT D.

$$W_{BO} = \frac{V_B}{ED} = \frac{6.2832 \text{ m/s}}{0.25 \text{ m}} = 25.133 \text{ rad/s}$$

ACCELERATION:

$$\frac{a_{D}}{D} = \frac{a_{B}}{B} = \frac{a_{B}}{D} = \frac{a_{B}}{B} + \frac{a_{DB}}{D} = \frac{(free)^{3}}{D}$$

$$\frac{a_{D}}{D} = \frac{a_{B}}{B} + \frac{a_{DB}}{D} = \frac{(g_{B})}{D} + \frac{(g_{B})}{D} = \frac{g_{B}}{D} = \frac{g$$

KINETICS OF PISTON

$$\frac{P}{\sqrt{D}} = \frac{P}{\sqrt{D}} = \frac{m_P a_D = (1.8 \text{ Rg})(236.86 \text{ m/s}^2)}{D = 426.35 \text{ N}}$$

FORCE EXERTED ON COMMECTING ROD AT D is:

KINETICS OF CONNECTING ROD: (NEGLECT WEIGHT)

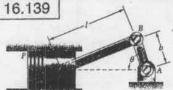
$$D=426.35N$$

$$D=\frac{B}{D}$$

$$D=\frac{B}{D}$$

$$D=\frac{B}{D}$$

$$D=\frac{B}{D}$$



GIVEN: WAG=600 rpm)

\$=250 mm, b=150 mm

mg=1.2 feq, mp=1.8 feq

B=0

FIND: FORCES ON

BD AT B AND D

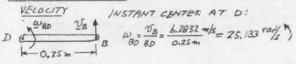
KINEMATICS: CIZANK AB:

QB WAB WAB = 600 rpm, (217) = 62.832 val/s)

B -0.1m A aB = (AB)WAB = (0.1 m)(62.832 val/s)

aB = 394.78 m/s^2 -

ALSO: Ng= (AB) WAB = (OIM) (62,882 Yad/s) = 6,2832 m/s + CONNECTING ROO BD1



ACCELERATION

$$\frac{a_{D}}{D} = \frac{a_{B}}{B} = \frac{a_{B}}{D} = \frac{a_{B}}{B} + \frac{a_{O/B}}{D} = \frac{(fix a)}{B}$$

$$\frac{a_{D}}{D} = \frac{a_{B} + a_{O/B}}{B} = [a_{B} \rightarrow] + [(BO)\omega_{ab}^{2} \rightarrow]$$

$$\frac{a_{D}}{A} = [394.78 \text{ m/s}^{2} \rightarrow] + [(0.25 \text{ m})(25.133 \text{ ra.}^{4/s}) \rightarrow]$$

$$Q_{D} = [394.78 \text{ m/s}^{2} \rightarrow] + [157.92 \text{ m/s}^{2} \rightarrow] = 552.70 \text{ m/s}^{2} \rightarrow$$

$$\frac{a_{D}}{B} = \frac{1}{2}(a_{B} + a_{D}) = \frac{1}{2}(394.78 \rightarrow + 552.70 \rightarrow) = 473.74 \text{ m/s}^{2} \rightarrow$$

KINETICS OF PISTON

FORCE EXERTED ON CONNECTING ROD AT D is:

D= 994.86N -

KINETICS OF CONNECTING ROD (NEGLECT WEIGHT)

$$D = 994.86N$$

$$B = D$$

$$D = D$$

$$E = D$$

$$E = D$$

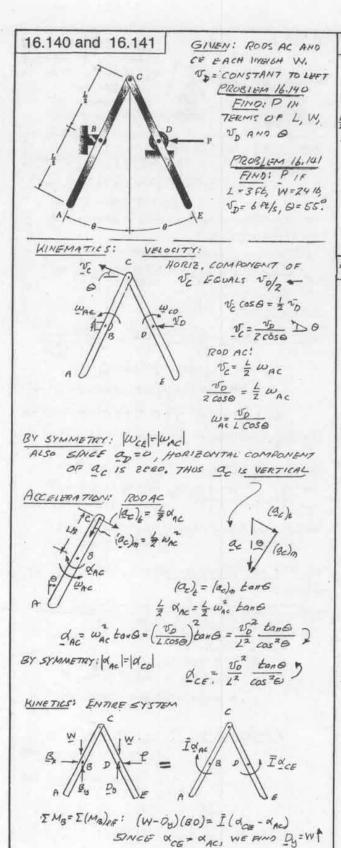
$$E = D$$

$$E = 0$$

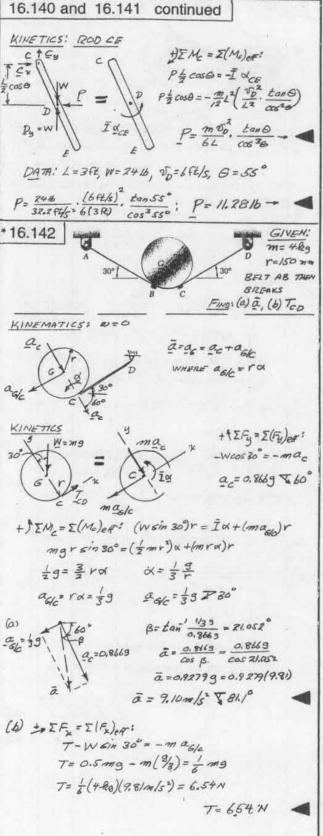
$$E = 994.86N = (1.2 ftg)(473.74m/s^2)$$

$$E = 994.86N + 568.44N = 1563.3N$$

FORCES ACTING ON CONNECTING ROD



(CONTINUED)







GIVEN:

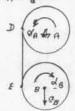
DISK OF MASS TO AND RADIUS Y PIN AT C IS REMOVED

FINDS

(a) da amo da (b) TENSION IN CHAIN

(a) aB

KINEMATICSI WA=WR=0



$$a_{b} = rd_{A} + a_{E} = a_{D} = rd_{A} + a_{E} = (rd_{A} + rd_{B}) + a_{E} = r(d_{A} + d_{E}) + a_{$$

KINETICS: DISKA:

+)
$$EM_A = E(M_A)eR$$
:
 $Tr = \overline{I}d_A$
 $Tr = \frac{1}{2}mr^2d_A$
 $d_A = \frac{2T}{mr^2}$) (1)

DISK B:

+) [Mg = [(Mg)ef1 Tr= IX Tr= 1mr dp d8=272 (2)

FROM (1) AND (2) WE NOTE THAT dA = dB +) IM= [(ME) + F: Wr = Id= (mag) r Wr= 1 mrdg + mr(d4+dg)r

Wr= 5 m 200

VR=====)

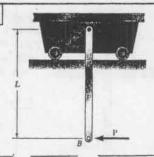
SUBSTITUTE FOR dA INTO (1):

$$\frac{Z}{S} = \frac{Z}{r} = \frac{ZT}{mr} \qquad T = \frac{1}{5} mg$$

$$\alpha_{B} = r(\alpha_{A} + \alpha_{B}) = r(2\alpha_{A}) = 2r(\frac{2}{5}\frac{9}{7})$$

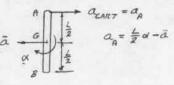
$$\alpha_{B} = \frac{4}{5}9\sqrt{\frac{2}{5}}$$

*16.144



GIVEN! CART OF MASS M ROD OF MASS on CART AT REST WHEN P IS APPLIED FIND: aA aB

KINEMATICS:

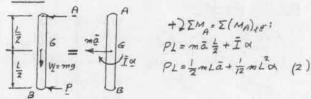


KINETICS: CART

$$A = max = m(\frac{L}{2}d - \bar{a})$$

$$\pm \sum_{k} \sum$$

ROD AB:



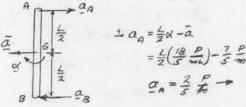
+ IFy = E/Fx)ex: P+A=mā P+m(= x-a)=ma FROM (1) (3) P = 2m a = 1 m L a

MULTIPLY (3) BY 1: 1-PL = 1 mL a - 12 mL d (4) ADD (2) AND(4): 7 PL = 5 mLa a= 7 P +

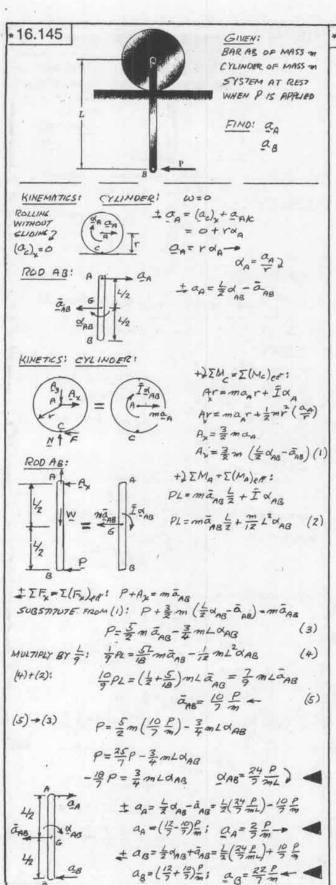
SUBSTITUTE (5) IN TO (3):

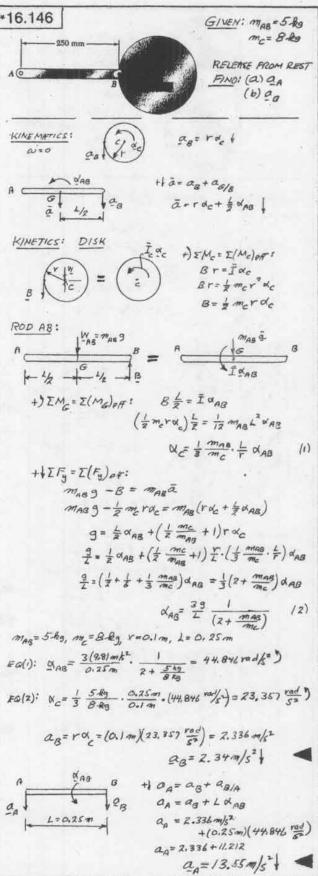
$$P = 2m\left(\frac{7}{5}\frac{P}{m}\right) - \frac{1}{2}mL \propto$$

0= 18 P 7

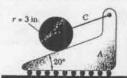


= a = = = = = = = $=\frac{1}{2}\left(\frac{18}{5}\cdot\frac{P}{mL}\right)+\frac{7}{5}\frac{P}{m}$ a= 16 P









GIVEN: WA = 6/B

WA = 4/B

AFTER CORD IS CUT

CYLINDER ROLLS,

FIND: (a) QA

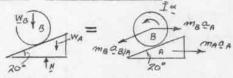
(b) X

KINEMATICS! WE RESOLVE aB INTO a AND A COMPONENT

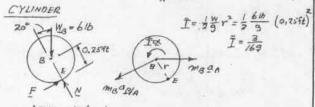


QB = QA + ABJA
WHERE ABJA = YA, SINCE THE
CYLINDER ROLLS ON WEDLE A.
ABJA = (0.25fb) &

KINETICS: CYLINDER AND WEDGE



 $\pm \sum F_{\chi} = \sum (F_{\chi})_{ef} : O = m_{A} a_{A} + m_{B} a_{A} - m_{B} a_{B|_{A}} \cos 20^{\circ}$ $O = \frac{(4+6)!6}{3} a_{A} - \frac{6!6}{3} \left(\frac{3}{12} f_{2}\right) \times \cos 20^{\circ}$ $a_{A} = (0.15 \cos 20^{\circ}) d. \qquad (1)$



+) [M= = [(M=)ex: (616) sin 20° (0.2512) =] 0 + mg o Bla (0.25ft) -mg o cos 20° (0.25ft)

1.5 sin 20° = $\frac{3}{16(32.2)}d_1 + \frac{616}{32.2}(0.25d)(0.25) - \frac{616}{32.2}d_A \cos 26^6(0.25)$

a51303 = 0.005824 + 0.01165d - 0.04378 ap

SUBSTITUTE FROM (1):

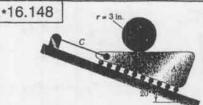
0.51303 = 0.01747 x - 0.04378 (0.15 cos 20") x

0.51303 = (0.01747 -0.00617) &

01 = 45.41 rad/62

d = 45.4 rad/s")

EQ(1): $a_{\mu} = (0.15 \cos 20^{\circ}) d$ = $(0.15 \cos 20^{\circ})(45.41)^{\circ}$ $a_{\mu} = 6.401 \text{ ft/s}^{2}$ $a_{\mu} = 6.40 \text{ ft/s}^{2}$



GIVEN: WB = 6 15

WA = 4 16

AFTER CORD IS CUT

CYLINDER ROLLS

FINO: (a) AA

(L) d

KINEMATICS! WE RESOLVE QB INTO QA AND A HORI-

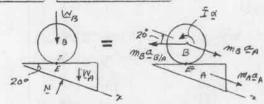
Paya B QA

(NHERE CL BYA = Y d, SINCE THE
CYUNOER B ROLLS ON WEDGE A.

BYA = (0,2512)01

a = a + a B/A

KINETICS: CYLINDER AND WEDGE:



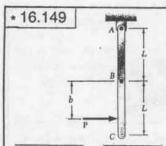
 $\begin{array}{l} +\sum_{k} =\sum_{k} (F_{k})_{eK}; \\ (W_{A}+W_{B})\sin 20^{\circ} = (m_{A}+m_{B})a_{A}-m_{B}a_{B|A}\cos 20^{\circ} \\ (10.16)\sin 20^{\circ} = (\frac{10}{3})a_{A}-(\frac{6}{3})(0.25\alpha)\cos 20^{\circ} \\ a_{A}=g\sin 20^{\circ} +\frac{6}{10}(0.25)\cos 20^{\circ} d \\ a_{A}=g\sin 20^{\circ} +0.15\cos 20^{\circ} d \end{array}$ (1)

 $(YLINDER: +) IM_{\bullet} = \Sigma(ME)_{eff}$ $O = \bar{I} \propto + (m_{\bullet} \alpha_{B/A})(0.254t) - (m_{\bullet} \alpha_{A} \cos 20^{\circ})(0.254t)$ $O = \frac{1}{2} \frac{6/b}{3}(0.254t) + \frac{6b}{3}(0.25a)(0.25) - \frac{6/b}{3} \alpha_{A} \cos 20^{\circ}(0.25)$ $O = \frac{1}{3} \Big[0.1875 + 0.375 + 1.4095 \alpha_{A} \Big]$ $Q = 0.5625 + 1.4095 \alpha_{A}; \quad d = 2.506 \alpha_{A} \qquad (2)$

SUBSTITUTE FOR & FROM (2) INTO (1): $a_q = g \sin 20^\circ + 0.15 \cos 20^\circ (2.506 a_A)$ $a_q = 11.013 + 0.3532 a_A$ $(1-0.3532) a_A = 11.013$

an= 17.027 ft/s2 an= 17.03 ft/s2 \$20°

FO(2) $\forall = 2.506 \, \alpha_{\rm R}$ = 2.506 (17.027) $\forall = 42.7 \, nod/6^2$ $(x = 42.7 \, rod/6^2)$



GIVEN: P= 20 N MAB= MBC = M = 3 As L = 500 mm b = L = 500 mm

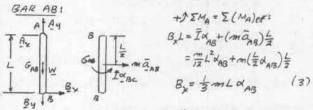
FIND! XAB AND DEC.

LINEMATICS! ASSUME d_{AB} , d_{BC} , AND $\omega_{AB} = \omega_{BC} = 0$ LIZ $AB = \frac{1}{2} d_{AB}$ $AB = \frac{1}{2} d_{AB}$

KINETICS: BAR BC +
$$\int \Sigma M_g = \Sigma (M_g)_{eff}$$
:

 $B_g = \frac{1}{2} M_g = \Sigma (M_g)_{eff}$:

 $PL = \overline{I} M_{gc} + (m \overline{a}_{gc}) \frac{L}{2}$
 $I M_g = \frac{1}{2} M_{gc} + (m \overline{a}_{gc}) \frac{L}{2}$
 $I M_g = \frac{1}{2} M_{gc} + (m \overline{a}_{gc}) \frac{L}{2}$
 $I M_g = \frac{1}{2} M_{gc} + (m \overline{a}_{gc}) \frac{L}{2}$
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 $I M_g = \frac{1}{2} M_{gc} + (m \overline{a}_{gc}) \frac{L}{2}$
 $I M_g = \frac{1}{2} M_{gc} + (m \overline{a}_{gc}) \frac{L}{2}$
 $I M_g = \frac{1}{2} M_{gc} + (m \overline{a}_{gc}) \frac{L}{2}$



ADD (2) AND(3): $P = \frac{4}{3}mL \alpha_{AB} + \frac{1}{2}mL \alpha_{BC}$ (4) SUBTRACT (1) PROM (4)

0 = 5 ml dag + 5 ml dgc

SUBSTITUTE FOR MECIN (1): dBc=-5 dAB (5)

P= = 1 ml dAB + 3 ml (-5 dAB) = - 7 ml dAB

 $AB = -\frac{6}{7} \frac{P}{mL} \qquad (6)$

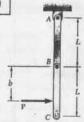
IEQ(S) $\alpha_{BC} = -5\left(-\frac{6}{7}\frac{P}{mL}\right)$ $\alpha_{BC} = \frac{30}{7}\frac{P}{mL}$ (7)

DATA: L=0.5m, m=3kg, P=204

EG(6): $d_{AB} = -\frac{6}{7} \frac{20 \text{ N}}{(3 \cdot 20)(0.5 \text{ m})} = -11.43 \text{ rod/s}^2$ $d_{AB} = 11.43 \text{ rad/s}^2$

 $Eo(7): \alpha_{8c} = \frac{30}{7} \frac{20N}{(3 Re)(0.5 m)} = 57.14 rad/s^{2}$ $\alpha_{8c} = 57.1 rad/s^{2}$

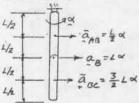
* 16.150



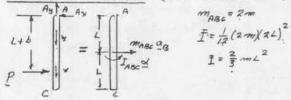
GIVEN: P=20N MAB=MBC=M= 3-R9 L=500mm

FINDI (a) DISTANCE B FOR WHICH BAIS MOVE AS A SINGLE RIGIN BODY (b) & OF BARS

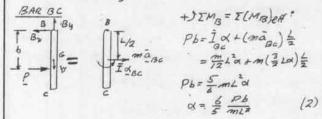
KINEMATICS! WE CHOOSE & = dAB = dBC)



KINETICS: BARS AB AND BC (ACTINE AS RIGIN BODY)



+) $\sum M_{A} = \sum (M_{A})_{eA}$: $P(L+b) = \sum_{ABC} x + m_{ABC} x_{B} L$ $P(L+b) = \frac{2}{3} m_{L}^{2} x + (2m)(Lx)L$ $P(L+b) = \frac{8}{3} m_{L}^{2} x$ (1)



SUBSTITUTE FOR & INTO (1)

$$P(L+b) = \frac{8}{3} mL^2 \left(\frac{b}{5} \frac{\rho b}{mL^2}\right)$$

 $PL + Pb = \frac{16}{5}Pb$; $L = (\frac{16}{5}-1)b = \frac{11}{5}b$

b= 51

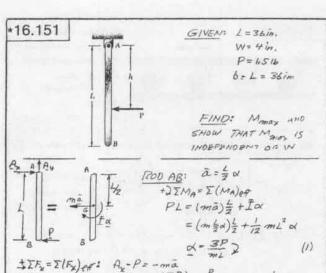
 $EQ(2) \propto = \frac{6}{5} \frac{P}{mL^2} \left(\frac{5}{11} L \right) \propto = \frac{6}{11} \frac{P}{mL}$

DATAI L= 0.5m, m = 3 kg, P= 20 N.

(a)
$$b = \frac{5}{11}L = \frac{5}{11}(500 \text{ mm}); b = 227 \text{ mm}$$

(6)
$$\forall = \frac{6}{11} \frac{P}{mL} = \frac{6}{11} \frac{20N}{(3R_0 \times 0.5m)} = 7.273 \text{ rad/s}^2$$

$$\forall = 7.27 \text{ rad/s}^2$$



PORTION AS OF ROD:

EXTERNAL FORCES: Ax. WAL, AXIAL FORCE FJ.,

SHEAR VJ, AND BENDING MUMENT MY

EFFECTIVE FORCES! SINCE ACCELERATION AT MY

POINT IS PROPORTIONAL TO DISTANCE FROM A, EFFECTIVE

FORCES ARE LIMEARLY DISTRIBUTED. SINCE MASS PER

UNIT LENGTH IS 71/L, AT POINT V WE FIND

Ax==P -

$$A = \frac{1}{2}$$

$$X = \frac{1}{2}$$

Az= P-m= d= P-m= (3P)=-=;

$$+ \chi \sum M_{j} = \sum (M_{j})_{p,p} : M_{j} - A_{\gamma} \chi = -\frac{1}{2} \left(\frac{3P\chi}{L^{2}} \right) \chi \left(\frac{2\chi}{2} \right)$$

$$M_{j} = \frac{1}{2} P\chi - \frac{1}{2} \frac{P}{L^{2}} \chi^{3} \qquad (2)$$

For
$$M_{\text{may}}$$
:
$$\frac{dM_{\text{j}}}{d\chi} = \frac{P}{Z} - \frac{3}{Z} \frac{P}{L^2} \chi^2 = 0$$
$$\chi = \frac{L}{\sqrt{3}} \tag{3}$$

SUBSTITUTING INTO (2)

$$(M_J)_{max} = \frac{1}{2} \frac{PL}{V_3} - \frac{1}{2} \frac{P}{L^2} \left(\frac{L}{V_3}\right)^3 = \frac{1}{2} \frac{PL}{V_3} \left(\frac{2}{3}\right)$$

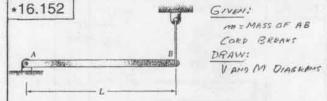
$$(M_J)_{max} = \frac{PL}{3V_3}$$

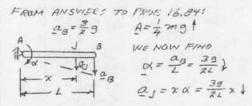
$$(4)$$

NOTE: EQS. (3) AND (4) ARE INDEPENDENT OF W

$$EG(3)$$
: $\chi = \frac{L}{V3} = \frac{36 \, \dot{m}}{V3} = 20.28 \, \dot{m}$

Mmax = 10.39 16-in. 20.8 in. BELOWI A.





PORTION AJ OF RODS

EXTERNAL FORCES: REACTION A, DISTRIBUTED LOAD PER UNIT LENGTH MB/L, SHEAR VI, BENNING MOMENT MJ.

EFFECTIVE FORCES: SINCE a ~ 4, THE EFFECTIVE FORCES ARE LINEARLY DISTRIBUTED. THE EFFECTIVE FORCE PER UNIT LENGTH AT J IS:

$$\frac{2n}{L} \alpha_{J} = \frac{2n}{L} \chi = \frac{3mg}{2L^{2}} \chi$$

$$\frac{1}{L} \alpha_{J} = \frac{2n}{L} \chi = \frac{3mg}{2L^{2}} \chi$$

$$\frac{1}{L} \alpha_{J} = \frac{2n}{L} \chi = \frac{3mg}{2L^{2}} \chi$$

$$\frac{1}{L} \alpha_{J} = \frac{2n}{L} \chi = \frac{3mg}{2L^{2}} \chi$$

$$+V \sum F_{g} = \sum \left(F_{g} \right)_{eff} : \frac{m3}{L} \chi - \frac{m9}{4} + V_{J} = \frac{1}{2} \left(\frac{3m9}{2L^{2}} \chi \right) \times V_{J} = \frac{m9}{4} - \frac{m9}{L} \chi + \frac{3}{4} \frac{m9}{L^{2}} \chi^{2} + \frac{1}{2} \sum \left(\frac{3m9}{2L^{2}} \chi \right) \times \left(\frac{V}{3} \right) \times V_{J} = \frac{m9}{4} \chi - \frac{1}{2} \frac{m9}{L^{2}} \chi^{2} + \frac{1}{4} \frac{m9}{L^{2}} \chi^{3} \times V_{J} = \frac{1}{4} \frac{m9}{L^{2}} \chi^{2} + \frac{1}{4} \frac{m9}{L^{2}} \chi^{3}$$

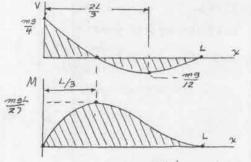
FINO
$$V_{man}$$
: $\frac{dV}{dx} = -\frac{mg}{L} + \frac{3}{2} \frac{mg}{L^2} x = 0$; $x = \frac{2}{3}L$

$$V_{min} = \frac{mg}{4} - \frac{mg}{L} (\frac{2}{3}L) + \frac{3}{4} \frac{mg}{L^2} (\frac{2}{3}L)$$
; $V_{min} = -\frac{mg}{L^2}$

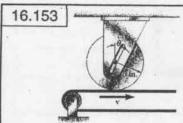
FIND M_{max} WHERE
$$V_j = 0$$
: $V_j = \frac{mg}{4} - \frac{mg}{2}x + \frac{3}{4}\frac{mg}{2}x^2 = 0$
 $3x^2 - 4Lx + L^2 = 0$
 $(3x - L)(x - L) = 0$ $x = \frac{L}{3}$ AND $x = L$

$$M_{may} = \frac{mq}{4} \left(\frac{L}{3} \right) - \frac{1}{2} \frac{mq}{L} \left(\frac{L}{3} \right)^2 + \frac{1}{4} \left(\frac{mq}{L} \right)^2 \left(\frac{3}{3} \right)^3 = \frac{mq}{27}$$

$$M_{min} = \frac{mq}{4} L - \frac{1}{2} \frac{mq}{L} L^2 + \frac{1}{4} \frac{mq}{2} L^3 = 0$$

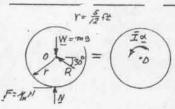


May 27 AT & FROM A



GIVEN: G= 30° M= 0.20

FIND & WHILE



 $\pm \sum F_{N} = \sum (F_{N}) e F$ $+KN - R\cos\theta = 0$ $R\cos\theta = 4KN \quad (1)$ $+ \sum F_{N} = 2[F_{N}) e F$ $R\sin\theta + N - mg = 0$ $R\sin\theta = mg - N \quad (2)$

DINOE(2)8Y(1): tand = mg-N ; 0.5774= mg-N 0.2N 0.1155N= mg-N; N= mg = 0.8965 mg

+) [M = Z(Mo)ex: +/2 N T = I X (0.2)(0.8965mg) T = \frac{1}{2} mr 2

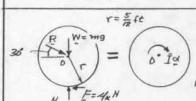
X = 0.35858 \frac{9}{V} = 0.35858 \frac{32.254/5^2}{(5/17 ft)} = 27.71 \frac{ra-1/2}{5}

X = 27.7 rad/52)

16.154

GIVEN: 6 = 300 4x = 0.20

FINDI & WHILE SLIPPING OCCURS



 $\pm \sum F_x = \sum (F_x)_{xx}!$ $R\cos \theta - M_x N = 0$ $R\cos \theta = -M_x N \quad (1)$ $+ \sum F_y = \sum (F_y)_{xx}!$ $R\sin \theta + mg - N = 0$ $R\sin \theta = N - mg \quad (2)$

DIVIDE (2) BY(1) 1 tan 8 = N-m9 ; 0.5774 = N-m9

0.1155N=N-mg: N= mg = 1.1306 mg

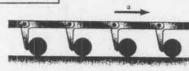
+) IMo = Σ(Mo)eff: 4/κNr = Īα (0.2)(1.1306mg) r = -mr²α

(x = 0.4522 \frac{g}{r} = 0.4522 \frac{32.7 (4/52)}{(5/0.46)}

X = 34,948 vad/5"

X=34,9 rad/s")

16.155



GIVEN: CYLINDERS

FIND: (0-) MAXIMUM Q

FOR ROLLING WITH

IND SLIDING

(W) MINIMUM Q

FOR CYLINDER TO MOVE - WITH NO ROTHTING

(a) CYLINDER ROLLS WITHOUT SLIDING: a=ra or a= =

$$\frac{B_{b}=4P}{B_{b}=p} \cdot \frac{B}{B} \cdot \frac{B}{B} = 0$$

PIS HORZ.

COM POWENT OF

FORCE ARM

EXERTS ON CYLHOR

+) $\sum M_{\rho} = \sum (M_{\rho})_{eq} : Pr - (N_{\rho}P)_{r} = \sum \alpha + (m\bar{\alpha})_{r}$ $P(1-q)_{r} = \sum mr(\bar{\alpha})_{r} + (m\bar{\alpha})_{r}$

$$P = \frac{3}{2} \frac{m\bar{a}}{(1-d)} \tag{1}$$

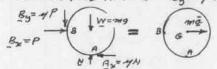
 $+f \Sigma F_y = 0! N - y, P - mg = 0$ (2)

 $\Rightarrow \Sigma F_{x} = \Sigma (F_{x})_{eff}: P - \Psi N = ma$ SOLVE(2) FOR N AND SUBSTITUTE FOR N INTO (2). $P - \Psi^{2}P - \Psi mg = ma$ (8)

SUBSTITUTE P FROM (1): $(1-q^2)\frac{3}{2}\frac{m\bar{a}}{(1-q)}-q mg=m\bar{a}$ $3(1+q)\bar{a}-2+q=2\bar{a}$

(b) CYLINDER TRANSLATES: X=0

ASSUME SCIOING IMPENIS AT A: Ax = MY P



+) $\Sigma M_{A} = \Sigma (M_{A})_{eff}$: $Pr - gPr = (m\bar{a})r$ $P(i-q)r = m\bar{a}r$ $p = m\bar{a}$

 $P = \frac{m}{1-q}$ $\pm \sum_{k=1}^{\infty} \sum_{k=1}^{\infty} (F_{k})_{k+1} P - y_{k} = m\bar{a}$ (5)

+ 1 E Fy = [(Fy) eft | N - 4/P - mg = 0 (6) Solve (5) FOR N AND SUBSTITUTE FOR N INTO (6).

SUBSTITUTE FOR P FROM (4):

ma (1-42) - 4mg = ma

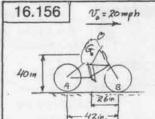
P-Py2-ymg=ma

ā(1+4)-49=ā
ā4-49=0

ā=9

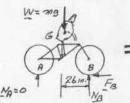
SUMMARY: a < 24 9: ROLLING

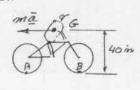
1+34 9 < 0 < 9 : RETATING AND BLIDING



FIND I SHORTEST STOPPING DISTANCE IF CYCLIST IS NOT TO BE THROWN OVER FRONT WHEEL

WHEN CYCLIST IS ABOUT TO BE THROWN OVER THE FRONT WHEEL, NA =0



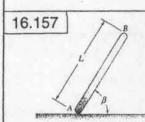


+) $IM_B = I(M_B)_{eff}$: $mg(26in.) = m\bar{a}(40in.)$ $a = \frac{26}{40}g = \frac{26}{40}(32.25t/s^2) = 20.93 ft/s^2$

UNIFORMLY ACCES FRATED MOTIONS

1/6 = 20 mpb = 29,333 FH/52 152-163 = 205: 0-(29,233 FH/5) = 2(-20.93/5)5 5=20.555 ft

5=20.6 St



GIVENI \$=70° UNIFORM ROD

RELEASED FROM REST.

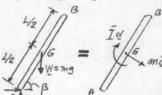
FRICTION IS SUFFICIENT

TO PREVEND SLIDING AT A.

FIND: (a) &

(b) NA, (x) FA

WE NOTE ROO ROTATES ABOUT A. WED



+) $\Sigma M_A = \Sigma (M_A) e \varphi$: $mg(\frac{1}{2} \cos \beta) = \tilde{I} \propto + (m\tilde{a}) \frac{1}{2}$ $\frac{1}{2} mg L \cos \beta = \frac{1}{12} mL^2 \alpha + (m \frac{1}{2} \alpha) \frac{1}{2}$ $= \frac{1}{3} mL^2 \alpha$

 $\alpha = \frac{3}{2} \frac{g \cos \beta}{L} \tag{1}$

I = 1/2 ml2

 $\pm \sum F_2 = \sum (F_2)eF$: $F_A = m\bar{a} \sin \beta$ $F_A = m \frac{1}{2} \alpha \sin \beta = m \frac{1}{2} \left(\frac{3}{2} \frac{9\cos \beta}{L}\right) \sin \beta$ $F_A = \frac{3}{4} mg \sin \beta \cos \beta$ (2)

(CONTINUED)

16.157 continued

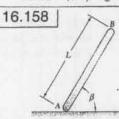
+† $IF_y = I(F_y)_{eff}$: $N_q - mg = -m\bar{\alpha}\cos\beta = -m(\frac{1}{2}\alpha)\cos\beta$ $N_q - mg = -m\frac{1}{2}(\frac{3}{2}\frac{3\cos\beta}{L})\cos\beta$

 $N_{\beta} = mg(1 - \frac{3}{4}\cos^2\beta)$ (3

FOR $\beta = 70^{\circ}$; $\alpha = \frac{3}{2} \frac{g \cos 20^{\circ}}{L}$; $\alpha = 0.513.\frac{g}{L}$

(b) Fa(3): NA=mg(1-3/cas 70); NA=0.912mg+

(e) FA(2) FA = 3 mg sin70 cos 70; FA = 0.241 mg -



GIVEN: B=70°, UNIFORM
ROD RELEASED FROM REST.
FRICTION AT SURFACE
EQUALS ZERO,
FIND: (a) K, (b) ā,
(c) REACTION AT A.

 $+ V \Sigma F_y = \Sigma (F_y) e^{\frac{\pi}{4}};$ $mg - A = m\bar{\alpha}_y = m \left(\frac{L}{2}\alpha \cos\beta\right) \qquad (1)$ $V \Sigma M_G = \Sigma (M_G) e^{\frac{\pi}{4}};$

 $A\left(\frac{1}{2}\cos\beta\right) = \bar{I}\alpha = \frac{1}{12}mL^{2}\alpha$ $A = \frac{mL}{6}\frac{\alpha}{\cos\beta} \tag{2}$

SUBSTITUTE (2) INTO (1): $mg - \frac{mL}{6} \frac{\alpha}{\cos \beta} = m \frac{L}{2} \alpha \cos \beta$

 $g = \left(\frac{L}{2}\cos\beta + \frac{L}{6\cos\beta}\right) \propto$

 $g = \frac{2}{6} \left(3\cos\beta + \frac{1}{\cos\beta} \right) \times$

 $g = \frac{L}{L} \left(\frac{3\cos^2 \beta + 1}{\cos \beta} \right) \propto \frac{69}{L} \left(\frac{\cos \beta}{1 + 3\cos^2 \beta} \right)$

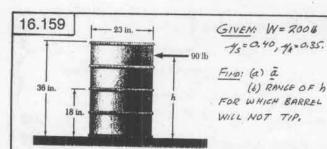
 $\hat{a} = \frac{L}{2} \propto \cos \beta = \frac{L}{2} \left(\frac{69}{L} \cdot \frac{\cos \beta}{1 + 3\cos^2 \beta} \right) \cos \beta = 39 \left(\frac{\cos^2 \beta}{1 + 3\cos^2 \beta} \right) = \frac{39}{6} \left(\frac{\cos^2 \beta}{L} \cdot \frac{\cos \beta}{1 + 3\cos^2 \beta} \right) \cos \beta = \frac{39}{1 + 3\cos^2 \beta}$

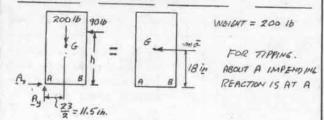
FOR B=70°:

(a) $d = \frac{69}{L} \frac{\cos 36^{\circ}}{1+3\cos^{2}70^{\circ}};$ $d = 1.519 \frac{9}{L}$

(6) \(\bar{a} = 39 \) \(\frac{\cas^270}{1 + 3\cas^270}\); \(\bar{a} = 0.260\) g \(\bar{q}\)

(a) A=mg 1 1+3cos2700 ; A=0.740 mg t





Ify = [(fy) or: Ay - 200 16 = 0; Ay = 200 16 t FOR SLIDING: Az = 1/2 Ay = 0.35(200) = 70 16 ->

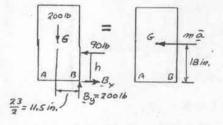
\$F = 5/6) = 20 16 -2

$$\frac{+\sum_{k=1}^{n}\sum_{(k)\neq k}: 901b-A_{2}=ma}{901b-701b=\frac{2001b}{5}\bar{a}}$$

$$\bar{a}=\frac{201b}{2001b}g=0.1g=0.1(32.2) \quad \bar{a}=3.22 Rl_{5}^{2}=4$$

+)
$$\Sigma M_A = \Sigma (M_A)_{eff}$$
: $(9016)h - (20016)(\frac{M_1S}{12}ft) = m\bar{a}(\frac{18}{12}ft)$
 $90h - 191.67 = \frac{70016}{32.2}(3.22ft/s)(1.5ft)$
 $90h - 191.67 = 30$; $90h = 221.67$
 $h = 2.463ft$ $h = 29.6 in$

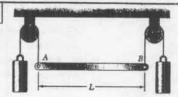
FOR TIPPING IMPENDING ABOUT B, REACTION IS AT B



+)
$$\Sigma M_B = \Sigma (M_B)_{eff}: (90h) + (2001b) \frac{N.S}{12}(1) = ma (\frac{18}{12} ft)$$

 $90h + 191.67 = \frac{2001b}{32.2} (3.22 ft/s^2) (1.5 ft)$

THUS RANGE FOR NO TIPPING IS h<29.6 in



16.160

BAR AB: WELDITS

BAR AB: W

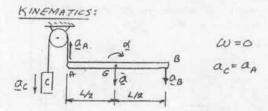
COUNTER WELDITS

= \frac{1}{2}W.

IMMEDIATELY

AFTER WIRE

FINO: (a) ap, (b) aB.



KINETICS: COUNTERWEIGHT M=MASS OF BAR AB

$$\frac{1}{2}W =$$

$$\frac{1}{2}W - T = m_{c}a_{c} = \frac{1}{2}ma_{A}$$

$$\frac{1}{2}m_{g} - T = \frac{1}{2}ma_{A}$$

$$T = \frac{1}{2}m(g - a_{A}) \qquad (1)$$

KINETICS BAR AB

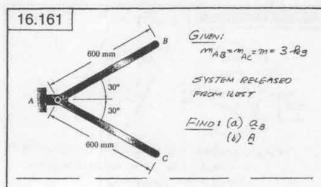
$$+ \int \Sigma M_g = \Sigma (M_g)_{pp}$$
: $T_{2}^{\prime} = \bar{I} \propto \frac{1}{2} m(g - a_{1}) \frac{1}{2} = \frac{1}{12} mL^{2} α$

$$3g - 3a_{1} = L α$$
(3)

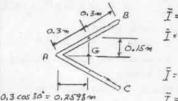
ADO EQS.(2) ANO(3):
$$4g = 2L\alpha$$
 $\alpha = \frac{2g}{L}$

$$\bar{a} = (\frac{1}{2} L \alpha - a_A) = \frac{1}{2} L(\frac{29}{L}) - \frac{1}{3}9; \qquad \bar{a} = \frac{2}{3}9 + \frac{1}{3}$$

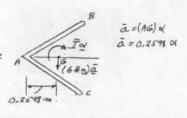
$$a_{B} = (L \times -a_{A}) = L \left(\frac{29}{L} - \frac{1}{3} 9 \right);$$
 $a_{B} = \frac{5}{3} 9$



CENTER OF MASS AND I'



$$\vec{I} = 2 \left[\vec{I}_{AB} + m_{AB} (0.15 m)^{2} \right]
\vec{I} = 2 \left[\frac{1}{12} (3 kg) (0.6 m)^{2} + (3 kg) (0.15 m)^{2} \right]
\vec{I} = 2 \left[0.09 kg m^{2} + 0.0675 kg m^{2} \right]
\vec{I} = 0.315 kg m^{2}$$



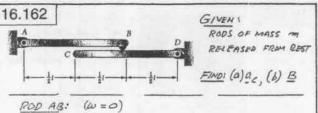
= 21.24 rad/52] a=0.2698(21.24)=5.518 m/62

 $\pm \sum F_{x} = 0: \quad A_{x} = 0$ $+ \uparrow \sum F_{y} = \sum (F_{y})_{AF}: \quad A_{y} - 2(3 + 9)(9.81 m/s^{2}) = -(6 + 9)\bar{x}.$ $A_{y} - 58.86 = -(6 + 9)(5.518 m/s^{2}).$ $A_{y} - 58.86 = -33.11 \quad A_{y} = 25.75 N.$

SINCE Av=0, A=25.8 HT



a = 12.74 9/5= \$ 60°



 $\frac{A_{x}}{A} + \frac{1}{A_{y}} + \frac{1}{A_{y}} + \frac{1}{A_{y}} = \frac{1}{A_{y}} + \frac{1}{A_{y}} +$

+) $\Sigma M_A = \Sigma (M_A)eA^+$ $mg(\frac{d}{Z}) - BP = \widetilde{L}\alpha + m\widetilde{\alpha}_{AB}(\frac{d}{Z})$ $\frac{1}{2}mgP - BP = \frac{1}{2}mL^2\alpha + m(\frac{d}{Z}\alpha_{AB})\frac{d}{Z}$ $\frac{1}{2}mgP - BP = \frac{1}{3}mP^2\alpha_{AB}$ (1)

ROD CD: $(\omega = 0)$ $C = \frac{B}{6} + \frac{\ell/2}{D} = C = \overline{D} \times \overline{D}$ $C = \frac{\overline{D} \times CD}{6 \times M} = \overline{D} \times \overline{D}$

 $m\bar{a}_{co} = m\frac{f}{2} \, \alpha_{co}$ +) $IM_D = I(M_0)eR$: $mg(\frac{f}{2}) + B(\frac{f}{2}) = \bar{I} \, \alpha_{co} + m \, \alpha_{co} \, \frac{f}{2}$ $\frac{1}{2} \, mg \, \ell + \frac{1}{2} B \ell = \frac{1}{12} \, m\ell \, \alpha_{co} + m(\frac{f}{2} \alpha_{co}) \, \frac{f}{2}$

MULTIPLY BY 2: $mgl + Bf = \frac{2}{3}mld_{c0} \qquad (2)$

ADD (1) AND (2): $\frac{3}{2}$ mg $\ell = m\ell^2 \left(\frac{1}{3} \alpha_{ab} + \frac{2}{3} \alpha_{co} \right)$ (3)

MULTRLY BY 3: $A_{AB} + 2 A_{CD} = \frac{9}{2} \frac{3}{2}$ (4)

KINEMATICS: WE MUST HAVE $A = \frac{1}{2} \Rightarrow B \Rightarrow CO \qquad (QB) = (AB)$ $C \Rightarrow B \Rightarrow CO \qquad (QB) = \frac{1}{2} \Rightarrow CO \qquad (S)$ $C \Rightarrow B \Rightarrow CO \qquad (AB) = \frac{1}{2} \Rightarrow CO \qquad (S)$

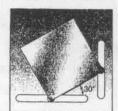
SV2STITUTE FOR α_{AB} FROM (S) INTO (4) $\frac{1}{2} \alpha_{c0} + 2\alpha_{c0} = \frac{9}{2} \frac{9}{R}$ $\frac{5}{2} \alpha_{c0} = \frac{9}{2} \frac{9}{R} ; \qquad \alpha_{c0} = 1.8 \frac{9}{R} \qquad (6)$

(a) ACCELERATION OF C: $a_{c} = l(1.8\frac{3}{4}); \quad a_{c} = 1.89 + 4$

(b) FORCE ON KNOB B: SUBSTITUTE FOR d_{CD} FROM (6) INTO (2) $mgl + Bl = \frac{2}{3} ml^2 (1.8 \frac{9}{4})$ B = 1.2 mg - mg

(ON ROD AB): B= 0.2 mg +

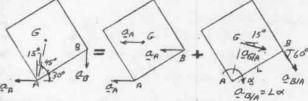
16.163



GIVEN: SQUARE PLATE
OF SIDE L= 150 00000
AND M= 2.5 Rg IS
RELEASED FROM REST

FIND (a) & (b) A

KINEMATICS:

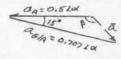


PLANE MISTON = TRAVISLATION + ROTATION

agl= an++ ag/4 760°

049= 100 5 in 30"
049= 100 5 in 30"

 $\bar{\alpha} = a_n + a_{6/n} = 15^\circ$ $\bar{\alpha} = La \sin 30 + \frac{La}{\sqrt{2}} = 15^\circ = 0.5 La + 0.707 La = 15^\circ$



1AW OF COSINBS

\$\bar{a}^2 = a_A^2 + a_{6/A}^2 - 2a_A a_{6/A} \cos /5^\end{a}^2 = (0.52\alpha)^2 + (0.7072\alpha)^2

-2(0.52\alpha)(0.7072\alpha) \cos /5^\end{a}^2 = 2\alpha'(0.75+0.5-0.683\alpha)

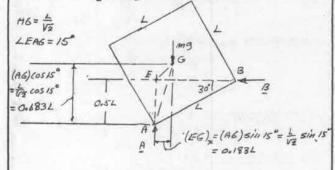
a=12 (0,06699); a=0.2588210

LAW OF SINES

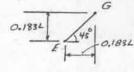
 $\frac{\bar{a}}{\sin s} = \frac{a_{6/4}}{\sin g}; \sin g = \frac{a_{6/4}\sin s}{\bar{a}} = \frac{0.70714}{0.2598214} \sin s$ $\sin g = 0.707; g = 135$ $\bar{a} = 0.2588 Ld \ \ 45$

KINETICS (W=0)

WE FIND THE LOCATION OF POINT E WHERE LINES OF ACTION OF A AND B INTERSECT.



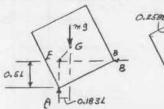
(EG) = 0.6829L - 0.5L = 0.183L



E6=(0.1834) V2 = 0.25881

(CONTINUED)

16.163 continued



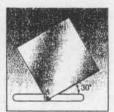
+) $EM_A = E(M_A)_{BB}$: $mg(0.183L) = \bar{I}\alpha + (ma)(0.2588L)$ $0.183mgL = fmL^2\alpha + m(0.2588La)(0.2588L)$ $0.183gL = L^2\alpha(f+0.06198)$

0.183 $\frac{9}{2} = 0.7336 \times 1$ $\alpha = 0.7834 \frac{9}{2}$ $\alpha = 0.7834 \frac{9}{3}$ $\alpha = 51.2 \frac{\text{rad/s}}{\text{5}^2}$

+ \$ EFg = I(Fg)eff: A - mg = - mā sin45" = -m(0.2588 Ld) sin45" = -m(0.2588 Ld) (0.7834 =) sin45"

A-mg = 0.1434m 9 A= 0.8566 mg= 0.8566 (2.540) (9.81 m/5") = 21.01X A= 21.01

16.164



GIVEN: SQUARE PLATE

OF SIDE L=150 mm

AND m=2.5 Ptg IS

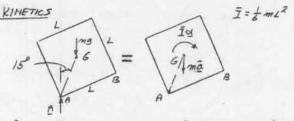
RELEASED FROM REST.

Fino: (a) & (b) A

700 = 6 170<u>0</u>)

SINCE BOTH A AND MY ARE VERTICAL, ax=0 AND & IS &

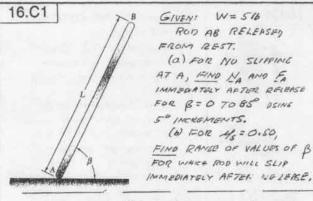
KINEMATICS $A6 = \frac{1}{113} \text{ Viso,} \quad a_{6/a} = (A6) \text{ d. } 15^{\circ}$ $a_{6/a} = \frac{1}{12} \text{ d.} \quad a_{6/a} = \frac{1}{12} \text{ d. } 15^{\circ}$ $a_{6/a} = \frac{1}{12} \text{ d. } 15^{\circ}$



+) $IM_{q} = I(M_{q})_{eff}$: $mg(A6)_{ein}_{is} = I\alpha + m\bar{\alpha}(A6)_{ein}_{is}$ $mg(\frac{L}{\sqrt{2}})_{ein}_{is} = \frac{1}{6}mL^{2}\alpha + m(0.183La)(\frac{L}{\sqrt{2}})_{ein}_{is}$ $\alpha.183\frac{3}{2} = (\frac{1}{6} + 0.033494)$

0.183 = 0.2002 d; d=0.943 = 0.9143 \frac{9}{2} = 0.9143 \frac{9.81 m/s^2}{0.15 =}

 $\begin{array}{c}
(A = 57.8 \text{ rad/s}^2) \\
+ \dagger \Sigma F_y = \Sigma (F_y)_{eff} : A - mg = -m\bar{a} \\
A - mg = -m(0.183La) = -m(0.183Lk(0.9143\frac{9}{L})) \\
A - mg = -0.1673mg; A = 0.8326 mg \\
A = 0.8324(2.5 \text{ Re})(9.81m/s^2); A = 20.4 N & 4.5 \text{ Re}
\end{array}$



WE NOTE THAT ROD RETATES ABOUT AND THAT

$$\tilde{I} = \frac{1}{12} m L$$

$$\tilde{A} = \frac{1}{12} m L$$

$$\tilde{A$$

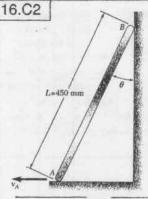
+)
$$I(M_A) = I(M_A)_{PR}$$
; $mg(\frac{1}{2}\cos\beta) = I\alpha + m\bar{\alpha}(\frac{1}{2})$
 $\frac{1}{2}mgL\cos\beta = \frac{1}{2}mL^2\alpha + m(\frac{1}{2}\alpha)\frac{1}{2}$
 $= \frac{1}{3}mL^2\alpha$
 $\alpha = \frac{3}{2}(\frac{9}{L}\cos\beta)$ (1)

+
$$\uparrow IF_g = \Sigma(F_g)eH$$
: $N-mg = m\bar{\alpha} \cos\beta$
 $= m \frac{1}{2} \cot \cos\beta = m \frac{1}{2} \left(\frac{3}{2} \frac{3}{2} \cos\beta\right) \cos\beta$
 $N = mg \left(1 - \frac{3}{4} \cos^2\beta\right)$ (3)

OUTLINE OF PROGRAM!

- (a) FOR β = D TO 85° AT S'INCREMENTS, DETERMINE
 F (from EQ(2)) AND N (from EQ.(3), ALSO
 DETERMINE REQUIRED VALUE OF MY = F/N
- (b) USE SMALLER INCREMENTS TO FIND TWO VALUES OF B CORRESPONDING TO ME = 0.50.

beta	F	N	mu	7s1ip?	
0.000	0.000	1.250	0.000	no slip	
5.000	0.326	1.278	0.255	no slip	
10.000	0.641	1.363	0.470	no slip	
15.000	0.938	1.501	0.624	slip	
20.000	1.205	1.689	0.714	slip	
25.000	1.436	1,920	0.748	slip	
30.000	1.624	2,166	0.742	slip	
35.000	1.762	2.484	0.709	slip	
40,000	1.847	2.799	0.660	slip	
45,000	1.875	3.125	0.600	slip	
50,000	1.847	3,451	0.535	slip	
55.000		3.766		no slip	
80,000	1.624	4.063	0.400	no slip	
65.000	1.436	4.330	0.332	no slip	
70.000	1.205	4.561	0.264	no slip	
75.000	0.938	4.749	0.197	no slip	
80,000	0.641	4.887	0.131	no slip	
85.000				no slip	
Seek	start of	range			
10.810		1.382		no slip	
10.820	0.691	1.382	0.500	slip	
Seek	end of re	ange			-
52.620	1.809	3.618	0.500	slip	
52,630		3.618			
52.640		3.819			
221040	11000	2.00	0.000		



GIVEN: m=5kg NA=1.5m/s 4 DA=0,

FIND:
NORMAL RECTIONS AT
A AND B FOR Ø=0 7050°
USING 5° INCREMENTS.
VALUE OF Ø AT WHICH
END B LOSES CONTACT

WITH WYALL.

KINEMATICS: WELDCITY: WELD

 $(a_{\alpha/n}) = L\omega^{2}$ $\bar{a} = \frac{L\omega^{2}}{\cos \Theta} \downarrow$ $(a_{\beta/n}) = L\omega^{2}$ $\bar{a} = \frac{1}{2}(a_{A} + a_{B}) = \frac{1}{2}a_{B}; \ \bar{a} = \frac{L\omega^{2}}{2\cos \Theta}$ (2)

KINETICS: 44 A = 0 $A = m(g - \bar{a})$ $A = m(g - \bar{a})$

+) $IM_A = I(M_A)\rho f$: $B(L\cos 6) - mg(\frac{1}{2}\sin 6) = -\tilde{I}\alpha - m\tilde{\alpha}(\frac{1}{2}\sin 6)$ $B = \frac{m(g-\tilde{\alpha})\frac{1}{2}\sin 6 + \frac{1}{2}mL^{\frac{1}{2}}\alpha}{L\cos 6}$ (4)

OUTUNE DE PROBLEMM: DATA M=5 Ag, L=0.45 M.

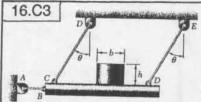
FOR EACH VALUE OF & EVALUATE W AND Q.

THEN USE W AND Q TO EVALUATE A AND B.

USING SMALLET INCREMENTS FIND VALUE

OF & FOR WHICH B=0.

theta	omega	alpha	Æ	A	8
deg.	rad/s	rad/s2	m/s²	Ñ	N
0.000	3.333	0.000	2.500	36,550	0.000
5.000	3.346	0.980	2.529	36,406	1.408
10.000	3.385	2.020	2.817	35.963	2.786
15.000	3.451	3.191	2.774	35,180	4.094
20.000	3.547	4.580	3.013	33.986	5.271
25,000	3.678	6.308	3.358	32.259	6.218
30.000	3.849	8.553	3.849	29,805	6.752
35.000	4.069	11.595	4.548	26,309	6.557
40.000	4.351	15.888	5.561	21,243	5.024
45.000	4.714	22.222	7.071	13.695	0.955
50,000	5.186	32.049	9.413	1.984	-8.166
	Find theta	for B = 0			
45.747	4.777	23.420	7.357	12.265	0.002
45.748	4.777	23.422	7.357	12.264	0.001
45.749	4.777	23.423	7.358	12.262	-0.001



GIVEN: 6=8m, h=6m.

30-16 CYLINDER

10-16 PLATFORM

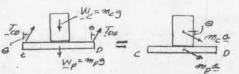
AFTER AB IS CUT,

FINO: MANY FOR INHIEN

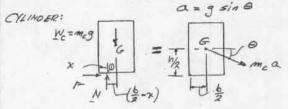
CYLINDED DOES NOT SLIP

FOR 6=0 TO 30° USING

5° INCREMENTS. THEN FOR 45 = 0.60, FIND & FOR WHICH SLIPPING IMPENDS. IN ALL CASES, CHECK WHETHER CYLINDER TIPS.



+ > IF = IFeq: (me+mp) gsing = (me+mp) a



RESULTANT OF FORCES EXERTED BY PLATFORM ON TO CYLINDER ACTS AT DISTANCE & FROM CORNER,

$$\begin{array}{ll}
+\sum F_{x} = \sum F_{x} | eq : F = me a \cos \theta \\
+\sum F_{y} = \sum (F_{y}) eq : N = me g = -me a \sin \theta \\
N = me (g - a \sin \theta) \\
(4_{x}) = \frac{F}{N} = \frac{Me}{m_{x}(g - a \sin \theta)} = \frac{(g \sin \theta) \cos \theta}{g - (g \sin \theta) \sin \theta} \\
4_{x} = \frac{\sin \theta}{1 - \sin^{2} \theta}
\end{array}$$

+) $ZM_0 = \Sigma(M_0)_{z=1}$ in $g(\frac{b}{z}-x) = m_0 \cos(\frac{b}{z}) + m_0 \cos(\frac{b}{z}-x)$ $\frac{1}{2}(\frac{b}{z}-x) = (\frac{1}{2}\sin\theta)\cos(\frac{b}{z}+(\frac{1}{2}\sin\theta)\sin(\frac{b}{z}-x)$

CYLINDER TIPS IF x <0; tand > b = 8 in ; 0 > 53.10

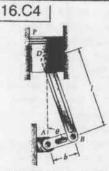
EVALUATE of AND & FOR EACH VALUE OF G.

PILINT MY AS MINIMUM VALUE OF MY FOR NO SLIDING.

theta	x	mu req.	?slip?	?tip?
0.000	4.000	0.000	no slip	no tip
5.000	3.738	0.087	no slip	no tip
10.000	3.471	0.176	no slip	no tip
15.000	3.196	0.268	no slip	no tip
20.000	2.908	0.364	no slip	no tip
25.000	2.601	0.466	no slip	no tip
30.000	2.268	0.577	no slip	no tip
35.000	1.899	0.700	slips	no tip

-- Find theta for mu = 0.60 -

30.960 2.200 0.5999 30.980 2.199 0.6004



GIVEN ENGINE EYSTEM OF PILOB IS.C3.

WAR = 1000 Mm), MAR = 0 l = 160 mm, b = 60 mm mp = 2529, MRD = 329

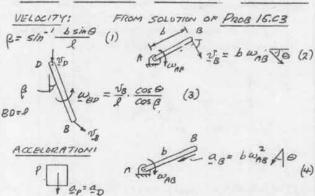
FIND: COMPONENTS OF

DYNAMIC REACTIONS ON BD

AT PONTS B AND D FOR

B= 0 TO 180 USING 100

INCREMENTS.



$$D = R \qquad = \frac{\lim_{n \to \infty} \sin_n - a_n \cos \theta}{R \cos \beta} \qquad (5)$$

$$\lim_{n \to \infty} \frac{1}{2} \int_{\mathbb{R}^n} dx dx = \frac{\lim_{n \to \infty} \sin_n - a_n \cos \theta}{R \cos \beta} + \lim_{n \to \infty} \cos \beta + \lim_{n \to \infty} \sin \beta \qquad (6)$$

KINETICS I WE FIRST FIND \$\overline{a}_{\text{X}} AND \$\overline{a}_{\text{Y}} \= a_{\text{B}} \column{c} \overline{a} \text{AND} & \overline{a} \te

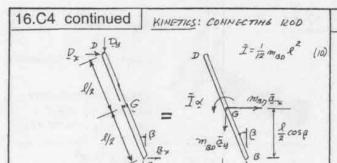
SINCE G IS AT THE MIDDLE OF BD $\pm \bar{a}_{\chi} = \frac{1}{2} \langle \bar{a}_{g} \rangle_{\chi} \tag{7}$

$$+ \Sigma F_g = \Sigma(F_g)_{eff}$$

$$O_g = m_p a_D$$
(9)

NOTE: SINCE WE SEEK THE DYNAMIC REACTIONS, WE OMIT THE WEVENT OF THE PISTON AND CONNECTINE ROD

(CONTINUED)



+ $\sum M_a = \sum (M_B)_{e,\theta}$: $D_x \cdot l \cos \beta \cdot D_y \cdot l \sin \beta =$ $- \hat{L} \propto + m_{\theta,\theta} a_x (\frac{1}{2} \cos \beta) - m_{\theta,\theta} (\frac{1}{2} \sin \beta)$

Divide BY & Save Fore
$$D_x$$

$$D_x = D_y \frac{\sin\beta}{\cos\beta} - \frac{\tilde{I} \alpha}{2\cos\beta} + \frac{m_{BP}}{2} \tilde{\alpha}_x - \frac{m_{BP}}{2} \tilde{\alpha}_y \frac{\sin\beta}{\cos\beta}$$

$$D_z = Q_y \tan\beta - \frac{\tilde{I} \alpha}{2\cos\beta} + \frac{m_{BP}}{2} \tilde{\alpha}_x - \frac{m_{BP}}{2} \tilde{\alpha}_y \tan\beta$$
[1]

$$\pm \sum F_{\chi} = \sum (F_{\chi})_{eff}:$$

$$B_{\chi} + D_{\chi} = m_{RD} \tilde{\Delta}_{\chi}$$

$$B_{\chi} = m_{RD} \tilde{\Delta}_{\chi} - D_{\chi} \qquad (12)$$

$$+\psi \cdot \Sigma F_{y} = Z(F_{y,1}e^{qz};$$

$$B_{y} + D_{y} = m_{xx}D\bar{a}_{y}$$

$$B_{y} = m_{xx}D\bar{a}_{y} - D_{y} \qquad (13)$$

BUTLINE OF PROGRAM:

ENTER DATA:
$$W_{AB} = 1000 \text{ rpm} \left(\frac{2\pi}{66}\right) = \frac{100}{3} \pi^{\text{rad}/5}$$
 $m_{\phi} = 2.5 \text{ kg}, \quad m_{B}o = 3 \text{ kg}$
 $L = 0.1 m_{\phi}, \quad b = 0.06 \text{ m}$

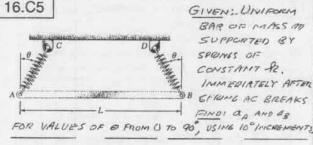
PROLUBING, IN SEQUENCE, EQS. (1) THROUGH (12).

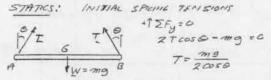
EVALUATE AND PRINT By, By, By, AND Dy FOR
VALUES OF & FROM O TO 180° AT 5°

INCREMENTS:

Positive directions of force components are: DOWN and TO THE RIGHT

theta	Bx	Ву	Dx	Dy
deg	N	N	N	N
0	0.00	4605.82	0.00	-2261.78
10	108.19	4497.37	-279.57	-2203.38
20	182.74	4177.66	-520.30	-2031.39
30	194.59	3663.87	-688.07	-1755.71
40	124.19	2985.61	-758.59	-1393.47
50	-33.57	2185.52	-722.48	-969.45
60	-265.48	1318.43	-589.25	-515.59
70	-539.76	447.34	-387.68	-68.61
80	-811.62	-364.52	-160.35	334.94
90	-1034.81	-1064.65	47.85	665.41
100	-1174.86	-1621.34	202.89	906.22
110	-1217.65	-2028.11	290.21	1056.59
120	-1169.20	-2300.43	314.47	1129.34
130	-1048.31	-2466.79	292.25	1145.24
140	-B77.31	-2558.80	242.90	1126.72
150	-675.57	-2604.17	182.09	1093.40
160	-456.95	-2623.56	119.39	1060.08
170	-230.09	-2630.38	58.71	1036.51
180	-0.00	-2631.89	0.00	1028.08





KINETICS! JUST AFTER AC BREAKS

$$A = \frac{1}{W \cdot mg} = \frac{1}{4} = \frac{ma_v}{6} = \frac{1}{ma_y} = \frac{1}{6} = \frac{1}{$$

Wing * $\frac{1}{2}$ $\pm \sum F_{x} = \sum F_{y|eq} \cdot T_{ein} \Theta = m \tilde{\alpha}_{x}$ $\frac{mg}{2\cos \theta} \sin \theta = m \tilde{\alpha}_{x}$ $\tilde{\alpha}_{x} = \frac{1}{2}g \tan \theta = (1)$

+
$$\sqrt{\Sigma E_g} = \Sigma (E_g)_{eg}$$
: $mg - T \cos \varepsilon = m a_g$

$$mg - \frac{mg}{T \cos \varepsilon} \cos \varepsilon = m a_g$$

$$\bar{a}_g = \frac{1}{2}g \downarrow \qquad (2)$$

+)
$$\sum M_g = \sum (M_g)_{pq}$$
; $(T\cos 6) \frac{1}{2} = \frac{1}{12} ML^2 d$
 $\frac{m9}{2\cos 6} \cos 6 \frac{1}{2} = \frac{1}{12} ML^2 d$ (3)

$$(a_{R})_{y} \stackrel{\triangle}{\wedge} (a_{R})_{y} \qquad (a_{R})_{y} \qquad (a_{R})_{y} = \bar{a}_{x} = \frac{1}{2}g \tan \theta \ (4)$$

$$(a_{R})_{y} \stackrel{\triangle}{\wedge} (a_{R})_{y} \qquad (a_{R})_{x} = \bar{a}_{y} = \frac{1}{2}g \tan \theta \ (5)$$

$$+\sqrt{(a_A)_g} = \bar{\alpha}_g + \frac{1}{2}\alpha = \frac{1}{2}g + \frac{1}{2}(\frac{3g}{2}) = 2g\sqrt{(b)}$$

$$+\uparrow (a_0)_y = -\bar{a}_y + \frac{1}{2}\alpha = -\frac{1}{2} + \frac{1}{2}(\frac{39}{2}) = g\uparrow$$
 (7)

END A!
$$\beta = \tan^{-1} \frac{(a_A)_y}{(a_A)_x}$$
; $a_A = \frac{(a_A)_x}{\cos \beta}$ (8,9)

END B:
$$\chi = ton^{-1} \frac{(a_B)y}{(a_B)x}$$
; $a_B = \frac{(a_B)x}{\cos x}$ (N,11)

OUTLINE OF PROGRAM:

PROBRAM, IN SEQUENCE, EGS. (1) THROUGH (11).
EVALUATE AND PRINT and, B, and FOR
VALUES OF & FROM O TO 900 USINE
10° INCREMENTS.

theta	[aA	beta]	[aB	gamma]	
0.000	2.000	90.000	1.000	90.000	
10.000	2.002	87.476	1.004	84.962	
20.000	2.008	84.801	1.016	79.686	
30.000	2.021	81.787	1.041	73.898	
40.000	2.044	78.153	1.084	67.240	
50.000	2.087	73.409	1.164	59.210	
60.000	2.179	66.587	1.323	49.107	
70.000	2.426	55.516	1.699	36.052	
80.000	3.470	35.196	3.007	19.425	
90.000	infinite	0.000	infinite	0.000	

17.1 GIVEN: 6000-16 FLYWHEEL, R= 36in.,

FIND! MAGNITUDE OF COUPLE DUE TO FRICTION KNOWING FLYWHELL ROTATES ISOU REVOLUTIONS WHILE CONSTING TO REST.

 $\omega_{s} = 300 \text{ rpm} \left(\frac{2\pi}{6}\right) = 10 \text{ Tr} \text{ rad}$ $\vec{1} = m \cdot \vec{R}^{2} = \frac{6000 \text{ ib}}{32.2 \text{ ft/s}^{2}} \left(3 \text{ ft}\right)^{2} = 1677 \text{ ib} \cdot \text{ ft} \cdot \text{ s}^{2}$ $T_{r} = \frac{1}{2} \vec{1} \omega_{s}^{2} = \frac{1}{2} \left(\frac{1677}{100}\right)^{2} = 927,600 \text{ ft} \cdot \text{ ib}, T_{2} = 0$ $U_{1-2} = -M \Theta = -M \left(\frac{1500 \text{ rev}}{2\pi}\right)^{2} \left(2\pi \frac{\text{ red}}{\text{ rev}}\right) = -9424.7 \text{ M}$

T,+U,-=T2: 827,600-9424.7 M = 0

M=87,81 16.62 M=87,816-ft

17.2 GIVEN: 50-Rg ROTOR, R= 180mm

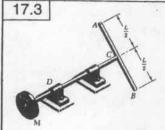
W= 3600 rpm, M= 3.5 N.m

FIND: NUMBER OF REVOLUTIONS AS

ROTOR COASTS TO REET

 $CU_0 = 3600 \text{ spm} \left(\frac{2\Pi}{60}\right) = 120 \pi \cdot \text{nod/s}$ $\tilde{I} = m \tilde{R}^2 = (50 \text{ Rg})(0.180 \text{ m})^2 = 1.620 \text{ kg·m}^2$ $T_1 = \frac{1}{2} \tilde{I} w_0^2 = \frac{1}{2} (1.620)(120 \pi)^2 = 115.12 \text{ kJ}, T_2 = 0$ $U_{102} = -M\Theta = -(3.5 \text{ N·m})\Theta$

 $\frac{T_{1}+U_{1-2}=T_{2}}{\Theta=32.891\times10^{3}}\cdot (3.5 \text{ M.m})\Theta=0$



GIVEN: 8-16 DISK OF 9-in, DIAMETER ROD AB WERNS 316/A M=416-ft

FIND: LENGTH L IF
W IS 300 YPM A FTER
2 REVOLUTIONS

 $Y = 4.6 \text{ in}, = \frac{3}{8} \text{ ft}$: $\omega = 300 \text{ rpm} \left(\frac{2\pi}{60}\right) = 10\pi \text{ rad/s}$

$$\begin{split} & \mathcal{W}_{RGR} = 8/b, \quad \mathcal{W}_{ROO} = (3/b/f2)L \\ & \bar{I} = \frac{1}{2} m_{BSR} r^2 + \frac{1}{12} m_{ROO} L^2 \\ & = \frac{1}{2} \frac{8}{3} \left(\frac{3}{8}\right)^2 + \frac{1}{12} \frac{3L}{9} L^2 = \frac{1}{3} \left(\frac{9}{16} + \frac{L^3}{4}\right) \\ & \mathcal{T}_{I} = 0, \quad \mathcal{T}_{Z} = \frac{1}{2} \bar{I} W^2 = \frac{1}{29} \left(\frac{9}{16} + \frac{L^2}{4}\right) (10\pi)^2 \\ & \mathcal{U}_{I-2} = M\Theta = (9/b \cdot f2) (2rev) \left(\frac{2\pi rad}{7er}\right) = 16\pi \\ & \overline{\mathcal{T}_{I}} + \mathcal{U}_{I-2} = \overline{\mathcal{T}_{Z}}; \quad O + 16\pi = \frac{1}{29} \left(\frac{9}{16} + \frac{L^3}{4}\right) (10\pi)^2 \\ & \frac{16\pi(29)}{(10\pi)^2} = \frac{9}{16} + \frac{L^3}{4} \\ & 3.2799 = \frac{9}{16} + \frac{L^3}{4}; \quad \frac{L^3}{4} = 2.717 \end{split}$$

13= 10.869 St3.

L = 2.22 ft

17.4 b

GIVEN: WO = 0

I DISK = IO

WE WEIGHT / UNIT LENGTH

OF ROD

EINO: LENGTH L FOR
MAXIMUM OF AFFER
COUPLE M IS APPLIED FOR
ONE REVOLUTION

 $T_{z} = \frac{1}{2} \left(I_0 + \frac{1}{12} \frac{\omega L}{9} L^2 \right) \omega_z^2$

 $U_{1\rightarrow2} = M \Theta = M(2\pi rad)$ $T_1 + U_{1-2} = T_2 = O + 2\pi M = \frac{1}{2} \left(I_0 + \frac{\omega L^3}{12g} \right) \omega_z^2$

 $\omega_2^2 = \frac{477 M}{I_0 + \frac{w \, 2}{129}}$

 $V_A = \frac{L}{2} \; \omega_2 \; : \quad V_A^2 = \frac{L^2}{4} \; \omega_2^2 = \frac{\gamma r \, M L^2}{I_o + \frac{\omega r \; L^3}{129}}$

DIFFERENTIATING WITH RESPECT TO L,

 $2\sqrt{A}\left(\frac{d\sqrt{A}}{dL}\right) = \left[2L\left(I_0 + \frac{\omega L^3}{129}\right) - L^2\left(\frac{3\omega L^2}{129}\right)\right] \frac{\pi M}{\left(I_0 + \frac{\omega L^3}{129}\right)^2}$

 $\frac{c^{1}\sqrt{1}A}{c^{2}L} = 0: \quad 2L\left(I_{0} + \frac{\omega L^{2}}{129}\right) - L^{2}\left(\frac{3\omega L^{2}}{129}\right) = 0$ $2I_{0}L - \frac{\omega L^{4}}{129}: \qquad L^{3} = \frac{249}{129}$

17.5 GIVEN: 300-Re PUNCHING MACHINE FLYVIAELL, REQUIRES 2500 J.

FIND: (a) Wy IMMEDIATELY AFTER A FUNICH

(b) IF M= 25 H·m, FIND REVOLUTIONS

BEFORE W IS ABAM 200 Y IN.

I=mk2= (300 ks)(0.6m)2=108 kg·m2

(0.) $\omega_{i} = 300 \text{ rpm} \left(\frac{2\pi}{60} \right) = 10\pi \text{ rod/s}$ $T_{i} = \frac{1}{2} m \omega_{i}^{2} = \frac{1}{2} \left(108 \cdot \frac{1}{10} \cdot \frac{$

T2 = 1 mw = = 1 (108 29 m) w

T,+U,+2=T2: 53.296.AJ -2,5AJ = = (108 Ag *on) W=

W2=30-67 rod/s (60) = 292,9 ym W2=293 ypm

(b) U2+1 = M6 25001 = (25 N·m) 0 0 = 100 rad (\frac{rev}{2n rad}) = 15.9155 rev 0 = 15.92 rav 17.6 GIVEN: (U, = 360 YPM OF PUNICHING MACHINE FLYWHEEL. EACH PUNICH REQUIRES 1500 ft.16.

AFTER EACH PUNICH (U) = 0.95 W).

FINO: (a) I OF FLYWHEEL

(b) REVOLUTIONS REQUIRED FOR

ANGULAL VELOCITY TO AGAIN BIE 360 YPM IF

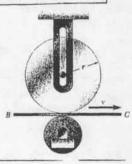
CONSTANT 1816-Ft COUPLE IS APPLEO

(a) $(\omega_{i} = 360 \text{ rpm} \left(\frac{2\pi}{86}\right) = 12\pi \text{ rod} k$ $(\omega_{2} = 0.95 \omega_{i} = 0.94 (12\pi \text{ rod} k)) = 11.4\pi \text{ rad} k$ $(\omega_{1} = \frac{1}{2}\bar{1}\omega_{1}^{2} = \frac{1}{2}\bar{1}(12\pi)^{2}$ $(\omega_{1} = \frac{1}{2}\bar{1}\omega_{2}^{2} = \frac{1}{2}\bar{1}(11.4\pi)^{2}$ $(\omega_{1} = \frac{1}{2}\bar{1}\omega_{2}^{2} = \frac{1}{2}\bar{1}(11.4\pi)^{2}$

 $\frac{T_1 + U_{1 \to 2} = T_2 : \frac{1}{2} \bar{I} (12\pi)^2 - 1500 = \frac{1}{2} \bar{I} (11.4\pi)^2}{\bar{I} = \frac{2(1500)}{\pi^2 (12^2 - 11.4^2)} = \frac{3000}{138.57} = 21.649 \cdot 16 \cdot 12 \cdot 15^2}{\bar{I} = 21.66 \cdot 12 \cdot 15^2}$

(b) $U_{2\rightarrow 1} = M\Theta$: |500%, $ft = (18 ft \cdot 16)G$ $G = 83,32 \text{ rod} \left(\frac{\text{rev}}{2\pi \text{ rod}}\right) = |3,263 \text{ rev}$ $\Theta = |3,26 \text{ rev}$

17.7 and 17.8



GIVEN: DISW PLACED ON EFLT

WHEN WOO, COEFFICIENT OF

KINGTIC FRICTION = 4/5.

FIND: REVOLUTIONS BEFORE

W = CONSTANT,

PROBLEM 17.7:

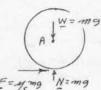
IN TERMS OF S, Y, AND MS

PROBLEM 17.8:

FOR Y = 6/1, S = 40 ft/s,

AND MS = 0.20.

CAN'T FORCE DOWN WORK IS F. SINCE ITS MOMENT ABOUT A IS METF, WE HAVE



U,-2=MB= rFB= r(45mg)0

ANGULAR VELDELTY BECOMES CONSTANT WHEN $W_2 = \frac{\nabla}{r}$

 $T_{1} = 0$ $T_{2} = \frac{1}{2} \tilde{I} w_{2}^{2} = \frac{1}{2} \left(\frac{1}{2} m r^{2} \right) \left(\frac{V}{r} \right)^{2} = \frac{m V^{2}}{4}$

Ti+U1-2= Tz: 0+r-4, mg 0 = mer 2

 $\Theta = \frac{\pi^2}{4r4kg} \text{ rad} \qquad \Theta = \frac{5}{8\pi r4kg} \text{ rev}$

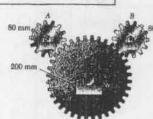
PROBLEM 17.8: r= 0.5ft, 45 = 0.20, 1 = 40 ft/s

G = (40 ft/s)2 BR(0.5ft)(0.20)(32.2 [1/52]

0=19.77 rev

NOTE: RESULT IS IMPERIMENT OF W.

17.9 and 17.10



GIVEN: M_= 12Ag, Az= 150mm

M==mg= 2.4Ag

m==mg= 2.4Ag

m= 10 N·m

(=INO: (a) REVOLUTIONS

OF C AS W_C INCREASES

I=ROM 100 PPM TO 450 PPM

(b) TANGENTIAL FORCE ON A

PROBLEM 17.19

M IS APPLIED TO GEAR C

PROBLEM 17.10

MIS APPLIED TO GEAR B

80 mm (A 200 mm)

 $\omega_{A} = \omega_{B} = \frac{200 \text{ mm}}{80 \text{ mm}} \omega_{c} = 2.5 \omega_{c}$ $(\omega_{c})_{c} = 100 \text{ rpm} \left(\frac{2\Pi}{60}\right) = 10.472 \text{ rad/s}$ $(\omega_{c})_{c} = 450 \text{ rpm} \left(\frac{2\Pi}{60}\right) = 47.124 \text{ rad/s}$

WORK AND ENERGY $\bar{I}_{A} = \bar{I}_{B} = m \hat{R}^{2} = (2.4 Rg)(0.06 m)^{2} = 8.64 \times 10^{-3} Rg \cdot m^{2}$ $\bar{I}_{C} = m \hat{R}^{2} = (12 Rg)(0.150 m)^{2} = 0.270 Rg \cdot m^{2}$

POSITION /: $(\omega_c)_{r} = 10.472 \text{ rad/s};$ $(\omega_n)_{r} = (\omega_8)_{r} = 2.5(\omega_c)_{r} = 26.18 \text{ rad/s}$ $T_{r} = 2\left[\frac{1}{2}\hat{I}_{A}(\omega_n)_{r}^{2}\right] + \frac{1}{2}\hat{I}_{c}(\omega_c)_{r}^{2}$ $= 2\left[\frac{1}{2}(8.64 \times 10^{3})(26.18)^{2}\right] + \frac{1}{2}(0.270)(10.472)^{2} = 20.726 \text{ J}$

 $\frac{Position 2}{(\omega_{e})_{2} = 47.124 \text{ rad/s}} = (\omega_{e})_{2} = 47.81 \text{ rad/s}$ $(\omega_{e})_{2} = (\omega_{f})_{2} = 2.5(\omega_{e})_{2} = 117.81 \text{ rad/s}$ $T_{2} = 2\left[\frac{1}{2}(8.64 \text{ xio}^{2})(117.81)^{2}\right] + \frac{1}{2}(0.210)(47.124)^{2} = 419.71 \text{ J}$

PROBLEM 17.9: M=10N. M APPLIED TO GEAR C

Ti+U1=2=T2: 20.765+100=419.715

G= 39.90 rad G= 6.35 rev

G=A17 A: $\Theta_{A} = 2.5 \Theta_{C} = 2.5(39.90) = 99.75 rad

A T, + U_{-2} = 72: <math>\frac{1}{2}m_{A}(w_{A})^{3} + F(0.08)\Theta_{A} = \frac{1}{2}m_{A}(w_{A})^{3}$ 0.08 at $\frac{1}{2}(B.69.40^{3})(26.18)^{2} + F(0.09)(97.80) = \frac{1}{2}(B.64 \times 10^{3})(1/7.81)^{2}$

2.961+7.98 F = 59.96 F = 7.14 N F = 7.14 N A

PROBLEM 17.10: M=10 N·m APPLIED TO BEAR B

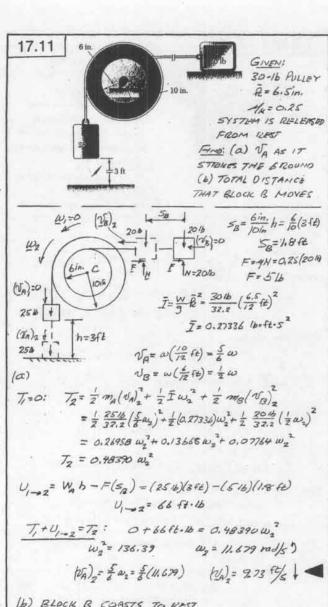
NOTE I AMBULAR SPEEDS AIRLE SAME
AS IN PROBLEM, THUS T, AND T2 AIRE ALSO THE SAME
T,= 20.726 J $T_2=419.71$ J

WE HAVE $U_{1-2}=M$ $\Theta_3=10$ Θ_8 $T_1+U_{1-2}=T_2$: 20.726 J + 10 $\Theta_3=419.71$ J

68= 39.90 rad 68= 2.56; 39.90 rad= 2.56; 6= 15.96 rad 6= 2.54 rev

F=12.86M

F=17.86H





TOTAL ENERGY OF BLOCKS AND PULLEY JUST BEFORE IMPACT = 66 ft. 16 KINETIC ENERGY OF BLOCK A JUST BEFORE IMPACT

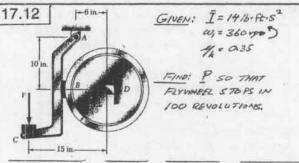
 $T_A = \frac{1}{2} \frac{W_A}{9} (\tilde{v_h})_2^2 = \frac{1}{2} \frac{25.16}{32.2} (9.73 \text{ ft/e})_2^2 36.75 \text{ ft-6}$

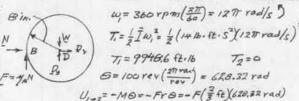
AFTER BLOCK A STRIKES THE GROOMD, WE FIND THAT THE KINETIC ENGREY OF THE PULLBY C AND BLOCK B 15 T = 66 ft. 16 - 3675 ft. 16 = 29.25 ft-16

FOR SYSTEM TO STOP, 29.25 ft. 16 OF ENERGY MUST BE DISSIPATED BY THE FRICTION FORCE, F= 516.

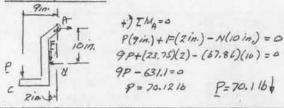
29.25ft.16 = (516)d d=5.85ft

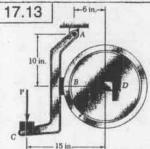
TO FIND TOTAL DISTANCE MICHED BY B. WE ADD SB=1,89t. 70TAL DISTANCE=1.8+5.85 = 7.65 ft





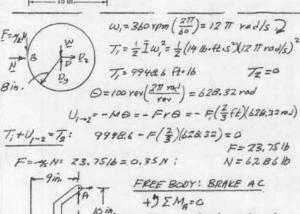
9948.6 - F(=) 628.32)=0 T, + U, = 72: F=23.75 16 F= 4/5 N: 23.25/6 = (0.35)N; N= 67.86 16 FREE BODY: BRAKE AC;





GIVEN: I = 14 16. Ft.5 W, = 360 mm 1x=0.35

FIND: P SO THAT FLY WHEEL STOPS IN 100 REVOLUTIONS



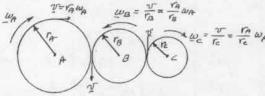
17.14 and 17.15



GIVEN: FRICTION
DISKS A, B, AND C
ARE MADE OF SAME
MATERIAL AND
HAVE SAME
THICKNESS

PROBLEM 17.14: FIND: EXPRESSION FOR WA AFTER
THE COUPLE M IS APPLIED FOR ONE REVOLUTION
PROBLEM 17.15: FIND: REVOLUTIONS OF A REGUINED
FOR WA = 150 rpm WHEN M = 6016.10,
YA = 8 in, YB = 6 in, YB = 4 in, AND WA = 12 10.

DENOTE VELOCITY OF PERIMETER EY OF.



DENOTE WASS DENSITY OF MATERIAL BY ρ AND THICKNESS OF DISKS BY t.

THEN MASS OF A DISK IS $m = (\text{VOLUME}) \rho = (\pi r^2 t) \rho$ AND $\bar{I} = \frac{1}{2} m r^2 = \frac{\pi r t}{r} r^4$

KINETIC ENERGY: T= Z = Iw2

$$T = \frac{1}{2} \left(\frac{\pi \rho t}{2} \right) \left[r_A^{\mu} \omega_A^2 + r_B^{\mu} \omega_B^2 + r_C^{\mu} \omega_C^2 \right]$$

$$= \frac{1}{2} \left(\frac{\pi \rho t}{2} \right) \left[r_A^{\mu} \omega_A^2 + r_B^{\mu} \left(\frac{r_A}{r_B} \right)^2 \omega_A^2 + r_C^{\mu} \left(\frac{r_A}{r_C} \right)^2 \omega_A^2 \right]$$

$$T = \frac{1}{2} \left(\frac{\pi \rho t}{2} \frac{\omega_A^2}{r_A} \right) r_A^2 \left[r_A^2 + r_B^2 + r_C^2 \right]$$

$$WORK; \quad U = M \Theta \qquad \omega_i = 0; \quad \omega_2 = \omega_A.$$

$$T_i + U_{i-2} = T_2: \quad O + M \Theta = \frac{\pi \rho t}{4} \omega_A^2 r_A^2 \left[r_A^2 + r_B^2 + r_C^2 \right]$$

$$\frac{Rassism 17.17}{r_A} \quad for \theta = MT: \quad M(2\pi) = \frac{\pi \rho t}{4} \omega_A^2 r_A^4 \left[1 + \left(\frac{r_B}{r_A} \right)^2 \left(\frac{r_C}{r_A} \right)^2 \right]$$

$$\omega_A^2 = \frac{8 M_0}{\ell t r_A^{\gamma} \left[1 + \left(\frac{r_B}{r_A}\right)^2 + \left(\frac{r_C}{r_A}\right)^2\right]}$$

PROBLEM 17.15: RECALL THAT MA=TIVATP AND WRITE

 $\mathcal{T} = \frac{1}{4} \left(\pi r_A^2 t_P \right) \left(r_A^2 + r_B^2 + r_c^2 \right)$ $\mathcal{T} = \frac{1}{4} \left(\frac{W_A}{\theta} \right) r_A^2 \left[1 + \left(\frac{r_B}{r_A} \right)^2 + \left(\frac{r_c}{r_A} \right)^2 \right] \omega_A^2$

DATA: WA= 150 rpm (3T) = 511 rad/s

WA= 1216, YA=8 in, YB=6 in., YC=4 in., M=60 16. in. = 5 ft. 16

U1-2= MO= (5 Ft-16) G

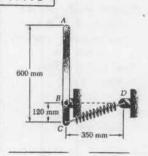
$$\frac{T_{1} + U_{1-2} = T_{2}}{-C + 50} = \frac{1}{4} \left(\frac{12.16}{32.2}\right) \left(\frac{9}{12}ft\right)^{2} \left[1 + \left(\frac{6in}{8im}\right)^{2} + \left(\frac{4in}{8im}\right)^{2}\right] \left(5\pi\right)^{2}$$

$$50 = 0.041408 \left[1 + \frac{4}{9} + \frac{1}{4}\right] (5\pi)^{2}$$

$$50 = 18.518 ; 0 = 3.704 \text{ rad} \left(\frac{rev}{2\pi rad}\right) = 0.5894 \text{ rev}$$

$$0 = 0.589 \text{ rev}$$

17.16



GIVEN: 4-RO RODAC

SPRING: R= 400N/m

WISTRETENED LENGTH

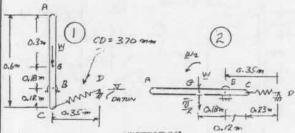
= 150 MM.

ROD IS KELEAKED FROM

12PST.

FIND: U) AFTER ROD

HAS ROTATED 90"



POSITION 1: UNSTRETCHED

SPANTS: 2,= CD - (150 mm) = 370 - 150 = 250 mm = 0.22 m Ve = 2 hx; = 2(400 Mp (0.22 m) = 7. 3 J 51144179 V = When man = (480)7.81ml 2(0.18 m) = 7.068 J

511A41T4 Va = Wh = man = (480)(9.81m/s)(0.16m) = 7.063) V = Ve + 49 = 9.68) +7.063) = 16.743

CONTRE ENERGY: 7, ED

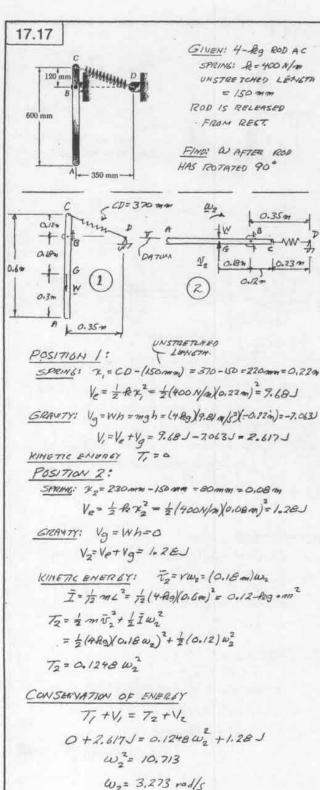
POSITION 2: SPRING: Vz=230mm - 150mm = 80mm + 0,08 m Ve= 1 k v2 = 1/400 N/m (0.08m) = 128 J

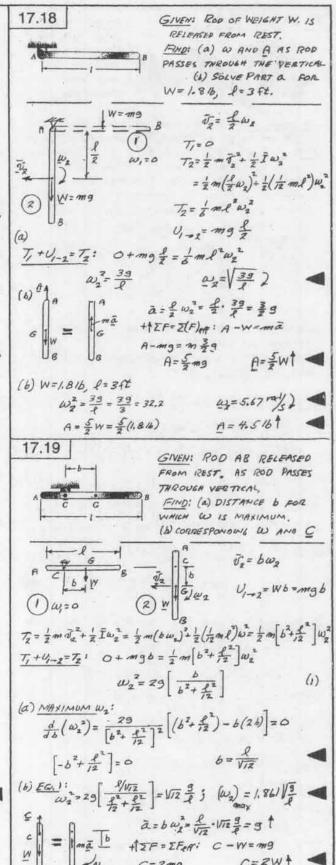
GRAVITYI $V_g = Wh = 0$ $V_z = V_c + V_g = 1.22 J$

KINETIC EMERGYI $\bar{V}_{Z} = r\omega_{\chi} = (0.18m)\omega_{\chi}$ $\bar{I} = \frac{1}{12}mL^{2} = \frac{1}{12}(4R_{0}\chi_{0.6m})^{2} = 0.12R_{0} \cdot m^{2}$ $T_{Z} = \frac{1}{2}m\bar{V}_{2}^{2} + \frac{1}{2}\bar{I}\omega_{2}^{2}$ $= \frac{1}{2}(4R_{0}\chi_{0.18}\omega_{2})^{2} + \frac{1}{2}(0.12)\omega_{2}^{2}$ $T_{Z} = 0.1248\omega_{2}^{2}$

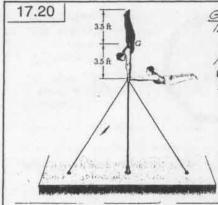
CONSERVATION OF ENERGY: $T_1 + V_1 = T_2 + V_2$ $O + 16.743J = 0.1248 w_2^2 + 1.28J$ $w_2^2 = 123.9$ $w_3 = 11.131 \text{ rad/s}$

W2=11.13 rad/5)





W= 3,27 rad/s)



GIVEN:

180:16 GYMMAST WITH

\$\bar{A} = 1.5 Pt\$

HE IS ROTATING

VERY ELOWLY (W, \(\pi\))

IN POSITION SHOWN,

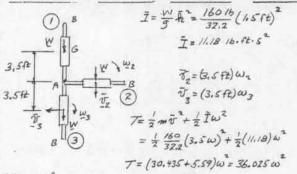
FIND! W AND

FORCE EXERTED

ON MIS MANOS

AFTER HE HAS

(\(\pi\)) 70°, (\(\pi\)) 180°



 $\frac{(a)6 = 90^{6}}{T_{1} = 0}; T_{2} = 36.025 \omega_{2}^{2}$ $U_{1-2} = W(3.5fi) = (16016)(3.5fi) = 560 \text{ ft-16}$ $T_{1} + U_{1-2} = T_{2}: O + 560 = 36.025 \omega_{2}^{2}$

 $\omega_2^2 = 15.545$ $\omega_2 = 3.94 \text{ ress}$

+) $\Sigma M_A = \Sigma (M_A)_{eff}$; (160)(3,5) = $\frac{W}{3}$ (3,50)(3,5) + \tilde{I} × 560 = $\frac{160}{32.2}$ 3,5° × + 11.18 ×

560 = 72.05 × × = 7.772 rad/s2

 $+\uparrow z_g = z(F_g)_{eR}$: $A_g = 16016 = -m(3.60)$ $A_g = 160 = -\frac{160}{32.2}(2.5)(7.772)$

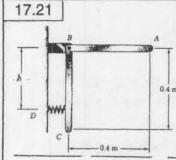
Ay-160=-135.17 Ay = 24.837 16 t

 $A_{x} = 270.316$ $\beta = \tan^{-1} \frac{24.837}{270.3} = 5.247^{6}$ $A = \frac{A_{y}}{\cos \beta} = \frac{270.3}{\cos 5.247^{6}} = 271.4816$ $A = 271.16 5.2^{6}$

(CONTINUED)

17.20 continued (6) $\Theta = 180^{\circ}$: $T_{i} = 01$ $T_{3} = 36.025 \omega_{3}^{2}$ $U_{i+3} = W(2 \times 3.5 \text{ ft}) = (16016)(7 \text{ ft}) = 1/20 \text{ ft} - 16$ $T_{i} + U_{i+3} = T_{3}$: $O + 1/20 = 36.025 \omega_{3}^{2}$ $\omega_{3}^{2} = 31.09$ $\omega_{2}^{2} = 5.58 \text{ rod/s}$

 $= \int_{\mathcal{B}} \frac{1}{160 \cdot 16} = \int_{\mathcal{B}} \frac{1}{$



GIVEN: TWO MODS EACH

OF MASS ON ARE WELDED

TOGETHER AND PRESSED

AGAINST SPRING AT D.

AFTER RELEASE RORS

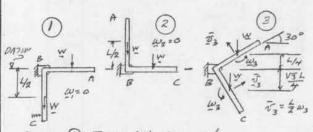
ROTATE THROUGH MIRY.

ANGLE DE 90°

FIND: ANGULAR VELOCITY

WHEN AB FORMS 30°

INTH HORIZONTAL.



POSITION (1) T,= 0, (Ve), (Vg) =- W = POSITION (2) T2=0, (Ve) = 0, (Ve) = + W =

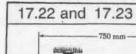
 $\sqrt{1/+V_1} = T_2 + V_2$: $O + (V_e)_1 - w \frac{L}{2} = O + w \frac{L}{2}$ $(V_e)_1 = wL$

POSITION (3): $|V_0|_3 = 0$; $|V_0|_3 = W(\frac{L}{4}) - W(\frac{V_3}{4}) = -0.183 \text{ WL}$ $T_3 = 2 \left\{ \frac{1}{2} m \tilde{V}_3^2 + \frac{1}{2} \tilde{I} \omega_3^2 \right\}$ $= 2 \left\{ \frac{1}{2} \frac{W}{4} (\frac{L}{2} \omega_2)^2 + \frac{1}{2} (\frac{L}{12} \frac{W}{3} L^2) \omega_3^2 = \frac{1}{3} \frac{W}{4} L^2 \omega_3^2 \right\}$

 $T_{1} + V_{1} = T_{3} + V_{3}:$ $O + (V_{e})_{1} + (V_{9})_{1} = T_{3} + (V_{9})_{2}$ $O + WL - \frac{1}{2}WL = \frac{1}{3}\frac{W}{9}L^{2}W_{3}^{2} - 0.183WL$ $WL(1 - \frac{1}{2} + 0.183) = \frac{1}{3}\frac{W}{9}L^{2}W_{3}^{2}$ $W_{2}^{2} = 3(0.683)\frac{9}{L} = 2.049\frac{9}{L} \qquad \omega_{2} = 1.431\sqrt{\frac{9}{L}}$

For L=0,4m; W3=1,43) \ 9.81 m/s= 7.086 rad/s

W3 = 7.09 rad/s)



GIVEN: MAB = 6-89

1.8-89 SEMICIRCULAR

DISK.

SPRING OF R=160 M/m

DISTRETCHED WHEN

AB IS HORIZONTAL

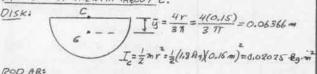
IF SYSTEM IS

RELEASED FROM REST.

PROBLEM 1722: WITH SPRING ATTACHED PROBLEM 1722: WITH SPRING ATTACHED PROBLEM 1723: SPRING PREMIONED

MOMENT OF INERTIA ABOUT C.

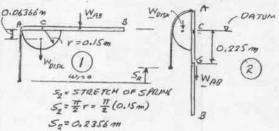
150 mm



 $I_c = \bar{I} + m_1 \bar{\chi}^2 = \frac{1}{12} (6.Re)(0.75 m)^2 + (6.Re)(0.225 m)^2$ = 0.28125 + 0.30375 = 0.585 Rg·m²

TOTAL I OF ASSEMBLY:

I = 0.02025+0.585 = 0.60525 Ag. m 2

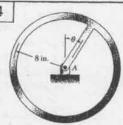


 $\frac{Postnon}{Postnon} J': \quad T_1 = 0, \quad V_1 = W_{DISIK} (-0.06366 m)$ $V_2 = (1.8 \cdot 29)(9.81)(-0.06336) = -1.1188 J$ $\frac{Postnon}{Postnon} Z: \quad (V_2)_2 = \frac{1}{2} - 2 \cdot 5 \cdot \frac{1}{2} = \frac{1}{2} (160 \mu/m) (0.23 \times m)^2 = 4.44 J$ $(V_3)_2 = W_{AB} (-0.225 m) = (6 \cdot 29)(9.81)(-0.225) = -13.24 J$

FOR MOR CENTROIDAL ROTATION INE USE EQ.(17.10) $T_2 = \frac{1}{2} I_c w_2^2 \cdot \frac{1}{2} (0.60525) w_2^2 = 0.3026 w_2^2$

PROBLEM 17.22: $T_1 + V_1 = T_2 + V_2$ $O - 1.1188 J = 0.3026 w_2^2 + 4.44 J - 13.24 J$ $7.681 = 0.3026 w_2^2$ $w_2^2 = 25.38 \qquad w_2 = 5.04 \text{ rad}$

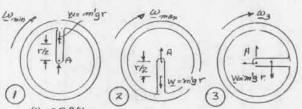
17.24



GIVEN: ASSEMBLY MADE OF OF =0.25 lb/ft ROD, KHOWING THAT WITH = 0.80 WMAX [FIND: (a) WMAX (b) W WHEN G=90°.

DENOTE MASS PER UNIT LENGTH BY m' AND RADIUS BY r $I_A = I_{ROD} + I_{RINE} = \frac{1}{3} (m' +) r^2 + (271 rm') r^2 = 6.6165 m' r^3$

FOR HON CENTROIDAL ROTATION: THE KINETIC ENERGY OF THE ASSEMBLY IS \$ 1 A W2



(a) $\omega_{min} = 0.8 \omega_{max}$ $\frac{T_1 + V_1 = T_2 + V_2}{f} : \frac{1}{2} I_A \omega_{min}^2 + mlg r \frac{r}{A} = \frac{1}{2} I_A \omega_{max}^2 - mlg r \frac{r}{2}$ $\frac{1}{2} I_A (\omega_{max}^2 - \omega_{min}^2) = mlg r^2$ $\frac{1}{2} 6.6/65 m^2 r^3 (1 - 0.8^2) \omega_{max}^2 = mlg r^2$

 $W_{max}^2 = 0.83\%5 \frac{3}{F} = 0.83965 \frac{32.2 \text{ PMc}^2}{(9/24)} = 40.555$

(b) $T_2 + V_2 = T_3 + V_3$: $\frac{1}{2} I_0 w_0^2 - m_0^2 r(\frac{r}{x}) = \frac{1}{2} I_0 w_0^2$ $\frac{1}{2} (6.6165 \text{ m}^3 r^3) (0.83365 \frac{9}{r}) - \frac{m_0^2 r^2}{r^2} = \frac{1}{2} (6.6165 \text{ m}^3 r^3) w_0^2$ $Z.7786 \text{ m}^3 g r^2 - 0.5 \text{ m}^3 g r^2 = 3.3162 \text{ m}^3 r^3 w_0^2$ $w_0^2 = \frac{7.27736}{3.3162} - \frac{9}{r} = 0.6885 \frac{9}{r}$ $w_0^2 = 0.6885 \frac{37.2 \text{ ft/s}^2}{(9/2 \text{ ft/s})} = 33.26$

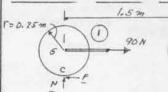
(%) (t) (%) (a) = 5.77 rad/s

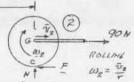
NOTEL RESULTS ARE INDEPENDENT OF WEIGHT PER UNIT LENGTH OF THE ROD USED TO MAKE THE ASSEMBLY



GIVEN: 20-Rg ROLLER
ROLLS WATHOUT SLIPPING
FINO: (a) TAFTER 1.500
motion.

(b) FRATION FORCE REGORDED TO FRENCHT SLIPPING.





INSTANT CENTER AT C: THUS F DOES NO INORM $T_{i} = 0$ $U_{i \to 2} = (90N)(1.5m) = 135J$ $T_{2} = \frac{1}{2}m\bar{x}_{2}^{2} + \frac{1}{2}\bar{1}w_{2}^{2} = \frac{1}{2}m\bar{x}_{3}^{2} + \frac{1}{2}(\frac{1}{2}mv^{2})(\frac{\bar{v}_{2}}{r})^{2}$ $T_{2} = \frac{3}{4}m\bar{x}_{2}^{2} = \frac{3}{4}(204a)(\bar{v}_{2}^{2}) = 15\bar{x}_{2}^{2}$

 $T_1 + U_{1 \to 2} = T_2$: $0 + 135J = 15 \tilde{v}_2^2$

 $\bar{v}_z^2 = 9$ $\bar{v}_z = 3m/s \rightarrow$

(b) CONSIDER MOTION AROUT MASS CENTER, $T_1 = 0$ $T_2 = \frac{1}{2} \tilde{I} w_2^2 = \frac{1}{2} \left(\frac{1}{2} m r^2 \right) \left(\frac{\tilde{V}^2}{r} \right)^2 = \frac{1}{4} m \tilde{V}_2^2$ $V_{1 \to 2} = F(1.5 m)$

 $T_1 + U_{1-2} = T_2$: $O + 1.5 F = \frac{1}{4} m \bar{U}_2^2$ $1.5 F = \frac{1}{2} (20 \log)(3 m/s)^2$:

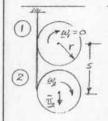
F= 30/6 -

17.26 and 17.27



GIVEN: OBJECT SHOWN
IS LECEASED FROM REST
FIND: V AFTER DOWNWARD
MOVEMENT S
PROBLEM 17.26
FOR A CYLINDER
PLOBLEM 17.27

FOR A THIN-INALLED PIPE



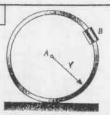
 $\hat{R} = RADIUS OF SYRATION$ $\hat{T} = r\omega \qquad \omega = \overline{T}$ $T_{1} = 0$ $T_{2} = \frac{1}{2}m\overline{V}_{2}^{2} + \frac{1}{2}\overline{1}\omega_{2}^{2}$ $= \frac{1}{2}m\overline{V}_{2}^{2} + \frac{1}{2}(m\overline{R}^{2})(\overline{Y})^{2}$ $T_{2} = \frac{1}{2}m(1 + \frac{\overline{R}^{2}}{r^{2}})\overline{V}_{2}^{2} \qquad V_{1} = mgS$

 $\frac{T_1 + U_{1-2} + T_2}{\bar{V}_2} = \frac{1}{2} m \left(1 + \frac{R}{V^2}\right) \bar{V}_2$ $\bar{V}_2 = \frac{295}{1 + \frac{R}{V^2}} \qquad (1)$

 $\frac{P_{ROSLEM} 17.25: CYLINDER}{\tilde{V}_{1}^{2} = \frac{795}{1 + \frac{1}{2}} = \frac{495}{3} \qquad \tilde{V}_{2} = \sqrt{\frac{495}{3}}$

 $\frac{PROBLEM 17.27! \ THIM-WALLED PIPE \ 4z^{2} \ v^{2}}{\hat{V}_{2}^{2} = \frac{295}{11!} = 95 \qquad \hat{z}_{2} = \sqrt{95} \ \downarrow$

17.28



GIVEN: HOOP OF MASS IN

ROLLS TO RIGHT. WITH

COLLAR BO = MASS IN AT

TOP W=W, AND AT

BOTTOM W = 3 W, L.

FIND: W, IN TERMS

OF G AND F.

$$\begin{split} & \mathcal{V}_{1 \to 2} = \mathcal{W}(2r) * mg/2r) = 2 mg r \\ & \mathcal{T}_{1} = \frac{1}{2} m N_{A_{1}}^{2} + \frac{1}{2} \bar{I} \omega_{1}^{2} + \frac{1}{2} m N_{B_{1}}^{2} \\ & = \frac{1}{2} m (r\omega_{1})^{2} + \frac{1}{2} (mr^{2}) \omega_{1}^{2} + \frac{1}{2} m (2r\omega_{1})^{2} = 3 m r^{2} \omega_{1}^{2} \\ & \mathcal{T}_{2} = \frac{1}{2} m N_{A_{2}}^{2} + \frac{1}{2} \bar{I} \omega_{2}^{2} + \frac{1}{2} m N_{B_{2}}^{2} \\ & = \frac{1}{2} m (r\omega_{1})^{2} + \frac{1}{2} m r^{2} \omega_{2}^{2} + 0 = m r^{2} \omega_{2}^{2} \end{split}$$

 $T_1 + U_{1-2} = T_2$: $3m r^2 w_1^2 + 2mg r = m r^2 w_2^2$ GIVEN! $w_2 = 3w$, $3m r^2 w_2^2 + 2mg r = m r^2 (3w)$ $2mg r = 6m r^2 w_1^2$; $w_1^2 = \frac{9}{2r}$; $w_2 = \sqrt{9}/3r$

17.29



GIVEN: HALF SECTION
OF PIRE OF MASS M,
RELEASED FROM RUST,
AFTER ROLLING THROUGH

FIND: (a) as (b) REACTION

 $\begin{array}{c|c}
\hline
 & \omega_1 = \overline{V_1} = 0 \\
\hline
 & \overline{V_2} = (AG)\omega_2 = r(1 - \frac{2}{H})\omega_2 \\
\hline
 & \overline{V_3} = 0 \\
\hline
 & \overline{V_2} = (AG)\omega_2 = r(1 - \frac{2}{H})\omega_2 \\
\hline
 & \overline{V_3} = 0 \\
\hline
 & \overline{V_3} = 0$

$$\begin{split} \widehat{J} &= m r^{2} - m(06)^{\frac{1}{2}} = m r^{2} - m\left(\frac{2r}{\pi}\right)^{2} = m r^{2}\left(1 - \frac{4}{\pi^{2}}\right) \\ T_{2} &= \frac{1}{2}mn_{2}^{2} + \frac{1}{2}\widehat{I}\,\omega_{2}^{2} = \frac{1}{2}m(1 - \frac{2}{\pi})^{2}r^{2}\omega_{2}^{2} + \frac{1}{2}mr^{2}\left(1 - \frac{4}{\pi^{2}}\right)\omega_{2}^{2} \\ (a) &= \frac{1}{2}mr^{2}\left[\left(1 - \frac{4}{\pi} + \frac{4}{\pi^{2}}\right) + \left(1 - \frac{4}{\pi^{2}}\right)\right] = \frac{1}{2}mr^{2}\left(2 - \frac{4}{\pi}\right)\omega_{2}^{2} \\ \overline{I_{1} + U_{1-2} = T_{2}} : O + mg\frac{2r}{\pi} = \frac{1}{2}mr^{2}\left(2 - \frac{4}{\pi}\right)\omega_{2}^{2} \\ \omega_{2} &= \frac{2}{\pi\left(1 - \frac{2}{\pi}\right)}\frac{9}{r} = 1.7519\frac{9}{r} & \omega_{2} = 1.324\sqrt{9}r \end{split}$$

(6) KINEMATICS SINCE Q MOVES HURIZOTTALLY, (Q0) y=0 $a_{11} = (06) w_{2}^{2} = \frac{2r}{\pi} (1.7519 \frac{9}{r}) = 1.11539 \frac{1}{1}$

 $\frac{\text{KINETICS}}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$ $\frac{1}{6}$

+ | [Fy = I(Fy) = A - mg = 1.1153 mg; A = 2.12 mg





GIVEN: 14-16 CYLINGERS

OF 5-IM, PROJUS.

PRUBLEM 17:30:

[Wab= 30 rod/s])

FIND: (a) DISTANCE A

WILL HISE REFORE WB=5M4s)

(b) TENSION IN CORN A-B

HICELEM 17:31: SYSTEM IS

RELEASED FROM NEST

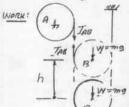
FIND: (a) VA AFTER 3 St OF

MOTION. (b) T IN COPED A-B



$$\begin{aligned} \mathcal{V}_{O} = r \omega_{g} & \omega_{g} = \frac{\mathcal{V}_{O}}{2r} = \frac{r \omega_{g}}{2r} = \frac{1}{2} \omega_{g} \\ \overline{\mathcal{V}}_{g} = r \omega_{g} = \frac{1}{2} r \omega_{g} \end{aligned} \tag{I}$$

KINETIC EMERGY: $T = \frac{1}{2} a n \tilde{\eta}_A^2 + \frac{1}{2} \tilde{I} \omega_A^2 + \frac{1}{2} \tilde{I} \omega_B^2$ $T = \frac{1}{2} m \left(\frac{1}{2} r \omega_B^2 \right) + \frac{1}{2} \left(\frac{1}{2} m r^2 \right) \left(\frac{1}{2} \omega_B^2 \right) + \frac{1}{2} \left(\frac{1}{2} m r^2 \right) \omega_B^2 = \frac{7}{16} m r^2 \omega_B^2$



SINCE CORD IS INEXTENSIBLE, WORK IS DONE ONLY BY
THE WEIGHT OF CYLINDER B

Um --Wh =-mgh

r= 512

PROBLEM 17.30: $(w_{a})_{z} = 30 \text{ rod/s}$); $(w_{E})_{z} = 5 \text{ rod/s}$) $T_{1} + U_{1-z} = T_{2}$: $T_{2} - mr^{2}(w_{a})_{z}^{2} - mg h = T_{2} - mr^{2}(w_{a})_{z}^{2}$ $h = \frac{7}{16} \left(\frac{5}{72} t_{1}\right) \frac{(30)^{2} - (5)^{2}}{32.2 t_{1}} = 2.004 \text{ ft}$ h = 2.06 ft

TENSION TAB: WE NOTE THAT POINT D MOVE TWICE

THE DISTANCE THAT A MOVES

VI-2-TAB(24)

THE DISTANCE THAT A MOVES $V_{1-\frac{\pi}{2}} - T_{AB}(24)$ FOR ONLY CYLINGER B, $T = \frac{1}{2} \tilde{L} \omega_{B}^{2}$

 $T_{1}+U_{1-2}=T_{2}: \frac{1}{2}I(\omega_{B})_{1}^{2}-2hT_{AB}=\frac{1}{2}I(\omega_{B})_{2}^{2}$ $T_{AB}=\frac{1}{4}I[(\omega_{B})_{1}^{2}-(\omega_{B})_{2}^{2}]\frac{1}{h}=\frac{1}{4}(\frac{1}{2}mr^{2})\frac{(\omega_{B})_{1}^{2}-(\omega_{B})_{2}^{2}}{\frac{7}{16}\frac{r^{2}}{9}[(\omega_{B})_{1}^{2}-(\omega_{B})_{2}^{2}]}$ $T_{AB}=\frac{1}{2}\frac{1}{2}mg=\frac{2}{3}W=\frac{2}{3}I(\omega_{B})$ $T_{AB}=\frac{1}{2}\frac{1}{2}mg=\frac{2}{3}W=\frac{2}{3}I(\omega_{B})$ $T_{AB}=\frac{1}{3}\frac{1}{2}mg=\frac{2}{3}W=\frac{2}{3}I(\omega_{B})$

NOTE: T_{AB} IS INDEPENDENT OF $(\omega_B)_A ANO(\omega_B)_2$ PROOLEM 17.31 $(\omega_B)_1 = 0$, h = 3 ft, $r = \frac{5}{12}$ ft

SINCE h AND $\tilde{\tau}_A$ ARE NOW DOWNWARD, U = +Wh = + mgh AND E_a , 2 15' $h = -\frac{7}{16} \frac{r^2}{9} \left[(\omega_B)_1^2 - (\omega_B)_2^2 \right]$ $3fz = -\frac{7}{16} \left(\frac{5}{12} \text{ ft} \right)^2 \frac{1}{32.2 ft} \frac{1}{16} \left[0 - (\omega_B)_2^2 \right]$

 $(\omega_B)_2^2 = 1271.8$ $|\omega_B|_2 = 35.66 \text{ rad/s}$ Foli) $\bar{V}_{A^2} = \frac{1}{2} r (\omega_A) = \frac{1}{2} (\frac{5}{12} \text{ fc})(35.66 \text{ rad/s}) = 7.430 \text{ fds}$ $\bar{V}_{A^2} = 7.43 \text{ fds}$

TEMSON TAB: SINCE TAB IS INDEPUNDENT OF VELEUTY,
INE AGAIN HAVE TAB = 416

17.32 GIVEN:

GIVEN: THECE SAS

MA = 6 RS

MA = 1.5 RS

SYSTEM IS RELEASED

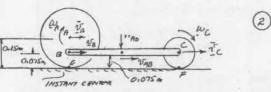
FROM IZEST

TS MM FIND! VAS AFTER

DISK A HAS

ROTATED 90°.

0.155m $\frac{1}{2}(0.15m + 0.075m) = 0.1125m$



 $\overline{V}_B = \overline{V}_{AB}$ $\omega_A = \frac{\overline{V}_B}{BE} = \frac{\overline{V}_{AB}}{0.075m}$ $\overline{V}_A = 2\overline{V}_B = 2\overline{V}_{AB}$ $\overline{V}_C = \overline{V}_{AB}$ $\omega_C = \frac{\overline{V}_C}{CF} = \frac{\overline{V}_{AB}}{0.075m}$

 $U_{l-2} = V_{l}(0.1125m - 0.075m) = (52g)(9.8)(0.0375m)$ $U_{l-2} = 1.8394 J$

 $T_{1} = 0$ $T_{2} = \frac{1}{2} m_{A} \vec{v}_{A} + \frac{1}{2} \vec{I}_{A} w_{A}^{2} + \frac{1}{2} m_{AB} \vec{v}_{AB}^{2} + \frac{1}{2} \vec{I}_{B} w_{B}^{2}$ $= \frac{1}{2} \left[(64e)(2v_{AB})^{2} + \frac{1}{2} (68e)(6.15e) \left(\frac{v_{AB}}{0.075} \right)^{2} + (54e)v_{AB}^{2} + (1.54e)(v_{AB})^{2} + \frac{1}{2} (1.54e)(0.075e) \left(\frac{v_{BB}}{0.075} \right)^{2} \right]$ $= \frac{1}{2} \left[24 + 12 + 5 + 1.5 + 0.75 \right] \vec{v}_{AB}^{2}$

= = = 2 24 + 12 + 5 + 1,5 + 0.73 VA6

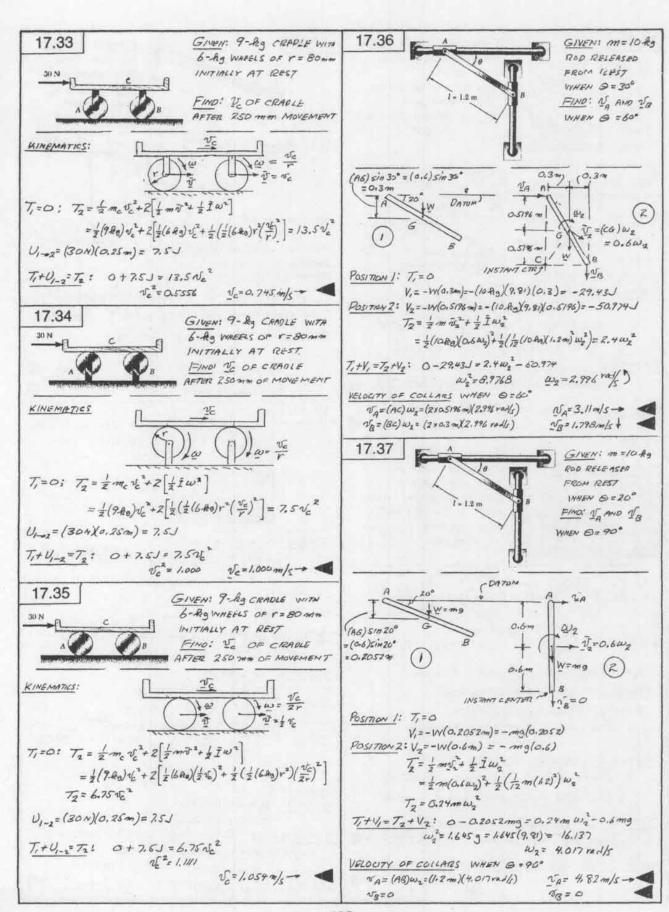
7= 21.625 VAB

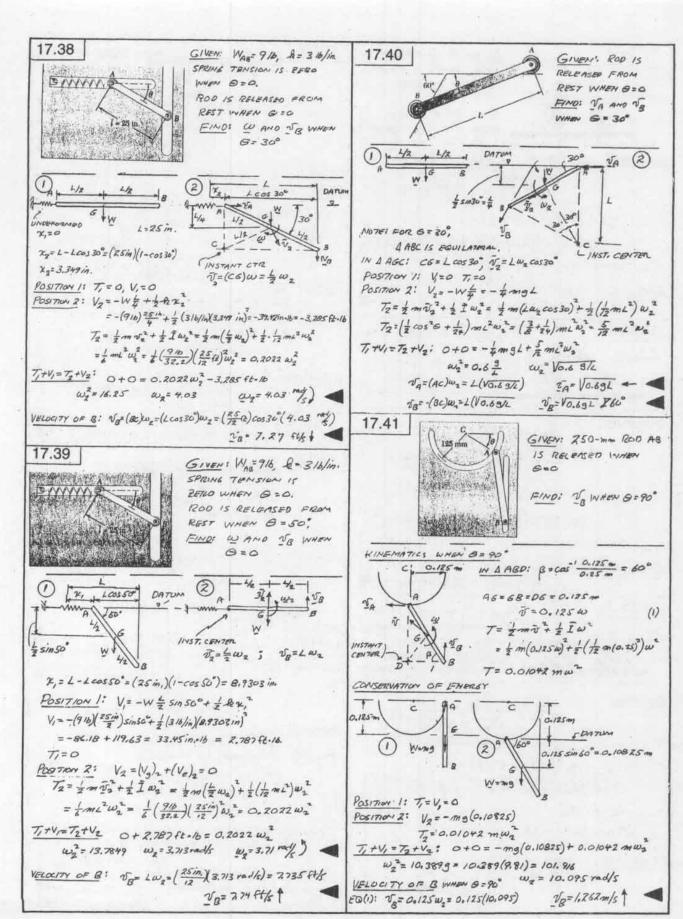
WORK ENERGY

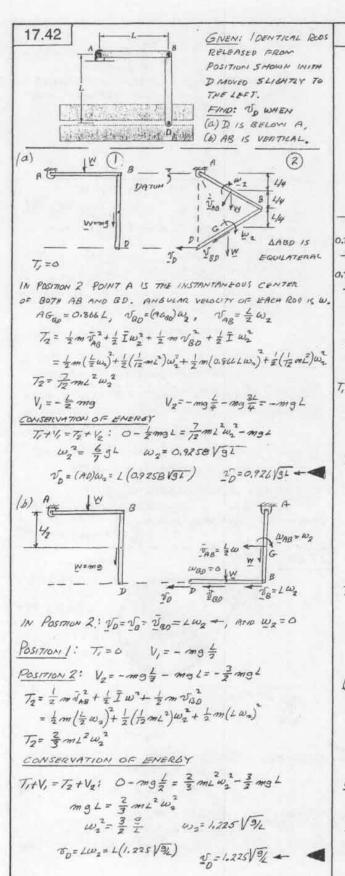
Ti+ U1-2 = T2

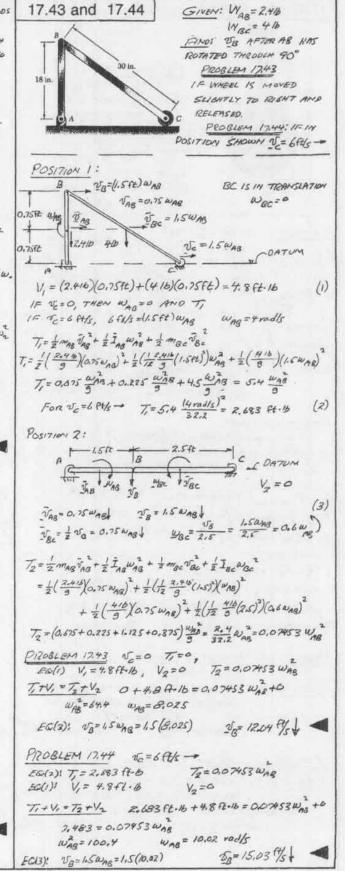
0+1,8394J = 21,625 JAB

VAB = 0.08506 VAB = 0.2916 mg











GIVEN: mag = 4-kg munege = 16-kg R = 160mm

PROBLEM 17.45:

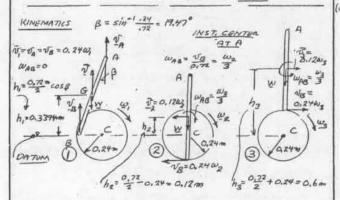
IF W = BOYPM),

FIND: W WHEN B IS

DIRECTLY BELOW C

PROBLEM 17.46:

FIND: W, SO ANGLIAR
VELICITY IS THE SAME IN
FOSTEN SHOWN AND WHEN
B IS DIRECTLY ABOVE C.



POTENTIAL ENERGY:

 $V_1 = Wh_1 = mgh_1 = (4+2g)(9.8)(0.339+m) = 13.318 \ J$ $V_2 = Wh_2 = mgh_2 = (4+2g)(9.8)(0.12m) = 4.709 \ J$ $V_3 = Wh_3 = mgh_3 = (4+2g)(9.8)(0.6m) = 23.544 \ J$

KINGTIC ENERGY $\bar{l}_{e} = I_{\text{FLYWHERE}} = (16 \cdot 80)(0.18 \, \text{m})^{2} = 0.5184 \cdot 8g \cdot m^{2}$ $\bar{l}_{AB} = \frac{1}{2} m_{AB} L^{2} = \frac{1}{2} (4 \cdot 80)(0.72 \, m)^{2} = 0.1728 \cdot 8g \cdot m^{2}$ $T_{e} = \frac{1}{2} \bar{L}_{ab} u_{e}^{2} + \frac{1}{2} m_{AB} \bar{v}_{e}^{2} = \frac{1}{2} (0.5184) w_{e}^{2} + \frac{1}{2} (4)(0.74 w_{e}^{2})$

 $= 0.2592 \, \omega_1^2 + 0.1152 \, \omega_2^2 \qquad \qquad T_1 = 0.3744 \, \omega_1^2$ $T_2 = \frac{1}{2} \, \tilde{L}_2 \, \omega_2^2 + \frac{1}{2} \, m_{AB} \, \tilde{v}_2^2 + \frac{1}{2} \, \tilde{L}_{AB} \, \omega_{AB}^2$

 $= \frac{1}{2}(0.5784)\omega_2^2 + \frac{1}{2}(4)(0.12\omega_2)^2 + \frac{1}{2}(0.1728)(\frac{\omega_2}{3})^2$ $= 0.2592\omega_2^2 + 0.0285\omega_2^2 + 0.098\omega_2^2 \qquad 7_2 = 0.2978\omega_2$

 $T_3 = SAME$ COEFFICIENT AS T_2 : $T_3 = 0.2976 \, w_3$

PROBLEM 17.45: PORTION 1 TO POSITION 2

W= 60 rpm (211) = 211 rad/s

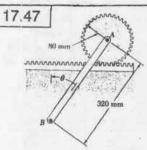
 $7. + V_1 = 72 + V_2$: 0.374 + $\omega_1^2 + 13.318J = 0.2978 \omega_2^2 + 4.769J$ 0.3744(211) + 13.318 = 0.2976 $\omega_2^2 + 4.769$

0.2976 $\omega_{2}^{2} = 23.39$ $\omega_{2}^{2} = 78.60 \text{ rad f}$ $\omega_{2} = 8.865 \text{ rad/s} \left(\frac{60}{20}\right)$ $\omega_{2} = 84.7 \text{ rg/m}$

PROBLEM 17.46: POSITION 1 TO POSITION 3 WITH W, = W3

 $T_1 + V_1 = T_2 + V_3$: 0.3744 $w_1^2 + 13.318J = 0.2976 w_2^2 + 23.544J$ 0.3744 $w_1^2 + 13.318 = 0.2976 w_2^2 + 23.544$

0,0768 ω, = 10,226 ω, = 133.2 ω, = 11,54 rads (60/20) ω, = 110.2 γρης



GIVEN:

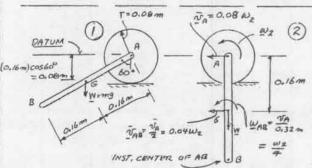
5-Rg GEAR, R=60 mm

4-Rg ROD AB

545TEM IS RELEASED

FROM REST WHEN Q=60°

FIND:



POSITION 1: T, = 0 V, = - W(0.08m) = - (4 &g)(9.81)(0.08) = -3.139.

POSITION 2: $T_2 = \frac{1}{2} m_A T_A^2 + \frac{1}{2} \bar{I}_A w_2^2 + \frac{1}{2} m_{AB} v_B^2 + \frac{1}{2} \bar{I}_{AB} w_{AB}^2$ $T_2 = \frac{1}{2} (5 + 3)(0.08 w_2)^2 + \frac{1}{2} ((5 + 3)(0.08)^2) w_2^2$

 $+\frac{1}{2}(4 \log z_0) + \frac{1}{2}(\frac{1}{12}(4 \log z_0))(\frac{\omega_2}{4})$

 $T_2 = 0.06 \omega_2^2 + 0.009 \omega_2^2 + 0.0032 \omega_2^2 + 0.00107 \omega_2^2 = 0.02927 \omega_2^2$ $V_2 = -W(0.16m) = -(4.8g)(9.8)(0.16) = -6.278 J$

7,+V,=72+V2: 0-31391=0.02927w2-6.2781

 $w_2^2 = 107.26$ $w_2 = 10.357$ rad/s $v_{E00} = 0.08 w_2 = 0.08 (10.357) = 0.829 m/s$ $\tilde{\eta}_0 = 829 mm/s = 0.08 (10.357) = 0.829 m/s$

17.48 30 mm

GIVEY:

WA = 22.5 HZ

MOTOR

DEVELOPS 3kW

EIND:

(a) MA

(b) MA

 $W_{A} = 22.5 \, H \, 2 \left(\frac{2 \, \Pi \, rad}{c \, yc \, l^{a}} \right) = 45 \, \Pi \, rad \, l_{5}$ $V_{A} \, w_{A} = V_{B} \, w_{B} : \quad (0.03 \, m) \, (45 \, \Pi \, rad \, l_{5}) = (0.160 \, m) \, w_{B}$ $w_{B} = 7.5 \, \Pi \, rad \, l_{5}$

(a) <u>PULLEY A</u>: POWER = MA WA
3000 IN = MA (45 T rad/s)

Mp = 21.2 N-m

(6) PULLEY B: POWER = MBWB

3000 W = MB(7.5 Th rad/s)

MR = 127, 3 Nom

17.49

GIVEN: MAXIMUM COUPLE THAT CAN BE APPLIED TO A SHAFT IS 15,5 kip . in. FIND! MAXIMUM HORSEPOWER THAT CAN BE TRANSMITTED AT (a) 180 pm, (b) 480 pm.

M = 15.5 kip in = 1.2917 kip ft = 1291.7 16.ft

(a) w=180 rpm (20) = 6TT rod/s

POWER = MW = (1291,7 16. Ft)(611 rad 6) = 24,348 12.16

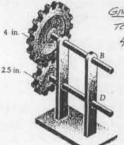
HORSE POWER = 24,348 = 44,3 bp

(b) w=480 rpm (2#)= 1671 rad/s

POWER = MW = (1291.716.12)(1871 red/s) = 64930 11.16

HORSEPOWER = 64930 = 1/8.1 hp

17.50



GIVEN MOTOR ATTACHED TO CHAFT AB DEVELOPES 4.5 kp WHEN WAS 720 ypm

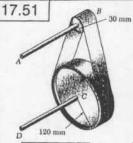
> FIND: MACHITURE OF COURT EXERTED ON (A SHAFT AB (d) SHAFT CO

(a) SHAFT AB: WAR = 220 pm (2T) = 75.398 rad/s POWER = 4.5 hp (550 82 8/6) = 2475 12.10/5

POWER = MAG WARS 2475 Ft. W/s = MAG (25.398 rouls) MAR = 32,826 /b.fz MA=32.8 16-ft

(b) SMART CO: WG = TA WAS = 410. (75.396) = 120,64 rad/s

POWER = Mcowep; 2475 ft. W/s = Mco (120,64 ran/s) M= 20.5 16. ft



GIVEN: 2.4 FW TO BE TRANSMITTED FROM A TO D ALLOWABLE COUPLES ARE MAB = 25 N.m Meo = 80 N.m FIND: RESURED MINIMUM SPEED OF SMAFT AB

SHAFT AB: POWER = MAR WAR 2400 W = (25 Nom) WAR

WAR 96 rads

SMAFT CO: POWER = MED WED 2400 W = (80 N.m) Wm Up= 30 rolls Fore was soralls, was To was 120 mm (30 rods)

War = 120 rodls

WE CHOOSE THE LARGER WAR (12000 N/s) (60) WAB = 1146 rpm

17.52

GIVEN: 30-Rg ROTOR WITH A = 200 mm CORST TO REST IN SIBNON FROM INITIAL ANGULAR VELOCITY OF 3600 PPM. FIND! MAGNITURE OF COURSE DUE TO FRICTION

I=mh = (30 kg/0.2m) = h2 Ag m2, W, = 3600 rpm (20) = 377 rad/s

SYST. MUMENTA, + SYST. EXT. IMP = SYST. MOMENTA +) MOMENTS ABOUT A: Iw, -ME = 0 (1.2 kg m (377 racks) - M (5.3 min x 605) = 0

M = 1.423 Non

17.53

GIVENI 4000-16 PLYWHEEL WITH A = 2711. COASTS TO REST FROM ANGULAR VELOCITY OF 450 YPM. FRICTION COUPLE IS OF MAGNITUDE 125 16-10. FIND! TIME REQUIRED TO COAST TO REST

I=m-E= (4000/6) (27 in) = 628.88 16.ft.5 w = 450 rmn (211) = 47.125 rad/s M= 12516-in = 10.417 16. Ft



SYST. MONENTA, + SYST. EXT. IMP, = SYST. MOMENTA +) MOMENTS ABOUT A: IW, - Mt = 0 t = 1w, = (678.80 16-(2-5)(47.125 rods) = 2845 s 10.417 16. Ft

> t=28455 (nin) £= 47.4 mm.

17.54



GIVEN: Wa= 816, 17=3in., 18=4.5in. DISKS OF SAME MATERIAL AND THICKNESS. M=2010.in., W,=0 FIND: TIME UNTIL Wa= 960 YPM

Wa = (10) WA = (4.5 in.) (8 10) = 18 16 I=IA+IB= 1 810 (3 A)+1 1816 (45 ft) = 0.0470716. Ft.5 W2 = 960 rpm (21 50) = 100.53 rad/s, M=2016-in = 1.667 16-ft

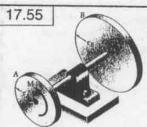
$$\begin{pmatrix}
\hat{I}_{\omega_{i}=0} \\
\hat{C}
\end{pmatrix} + \begin{pmatrix}
\hat{M}_{i} \\
\hat{C}
\end{pmatrix} = \begin{pmatrix}
\hat{I}_{\omega_{2}} \\
\hat{C}
\end{pmatrix}$$

SYST. MOMENTA, + SYST. EXT. IMP, = SYST. MOMENTA2

+) MOMENTS ABOUT C: O+ ME = IW, t= 1 w2 = (0.04707 10-4.5)(100.53 rad/s) 1.667 16. Ft

t = 2.839 5

£=2.845



GIVEN: Ma=3.83, Ya=100 mm,
YB=125 mm, DISKS OF SAME
MATERIAL AND THICKNESS.

CU,=200 Ypm, Wa=800 Ypm

Lina=35.

FIND: MAGNITUDE OF

COURE M

ms= (ra)2 ma= (125 mm 2 3 kg = 4.6875 kg

$$\begin{split} \bar{I} &= \bar{I}_{A} + \bar{I}_{B} = \frac{1}{2} \left(3 \, \Re g \right) \left(0.1 \, m)^{2} + \frac{1}{2} \left(4.6875 \, \Re g \right) \left(0.125 \, m \right)^{2} = 0.05162 \, \Re g \cdot m^{2} \\ \omega_{s} &= 200 \, v pm \left(\frac{27}{60} \right) = 20.944 \, rad/s \; ; \; \omega_{z} = 800 \, v pm \left(\frac{27}{60} \right) = 83.716 \, rad/s \end{split}$$

$$\begin{pmatrix}
\hat{I}_{\underline{\omega}_1} \\
\hat{G}
\end{pmatrix} + \begin{pmatrix}
\hat{M}_{\underline{\omega}_1} \\
\hat{G}
\end{pmatrix} = \begin{pmatrix}
\hat{I}_{\underline{\omega}_2} \\
\hat{G}
\end{pmatrix}$$

SYST. MOMENTA, + SYST. EXT. IMP = SYST. MONEYTA2

+) MOMENTS ABOUT G: $\tilde{I}\omega_1 + Mt = \tilde{I}\omega_2$ $M = \frac{\tilde{I}}{t}(\omega_2 - \omega_1) = \frac{0.05/62 \cdot k_3 \cdot m^2}{25}(83.7\% \text{ roll}_5 - 20.744 \text{ roll}_5)$

M=1.081 N.m

17.56



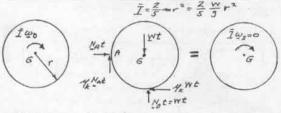
GIVEN STABLE OF WEIGHT W

MR = COEF. OF KINETIC PRICTION

FIND: EXPRESSION FOR TIME

REGULARD FOR STABLE TO

COME TO REST.



SYST. MOMENTA, + SYST. EXT. IMP_{r-2} + SYST. MOMENTA₂ $+y^{\dagger}$ COMPONENTS: $0 + N_0 t + M_0 N_0 t - W t = 0$ (1) +x COMPONENTS: $0 + N_0 t - M_0 N_0 t = 0$ (2)

FROM EQ(2): NA = M/K NQ (3) SUBSTITUTE INTO EQ(1): Nat +1/K (M/K NQ)t -Wt

$$N_B = \frac{1}{1 + 4k^2} \mid N$$

EG(2):

+) MOMENTS ABOUT 6: $\bar{I} w_0 - (\eta_0 N_0 t) r - (\eta_0 N_0 t) r = 0$ $\frac{2}{5} \frac{W}{9} r^2 w_0 - \frac{4\kappa^2}{1 + 4\kappa^2} rWt - \frac{4\kappa}{1 + 4\kappa^2} rWt = 0$ $\frac{2}{5} \frac{V}{9} w_0 - \frac{4\kappa + 4\kappa^2}{1 + 4\kappa^2} t = 0$

t= 1+42 · 2 / 3 wo

17.57



GIVEN: m= 3 Ag, r=125 mm

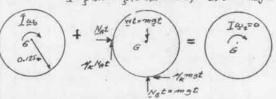
W= 90 rad/s

Mk= 0.10

FIND: TIME RESOURCE FOR

Street to come to REST

I= = = = = = = = (240/0.125m) = 18.75 ×10 3 Agran



=YST. MOMENTA, + SYST. EAT. IMP, = SYST. MOMENTA2 +y1 COMPONENTS: $O + N_0 t - M_0 N_0 t = 0$ (1) +1 COMPONENTS: $O + N_0 t - M_0 N_0 t = 0$ (2) EO(2): $N_0 = M_0 N_0$ EO(1): $N_0 t - M_0 (M_0 N_0) t - m_0 t = 0$

 $N_{\mathcal{B}} = \frac{mg}{1+q_{\mathcal{K}}^2} = \frac{(2+g)}{1+(6+io)^2} = 29/39 \text{ N}$ $N_{\mathcal{A}} = -q_{\mathcal{K}}N_{\mathcal{A}} = 0.1(29/39N) = 2.9/39N$

+) MOMENT ABOUT 6: $\overline{I} = -\frac{1}{4} N_A t^2 t^2 - \frac{4}{4} N_B t^2 t^2 = 0$ $t = \frac{I \omega_0}{4 k^2 (N_B + N_B)} = \frac{(18.75 \times 10^{-3} t \cdot 9 \cdot m^2)(90 \cdot ra.l/t)}{(0.105 \times 0.125 m)(29.139N + 2.9139N)}$ t = 4.212 s

17.58 and 17.59

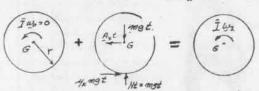


GIVENI DISK AT REST PLACED IN CONTACT WITH BELT.
COEF. OF KINETK FRICTION = 44.

PROBLEM 17.59: FOR Y= 510,

W= 510, V= 50 (L/5, 4/6 = 0.20.

Wama Istmr



SYST. MOMENTA, + SYST. EXT. IMP, 2 = SYST. MOMENTA

+) MOMBUTS ABOUT 6: $0 + (-\eta_k mgt)r = \bar{1}\omega$, (1) FINAL ANGUAR VELOCITY: $v = r\omega$,; $\omega_2 = \sqrt[3]{r}$

EQ(1): (4kmgt)r= 1mr2(3) PROBLEM 17.58: t= 294k

NOTE: RESULT IS INDEPENDENT OF IN AND F. PROBLEM 17.59: DATA: N=50 FUS, 4/k=0.20

t= 7 . 50 Fels ; t=3,885

17.60 and 17.61



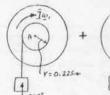
GIVENT 350- By FLYWHEFL OF R= 600 mm.

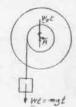
W, = 100 rpm] WHEN POWER IS CUT OFF MAID SYSTEM

FROMEN 17.600 FINDS TIME REQUIRED FOR SYSTEM TO COMME

PROBLEM 17.61: FIND TIME 111712 Was = 46 x pm }

T=mk = (350 kg)(0.6 m) = 126 from







SYST, MOMENTA, + SYST. EXT. IMP - SYST. MOMENTA;

+ MOMENTS ABOUT A: most r+IW-most r= migr+IW2 SUBSTITUTE: TIFFW, AND TE YOU.

(mr2+1)w, -mgtr = (mr2+1)w2

 $t = \frac{6.075 + 178}{224.87} (\omega_1 - \omega_2)$ $t = 0.49824 (\omega_1 - \omega_2)$

PROBLEM 17.60:

W=100 rpm (20)= 10.472 rolls w2=0

EG(1): t=0.48864(10.472 rod/c) t=5.225

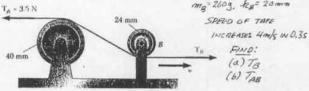
PROBLEM 17.61: 00, = 100+pre (20) = 10.472 +0.1/5 wg = 40 rfm (27) = 44,189 rad/s

EG(1): t=0.49844(10.472-4.189)

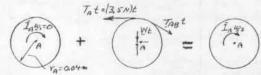
£= 3135

17.62

GIVEN: MA=6009, RA=32 mm mg= 260g, kg= 20mm



In=m, Ra = (0.6 Rg(0.032m)2 = 614.4 ×10.6 Feg.m2 IB = mg Ra= (0.26 Ray 0.020m)= 104 × 10-6 feg m2 DRUM A: ASSUME W, = 0 THEN Wa= T = 4m/s = 100 rolls



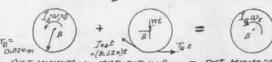
+] MOMENTS ABOUT A: C) -3,5t(0,04m) + TABt(0,04m)= IA W2 -3.5(0.35×0.04)+Tm3(0.35×0.04)=(614.4×10 (100rad/s) -0.042 + 0.012 TAB = 0.06144 708=8.62 H

(CONTINUED)

17.62 continued

DRUM B WE RECALL: TAS 8.62N

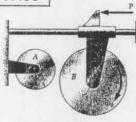
w2 = AV = 4 m/s = 166.67 red/s



SYST. MOMENTA, + SYST. PAT. IMP 1-2 = SYST. MOMENTA, +) MOMENTS ABOUT B: O + Tatr - Tastr = I wa TB(0.35 (0.05400) - (8.62N)(0.35)(0.02400) - (104210 fg-00) (166.67 00)

TR= 11.03 N

17.63



GIVEN: DISK A IS AT REST WHEN DICKS A AND B ARE BROUGHT INTO CONTACT

SHOW THAT DINAL WA DEPENOS ON ONLY Ub AND ma

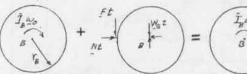
DISK A: (Wa) 0+0

$$\frac{\overline{I_{a}}(u_{a})_{a}=0}{a^{2}r_{a}} + \left(\overline{I_{a}}\right)^{NE} = \left(\overline{$$

SYST, MONENTA, + SYST, MOMENTA,

+) MOMENTS ABOUT A: O+(PE) r= IA WA (1)

DYSK B: (WB) = WO



+) MOMENTS ALLOW 1 2: I WE (Ft) " = I we SUBSTITUTE FOR FE FROM EQ(1) I wo - I w To = I we (2)

FOR FINAL ANGULAR VELOCITIES! VAN = TBW; Wa = TB WB

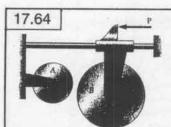
$$\frac{\bar{I}_{E}\omega_{b} - \bar{I}_{A}\omega_{B}\left(\frac{v_{B}}{v_{A}}\right)^{2} = \bar{I}_{E}\omega_{B} }{\omega_{B}^{2} \frac{\omega_{b}}{1 + \frac{\bar{I}_{A}}{I_{B}}\left(\frac{v_{B}}{v_{A}}\right)^{2}}$$

$$(3)$$

BUT FOR UNIFORM DISUS: $\frac{\vec{I}_A}{\vec{I}_B} = \frac{\frac{1}{2}m_A r_A}{\frac{1}{2}m_B r_B^2} = \frac{m_A}{m_B} \left(\frac{r_A}{r_B}\right)^2$

SUBTITUE INTO EG(1):

THUS, WE DEDETIES ON ONLY WE AND THE



GIVEN: W= 7.54, FA= 6 in. Wa 104, 18= 8in. Wa= 900 Fpm

FIND! (a) FINAL WA AND WB (6) I MODESE OF FRICTION FORCE EXERTED ON DISK A.

In = 1 WAY = 1 750 (6 12 12) = 7.5

I = 1 Wars = 1 1016 (8 12) = 20

DISKA:



SYST MOMBITA, + SYST. GXT. IMP, = SYST. MOMBATA. +) MOMENTS ABOUT A: O+(FX) "A = IWA (FE)(0,552)= 7.5 ap

DISK B: (Wg) = W = 900 YPM (21) = 30TT rad/s







(2)

SYST. MOMENTA, + SYST. EXT. IMP = 2 = SYST MOMENTAS +] MOMENTS ABOUT E: Ig(Wo) - (FE) YO = Ig WB 20.30T - (FEX (1) = 30 WB

SURSATUTE FROM EQ(1):

20 - 3017 - / 25 wn) (12) = 20 wg 3071 - 0,5675 WA = WB

FIMAL VELOCITIES OCCUR WHEN;

VANA = VB Wg; Wg = VA WA = 6in. WA = 0.75 WA (3)

SUBSTITUTE FOR WB FROM (2) INTO (3)

3077-0,5625 NA = 0,75 WA

30T = 1.3/25 WA WA= 71.807 nody WA=71.807 nod/s (60) Wa= 685.7 mm

EQ(3) WB = 0.75 WA = 0.75 (685.74pm) WB = 574.3 xpm

IMPULSE OF FE EXERTED ON DISK A.

Ft = 7,5 WA = 7,5 (71807 rodb) EO(1) Ft = 4.18 16:5

> W= 686 rpm 5 WB= 514 4100)

17.65



SHOW THAT SYSTEM OF MOMENTA IS EQUIVALENT TO A SINGLE VECTOR AND EXPRESS THE DISTANCE FROM 6 TO THE LINE OF ACTION OF THE VECTOR IN TERMS OF A, T, AND W.

+ 9 MOMENTS ABOUT 6 Iw=(mv)d d = Iw = mik

17.66

SHOW THAT SYSTEM OF MOMBUTA IS EQUIVALENT TO MEW LOCATED AT P WASTE 6P= 42/F

mitemiew +) MOMENTS ABOUT G Iw=(mrw)6P

17.67 FOR A RIGID SLAS IN PLANE MOTION, SHOW THAT HA IS EQUAL TO IAW, IF AND OMEY IR (a) A IS THE MASS CENTER, (b) A IS THE INSTANTANEOUS CENTER DE ROTATION, (c) To IS DIRECTED ALONG WINE AG.



FOR GENERAL PLANE MOTION V = VA +N = VA + WXY6/A

SYSTEM OF MOMENTA-

MOMENTS ABOUT A Ha = I W+ Yola x mis HA= IW + m TUAX (TA+WXYSIA) HA = Iw + mrain x VA + mrain x (w + YEIA)

SINCE WILYGLA THE TRIPLE VECTOR PRODUCT CAN BE WRITTEN: YEINX (W+YEIN) = YEIN W THOS HA= IW+ mrala + IA + mrana

BY PARALLEL-AXIS THELOSOMS In = I + m Y WAR WE NOW HAVE HA = IAW + MYSIA YTA .. HA = IA W, ONLY WHEN YOUN XON = 0 (a) TOLA = 01 A COUNCIDES WITH & (b) VA= 0: A IS INSTANT. CENTER

(C) SIA AND VA ARE COMMENCE VA IS DIRECTED ALONG AE.

17.68



GIVEN: IMPUSIVE FORCE F IS

REPLIED TO SLAB.

SHOW THAT: (a) INST. CENTER

IS AT C AND GC = \$\frac{1}{2}\beta f.

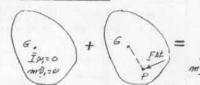
(b) IF F WEEL MENLED AT C

THEN P IS THE INST. CENTER

At TIME OF APPLICATION OF F AT THE CONTER

OF PERCUSSION P.

(a)



SYST. MOMENTA, + 5YST, EXT, $IMP_{rel} = 5YST$, $MOMENTA_2$ +/COMPONENTS: $FDt = m\bar{\chi}_2$; $\bar{\chi}_2 = \frac{FDt}{m}$ (1)

+) MOMENTS ABOUT 6: $(Fat)(6P) = \overline{1}\omega_{\theta}$ $\omega_{2} = \frac{FAt}{\overline{1}}(6P) = \frac{FAt}{mh^{2}}(6P) \qquad (2)$



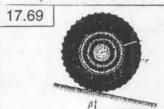
KINEMATICS: THE INSTANTANEOUS CONTER MUST BE LOCATED ON A LIME L TO $\bar{\nu}_{x}$, THAT IS, ON GP. ALSO, $\bar{\nu}_{z}$ = (60) $\bar{\nu}_{y}$ GC= $\frac{\bar{\nu}_{x}}{\bar{\nu}_{z}}$

SOBSTITUTE FROM (1) AIII (2) $GC = \frac{FAL}{m}$ $GC = \frac{\frac{1}{4}}{6P}$



WE NOW ASSUME THAT F IS
ARRIVED SO THAT THE NEW CENTER
OF PETICUSSION P' IS LOCATED AT C.
FROM PART Q, WE NOTE
THAT NEW INST. CENTER WILL BE
LOCATED AT C! WHERE $GC' = \frac{k^2}{6P^2} = \frac{k^2}{GC} = \frac{k^2}{k^2/CD} = 6P$

THUS NEW INSTANTANEOUS CENTER IS LOCATED AT P

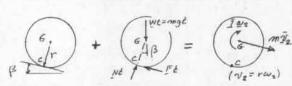


GIVEN; R = RADIUS OF GYRATION T, = 0

FINO: (a) \$\vec{\pi}_2\$ AT TIME \$\tau\$

(b) \$A_{\vec{\pi}_2}\$ REGURED

TO PREVENT SUPANI



SYST, MOMENTA, + SYST, EXT, IMP₁ = SYST, MOMENTA₂ + $\frac{1}{2}$ MOMENTS ABOUT C: (Wt sing) $r = \overline{1} w_1 + m \overline{v}_2 r$ $mgt sing = m - \overline{e}^2 w_2 + m r^2 w_1$ $w_2 = \frac{rgt}{r^2 + \overline{R}^2}$ (1)

(CONTINUED)

17.69 continued

(a) $\overline{y}_{z} = rw_{z}$; $\overline{y}_{z} = \frac{r^{2}}{r^{2} + R^{2}} gt sin \beta \sqrt{g}$ (b) $+\sqrt[3]{components}$; $Nt - mgt cos \beta$ $N = mg cos \beta$ $N = mg cos \beta$ $N = mg cos \beta$ $N = \frac{1}{rt} w_{z} = \frac{mk^{2}}{rt} \cdot \frac{r^{2}gt}{r^{2} + k^{2}} sin \beta = \frac{k^{2}}{r^{2} + k^{2}} mg sin \beta$ $N = \frac{1}{rt} w_{z} = \frac{mk^{2}}{rt} \cdot \frac{r^{2}gt}{r^{2} + k^{2}} sin \beta = \frac{k^{2}}{r^{2} + k^{2}} mg sin \beta$ $N = \frac{1}{rt} w_{z} = \frac{k^{2}}{r^{2} + k^{2}} sin \beta = \frac{k^{2}}{r^{2} + k^{2}} sin \beta$

17.70

GIVEN) Y = 1.5 in.

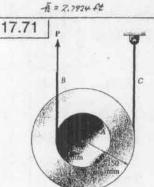
WHEEL STARTS INDOM REST
AND ROLLS WITHOUT SLIDING. $\overline{\chi}_{z} = 6$ in./s AT $\pm = 30$ s.

Find: $\overline{A}2$

 $\frac{\sqrt{1} = \mu_1 = 0}{\sqrt{2} \cdot \frac{1}{\sqrt{2}}} = \frac{\sqrt{1} \cdot \frac{1}{\sqrt{2}}}{\sqrt{2}} = \frac{1$

SYST. MOMENTA, + SYST. EXT. $IMP_{r=1} = SYST$ MOMENTA n+) MOMENTS ABOUT $c: nngt(rsing) = \overline{1} \omega_2 + m \overline{v}_2 r$ $fingtr sing = nn \overline{n}^2 \omega_2 + nn r^2 \omega_2$ $gt r sing = (-\overline{n}^2 + r^2) \omega_2$ $gt r sing = (-\overline{n}^2 + r^2) \omega_2$ (1) $DATA: r = \overline{g} / t, \quad \overline{v}_2 = 6 in / s = 0.5 \text{ ft/s}, \quad t = 30.5$ $\omega_2 = \frac{\overline{n}_n}{r} = \frac{\overline{n}_n sit/s}{\sqrt{n}} = 4 rodb$

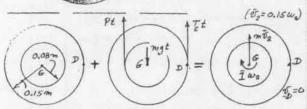
 $ES(1) = \frac{1}{32.247/5}(306)(\frac{1}{6}4)\sin 15^{\circ} = \left[\frac{1}{16} + (0.12546)^{\circ}\right](4rad/6)$ $= \frac{1}{16} + (0.015625 = 7.8131) = \frac{1}{16} = 7.7975$



GIVEN: M=3.kg, k=100 mm PULLEY IS 47 REST WHEN P=24 N IS APPLIED TO B

-B = 2.79 ft

END: (A) T AFTER 1.55 (W TENSION IN COME C



SYST. MOMENTA, + SYST EXT INP, <math>-2 = SYST. MOMENTA 2 +) MOMENTS ABOUT D: $PL(0.08+0.15) - mgt(0.15) = \overline{1} \omega_2 + m\overline{v}_1(0.15)_2$ $(24N)(1.55)(0.22) - (34g)(7.81)(1.55)(0.15) = (34g)(0.16)(\omega_2 + [34g)(0.16)(\omega_2 + [$

17.71 continued

WE HAVE FOUND J- 2.58/m/st

+ COMPONENTS: Pt + Tet - mgt = mis (24MX1.55) + Te(1.55) - (3kg)(9.81)(1.55) = (3kg)(2.551mm/s) 36+1.5Te - 44.145= 7.653 45 Te= 15.798

TE= 10.53 N

17.72



GIVEN! TWO 1416 CHINDERS OF RADIUS r= 5 in. SYSTEM IS RELEASED FROM REST WHEN t=0 FIND: (a) No AT #= 35. () TENSON IN BELT COMNECTING CYLINDERS



KINEMATICS CYLINDER & INSTANT. CENTER OF B IS AT C. Va= rwa V=V=zrwg CYLINOUTE A W = 200 = 200 P

CYLINDER A:



SYSTI MOMENTA, + SYST, BYT, IMP, = SYST, MOMENTA

+) MOMBETT ABOUT A: (QE) = I WO (at) r = = mr2(2 mg)

Qt=mrwg

CYLINDER B:





10

t) MOMENTS ABOUT C: (mgt)r-(Ot)2r= Iug+migr mgtr-(at)(2+) = 1 mr2 +m(rwg)r 121 mgt-2(ot) = 3mrwg

SUBSTITUTE FOR (Qt) FROM(1) mgt - 2(mray) = 3 mrug WR = = = 9#

> 1 = 2 gt TR= TWB;

Ot=mrug; Qt=mr(2 9t) FO(1): Q= = = = = W

DATA: W= 1416, t= 35

(a) $\sqrt{N_B} = \frac{2}{3}gt = \frac{2}{3}(32.274/5^2)(35)$ De= 27.6 P/s +

(b) Q= = = = = = (1416) = 4/6 TENSION IN CONNECTING BELT = 416 17.73



GIVEN! TWO 14-16 CYLINDERS OF RADIUS SIA. HATTALLY WA= 30 rod/s) FINDS (a) TIME REGULED FOR WA TO SE REDUCED TO WA = 5 rads)

(6) TENSION IN BELT CONNECTING CYLINDERS

B TB

KINEMATICS: CYLINDER B INSTANT. CENTER OF B IS AT C. V= rwg Vp=T= 2rws

CYLINDER A:



SYST, MOMENTA, + SYST. FAT., IMP, +2 = SYST. MOMENTA + J MOMENTS ABOUT A: I(u), - (GE) = I (u)= (Qt) = = 1 mr2 [(wn), - (win)] (QE) = 1 mr 2(WB), - 2(WB) (1) Qt= mr (wB), - (WB)2

CYLINDER B:



SYST. MOMENTA, + SYST. EST. IMP,-2 = SYST. MOMENTA, +2 MOMENTS ABOUT C:

 $\bar{I}(\omega_{\theta})$, + $m(\tau_{\theta})$, $r + \varphi t(zr) - fmgt)r = <math>\bar{I}(\omega_{\theta})$ + $m(\tau_{\theta})$ rSuasmost I = 2 mr, (vg)= r(va), AIN (va)= r(wa);

Qt(2r)-(mgt)r= mr2 [(wa)2-(a8)]+ 1/2 mr2 [(wa)2-(a8)] 2Qt -mgt = 3 mr (wB) 2 - (wB), (2)

SUBSTITUTE FOR QT FROM (1):

2mr (wg), -(wg)2 -mgt = = mr (wg)2-(wg),

t= 7 5 [(w8), -(w8)2] (3) SUBSTITUTE FOR & FROM (3) INTO (1)

Q (7/2 9 ((wa), -(wa)) = mr ((wa), -(wa))

14)

DATA: (WA) = 30 rad/s -> (WB) = = 1(WA) = 15 rads (wa) = 5 rads - (wa) = 1 (wa) = 2,5 rads W= 1416, Y= Sin. = 5/12 Ft

t= 7. (5/12 (1)) [15 radk - 2.5 radk] (a) EG(3):

Z=0.566 5 t=0.5661 s

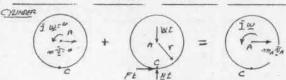
(b) FG(4): Q= = = (1410)= 416 TENSION IN CONNECTING BELT = 4 16



CYUMPER: ME BLO, Y= 2400000 CAROMSE: ME = 3-29

SYSTEM AT REST WHEN
P= 10 N APPLIED FOR 1,25

Fire: (a) VB, (b) VA



STUT MOMENTA, + SYST. EXT. IND $_{1-2}$ = SYST. MOMENTA? +) MOMENTS ABOUT C: $Q = \overline{1}\omega - m_A \overline{V}_A Y$ $\omega = \frac{m_A \overline{V}_A Y}{\overline{1}} = \frac{m_A \overline{V}_B Y}{4m_A Y^2}$; $\omega = \frac{2 \overline{V}_A}{Y}$ (1)

+ COMPONENTS: Ft = MAT

CARRIAGE:

± components: Pt - Ft = mg 58

$$Pt = m_A \dot{V}_A = m_B v_B$$

$$Pt = m_A \dot{V}_A + m_B v_G \qquad (2)$$

KINEMATICS! ASSUME ROLLING

$$\bar{v}_A = \frac{Pt}{m_A + 3m_B}$$
 (5)

DATA: m= 8-kg, m= 3-kg P=10N, t=1.25

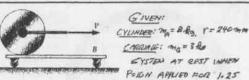
$$rac{7}{4} = \frac{(10 \text{ N})(1.25)}{8 \cdot 89 + 3(3 \cdot 49)} = \frac{12}{17} \text{ m/s}$$

V = 0.706 m/s -

$$V_{g} = 3 V_{A} = 3 \left(\frac{12}{17} m/s\right) = \frac{34}{17} m/s$$

$$\bar{V}_{g} = 2.12 m/s \rightarrow$$

17.75



FIND: (a) 28, (b) 7A

CYLINOGA + WE PE



(6)

SYST. MOMERTA, + SYST. EYT. $IMP_{rel} = SYST$ MOMERTA 2 +2 MOMERTS ARGUT A; $(Fz)r = \overline{1}u$ $(Fz)r = \frac{1}{2}m_{rel}r^{2}u$

$$Ft = \frac{1}{2} m_T u \qquad (1)$$

$$Pt = Ft = m_T \tilde{e}_1 \qquad (2)$$

 \pm , CONFORMATS: $Pt - Ft = m_A \tilde{v}_A$ (2) t) MUMBETS ABOUT C: $Pt = \tilde{I} \omega + m_A \tilde{v}_A r$ (3)

CARRINE;

mage

fe mage

mage

evst. Momenth, + Syst. Fat. IMP, = 2 = Eyst. Momenth.

L. COMPONENTS;

Fe = mage

KINEMATICS ASSUME ROLLING

SURSIDE FOIL) = FOIL): Pt - $\frac{1}{2}m_{\mu}u = m_{\mu}v_{\mu}$ $\bar{v}_{A} = \frac{Pt}{m_{\mu}} - \frac{1}{2}r\omega$

SUBSTITUTE
$$EG(i) \rightarrow EG(s)$$
:
$$(7)$$

 $\frac{1}{2}m_{0}rw = m_{0}(v_{0}-rw)$ Substitute $EG(\delta) \rightarrow EG(\delta)$:

$$\frac{1}{2}m_{A}r\omega = m_{B}\left(\frac{Pt}{m_{A}} - \frac{1}{2}\gamma\omega - r\omega\right)$$

$$\left(\frac{1}{2}m_{A}r + \frac{3}{2}m_{B}r\right)\omega = \frac{m_{B}}{m_{B}}Pt$$

$$\omega = \frac{2Pt}{r} \left(\frac{m_A r}{m_A} + \frac{2m_B}{m_A} r \right)$$

$$\omega = \frac{2Pt}{r} \left(\frac{m_B}{m_A} \right) \frac{1}{m_A + 3m_B}$$
(8)

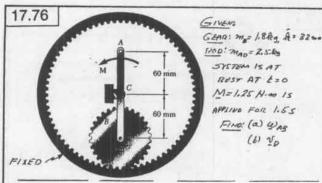
DATA: ma = 8-29, ma = 3-29
P=10N, E=1.25, Y=0.24m

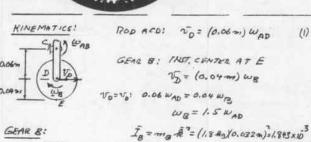
$$EG(\theta): \qquad \Omega = \frac{2(\log N, 2s)}{0.24m} \cdot \frac{2.489}{8.49} \cdot \frac{1}{8.49 + 3(2.424)}$$

$$U = \frac{37.5}{10} \text{ rad/s}$$

$$Ea(6)$$
; $\tilde{v}_{A} = \frac{Pt}{m_{A}} - \frac{1}{2} r \omega = \frac{(ION)(1.25)}{6Rg} - \frac{1}{2}(0.24m)(\frac{37.5}{12} radje)$
 $\tilde{v}_{A} = 1.5 - 0.2647 = 1.235 cm/s$

UB= 0,706 m/s ->





$$\frac{GEAR B:}{I_B = m_B R^2 = (1.8 R_0)(0.032 R_0) = 1.885 \times 10^3}$$

$$\frac{I_B = m_B R^2 = (1.8 R_0)(0.032 R_0) = 1.885 \times 10^3}{R_0 \cdot m^2}$$

$$+ Dt D = I_0 I_0 I_0$$

$$I_0 I_0 I_0$$

$$I_0 I_0$$

$$I_0$$

= SYST. MOMENTA, + SYST. ETT IMP, = SYST. MOMENTA,
+) MOMENTS ABOUT E: (Dt)r = Ig wg + mg vg r

Dt(0.04m) = (1.843×10⁻³kg·n²) wg + (1.8kg×0.04m) wg

Dt = 0.11808 (1.5 w/AD) = 0.1771 WAD (2

 $\frac{ROD \ ACD}{ACD}: \ \bar{I}_{AD} = \frac{1}{12} m_{AD} (AD)^{2} = \frac{1}{12} (2.5 \frac{2}{12}) (0.12 m)^{2} = 3 \times 10^{-3} \frac{2}{12} g. m^{2}$ $I_{AD} = 0.06 m_{AD} =$

SYST, MOMENTA, + SYST. EXT. IMP, -2 + SYST. MOMENTAZ

+) MOMENTS AROUT C: $Mt - (Dt)(0.06m) = \bar{I}_{AO} \, \omega_{AO}$ $(1.25 \, N \cdot m)(1.56) - (Dt)(0.06m) = (3 \times 10^{-3} \, eg. \, m^2) \omega_{AO}$ $1.875 - 0.06 (Dt) = 3 \times 10^{-3} \, \omega_{AO}$

SURSTITUTE FOR DE FROM EQ(2)

1.875-0.06 (0.1771 WAD) = 3×10 WAD

WAD= 137.6 red/)

FG.(1): Vp = (0.06m) wp = (0.06m)(1326 radk)

17.77



GIVENS SPARE OF RADIUS TO PLACED ON FLOOR (AT t=0) WITH T=0 AND W=U_0).

COEF OF KINETH FRAKTION = 4/2 FINDS (a) TIME t, WHEN ROLLING WITHOUT SCIONS STARTS

() TANO W AT Z=t,

(G miles

SYST. MOMENTA + SYST. EXT. IMP = SYST. MOMENTA

 $+\dagger y$ comparents: $N\xi_{-}W\xi_{+}o$ N=W=mg (1)

 \pm , \times COMPONENTS: $Ft_1 = m\overline{\nu}_2$ (2) \pm) MOMENTS ABOUT 6: $\pm \omega_1 - Ft + \bar{1}\omega_2$ (3)

+) MOMENTS ABOUT G: $\bar{I}\omega_0 - FERT_1 = \bar{I}\omega_2$ (3)

 $\frac{EQ(3)!}{\frac{2}{3}mr^{2}w_{0} - (4_{h}m_{9})rt_{1} = \frac{2}{5}mr^{2}w_{2}}{w_{2} = w_{0} - \frac{5}{2}\frac{4_{h}gt_{1}}{r}}$ (5)

SLIDING STOPS WHEN $\overline{v}_2 = ru$ $\frac{4}{2} = ru - \frac{5}{2} \frac{4}{8} gt,$ $\frac{7}{2} \frac{4}{8} gt = ru - \frac{5}{2} \frac{4}{18} gt,$ $\frac{7}{2} \frac{4}{8} gt = ru - \frac{5}{2} \frac{4}{18} gt,$

Ea(4): \$\vec{1}_2 = 1/49 t = 1/49 (\vec{2} + 1/49) \$\vec{1}_2 = \vec{2}{3} \tau \vec{1}{2} - \vec{1}{2}

Eals): $\omega_z = \omega_o - \frac{5}{2} \frac{4\kappa^3}{\Gamma} \left(\frac{2 \text{ rws}}{7 - 4\kappa^3} \right)$ $\omega_z = \frac{2}{7} \omega_e \lambda$

17.78

GIVEN: SPARE OF RADIUS Y
PLACED OM FLOWN WITH
VELUCITYS SHOWN, F
FINAL VELOCITY IS TO BE ZEAD,
FINO: (a) BUD IN TERMS OF TIME Y.
(W) TIME REQUIRED TO COME TO BET

Two + Givt = Two or of the state of the stat

SYST. HOMENTA, + SYST. EXT. IMP == + SYST. MOMENTAL + (4 COMPONENTS: NE-WE = N=W=mg (1)

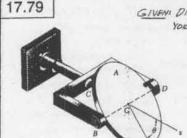
* X COMPONENT: mino - Ft = 0 Ft = mino (2)

+) MOMENTS ABOUT 6: 146-(1=6) == 0 [31

SURSTITUTE FOR Ft (from EQ 2) AND $\overline{I} = \frac{2}{5}mr^4$

EG.(3): $\frac{2}{5}mr^2w_0 - (m\tau_0)r = 0$ $w_0 = \frac{5}{2} \cdot \frac{\tilde{V}_0}{r}$

 $\frac{FG(2)}{F} = \frac{m\vec{v_0}}{f_{\text{mg}}}; \quad t = \frac{\vec{v_0}}{f_{\text{kg}}}$



GIVEN: DISK: W= 2.510, r = 4in. YOKE: Wy=1.510, &= 3 in. WHEN 6=0, W= 120199

FIND: WY WHEN 5 = 90°

ROTATION ABOUT & AXIS:



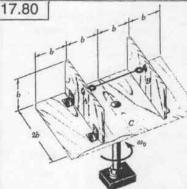
SYST. MOMENTA, + SYST. EXT. IMP, = SYST. MUMENTA? WE HAVE CONSERVATION OF AMOUNTED MOMERITUM ABOUT THE 2 AXU. Iw= Iw,

$$\begin{split} \vec{I}_{j} &= \vec{I}_{j0kg} + \vec{I}_{DSE,600} = m_{y} \frac{1}{k} + \frac{1}{4} m_{y}^{x} \\ &= \frac{1.576}{9} \left(\frac{3}{12} f_{k}\right)^{2} + \frac{1}{4} \frac{2.576}{9} \left(\frac{4}{12} in\right)^{2} \\ &= \frac{0.09375}{5} + \frac{0.06944}{9} = 0.16319 \frac{1}{9} \end{split}$$

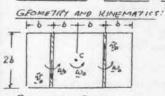
$$\begin{split} \bar{I}_2 &= \bar{I}_{YOLE} + \bar{I}_{DSL, 6 = 90^c} = m_Y \bar{A}_L^2 + \frac{1}{2} m_D r^2 \\ &= \frac{1.516}{3} \left(\frac{3}{12} f_L\right)^2 + \frac{1}{2} \frac{2.516}{3} \left(\frac{4}{12} m_L\right) = 0.23264 \frac{1}{9} \end{split}$$

EQ(1): 0.16319 (120 pm) = 0.23264 g 102

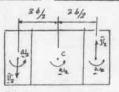
W= 84,2 ypm



GIVEN: PANELS AND PLATE ARE MADE OF SAME MATERIAL AND ARLE OF SAME THICKNESS. IN THE POSITION SHOWN ANGUAL VELUCITY = WE FIND: AFTER WIDE BREAKS ANGULAR VECOCITY WHEN PANGLS HAVE COME TO REST



PANCIS IN UP PUSITION 150 = b w.



AGAINST PLATE

PANELS IN DOWN POSITION できるめい。

LET P=MASS DENSITY, Z= THICKNESS PLATE: mplate = (t (2+ X+ b) = 8pt 8 I plate = /2 (8/26) (26)2+(46) = 100 ptb = 40 ptb

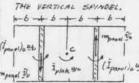
(CONTINUED)

17.80 continued

EACH FANEL: mpone = pt (b) 2h = 2pt 6 PAINEL IN UP POSTIONS (Iponel) = 1/2 (2pt 6")(26) = 8/2 pt 6" = 3/2 pt 6"

PRINTE IN DOWN POSITION (I ponel) = 1/2 (2pt 6) 62+(20) = 1/2 pt 64 = 6 pt 64

WE HAVE CONSERVATION OF ANGULAR MUMBERTON AGOUT



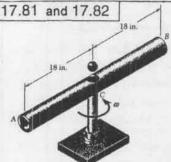
FINAL MOMENTA

INITIAL MONEHTA

+) MOMENTE REDUT C: $\bar{I}_{plain} \omega_o + 2 \left[(\bar{I}_{panel})_o \omega_o + m_{panel} \omega_o(b) \right] = \bar{I}_{plain} \omega_o + 2 \left[(\bar{I}_{panel})_o \omega_o + m_{panel} \omega_o + (\frac{3b}{2}) \right]$

40 pt 6 w + 2 [3 pt 6 w + 2 pt 6 (6 m) 6] $=\frac{40}{3}ptb^{4}w_{0}+2\left[\frac{5}{6}ptb^{4}w_{1}+2ptb^{2}(\frac{3}{2}bw_{0})(\frac{3}{2}b)\right]$

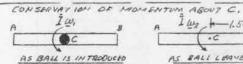
[40+4+4] pt64 Wo = [40+9 pt64 W, W= = 2 Wo 56 w = 24 w ; w = 56 2(24) wo



GIVEN: 4-16 TURE AB INITIALLY W= 8 rad/s BALLS INTRODUCED TO TUBE PROBLEM 17.82:

Fino: (0) W AS A 0.8-16 BALL LEAVES TURE (b) as AS A SECOND O.R-16 GALL LEAVES TUSS. PROBLEM 17.83:

FIND: a AS A SINGLE 1.6-16 BALL LEAVES TURE



1 w2 + 1.5A - moto AS BALL LEAVES TUBE

) MOMENTS ABOUT C: IN = IN + m. Va (1.5 ft) (1) I=1/2 (3A)=3 No= (1.592)w.

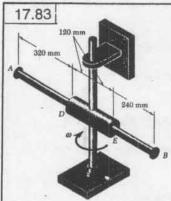
 $\begin{array}{ll} \frac{P_{ROB\,IFM}\left/7.82\right:\left(\alpha\right)\;F_{IRST}\;O.8\cdot Ib\;BALL,\;\;\left(\omega\right):=g\cdot valls\\ F_{G}(I):\;\;\;\frac{3}{3}\left(8\;md_{K}\right)=\frac{3}{9}\;\omega_{2}+\frac{0.8\;b}{9}\left(I.S\,\omega_{2}\right)\left(I.S\,f\right) \end{array}$

24 = (3 + 18) 02 Wo= 5 rads AS FIRST BALL LEAVES TUBE: W=5 red/s

(b) SECOND O.B-16 BALL W,= 5 radb Ea(1): 3/5 rad/s) = 3/4 wg + 0.8 % (1.5 a) x1.5(2) w2 = 3,125 radb 15 = (3+1.8) W2 AS SECONO BALL LEAVES TUBE: W=3,17510cl/s

PROBLEM 17.83: A 1.6-16 BALL IS INTRODUCED, W. = 8 radt

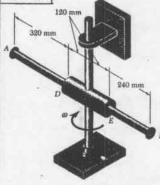
3 (Evad/s) = 3 00 + 1.616 (1.544)(1.5) EQU Wa = 3.636 radb 24 = (3+3.6) W2 AS 1.6-16 BALL LEAVES THE TUBE: W=3.64 rod/s



GIVEN: 3-Ag ROD AB FOR CYLINDER DE: 1 = 0.025 Ag.m. IN POSITION SHOWN: W= 40 rods Ano

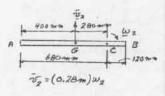
END B OF ROD IS MOUNTE TOWARD E AT 76 mm/s.

FIND: ANGULAR VELOCITY OF ASSEMBLY AS END B STRUKES CYLINDER AT E.



KINEMATICS AND GEOMETRY 360 mm 15,= (0.04 m) w, = (0.04 m) 40 rads

V = 1.6 mg

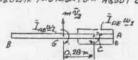


INITIAL POSITION

FINAL POSITION

WE HAVE CONSETTUATION OF AMBULAR MIGHTUM ABOUT C.



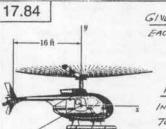


+) MOMENTS AROUT C: Ins 1/2 (3 Rayo, 8 m) = 0.16 - Rg . m = Ing w, + m v, (0.04m)+Inw, + Inw,+m v(0.28m)+I wa

(a16 kg·mi X 40 rads) + (3 kg X / 6 m/s X 0.04 m) + (0.025 kg·mi) (40 rad/s) = (0.16-fig-m) w,+(3-fig)(0.28/4)(0.28)+(0.025 kg-m) w,

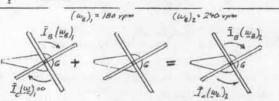
(6.4+0.192+1.00) = (0.16+0.2352+0.025) aug

7.592= 0.4202 wz; w= 18.068 rad/s; W= 18.07 rad/s



GIVEN: I CAB = I = 650 Ib. ft. 5 EACH BLADE WEIGHS 55 16 INITIAL ANGULAR VELOCITY OF CAR = 2ERO .

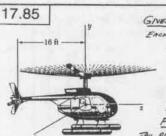
FINDS WE AS WRINDES IS INCREASED FROM 1807pm 70 240 rm



SYST. MOMENTA, + SYST. EXT. IMP, -2 = SYST. MOMENTA In 4 3 38.2 (1452) = 446. + 16. (2.5° 8 1 - 38.8 (Wa) = (No 7 = 8 | Wa) = (Wa) + 240 ym (NOTE: W 15 OF 81.40 ym 70 CAR +) MOMENTS ABOUT GI IB(WE), + 0 = IB(WE) 2+ Ic(WE)2 (446.4 10.77 5) (180mm) = (446.416.66.5) (0+)+240mm] + (650 10.17.5) (04)3

(wc) = 26784 1096.4

(W CAB) = 24.4 rpm



GIVEN I = 1 = 650 16-ft-5" EACH BLADE WEIDE 55 16 WCAB = 1250 16

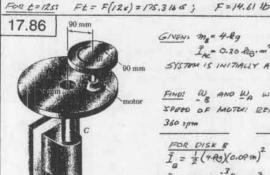
TAIL PROPELLER PREVENTS ROTATION OF CAB AS NO OF BLACKS IS INCREASED PROM 180 rpm TO 240 rpm IN 125. FIND: FORCE EXERTED ST TAIL PROPELLER AND FINAL T.

 $\bar{I}_{n}(\omega_{n})$

SYST. MOMENTA, + SYST. EXT. IMP, = = SYST. MOMENTA? +) MOMENTS ABOUT 6: $\tilde{I}_{a}(\omega_{8})_{i}$ + $F \pm (164) = \tilde{I}(\omega_{8})_{2}$ (1)

+ COMPONENTS: O+ Ft = m V2 (2) I=4 1 5510 (14/2) = 446.4 16.52.52 m=m2+m8=1/322 [125010+4(5510)] = 45.65 10.5/A (wo)= 180 rpm 21 = 18.85 rad/s 3 (wg)= 240 rpm 21 = 25.13 rad/s EGILIST (446.4 16-12-58 (18.85 mark) + (FEXICRE) = (446.4 16-17-52 × 25,13 vadk)

Ft= 175.3 16.5 EG(2): 175.316.5 = (45.65 16.57A) N; T=3.84 H/5



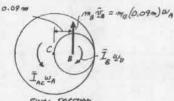
GIVEN: M= 4-Rg In = 0.20 Rg.m2 SYSTEM IS INITIALLY AT REST

FIND: WE AND WA WHEN SPEED OF MOTOR REACHE 360 rpm

I = = (4.89)(0.09m) I = 16,2 × 10 Rg·m

WE HAVE CONSERVATION OF ANGULAR MOMENTUM ABOUT SHAFT C





INITIAL POSITION

+) MOMENTS ABOUT C: Inw + mg vg (0.09 -) - Inwg (0.20 29 m2) w, + (4 29) (0.09 m, (0.09) - (16.2×10 25 m2) w 0,2324 WA - 0,0162 Wg = 0 WR = 14,346 WA (1)

WMOTOR = WATWA

360 vpm = WA + 14.346 WA

WA = 23,46 ymm

Wa= 23.5 rpm WB = 337 10%

EGG) WB = 14.346(23.46) = 336.55 774



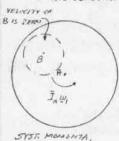
GIVEN: FOR 200-MM PRADIUS PLATFORM - RIM UNIT:

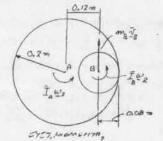
ma= 5 Ag, R = 175 mm

INITIAL ANGULAR VELUCITY I WAS SOYD DISK! MR=3Ag, Te= Forming DISK PLACED, WITH NO VELOCITY, ON PLATEORM

FIND, FINAL ANGULAN VELOCITY

INE HAVE CONSERVATION OF AUGULAN MONIGATUM AGOUT SHAFF





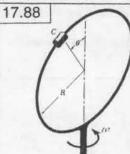
+) MOMENTS AGOUT A! I AW = I AW + I AW + MATE (0.12 m) (1) In = 111 A2 = (5 + 9)(0,175 m) = 0.153125 Ag. m In = 2 mare = = 1 (3 kg (0.08 m) = 9.6 x 10 kg mi2 2 = 10.12m as

Edin (0.153125 Ag = 2) 0, = (0.153125 Agai) 14, 1 (9.6×10 3 kg/mi) 14, + (38g)(0.12 m) wz

0.153125 01, = 0,20593 412

W= 0.7436 W= 0.7436 (50 1pm)

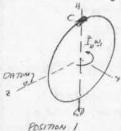
10=37.2 rpm

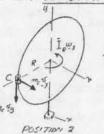


GIVENI 2-Ry COLLAR C PING: m = 3-89 R = 250 mm WHEN 0=0, W= 35 rod/s AND 25 = 0

FIND: (a) W WHEN 8=900 (b) VELOCITY OF COLLAN QULATINE TO RINK WHEN 6 = 90°

INE HAVE CONSERVATION OF ANSULAR MODALITURA PROUT VERTICAL Y AXIS AND CONSERVATION OF EMERSY





CONSMINATION OF ANGUAR INCOMENITURA

MOMENTS AROUT Y AxIS: I w = I w = + m. Vic ZmaRw, = +mRwy + mRwy m 12 w, = (mx+2mc) 12 w,

W2= mR W, (1)

(CONTINUED)

17.88 continued

T = \$ I w = = 2 (2 ma R) w = + ma R w, V, = We R = meg R $T_2 = \frac{1}{2} \hat{I}_R w_2^2 + \frac{1}{2} m_e \left(\hat{r}_2^2 + \hat{r}_3^2 \right) = \frac{1}{4} m_R R^2 w_1^2 + \frac{1}{2} m_e R^2 w_2^2 + \frac{1}{2} m_e R^2 w_3^2 + \frac{1}{2}$

CONSTIVATION OF ENERGY: T, + V, = T2 + Y2 = mn 12 w, + mg R = (+mp + 1 me) 12 w, + 1 mm Ny (e)

DATA: me=2kg, me= = kg R=0,25m, W,=35 valls EQ.(1) W2= 349+2(240) (35 red/s)

EG(2): 1(3 kg)(0.25m)(35 rods) + (2 kg)(9.81m/s)(0.25m) = [1/3 Rg) + 1/2 kg) (0.25m)(15 m/b) + 1/2 Re) vg

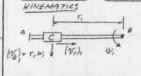
57.422 + 4.905 = 24.609+5 of Ny = 37.716;

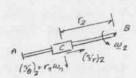


GIVEN: IN POSITION SHOWN W, = 1.5 rad/s, (5,) = 1.5 m/s

me=8-Rg, FOR ROD AND SARING I = 1,2 kg.m.

FINO: (a) MINIMUM DISTANCE RETWEEN CAIDB, (6) COREST-PONDING ANGULAR VELLKITY





KINETIES: SINCE MOMENTS OF ALL FORCES ALOUT & ARE ZERO, WE HAVE: (Hg),=(Hg)2: IBW, + mc (NG), = IDW2+mc(NG)2+2 (In+mer,) w, = (In+mer2) w2 (1)

[1.2 Ag. mi+ (8 Ra)(0.6m)2 (1.5 on/s) = 1.2 Ag. mi+ (8 Ra) 12 Wg CONSERVATION OF EMERGY SINCE VIEVE, WE HAVE TI-TA

TI= 18 W1 + 1 me (V0) + 1 me (SV) = \$ (1.2 Rg. mi)(1.5 rod/s) + \$ (8 ka/0.6 mi)(1.5 rad/s) + \$ (6 ka)(1.5 m/s) 7=13.591

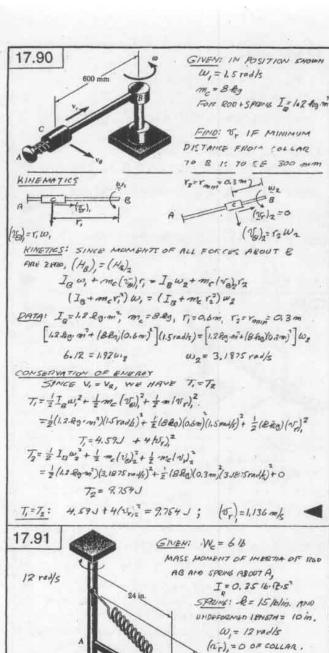
To = 1 Ip w2 + 1 mc (20) 2 + 2 mc/10) = 1 (1.2 fig.m) w + 1 (8 fig) (w + 1 (8 fig) (7 r) = To=0,6 w2 + 4 10 w2 + 4 (21)2

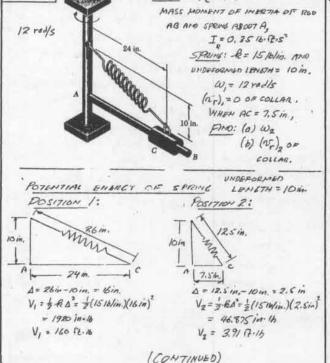
13.69= (0.6+472) 12 +4/27) (3) T,= T2: (4)

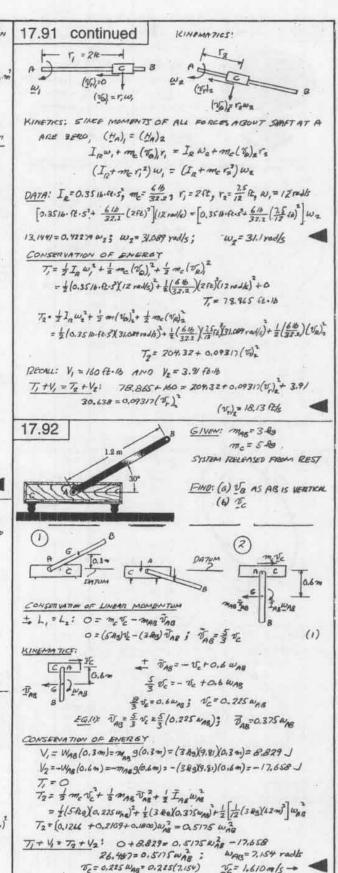
6.12 = (12+8 r1) Wz ; Wz = 6.12 = 3.06 EQ.(1): EG(s) /3.59 = $(0.6+4r_2^4)\left[\frac{3.06}{0.6+4r_2^4}\right]^2 + 4(v_1)_2^4$

FOR THUMAN WE HAVE IN = 0 $|3.59 = \frac{(3.06)^2}{0.6 + 4r_3^2}$; 8.154 + 54.36 7 = 9.364 Y= 22.25x 10-3 mi 12= 0.1492 m r= 149,2mm

3.06 3,06 FG(9): $W_2 = \frac{3.06}{0.6 + 4\Gamma_2} = \frac{3.06}{0.6 + 4(22.25 \times 10^{-3})} = 4.44) \text{ val} f$ W= 4.44 rad/s

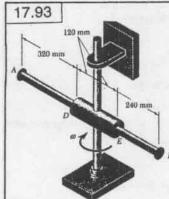






UB = Vc + (AB) WAR = -1.610 + (1.20) 7,154 rails)

UR= 6.975 m/s -



GIVEN: 3-kg POD AB

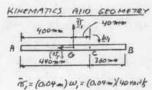
FOR CYLINDER DE: I=0.025.Rg. of

IN POSITION SHOWAR

GU = 40 valle AND

W=40 rad/s AND END B OF ROD IS MOVING TOWARD E AT 75 ma/s

FIND VELOCITY OF AB RELATIVE TO DE AS END B STAKES END E OF THE CYLINDER



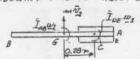
7 = 1.6 mg

INITIAL POSITION

FINAL POSITION

WE HAVE CONSERVATION OF ANGULAR MIGHTUM ABOUT C.





+) MOMENT AROUT C: Ins 1/2 (3 Re)(0.8 m) = 0.16-Rg · m 2

Ins u, + m v, (0.04m) + Ins u, = Ins u, + m v(0.28m) + Ins u,

(0.16 Rg · m² (4016 16) + (2 Rg)(1.6 m le)(0.04 m) + (0.025 Rg · m²)(40 rod)(1)

 $= (0.16 + ig \cdot m^2) w_2 + (3 + ig \times 0.28 + ig \times 0.28) + (0.025 + ig \cdot m^2) w_3 + (3 + ig \times 0.28 + ig \times 0.28) + (0.025 + ig \cdot m^2) w_3$

(6.4+0.192+1.00) = (0.16+0.7352+0.025) W2 7,592=0.4902 W2; W2=18.068 rad/s; W=18.07 rad/s

CONSERVATION OF ENERGY $(V_r) = 0.075 \text{ m/s}$ $V_i = V_2 = 0$ $T_i = \frac{1}{2} \vec{L}_{DE} \alpha_i^2 + \frac{1}{2} \vec{L}_{AB} \omega_i^2 + \frac{1}{2} m_{AB} \vec{v}_i^2 + \frac{1}{2} m_{AB} (\vec{v}_r)_i^2$ $= \frac{1}{2} (0.021 \cdot B_3 \cdot m^2) (40 \text{ rad/s}) + \frac{1}{2} (0.16 \cdot B_3 \cdot m^2) (40 \text{ rad/s})^2$ $+ \frac{1}{2} (3 \cdot B_2 X_{10} b m/s)^2 + \frac{1}{2} (3 \cdot B_2 X_{10} (0.075 m/s)^2$

 $T_1 = 20J + 128J + 3.84J + 6.006J = 151.85J$ $\tilde{V}_2 = (0.28m)w_2 + (0.28m)(18.068 \text{ rad/s}) = 5.059 \text{ m/s}$ $T_2 = \frac{1}{2}\tilde{I}_{DE}w_1^2 + \frac{1}{2}\tilde{I}_{AB}w_2^2 + \frac{1}{2}m_{AB}\tilde{V}_2^2 + \frac{1}{2}m_{AB}(1v)_2^2$ $= \frac{1}{2}(0.025 \log m)(18.068 \text{ rad/s})^2 + \frac{1}{2}(6.16 \log m)(18.068 \text{ rad/s})^2$ $+ \frac{1}{3}(3 \log 3.5059 \text{ m/s})^2 + \frac{1}{3}(3 \log 3)(1v)_1^2$

 $T_2 = 4.081 J + 26.116 J + 38.391 J + 1.5 (N_y)_2^3$ $T_2 = 68.587 J + 1.5 (N_y)_2^3$

 $T_1+V_1=T_2+V_2$: $|51.85J+0=60.582J+1.5(v_1)_1^2$ $\theta 3.263=1.5(v_1)_2^2$ $|\alpha_{r_1}^*|=7.45 \text{ m/s}$ 17.94 18 in. 8 in. 8

GIVEN: 4-16 TUBE AB
INITIALLY W. = Bracks

AN AS-16 BALL IS
INTRODUCED INTO TUBE
AND LEAVE TUBE AT B.
A SECOND DIS-16 BALL
IS THEN PUT INTO TUBE
FIND! VELOCITY OF
ENERGALL CECATINE TO
TUBE AS IT LEAVES THE

) MUMENTS ABOUT C: $\bar{I}_{N_1} = \bar{I}_{N_2} + m_1 V_6 (1.5 \text{ ft})$ (1) $V_6 = (1.5 \text{ ft}) w_1$ $\bar{I} = \frac{1}{12} \frac{416}{3} (3 \text{ ft})^2 = \frac{3}{3}$

FIRST 0.8.16 BALL, $\omega_1 = B \text{ rad/s}$ FG(1): $\frac{3}{3}(B \text{ rad/s}) = \frac{3}{3} \omega_2 + \frac{0.86}{9}(1.5\omega_2)(1.564)$ $24 = (3 + 1.8)\omega_2$ $\omega_2 = 5 \text{ rad/s}$ AS FIRST BALL LEAVES TURE: $\omega = 5 \text{ rad/s}$ $\frac{SECGNO 0.8-16 \text{ BALL}}{3}(5 \text{ rad/s}) = \frac{3}{3} \omega_2 + \frac{0.86}{9}(1.5\omega_2)(1.564)$

 $Eo(i): \frac{2}{3}(5 \operatorname{rad} s) = \frac{1}{3}(\omega_2 + \frac{1}{3})(15 \omega_2)(15 \omega_2)(15 \omega_3)$ $15 = (3+1.8) \omega_2 \qquad \omega_2 = 3.125 \operatorname{rad} s$ As SECOND BALL LEAVES TUBE: $\omega = 3.125 \operatorname{rad} s$

CONSTRUATION OF ENERGY

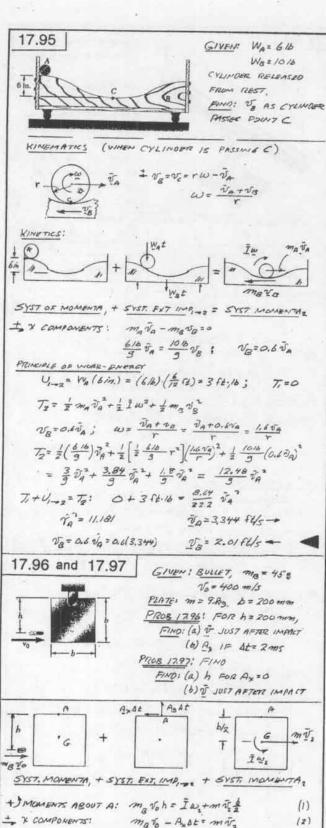
 $\begin{aligned} & \underbrace{F/RS7 \ \&n\mu}_{V_1=0} : \ \omega_s = 8 \text{ rod} k , \ \omega_2 = 5 \text{ rod} f \\ & V_1=0, \ T_1 = \frac{1}{2} \overline{I} \, \omega_s^2 = \frac{1}{2} \left(\frac{2}{9}\right) \left(8 \text{ rod}/s\right)^2 = \frac{96}{9} \\ & V_2=0, \ T_2 = \frac{1}{2} \overline{I} \, \omega_s^2 + \frac{1}{2} \, m \, V_0^2 + \frac{1}{2} \, m \, V_1^2 \\ & = \frac{1}{2} \left(\frac{3}{9}\right) \left(5 \text{ rod} k\right)^2 + \frac{1}{2} \left(\frac{0.8 \, \text{b}}{9}\right) \left(\frac{3}{8}\right)^2 \left(5 \text{ rod}/s\right)^2 + \frac{1}{2} \left(\frac{0.8 \, \text{b}}{9}\right) \left(\frac{3}{8}\right)^2 \\ & T_2 = \frac{32}{9} \frac{5}{9} + \frac{27.5}{9} + \frac{9}{9} \cdot V_1^2 \end{aligned}$

 $T_r + V_r = T_r + V_s$: $\frac{96}{3} + 0 = \frac{1}{9} \left(325 + 22.5 \right) + \frac{0.4}{5} \nabla_r^2 + 0$ $V_r = 9.49 \text{ fels}$

SECOND BALL W, = Stadls, Wa = 3.125 rails

 $V_{1}=0, \ T_{1}=\frac{1}{2}\tilde{1}w_{1}^{2}=\frac{1}{2}\left(\frac{3}{9}\right)(5vods)^{\frac{1}{2}}=\frac{37.5}{9}$ $V_{2}=0, \ T_{2}=\frac{1}{2}\tilde{1}w_{1}+\frac{1}{2}m_{1}\eta_{0}^{2}+\frac{1}{2}m_{1}\eta_{1}^{2}$ $=\frac{1}{2}\left(\frac{3}{9}\right)(3.125mds)^{\frac{1}{2}}+\frac{1}{2}\left(\frac{0.8}{9}V_{1}.5\right)(3.125mds)^{\frac{1}{2}}+\frac{1}{2}\left(\frac{0.8}{9}V_{1}V_{1}^{2}\right)^{\frac{1}{2}}$ $=\frac{14.687}{9}+\frac{8.787}{9}+\frac{0.47}{9}V_{1}^{2}$

 $T_r + V_r = T_2 + V_2$: $\frac{37.5}{9} = \frac{14.487}{9} + \frac{8.789}{9} + \frac{0.4}{9} v_r^2$ $v_r^2 = 35.156$ $v_r = 5.93 \text{ ft/s}$

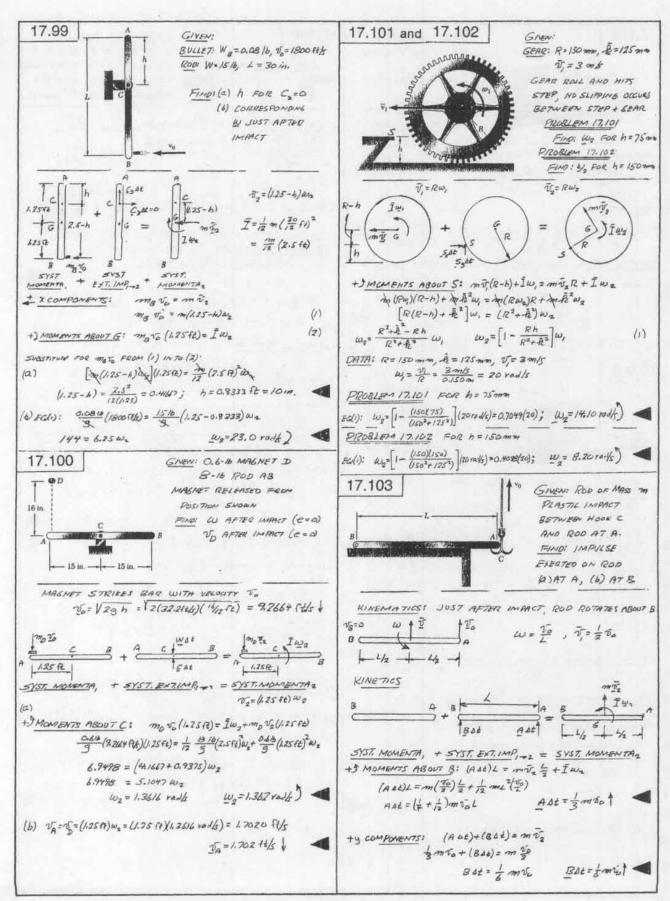


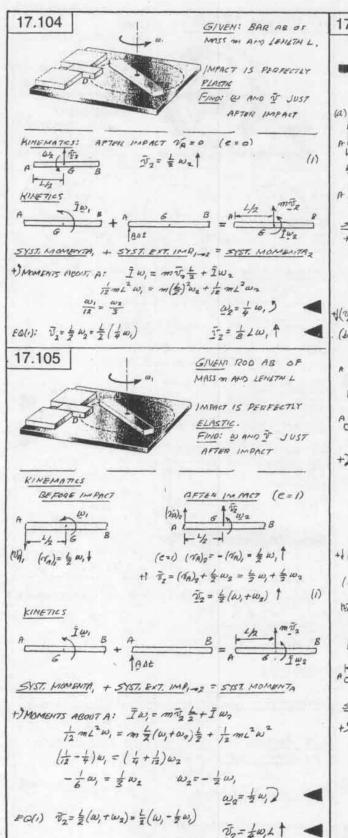
RUTATION ABOUT A: V= 0 w I= 1 mb

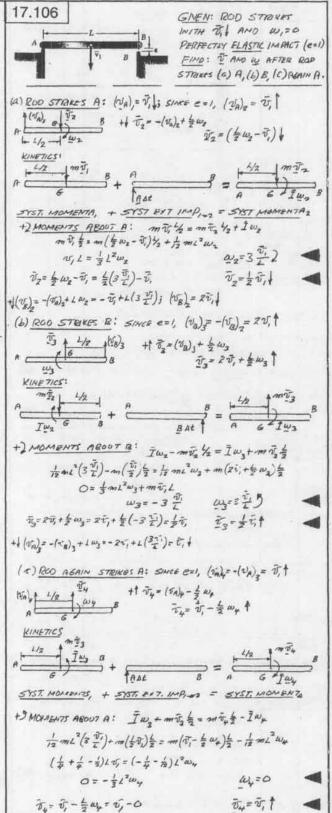
(CONTINUED)

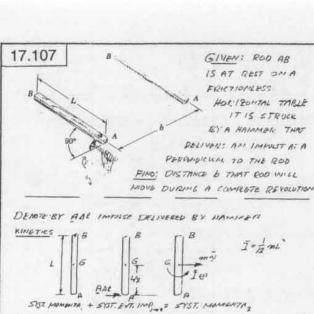
17.96 and 17.97 continued SUBSTITUTE FROM (3) INTO (1) mg voh = + mb w2 + m (+ 6 w2) 1/2 b ma Toh = 5 m 62 W2 EQ(2): AxAt=mav-mv. DATA: mg=0.045 kg, %=400 m/s, b=0.2m, m= 9.kg, st=0.0025 PROBLEM 17.96 FOR h= 0.2m EG(+) (0.045 Rg)(400 mg/s (0.2 m) = 5 (9 kg (0.20m) Wg W= 24 rad/s (a) $\bar{V}_{2} = \frac{b}{2} w_{2} = \frac{0.2 \text{ m}}{2} (2 + rad/s)$ T= 2.4m/s -(6) EO(5) Ax(0.0025) = (0.045 fig) 400 m/s) - (9 fig)(2.4 m/s) 0.001A=18-21.8 A=-1.8.RN PRUBLEM 17.97 FOR Ax=0 FO(5) mate = mt2; mto=m(2 w2); w= 2 ma 20 SUBJITUTE INTO (4) mg V h = 5 m b (2 mg Vo) h= 5 b= 5 (200 mm) h = 166.7 on m (b) mg to 2 mit; (0.045 kg) (+00 m/s) = (9.89) t; \$ = 2m/5 17.98 GIVEN: BULLET, WB= 0.0816, Vo=1800 9/5 1200: W= 1516, L = 30 in. h= 12 in. FINO: (a) W JUST AFTER MAPACT (6) C FOR St= 0,0015 I= 12 mL2 Soat = 1/2 (1510) (30 4) 1=0.24262 16.92.52 == (0.25ft)wz SYSTI + SYST. = SYSTI MOMENTA, + EXTENDED = MOMENTA, +) MONERIS ABOUTC: mg 16(214) = I w2 + m 7, (0:25ft) (3.0816) (1800 (4/5)/1556)= (0.24212 10.12.5) W2+(1516)(0.25) W2 6.708 = (0,24262+0.02911) Wa (0) Wa = 24.686 radf W== 247 radk) (b) + & COMPONENTS: CxAt-ma To = -mitz Cast = MA To - m (0.25 ft) Na = (30.8 b) (1800 Ab) - (15/b) (0.25 P) (24,686 rath) Cx St= 1.597 16:5 At = 0.0015 C2 (0.0015) = 1.597 16.5 Cy= 159716-

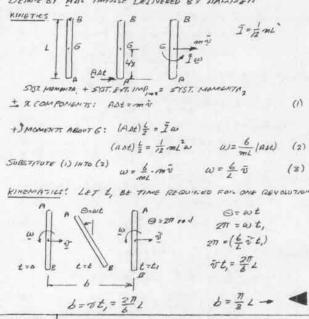
(3)









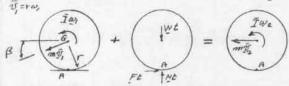




GNENI SPHERE ROLLS AND HITS HORIZONTAL SUNFACE. AFTER SLIPPING IT STARTS ROLLING AGAIN

FIND: V2 AND W2 AS IT ROLLS TO THE LEFT

> POSITION 2, SPHERE HAS RESUNIED ROLLING, To = YW.



SYST. MOMENTA, + SYST. ENT. IMP, = SYST. MOMENTA +) MOMENTS ABOUT A: I w.+ (mov, rosp) r= I w.+ movy 3-mr2w,+(mrw, cosp)r=3-mr2w,+mr2wa)r (=+cosa)w,= = = w2

un= = = 1/2+5cosB) a, 5

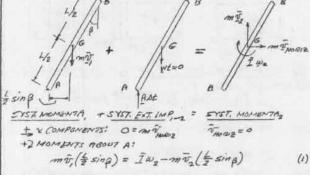


17.109 and 17.110

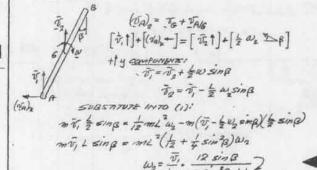
GINEN: ROD AR STRIKES FRICTIONLESS SURFACE WITH THE VELOCITY SHOWN.

DERIVE AN EXPRESSION FOR W IMMEDIATELY AFTER

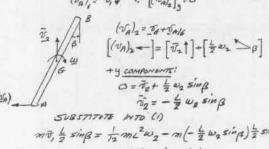
PROBLEM 17.109 ASSUME PERFECTLY ELASTIC IMPACT, (e = 1) PROBLEM 17,110 ASSUMIE PERFECTLY PLASTIC IMMET (@=0)



PROBLEM 12109: FLASTIC IMPACT AT A (e=1) : ((vA)2 = v,4 (VA) = V, 4 KINEMATICS;

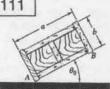


PIZOBLEM 17.110: PLASTIC IMPACT (e=0) (VA) = V, 1 : (VA) =0



mit, = sing = 12 ml2w2 - m(- = wasing) = sing mit = sing = m12 (12 + 4 sin 2) wa





GIVEN: UNIFORM CRATE

IS RELEASED FROM REGT.

IMPACT AT B IS PERFECTLY
PLASTIC.

FIND I SMALLEST VALUE OF & B FOR WHICH COLDNER A REMAINS IN CONTRCT WITH FLOOR

WE CONSIDER THE LIMITING CASE WHEN THE CRATE IS JUST READY TO ROTATE ABOUT B. AT THAT INSTANT THE VELOCITES MUST BE ZERO AND THE REACTION AT CORNER A MUST BE ZERO.



$$\begin{array}{c|c} & C & \downarrow & \downarrow \\ & -G & \downarrow & \downarrow \\ & & A & \downarrow \\ & & & A & \downarrow \\ & & & & A & \downarrow \\ & & & & & A & \downarrow \\ & & & & & & A & \downarrow \\ & & & & & & & A & \downarrow \\ & & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & & \\ & & \\ & & & \\ & & & \\ & & \\ & & & \\ & & & \\ & & \\ & & & \\ & & \\ & & & \\ & \\ & \\ & & \\ & \\ & \\ & & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ & \\ &$$

SYST MOMENTA, + SYST, EXT. IMP₁₋₂ = SYST, MOMENTA₂ +2 MOMENTS ABOUT B \bar{I} ω , + $(mr\bar{v})$, $\frac{b}{2}$ - $(m\bar{v})$, $\frac{a}{2}$ +0 =0 (1)

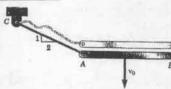
NOTE: $\sin \phi = \frac{b}{\sqrt{a^2 + b^2}}$, $\cos \phi = \frac{a}{\sqrt{a^2 + b^2}}$

V = (A6) w = 1/2 / a2+62 w,

THUS: $(m\bar{v}_i)_{\chi} = (m\bar{v}_i) \sin \phi = \frac{m}{2} |a^2 + b^2 \omega_i \frac{b}{\sqrt{a^2 + b^2}} = \frac{1}{2} mb\omega_i$ ALSO, $(m\bar{v}_i)_{\chi} = (m\bar{v}_i) \cos \phi = \frac{1}{2} m\alpha\omega_i$ $\bar{T} = \frac{1}{2} m(a^2 + b^2)$

 $\frac{20(1)!}{12}m(a^{2}+b^{2})\omega_{1}+\frac{1}{2}(mb\omega_{1})\frac{b}{2}-\frac{1}{2}(ma\omega_{1})\frac{a}{2}=0$ $\frac{1}{3}mb^{2}\omega_{1}-\frac{1}{6}ma^{2}\omega_{1}=0$ $\frac{b^{2}}{2}=2$ $\frac{b}{2}=\frac{1}{\sqrt{2}}$

17.112



GIVEN: ROD OF LENGTH L.
"ASSUMING PERFECTLY
PLASTIC IMPACT,

FIND: W AND & JUST AFTER CORD BECOMES TAUTI

KINEWATICS (JUST AFTER IMPRIET) LET $\Theta = + \cos^{-1} \frac{1}{2}$ $|V_A|_2 = V_S + V_{AB}$ $|V_A|_2 = V_S + V_S + V_S$ $|V_A|_2 = V_S + V_S + V_S$ $|V_A|_2 = V_S + V_S + V_S$ $|V_A|_2 = V_S + V_S + V_S$

17.112 continued

 $A = \begin{bmatrix} 1/2 & 1/2$

SYST, MOMENTA, + SYST. EXT. IMP, -2 = SYST. MOMENTA?
+2 MOMENTS AROUT A

mode = Iw+mode =

 $m\vec{v}_0 = \frac{1}{2} m^2 \omega + m\vec{v}_0 = \frac{1}{2}$ $\vec{v}_0 = \frac{1}{2} L \omega + \vec{v}_0 \qquad (2)$

+ / components:

 $v_0 = v_x \tan \theta + v_y$ $(1) - (2) \quad v_0 = (\bar{v}_y - \frac{1}{2}\omega) \tan^2 \theta + \bar{v}_y$ (3)

 $\tilde{N}_{0} = \tilde{\pi}_{y} \left(1 + \tan^{2} \theta \right) - \frac{1}{2} \omega \tan^{2} \theta$ $\omega = \frac{2}{L} \left(\tilde{\pi}_{y} - \frac{1 + \tan^{2} \theta}{\tan^{2} \theta} - \frac{\tilde{\pi}_{0}}{\tan^{2} \theta} \right) \tag{4}$

 $(\psi) \to (2) \quad \tilde{V}_0 - \frac{L}{6} \cdot \frac{Z}{L} (\tilde{V}_y \frac{1 + L s_n^2 G}{L s_n^2 G} - \frac{\tilde{V}_0}{L s_n^2 G}) + \tilde{V}_y$ $\tilde{V}_U = V_y (1 + \frac{1}{3} \frac{1 + L s_n^2 G}{L s_n^2 G}) - \frac{\tilde{V}_0}{3} \frac{\tilde{V}_0}{L s_n^2 G}$ $3 V_0 L s_n^2 G = V_y (1 + 4 L s_n^2 G) - V_0$ $V_y = \frac{1 + 3 L s_n^2 G}{1 + 4 L s_n^2 G} = V_0$ (6)

 $(5) \rightarrow (2) \quad \tilde{V}_0 = \frac{L}{6} \cdot \omega + \frac{1+3 \tan^3 6}{1+4 \tan^3 6} \tilde{V}_0$ $\omega = \frac{6}{L} \left[1 - \frac{1+3 \tan^3 6}{1+4 \tan^3 6} \right] \tilde{V}_0 = \frac{6}{L} \left[\frac{1+4 \tan^3 6}{1+4 \tan^3 6} - \frac{1-3 \tan^3 6}{1} \right]$ $\omega = \frac{6}{L} \frac{\tan^3 6}{1+4 \tan^3 6} \tilde{V}_0 \qquad (6)$

(6) No(5) \rightarrow (1) $\overline{\mathcal{I}}_{X} = (\overline{\mathcal{I}}_{Y} - \frac{1}{2}\omega) \tan 6$ $\overline{\mathcal{I}}_{X} = \left[\frac{1+3\tan^{2}6}{1+4\tan^{2}6}\,\overline{\mathcal{I}}_{0} - \frac{1}{2}, \frac{6}{L}, \frac{\tan^{2}6}{1+4\tan^{2}6}\,\overline{\mathcal{I}}_{0}\right] \tan 6$ $\overline{\mathcal{I}}_{X} = \frac{\tan 6}{1+4\tan^{2}6}\,\overline{\mathcal{I}}_{0}$ (7)

DATA: \$ 2 A 6 TO B

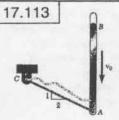
tan 8 = 1/2

Eq.(6): $\omega = \frac{6}{L} \frac{\tan^2 6}{1 + 4 \tan^2 6} \, \tilde{v}_0 = \frac{6}{L} \frac{\cos^2 2}{1 + 4 (\cos^2 2)} \, \tilde{v}_0 = \frac{4.5}{2} \, \frac{v_0}{c}$ $\omega = \frac{3}{4} \, \frac{v_0}{L} \, 2$

EG(2): $\vec{V}_{X} = \frac{\tan 6}{1 + 4 \tan 6} \vec{V}_{0} = \frac{0.5}{1 + 4 (0.5)^{2}} \vec{V}_{0} = \frac{0.5}{2} \vec{V}_{0}$

 $\frac{E_{G}(s)}{\sqrt{y}} = \frac{1+3\tan^{2}\theta}{1+4\tan^{2}\theta} v_{o} = \frac{1+3/6.0^{2}}{1+4(0.0)^{2}} v_{o} = \frac{1.15}{2} v_{o}$ $\sqrt{y} = \frac{7}{8} v_{o} \downarrow$

CHECK: EQ() $\vec{v}_{x} = (\vec{v}_{y} - \frac{1}{2}\omega) \tan 6$ $= (\frac{2}{3}\vec{v}_{w} - \frac{1}{2} \cdot \frac{3}{4}\frac{\vec{v}_{w}}{2})(0, s)$ $= (\frac{2}{3}\vec{v}_{a}) \cdot s$ $\vec{v}_{y} = \frac{1}{4}\vec{v}_{a}$



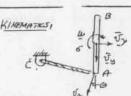
GIVEN; ROD AB OF LENGTH L.

ASSUMING PETERECTLY

PLASTIC IMPACT

FIND: W AND I IMMEDIATELY

AFTER COID WECOMES TAUT.



 $\vec{v} = \vec{v}_A / \phi + \frac{1}{2} \omega - \omega$ $\vec{v}_{\pm} = (\frac{1}{2} \omega - \vec{v}_A \sin \phi) \rightarrow \omega$ $\vec{v}_{\pm} = \vec{v}_A \cos \phi$

KINETICS !

SYST, MOMENTA, +SYST, EXT, IMP, = SYST, MACHENTA

+) MOMENTS ABOUT A: $O = \overline{L}\omega + m\overline{N}_2 \frac{L}{2}$ $O = \frac{1}{2}mL^2\omega + m(\frac{L}{2}\omega - N_0 N_0 \frac{L}{2} \sin 6) \frac{L}{2}$ $O = \frac{1}{3}N_0 L^2\omega - N_0 N_0 \frac{L}{2} \sin 6$ $\omega = \frac{3}{2} \frac{N_0}{2} \sin 6$ (1)

+ $\int components$: $\int v_0 cos\theta = \int v_0 cos\theta$

(2) \rightarrow (1) $\omega = \frac{3}{2L} \left(\sqrt{\cos s} + \frac{L}{2} \omega \sin \theta \right) \sin \theta$ $\omega = \frac{3\sqrt{6}}{2L} \cos \theta \sin \theta + \frac{3}{4} \omega \sin^2 \theta$ $\omega = \frac{3\sqrt{6}}{2L} \frac{\cos \theta \sin \theta}{1 - \frac{3}{4} \sin^2 \theta}$

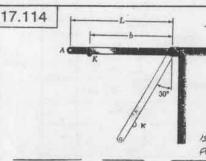
For $a = ton^{-1}o.5$, $cas a = \frac{2}{\sqrt{5}}$ And $sin6 = \frac{1}{\sqrt{5}}$ $\omega = \frac{3}{2} \frac{\tau_0}{L} \cdot \frac{(\frac{2}{\sqrt{6}})^{1/\sqrt{6}}}{(-\frac{3}{4}(\frac{1}{\sqrt{6}})^{2})^{2}} = \frac{3}{5} \cdot \frac{1}{0.85} \frac{v_0}{L}$ $\omega = 0.7059 \frac{v_0}{L}$ $\omega = 0.705 \frac{\tau_0}{L}$ $\omega = 0.705 \frac{\tau_0}{L}$

 $EQ(2) \cdot \sqrt{1}_{A} = \sqrt[3]{6} \cos 6 + \frac{1}{2} \cos 8 \sin 6$ $= \sqrt[3]{1}_{0} + \frac{1}{2} (0.7059 \sqrt[3]{6}) \frac{1}{\sqrt{2}}$ $= (0.8944 + 0.1578) \sqrt[3]{6}$

 $\vec{v}_A = 1.0522 \vec{v}_0$ $\vec{v}_{12} = \frac{1}{2} \omega - \vec{v}_A \sin e = \frac{1}{2} (0.2059 \frac{\vec{v}_0}{2}) - (1.0572 \vec{v}_0) \frac{1}{15}$ $= (0.35295 - 0.47059) \vec{v}_0 = -0.11764 \vec{v}_0$

Ty=0.1176 Vo ← ◀
Ty=0.1176 Vo ← ◀

Ty= V4 cos6= (1.0522 Vo) = 0.9411 Vo Ty= 0.941 Vo ↓



GNEY: ROD IS

RELEASED FROM

POSITION SHOWN

AND REBOUNDS TO

30° WITH THE VEETICAL.

FIND: (a) COST. OF

RESTITUTION, (b)

SHOW THAT REBOUND

IS INDEPENDENT OF

POSITION OF KNOB K

AFTOR IMPACT; $\tilde{V}_3 = \frac{L}{2}w_3$ $\tilde{V}_$

Ti+V= T2+V2: 0= fm2 =- m3 =; u=3 =.

 $V_{3} = -W_{2}/2 = -mg^{\frac{1}{2}}$ $V_{4} = -W_{2}/2 = 30^{3}$ $V_{3} = \frac{1}{2} \tilde{L} u_{3}^{2} + \frac{1}{2} m \tilde{v}_{3}^{2} = \frac{1}{2} (\frac{1}{2} m \tilde{L}^{2}) u_{3}^{2} + \frac{1}{2} m (\frac{1}{2} u_{3})^{2} = \frac{1}{6} m \tilde{L}^{2} u_{3}^{2}$ $T_{4} = 0$ $T_{4} = 0$

 $T_3 + V_2 = T_4 + V_4$: $\frac{1}{6} mL^2 \omega_3^2 - m_0 \frac{L}{2} = 0 - m_0 \frac{L}{2} \cos 36^4$ $\omega_3^2 = 3\frac{9}{2} \left(1 - \cos 36^4\right).$

MARACT FOR SELECTION OF SELECTI

 $(v_{\kappa})_2 = rw_2 = r\sqrt{3}\frac{5}{L}$ $(v_{\kappa})_2 = rw_2 = r\sqrt{3}\frac{3}{L}(1-\cos 30^{\circ})$

 $C = \frac{(V_K)_3}{(V_K)_4} = \frac{r\sqrt{3\frac{9}{2}(1 - \cos 3\theta^4)}}{\sqrt{\sqrt{3\frac{9}{2}}}}$

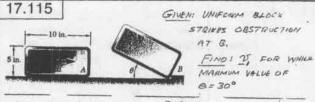
e=V(1-cas36*

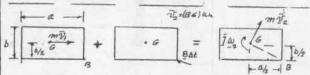
e= V.(1-cos 300) = V1-0.86603

e=0.366

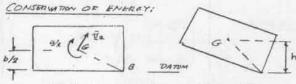
WIE NOTE THAT RESULT IS INDEPENDENT OF THE POSITION OF THE KNOB.

(2)





SYST. NOMENTA, $+ \leq YST, EYT, IMP_{1}=2^{-1} \leq YST, MOMENTA_{2}$ +) MOMENTS ALCOUT e! $m \bar{v}_{1} = \bar{I} \omega_{0} + m \bar{v}_{1} = (86)$ $m \bar{v}_{1} = \bar{I} \omega_{0} + m \bar{v}_{1} = (86)^{2} \omega_{0}$ $36^{\frac{1}{2}} = (4/2)^{\frac{1}{2}} + (4/2)^{\frac{1}{2}} = \frac{1}{4}(a^{2} + b^{2})$ $\bar{I} = \frac{1}{12} m(a^{2} + b^{2})$ $d_{1} = \frac{1}{12} m(a^{2} + b^{2}) \omega_{2} + d_{1} = \frac{1}{4}(a^{2} + b^{2}) h_{2}$ $\bar{v}_{1} = \frac{1}{2} + \frac{1}{3}(a^{2} + b^{2}) \omega_{2} + d_{2} = \frac{3}{2} + \frac{b}{a^{2} + b^{2}} \bar{v}_{1}$ $DA7A! = \frac{10}{12} ft = \frac{5}{12} ft$ $\omega_{2} = \frac{3}{2} \frac{5/n}{12} = \bar{v}_{1} + \frac{3}{2} \frac{5}{12} = 0.770 \bar{v}_{1}$ (1)



$$T_{i} = \frac{1}{2} I w_{2}^{2} + \frac{1}{2} m v_{2}^{2} = \frac{1}{2} I m (a^{2} + b^{2}) w_{2}^{2} + \frac{1}{2} m \frac{1}{4} (a^{2} + b^{2}) w_{2}^{2}$$

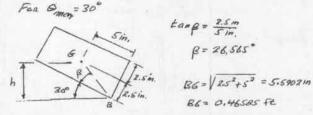
$$V_{i} = w \frac{b}{2} = mg \frac{b}{2}$$

$$V_{2} = w h = mg h$$

$$T_{2} = 0$$

$$T_1+V_2=T_2+V_2$$
: $\frac{1}{6}d\gamma(a^2+b^2)u_2^2+d\gamma_3\frac{b}{2}=d\gamma_3h$

$$\omega_2^2=\frac{6(h-b/2)}{(a^2+b^2)}5$$
 (2)

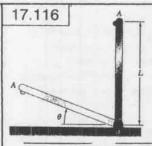


h=(B6)ton(30"+p)=(0.46585f6)sin(30"+26.565") h=0.38876 14

$$EG(2): W_{2}^{2} = \frac{6(h-b/2)}{9(a^{2}+b^{2})} = \frac{6(0.3887k-\frac{7.5}{12})}{\left[\left(\frac{10}{12}\right)^{2}+\left(\frac{5}{12}\right)^{2}\right]} 32.7 = 40.156$$

$$W_{2} = 6.237 \text{ rad/s}$$

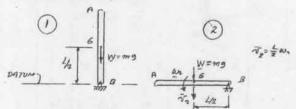
EQ(1):
$$W_2 = 0.720 \, \tilde{V}_1$$
; 6.837 = 0.720 \tilde{V}_1 , $\tilde{V}_2 = 8.80 \, \text{ft/s}$



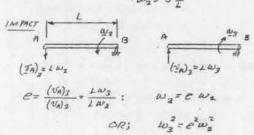
GIVEN: ROD AB IS SIVEN SLIGHT NUDGE AND ROTATES COUNTERCLOCKWISE AITS SURFACE AND REBOUNDS C=040

FIND: MAXIMUM & OF REBOUND.

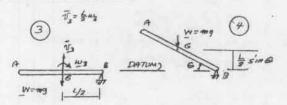
CONSERVATION OF ENERGYS



 $\overline{T}_{1} = 0 \quad V_{1} = m6 \frac{1}{2}$ $\overline{T}_{2} = \frac{1}{2} \overline{I} \omega_{1}^{3} + \frac{1}{2} m \overline{V}_{2}^{3} = \frac{1}{2} \cdot \frac{1}{12} m \overline{L}^{2} \omega_{2}^{3} + \frac{1}{2} m (\frac{L}{2} \omega_{1})^{2} = \frac{1}{6} m \overline{L}^{2} \omega_{2}^{2}$ $\overline{T}_{1} + V_{1} = \overline{T}_{2} + \overline{V}_{2}^{2} : 0 + m6 \frac{L}{2} = \frac{1}{6} m \overline{L}^{2} \omega_{2}^{3} + 0$ $\underline{W}^{2} = 3 \frac{6}{2}$



CONSERVATION OF ENERGY

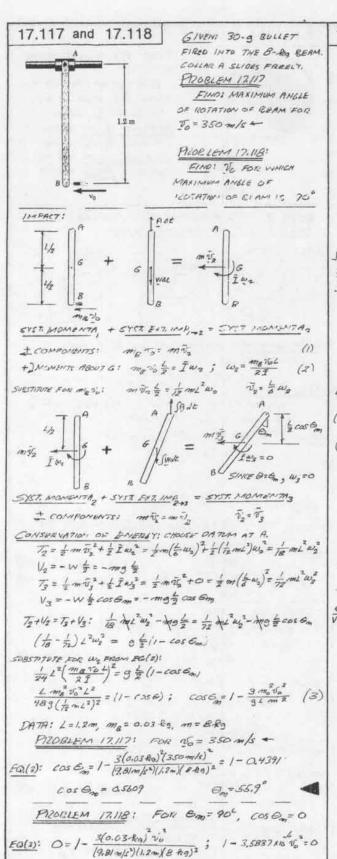


$$\frac{7_3 + V_3 = 7_4 + V_4}{\frac{1}{6} mL^2 (c^2 w_1^2)} = mg \frac{L}{2} sinG$$

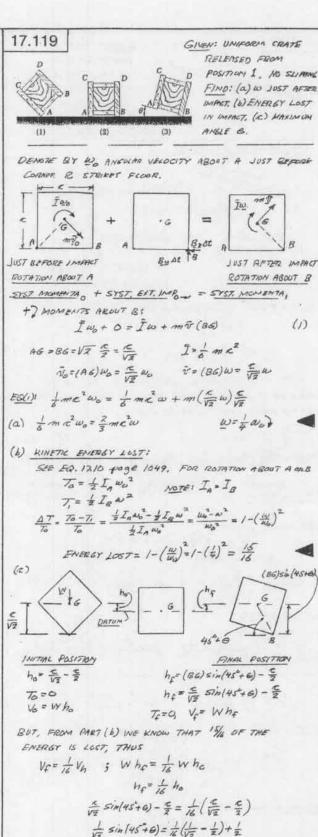
$$\frac{1}{6} mL^2 (c^2 w_1^2) = mg \frac{L}{2} sinG$$

$$\frac{1}{6} mL^2 e^2 (3 \frac{g}{L}) = mg \frac{L}{2} sinG$$

$$sinG = e^2$$



No = 279,04 x103

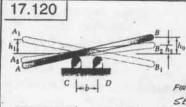


 $sin(45°+6) = \frac{2 + 15V2}{32} = 0.72541$

B=1.50°

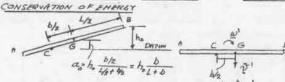
4540= 46.503°

20= 528 m/s - ■



GIVEN: ROD OF LENGTH L= 30 in. b=5in. ROD 15 RELETED WHEN h = 4 in. FIND: (a) h, AFTER

FIRST IMPACT. (6) ha AFTER SECOND IMPACT.



POSITION "PRIME" = \$ W POSITION O" V=0; T= 1 Iw + 1 mo Vo= mgao; To=0 ア==(たかん)い++か(冬い)2

T== mw1 (12+362) Totlo= T'+V': 0+ mg a = 1/2 man (12+362)

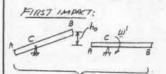
1 3 ho 1+ b = w (12+362) (w)= 24 1 b ho (L+b)(12+362) (1)

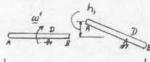
NOTE THIS EXPRESSION ALSO RELATES THE HEIMT THE END OF THE ROD RISES WHEN ANEVLAR VELOUTY W OCCURS WHEN ROD IS HORIZONTAL



POSITION "PRIME" POSITION "DOVALE PRIME V'= & w' v"= \$ 10"

+) MOMENTS ABOUT D: エル'ーかで'き= エル"+かず号 12 m L2w' - m (\$) w' = 19 m L w" + m (\$) w" $\omega'' = \frac{L^2 - 36^2}{L^2 + 36^2} \, \omega'$ (2)





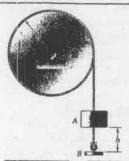
FO(i): $(w')^{\frac{2}{2}} \frac{249b}{(L+b)} \cdot \frac{ho}{(L^2+3b^2)}$ (co") = 243b · h.

 $(\omega'')^{\frac{2}{2}} \frac{(L^2 - 36^2)^2}{(L^2 + 36^2)^2} (\omega')^2$ $\frac{\begin{bmatrix} 248b \\ (L+b) \end{bmatrix} \cdot \frac{h_1}{(L^2+3b^2)} = \frac{(L^2-3b^2)^2}{(L^2+3b^2)^2} \cdot \frac{\begin{bmatrix} 249b \\ (L+b) \end{bmatrix} \cdot \frac{h_0}{(L^2+3b^2)}$ h, = [12-362] 2ho (3)

SECOND IMPACT: ho-h, h, -he h= [12-362] ho

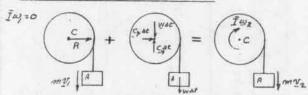
 $\frac{L^2 - 3b^2}{L^2 + 3b^2} = \frac{30^2 - 3(5)}{20^2 + 3(5)}^2 = 0.84615$ DATA: ho=4in, 1=30in, d=6in.

EQ(3): h,=(0.8+45)(4 in,) = 2.8639 in. EO(4): hz = (0.8484) (4in) = 2.0505in. h,=2.86 in h== 2,05in. 17.121 and 17.122



GIVEN: 3-16 COLLAN A DIROFS h=15 in. 8-16 DISK OF PROJUS R= 9%. IMMEDIATELY AFTER IMPACT FINO: (a) JA, (b) W PROBLEM 12.121 ASSUME PERFECTLY PLASTIC IMPACT. PROBLEM 17,122 ASSUME PERFECTLY ELASTIC IMPACT.

COLLAR A FALLS A DISTANCE h: v, = 129 h PRINCIPLE OF IMPULSE - MOMENTUM



SYST MOMENTA, + SYST, ETT. IMP, -2 = SYST, MOMENTA? + 2 MOMENTS AROUT C:

mr, R = Iwa + m voR (1)

PROBLEM 12.121

PLASTIC IMPACT C=0 V2=RW2; W2= V2 M=MASS OF DISK; I= +MR2

m V, R = 1 MR2 (1/2) + m = 2R EQ.(1): mv, = = M V2 + mV2

> V2 = 2 m V, (3)

DATA: m= 216; M= 916; h=15in. V, = V29h = V 2 (32.2 FUS?) (15 fe) = 8.972 ft/s

FG(3) $T_a = \frac{2(3/6)}{2(3/6) + \frac{6}{3}} (8.972 G/6) = 3.845 fd/s$ $Y_a = 3.$ V2 = 3.85 14/5 +

Wy= \frac{\su_2}{R} = \frac{3.845 \tels}{\left(\frac{7}{2} \tels)} = 5.127 \text{ rady} (wg = 5.13 rod/s)

PROBLEM 17,1221

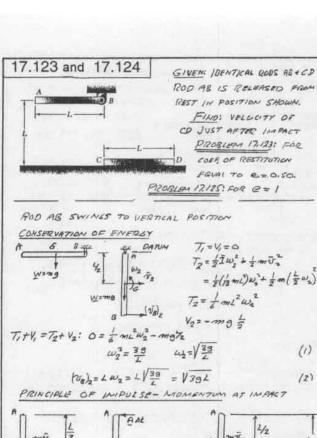
ELASTIC IMPACT e=1 (VB) = (VA) = (VA), -(Va), (VB) = 0 ; (VB) = RWO: RW2-V2=V, 3 W2=(V1+V2)/R (2)

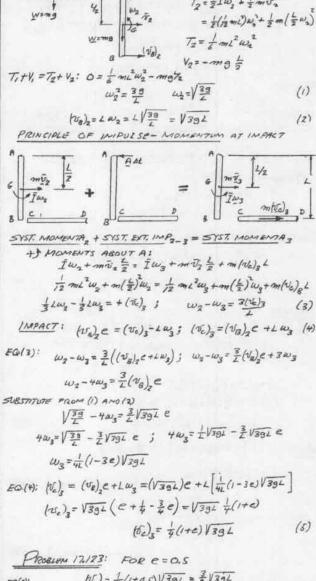
EQ(1): m v, R = 1 MR2 (V, + V2) + m V2 R mv, R= &MRV, + &MRV2 + mV2R

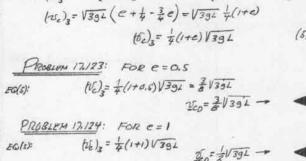
DATA: m= 21/9; M= 84/9; h= 15in., V,=8.972 ft/s

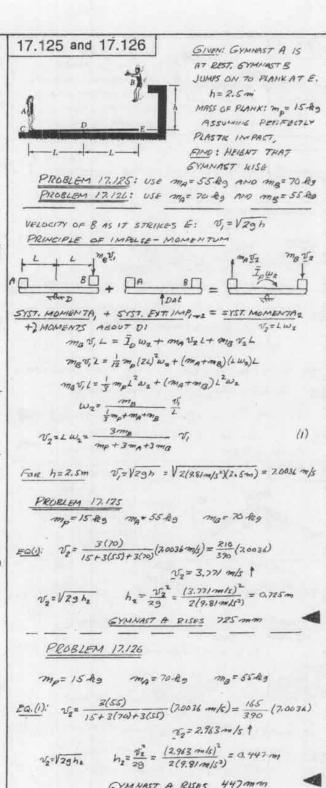
EQ.(4) $\sqrt{2} = \frac{2(\frac{3}{2}) - \frac{9}{2}}{2(\frac{3}{2}) + \frac{9}{2}} (8.972 \, \text{Fe/s}) = -1.2817 \, \text{Pe/s}$ 15=1.282 ft/s

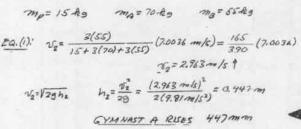
EQ(1): W2 = 10.254 radf (3 ft) Wz= 10,25 rad/5)

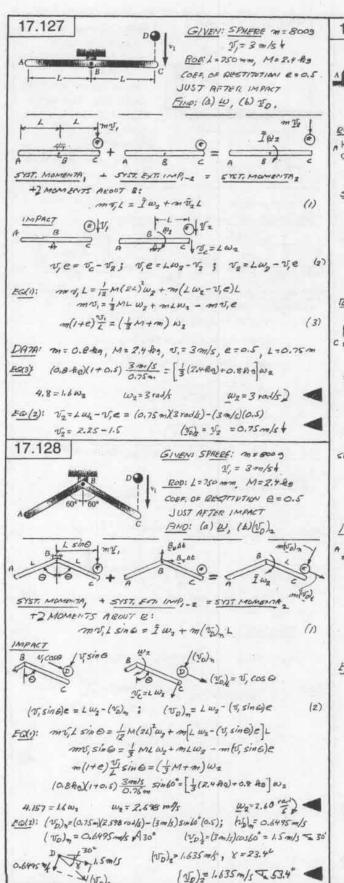


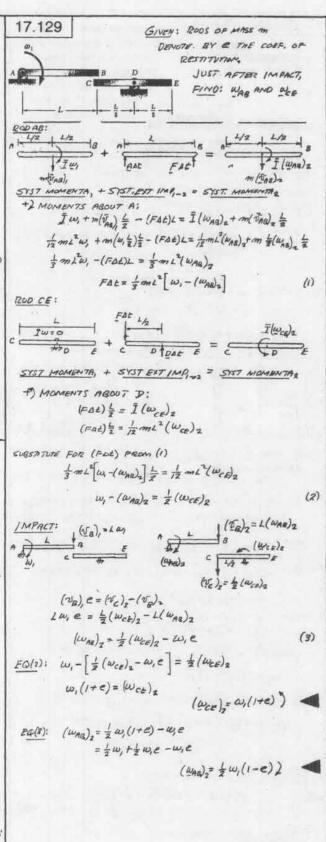


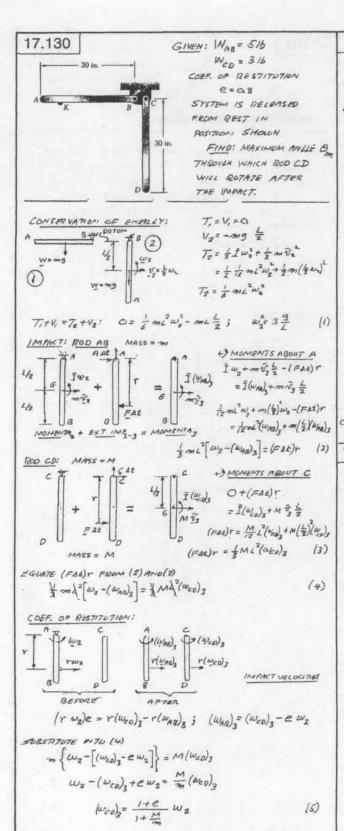












(CONTINUED)

17.130 continued DATA: m= (510)/9 M= (316)/9 E=0.8 L=2.5ft Es(1): $w_q^2 = \frac{39}{L} = \frac{3(32.2 \, \text{A/r})}{2.5 \, \text{ft}}$ W= 6.216 rad/s) $E_{G}(5)$ $(\omega_{c0})_{3} = \frac{1+e}{1+\frac{M}{2}} \omega_{2} = \frac{1+0.8}{1+\frac{3}{2}} 6.26 \text{ rad/s} = 6.993 \text{ rad/s}$ CONSERVATION OF ENERGY ROD CO: MASS = M W=M3 | YE C W (DATUM (Wes) 2 $T_3 = \frac{1}{2}\bar{I}(\omega_{e0})^2 + \frac{1}{2}M\bar{x}^2 = \frac{1}{2}\frac{M}{12}(\omega_{e0})^2 + \frac{1}{2}M(\frac{L}{2})(\omega_{e0})^2$ T3 = = ML2(WED) V==-Mg = Vu=+Mgh 74=0 T3+V3=T4+V4 1 ML 2 (WED) - Mg = Mgh h + L = L2 (Wep) 2 $h + \frac{2.5 \, \text{ft}}{2} = \frac{(2.5 \, \text{ft})^2}{6 \, (32.2 \, \text{ft/s}^2)} \, (6.993 \, \text{mad/s})^2$ h+1.25ft = 1.582ft h = 0.332 ft B= 15.40 @=105.4° Om= 90°+ 15.4° h=0332FE 17.131 GIVEN: SPHERE A POLLS AND STRIKES SPHETOE B. ASSUME PERFECTLY ELASTIC IMPACT AND DENOTE COEF, OF KINETK FIRICTION BY Mr. FIND: JUST AFTER IMPACT (a) WE AND IT OF EACH SOMERE. (6) FINAL VELOCITY OF BACK SPHERE. (a) IMMEDIATELY AFTER IMPACT SPHERE A WAS ROLLING ", = YW, I(WA)2 FALTO SPHERE B:

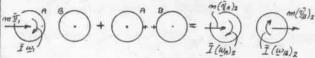
+) MOMENTS ABOUT G Iw, = I(WA)= (WA)= W.)

FALZO I(wa)=6 +) MOMENTS ABOUT G: (wB) = 0

(CONTINUED)

17.131 continued

CONSIDER BOTH SPHERES AS A SYSTEM



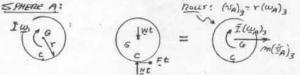
+ SYST. EXT. IMP = 5YST, MOMENTAZ I COMPONENTS! m v, = m(va)2+m(va)2

$$\tilde{v}_{r} = (v_{n})_{r} + (v_{n})_{z} \tag{1}$$

RELATIVE VELOCITIES (e=1) V, = (VB) - (VA) 2 (v) - (v) = ev = v,

ADD EQS. (1) AND (2):
$$2\bar{v}_1 = 2(\bar{v}_2)_2$$
; $(v_2)_2 = \bar{v}_1 \rightarrow v_1 = v_2$
SUBTRAT EQ(1) FROM FQ(1): $c = 2(\bar{v}_2)_2$; $(\bar{v}_3)_3 = 0$

(b) MOTION AFTER SPHERES START ROLLING UNIFORMLY NOTE: TIME INTERVAL IS NOT SMALL AND IMPUSES OF FRICTION FORCES MUST BE INCLUDED



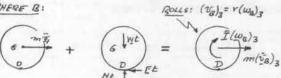
SYST, MOMENTA, + SYST. EXT. IMP = SYST. MOMENTA

+3 MOMENTS ABOUT C:
$$\bar{I}\omega_1 = \bar{I}(\omega_n)_3 + m(\bar{v}_n)_3 \Upsilon$$

 $\frac{2}{5}mr^2\omega_1 = \frac{2}{5}mr^2(\omega_n)_3 + mr^2(\omega_n)_3$
 $(\omega_n)_2 = \frac{2}{5}\omega_1$

(VA)=ア(WA)=ラアW,=ラび,

SPHERE B:



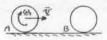
SYST MOMBITA + SYST EAT IMP2-3 = SYST, MOMBITA3 +) MOMENTS ABOUT DI mv, r = I(WB)3+m(TB)3 r

$$m\eta_{r} = L(u_{g})_{3} + m(\eta_{g})_{3} + mr^{2}(w_{g})_{3}$$

$$(\vec{v}_{0})_{2} = r(\omega_{p})_{2} = r(\frac{5}{7}\vec{v})$$
 $(\vec{v}_{0})_{3} = \frac{5}{7}\vec{v}, -$

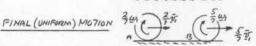
SUMMARY

INITIAL MOTION

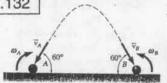


JUST AFTER IMPACT





17.132



GIVEN: BALL BOUNCES AS SHOWN. 4=40, VA=26 FIND: WO IN TERMS OF TO AND T

SINCE THE LINEAR MID ANGULAR VELOCITIES ARE CHANGED DURING A SHORT INTERVAL AT, BOTH THE NORMAL AND FRICTION FORCES ARE IMPULSIVE WE ASSUME THAT NO SLIPPING OCCURS.

FOR THE VELOCITY OF THE BALL TO BE REVERSED AT EACH IMPACT, WE MUST HAVE AT POINT A.

BEFOREIMPACT

AFTER IMPACT

$$\underline{\underline{\underline{\underline{T}}}}_{M} \underbrace{\underline{\underline{T}}}_{M} \underbrace$$

SYST. MOMERTA, + SYST. EXT. IMP = SYST. MOMENTAL

+ 2 MOMENTS AROUT C: I wa- (m va cos60) r = - I wa + (mi cos60) r

SUBSTITUTE: WA = WA = WO NA = NA = 76

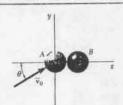
$$Z\bar{I} w_o = 2(mv_o \cos 60^\circ) r$$

$$w_o = \frac{(mv_o \cos 60^\circ) r}{\bar{I}}$$

$$w_o = \frac{mv_o (\frac{1}{2}) r}{\frac{2}{5} mr^3}$$

$$w_o = \frac{5}{4} \frac{v_o}{r}$$

17.133 and 17.134



GIVEN: BALL A IS ROLLING WITHOUT SLIPPING YVERN IT HITS BALL B. COEF, OF WHAT THE FAIRTION IS 4/6.
ASSUMING PERFECTLY FIRSTK IMMET,

FIND: (a) \$ AND WOF

FACE GAIL, (b) \$8 AFTER

17 STANTS ROLLING

PROBLEM 17.134! FIND EQUATION OF PATH OF BALL A

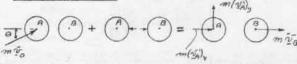
PROBLEM 17.133

(a) MOTION IMMEDIATELY AFTER IMPACT

FRICTION FARCES ARE NON IMPULSIVE, THUS ANGULAR MOMENTUM (AND THUS W) OF EACH BALL IS UN CHANGED. WE HAVE W'S = 0

AND SINCE BALL A WAS ROLLING:

LOOKING DOWNIWARD



SYST. MOMENTA, $+ SYST. EXT. IMP_{O \to 1} = SYST. MOMENTA,$ $\pm comparents: m \bar{\tau}_0 \cos \theta = m(\bar{\tau}_0^i)_0 + m \bar{\tau}_0^i \qquad (1)$

+1 components OF RAIL A: $m\vec{x}_0 \sin \theta = m(\vec{x}_0)$ $|\vec{x}_0|_{\mu} = \vec{x}_0 \sin \theta$ (2)

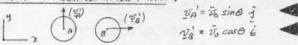
FOR ELASTIC IMPACT
$$e=1$$

$$v'_{a} - (v'_{a})_{x} = \bar{v}_{a} \cos 6 \qquad (3)$$

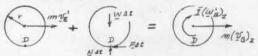
SOLVING SIMULTANEOUSLY Eas(1) AND (2)

$$(\vec{v}_{A}) = 0$$
 $(\vec{v}_{B}) = \vec{v}_{0} = cos\theta$ (4)

MOTION IMMEDIATELY AFTER IMPACT:



(b) FINAL VELOCITY OF BALL B:



SYST, MOMENTA, + SYST. EXT. IMP_= z = SYST. MOMENTA z = 1 MOMENTS ABOUT D: $m 2g'r = 1(w_0')_2 + m(v_0)_2 r$ (5)

WE RECALL! V' = V, cosa ANO I = 2 mr

RALL ROLLS: (NG) = r(Wg)2

EQ(S): $mr\tilde{v}_0 \cos \theta = \frac{2}{5}mr\tilde{v}(w_8)_2 + mr(w_8)_2 r$ $[w_8]_2 = \frac{5}{7}\frac{\tilde{v}_4}{r}\cos \theta \qquad [\tilde{v}_8]_2 = r(w_8)_2 \rightarrow (v_8)_2 = \frac{5}{7}\tilde{v}_0\cos \theta \tilde{L}$

(CONTINUED)

17.133 and 17.134 continued

WE ASSUME THAT BALL A ROLLS WITHOUT SLIPPING IN Y DIRECTION (May = (Wa) x X PR = - (To Sind i) X I-R = + TO SIND I
ASSUMPTION IS CORRECT

THE 4 COMPONENT OF VECTOR IS CONSTANT AND THUS THE 4 COORDINATE AT ANY TIME t is $y = (v_A)t = (\tilde{v}_0 \sin \theta)t$

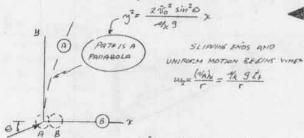
= COST. OF KINSTIL PRETION BETWEEN BALLS AND TABLE
BALL A ROLLS AND SLICES IN THE Y DIRECTION

$$\tilde{I}\left(\frac{\tilde{\chi_p}}{r}\cos\theta\right) \xrightarrow{\left(\frac{1}{r}\cos\theta\right)} + \left(\frac{1}{r}\cos\theta\right) + \left($$

± x composents: 0 + 4/mg t = m(in),

 $(a_{A})_{x}^{2} = 4\chi_{3}^{2} = constant \qquad x = \frac{1}{2}at^{2} \qquad x = \frac{1}{2}\pi_{A}3t^{2}$ (2)

Eliminate t Between EGS (1) A+O(2) $t = \frac{1}{2} \frac{1}{\sqrt{2}} \frac{3}{\sqrt{2}} \frac{3}{$



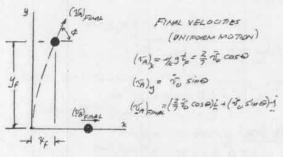
+) MOMENTS ABOUT 6: I (TEOSO) - (7, mgt) = I N2

= mr (TEOSO) - 4, mgt = 2 mr 2 (12, gt 4)

= 16,000 = 3 12 ts

ROLLING WITHOUT SLIDING BEENS WHEN $t_f = \frac{2}{7} \cdot \frac{\overline{V}_0 \cos \theta}{\sqrt{k} \cdot \theta}$ $EQ.(2): \quad \gamma_f = \frac{1}{2} \gamma_b \cdot 3 t_f^2 = \frac{2}{47} \cdot \frac{\overline{V}_0^2 \cos^2 \theta}{\sqrt{k} \cdot \theta}$

Ew(1): y= 10 sin 6 to = 2 10 sin 6 cos

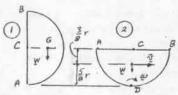


(VA) = (VB) = 5 Vo cos 8 1

17.135

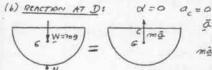


GIVEN: UNIFORM HEMISPHENE IS RELEASED FROM REST AND ROLLS WITHOUT SLIDING, AFTER HEMISPHERE ROLLS THROUGH 90, FINO: (a) W (b) NORMAL REACTION.



(a) WORK- ENERGY: U1=2=W(3+)=3 mgr $T_2 = \frac{1}{2} \tilde{I} \omega^2 + \frac{1}{2} m \tilde{v}^2 = \frac{1}{2} (\frac{2}{5} - \frac{7}{26}) m r^2 \omega^2 + \frac{1}{2} m (\frac{5}{8} r) \omega^2$ 7= 13 71 FW

T,+ U,-2=72: 0+3 mgr = 13 mrw w=1.074 3/2)



a=a=+a===(3+)w1+ ma=m/37009

+ | EF = EFeft: N-mg = m (3 r) w N= mg+m(3 r)(16 9)= 149 mg

17.136

N=1.433 mg 1 GIVEN: 4 = 0.15 10 = 25 m/s V CYLINDER IS AT REST WHEN PLACED FINO: (a) NUMBER OF PRUMINTIALIE

REFORE CYLINDER REACHES CONSTANT VELOCITY. (b) TIME REQUIRED TO REACH CONSTANT VELOUTY.

WHILE SLIPPING OCCURS:

+ I I F4 = 0 NOOSB-MNSIMB-Mg=0 N= mg
casB- of sin B (1)

SLIPPING GCCURS UNTIL:

WORK-EMERGY ME FY = MOMENT OF F ABOUT A. U = MD = Fr0 = 4, Nr0

Ti+U1-2= Tz: O+1/2Nr0 = + mv2 0= 1 mv 1 = 1 mv cosp-4/2 sing (CONTINUED)

17.136 continued

PRINCIPLE OF IMPULSE - MOMENTUM



SYST. MOMENTA, + SYST. EXT. IMP, -2 = 5YST. MOMENTAR +) MOMENTS ABOUT A: FLY = IN -4x NEr = = = mr (2)

SUBSTITUTE FOR HI 4/ (cosp - 4/ sma) tr = 1 mrv

(3) t= 1 1/9 (cosp - 4/4 sine)

 $\frac{(DA7A^2 - 4)_R = 0.15}{(B = 0.15)} = 25^{\circ}, \quad 15 = 25 \text{ mils}, \quad r = 0.12 \text{ m}$ $\frac{(A7A^2 - 4)_R = 0.15}{(A(A)^2 - 4)^2} = \frac{(A7A^2 - 4)_R + (A7A^2 - 4$

0= 745,86 red (REY) ; B= 118. Trevolutions

Fials): t = 25 m/s [cos 25°-(0.15) sin 25]; t=7.165

17.137

GNEH! 4x = 0.15 V= 25 m/5 CYLWORR AT REST PLACED ON BELT UNTIL MOTION BECOMES UNIFORM FIND: (4) NUMBER

OF REVOLUTIONS REQUIRED. (6) TIME INTERML REQUIRED

W=mg WHILE SLIPPING OCCURS! E=XN + | ITY=OI NOOSB + 4 NSINB - Mg=0 N= COSB+ 1/2 SIMB (1)

FOR CYLINDERS SLIPPING OCCURS UNTE WET WORK-ENERSY: ME Fr = MOMENT OF F AGOUT A. U1-Z= MAG= Fro = MAYO Ti=0: Ta= \$ 1 w2= \$ (\$ mr2)(\$) = +mv2

TitU,== Ta: O+1/2N+0= +m52 6= + mv2 1 = 1 mv2 cosp+4 sino @= 4 . V= (cosp +4x sine) (2)

PRINCIPLE OF IMPULSE-MOMENTUM



+2 MON ENTS ABOUT A! -4xNtr==mr2(V)

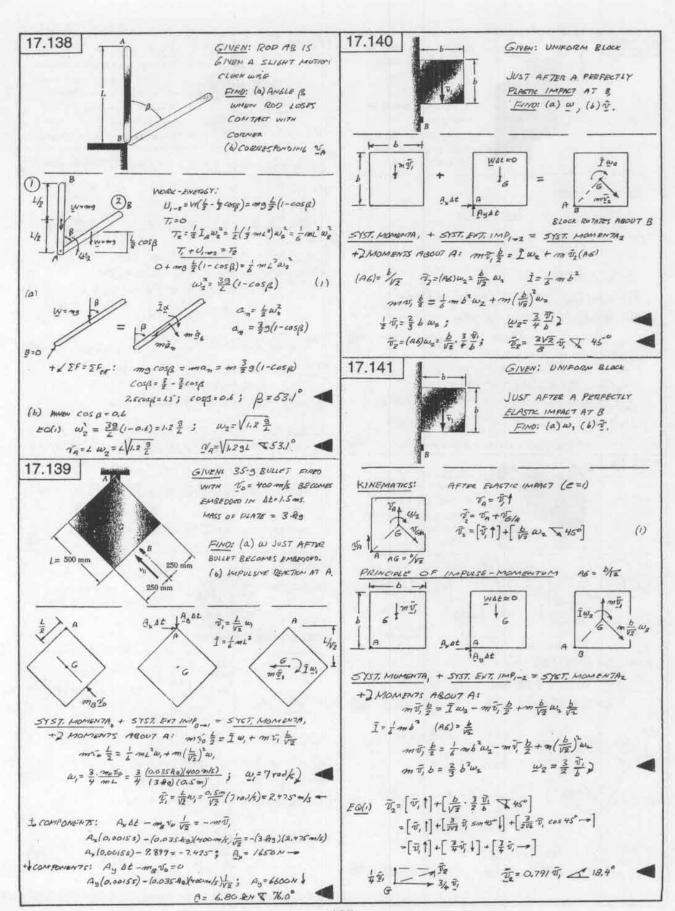
SUBSTITUTE FUR M

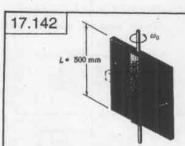
Mx (cosp+4x sing) tr = 1 mrs t= 1 v (cosp+4ksing) (3)

DATA: 4/2 = 0.15, B = 25", V= 25 m/s, r= 0.12m

EQ(2): $\Theta = \frac{(25 \text{ m/s})^2}{4(0.15)(0.12 \text{ m})(9.81 \text{ m/s}^2)} \left[\cos 25^6 + (0.15) \sin 25^6\right]$ 0= 136.6 revolutions 6 = 858.05 rad (70 rad)

6= 4 12 (cosp-1/2 sing) (2) Ea(s): t= 25mls 2(0,15) (9.81 m/s) [cos25*1(0.15) sin 25*] t=8.245





GIVEN: 3-AS BAR AB 4-kg PLATE W= 120 rpm

AFTER BAR SWUNG TO HODIZON TAL, FIND: (a) W , (b) EVERLY LOST DURING PLASTIC IMPACT AT C

(a) LOOWING DOWN WARD

CONSERVATION OF AMEDIAR MUMERITUM ABOUT SHAFT $I_{\omega} = I_{\omega}$

I= IPIATE 12 (4-Rg)L2

I,= Iping+ I ang= 1/2 (4Ag)12+ 1/2 (3Ag)12= 1/2 (7Ag)12

EQ(1): 12 (4Ag) L2 (120 rpm) = 1 (7 kg) L2 W,

W= \$ (120 vpm) W,= 68.6 rpm

(b) ENERGY: (WE MUST USE rad/s) , Up = 120 rpm (27) = 4 17 rack = 12,566 radt W,= 4 40 = \$ (411 rad/s) = 7.18/ rad/s To = \$ Io Wo2 = 1 (4 AgX0.5 m)2 (12.566 reals) = 6.580 J Ti= 121 w, = = = [(7.89)(0.5m) 2 (7.181 rack) = 3.760)

ENERGY LOST = 6.580 J-3.7601 = 2.82 J

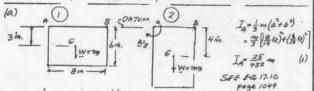
17.143

(6)



GIVEN: PINB 15 REMOVED AND PLATE SWINGS ABOUT A FINO: (a) W AFTER 90° ROTATION. (b) MAXIMUM W

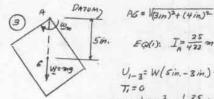
ナッカエルン



U,=== W (4in.-3in,)=mg(1/2 ft) Ti=0: T== +I, w= + 25 +w"

W= 18 9

w= 18 3 = 18 (322) = 23.184 W= 4.81 rad/s)



AG = 1(3/n)2+ (4/n)= = 5 in.

U1-3= W(5in. - 3in.) = mg/2 fd

 $T_2 = \frac{1}{2}I_2\omega_3^2 = \frac{1}{2}\frac{25}{482}\omega_3^2$

T,+U,-3=T3 1x9(3)= = = 25 mm3 17.144

GIVEN; DISK OF SAME THICKNESS AND SAME MATERIAL DISIC B IS AT REST LANGE IT IS DROPPED ON DISK A KNOWING WA = 1816 FIND: (a) FINAL W OF DISKS (6) CHANGE IN KINETIC ENERGY

FOR A DISK; m=pv=pn+2; Î= = m+== = [pn+2) += = pnt+

Inwa Iwg=

SYST, MOMERITA + SYST. EXT. IMP, == SYST. + MOMENTS ABOUT 6: In Wa = In W4 + I W4 1 pne ra wa= (1 pnera + 1 pmera) was WE = THY WA

Wy = (6 in)+ (4 in)+ (500 ypm) = 417.526 ypm

(b) ENERGY: Wa=mag=pgntra Wa=mag=pgfit ra Wa=1816; W13=(V8) WA= (4/n.) (1816) = 816

INITIAL KINETIC EMERGY WA= 500 pm (20) = 52.36 rad/f T,= = 1, WA = 1 2 1 100 (6/2 ft) (52.36 rodk) = 95.284 ft.16 W= 417526 rpn (211) = 43.723 rad/s

 $T_2 = \frac{1}{2} \left(\tilde{I}_B + \tilde{I}_B \right) u_p^2 = \frac{1}{2} \left[\frac{1}{2} \frac{1816}{32.2} \left(\frac{6}{12} \text{ GE} \right) + \frac{1}{2} \frac{816}{32.2} \left(\frac{44}{12} \text{ TE} \right) \right] \left(43.773 \text{ rad/s} \right)^2$ To= 79.985 ft.16

ENERGY LOSS: AT=7,=77,785 A.16-95.284 (F.16 17=-15,80 Ft.16

17.145



FIND: DISTANCE h IF BALL IS TO START ROLLING WITHOUT SLIDING

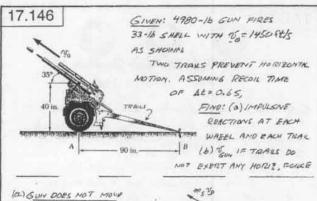


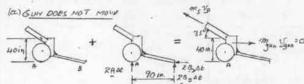
SYST. MOMENTA, + SYST. EXT. IMP, = SYST. MOMENTA2

+] MOMENTS ABOUT 6: (PAE) h = I w2 (1) - COMPONENTS: Pat = m V. (2)

DIVIDE EG(1) BY EG(2) MEMBER BY MEMBER h= 1 . 11

FOR ROLLING DITTO





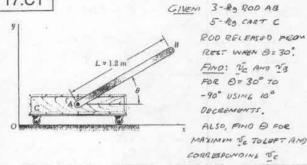
SYST. MOMENTA, + SYST, FIT. IMP, = SYST, MOMENTA: + COMPONENTS: 28, At = ms Vo cos 350 2B, (0.65) = (384) (1450 ft/s) cos 35° B= 1014 16 -By= 1014.416

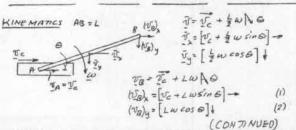
+) MUMENTS AROUT A: 28,00 (90 in.) = (m, v, cos 35°) (40 in) 2B, (0.65) (40in) = (331) (1460 (4/5) cos 35" (40 in) Bn= 450.816 By=45/16+

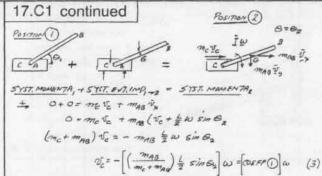
+1 COMPONENTS: ZADE + 2By Ot = ms 16 sin 35" 2(A+BB)(0.6=) = (3316) (145016) sin 35° A+B= 710.3 16 A= 259 15 t A+450,810=710.316

(b) TRAILS NOT EMBEDDED By=0, VEW +0 + COMPONENTS: 0+0 = - ms 20 5/n 35" + mgun 25 au 0 = - (33/6) (1450 (45) cas 35" + (4980/6) Ngan

2 que 7.87/ 12/5 25 = 7.87 St/s -+ 17.C1







CONSERVATION OF ENERSY Wang

IZH WASL 1/2=mg = sin 03 V,=mg = sin 0, Ti=0 T2= = = mc va + = IW+ = mAB v = 1 2 = V2 + Vy2 = (TE + = wsin 0) + (= 4 cos 0) T= 12+Lwsie 6+ 1220 = [(coppe())2+ (coppe())Lsing + 4 W V = [COFFE @] W" (4)

To = = {m (coeff())+ I+ mag (coeff(2))} w" 72 = 1 (COPF#3) W2 (5)

T, +V, -T2+V2 0+mas 2 sing = 2 [COFF 3] W+ mas 2 518 62 W= MABBL (5100, -5100) (6) COEFF 3

OUTLINE OF PROGRAM:

ENTER DATA: L=1.20, m=5-ho, mAD=3-hg, O,=30°. PROGRAM IN SEQUENCE EQS. (3), (4), AND (5) WHICH CONTAIN THE THREE COEFFICIENTS, THEN PROGRAM EGS, (1) AND (2) THAT INVOLVE (TA), AND (Sa), FUALUATE AND PRINT

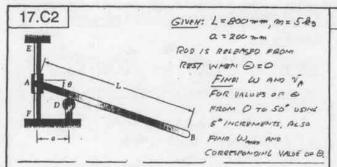
0, w, v, v, (Va), (VB)4

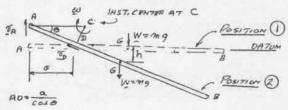
Linear velocities positive to the right and up Omega positive clockwise

theta	omega	vAB=0	vC	vBx	vBy
deg.	rad/s	m/s	m/s	m/s	m/s
30.00	0.000	0.000	0.000	0.000	0.000
20.00	2.002	1.157	-0.154	0.667	-2.257
10.00	2.841	1.689	-0.111	0.481	-3.358
0.00	3.502	2.101	0.000	0.000	-4.202
-10.00	4.082	2.427	0.159	-0.691	-4.824
-20.00	4.621	2.672	0.356	-1.541	-5.211
-30.00	5.136	2.837	0.578	-2.504	-5.338
-40.00	5.631	2.923	0.814	-3.529	-5.177
-50.00	6.098	2.933	1.051	-4.555	-4.704
-60.00	6.516	2.881	1.270	-5.502	-3.910
-70.00	6.854	2.795	1.449	-6.279	-2.813
-80.00	7.076	2.715	1.568	-6.795	-1.475
-90.00	7.154	2.683	1.610	-6.975	-0.000

Find maximum velocity of cart to the left. 19.70 2.0315 1.1760 -0.15408479 -0.15408483 1.1766 19.68 2.0335 1.1772

(VE) TO LEFT = 0.1541 m/s WHEN 6= 19.7"





$$CD = AD to -G = a \frac{tanG}{cosG} ; AC = \frac{AD}{cosG} = \frac{a}{cos^2G}$$

$$\frac{MASS MOMENT OF INENTIA ABOUT INST, CONTEX}{I_c = \overline{I} + m \left[(cD)^2 + (0S)^2 \right]}$$

$$\hat{I_c} = \frac{1}{12} mL^2 + m \left[a^2 \frac{tanG}{cos^2G} - \left(\frac{L}{2} sinG - a tanG \right)^2 \right]$$
 (1)

CONSTRUATION OF ENERGY

T= 1 I w (SAG EQ 17.10 page 1049)

$$\frac{T_1 + V_1 = T_2 + V_2}{\omega^2} = \frac{1}{2} I_{ew}^2 - meh$$

$$\omega^2 = \frac{2msh}{I_e} \qquad \omega = \sqrt{\frac{2msh}{I_e}}$$
(2)

VELOCITY OF A: VA= (Ac) W

OUTLINE OF DROGRAM

PROGRAM IN SEQUENCE, AD, D6, h, CD, AC, I_c , W, V_A . EVALUATE AND PRINT Θ , h, W, AND V_A FOR VALUES OF Θ FROM O TO 53° AT S° INTERVALS.

L = 800 mm	a = 20	O mm m	= 5 kg
		(5.000)	
theta	h	omega	VΑ
deg	mm	rad / s	m/s
0.000	0.000	0.000	0.000
5.000	17,365	1.911	0.385
10.000	34.194	2,680	0.553
15.000	49.938	3.235	0.693
20.000	64.014	3.648	0.826
25.000	75.786	3.934	0.958
30.000	84.530	4.079	1.088
35.000	89.389	4.051	1.208
40.000	89.295	3.811	1.299
45.000	82.843	3.325	1.330
50.000	68.067	2.592	1.255
***********	+++++++++	+++++++++++	++++++
Find thet	a for max o	mega	
theta	h	omega	
doca	200	Omeganias	

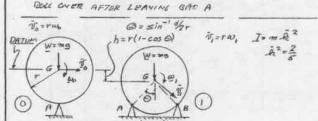
theta deg	h mm	omega _{Mas}
31.810	86.788	4.0907731056
31.820	86.799	4.0907735825
31.830	86.810	4.0907735825
31.840	86.821	4.0907731056
++++++++	++++++++	****************

17.C3



GIVEN: 10-in. RADIUS SPHERE POLLS OFFER BARS
WITHOUT SLIPPING. KNOWING THAT WELLS TOUGHT DE PROPERTY PLATIC IMPROTS, FOR d=1 in.
TO 6in. USING O.S-in. INCREMENTS,

FIND: (A) W, AS G PASSES DIRECTLY ABOVE B



CONSENTION OF EXPERTY $V_0 = 0; \quad T_0 = \frac{1}{2} \hat{L} w_0^2 + \frac{1}{2} m v_0^2 = \frac{1}{2} m v_0^2 + \frac{1}{2} m v_0^2 = \frac{1}{2} m \left(\frac{1}{4c} + r^2\right) w_0^2$ $V_1 = -mgh; \quad T_2 = \frac{1}{2} m \left(\frac{1}{4c} + r^2\right) w_0^2 = \frac{1}{2} m \left(\frac{1}{4c} +$

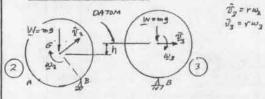
AFTER IMPACT AT B: SPIECE ROTATES AROUT B



BEFORE IMPACT AT B AFTER IMPACT AT B

+2 MOMBETTS ABOUT B: $\tilde{I}w_1 + (mv_1\cos 2\theta)r = \tilde{I}w_2 + mv_1r$ $m\tilde{R}^2w_1 + mv_1\cos 2\theta w_1 = m\tilde{R}^2w_2 + mr^2w_2$ $w_2 = \frac{\tilde{R}^2 + r^2\cos 2\theta}{\tilde{R}^2 + r^2}w_1$ (2)

SPHERE ROTATES ABOUT B UNTIL 6 IS ABOVE B



$$V_{2} = -mgh; T_{2} = \frac{1}{2}m(-k^{2}+r^{2})\omega_{2}^{2}$$

$$V_{3} = 0 ; T_{3} = \frac{1}{2}m(-k^{2}+r^{2})\omega_{3}^{2}$$

$$\frac{T_{2} + V_{2} = T_{3} + V_{3}:}{\frac{1}{2}m(-k^{2}+r^{2})\omega_{2} - mgh = \frac{1}{2}m(-k^{2}+r^{2})\omega_{3}^{2}}$$

$$\omega_{3}^{2} = \omega_{2}^{2} - \frac{2gh}{k^{2}+r^{2}}$$
(3)

(CONTINUED)

17.C3 continued

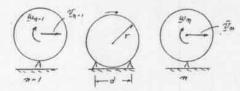
INF MAVE FOUND:
$$\omega_1^2 = \omega_0^2 + \frac{23h}{\lambda^2 + r^2}$$
 (1)

$$\omega_{z} = \frac{-\hat{k}^{2} + r^{2} \cos 2\Theta}{-\hat{k}^{2} + r^{2}} \omega,$$
 (2)

$$\omega_3^2 = \omega_2^2 - \frac{2gh}{\bar{h}^2 + r^2}$$
 (3)

W3 IS ANGULAR VELOCITY OF SPHERE AS & PASSES
OVER B. (THIS IS SHOWN AS W, IN PROBLEM FILIDE)

AS SPHERE ROLLS FROM THE (N-1) HEAR TO THE THE AND BAR.



ENTER DATA: V = 10 ft, Wo = 1.5 voils = 2 = 0.4

FOR
$$d = \frac{1}{12} ll$$
 TO $\frac{l}{12} lt$; INCREMENT $\frac{O.S}{12} ft$
 $W_n = W_0$

FOR $m = 1$ TO 1000 SINCREMENT = 1
 $\Theta = Sin^{-1} (\frac{d}{2}r)$
 $h = r(1 - cos \Theta)$
 $W_i = \left\{ w_m^2 + 2s h / (\frac{1}{R}^2 + r^2) \right\}^{V_2}$
 $W_2 = \left\{ (\frac{1}{R}^2 + r^2 cos 26) / (\frac{1}{R}^2 + r^2) \right\}^{V_2}$
 $W_3 = \left\{ W_i^2 - 2g h / (\frac{1}{R}^2 + r^2) \right\}^{V_2}$
 $IF n = 1$ PRINT W_3 (G IS ABOVE B)

 $IF n = 1$ PRINT W_3 (G IS ABOVE B)

 $IF M_3 < O \rightarrow STOP$,

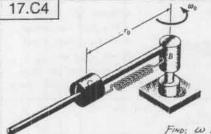
 $(n) IS NUMBER OF EARS$

ROLLED OVER)

r = 10.000 in. omega0 = 1.500 rad/s

1 - 10,000	Tit. Omogao	- 1.500 Tau/a
Distance between bars in.	omega when G is over B rad/s	Number of bars sphere rolls over
1.0 1.5 2.0 2.5 3.0 3.5 4.0 4.5 5.0	1.494 1.487 1.476 1.460 1.438 1.409 1.370 1.319 1.252 1.164	491 169 76 40 23 14 9 6
6.0	1.047	2

NOTE: FOIR d=7 in, SPHERO FAILS TO REACH
A POSITION WITH & ABOVE B



SPRING: MC = 2.5 &g

SPRING: & = 750 N/m

UN STOUTCHED

LUNGTH: 6= 500 mm

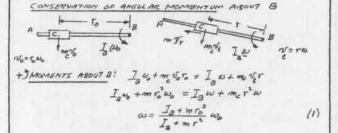
1000 AND HUS:

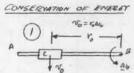
IB = 0.3 &g. m²

INITIALLY: Y = 500 mm

We 10 ralls

FIND: W AND TO FAR FOR VALUES OF F FROM 500 MM TO 700 MM AT 25- MM INCREMENTS. ALSO FIND FINEX.





or store of a

SPRING UNDEFERMED

STRETCH OF SPRING D=(r-t)

$$\begin{split} &\mathcal{T}_{1} = \frac{1}{2} I_{B} w_{0}^{2} + \frac{1}{2} m v_{0}^{2} = \frac{1}{2} I_{B} w_{0}^{2} + \frac{1}{2} m v_{0}^{2} u_{0}^{2} = \frac{1}{2} (I_{0} + m v_{0}^{2}) u_{0}^{2} \qquad V_{1} = \mathcal{S} \\ &\mathcal{T}_{2} = \frac{1}{2} I_{0} w^{2} + \frac{1}{2} m v_{0}^{2} + \frac{1}{2} m v_{0}^{2} + \frac{1}{2} \mathcal{H} d^{2} \\ &= \frac{1}{2} I_{B} w^{2} + \frac{1}{2} m v_{0}^{2} u^{2} + \frac{1}{2} m v_{0}^{2} \\ &\mathcal{T}_{2} = \frac{1}{2} (I_{B} + m v_{0}^{2}) w^{2} + \frac{1}{2} m v_{0}^{2} \\ &\mathcal{T}_{1} + V_{1} = \mathcal{T}_{2} + V_{2} : \frac{1}{2} (I_{B} + m v_{0}^{2}) u_{0}^{2} = \frac{1}{2} (I_{B} + m v_{0}^{2}) u^{2} + \frac{1}{2} m v_{0}^{2} + \frac{1}{2} \mathcal{H}(r - r_{0})^{2} \\ &\mathcal{T}_{\Gamma} = \left\{ \frac{1}{2} \left[(I_{B} + m v_{0}^{2}) u_{0}^{2} - (I_{B} + m v_{0}^{2}) u_{0}^{2} - \mathcal{H}(r - r_{0})^{2} \right] \right\}^{V_{2}} \end{split}$$

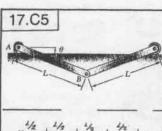
OUTLINE OF PROGRAM:

ENTER OATA: m=2.5-kg, I=0.3 kg·m, 1=0.5m, R=750H/m and W=10 radf

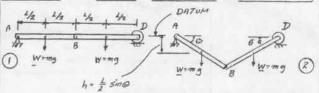
PROGRAM EQ(1) AND THEN EQ(2). EVALUATE AND PRINT LU AND V_{ξ} FOR VALUES OF Y FROM 0.5 mm 70 0.7 mm AT 0.025 m INCREMENTS. THEN SEEK YMAX WHERE V_{ξ} =0

r mm		omega rad/s	v	radial m/s
500.00 525.00 550.00 575.00 600.00 625.00 650.00		10.000 9.352 8.757 8.211 7.708 7.246 6.820 6.428		0.000 1.486 1.962 2.221 2.341 2.346 2.239 2.007
700.00		6.066		1.599
 Find	r	maximum	(where	vr = 0)
r		omega	U	radial]~2

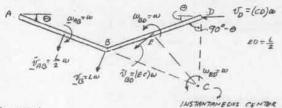
r omega [v radial]^2 mm rad/s 731.75 5.645 0.0014211 731.76 5.645 0.0004968 731.77 5.645 -0.0004275 731.78 5.645 -0.0013555



GIVEN: L = 30 in. BARS ARE RULEAS NO FROM REST WHEN 6 = 0. FIND: WAR AND ID FOR VALUES OF & FROM O TO 90" USING 10" INCRE MONTS



KINEMATICS OF POSITION 2:



AB= BC=L

$$\frac{1 \times \Delta CED}{(1 \times \Delta CED)^2 + (1 \times \Delta CED)^2 + (2 \times \Delta CED)^2 +$$

CONSERVATION OF ENERGY

$$\begin{split} V_1 &= O & T_1 = O \\ V_2 &= -2mg \left(\frac{1}{2} \sin \Theta \right) = -mg L \sin \Theta \\ T_2 &= \frac{1}{2} \cos \tilde{\eta}_{RB}^2 + \frac{1}{2} I \omega_{RB}^2 + \frac{1}{2} m \tilde{\chi}_{RB}^2 + \frac{1}{2} I \omega_{RB}^2 \\ &= \frac{1}{2} \cos \left(\frac{L}{2} \omega \right)^2 + \frac{1}{2} \left(\frac{1}{12} m \tilde{L}^2 \right) \omega^2 + \frac{1}{2} \cos \left(\frac{EC}{2} \right)^2 + \frac{1}{2} \left(\frac{1}{12} m \tilde{L}^2 \right) \omega^2 \\ T_2 &= \left[\frac{1}{8} + \frac{1}{29} + \frac{1}{2} \left(\frac{EC}{2} \right)^2 + \frac{1}{24} \right] m \tilde{L}^2 \omega^2 \\ &= \frac{1}{24} \left[5 + 12 \left(\frac{EC}{2} \right)^2 \right] m \tilde{L}^2 \omega^2 \end{split}$$

T, +4, = T2 +42

$$0 + 0 = -mgL \sin \Theta + \frac{1}{24} \left[5 + 12 \left(\frac{EC}{L} \right)^2 \right] mL^2 \omega^4$$

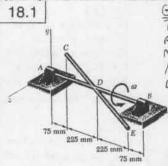
$$\omega = \left[\frac{249}{L} \cdot \frac{\sin \Theta}{5 + \left(\frac{EC}{L} \right)^2} \right]^{1/2}$$
(3)

VELOUTY OF D:
$$V_D = (co) \omega$$
 (4)

OUTLINE OF PROGRAMI

ENTER L= 30in = 2.5ft, g= 32.2ft/s2 PROGRAM, IN SEGUENCE, EGS. (1), (2), (3), AND (4) EVALUATE AND PRINT W AND VO FOR VALUET OF & FROM O TO 90° USING 10° INCREMENTS.

theta deg.	omega rad/s	vD . ft/s
0	0.0000	0.0000
10	2,4806	2.1537
20	3,1277	5.3487
30	3.3226	8.3066
40	3.3302	10.7031
50	3.2746	12.5423
60	3.2088	13.8945
70	3.1544	14.8210
80	3.1198	15.3622
90	3.1081	15.5403



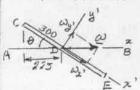
TWO UNIFORM RODS AB AND CE ARE WELDED AT MIDPOINTS D.

MASS OF EACH ROD = 1.5 kg LENGTH = 600 mm

ASSEMBLY HAS CONSTANT ANG. VEL W= 12 rad/s.

FIND: ANG MOMENTUM HD

SINCE ROD AB HAS MOM, OF INERTIA 20 ABOUT AXIS OF RUTHTION, ONLY ROD CE CONTRIBUTES TO ANGULAR MOMENTUM.



SINCE CD = 300 mm, $\cos \theta = \frac{225}{300}$ 0 = 41,41 USING THE PRINCIPAL CENTROIDAL AXES 2' 4' E, WE HAVE

$$\omega_{x} = \omega \cos \theta$$

$$\omega_{y} = \omega \sin \theta$$

$$\omega_{x} = 0$$

$$\bar{I}_{z} = 0$$

$$\bar{I}_{y} = \frac{1}{12} m \ell^{2}$$

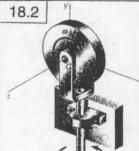
$$\bar{I}_{z} = \frac{1}{12} m \ell^{2}$$

$$H_z = \overline{I}_2 \omega_2 = 0$$

Hy, = Iy, W, = 1 mlwsind

= $\frac{1}{12}$ (1.5 kg)(0.6 m) (12 rad/s)sin41.41° = 0.357

H = 0.357kg·m3/5; 0= 48.6°, 0y = 41.4°, 0z = 90°



GIVEN:

THIN, HOMOGENEOUS DISK OF HASS M AND RADIUS & SPINS AT CONSTANT RATE WI. FORK-ENDED ROD SPINS AT CONSTANT RATE W2.

ANGULAR MOMENTUM H OF DISK.

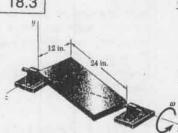
SINCE THE A y, & AXES ARE PRINCIPAL CENTROIDAL AXES, WE CAN USE EUS. (18.10) WITH Iz= Iy= 4m 22, Iz= = 1m2 Wx=0, Wy=Wz, Wz=W,

AND WRITE

$$\begin{aligned} H_{\chi} &= \bar{I} \omega_{\chi} = 0 \\ H_{\chi} &= \bar{I}_{\chi} \omega_{\chi} = \frac{1}{4} m t^{2} \omega_{\chi} \\ H_{z} &= \bar{I}_{z} \omega_{z} = \frac{1}{2} m t^{2} \omega_{1} \end{aligned}$$

$$H_6 = \frac{1}{4} m \dot{v} (\omega_z j + 2\omega, \underline{k})$$

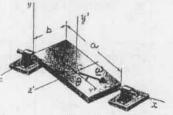
18.3



GIVEN!

RECTANGULAR PLATE SHOWN WEIGHS 18 16 AND ROTATES WITH CONSTANT W= 5 rad/s FIND:

ANGULAR MOMENTUH H ABOUT MASS CENTER G.



WE USE THE PRINCIPAL CENTROIDAL -XES Gx'y'=' WE HAVE W, = WCOSA Wa' = - wsind

I, = 1/2 111 b, I, = 1/2 m(a+b), I, = 1/2 ma

USING EQS. (18.10)

 $H_{x} = I_{x}, \omega_{x} = \frac{1}{12} mb^{2} \omega \cos \theta$

 $H_{y'} = \bar{I}_{y}, \omega_{y'} = 0$

 $H_{2'} = \vec{I}_2, \omega_2, = -\frac{1}{12} \text{ main sind}$

H = H2, L' + H3, 1 + H2, 1

WHERE L', j', K' ARE THE UNIT VECTORS ALONG THE x, y, Z' AXES.

Hg= 1mbwcosoi' - 1mazwsinok'

TO RETURN TO THE ORIGINAL 2, Y, & AXES, WE NOTE THAT

 $\underline{l}' = \underline{l} \cos \theta + \underline{k} \sin \theta$ k'=-i sind + Kcos H

THEREFORE

 $H_{G} = \frac{1}{12} m b^{\epsilon} \omega \left(as^{*0} i + cososinok \right) + \frac{1}{12} m a^{2} \omega \left(sin^{2} b i - sinows o k \right)$

 $\underline{H}_{c} = \frac{1}{n} m \omega \left[(a^{2} \sin^{2}\theta + b^{2} \cos^{2}\theta) \underline{i} - (a^{2} - b^{2}) \sin\theta \cos\theta \underline{k} \right]$

GIVEN DATA: 111 = (10/b) (32.2 ft/s') = 0.55901 16.5/ft

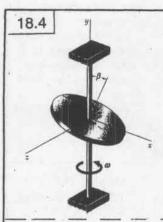
a = 24in. = 2 ft b = 12 in. = 1 ft tan 0 = = 0.5 0 = 26.565°

71+451 H = 1/2 (0.55901 /65/ft) w [(4 sin 26.565"+ cos 26,565") 1-(4-1) sin 26.565° cos 76.565° K7(ft)

H_=(0.046584 16.5 / ft) w (1.600 1. 1.200 E)(ft) HG = [(0.074534 16. ft. s2) i - (0.0 55901 16. fts 5) K] W (1) LETTING W=5 rad/s.

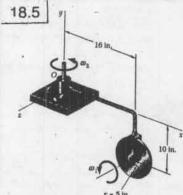
H6 = (0.372716.ft.s)i-(0.274516.ft.s)k (2)

H = (0.373 lb. ft.s) i - (0.280 lb. ft.s) K



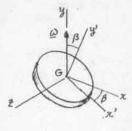
GIVEN:
HOMOGENEOUS DISK OF MASS OF AND RADIUS & MOUNTED ON SHAFT AB WITH $\beta = 25^{\circ}$.
SHAFT ROTATES WITH CONSTANT ω .

FIND: ANGLE & FORMED BY AB AND ANG. MOHENTUM H OF DISK ABOUT G.



GIVEN:
HOMOGENEOUS DISK OF
WEIGHT W = 8 LB
ROTATES AT CONSTANT
RATE $\omega_i = 12 \text{ rad/s}$.
ARM OA ROTATES AT
CONSTANT RATE $\omega_2 = 4 \text{ rad/s}$

FIND: ANGULAR MOMENTUM HA OF DISK ABOUT ITS CENTER A.



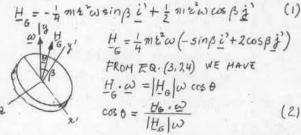
WE USE THE PRINCIPAL CENTROIDHL AXES GX & E.

WE HAVE: $\widetilde{I}_{x} = \widetilde{I}_{x} = \frac{1}{4} m z^{2}$ $\widetilde{I}_{y}, = \frac{1}{2} m z^{2}$ $\omega_{x} = -\omega \sin \beta$ $\omega_{y} = \omega \cos \beta$

USING EQS. (18.10): $H_{z} = \overline{I}_{z}, \ \omega_{z}, = -\frac{1}{4}mz^{2}\omega \sin\beta$ $H_{y} = \overline{I}_{y}, \ \omega_{y}, = \frac{1}{2}mz^{2}\omega \cos\beta$ $H_{z} = \overline{I}_{z} \ \omega_{z} = 0$

WE HAVE HG = H, L'+ Hz, j'+ Hz K

WHERE i', i', K ARE THE UNIT VECTORS ALONG THE X'Y' AXES.



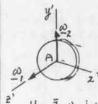
BUT $H_G \cdot \omega = \frac{1}{4} m t^2 \omega \left(-\sin \beta \vec{i} + 2\cos \beta \vec{j} \right) \cdot \omega \vec{j}$ OR, OBSERVING THAT $\vec{i} \cdot \vec{j} = -\sin \beta$ AND $\vec{j} \cdot \vec{j} = \cos \beta$, $H_G \cdot \omega = \frac{1}{4} m t^2 \omega^2 \left(\sin^2 \beta + 2\cos^2 \beta \right)$

 $= \frac{1}{4} m k^{2} \omega^{2} (1 + \cos^{2} \beta)$ ALSO $|H_{G}| \omega = \frac{1}{4} m k^{2} \omega^{2} \sqrt{\sin^{2} \beta + 4\cos^{2} \beta}$ (3)

 $= \frac{1}{4} m z^{2} (0^{2} \sqrt{1 + 3 \cos^{2}/3})$ (4)

SUBSTITUTING FROM (3) AND (4) INTO (2), $\cos \theta = \frac{1 + \cos \beta}{\sqrt{1 + 3} \cos^2 \beta}$

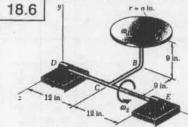
FOR \$ = 25°, cos 8 = 0.9786 \$ = 11.88°



WE USE PRINCIPAL CENTRUIDAL AXES AZ'y'z'. WE HAVE $\overline{I}_{x} = \overline{i}_{y} = \overline{i}_{y} m s^{2} \qquad \overline{I}_{z}, = \overline{i}_{z} m s^{2}$ $\omega_{z} = 0, \quad \omega_{y} = \omega_{z}, \quad \omega_{z} = \omega_{z}$ FROM EQS. (18.10):

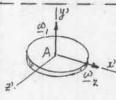
 $\begin{array}{ll} \dot{z}' & H = \bar{1}_{z}, \omega_{z}, \dot{i} + \bar{1}_{g}, \omega_{y}, \dot{j} + \bar{1}_{z}, \omega_{z}, \dot{k} = \frac{1}{4}mz^{2}(\omega_{z}\dot{j} + 2\omega_{z}\dot{k}) \\ GIVEN \ DATA: \ m = \frac{W}{g} = \frac{81b}{32.775} = 0.24845 \ lb.s'/ft \\ z = 5 \ in. = \frac{5}{12} \ ft, \ \omega_{1} = 12 \ rad/s, \ \omega_{2} = 4 \ rad/s \end{array}$

 $\frac{H_{A} = \frac{1}{4}(0.24845)(\frac{5}{12})^{2}[4j + 2(12)k]}{=(0.043133 /b.ft.s)j + (0.25880 /b.ft.s)k}$ $\frac{H_{A} = (0.0431 /b.ft.s)j + (0.259 /b.ft.s)k}{(0.259 /b.ft.s)k}$



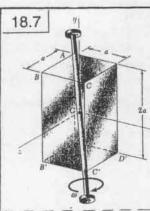
GIVEN:
HOMOGENEOU: DISK
OF WELGHT W= 616
ROTHES AT CONSTANT
RATE W_= 16 rad/s.
SHAFT DCE RETHIES
AT CONSTANT RATE
W_= 8 rad/s.

FIND: ANE, MINISHTH HAOF DISK ABOUT ITS CENTER A.



WE USE PRINCIPAL CENTROIDAL AXES AZYZZ, WE HAVE $\vec{I}_{x} = \vec{I}_{z} = \frac{1}{4}mz^{2}$, $\vec{I}_{y} = \frac{1}{2}mz^{2}$ $\omega_{z} = \omega_{z}$, $\omega_{y} = \omega_{1}$, $\omega_{z} = 0$ FROM EGS. (18.10):

 $\begin{array}{l} H_{A} = \overline{f}_{A} \omega_{A} \cdot i + \overline{f}_{A} \omega_{A} \cdot i + \overline{f}_{A} \omega_{A} \cdot k = \frac{1}{4} m_{A} \cdot (\omega_{A} \cdot i + 2\omega_{A} \cdot j) \\ \text{GIVEN DATA: } m_{1} = \frac{11}{3} = \frac{6b}{32271/52} = 0.186335 \ lb \cdot 5^{\circ}/fl \\ z = 8 \ in_{1} = \frac{2}{3} + t, \ \omega_{A} = 16 \ rad/s, \ \omega_{A} = 5 \ rad/s \\ H_{A} = \frac{1}{4} (0.186335) (\frac{2}{3})^{\circ} [8 \cdot k + 2(16) \cdot j] \\ = (0.165631 \ lb \cdot ft \cdot s) \cdot i + (0.66352 \ lb \cdot ft \cdot s) \cdot j \\ H_{A} = (0.1656 \ lb \cdot ft \cdot s) \cdot i + (0.663 \ lb \cdot ft \cdot s) \cdot j \end{array}$



SOLID RECTAI/ GULAR RICHL-LELE PIPED SHOULD, F 1/455 M. IT NOTH TES ABOUT 173 DIATIONAL AS' AT CONSTANT SINE W.

FIND:

(a) MAGNITUDE OF HUSTING

(b) ANGLE THAT HE FINE WITH AC'.

WE DENOTE BY \bar{I}_{x} , \bar{I}_{y} , \bar{I}_{z} THE PRINCIPAL CENTROLDAL MOMENTS OF INTERTIA. WE HAVE $\underline{\omega} = \omega \frac{-a\underline{i} + 2a\underline{j} - a\underline{k}}{a\sqrt{6}} = \frac{\omega}{\sqrt{6}} \left(-\underline{i} + 2\underline{j} - \underline{k} \right) \tag{1}$

 $\underline{H}_{\alpha} = \overline{J}_{\alpha} \omega_{\alpha} \underline{i} + \overline{I}_{y} \omega_{y} \underline{j} + \overline{I}_{z} \omega_{z} \underline{k} = \frac{\omega}{\sqrt{6}} \left(-\overline{I}_{z} \underline{i} + 2\overline{I}_{y} \underline{j} - \overline{I}_{z} \underline{k} \right)$

COMPUTATION OF THE MOMENTS OF INEXTIA: $\overline{I}_{z} = \overline{I}_{z} = \frac{1}{12} m (a^{2} + 4a^{2}) = \frac{5}{12} m a^{2}$ $\overline{I}_{y} = \frac{1}{12} m (a^{2} + a^{2}) = \frac{1}{5} m a^{2}$

SUBSTITUTE INTO (2);

 $\frac{H}{6} = \frac{\omega}{\sqrt{6}} \left(-\frac{5}{12} ma^{2} \dot{i} + \frac{2}{6} ma^{2} \dot{j} - \frac{5}{12} ma^{2} \dot{k} \right)$ $\underline{H} = \frac{ma^{2} \omega}{12\sqrt{6}} \left(-5 \dot{i} + 4 \dot{j} - 5 \dot{k} \right)$ (3)

(a) $|H_{G}| = \frac{ma^{2}\omega}{12\sqrt{6}}\sqrt{25+16+25} = ma^{2}\omega\frac{\sqrt{11}}{12}$ (4)

14 = 0,276 mais

(b) FRUM EQ. (3.24) WE HAVE $H_G \cdot \omega = |H_G| \omega \cos \theta$

$$\cos \theta = \frac{H_6 \cdot \omega}{|H_6| \omega} \tag{5}$$

RECALLING (1) AND (3):

 $\frac{H_{6} \cdot \omega = \frac{ma^{2}\omega}{12\sqrt{6}} \left(-5\underline{i} + 4\underline{j} - 5\underline{k}\right) \cdot \frac{\omega}{\sqrt{6}} \left(-\underline{i} + 2\underline{j} - \underline{k}\right)}{-\frac{ma^{2}\omega^{2}}{72} \left(5 + 8 + 5\right) = \frac{1}{4} ma^{2}\omega^{2}} \tag{6}$

RECALLING (4): $|\underline{H}_{G}|\omega = \frac{\sqrt{11}}{12} ma^{2}\omega^{2}$ (7)

SUBSTITUTING FROM (6) AND (7) INTO (5):

 $\cos \theta = \frac{1/4}{\sqrt{11}/12} = \frac{3}{\sqrt{11}} = 0.90453$ $\theta = 25.239$

 $\theta = 25.2^{\circ}$

18.8 GIVEN; SOLID PARALLELEPIPED OF PRIE.

18.7 IS REPLIACED BY HOLLOW ONE MADE OF 6 THIN METAL PLATES.

FIND: (a) MAGNITUDE OF ANG, MOMENTUM HG.

WE DENOTE BY I, I I THE PRINCIPAL CENTROIDAL MOMENTS OF INTERTIAL WE HAVE

 $\omega = \omega \frac{-a\underline{i} + 2a\underline{j} - a\underline{k}}{a\sqrt{6}} = \frac{\omega}{\sqrt{6}} \left(-\underline{i} + 2\underline{j} - \underline{k} \right) \tag{1}$

 $H_{G} = \overline{I}_{z} \omega_{z} \underline{i} + \overline{I}_{z} \omega_{z} \underline{i} + \overline{I}_{z} \omega_{z} \underline{k} = \frac{\omega}{\sqrt{6}} \left(-\overline{I}_{z} \underline{i} + 2\overline{I}_{z} \underline{j} - \overline{I}_{z} \underline{k} \right) (2)$

COMPUTATION OF MOMENTS OF INERTIA:

EACH OF THE TWO SQUARE PLATES HAS A MASS EQUAL TO M/ID AND EACH OF THE RECTANGULAR PLATES HAS A MASS EQUAL TO M/S. USING THE PARALLEL.

(1) AXIS THEOLEM WHEN WEELES, WE BETAIN:

1), Ī _z	Īg	Ī
SQUARE PLATES	$\frac{\frac{2m}{10}(\frac{a^2}{12} + a^2)}{= \frac{13}{60} ma^2}$	$\frac{2m}{10} \cdot \frac{a^2}{6} = \frac{ma^2}{30}$	13 ma
RECTANG. PLATES // yz PLANE	$\frac{2m}{5} \frac{a^2 + 4a^2}{12}$	$\frac{2m}{5} \left[\frac{a^t}{12} + \left(\frac{a}{2} \right)^t \right]$ $= \frac{2}{15} ma^2$	$\frac{2m}{5} \left[\frac{\alpha^2}{3} + \left(\frac{\alpha}{2} \right)^2 \right]$ $= \frac{7}{30} m\alpha^2$
REC. PL.// 24 PLANE	7 1	2 mat	1 mak
sums	37 ma	9 m at	37 mat 60

(3) SUBSTITUTE THE VALUES OBTAINED FOR I, I, I,

INTO (2):

$$\underline{H}_{6} = \frac{m \, a^{2} \, \omega}{60 \, \sqrt{6}} \left(-37 \, \underline{i} + 36 \, \underline{j} - 37 \, \underline{k}\right) \tag{3}$$

(a)
$$|H_6| = \frac{ma^2\omega}{60\sqrt{6}}\sqrt{(37)^2 + (36)^2 + (37)^2} = \frac{ma^2\omega}{60\sqrt{6}}\sqrt{4034}$$
 (4)

$$|\underline{H}_{G}| = 0.432157 \, \text{mad}, \quad |\underline{H}_{G}| = 0.432 \, \text{mad}$$

(b) WE RECALL EQ. (5) IN SOLUTION OF PROB. 18.7:

$$cce\theta = \frac{16 \cdot \omega}{116 \cdot \omega}$$
 (5)

RECALLING (1) AND (3) ABOVE :

$$\frac{H}{6} \cdot \omega = \frac{ma^2\omega}{60\%} \left(-37\underline{i} + 36\underline{j} - 37\underline{k} \right) \cdot \frac{\omega}{16} \left(-\underline{i} + 2\underline{j} - \underline{k} \right) \\
= \frac{ma^2\omega^2}{360} \left(37 + 72 + 37 \right) = \frac{146}{360} \, ma^2\omega^2 \quad (6)$$

RECALLING (4) A POVE:

$$|H_{6}|\omega = \frac{\sqrt{4034}}{60\sqrt{6}} m a^{2} \omega^{2}$$
 (7)

SUBSTITUTING FROM (6) AND (7) INTO (5):

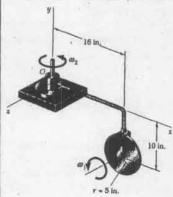
$$\cos \theta = \frac{146}{360} \frac{60 \, \text{V}_{6}}{\sqrt{4034}} = \frac{146}{\sqrt{6 \times 4034}} = 0.95845$$

$$\theta = 20.208^{\circ}$$

0=20.2°

18.9 GIVEN; DISK OF PROB. 18.5 WITH W=816, $\omega_1 = 12 \text{ rad/s}$, AND $\omega_2 = 4 \text{ rad/s}$.

FIND: ANGULAR MOMENTUM HO ABOUT POINT O.



WE USE EQ. (18.11): $\frac{H_0}{E} = \frac{\overline{E} \times m\overline{U}}{E} + \frac{H_0}{E} \qquad (1)$ WHERE $\overline{\underline{z}} = \underline{\underline{v}}_{A} = (\frac{16}{12}ft)\underline{i} - (\frac{10}{12}ft)\underline{j}$ $\overline{\underline{z}} = \underline{\underline{v}}_{A} = (\frac{4}{3}ft)\underline{i} - (\frac{5}{6}ft)\underline{j}$ $\frac{111}{2} = \frac{W}{3} = \frac{816}{32.765}.$ $= 0.24845 |\underline{b} \cdot \underline{s}^2/ft$ $\overline{\underline{U}} = \underline{\underline{U}}_{A} = \underline{\underline{U}}_{A} \times \underline{\underline{v}}_{A}$ $= (4rad/s)\underline{\underline{j}} \times (\frac{4}{3}\underline{i} - \frac{5}{6}\underline{\underline{j}})$ $\overline{\underline{U}} = -(\frac{16}{3}ft/s)\underline{\underline{k}}$

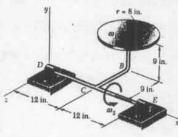
TROM THE SOLUTION OF PROB. 18.5, WE RECALL THAT $H_G = \frac{H}{A} = (0.0431 \text{ bift-3}) + (0.259 \text{ bift-5}) + (0.2$

 $\frac{H}{o} = \left(\frac{4}{3}i - \frac{5}{6}i\right) \times 0.24845 \left(-\frac{16}{3}k\right) + 0.043ij + 0.259k$ = 1.7668j + 1.1042i + 0.043ij + 0.259k

HD = (1.104 16.ft.s) i + (1.810 16.ft.s) j + (0.259 16.ft.s) k

18.10 GIVEN: DISK OF PROB. 18.6 WITH W = 616, $\omega_1 = 16$ rad/s, AND $\omega_2 = 8$ rad/s.

FIND: ANGULAR MOMENTUM HO ABOUT POINT D.



WE USE EQ. (18.11) WITH RESPECT TO D: $H = E \times m \bar{U} + H_G$ (1) WHERE $\bar{z} = E_A = (1ft)L + (0.75ft) + (0.75ft) + (0.75ft) = (0.75ft) = 0.186335 |b.5^2/ft|$

 $\overline{\mathcal{D}} = \mathcal{V}_{A} = \mathcal{W}_{2} \times \mathcal{Z}_{A} = (8 \operatorname{rad/s}) \underline{i} \times (\underline{i} + 0.75 \underline{j} - 0.75 \underline{k})$ $\overline{\mathcal{D}} = (6 + t/s) \underline{j} + (6 + t/s) \underline{k}$

FROM THE SOLUTION OF PROB. 18.6, WE RECALL THAT HG = HA = (0.1656 16.ft.s) + (0.663 16.ft.s) &

SUBSTITUTING INTO (1);

 $H_{D} = (i + 0.75 j - 0.75 k) \times 0.186335 (6j + 6k) + 0.1656 i + 0.663 j$

= 1.1180 k -1.1180 j +0.8385 i +0.8385 i +0.1656 i +0.663 j

H = (1.843 16.44.5) i - (0.455 16.41.5) j + (1.11816.41.5) k

18.11



FROJECTILE WITH M=30 kg

\$\bar{k}_{\pi} = 60 \text{mm}, \$\bar{k}_{\pi} = 250 \text{mm}, \$\\
RNGLE \text{B} = 5°, ANG. MOM. \$\\
\begin{array}{c} \text{H} = (320 g \cdot m^2/s) \bar{\text{L}} - (9g \cdot m^2/s) \bar{\text{L}} & \\
RESOLVE & RESOLVE & NATO COMPONENTS

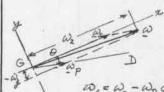
(a) ALONG GX (RATE OF SPIN) (b) ALONG GD (RATE OF PRECESSION)

BECHUSE OF AXISPIMETRY OF PROJECTILE, THE 2

AND Y AXES ARE PRINCIPAL CENTROIDAL AXES, $\vec{I}_z = m\vec{k}_z^2 = (30 \text{ kg})(0.060 \text{ m})^2 = 0.108 \text{ kg} \cdot m^2$ $\vec{I}_y = m\vec{k}_z^2 = (30 \text{ kg})(0.250 \text{ m})^2 = 1.875 \text{ kg} \cdot m^2$ GIVEN: $H_z = 0.320 \text{ kg} \cdot m^2/s$, $H_y = -0.009 \text{ kg} \cdot m^2/s$ PROM EOS. (18.10): $\omega_z = \frac{H_z}{I} = \frac{0.320 \text{ kg} \cdot m^2/s}{0.108 \text{ kg} \cdot m^2} = 2.9630 \text{ rad/s}$

 $\omega_{y} = \frac{H_{b}}{\bar{I}_{g}} = \frac{-0.009 \, kg \cdot m^{2}/s}{1.875 \, kg \cdot m^{2}} = -0.00480 \, \text{rad/s}$

THUS: \(\omega = (2,9630 rad/s) \overline{i} - (0.00480 rad/s) \overline{j} \)
WE MUST NOW RESOLUE (INTO OFLIQUE COMPONENTS ALONG GZ AND GD.



WE NOTE THAT $-\omega_y = \omega_p \sin \theta$ $\omega_p = \frac{-\omega_y}{\sin \theta} = \frac{+0.00480}{\sin 5^{\circ}}$ $\omega_p = 0.055074 \text{ rad/s}$

 $\omega_s = \omega_z - \omega_p \cos \theta = 2.9630 - 0.055074 \cos 5$ = 2.408 rad/s

ANSWERS: (a) w= 2.91 rad/s. (b) wp=0.0551 rad/s

18.12 GIVEN: PROJECTILE OF PROP. 18.11.
ADDITIONAL DATA: \$\vec{v} = 650 m/s.

FIND: ANG. MOM. HA. (RESOLVE INTO X, Y, & COMP.)

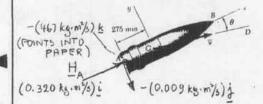
RESOLVE \$\overline{V}\$ INTO RECTANS, COMP. ALONG ZANDY AXES. \$\overline{V} = (650 m/s)(0055° \overline{L} - 5in5° \overline{L}) = (647.53 m/s)\overline{L} - (56.65 m/s)\overline{L}\$

USING EQ. (18.11) AND RECALLING DATA FROM PRUB, 18.11,

 $\frac{H}{A} = \overline{e} \times m \overline{U} + \underline{H}_{G} \\
= (0.275 \text{ m}) \underline{i} \times (30 \text{ kg}) [(647.58 \text{ m/s}) \underline{i} - (56.65 \text{ m/s}) \underline{j} + (0.320 \text{ kg} \cdot \text{m}^2/\text{s}) \underline{i} - (0.009 \text{ kg} \cdot \text{m}^2/\text{s}) \underline{j}$

= -(467.57 kg·m³/s)k +(0.320 kg·m³/s)i-(0.009 kg·m³/s)j

H= (0,320 kg·m/s)i-(0.009 kg·m/s)j-(467kg·m/s)k



18.13

(a) Show that the angular momentum H_H of a rigid body about point B can be obtained by adding to the angular momentum HA of that body about point A the vector product of the vector rAB drawn from B to A and the linear momentum mv of the body:

$$\mathbf{H}_{H} = \mathbf{H}_{A} + \mathbf{r}_{A/B} \times m\widetilde{\mathbf{v}}$$

(b) Further show that when a rigid body rotates about a fixed axis, its angular momentum is the same about any two points A and B located on the fixed axis $(H_A = H_B)$ if, and only if, the mass center G of the body is located on the fixed axis.

(a) USING EQ. (18.11) TO DETERMINE H AND THEN

$$\underline{H}_{A} = \underline{L}_{G/A} \times m \, \underline{v} + \underline{H}_{G} \tag{1}$$

$$\underline{H}_{B} = \underline{t}_{6/B} \times m \, \underline{t} + \underline{H}_{C} \tag{2}$$

SUBTRACTING (1) FRUM(2)

A
$$\frac{2G/A}{B}$$
 G $H_B - H_A = \left(\frac{E}{G/B} - \frac{E}{G/A}\right)^{\frac{1}{2}}$ $H_A = \left(\frac{E}{G/B} - \frac{E}{G/A}\right)^{\frac{1}{2}}$ $H_A = \left(\frac{E}{G/B} - \frac{E}{G/A}\right)^{\frac{1}{2}}$ $H_A = \frac{E}{G/A}$ $H_A = \frac{E}{G/A}$

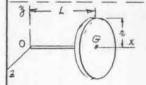
(b) IT FILLOWS FRIM EN. (3) THAT HA = HR IF AND CIVILY IF. EA/axmV=0 (4)

BUT, DENOTING BY AND THE UNIT VECTOR ALONG
THE FIXED AXIS, WE HAVE $\overline{\Psi} = \omega_A^2/8 \times \frac{1}{2}$ EU (4) YIELDS X m (W] A/B X EG/A)

WE NOTE THAT ENE IS PERPENDICULAR TO WARE TO AND, THUS, NOT PARALLEL TO IT. THEREFORE, THIS SECOND VECTOR MUST BE ZERD, WHICH WILL OCCUR IF EG/A IS PARALLEL TO 2AB, THAT IS IF, AND ONLY IF, G IS LUCATED ON AB.

18.14 GIVEN: DISK OF SAMPLE PROB. 18.2 AND ANSWERS TO PART & OF THAT PROBLEM:

FIND: ANG. MOMENTUM HOUSING EQ. (18.11), AND VERIFY THAT RESULT IS SAME AS IN PART LOFS, P. 10.2

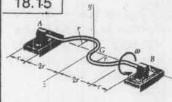


EQ. (18.11): H = Exmo + H. = Lixmrw, k+ 1 mrw (1-21)

H = - mr [w, j + 1 mr w, i - 1 mr w, e = 1 m 2 w, i-mL & w, j - 4 m 2 2 w, j H = 1/2 m 2 w, i - m (1 + 1/4 22)(2W1/1) j

WHICH IS THE ANSWER OBTAINED IN PART b OF SAMFLE PROB. 18.2.

18.15



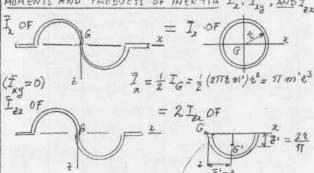
GIVEN:

SHAFT OF MASS M, MADE OF KOD OF UNIFORH CROSS SECTION ROTHTE WITH CONSTANT ANG. VEL. W.

(A) ANG. MOM. H., (b) ANGLE FORMED BY H AND AXIS AB.

NASC LAK (1817 : FIGTH = M) = am = t + T/2 + T/2 + Z = E(T+1)2

MOMENTS AND PRODUCTS OF INTERTIA I2. IN AND I2



THUS: I2 = 2 (11) TE) 21 21 = 2 11/12 (2)(22) = 4 m1 x3

(a) ANGULAR MOMENTUM HA

WE USE ERS. (18.7) SINCE OJEW, OJE D, WEHAVE H,= I, w = 17m'20 (1)

$$H_y = -\overline{1}_{xy} \omega = 0$$

$$H_z = -I_{zx} \omega = -4m^2 t^3 \omega \tag{2}$$

$$H_G = H_z i + H_z k = m' \epsilon^3 \omega (\pi i - 4k)$$

OR, RECALLING THE EXPRESSION OBTAINED FOR m';

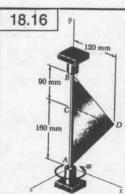
$$\frac{H}{G} = \frac{m e^{2} \omega}{2(\pi + i)^{2}} (\pi i - 4 \underline{k}) =
= m e^{2} \omega \left[\frac{\pi}{2(\pi + i)} - \frac{2}{\pi + i} \underline{k} \right]
\underline{H} = m e^{2} \omega \left(0.379 i - 0.483 \underline{k} \right)$$

(b) ANGLE FOUNTED EY H, AND AXIS AL.

DENOTING BY & THAT ANGLE, WE HAVE $tan \theta = \frac{|H_2|}{H_1}$

AND, RECALLING (1) AND (2):

$$tan\theta = \frac{4m^2 \xi^3 \omega}{\pi \epsilon_0^3 \kappa^3 \omega} = \frac{4}{\pi} \qquad \theta = 51, 9$$



TRIANGULAR PLATE SHOWN HAS MASS m = 7.5 kg AND IS WELDED TO SHAFT AB. PLATE RUTATES AT CONSTANT RATE W= 12 rad/s.

FINDI

(a) ANG HOMENTUM HO (b) ANG. MOMENTUM HA (FIND I AND USE PROPERTY INDICATED IN PROB. 18,13a.)

D. 12 m 0,09 to 0.16

(a) WE DIVIDE PLATE INTO TWO RIGHT TRIANGLES AND COMPUTE THEIR PRODUCTS OF INERTIA. 11 = 1 (7,5 kg) = 2,7 kg $m_z = \frac{16}{27} (7.5 \text{ kg}) = 4.8 \text{ kg}$

FROM SAMPLE PROB. 9.6, WE RECALL

Ixy, MASS = m/2 bh (1/24 bh) = mbh

TRIANGLE 1: (Iz, y) = 1/2 (2.7 kg/0.12 m/0.09 m) = 2.43 x10 TRIANGLE 2: (Iziy) = 1/2 (4.8 kg)(0.12 m)(-0.16 m)=-7.68 ×10 THUS, FOR THE PLATE, IZy=(0.43-7.68) 10=-5.25×10 gm Izin =-5,25 ginz

WE NOTE THAT Iye = 0. MOMENT OF INTETTA T OF ENTIRE PLATE. Iy, MEER = 1 bh3, Iy, MASS = 1 bh (12 bh5) = 6 mh2 Iy = 1 (7.5 kg) (0.12 m) = 0.018 kg·n = 18g·n

ANGULAR MOMENTUM H

WE USE EQS. (18.13) TO DETAIN THE COMPONENTS Hx, Hy, Hz, OF Hc

Hz = - Iz 4 = - (-5,25 g.m2)(12 rod/s) = + 63.0 g.m2/s $H_g = I_g \omega = (18 \text{ g·m}^2)(12 \text{ rad/s}) = 216 \text{ g·m}^2/\text{s}$ H2, = - Ig2, W= 0

 $H_c = (63.0 g \cdot m/s) i + (216 g \cdot m/s) j$

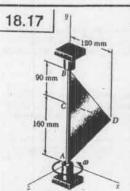
(b) ANGULAR MOMENTUM HA. WE APPLY THE EQUATION GIVEN IN PART a OF PROB. 18.13 TO POINTS A AND C.

Ha=Hc+2c/AXMU

WHERE 15C/A- (O.16 m) j. NOTING THAT THE DISTANCE FROM THE AXIS OF ROTATION AB TO THE MASS CENTER G OF THE PLATE 15 = = 1 (O,IZ m) = 0.04 M, WE HAVE m v = m (w x E) = (7,5 kg)(12 rad/s)j x (0,04m) i

=-(3.60 kg·m/s)k = -(3600g·m/s)kECIA X m = (0.16 m) j x (-3600 g.m/s) k = - (576 g.m/s) i SUBSTITUTING FOR HC AND EGIAXMID INTO (1):

HA = - (513g·m3/s) i + (216g·m3/s) j



GIVEN

TRIANGULAR PLAJE SHOWN HAS MASS M = 7.5 kg AND IS WELDED TO SHAFT AR. PLATE, RATATES AT CONSTANT KHT= W = 12 rad/s.

(a) ANG. MOMENTUM H (6) ANG. MOMENTUN HA (FIND IT AND USE PRIPERTY INDICATED IN PROS. 18.18 a.)

(a) SEE PART a OF SOLUTION OF FROB. 18.16. WE FIND

ANG. MOMENTUM H = (63,0g.m'/s) 1 + (216g.m'/s) }

(6) ANG. MOMESTINA H.

WE APPLY THE EQUATION SINEN IN FIRT & OF PRIB. 18.13 TO POINTS B AND C .

(1) Ha= Hc++G/8×mo

WHERE \$ C/B = - (0.09 m) &.
NOTING THAT THE DISTANCE FLOOR THE AXIS OF ROTHITION AE TO THE HASS CENTS & OF THE PLATE IS

E= + (0.12m) = 0.04m

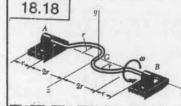
WE HAVE

mv=m(w×=)=(7.5kg)(12 md/s)jx (0.04m)i = - (3.60 kg·m/s) k = - (3600 g·m/s) k

* c/8 × m1 = - (0.09 m) j × (-3600 g·m/s) k = + (324 g. m/s) i

SUBSTITUTING FOR H AND EC/exmit INTO (1);

HB=(387g.nt/s)i+(216g.nt/s)+



GIVEN:

SHAFT OF PROB. 18.15

ANG. MOM, OF SHAFT

(a) ABOUT A

WE FIRST DETERMINE Ha. SEE SOLUTION OF AREA 18.15. WE FOUND H=mtw (0,379 1-0,483 K)

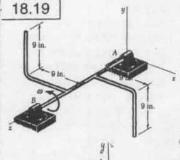
FROH EQ. (18.11) WE HAVE

HA = EGAXMVIH H= 1c/ex 11 + ++

BUT T=0 SINCE G IS LOCATED ON AXIS AB. THUS!

(a) AND (b): H=H=H=H=1112W(0.371i-0.183 k)

NOTE . THE RESULT CETAINED VERIFIES THE PROPERTY INDICATED IN PEOB. 18.13 b. NAMELY, THAT IF THE MASS CENTER & OF A BUTH RUTHTHIS ABOUT A FIRE TO A IS LOCATED ON THE ANIS, THE ANGULAR MONEUTIME IS THE SPITE AROUT ANY THE FOURTS OF THEIR MY IS.



GIVEN!
TWO L-SHAPED PHIS,
EACH WEIGHING 516,
HRE WEIGHING 516,
HRE WEIGHING 516,
THE 27-10, SHAPE 52,
LOTTING 1 MINISTERS
TOTAL 1 MINISTERS
TOTAL 1 MINISTERS
TOTAL 1 (a) HA
(b) ANGLE FAMED BY HA
ALL AB.

MOMENTS AND PRODUCTS OF INTRIA

FOR EACH NUMBERED ELEMENT: a = 9 in = 0.75 ft. $m = \frac{1}{2} (5.16)/9 = 2.5/9$

FOR 1 AND 4: I = I + md = 1 ma + m(a + a) = 4 ma

FOR 2 AND 3: $I_2 = \frac{1}{3}ma^4$ FOR ASSEMBLY: $I_2 = 2\left(\frac{4}{3}ma^2 + \frac{1}{3}ma^4\right)$ $I_3 = \frac{10}{3}ma^2$ PRODUCTS OF INTERTIA OF ASSEMBLY:

$$\begin{split} I_{22} &= (I_{x2})_1 + (I_{x4})_2 + (I_{x4})_3 + (I_{xx})_4 \\ &= \pi_1 \left(-a \right) (2a) + \pi_1 \left(-\frac{a}{z} \right) (2a) + \pi_1 \left(\frac{1}{z} \right) a + \pi_1 (a) a = -\frac{g}{z} \pi_1 a^2 \\ I_{x2} &= \pi_1 \left(\frac{a}{z} \right) (2a) + 0 + 0 + \pi_1 \left(-\frac{a}{z} \right) a = \frac{1}{z} m a^2 \end{split}$$

(a) ANGULAR MOHENTUM ABOUT A

WE USE FOL (18.13) to OBTHIN THE COMPONENTS OF #4.

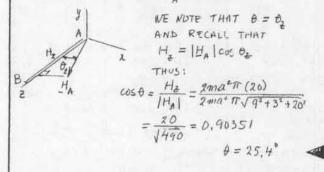
WE HAVE W2 = \$\omega = 360 \text{ rpm} = 6(277) = 12 17 \text{ rod/s}, \omega_2 = \omega_3 = 0

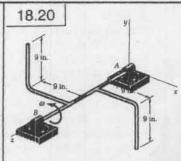
$$\begin{aligned} H_z &= -I_{x2} \omega_z : + \frac{3}{2} ma^2 (12\pi) = 18 \, ma^2 \pi \\ H_y &= -I_{y2} \omega_z : - \frac{1}{2} ma^2 (12\pi) = -6 ma^2 \pi \\ H_z &= I_z \omega_z : \frac{10}{2} ma^2 (12\pi) = 40 \, ma^2 \pi \end{aligned}$$

THUS: $\frac{H}{A} = \frac{H_2 i + H_3 i + H_2 k}{32.2} + \frac{1}{2} k = 2 ma \pi (9 i - 3 i + 20 k)$ = $2 \left(\frac{2.5}{32.2} \frac{5}{16} \cdot \frac{5}{16} + \frac{1}{2} \left(\frac{5}{16} \cdot \frac{5}{16} + \frac{1}{2} \cdot \frac{5}{16} + \frac{1}{2} \cdot \frac{5}{16} \right) + \frac{1}{2} \cdot \frac{5}{16} + \frac{1}{2}$

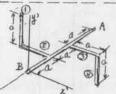
HA = (2.47 16. ft.s) i - (0.823 16. ft.s) j + (5,49 16. ft.s) k

(b) ANGLE & FORMED BY HA AND AB





GIVEN:
TWO L'SHAPED ARMS,
EACH WEIGHING 516,
ARE WELDED AT THE
ONE-THIRD POINTS OF
THE 27-IN SHAFT AB.
ASSEMBLY RUTATES AT
CONSTANT 360-FPM RATE.
FIND: (a) H
(b) ANGLE PORHED BY H
B
AND BA.



WE WILL USE AXES 2,3, & WITH

ORIGIN AT B.

MOMENTS AND PRODUCTS OF INEKTIH

FOR EACH NUMBERED ELEMENT: a = 9in = 0.75 ft $m = \frac{1}{2}(516)/g = 2.5/g$

FOR 1 AND 4; $I_z = \overline{I} + md^2 = \frac{1}{12}ma^2 + m\left(\frac{a}{4} + a^2\right) = \frac{4}{3}ma^2$ FOR 2 AND 3: $I_z = \frac{1}{3}ma^4$ FOR ASSENBLY: $I_z = 2\left(\frac{4}{3}ma^4 + \frac{1}{3}ma^4\right)$ $I_z = \frac{10}{3}ma^4$ PRODUCTS OF INTERTIA OF ASSEMBLY:

$$\begin{split} \mathbf{I}_{\mathbf{x}^{1}\overline{\mathbf{c}}} &= \left(\mathbf{I}_{\mathbf{x}^{1}\overline{\mathbf{c}}}\right)_{1} + \left(\mathbf{I}_{\mathbf{x}^{1}\overline{\mathbf{c}}}\right)_{1} + \left(\mathbf{I}_{\mathbf{x}^{1}\overline{\mathbf{c}}}\right)_{3} + \left(\mathbf{I}_{\mathbf{x}^{1}\overline{\mathbf{c}}}\right)_{4} \\ &= \sin(-a)(-a) + m\left(-\frac{a}{c}\right)(-a) + m\left(\frac{a}{c}\right)(-2a) + m(a)(-2a) = \frac{3}{2} mo^{2} \\ \mathbf{I}_{\mathbf{y}^{1}\overline{\mathbf{c}}} &= m\left(\frac{a}{2}\right)(-a) + 0 + 0 + m\left(-\frac{a}{c}\right)(-2a) = \frac{1}{2} ma^{2} \end{split}$$

() ANGULAR MOMENTUM ABOUT B

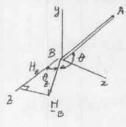
WE USE EWS. (18.13) TO OBTAIN THE COMPONENTS OF HB. WE HAVE $\omega_1 = \omega_2 = 360 \, \text{rpn} = 6(2\, \Omega) = 12 \, \text{Trad/s}, \ \omega_2 = \omega_3 = 0.$ $H_{z} = -I_{z,z} \ \omega_z = +\frac{3}{2} \, \text{ma}^2(12\, \Omega) = 18 \, \text{ma}^2 \Omega$ $H_{y,z} = -I_{y,z} \ \omega_z = -\frac{1}{2} \, \text{ma}^2(12\, \Omega) = -6 \, \text{ma}^2 \Omega$ $H_{z} = I_{z,z} \ \omega_z = \frac{10}{3} \, \text{ma}^2(12\, \Omega) = 40 \, \text{ma}^2 \Omega$

THUS: $H = H, \dot{b} + H, \dot{d} + H, \dot{k} = 2 ma^{2} \pi (9i - 3 \dot{d} + 20 \dot{k})$ = $2(\frac{2.5}{32.2} /b \cdot 5/ft)(0.75 ft)^{2} (\pi m db)(9\dot{b} - 3 \dot{d} + 20 \dot{k})$ = $(0.2744/b \cdot ft \cdot s)(9\dot{b} - 3\dot{d} + 20 \dot{k})$

H = (2,47 16. ft.s) i - (0.823 16. ft.s) j + (5.49 16. ft.s)k

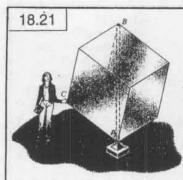
NOTE. THIS IS THE SAME ANSWERTHAT WAS OBTAINED FOR HA
IN PRUB. 18, 19, THIS COULD HAVE BEEN ANTICIPATED, SINCE
THE MASS CENTER & OF THE ASSEMBLY LIES ON THE FIXED
AXIS AB (CF. PROB. 18.13.6),

(b) ANGLE & FORMED BY HB AND BA



WE NOTE THAT $\theta = \overline{\Pi} - \theta_2$ AND RECALL THAT $H_2 = |H_B| \cos \theta_2$ THUS: $\cos \theta = \cos (\overline{\Pi} - \theta_2) = -\cos \theta_2$ $= -\frac{H_2}{|H_B|} = \frac{2m\alpha^2 \overline{\Pi} (2\alpha)}{2m\alpha^4 \overline{\Pi} \sqrt{q^2 + 2^2 + 2\Omega^4}}$

 $= -\frac{20}{\sqrt{490}} = -0.90351$ $\theta = 154.6^{\circ}$



HOLLOW CUBE CONSISTS OF SIX 5×5ft ALUMINUM SHEETS AND CAN POTE IS ABIDITY THTICAL DIAGONAL AB, STUDENT PUSHES CORNER C FOR 1,25 IN DIRECTION PERPENDICULAR TO PLANE ABC WITH FORCE OF 12,5 lb, CAUSING CUBE TO COMPLETE 1 REK IN 55. FIND: WEIGHT OF CUBE.

HINT: PERP. DISTANCE FLOOR C TO AB IS a 12/3. WHERE a 15 SIDE OF CUBE.



FOR CUBE, IAB = I SINCE THE ELLIBOID OF INFRITA AT G IS A SPHERE (SEC. 9. 17). FOR THE TWO HORIZONTAL FACES $(I_{DD'})_{H} = 2(\frac{m}{6})(\frac{a^{2}}{6}) = \frac{ma^{2}}{19}$ WHERE M = MASS OF CUBE

FOR THE FOUR VERTICAL PACES $\left(I_{pp}\right)_{y} = 4\left(\frac{m}{6}\right)\left[\frac{a}{12}+\left(\frac{a}{2}\right)^{2}\right] =$

FOR THE WHULE CUBE!

$$I_{AB} = I_{DD}, = (I_{DD})_{\mu} + (I_{DD})_{\nu} = \frac{ma^2}{1B} + \frac{2ma^4}{q} = \frac{5}{1B} ma^2$$

IMPULSE- MOMENTUM PRINCIPLE

ANG, IMPULSE ABOUT AB = FINAL ANG, MOMENTUM ABOUT AB (F Dt) a V2/3 = 5 ma2 W

GIVEN DATA: F= 12,5 16, Dt = 1,25, a= 5 ft, w= 20 rad SUBSTITUTE DATA AND m= W/g INTO (1):

 $(12.5 \text{ lb})(1.2 \text{ s})(5 \text{ ft})\sqrt{\frac{2}{3}} = \frac{5}{18} \frac{W}{32,2 \text{ ft/s}} (5 \text{ ft})^{t} (\frac{2 \text{ ft. rad}}{5 \text{ s}})$

SOLVING FOR W: W = 225,96 16

W= 226 16

18.22 GIVEN: ALUMINUM CUBE OF PROB. 18.21 16 REPLACED BY CUBE CONSISTING OF SIX

PLYWOOD SHEETS, WELGHING 20 16 EACH. STUDENT PUSHES CORNER C AS IN PROB. 18,21 (FOR 1,25 WITH 12,5-15 FORCE).

FIND: TIME REQUIRED FOR CUBE TO COMPLETE 1 REV.

SEE SOLUTION OF PROB. 18,21 FOR DERIVATION OF (Fat) a 12/3 = 5 ma w (1)

GIVEN DATA: F = 12.5 16, At = 1.25, a = 5 ft $m = \frac{W}{g} = \frac{6(201b)}{32,2 \text{ ft/s}^2} = 3.727 \text{ lb. s}^2/\text{ft}$

SUBSTITUTE DATA INTO (1): $(12.5 \text{ lb})(1.2 \text{ s})(5 \text{ ft})\sqrt{\frac{2}{3}} = \frac{5}{18}(3.727 \text{ lb} \cdot \text{s}^2/\text{ft})(5 \text{ ft})^* \omega$

SOLVING FOR W: W = 2.366 5 $C = \frac{2\pi}{10} = \frac{2\pi}{2.3665} = 2.65565$

2 = 2,665

GIVEN: 18.23 C= 180 mm F At

TWO CIRCULAR PLATES

EACH OF MASS M= 4 Kg, ARE RIGIDLY CONNECTED TO KOD AB OF NEGLIGIBLE MASS AND SUSPENDED AS SHOWN, AN IMPULSE

> FAt = - (2,4 N·s)k IS APPLIED AT D.

FIND: (a) VELOCITY DOF MASS CENTER G. (b) ANGULAR VELOCITY W OF ASSEMBLY.

COMPUTATION OF MOMENTS AND PRODUCTS OF INERTIA FOR UPPER PLATE: $I = \bar{I}_{x} + m d^{2} = m \left(\frac{1}{4} \xi^{2} + d^{2} \right) = (4 kg) \left[\frac{1}{4} (0.18 m)^{2} + (0.15 m)^{2} \right]$

Iy = Iy, + me" = 1 me" + me = 3 me" = 3 (4 kg)(0.18m)] = 0,1944 Kg·m"

 $I_z = \overline{I}_2 + m(\xi^z + d^2) = \frac{1}{4}m\xi^2 + m(\xi^2 + d^2) = m(\frac{5}{4}\xi^2 + d^2)$ =(4kg)[{2(0,18m)}+(0,15m)]=0,252kg·m"

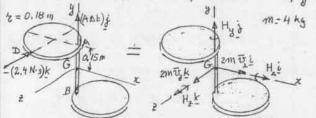
 $I_{xy} = m(-t)(d) = -mt d = -(4kg)(0.18m)(0.15m)$ $I_{xy} = -0.108 kg \cdot m^{2}$, $I_{xz} = 0$, $I_{zz} = 0$

FOR LOWER PLATE: WE OBTAIN THE SAME RESULTS THUS, FOR ASSEMBLY; WE DUUBLE RESULTS FOR UPPER

I = 0.2448 kg·m², I = 0.3888 kg·m², I = 0.504 kg·m² Izy=-0.216 kg·m2, Iy= 0, Izx=0

IMPULSE-MUMENTUM PRINCIPLE

WE NOTE THAT THE IMPULSIVE FORCES ARE F AND, POSSIBLY, THE FORCE AT A. ALSO, FROM CONSTRAINTS, U=Q



(a) VELUCITY OF MASS CENTER , EQUATE SUMS OF - (2,4N.5) k+ (ADF) = 2(4kg) (V2++ V3 k) VECTURS:

THUS: ALT =0, V=0, V=-0.3 m/s

V = - (0.300 m/s) k (L) ANGULAR VELOCITY, EQUATE SUMS OF MOMENTS ABOUT G:

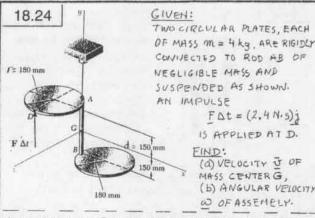
[(-0.10m)i+(0.15m)j]×(-2,4N·s)k= Hi +H,j+H,j+ -(0,432 kg·m²)j-(0,360 kg·m²)i= Hz i+ Hjj++jk $H_{s} = -0.360$, $H_{s} = -0.432$, $H_{s} = 0$ SUBSTITUTE FROM (1) AND (2) INTO EUS, (18.7):

 $H_{1} = \overline{1} \omega_{1} - \overline{1} \omega_{1} - \overline{1} \omega_{2} : -0.360 = +0.2448 \omega_{1} + 0.216 \omega_{2}$ $H_{2} = -\overline{1} \omega_{1} + \overline{1} \omega_{1} - \overline{1} \omega_{2} : -0.432 = +0.216 \omega_{1} + 0.3888 \omega_{2}$ $H_{3} = -\overline{1} \omega_{1} - \overline{1} \omega_{2} - \overline{1} \omega_{2} : 0 = \omega_{2}$ $0 = \omega_{2}$ (4)

SOLVE (3) AND (4): W= -0.96154, Wy = -0.57692

W = - (0.962 rad/s)2-(0.577 rad/s) }

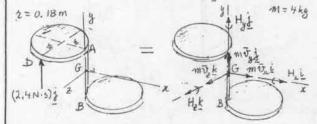
15)



COMPUTATION OF MOMENTS AND PRODUCTS OF INERTIA SEE SOLUTION OF PROB. 18.23 WHERE WE FOUND I = 0.2448 kg·m, I = 0.3888 kg·m, I = 0.504 kg·m In = - 0.216 kg·11, In= 0, In

IMPULSE- MOMENTUM PRINCIPLE

WE NOTE THAT THE CORD AT A WILL BECOME SLACK THUS, THE ONLY IMPULSIVE FORCE IS F.



(a) VELOCITY OF MASS CENTER

EQUATE SUMS OF VECTORSI

(2.4 N.s) = 2 (4 kg) (vz + vy + + vz +)

THUS: V=0, U=0.300 m/s, V2=0 v = (0,300 m/s) g

(b) ANGULAG /ELD=17Y

EQUATE SUMS OF MOMENTS ABOUT G:

[(-0.18 m) + (0.18 m) k] x (2.4 N/s) = H2 + H3 j+H2 k (0.432 kg·m²)(-k-i) = H2 (+ Hy i + H2k

THUS: Hz = - 0.432 kg·m², Hy = 0, Hz = -0.432 kg·m² (2) (a) VELOCITY OF MASS CENTER SUBSTITUTE FROM (1, AND (2) INTO ERS. (18.7):

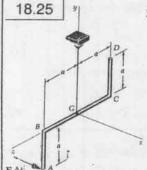
H, = I, 0) - I, 6) - I, 0) - 1, 0) : -0,432 = +0.2448 W, +0.216 Wy

Hy =- I, Wx + I, Wy - Iy, Wy. 0 = +0.216 W, +0.3888W Hz = - Ix = Wx - Iz = Wy + Iz Wz: -0.432 = +0.504 Wz

SOLVING (3) AND (4) SIMULTANEOUSLY,

Wz = -3,4616 rad/s, Wy = 1,9231 rad/s SULVING (5) FOR W2: W2 = -0.8571 rad/s THOS:

W = - (3.46 rad/s) i + (1.923 rad/s)j - (0.857 rad/s) K €



GIVEN:

UNIFORM BENT ROD OF MASS M IS SUSPENDED AS SHOWN. ROD IS IT AT A WITH IMPLIESE F At IN DIRECTION PERPENDICUAR TO PLANE CONTAINING ROD.

IMMEDIATELY AFTER IMPACT (a) VELOCITY I OF MASS CENTER

(b) ANGULAR VELOCITY W

COMPUTATION OF MOMENTS AND PRODUCTS OF INERTIA

PORTION BC: $(I_y)_{BC} = (I_y)_{BC} = \frac{1}{12} (\frac{m}{z})(2a) = \frac{1}{6} ma$, $(I)_{BC} = 0$ (Izy)BC = (Izz)BC = (Izz)BC = 0

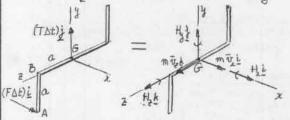
PORTIONS AB AND CD:

 $(1_{z_{AB}} = (1_{z_{CD}} = \overline{1} + (\frac{m}{4})d^2 = \frac{1}{2}(\frac{m}{4})a + \frac{m}{4}(a + \frac{a}{4}) = \frac{1}{3}ma$ $(I_z)_{AB} = (I_z)_{CD} = \frac{m}{V} a_z^2, \quad (I_z)_{AB} = (I_z)_{CD} = \frac{1}{3} \frac{m}{4} a^2 = \frac{1}{12} ma^2$ $(I_{xy})_{AB} = (I_{xy})_{CD} = 0, (I_{xz})_{AB} = (I_{xz})_{CD} = 0, (I_{zz})_{AB} = (I_{yz})_{CD} = -\frac{mo}{4}(0)\frac{u}{2})z - \frac{mo}{8}$

THE MOMENTS AND PRODUCTS OF INERTIA OF THE ROD ARE OBTAINED BY ADDING THE ABOVE VALUES:

$$\begin{split}
\bar{I}_{2} &= \frac{1}{6}ma^{2} + \frac{1}{3}ma^{2} + \frac{1}{3}ma^{2} = \frac{5}{6}ma^{4} \\
\bar{I}_{3} &= \frac{1}{6}ma^{4} + \frac{1}{4}ma^{4} + \frac{1}{4}ma^{4} = \frac{2}{3}ma^{4} \\
\bar{I}_{2} &= 0 + \frac{1}{12}ma^{2} + \frac{1}{12}ma^{4} = \frac{1}{6}ma^{4} = -\frac{1}{4}ma^{4}, \quad \bar{I}_{2} &= 0 \\
\bar{I}_{2y} &= 0, \quad \bar{I}_{32} &= 0 - \frac{1}{6}ma^{4} - \frac{1}{6}ma^{4} = -\frac{1}{4}ma^{4}, \quad \bar{I}_{2} &= 0 \\
IMPULSE-MOMENTUM PRINCIPLE
\end{split}$$

THE IMPULSES CONSIST OF FAt = (FAt) AND, POSSIBLY, AN IMPULSE (TABJAT G. BECAUSE OF CONSTRAINTS, To = 0.



EQUATE SUNIS OF VECTORS: (FAt) i+(Tot) j=mvi+mv k THUS: $\overline{v}_z = (F\Delta t)/m$, $\overline{v}_z = 0$, $T\Delta t = 0$ $\underline{\overline{v}} = (F\Delta t/m)\underline{i}$

(4) (b) ANGULAR VELOCITY

FOUNTE MOMENTS ABOUT G: (-aj +ak) x (FDt) i = H i + H j + H t a Fat (k+j) = Hz i + Hyj + Hz k

THUS: Hz=0, Hy=aFAE, Hz=aFAE SUBSTITUTE FROM (1) AND (2) INTO EQS. (18.7):

(3) Hz= I Wz- I Wy- I Wz: 0= = Ma W.

(2)

(1) Hy=-I, Wa+ I, Wy-I, W: aFat= = maily + f maily (5)

Hz=-Ix Wx-Ix Wy+IWz: aFbt=4 mawy+ 1 mawz (CONTINUED)

18.25 continued

WE REPEAT THE FOLLOWING ERS. :

W2=0 (3)

aFAt = = many + 4ma wz

(4) (5) a Fot = 1 ma wy + 2 ma wz

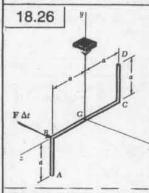
SOLVING (4) AND (5) SIMULTANEOUSLY, WE OBTAIN

$$\omega_y = -\frac{12}{7} \frac{F \Delta t}{ma}$$

$$\omega_{\pm} = \frac{60}{7} \frac{F\Delta t}{ma}$$

THUS:

 $\omega = (12 F\Delta t / 7 ma)(-\frac{1}{2} + 5 K)$



GIVENI

UNIFORM BENT ROD OF MASS M IS SUSPENDED AS SHOWN. ROD IS HIT AT B WITH IMPULSE FAT IN DIRECTION PERPENDIC-ULAR TO PLANE CONTAINING ROD.

FIND:

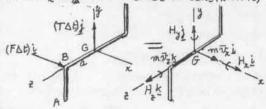
IMMEDIATELY AFTER IMPACT (a) VELUCITY OF MASS CENTER

(b) ANGULAR VELOCITY W

MOMENTS AND PRODUCTS OF INERTIA

SEE SOLUTION OF PROB. 18.25. WE OBTAINED]= {ma,]= = ma,]= = = ma,]= - + ma,]= = = 0 IMPULSE- MOMENTUM PRINCIPLE

THE IMPULSES CONSIST OF FAT = (FAT) + HAD, POSSIBLY, AN IMPULSE (TOE) AT G. BECAUSE OF CONSTRAINTS, 1/4 = 0.



(a) VELOCITY OF HASS CENTER

EQUATE SUMS OF VECTORS: (FDE) + (TDE) = mil + mil +

THUS: V= (FAt)/m, V= 0, TAt =0

U= (FAt/m) i

(6) ANGULAR VELOCITY

EQUATE MOMENTS ABOUT G:

akx (FAt) = Hi+ Hi+ H, j+ H, K

(aFAb) = Hz 1 + Hy j + Hz K

THUS: Hz = 0, Hx = aFAt, Hz = 0

SUBSTITUTE FROM (1) AND(2) INTO EQS. (18.7):

(3) H= Iwx-Ixwy-Izw: D=5 maw

(4) Hy=-Inw+Inwy-Inzw: aFat==mawy-4mawz

(5) Hz = - Izwz - Izwy + Izwz; 0 = - # mawy + 7 maw

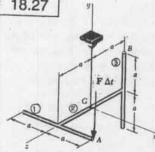
SOLVING (4) AND (5) SIMULTANEOUSLY, WE OBTAIN

 $W_2 = -\frac{36}{7}$

THUS:

w=(12 Fat/7ma)(2j-3K)





GIVEN:

THREE RODS, EACH OF MASS M AND LENGTH 20 ARE WELDED TO FORM ASSEMBLY. ASSEMBLY IS HIT VERTICALLY AT A AS SHOWN.

IMMEDIATELY APTER IMPACT (a) VELOCITY OF MASS CENTER (B) ANGULAR VELOCITY & OF ASSEMBLY,

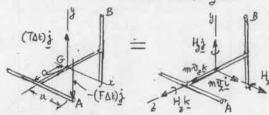
COMPUTATION OF MOMENTS AND PRODUCTS OF INERTIA $I_z = (I_z)_1 + (I_z)_2 + (I_z)_3 = ma^2 + \frac{1}{3}ma^4 + m(a^4 + \frac{a^4}{3}) = \frac{8}{3}ma^4$

(1) $I_{a} = (I_{a})_{1} + (I_{a})_{2} + (I_{a})_{3} = \frac{1}{3} m a^{2} + 0 + \frac{1}{3} m a^{4} = \frac{2}{3} m a^{4}$

In = 0, In = 0, In = 0

IMPULSE- MOMENTUM PRINCIPLE

THE IMPULSES CONSIST OF - (FAT) A ARCIED AT A AND (TAT) APPLIED AT G. BECAUSE OF CONSTRAINTS, Ty = 0 .



(a) VELOCITY OF MASS CENTER

EQUATE SUMS OF VECTORS: (TAL) - (FAt) = m TL + m T. K THUS: TAL = FAt, Uz=0, Uz=0. SINCE TEO FROM ABOVE,

(b) ANEULAR VELOCITY

EQUATE MOMENTS ABOUT G: (ai+ak) x (-Fat) j = H, i + H, j + H, k - (aFAt) k + (aFAt) i = Hci+Hyj+Hzk

THUS: H = aFat, H = 0, H = - aFat (2)

SINCE THE THREE PRODUCTS OF INTERTIA ARE ZERO, THE X, Y, AND & AXES ARE PRINCIPAL CENTROID AL AXES AND WE CAN USE EGS. (18. 10), SUBSTITUTING FROM (1) AND (2) INTO THESE EQUATIONS, WE HAVE

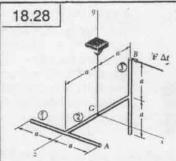
$$H_2 = \bar{I}_2 \omega_2$$
: $\alpha F \Delta t = \frac{8}{3} m \alpha^2 \omega_2 = \frac{3}{5} F \Delta t / 8 m \alpha$ (3)

$$H_y = \overline{I}_{\omega_y} : 0 = \frac{\omega}{3} m a^* \omega_y \qquad \omega_y = 0$$
 (4)

$$H_2 = \vec{I}_3 \omega_2$$
: $-a F \Delta t = \frac{2}{3} m d$ $\omega_2 = -3 F \Delta t / 2 m a$ (5)

4(1)

THEREFORE!



THREE RODS, EACH OF MASS ME AND LENGTH 20 FAI ARE WELDED TO FORM ASSEMBLY, WHICH IS HIT AT B IN DIRECTION OPPOSITE TO X AXIS. FIND:

IMMEDIATELY AFTER INDACT (a) VELOCITY OF MASS CENTER. (b) ANGULAR VELOCITY W.

COMPUTATION OF MOMENTS AND PRODUCTS OF INTERTIA

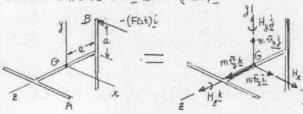
$$\begin{split} \vec{J}_{z} &= (\vec{I}_{z})_{1} + (\vec{I}_{z})_{2} + (\vec{I}_{z})_{3} = ma^{2} + \frac{1}{3}ma^{2} + m(a^{2} + \frac{a^{2}}{3}) = \frac{6}{3}ma^{4} \\ \vec{T}_{z} &= (\vec{I}_{z})_{1} + (\vec{I}_{z})_{2} + (\vec{I}_{z})_{3} = m(a^{2} + \frac{a^{2}}{3}) + \frac{1}{3}ma^{2} + ma^{3} = \frac{6}{3}ma^{4} \\ \vec{I}_{z} &= (\vec{I}_{z})_{1} + (\vec{I}_{z})_{1} + (\vec{I}_{z})_{1} = \frac{1}{3}ma^{4} + 0 + \frac{1}{3}ma^{6} = \frac{2}{3}ma^{4} \end{split}$$

$$\vec{I}_{zy} &= \vec{I}_{dz} = \vec{I}_{zz} = 0$$

$$(1)$$

IMPULSE - MONENTON PRINCIPLE

THE ONLY IMPULSE IS FAT = - (FAT) i.



(a) VEINC TO DEMASS CENTER

EQUATE STON TO VEIT SELL INTER

- (FOT) = 007 1 1 107 1 - 7 1 4 5=0, 5=0 --- (FΔt/m)i THUS: == Fatim,

(b) ANGBLAR - DOTY

EQUATE MEMENTS ABOUT 6: (aj-ak) x (-Fot) = Hz i + Hg + Hz K (aFot)k+(aFot)= 11, 1+ Hyj+ +2 \$

THUS: H=0, Hy=aFat, H= aFat

SINCE THE THEFE THUS IN THE PRINTER THE ZER .. THE Z, J. AND E AXES ARE PRINCIPAL CENTROLINE AXES AND WE CAN USE EQS. (18.10), SUBSTITUTING FROM (1) AND (2) INTO THESE FOUNTIONS, WE HAVE

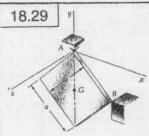
$$H_{\lambda} = \overline{I} \omega_{\lambda}$$
: $O = \frac{1}{2} m \alpha^{\dagger} \omega_{\lambda}$ $\omega_{\lambda} = 0$ (3)

$$H_2 = \overline{I}_3 \omega_a$$
; $\alpha F \Delta t = \frac{\theta}{3} m \alpha^4 \omega_3$ $\omega_3 = 3 F \Delta t / 8 m \alpha$ (4)

W2 = 3FAE/2ma H2= IW2: aFAt= = ma' W2

THEREFORE

ω= (3FΔt/87.a)(j+4k)



GIVEN:

SQUARE PLATE OF MASS M SUPPORTED BY BALL AND SUCKET WITH ANGULAR VELOCITY WAS WHEN IT STRIKES DESTINOTION AT B IN XX PLANE (0 = 0),

IMMEDIATELY AFTER IMPACT (a) ANE. VELOCITY CE PUTTE. (6) VELOCITY OF G.

ANGULAR MOMENTUM

BECAUSE OF SYMMETRY OF SQUARE PLATE, I IS THE SAME ABOUT ANY AXIS THROUGH & WITHIN XI PEANE. (CF. SEC. 9.17), I = 1/2 ma., IT FOLLOWS THAT H = 1/2 ma. (1) FOR ANY W.

VELOCITIES AFTER IMPACT SINCE E = O, CONNER B REMAINS IN CONTRACT WATER OTHERS ABOUT AL 2BA = - cos 45° i + sin 45° j = (- i + i)/V2 ω'=ω' 2BA = ω'(-++)/12 $\vec{v}' = \omega' \times \vec{z} = \left[\omega'\left(-\underline{i} + \underline{i}_{1/1}, \underline{i}_{1/2}\right) - \underline{i}\right) = \frac{1}{2}\omega'a k$

IMPULSE-MORIENTUM PRINCIPLE



EQUATING NUMERTS ABOUT LINE BAS H cos 450+0=HG+ >Ba (Exmin,) RECALLING (1), (2), (3), AND VALUE OF 2BA. 12 maw cos 45° = 12 maw + [(-i+1)/12].[-aix1nwak] $\frac{\omega_0}{12\sqrt{2}} = \omega^3 \left(\frac{1}{12} + \frac{1}{4} \right)$ (4)

(a) ANGULHR VELUCITY FROM (2) AND(4): 4 = - 1+2 (3) W= = W (-1+1)

(b) VELOCITY OF G. FROM (3) AND (4):

1) = 1 W Wo ak = 0.08839 Wak v=0.0884 wak

GIVEN: IMPACT DESCRIBED IN PROB. 18,29. FIND: IMPULSE ON PLATE AT (a) B, (b) A.

SEE SOLUTION OF PROB. 18.29 FOR IMPULSE - MUMENTUM DIAGRAM AND DETERMINATION OF W' AND T'. (a) EQUATING MOMENTS ABOUT A:

1/2 mat wo j + a (i-j) x Batk = 1/2 mat w' - a j x m v' SUBSTITUTING FOR W'AND T'AND PERFORMING PRIDUCTS 1/2 matu j - a Bat(j+i) = 1/2 mat 1 w (-i+j)-a xm wak = ma wo (- 16 + 16 2 - 16 4)

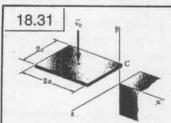
EQUATING THE COEFF. OF L: - a BAt = - 7 ma w

BAL = 0, 10312 miw. @ Bat = 0,1031 mwak

(b) EQUATING SUMS OF VECTORS:

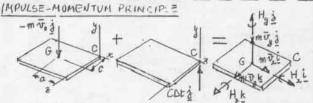
AAt + BAt = my, Abt = my'- Bbt = m(0.08839 W. ak) - 0.10312 mwak

ADE=- 0.01473 MWO ak



RECTANGULAR PLATE OF MASS M FALLING WITH IT AND NO ANG VELUCITY STRIKES OBSTRUCTION (E=0).

ANG, VELUCITY OF PLATE



z comp: $\vec{v}_{z} = 0$ z comp: \vec{v}_{z}

EQUATING MOMENTS ABOUT C: (-ai+ck) x (-m voj) = (-ai+ch) x m voj + Hzi+Hzi+Hzi+Hzi m vo (ak+cl) = -m vo (ak+ci) + Hzi+Hzi+Hzi+Hzi+Hzi SINCE e = 0, PLATE ROTATES ABOUT C IMMEDIATELY AFTER IMPACT

EWATE COEFF. OF UNIT VECTURES: W= 0 = WC = WC + W2 a) + Wak

EQUATE COEFF. OF UNIT VECTORS: $\omega_{j} = 0$, $\overline{\psi} = -(\omega_{z}C + \omega_{z}a)$ (4) USE EUS. (18.10):

$$H_{x} = \overline{I}_{x} \omega_{x} = \frac{1}{12} m(2c)^{2} \omega_{x} = \frac{1}{3} m c^{2} \omega_{x}$$

$$H_{y} = \overline{I}_{x} \omega_{y} = 0 \quad [BECAUSE OF(4)]$$

$$H_{z} = \overline{I}_{z} \omega_{x} = \frac{1}{12} m(2a)^{2} \omega_{x} = \frac{1}{3} m a^{2} \omega_{x}$$
(5)

SUBSTITUTE FROM (4) AND (5) INTO (3): $m\vec{v}_{0}(a\underline{k}+c\underline{i}) = +m(\omega_{x}c+\omega_{x}a)(a\underline{k}+c\underline{i}) + \frac{1}{3}mc^{2}\omega_{x} + \frac{1}{3}ma^{2}\omega_{x} + \frac{1}{3}ma^{2}\omega_{x} + mac\omega_{x}) + \frac{1}{3}mc^{2}\omega_{x} + mac\omega_{x} + \frac{1}{3}mc^{2}\omega_{x} + mac\omega_{x} + \frac{1}{3}mc^{2}\omega_{x} + mac\omega_{x} + \frac{1}{3}mc^{2}\omega_{x} + \frac{1}{3}mc^{2$

DIVIDE BY M AND EQUATE COEFF. OF UNIT VECTORS!

$$\frac{4}{3}c^{2}\omega_{\chi} + ac\,\omega_{z} = \overline{v}_{0}\,c \tag{6}$$

$$ac\,\omega_{\chi} + \frac{4}{3}\,a^{2}\omega_{\chi} = \overline{v}_{0}\,a \tag{7}$$

SOLVE (6) AND (7) SIMULTANEOUSLY:

$$\omega_2 = 3\bar{v_o}/7c$$
, $\omega_z = 3\bar{v_o}/7a$ $\omega = \frac{3}{7}\bar{v_o}(\frac{1}{c}i + \frac{1}{a}k)$

18.32 GIVEN: IMPACT DESCRIBED IN PROB. 18.31

(a) VELUCITY OF & IMMEDIATELY AFTER IMPACT, (b) IMPULSE ON PLATE DURING IMPACT.

(R) FROM SOLUTION OF PRUB. 19.3 1: EQS. (I) AND(4): $\vec{U} = \vec{U}_{3} \dot{j} = -\left(\omega_{x} c + \omega_{z} a\right) \dot{j}$ FROM ANSWER TO PROB. 18.31: $\omega_{z} = \frac{3\vec{V}_{0}}{7c}$, $\omega_{z} = \frac{3\vec{V}_{0}}{7a}$ THUS: $\vec{U} = -\left(\frac{3\vec{V}_{0}}{7} + \frac{3\vec{V}_{0}}{7}\right) \dot{j}$ $\vec{V} = -\frac{6}{7}\vec{V}_{0} \dot{j}$

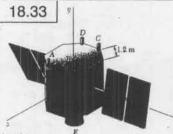
(b) PROM IMPULSE-MOMENTUM DIAGRAM OF PRUB. 18.31; EQUATING SUMS OF VECTORS:

$$-m \vec{v}_{0} \vec{j} + C \Delta t \vec{j} = m \vec{v}$$

$$C \Delta t \vec{j} = m \vec{v} + m \vec{v}_{0} \vec{j} = -\frac{2}{7} m \vec{v}_{0} \vec{j} + m \vec{v}_{0} \vec{j}$$

$$= \frac{1}{7} m \vec{v}_{0} \vec{j}$$

$$C \Delta t = \frac{1}{7} m \vec{v}_{0} \vec{j}$$



GIVEN: PROBE WITH

m= 2500 kg, k= 0.98 m

kg= 1.06 m, kz = 1.02 m;

500-N MAIN THE MICE E;

20-N THRUSTERS A,B,C,D

CAN EXPEL FUEL IN y DIR.

PROBE HAS ANG, VELOCITY

W= (0.040 rad/s) k

FIND:

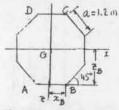
(a) WHICH TWO THRUSTERS THE DE REDUCE OF TO REDUCE OF TO ZERO,
(b) OPERATING TIME OF THESE THRUSTERS.

CO HOW LONG SHOULD E BE ACTIVATED IF \$ 15 TO FE UNCHINES

INITIAL ANGULAR MOMENTUM

$$\begin{split} H_{G} &= \bar{I}_{2} \omega_{1} \, \underline{i} + \bar{I}_{3} \omega_{3} \, \underline{i} + \bar{I}_{2} \omega_{2} \, \underline{k} = m \left(k_{x}^{2} \, \omega_{3} \, \underline{i} + k_{y}^{2} \, \omega_{3} \, \underline{i} + k_{z}^{2} \, \omega_{3} \, \underline{i} \right) \\ &= \left(2500 \, \text{kg} \right) \left[0.98 \, n_{3}^{12} \left(0.04 \, \text{rad/s} \right) \, \underline{i} + 0 + \left(1.02 \, \text{m} \right)^{2} \left(0.06 \, \text{reV/s} \, \underline{k} \right) \right] \\ &= \left(96.04 \, \text{kg} \cdot n_{3}^{2} / s \right) \, \underline{i} + \left(156.06 \, \text{kg} \cdot n_{3}^{2} / s \right) \, \underline{k} \end{split} \tag{1}$$

ANGULAR IMPULSE OF TWO 20-N THEUSTERS



LET US ASSUME THAT THE STESS A AND B WILL BE USED.

FROM GEOMETRY OF OCTOGON, $X_B = \frac{1}{2}A = \frac{1}{2}(1.2 \text{ m}) = 0.6 \text{ m}$ $R_B = \frac{1}{2}A + a \sin 4S^2 = 1.2071a$ = 1.44855 m $X_A = -X_B$ $R_A = R_B$

ANG. HAPULSE ABOUT $G = \underbrace{\epsilon}_{A} \times (-F \wedge t_{A})_{\dot{f}} + \underbrace{\epsilon}_{B} \times (-F \wedge t_{B})_{\dot{g}}$ $= (-x_{B} \cdot + z_{B} \cdot k) \times (-F \wedge t_{A})_{\dot{g}} + (z_{B} \cdot + z_{B} \cdot k) \times (-F \wedge t_{B})_{\dot{g}}$ $= x_{B} (F \wedge t_{A} - F \wedge t_{B})_{\dot{g}} + z_{B} (F \wedge t_{A} + F \wedge t_{B})_{\dot{g}}$

= (0.6 m)(FAt, -FAt,) E+(1.44853m)(FAt, + FAt,) i (2)

IMPULSE-MOIYENTUM PRINCIPLE

SINCE THE PINAL ANGULAR VELOCITY AND, THUS, THE FINAL ANGULAR MUNICUTUM MUST BE ZERO, THE SUM OF (1) AND (2) MUST BE ZERO. EQUATIONS THE COEFF. OF L AND K TO ZERO: (1.44853 m)(FDta+FDta)+96.04 kg·m/5 = 0 (0.6m)(FDta-FDta) + 156.06 kg·m/5 - 0

OR
$$F\Delta t_{A} + F\Delta t_{B} = -66.302 \text{ N·s}$$
 (3)
 $F\Delta t_{A} - F\Delta t_{B} = -260.1 \text{ N·s}$ (4)

SOLVING (3) AND (4) SIMULTANEOUSLY :

FAT = -163.20 N.S FATB = 96.90 N.S
THE FACT THAT FAT O INDICATES THAT THE DIAGONALLY
OPPOSITE THRUSTER SHOULD BE USED INSTEAD OF A THUS

(a) THRUSTERS BAND C

(b)
$$F\Delta t_B = 96.90 \cdot N.s$$
, $\Delta t_B = \frac{96.90 N.s}{20 N} = 4.84 s$
 $F\Delta t_C = 163.20 N.s$, $\Delta t_C = \frac{163.20 N.s}{20 N} = 8.16 s$

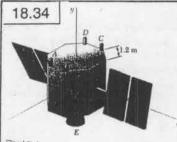
(C) IF THE VELOCITY \$\frac{1}{2}\$ OF THE MASS CENTER 15 TO BE UNCHANGED, THE RESULTANT OF THE LINEAR IMPULSES MUST BE ZERO.

$$-(F\Delta t_B)_{\frac{1}{2}} - (F\Delta t_C)_{\frac{1}{2}} + (500 \text{ N}) \Delta t_E_{\frac{1}{2}} = 0$$

$$-96.90 \text{ N} \cdot \text{S} - 163.20 \text{ N} \cdot \text{S} + (500 \text{ N}) \Delta t_E = 0$$

$$\Delta t_{\rm E} = \frac{260,1 \, \text{N} \cdot 5}{500 \, \text{N}} = 0.5202 \, \text{s}$$

At=0.520s



GIVEN: PROBE WITH

M = 2500 kg, Kz = 0.98 m,

kg = 1.06 m, Kz = 1.02 m;

500-N MAIN THRUSTER E;

20-N THRUSTERS A, B, C,D

CAN EXPEL PUEL IN J. DIR.

PROBE HAS ANG. VELOCITY

\(\Omega = (0.060 \text{ md/s}) \)

\(\omega = (0.040 \text{ ma/s}) \) k

FIND:

(a) WHICH TWO THRUSTERS SHOULD BE USED TO REDUCE CO TO ZERO

(b) OPERATING TIME OF THESE THRUSTERS,

(c) HOW LUNG SHOULD E BE ACTIVATED IF IT IS TO BE UNCHANGET

INITIAL ANGULAR MOMENTUM

$$\begin{split} & \stackrel{H}{G} = \stackrel{T}{I} \omega_{2} \stackrel{i}{L} + \stackrel{T}{I}_{g} \omega_{g} \stackrel{i}{L} + \stackrel{T}{I}_{g} \omega_{g} \stackrel{k}{L} = \pi I (k_{2}^{2} \omega_{3}, i + k_{2}^{2} \omega_{3} \stackrel{i}{d} + k_{2}^{2} \omega_{3} \stackrel{k}{L}) \\ &= (8500 \text{ kg}) [(0.98 \text{ m})^{4} (0.06 \text{ md/s}) \stackrel{i}{L} + 0 + (1.02 \text{ m})^{2} (-0.04 \text{ md/s}) \stackrel{k}{L}] \\ &= (144.06 \text{ kg} \cdot \text{m}^{2}/\text{s}) \stackrel{i}{L} - (104.04 \text{ kg} \cdot \text{m}^{2}/\text{s}) \stackrel{k}{L} \end{split}$$

ANGULIR INTRUISE OF TWO 20-N THRUSTERS

SEE SOLUTION OF PROB. 18, 33. ASSUMING THAT THEUSTERS A AND B ARE USED, WE FOUND

ANG, IMPULSE ABOUT G

= (0.6 m)(Fata-Fate) K+ (1.44853 m)(Fata+Fate) i (2)

IMPULSE-MOMENTUM PRINCIPLE

SINCE THE FINAL ANG. VELOCITY AND, THUS, THE FINAL ANG. NOMENTUM MUST BE ZERO, THE SUM OF (1) AND (2) MUST BE ZERO. EQUATING THE COEFF. OF i AND K TO ZERO, (1.44853 m) (FATA+FATA) + 144.06 kg·m²/s = 0

(0,6 m)(FAtA-FATB)-104.04 Kg. m/3=0

OR
$$F\Delta t_{A} + F\Delta t_{B} = -99.453 \text{ N.s}$$
 (3)

 $F \Delta r_A - F \Delta t_B = 173.40 \text{ N·s} \tag{4}$

SOLVING (3) AND (4) SIMULTANEOUS Y

$$F\Delta t_{A} = 36.974 \text{ N·s}$$
 $F\Delta t_{B} = -136.43 \text{ N·s}$

THE FACT THAT $F\Delta t_B < 0$ INDICATES THAT THE THRUSTER D, WHICH IS DIAGONIALLY CPRESITE TO B SHIVLD HE USED INSTEAD OF B. THUS:

(a) THRUSTERS A AND D

(6)
$$F\Delta t_{A} = 36.974 \text{ Ms}, \quad \Delta t_{A} = \frac{36.974 \text{ M/s}}{20 \text{ N}} = 1.84873$$

$$F\Delta t_{D} = /36.43 \text{ N/s}, \quad \Delta t_{D} = \frac{/36.43 \text{ N/s}}{20 \text{ N}} = 6.8215$$

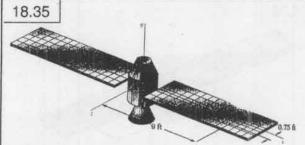
$$\Delta t_{A} = 1.8495; \quad \Delta t_{D} = 6.825$$

(C) IF THE VELUCITY TO OF THE MASS LENTER IS TO BE UNCHANGED, THE RESULTANT OF THE LINEAR IMPULSES MUST BE ZERO.

$$-(F\Delta t_{A})\underline{j} - (F\Delta t_{D})\underline{j} + (500N)\Delta t_{E}\underline{j} = 0$$

$$-36.974 \text{ N·S} - 136.43 \text{ N·S} + (500N)\Delta t_{E} = 0$$

$$\Delta t_E = \frac{173.40 \text{ N.3}}{500 \text{ N}} = 0.34685$$
 $\Delta t_E = 0.3475$



GIVEN:

PROBE WITH PRINCIPAL CENTRUIDAL AXES 2, 3, 4, AND W= 3000 lb, k= 1.375 ft, kg = 1.425 ft, ka = 1.250 ft. PROBE HAS NO ANG VELOCITY WHEN STRUCK ATA BY 5-02 METERRITE WITH VELOCITY RELATIVE TO PROBE

v = (2400 ft/s)i - (3000 ft/s)j + (3200 ft/s)k

METEORITE EMERGES ON OTHER SIDE OF PANEL MOVING.
IN SAME DIRECTION WITH SPEED REDUCED BY 20%
PIND: FINAL ANGULAR VELOCITY OF PROBE.

ANGULAR MOMENTUM OF METEORITE ABOUT G.

$$(H_{6})_{M} = \frac{16}{4} \times m_{M} \cdot \frac{1}{20}$$

$$= \left[(9+t) \cdot \cdot + (0.75+t) \cdot \cdot \cdot \right] \times \frac{(5/16) \, lb}{32.2 \, fr / 5} \cdot \left[(2400 \, fr / 5) \cdot \cdot - 3000 \cdot \frac{1}{3} + 3200 \cdot \frac{1}{3} \right]$$

$$= (9.705 \times 10^{-3} \, lb \cdot 5 / 5 \cdot l) \left(-27 \cdot k - 28.8 \cdot j + 1.8 \cdot j + 2.25 \cdot l \right) \left(10^{3} \, f \cdot l / 5 \right)$$

$$= (9.705 \, lb \cdot f \cdot t \cdot s) \left(2.25 \cdot i - 27 \cdot j - 27 \cdot k \right)$$

$$(H_{6})_{M} = (21.836 \, lb \cdot f \cdot s) \left(i - 12 \cdot j - 12 \cdot k \right)$$

$$(1)$$

FINAL ANGULAR MOMENTUM OF PROBE

$$(\underline{H}_{G})_{p} = \overline{\underline{I}}_{2} \omega_{x} \underline{i} + \overline{\underline{I}}_{y} \omega_{y} \underline{i} + \overline{\underline{I}}_{z} \omega_{z} = m(k_{z}^{2} \omega_{x} \underline{i} + k_{y}^{2} \omega_{y} \underline{j} + k_{z}^{2} \omega_{z} \underline{k})$$

$$= \frac{3000 \text{ lb}}{32.2 \text{ Pl}_{B}} \underline{[(1.375 \text{ ft})^{2} \omega_{x} \underline{i} + (1.425 \text{ ft})^{2} \omega_{y} \underline{j} + (1.250 \text{ ft})^{2} \omega_{z} \underline{k}]$$

$$= (176.15 \text{ lb.ft.s}^{2}) \omega_{x} \underline{i} + (189.19 \text{ lb.ft.s}^{2}) \omega_{y} \underline{j} + (1.250 \text{ ft})^{2} \omega_{z} \underline{k}$$

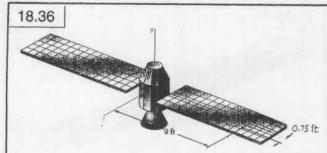
$$(145.57 \text{ lb.ft.s}^{2}) \omega_{z} \underline{k} \qquad (2)$$

WE EXPRESS THAT (HG)P = 020 (He)M RECALLING (1) AND(2):

EDUATING THE COEFF. OF THE UNIT VECTORSI

$$176.15 \omega_{\chi} = 4.367$$
 $\omega_{z} = 0.02479 \text{ rad/s}$
 $189.19 \omega_{y} = -52.406$ $\omega_{y} = -0.2770 \text{ rad/s}$
 $145.57 \omega_{z} = -52.406$ $\omega_{z} = -0.3600 \text{ rad/s}$

(0.0248 rad/s)i-(0.277 rad/s)j-(0.360 rad/s)k



PRUBE WITH PRINCIPAL CENTROIDAL AXES Z, y, =, AND W=3000 lb, K, = 1.375 ft, Ky=1.425 ft, K, = 1.250 ft.
PROLE HAS NO ANEVLAR VELOCITY WHEN STRIKE AT A BY 5-02 METEOCITE WHICH EMERGES ON OTHER SIDE OF PRINCIPAL MOVING IN SAME DIRECTION WITH SPEED REDUCED BY 25% FINAL ANGULAR VELOCITY OF PROSE IS

 $\omega = (0.05 \text{ rad/s})\dot{v} - (0.12 \text{ rad/s})\dot{j} + \omega_2 \text{ h}$ AND X COMPONENT OF CHANGE IN \dot{v} OF PROBE IS $\Delta v_2 = -0.675 \text{ in./s}.$

FIND: (a) Wa,

(b) RELATIVE VELOCITY & OF METEURITE WITH WHICH IT STRIKES FAMEL.

CONSERVATION OF LINEAR MOMENTUM IN X DIRECTION SINCE 25% OF LINEAR MOM, OF METEUKITE IS TRANS-FERRED TO PROFE:

0.25 (5/16) 16 (Vo) = 3000 16 AV

 $(V_0)_2 = 38.4 \times 10^3 \Delta V_2 = 38.4 \times 10^3 (-0.675 in./s) = 25.92 \times 10^3 in./s$ $(V_0)_2 = -2160 \text{ ft/s}$

CONSERVATION OF ANEULAR MOMENTUM AROUT G-INITIAL ANG. MAN. OF METEORITE:

(Ho) = & x m v = [(9+1)i+(0,75H)k] x (7/6/6) [(Va) 1+(Va) 1+(Va) 2]

RECALLING THAT (Va) = -2 160 ft/s AND USING DETERSINANT

$$(H_{G})_{M} = \frac{(5/16) B}{32.2 \text{ ft/s}}, \begin{vmatrix} \frac{1}{4} & \frac{1}{2} & \frac{E}{2} \\ -2/160 \text{ ft/s} & (42) & (42) \end{vmatrix}$$

 $I(H_G)_{44} = \frac{(5/16)/6}{32.2 + 175}$ [-0.75(V_0) $i - (1620 + 9(V_0)) + 9(V_0) k$

FINAL ANG, MONI, OF PROBE;

 $(H_G)_p = I_{\omega_2} \underbrace{i}_{i} + I_{\omega_2} \underbrace{k}_{i} + I_{\omega_2} \underbrace{k}_{i} = m(k_{\omega_2} \underbrace{i}_{i} + k_{\omega_2} \underbrace{i}_{i} + k_{\omega_2} \underbrace{k}_{i})$ $= \frac{3000 \text{ Ib}}{32.2 \text{ Hig.}} \underbrace{\{(1.3754)^{\frac{1}{2}}(0.05 \text{ fod/s}) \underbrace{i}_{i} - (1.4254)^{\frac{1}{2}}(0.12 \text{ fod/s}) \underbrace{i}_{i} + (1.25084)^{\frac{1}{2}} \underbrace{k}_{i} \}$ $(1.25084)^{\frac{1}{2}} \underbrace{\omega_2}_{i} \underbrace{k}_{i}$

SINCE 25% OF AMEDIAN MOM. OF METEORITE IS TRANSFERRED TO PROBE, (Hg) = 0.25(Hg) OR, RECALLING (1) AND (2):

3000 $[(1.375)^{2}(0.05)]_{i} - (1.425)^{2}(0.12)]_{j} + (1.250)^{2}_{0}, k$ = $0.25(5/16)[-0.75(5)]_{i} - (1620 + 9(5)]_{i} + 9(5)]_{j}$

FQUATE THE COEFF OF UNIT VECTORS:

① 203.59 = -0.0585944($\sqrt{6}$)y ($\sqrt{6}$)y = -4840 tt/s ② -731.03 = -126.56 - 0.70313($\sqrt{6}$)z ($\sqrt{6}$)z = 859.7 tt/s ③ 4687.5 ω_z = 0.70313(-4840) ω_z = -0.726 rod/s

MISWERS:

(a) $\omega_z = -0.726 \text{ rad/s}$ (b) $\Psi_0 = -(2160 \text{ ft/s})\underline{i} - (4840 \text{ ft/s})\underline{j} + (860 \text{ ft/s})\underline{k}$ 18.37 GIVENI

RIGID BODY WITH FIXED POINT O, ANG. VELOCITY WANGULAR MOMENTUM HO, AND KINETIC ENERGY T.
SHOW THAT: (A) H. W = 2T,

(6) 0 < 90°, WHERE O IS ANGLE EETWEEN WAND HO

 $= H_2 \omega_n + H_y \omega_q + H_z \omega_z \tag{1}$

SINCE X, Y, Z ARE PRINCIPAL AYES.

 $H_2 = I_2 \omega_2$ $H_3 = I_3 \omega_3$ $H_2 = I_2 \omega_3$

SUBSTITUTE INTO (1):

How = I, W, + I, W, + I, W, + I

BUT, FROM EQ. (18.20), T= 1 (I, w2+ I, w+ I, w)

 $\frac{H_0 \cdot \omega = 27}{(b) \text{ WE CAN EXPRESS THE SCALAR PRODUCT AS}}$ $\frac{H_0 \cdot \omega = H_0 \cdot \omega$

THUS: $\cos \theta = \frac{H_0 \cdot \omega}{H_0 \omega} = \frac{2T}{H_0 \omega} > 0$, SINCE T>0

SINCE CUS \$ > 0, WE MUST HAVE \$ < 900 (Q.F.D)



GIVEN!

RIGID BODY WITH FIXED POINT OF WE INSTANTANEOUS AND VELOCITY TO INTERIOR OF ABOUT LINE OF ACTION OL

(2)

SHOW THAT $T = \frac{1}{2}I_{0L}\omega^2$ (a) USING EQS. (9,46) AND (18,19),

(b) CONSIDER ING T AS THE SUM OF THE K.E. OF PARTICLES P.

(a) EQ. (18.19):

 $T = \frac{1}{2} (I_2 \omega_2^2 + I_y \omega_y^2 + I_z \omega_z^2 - 2I_{zy} \omega_z \omega_y - 2I_{yz} \omega_y \omega_z - 2I_z \omega_z \omega_z)$

LET $W_x = \omega \cos \theta_x = \omega \lambda_x$ $\omega_y = \omega \cos \theta_y = \omega \lambda_y$ $\omega_z = \omega \cos \theta_z = \omega \lambda_z$

SUBSTITUTE INTO EW. (18,19): T= = (1,2+1,2+1,2+2, 1,2,2+2,7,2,2-25,2) W

BUT, BY EN (9.46) OF SEC. 9.16, EXPRESSION IN PARENTHESES IS IOL. THUS:

 $T = \frac{1}{2} I_{OL} \omega^2$ (Q.E.D.)

(b) EACH PARTICLE P. DESCRIBES A CIRCLE OFRADIUS (C CENTERED DN DL WITH A SPEED V; = (" ω THEREFORE T-15 (Am.) V.2-15 (Am.) v.2.)2

 $T = \frac{1}{L} \sum_{i} (\Delta m_i) V_i^2 = \frac{1}{2} \sum_{i} (\Delta m_i) Q_i^2 \omega^2$ $= \frac{1}{2} (\sum_{i} Q_i^2 \Delta m_i) \Delta^2$

BUT E CLAM: = IDL

THEREFORE:

 $T = \frac{1}{2} J_{0L} \omega^2$

(Q, E, D.)

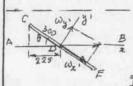
75 mm 225 mm 75 mm

ASSEMBLY OF PRUB. 18.1. FOR EACH ROD:

M=1,5 kg
LENGTH = GOUMM
ASSEMBLY ROTATES
WITH \(\omega = 12 \text{ rod /s} \).

FIND:

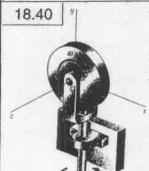
KINETIC ENERGY
OF ASSEMBLY.



USING PRINCIPAL AXES 2'8' 7:

 $\omega_{x} = \omega \cos \theta$ $\omega_{y} = \omega \sin \theta$ $\omega_{z} = 0$ $\vec{T}_{x} = 0, \quad \vec{I}_{y} = \frac{1}{\sqrt{2}} m \ell^{2}, \quad \vec{I}_{z} = \frac{1}{\sqrt{2}} m \ell^{2}$

EQ. (18.20): $T = \frac{1}{2} (\bar{I}_{2}, \omega_{2}, + \bar{I}_{3}, \omega_{3}, + \bar{I}_{4}, \omega_{2})$ $T = \frac{1}{2} (0 + \frac{1}{12} m \ell^{2} \omega^{2} \sin^{2} \theta + 0)$ $= \frac{1}{24} (1.5 \text{ kg}) (0.6 \text{ m})^{2} (12 \text{ rad/s})^{2} \sin^{2} 41.41^{2}$ T = 1.417 J



GNEN:

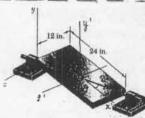
DISK OF PROB. IB. 2 OF MASS AND RADIUS & ROTATING AS SHOWN FIND:

KINETIC ENERGY OF DISK.

EQ.(18,20): $T = \frac{1}{2} (\overline{I}_{2} \omega_{1}^{2} + I_{3} \omega_{2}^{2} + J_{3} \omega_{2}^{2})$ $= \frac{1}{2} (0 + \frac{1}{4} m \epsilon^{2} \omega_{2}^{2} + \frac{1}{2} m \epsilon^{2} \omega_{1}^{2})$ $T = \frac{1}{8} m \epsilon^{2} (\omega_{2}^{2} + 2 \omega_{1}^{2})$

18.41 GIVEN: 18-16 RECTANGULAR PLATE OF PROB. 18.3 ROTATING WITH W = 5 tod/s ABOUT & AXIS.

FIND: KINETIC ENERGY OF PLATE



WE USE PRINC, CENTROIDHL AXES

GZY'E' WITH

tand = 12in = 0.5 A=26.565

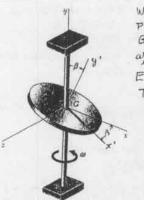
 $tan\theta = \frac{12 in}{24 in} = 0.5 \quad \theta = 26.565$

 $\overline{I}_{2}, = \frac{1}{12} \frac{(8b)}{9} (14t)^{2} = \frac{1.5}{9}$ $\overline{I}_{2}, = \frac{1}{12} \frac{181b}{9} (24t)^{2} = \frac{6}{9}$

T=0,932 ft. 16

EQ. (18,20): $T = \frac{1}{2}(\bar{1}_2, \omega_2^2, + \bar{1}_3, \omega_3^2, + \bar{1}_2, \omega_{21}^2)$ $T = \frac{1}{2}[\frac{1.5}{3}(5 \text{ rad/s})^2\cos^2 26.565^2 + 0 + \frac{6}{3}(5 \text{ rad/s})^2\sin^2 26.565^2]$ $= \frac{1}{2}\frac{11.5}{32.2}(5)^2(\cos^2 26.565^2 + 4 \sin^2 26.565^2)$ $= (0.58230 \text{ ft.} \text{ lb})(0.8 + 4 \times 0.2) = 0.9317 \text{ ft.} \text{ lb}$ 18.42

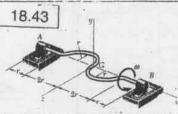
GIVEN: DISK OF PROB. 18.4. WITH P=25.



WE RESOLVE $\omega = \omega_{\hat{k}}$ A corls THE

PRINCIPAL CENTROLLAL AXES $G z_{\hat{k}}^{i} z_{\hat{k}}^{i} = \omega_{\hat{k}}^{i} = \omega_{\hat{k}}^{i} = \omega_{\hat{k}}^{i} + \omega_{\hat{k}}^{i} = \omega_{\hat{k}}^{i} = \omega_{\hat{k}}^{i} + \omega_{\hat{k}}^{i} = \omega_{\hat{k}}^{i} + \omega_{\hat{k}}^{i} + \omega_{\hat{k}}^{i} + \omega_{\hat{k}}^{i} = \omega_{\hat{k}}^{i} + \omega_{\hat{k}}^{i}$

T=0.228m2w



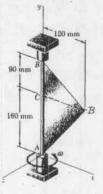
GIVEN:
SHAFT OF ARE 18.15
OF MASS M, ROTATING
WITH ANG, VEL. W.
FIND:
KINETIC ENERGY OF

MASS PER UNIT LENGTH = $m^2 = \frac{m}{2\ell + 2\pi t} = \frac{m}{2(\pi + 1)t}$ SINCE $\omega_z = \omega_z = 0$, F(2, (18, 19)) REDUCES TO $T = \frac{1}{2}I_2\omega^2$. BUT I_2 OF BOTH SERVICIACULTIKE PERTINDS OF KOD IS SAME AS OF FIRE CIRCLETAK MIDITARY IS, $I_2 = \frac{1}{2}(2\pi c n)^2 c^2 = 97c^2 m^2 = \pi c^2 \frac{m^2}{2(n+1)} = \frac{\pi c^2}{2(n+1)} \frac{m}{n}$ THEREFORE, $T = \frac{1}{2}\frac{\pi c^2}{2(n+1)} = \frac{\pi c$

18.44 GIVEN: TRIAINGLING PLATE OF PROB. 18, 16

OF MASS MI = 7.5 Kg WITH ANG, VEL. W= 12 rad/s

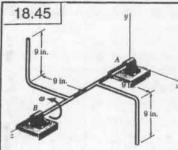
FIND: KINETIC ENERGY OF PLATE



SINCE $\omega_{\chi} = \omega_{z} = 0$, EQ.(18.19) REDUCES TO $T = \frac{1}{2} I_{\chi} \omega^{2}$ (1) BUT $I_{\chi, MRS} = \frac{1}{12} bh^{3}$. AND $I_{\chi, MRS} = \frac{m}{2bh} (\frac{1}{12} bh^{3}) = \frac{1}{6} mh^{2}$ WHERE m = 7.5 kg, h = CB = (0.12m)THUS $I_{\chi, MRSS} = \frac{1}{6} (7.5 kg) (0.12m)^{2}$ $= 18.00 \times 10^{-2} kg \cdot m^{2}$

SUBSTITUTING THIS VALUE FOR I AND 12 rad/s FOR W INTO (1), WE HAVE $T = \frac{1}{2} \left(18.00 \times 10^{3} \, \text{kg·m}^{2} \right) \left(12 \, \text{rad/s} \right)^{2}$

T= 1,296J



ASSEMBLY OF PROB. 18.19 WHICH ROTATES AT 360 Ppm. EACH L-SHAPED AKM WEIGHS 5 16.

FIND!

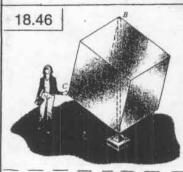
KINETIC ENERGY OF ASSEMBLY.

SINCE $\omega_{\chi} = \omega_{\chi} = 0$, EQ.(IB.19) REDUCES TO $T = \frac{1}{2} I_{\chi} \omega^{2}$ $T = \frac{1}{2} I_{\chi} \omega^{2}$ POR ONE ARM(OF MASS m); $I_{z} = (\overline{I}_{z}) + \frac{m}{2} d^{z} + (\overline{I}_{z})_{z}$ $= \frac{1}{12} \frac{m}{2} a^{z} + \frac{m}{2} (a^{z} + \frac{a^{z}}{4}) + \frac{1}{3} \frac{m}{2} a^{z} = \frac{5}{6} ma^{z}$ FOR BOTH ARMS: $I_{z} = \frac{5}{3} ma^{z} = \frac{5}{3} \frac{5}{22.2445^{z}} (\frac{3}{4} + t)^{c}$

= 0.14557 16.ft.s2

AND W= 360 rev = 360 29 rad = 12 17 rad/s

THUS: $T = \frac{1}{2} I_{\Delta} \omega^2 = \frac{1}{2} (0.14557 \text{ lb.ft.s}) (1271 \text{ rad/s})^2$ T = 103.5 ft. lb



GIVE 1:

HOLLOW 5x5+t ALUMINUM CURE OF PROB. 18.21.
5TUTE-IT PIE-TE COLITER C FOR 1.25 IN DIRECTION PERFONDIC! HE TO PLANE ARC WITH 12.5-16 FORCE, CAUSING CURE TO COMPLETE 1 REV IN 55.

FIND: KINETIC ENERGY

DIRECT COMPUTATION OF K.E. WE HAVE $\omega = (2\pi \text{ rad})/5s = 1.2566 \text{ rad/s}$ WE RECALL FROM PROB. 18.21 THAT AB IS A PRINCIPAL AXIS AND THAT IAB = $\frac{5}{18}$ ma* THUS, EQ.(18.4) YIELDS $T = \frac{1}{2} I_{AB} \omega^2 = \frac{1}{2} \frac{5}{18} \text{ ma}^4 \omega^2 = \frac{5}{36} \text{ m} (5 \text{ ft})^4 (1.2566 \text{ rad/s})^2$

BUT WE FOIND IN PROB, 18.21 THAT W = 226 16
THUS: $T = \frac{5}{36} \frac{2261b}{32,247/52} (54+)^2 (1.2566 red/s)^2 = 38,48 + 1.6$ T = 38,5 + 1.6

ALTERNATIVE SOL TION

WE NOTE THAT THE K.E. IMPARTED TO THE CUBE IS EQUAL TO THE WORK U DONE BY THE STORENT!

T=U132 = FAS

WHERE F = 12,5 IS AND AS = & UDt = & WEDT

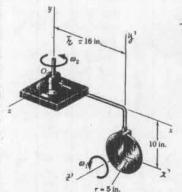
RECALLING THAT THE RADIUS & OF THE CIRCLE DESCRIBED

EY C IS (SEE HINT IN PROE. 18.21)

 $t = a \sqrt{2/3} = (5 + t) \sqrt{12/3} = 4.0825 + t$ WE HAVE $\Delta S = \frac{1}{2} (1.566 \text{ md/s}) (4.0825 + t) (1.2 s) = 3.078 + t$ AND $T = (12.51b) (3.078 + t) = 38.48 + t \cdot b$, $T = 38.5 + t \cdot b$ 18.47 GIVEN

AND ANGULAR VELOCITIES $\omega_1 = 12 \text{ rad/s}$ AND $\omega_2 = 4 \text{ rad/s}$.

FIND: KINETIC ENERGY OF DISK.

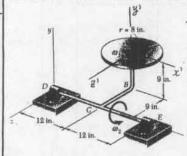


EQ. (18,17):

 $T = \frac{1}{2} m \overline{v}^{2} + \frac{1}{2} (\overline{1} S_{2}^{2} + \overline{1} S_{2}^{3})$ $= \frac{1}{2} m O_{2}^{2} \overline{z}^{2} +$ $+ \frac{1}{2} (O + \frac{1}{4} m z^{2} \omega_{2}^{2} + \frac{1}{2} m z^{2} \omega_{3}^{4})$ $= \frac{1}{2} \frac{B(b)}{32.2kf_{5}^{2}} [(4 \text{ rad/s})^{2} (\frac{1}{12}b)^{2}]$ $+ \frac{1}{4} (\overline{\frac{12}{12}}fz)^{2} (4 \text{ rad/s}) + \frac{1}{2} (\overline{\frac{12}{12}})^{2} [12)^{2}]$ = (O, 12422) [28.444 + + 0.6944 + 12.5] $= 5, 1724 + ft \cdot 16$ $T = 5, 17 + ft \cdot 16$

18.48 GIVEN:

AND ANGULAR VELOCITIES $\omega_1 = 16 \text{ rad/s}$ AND $\omega_2 = 8 \text{ rod/s}$. FIND: KINETIC ENERGY OF DISK.

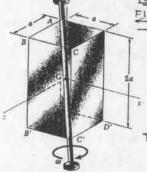


EW (18.17): $T = \frac{1}{2}mD^{2} + \frac{1}{2}(\vec{1}_{1}, x^{2} + \vec{1}_{1}_{1}, x^{2} + \vec{1}_{2}_{1}, x^{2} + \vec{1}_{2}_{1}, x^{2} + \vec{1}_{2}_{1}, x^{2} + \vec{1}_{2}_{2}, x^{2} + \vec{1}_{2}_{2}, x^{2}_{2})$ WHERE $\vec{v}^{2} = \Omega_{1}^{2}(AC)^{2}$ WITH $(AC)^{2} = (AB)^{2} + (BC)^{2}$ $(AC)^{2} = 2(\frac{9}{12}f^{2})^{2} = 1.125$ $\vec{1}_{N} = \frac{1}{4}mx^{2} = \frac{1}{4}m(\frac{B}{12}f^{2})^{2}$ = 0.1111m $\vec{1}_{N} = \frac{1}{2}mx^{2} = 0.222270$

7HUS: $T = \frac{1}{2} (m \bar{v}^2 + I_{\chi}, \omega_{\chi}^2 + I_{\chi}, \omega_{y}^2 + 0)$ $= \frac{1}{2} m [1.125 \omega_{\chi}^2 + 0.1111 \omega_{\chi}^2 + 0.2222 \omega_{\chi}^2]$ $= \frac{1}{2} \frac{6 \frac{16}{52.2 \text{ ft/s}^2} [1.236 (8 \text{ rad/s})^2 + 0.7212 (16 \text{ rad/s})^2]}{T = 12.67 \text{ ft. lb}}$

18.49 and 18.50

GIVEN: PARALLELEPIPED OF 18,49: PROB. 18.7 (SOLID) 18,50: PROB. 18.8 (HOLLOW) FIND: KINETIC ENERGY



SINCE G IS FIXED AND x,y, zARE PRINCIPAL AXES, USE (18,20): $T = \frac{1}{2} (I_{x} \omega_{x}^{2} + I_{y} \omega_{y}^{4} + I_{z} \omega_{z}^{2})$ WITH $\omega_{x} = -\frac{\alpha}{\sqrt{\alpha^{2} + \alpha^{2} + 4\alpha^{2}}} \omega$ $\omega_{x} = -\frac{\omega}{(6)}, \omega_{y} = \frac{z\omega}{\sqrt{b}}, \omega_{z} = -\frac{\omega}{\sqrt{b}}$ THUS: $T = \frac{1}{12} (I_{x} + 4I + I_{z}) \omega^{2}$ (1)

 $\frac{18.49}{5005717070} \text{ WE HAVE } I = I_{2} = \frac{1}{12} m \left[a^{2} + (2a)^{2} \right] = \frac{5}{12} m a^{2}, I_{3} = \frac{1}{6} m a^{2}$ $T = \frac{1}{12} m a^{2} \left(\frac{5}{12} + \frac{4}{6} + \frac{5}{12} \right) \omega^{2} = \frac{1}{12} m a^{2} \left(\frac{3}{2} \omega^{2} \right) \qquad T = \frac{1}{6} m a^{2} \omega^{2}$

(CONTINUED)

18.49 and 18.50 continued

WE RECH _ FROM THE PREVIOUS PAGE T=12(Ix+4Ix+I2) W2 (1)

18.50: SEE SOLUTION OF PROB. 18.8 FOR THE DETER-MINATION OF THE PRINCIPAL MOMENTS OF INERTIAS $I_{y} = \frac{37}{40} ma^{2}$ $I_{y} = \frac{9}{20} ma^{2}$ Iz = 37 ma SUBSTITUTE IN ER. (1) $T = \frac{1}{12} ma^* \left(\frac{37}{60} + \frac{4\times9}{30} + \frac{37}{60} \right) \omega^2 = \frac{146}{720} ma^2 \omega^2$

18.51

GIVEN: SQUARE PLATE OF PROB. 18.29 OF MASS M WITH ALLE VEL Was STRIKES BWITH e = Q FIND!

T=0,203 maw

KINETIC ENERGY LUST IN IMPACT.

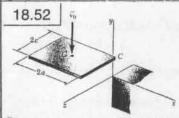
WE RECALL FROM PRUB. 18.29 THAT I = 1/2 7114 ABOUT ANY AXIS THROUGH G IN THE PLANE OF THE PLATE. KINETIC ENERGY BEFORE IMPACT

 $T_0 = \frac{1}{2} \overline{I} \omega_0^2 = \frac{1}{2} \left(\frac{1}{17} n | a^2 \right) \omega_0^2 = \frac{1}{24} m a^2 \omega_0^2$

KINETIC ENERGY AFTER IMPACT

PLATE ROTATES ABOUT AB. WE FOUND IN PROB. 18, 29 THAT $\omega' = \frac{1}{4\sqrt{2}} \omega_0$ AND $\overline{v}' = \omega'(a/2)$ THEREFORE, FROM EU. (18.17). $T = \frac{1}{2} m \sigma^{2} + \frac{1}{2} \bar{I} \omega^{2} = \frac{1}{2} m \omega^{2} (\frac{a}{2})^{2} + \frac{1}{2} (\frac{1}{12} m a^{2}) \omega^{2}$ $= \frac{1}{6} m a^2 \omega^2 = \frac{1}{6} m a^2 \left(\frac{\omega_0}{4\sqrt{2}} \right)^2 = \frac{1}{192} m a^2 \omega_0^2$

KINETIC ENERGY LOST $= \frac{1}{24} m a^2 \omega_0^2 - \frac{1}{192} m a^2 \omega_0^2 = \frac{7}{192} m a^2 \omega_0^2$



GIVEN: RECTHNEULAR MLATE OF PRUBS, 18.31 AND 18.32 OF MASS IM FALLING WITH VELOCITY TO AND W=0 HITS OBSTRUCTION (E=0) FIND: KINETIC ENERGY LOST IN IMPACT.

BEFORE IMPACT

T = 1 m 0

AFTER IMPACT

FROM PROB. (11.31): $\omega_{\chi} = 3 \, \overline{v}_0 / 7c$, $\omega_{\chi} = 0$, $\omega_{\chi} = 3 \, \overline{v}_0 / 7a$ FROM PROB. (18.32): $\overline{\psi} = -(6 \, \overline{v}_0 / 7) \, \underline{\dot{q}}$

PQ.(18. 17): T= + m v2++ (I, w2+ I, w+ I, w2) $= \frac{1}{2} m \left(\frac{6}{7} \vec{V_0} \right)^2 + \frac{1}{2} \left[\frac{1}{3} m c^2 \left(\frac{3}{7} \vec{V_0} \right)^2 + 0 + \frac{1}{3} m a^2 \left(\frac{3}{7a} \vec{V_0} \right)^2 \right]$ $=\frac{1}{2}m\overline{V_0}^2(\frac{1}{7})^2[36+3+3] = \frac{1}{2}m\overline{V_0}\frac{42}{49} = \frac{1}{2}\frac{6}{7}m\overline{V_0}$

76-T= 1m 76(1-6)

 $T_o - T = \frac{1}{14} m \overline{V_o}$

18.53 GIVEN ;

SPACE PROBE OF PROB. 18,35, WITH W = 3000 lb, K = 1,375 ft, Ky = 1,425 ft, Kz = 1,250 ft.

KINETIL ENERGY OF PROBE IN ITS MOTION ABOUT ITS MASS CENTER AFTER ITS COLLISION WITH METEURITE.

SEE SOLUTION OF PROB. 18,35 FOR DETERMINATION OF $\omega_{x} = 0.0248 \text{ rad/s}, \ \omega_{y} = -0.277 \text{ rad/s}, \ \omega_{z} = -0.360 \text{ rad/s}$ IN MOTION ABOUT G, G IS A FIXED POINT AND THE 12, 4, 2 AXES ARE PRINCIPAL AXES, WE USE EU (18,20); $T' = \frac{1}{2} (I_1 \omega_2^2 + I_2 \omega_2^2 + I_2 \omega_2^2) = \frac{1}{2} m (k_2^2 \omega_2^2 + k_2^2 \omega_2^2 + k_2 \omega_2^2)$

 $= \frac{1}{2} \frac{3000 lb}{32.2 t/s} \left[(1.375 ft \times 0.0248 rad/s)^{2} + (1.425 ft \times 0.277 rad/s)^$ + (1.250 ft x 0,360 rad/s)]

 $= \frac{1}{2} \frac{3000 \text{ lb}}{32.2 \text{ ft/s}^2} (0.3595 \text{ ft}^2/\text{s}^2) = 16.747 \text{ ft} \cdot \text{lb}$ T' = 16.7T'= 16,75 ft.16

GIVEN: 18.54

SPACE PROBE OF PROB. 18.36, WITH W= 3000 16, K,= 1.375 ft, Ky= 1.425 ft, K2= 1.250 ft. FIND:

KINETIC ENERGY OF PROBE IN 175 MOTION ABOUT 175 MASS CENTER AFTER ITS COLLISION WITH METEORITE.

SEE STATEMENT AND SOLUTION OF PROB. 19.36 FOR THE VALUES OF WX, WX, WZ AFTER COLLISION :

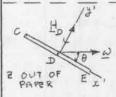
 $\omega_z = 0.05 \text{ rad/s}, \ \omega_y = -0.12 \text{ rad/s}, \ \omega_z = -0.726 \text{ rad/s}$ IN MOTION ABOUT G. G IS A FIXED POINT AND THE X, Y, Z AXES ARE PRINCIPAL MXES. WE USE EW. (18,20): $T' = \frac{1}{2} (I_2 \omega_x^2 + I_3 \omega_3^2 + I_2 \omega_3^2) = \frac{1}{2} m (k_x \omega_x^2 + k_3 \omega_3^2 - k_2^2 \omega_3^2)$

= \frac{1}{32.2 the [(1.375 \tex 0.05 rodk) + (1.425 \tex 0.12 rad/s) + + (6250 ftx 0,726 rad/s)2)

= 1 3000/b (0,8575 ft/sz) = 39,946 ft/b T'= 39,9 ft.16

GIVEN: ASSEMBLY OF PROB 18.1 18.55 FOR FACH ROD: M= 1,5kg, L= 600 mm ASSEMBLY ROTATES WITH W= 12 rad/s.

FIND! RATE OF CHANGE H OF ANG. MOMENTUM H



FROM PROB. 18.1: 0 = 41.41" USING PRINCIPAL HXES X'Y'&: W = W (cos D1'+ sin Dj') HD = 12 m l'w sin & j' EQ. (18.22') YIELDS HD = (HD) DXA + TXHD

BUT (HD) DE'B' = 0 AND Q = W.

HD = WXHD=W(costi'+sinti)x 12 ml'wsinti = $\frac{1}{12}$ mlw sindcoso $k = \frac{1}{24}$ mlw sin 20 k

WITH GIVEN DATA, HD = 1/2 (1.5 kg) (0.6 m) (12 rad/s) sin 82.82° K

H = (3.21 N·m) k

GIVEN: DISK OF PROB. 18.2. 18.56 GIVEN: DISK OF PROB. 18.6 WEIGHING 5 IL 18.60 FIND: RATE OF CHANGE HE OF HE WITH W. = 16 rod/s AND Wz = B rod/s. FIND! RATE OF CHANGE HA OF HA. FROM PROB. 18.2: W = W2 1 + W, K USING PRINCIPAL CENTROIDAL AXES AX" 'E' . HG = 4 m2 (Wzj + 2W, K) W= W2 1 + W1 & WE NOTE THAT THE ANGULAR VEWCITY OF THE けるまし、ひ、シャラ、ジャラ、シュー」からかとしょきからいは PRAME GXYZ 15 1 = W23 EQ. (18.22): EQ.(18.22): HA = (HA)Axy + + QXHA = 0 + W2i X HA $H_G = (H_G)_{GXYZ} + \Omega \times H_G = 0 + \Omega \times H_G$ = w2 ix (1 m + w2 i + 1 m + w)) = 1 m+ w, w2 t THUS: H = Wzjx + mr (wj+2w, r) WITH GIVEN DATAL = 1 32,2 this (9 ft) (16 rad/s) (8 rad/s) K H = + mrw, wi H = (5,30 lb.ft)k GIVEN: PLATE OF PROB. 18,3 WEIGHING 18.57 18/b, WHICH ROTATES WITH W=5 rad/6. 18.61 GIVEN: ASSEMBLY OF PRUB. 18.1. FIND: RATE OF CHANGE H. OF H. FOR EACH RUD: m=1.5 kg, l= 600 inii AT INSTANT CONSIDERED, W=(12 rad/s) i, X=(96 rad/s2) i. WE HAVE $\omega = (5 \operatorname{rad/s}) i$ FIND: RATE OF CHANGE SEE SOLUTION OF PRUB. 18.3 FOR THE DERIVATION OF EQ. (2): PROM PROB. 18.1: 0=41,41° Ha = (0,3727 /b.ft.s) i - (0,2795 /b.ft.s)k USING PRINCIPAL AXES X X 2: w=w(cosbi'+sinbj') EQ. (18,22): a = a (cosoi'+sinoj') H=(HG)xyz+ IXXHG = 0 + W X HG 7 OUT OF Hp = 12 me wsindj' THUS: H = (5 rad/s) i x [0,3727 | b.ff.s) i - (0,2795 | b.ff.s) k] HD) Dxij = 12 m l'w sindj' = 12 m l'asindj' H = (1.398 16.1t)} APPLY EQ. (18,22), OBSTRVING THAT Q = W : 18.58 GIVEN: DISK AND SHAFT OF PRUB. 18.4. $\dot{\mathbf{H}}_{\mathbf{D}} = (\dot{\mathbf{H}}_{\mathbf{D}})_{\mathbf{D}\mathbf{z}'\mathbf{y}'\mathbf{z}} + \dot{\mathbf{D}} \times \dot{\mathbf{H}}_{\mathbf{D}} = (\dot{\mathbf{H}}_{\mathbf{D}})_{\mathbf{D}\mathbf{z}'\mathbf{y}'\mathbf{z}} + \dot{\omega} \times \dot{\mathbf{H}}_{\mathbf{P}}$ FIND: RATE OF CHANGE H OF HG = 1/2 m (o sin o j + w (o so i + sin o j) x 1/2 m (w sin o j) USING THE PRINCIPAL AXES GX'y'Z. Inleasing; + 1 mlew cost sind k WE FOUND IN PROB. 18.4 THAT w = w (-singi'+ cospj') BUT j'= sinti + costj HG = 4me w (-sinp i'+2 wspj') H = 1 ml x sind (sinti + coso)+ EN (18.22): He = (HG)Gx, A, + To x He = 0 + M x He 12 ml w cos & sint k H= ω(-sinβi'+ cosβi') × +m v ω(-sinβi'+2 cosβj') H = 12 ml sind (& sind i+ acost j + cost k) = + mrw (-2sin Bcosp K+cospsing K) = - 18 m2 w sin 2 B k = - 18 m12 w sin 50° WITH GNEN DATA! m= 1.5 kg, l= 0.6m, W= 12 rad/s, Q= 96 rad/s. 0 = 41.41. H=-0,0958mEWK H_= 17 (1.5 kg X 0.6 m) sin 41.410 [(96 rad/si) sin 41.410 i + 18.59 GIVEN: DISK OF PAOB. 18.5 WEIGHING & Ib. (96 rad/s) cos 41,41 1 + (12 rad/s) cos 41,41 K WITH WI = 12 rad/s AND WZ = 4 rad/s. H = (1.890 Nim) 1 + (2.14 Nim) 1 + (3.21 Nin) k FIND: RATE OF CHANGE HE OF HA USING PRINCIPAL CENTROIDAL AXES HX'Y'E' GIVEN: ASSEMBLY OF PROB. 18. 1. 18.62 W= W2 1 + W, K FOR EACH ROD: MI = 1.5 kg, &= 600 mm. 日から、いたり、からきりないとことがなったとういといき AT INSTANT CONCIDENTO, W=(12 rad/s), X=-(96 rad/s) L. FIND: RATE OF CHANGE HOOF HO. EQ. (18.22): HA = (HA) Azy'; + Qx HA = 0 + WZ X X HA SUBSTITUTE GIVEN DATA INTO ER. (1) OF PROB. 18, 61. H = W2 j x (+ m + w, j + + m + w, k) = + n + w, w2 i Hp=12ml'sin9(xsin+i+xcosoj+w'cosok) H = 1/2 (1.5 kg) (0, 6 m) sin 41.41° [- (96 rad/s) sin 41.41"+ WITH GIVEN DATA: H= 1/2 816 (5 ft) (12 rad/s) (4 rad/s) & (-96 rab/s) cos +1,41° j + (12 md/s) cos 41,41° k) H = (1.035 /b. ft) i H = - (1.890 N·m) = - (2.14 N·m) + (3,21 N·m) k

18.63

GIVEN: AT INSTANT CONSIDERED, 18-16 PLATE OF PROB. 18.3 HAS W = (5 rod/s) L AND & = - (20 rad/52) i.

FIND : RATE OF CHANGE H OF H

SEE SOLUTION OF PROB. 18.3 FOR THE DERIVATION OF EQ.(1): H= [(0.074534 16.ft.5') = -(0.055901 16. H5') A) W (1)

SINCE IS = 0, WE HAVE

(HG) = (0.074534 1-0.0559 11K) a

SINCE D = W, EW. (18.22) YIELDS

HE (HE) HY + WX HG

= (0,074534i-0.0559016) a

+ W 1 × (0,074534 i-0.055901 k) W

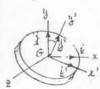
= 0.074534 Qi -0.055901 QK +0.055901 02 j

LETTING & = - 20 rad/s* AND W = 5 rad/s, H = 0,074534 (-20) 1 +0,055901 (5) 3 -0.055901 (-20) K

H = - (1,491 16.ft) i + (1,398 16.ft) j + (1,118 16.ft) k

GIVEN: AT INSTAIL CONSTRUCT. SHA! 18.64 OF PHE. 18.4 HAS ANGULA F TELOCITY w=w & AND AMERICA HECE TRATION &= a } FIND: RATE OF CHANGE H OF IT

SEE SOLUTION OF PROB. 18,4 FOR THE DETERMINATION OF HA. USING THE PRINCIPAL CENTROIDAL EXES



Gx'y'z, WE OBTHINTE EU.(1):

TO REVERT TO THE OKIGINAL

AXES GXYZ, WE OBSERVE 1' = 6 cos p - j sin p = i simp + i cosp

SUBSTITUTING INTO (1):

He = 1 min [-sinp (icos & - ising) +

2 cosp(ising+jasa)

= 4 me w sin,3 cosp i + (1+cos 3) j]

SINCE W=X

(HG)Gara = 4 me a [sin B cospi+ (Hcosp)]

WE USE EQ. (18.22) WITH Q= W = W

GXY=+ 12 × H = + +112 X [Sinface] +(1+cins)] + wjx + mrw[sing ws ti+(1+ws/p) jj

H= + m 4 a [sin B coc B i + (1+00 3/3) i] - + ans osinpea pk

LETTING B=250: H = 1 m2 x (0.38302 + 1.8214 j) - 4 0.4 (0.58302) K

H = m 2 (0,0958 x i + 0.455 x j - 0.0958 w k)

18.65 600 mm 0, = 900 mm 900 mm

GIVEN:

ASSEMBLY CONSISTING OF TWO TRIANGULAR PLATES, EACH OF MASS MI 5kg . WELDED TO VERTICAL SHAFT ASSEMBLY RUTATES WITH CONSTANT W = 8 rad/s.

DYNAMIC REACTIONS AT A

SINCE W= Wj, FQS. (18.7) YIELD H=-I, w, H, = I, w, H=-I, w (1)

MOMENTS AND PRODUCTS OF INERTIA:

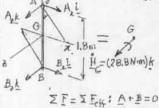
 $I_g = \lambda \left(\frac{m}{\Delta} I_g, AREA \right) = 2 \frac{4m}{4ab} \left(\frac{1}{12} a b^3 \right) = \frac{1}{3} m b^2$ [cf. front cover] $I_{23} = 2\left(\frac{m}{A}I_{24,AREA}\right) = 2\frac{m}{Lab}\left(\frac{1}{24}a^{2}b^{2}\right) = \frac{1}{6}mab$

Iy2 = 0 PROM EU.(1): H=- = - = mabw : + = m b" wj

 $\begin{array}{l} \exists a. (19.27): \\ \overset{\cdot}{H}_{G} = (\overset{\cdot}{H}_{G})_{G} \times \overset{\cdot}{H}_{G} = 0 + \omega \times \overset{\cdot}{H}_{G} = \omega_{\overset{\cdot}{J}} \times m \omega (-\frac{1}{6}ab\overset{\cdot}{L} + \frac{1}{3}b^{\overset{\cdot}{L}}\overset{\cdot}{J}) \end{array}$

H = 1 mabotk = 1 (5kg)(0.9m)(0.6m)(Brad/s)=(28,8 N·m)k

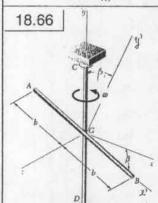
EQUATIONS OF NOTION: WE EQUATE THE SYLTEMS (F EXTERNAL AND EFFECTIVE FONCES.



IMB = Z(MB)ett: (1.8 m) = (Azi+Azk)=(20.8 Mm)k - 48 A, K + 1.8 Az L = 28.8 K

Az=-16N, Az=0 A = - (16.00 N) i

B = (16,00 N) i



GIVEN!

ROD AB OF MASS IN IS WELDED TO SHAFT CD, OF LENGTH 26 WHICH ROTATES AT CONSTANT RATE W.

FIND:

DYNAMIC REACTIONS AT C

USING THE PRINCIPAL AXES Gz'y'z:

I,=0, I,= I== + mb" Wz = - Wsing, Wz = W cos B, W = 0 H=I,w,i+I,w,i+I,w,k

Hg=1mbwcosBj

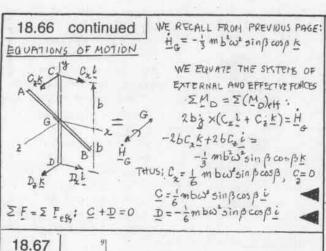
OR, SINCE = isinp+jcosp: H===mbwcosp(sinpi+cospj) (1) EQ.(18,22):

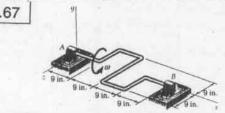
H=(HG) = + 12 × H= = 0 + 10 × H

= wj x 3 m bar coeps (sing i + co:) j)

= - 1 mbw sin, 1 cosps k

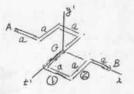
(CONTINUED)





CIVEN: 16-16 SHAFT WITH UNIFORM CROSS CE ROTATES AT CONSTANT RATE Q = 12 rad/s. FIND: DYNAMIC REPORTED AT A MILE B

MONENTS AND PRINCE : OF 1. FOTIA



WE DENOTE BY A THE LENGTH
OF AN ELEMENT IS NOT AND
EY ON ITS MASS. USING THE
CENTROIDEL AXES GZY'E:

= 2 mat + 4 (fmat) = 10 mat

Inj. = 0

 $I_{22i} = 2m\bar{z}_1^2 + 2m\bar{z}_2^2 = 2m(\frac{4}{5})a + 2ma(\frac{4}{5}) = 2ma^2$ $H_G = \bar{I}_2\omega_i - \bar{I}_2\omega_j - I_2\omega_k = \frac{10}{3}ma^2\omega_i^2 - 2ma\omega_k \qquad (1)$

FQ.(18.72): Hg=(Hg)gzy'++ 1xHg=0+WixHg

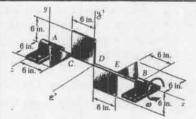
Ho = wix (10 mawi-2 mawk) = 2 maw = EQUATIONS OF MOTION

Ask G By H

 $\sum_{A} A = \sum_{A} (M_{A})_{eff}$: $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \stackrel{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel{!}{\underline{!}}) = \frac{!}{\underline{!}}_{G}$ $4a \stackrel{!}{\underline{!}} \times (B_{g} \stackrel{!}{\underline{!}} + B_{g} \stackrel$

 $\Sigma F = \Sigma F_{eff}$: A + B = 0 $A = -B = \frac{1}{2} ma \omega^* k$ $DATAi m = \frac{1}{6} \frac{W}{3} = \frac{1}{8} \frac{16.16}{32,217} = 0.062112 16.5\% ft$ a = 9 in = 0.75 ft $\omega = 12 rad/6$

THUS: $A = \frac{1}{2} (0.062112 / 6.5^{3}/ft) (0.75ft) (12 rad/s)^{2} = 3.354 / 6$ A = (3.35 / 6) k; B = -(3.35 / 6) k 18.68



GIVEN:
ASSEMBLY WEIGHS 2.7 Ib AND ROTATES AT CONSTANT
RATE $\omega = 240$ rpm
FIND: DYNAMIC REACTIONS AT A AND B

COMPUTATION OF MOMENTS AFID PRODUCTS OF INERTIA

WE US = THE CENTROIDAL AXES DXZ'E'.

FOR EACH SQUARE: $mI = \frac{1}{3} \frac{2.71b}{g} = \frac{0.9}{g}$ $I_{\chi} = \frac{1}{3} ma^{\alpha} = \frac{1}{3} \frac{0.9}{g} (\frac{1}{2} + 1)^{\alpha} = 0.075/a$ $I_{\chi} = m(\frac{a}{2})(-\frac{a}{2}) = -\frac{1}{4} \frac{0.9}{g} (\frac{1}{2} + 1)^{2} = -0.05625/g$, $I_{\chi} = 0$ FOR EACH TKIANCIE! $m = \frac{1}{6} \frac{2.71b}{g} = \frac{0.45}{g}$ $I_{\chi/MSS} = I_{\chi/AKEM} \frac{dI}{|A|} = \frac{1}{12} a^{\frac{4}{3}} \frac{m}{2} = \frac{1}{6} ma^{\alpha} = \frac{1}{6} \frac{0.45}{g} (\frac{1}{2})^{2} = \frac{0.0875}{g}$ $I_{\chi/AKEM} = A\chi' Z' + I_{\chi/2} = \frac{1}{2} a^{\frac{4}{3}} \frac{m}{2} - \frac{1}{3} + \frac{1}{72} a^{\frac{4}{3}} = -\frac{15}{72} a^{\frac{4}{3}} \left[\frac{1}{2} \frac{1}{3} \frac{n}{3} \right] + \frac{1}{72} a^{\frac{4}{3}} = -\frac{15}{12} a^{\frac{4}{3}} \left[\frac{1}{2} \frac{n}{3} \right] = 0$ $I_{\chi/AKEM} = I_{\chi/AKEM} = -\frac{15}{72} a^{\frac{4}{3}} \frac{m}{2} = -\frac{5}{12} ma^{\frac{4}{3}} = -\frac{5}{12} \frac{0.45}{g} \left(\frac{1}{2} \frac{n}{4} \right) = 0$ $I_{\chi/AKEM} = I_{\chi/AKEM} = -\frac{15}{72} a^{\frac{4}{3}} \frac{m}{2} = -\frac{5}{12} ma^{\frac{4}{3}} = -\frac{5}{12} \frac{0.45}{g} \left(\frac{1}{2} \frac{n}{4} \right) = 0$ $I_{\chi/AKEM} = I_{\chi/AKEM} = -\frac{15}{72} a^{\frac{4}{3}} \frac{m}{2} = -\frac{5}{12} ma^{\frac{4}{3}} = -\frac{5}{12} \frac{0.45}{g} \left(\frac{1}{2} \frac{n}{4} \right) = 0$ $I_{\chi/AKEM} = I_{\chi/AKEM} = I_{\chi/AKEM} = -\frac{15}{72} a^{\frac{4}{3}} \frac{m}{2} = 0$ $I_{\chi/AKEM} = I_{\chi/AKEM} = I_{\chi/AKEM} = -\frac{15}{12} a^{\frac{4}{3}} \frac{m}{2} = 0$ $I_{\chi/AKEM} = I_{\chi/AKEM} = I_{\chi/AKEM} = I_{\chi/AKEM}$

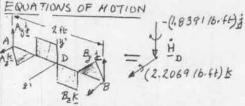
 $I_{\chi} = \frac{1}{2 \times 0.075} + 2 \times 0.01875 / g = 0.1875 / g$ $I_{\chi y} = \frac{2(-0.05625)}{g} = -0.1125 / g$ $I_{\chi z} = \frac{2(-0.046875)}{g} = -0.09375 / g$

H_ = I wi - I y wi - I z wk

 $\frac{H}{D} = (0.1875 \underline{i} + 0.1125 \underline{j} + 0.09375 \underline{k})(\omega/g)$ $E0.(18.22) : \frac{H}{D} = (\frac{H}{D})_{DX} \underline{y}_{2}^{1} + \underline{\Pi} \times \underline{H}_{B} = 0 + \omega \underline{i} \times \underline{H}_{G}$

SINCE D- 240 rpm = 817 rad/s, AND ix i=k, !xk=-i, H = (0.1129 K-0.09375 i)(817)/g. II = -(1.8391 /6.ft) i + (2.2069 /6.ft) k

(1)

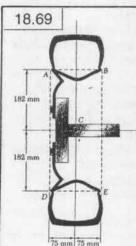


ZM = Z(Ma)ess: (2ft) i × (B, j+B, t) = -1.8391 j+2.2069 k 23 k-28 j = -1.8391 j + 2.2069 k

THUS: $B_y = \frac{1}{2}(2.2069) = 1.1034 \text{ lb}$ $B_z = \frac{1}{2}(1.8391) = 0.9196 \text{ lb}$

B = (1.103 lb) + (0.920 lb) k

 $\overline{Z} f = \overline{Z} f_{ess} : \underline{A} + \underline{B} = 0$ $\underline{A} = -\underline{B} \qquad A = -(1.10316) \hat{\underline{j}} - (0.92016) \underline{K}$



18-kg WHEEL IS ATTACHED TO BALANCING MACHINE.

WHEN MACHINE & FINS AT THE RATE OF 12.5 rev/s, WHEEL IS FOUND TO EXERT ON MACHINE A FORCE-CO. PLE - 12.211

CONSISTING OF

F = (160 N) & APPLIED ATC

 $M_C = (160N)_{\frac{1}{2}} APPLIEDA$ AND $M_C = (14.7 N.m)_{\frac{1}{2}} N.m$ FIND:

FIND:

(a) DISTANCE & FROM & AXISTO G,

AND IZY AND IZX;

(b) THE TWO CORRECTIVE MASSES

REQUIRED TO BALANCETHE WHELL

AND AT WHICH OF POINTS 4, B, C, D

THEY SHOULD BE PLACED.

(a) THE FORCES EXERTED ON THE WHEEL MUST BEEQUINILENT

$$\frac{\sum_{i=1}^{n} \frac{1}{(160 \text{ N})_{1}^{2}}}{\sum_{i=1}^{n} \frac{1}{(160 \text{ N})_{2}^{2}}} = \frac{\sum_{i=1}^{n} \frac{1}{(160 \text{ N})_{2}^{2}}}{\sum_{i=1}^{n} \frac{1}{(160 \text{ N})_{2}^{$$

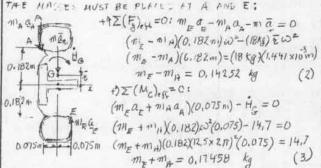
$$\begin{split} & \sum \underline{M}_{G} = \overline{Z}(\underline{M}_{G})_{eH} : - (14,7N_{i}m)\underline{K} = \underline{H}_{G} \qquad (1) \\ & \underline{B}UT \ \underline{H}_{G} = \underline{I}_{i}\underline{\omega}\underline{i} - \underline{I}_{i}\underline{\omega}\underline{i} - \underline{I}_{j}\underline{\omega}\underline{k} \\ & \underline{AND} \ \underline{H}_{G} = \underline{\omega} \times \underline{H}_{G} = \underline{\omega}\underline{i} \times (\underline{I}_{i}\underline{\omega}\underline{i} - \underline{I}_{j}\underline{\omega}\underline{j} - \underline{I}_{i}\underline{\omega}\underline{k}) \\ & \underline{H}_{G} = -\underline{I}_{j}\underline{\omega}^{2}\underline{K} + \underline{I}_{j}\underline{\omega}^{2}\underline{j} \\ & \underline{SUBSTITUTE} \ \underline{IN} \ (1): -14.7\underline{K} = -\underline{I}_{j}\underline{\omega}^{2}\underline{K} + \underline{I}_{j}\underline{\omega}^{2}\underline{j} \\ & \underline{THUS}: \underline{H_{i}TN_{i}m} \\ \underline{I}_{2j} = \underline{(12,5)27170}\underline{f}\underline{s}) = 2.3831 \times \underline{10}^{3}\underline{k}\underline{g}.m^{2} \ \underline{AND} \ \underline{I}_{2j} = 0 \end{split}$$

Izy=2.38 g·m², Iz=0

(b) WITH CORRECTIVE MASSES THE FORCES EXENTED

OF THE WHEEL ARE POUNTALENT TO TERM, PUR THE

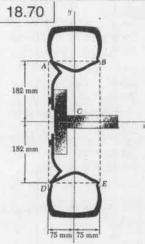
EFFECTIVE PIXES TO ALSO BE EVENTALENT TO TERM.



SOLVING (2) AND (3) SIMULTANEOUSLY:

MA = 16.034×10-3kg, m = 158,55×10-3kg

AT A AND E; m= 16.03g, m= 158.6g



GIVEN: 18-14 WHEEL IS ATTACHED

MECHANIC PINDS THAT A 170 g
MASS PLACED AT B AND A
56-3 MASS PLACED AT D ARE
NEEDED TO BALANCE WHEEL.

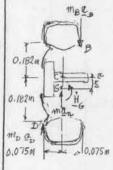
FIND:
BEFORE THE CORKELTIVE
MASSES HAVE GEEN ATTACHED:
(1) SICTANCE & FROM X AYIS
TO G, NOD IN ANTIBLE;
(b) THE FORCE-COUNTE OF THE
AT I EQUIVABLE TO THE FORE
EXERTED BY THE WHEEL ON THE

TO PALANCING MACHINE AND

SPING AT THE RATE OF ISTEVIS

(a) AFTER THE CURRECTIVE MASSES HAVE BEEN ADDED, THE SYSTEM OF THE EXTERINAL PLACES IS ZERU THEREFORE, THE SYSTEM OF THE EFFECTIVE FORCES MUST AUGU BE EQUINALENT TO ZERON SINCE THE LAKEER OF THE TWO MASSES IS PLACED ABOVE THE Z AXIS, THE MASS CENTER G OF THE UNBALANCED WHEEL ALIST ME BELOW THAT AXIS.

MACHINE.



 $+\frac{1}{4} \sum (F_g)_{eg} = 0: m \bar{\alpha}_n - m_g z_g + m_D a_D = 0$ $(18kg) \bar{z} \omega^2 - (0.170 kg)(0.182 m) \omega^2 + \\
+ (0.056 kg)(0.182 m) \omega^3 = 0$ $18\bar{z} = (0.170)(0.182) - (0.056)(0.182)$ $\bar{z} = 1.1527 \times 10^3 m \quad \bar{z} = 1.153 mm$ $+\frac{1}{2} \sum (m_d)_{eff} = 0:$ $\dot{H}_g - m_g a_g(0.075 m) - m_D a_D(0.075 m) = 0$ $\dot{H}_g = m_g b_B \omega^3(0.075) + m_D b_D \omega^3(0.075)$ $= (0.170 + 0.056)(0.182)(0.075) \omega^2$ $\dot{H} = 3.0849 \times 10^{-3} \omega^2 K \qquad (1)$

SINCE $m \bar{a}_{n}$ PASSES THRU C, $H_{c} = H_{c} = 3.0849 \times 15^{3} \omega^{2} k$ (2) BUT $H_{c} = I_{2} \omega \dot{i} - I_{2} \omega \dot{i} - I_{2} \omega k$ AND $H_{c} = \omega \dot{i} \times (I_{2} \omega \dot{i} - I_{2} \omega \dot{j} - I_{2} \omega \dot{k}) = -I_{2} \omega^{2} k + I_{2} \omega^{2} \dot{j}$ (3) EQUATING (2) AND (3), WE HAVE $-I_{2} = 3.0849 \times 10^{-3}$, $I_{2} = 0$ $I_{2} = -3.089 \text{ m}^{2}$, $I_{2} = 0$

(b) THE FORCE-COURSE SYSTEM EXERTED ON THE WHEEL BEPORE THE CURRECTIVE MAY SHAVE BEEN A THACHED IS EGUAL TO THE EFFECTIVE FORCES!

$$\begin{split} F &= m \, \bar{a}_n = m \, \bar{z} \omega^2 \underline{i} = (18kg)(1.1527 \times 10^3 m)(15 \times 2\pi \, rad/s)^2 \underline{i} = (184.3 \, N) \underline{i} \\ \underline{M}_c &= \underline{M}_g = \underline{H}_g = 3.0849 \times 10^3 (15 \times 2\pi \, rad/s)^2 \underline{k} = (27.4 \, N.m) \, \underline{k} \end{split}$$

THE FORCE COUPLE SYSTEM EXERTED BY THE WHEEL ON THE MACHINE BEFORE THE CORRECTIVE MASSER HAVE BEEN ATTACHED

18.71 a = 900 mm 900 mm

GIVEN:

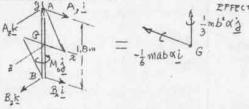
ASSETT Y OF PRUB. 18.65 CONSISTING OF TWO TRIANGULAR PLATES EACH OF MASS m=5ky + 15 AT REST WHEN A COUPLE OF MONENT Mo= (36N·m) & IS APPLIED TO SHAFT AB. (a) ANGULAR ACCELERATION OF ASSEMBLY, (b) INITIAL DYNAMIC

RENCTIONS AT A AND B.

SEE SOLUTION OF FROB. 10.65 FOR DERIVATION OF EQ. (2): H=- = mabwi+ = mb'wj

ER (18,22): # = (!! +) Gry + 1 × H = (!! +) Ery + 0 SINCE IZ = W = D WHEN COUPLE IS APPLIED, THUS H= = (Ha) Gyy = - 1 mab x 1 + 1 mb2 x j

OF MOTION : EQUIVALENCE OF A PILLE D AND EFFECTIVE FORCES.



Z MB = Z(MB)eff; (1,8m)j x (Azi+Azk)+Mj = - 1 mabori+ 1 m 502 -1.8A, K + 1.8A, i + Moj = - = ma6xi + for b xj EQUATING THE COEFF. OF i, 1, K:

- (1.8m) Az = 1 maba (4)
- Me = = 1 n. 6 x (5)
- A, = 0

(G) ANGULAR ACCELERATION SUBSTITUTING GIVEN DATA IN (5):

36 N·m = = (5 kg)(0,6 m) X

0 = 60.0 rad/5

(1) HUTIAL EVENTER PENCHAL EO (4): (1.8 m) Az = - 1/6 (5 kg) (0.9 m) (0.6 m) (60 rad/s)

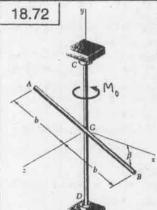
Az = - 15,00 N

RECH! ING Za (1), A=0, A = - (15,00 N) k

ZF = Z(E) = ;

A+B=0, B=-A.

E = (15,00N)K



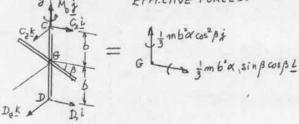
GIVEN:

ASSEMBLY OF PROB. 18.66. CONSISTING OF ROD OF MASS M WELDED TUSHAFT OD OF CENGTH 26. ASSEMBLY IS AT REST WHEN COUPLE OF MOMENT MO = MO 15 APPLIED TO SHAFT CD FIND: (a) ANGULAR ACCELERATION OF ASSEMBLY, (b) INITIAL DYNAMIC REACTIONS AT CAND D.

SEE SOLUTION OF PROB. 18,66 FOR DERIVATION OF ED. (1): H===mbwcosp(sinpi+cospj) (1)

EQ. (18.22): HG = (HG) Gays + 11 × HG = (HG) Gays + 0 SINCE Q= WEN COUPLE IS APPLIED. THUS $H_G = (H_G)_{GXYY} = \frac{1}{3} mb \propto cos \beta (sin \beta i + cos \beta j)$

EQUATIONS OF MOTION: EQUIVALENCE OF APPLIED AND EPPECTIVE FURCES.



ZM = Z(MD)ess: 26 x (C2 + C2 +) + Moj = + mb x sin san si + 1 mb x cospa -26 C2 K +26 C2 i +Moj = 1 mbasin scospi + 1 mbacospi EQUATING THE COFFF. OF 1, d, K:

- 1 26 C2 = 1 mba sinpossis (3)
- 1 Mo = 1 m + 2 x cos 2 B (4)
- (R) C, =0 (5)

(a) ANGULAR ACCELERATION

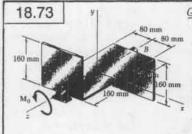
FROM EQ. (4): x = 3 M / m 6 cos B

(b) INITIAL DYNAMIC REACTIONS

PROM =0 (3): C2 = 1 mb of sin B cosp = 1 mb sin B cosp (3Mo/mbas p) Cz = (Mo/2b) tan/s

RECALLING EQ. (5), C = 0, C = (Mo/2b) tan 15 k

ΣF= Z(F)+4: C+D=0, D=-C D=-(Ma/2b)tan/k



2,4-kg COMPONENT SHOWN IS AT REST WHEN COUPLE M = (0,8 Nim) k IS APPLIED TO IT.

FIND!

(a) ANS. ACCELERATION (L) DYNAMIC REPCTIONS AT A AND 8 IMMEDIATELY AFTER COUPLE IS APPLIED

COMPUTATION OF MOMENTS AND PRODUCTS OF INERT A



TUTAL MASS = 111 = 2.4 kg, a=160 mm PORTO 11: $I_2 = \frac{1}{12} \left(\frac{mn}{2} \right) a^2 = \frac{1}{24} ma^2, \quad I_{32} = I_{23} = 0$ PORTIONS 2 AND 3: $I_{1} = 2\left(\frac{m}{4}\right)\left[\frac{a^{2}}{6} + \left(\frac{a}{2}\right)^{2}\right] = \frac{5}{14} + 1a^{2}$ $I_{yz} = 2\left(\frac{m}{4}\right)\left(\frac{a}{2}\right)a = \frac{1}{4}ma$, $I_{xz} = 0$

COMPENENT: I2= 1 ma + 5 ma = 1 ma, I3 = 4 ma, I2= 0

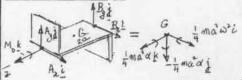
ANGULAR HOMENTUM:

H = - I = w i - I = w i + I = w k = 0 - 1 maw j + 4 maw k H= 4 Naw(-j+t)

RATE OF CHANGE:

EQ. (18.22) TIELDS, SINCE Q= WE, H = (H6) Gyy + WK x H6 = 1 maa (-j+k) + wkx + maw (-j+k) = = = mad(-i++)++ mawi $H = \frac{1}{4} ma^2 (\omega_{\underline{i}} - \alpha_{\underline{j}} + \alpha_{\underline{k}})$

EQUATIONS OF MUTICA



IM= = (MB) est: 20 Kx (Az 1 + Ay) + Mok = H 20 Ant - 20 Ay i + Mok = 1 maw 2 - 4 max + + 4 max + (3)

(a) ANG. ACCELERATION

a = 52,1 rad/s

(b) DYNAMIC REACTIONS

EQUATE COEFF. OF & IN (3): $2aA_{x} = -\frac{1}{4}mdx = -\frac{1}{4}md\frac{4Mu}{200} = -M_{0}$

 $A_3 = -\frac{M_0}{2a} = -\frac{0.8 \text{ N/m}}{2(0.16\text{m})} = -2.50 \text{ N}$

FRUATE CUEFF, OF & IN (3)!

Ay = - 1 maw (6) -2 a Ay = 1 ma w

A =- (2.50N) 1 1/HUS; SINCE W = 0, Ay = 0; THUS:

B = (2,50N)i . ZF = Z(F)+1: A+B=0, B=-A,

18.74

GIVEN: COMPONENT OF PROB. 40, 73. FIND: DYNAMIC REACTIONS AT A AND B AFTER ONE FULL REJOLUTION

SEE SOLUTION OF PRUB 12.73 FOR DERIVATION OF EWS. (4) (5), AND(6) FRON EQ.(4), 0 = 52.083 rad/5.

PUR ONE PULL REVOLUTION, 0 = 271 rad

FROM EQS. (15.16):

ω = 2 α θ = 2 (52,083 rad/s')(271 rad)= 654.49 rad/s EQ. (5): Az=-2,50N

=Q (6): Ay = - 1/8 (2.4 kg)(0.16 m)(654.49 rad/3)=-31.4N
THERE FORE:

A = -(2,50N)i - (31.4N)j ; B = -A = (2,90N)i + (31.4N)j

18.75

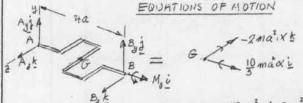
GIVEN: 16-16 SHAFT OF PROB 18.67 IS ATREST WHEN A COUPLE MO IS APPLIED TO IT, CAUSING ANGULAR ACEL. K= (20 rad/s) i. FIND: (a) COUPLE Mo.

(b) DYNHALL REACTIONS AT A AND B INMEDIATELY AFTER M. IS APPLIED.

SEE SOLUTION OF PRUB. 18.67 FOR DERIVATION OF H = 10 mawi - 2 maw k

EQ. (18.22): HG = (HG)GZHZ + 1- × HG = (HG)GZHZ + 0

 $H_G = \frac{10}{3} \text{ma}^2 \times i - 2 \text{ma}^2 \propto k$ SINCE W= X: WHERE a=9in:=0,75ft AND m= = (1616)=216



IM = Z(M) = +: 4aix(By j+B, k)+Mi= 10 max 6-2 max 6 4a \$ K-4a B j + Moi = 10 max i-2 max (3)

(a) COUPLE M. ERUATE COFF. OF ! IN ER (3):

 $M_0 = \frac{10}{3} ma^2 \times = \frac{10}{3} \frac{2/b}{32.2 Hb} (0.75H)(20 rad/s) = 2.329. lb.ft$ M=(2,33 16.ft)i

(b) DYNAMIC REACTIONS AT t=0

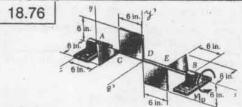
B2=0 EQUATE COFFF OF & IN EU. (3):

EQUATE COEFF. OF & IN EQ. (3): 4a By = - 9 ma' x 9 h

By = - = ma \(= - \frac{1}{Z} \frac{32,24/5}{32,24/5} \) (0,75 ft) (20 \(\text{rad/s}^4 \) = -0,466 lb THEREFORE: B = - (0,466 18) j

A =- B = (0,466 16)j ZF = Z(F): A+B=0

A = (0,466 16); B=- (0,466 16);



GIVEN!
THE 2.7-16 ASSEMBLY OF PROB. 18.68 IS AT REST
WHEN A COUPLE M. IS APPLIED TO AXLE AF CAUCING
AN ANGULAR ACCELERATION & = (150 rad/s);
PIND: (a) THE COUPLE M.

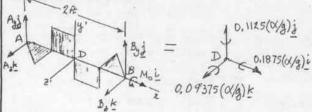
(b) THE DYNAMIC REACTIONS AT A AND B IMMEDIATELY AFTER M. B APPLIED.

SEE SOLUTION OF PROB. 18,68 FOR DECIMATION OF FW.(1):

 $\frac{H}{D} = (0.1875 \stackrel{!}{L} + 0.1125 \stackrel{!}{d} + 0.09375 \stackrel{!}{K})(\omega/g) \qquad (1)$ WHERE THE NOISERICAL WILLIES ARE EXPRESSED IN 16.542 $EQ.(18.22): \stackrel{!}{H} = (\stackrel{!}{H})_{DZ_{3}^{1/2}} + \Omega \times H = (\stackrel{!}{H})_{DZ_{3}^{1/2}} + 0$

SINCE à= a: H= (0,1875 i+0,1125 j+0.09375)(x/) (2)

EQUATIONS OF MOTION



 $ZM_A = Z(M_A)_{eff}$: $(2ft)i \times (B_j \hat{a} + B_j \hat{b}) + M_0 \hat{b} = 0.1875(a/g)\hat{b} + 0.1125(a/g)\hat{b} + 0.09275(a/g)\hat{b}$

0.09275 (24g) k (21t) By K-(21t) By 1 + Moi = 0.1875(4g) i+0.1125(2g) j+0.09375(4g) K

(a) <u>COUPLE</u> M_0 EQUATE CORPT. OF \underline{i} IN EQ. (3): $M_0 = 0.1875 (\alpha/g) = (0.1875 16.41^2) \frac{150 \text{ rad/g}^2}{32.2 \text{ ft/s}^2} = 0.873 \text{ lb-ft}$ $M_0 = (0.873 \text{ lb-ft}) \underline{i}$

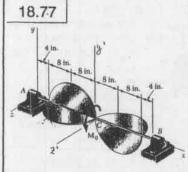
(b) DYNAMIC REACTIONS AT += 0.

(24) By = 0,09375 (x/g)=(0.09375 16.44) 150 rads = 0,43672 16.44 (a) COUPLE MO EQUATE COEFF. OF

 $B_y = 0.218 \text{ lb}$ $EQUATE COEFF, OF <math>\frac{1}{2}$ IN $EQ(\frac{1}{2})$. $-(2ft)B_z = 0.1125 (a/g) = (0.1125 \text{ lb.ft}) \frac{150 \text{ rad/s}^2}{32.2 \text{ ffs}^2} = 0.52 \text{ to 7 lb.ft}$ $B_z = -0.262 \text{ lb}$

THUS: B = (0.218 B)j - (0.262 B)k $\Sigma F = \Sigma (F)_{eff} : A + B = 0, A = -B$

A = - (0,21816) j + (0,26216) k



GIVEN:

ASSEMBLY WEIGHS
12 IL AND CONSISTS OF
4 SEHICIRCULAR PLATES.

4 SEHICIRCLAR FLATES.
ASSEMBLY IS MITTER
AT t = 0 WHEN COUPLE

Mo IS APPLIED FOR
ONE FULL PERMUTION
WHICH LASTS 25.

FIND: (a) THE COUPLEMON (b) THE DYNAMIN KENTION AT A MAN! B AT t = 0

MASS OF ASSEMBLY = $mL = \frac{12 \text{ lb}}{32,249} = 0.37267 \text{ lb.s}^2/4c$ RADINS OF SEMICIRCULAR PLATES = $E = 8 \text{ in.} = \frac{2}{3} \text{ ft}$ HOTENTS AND PROJECTS OF INEFT IA

POR ASSEMBLY: $I_2 = 2\left(\frac{mL}{2}\right)\frac{L}{2} = \frac{1}{4}mL^2$ FOR RACH VERTICAL PLATE: $I_{33} = \frac{m}{4}z\sqrt{g} = -\frac{m}{4}L\left(\frac{4L}{3T}\right)$

FOR EACH HURIZONTH PLATE: $I_{zy} = 0 \qquad I_{zz} = -\frac{4012}{3\pi}$ FOR ASSETTION: $I_{xy} = I_{zz} = 2\left(-\frac{mz}{3\pi}\right) = -\frac{2mt^2}{3\pi}$ ANGULAR

ANGULAR MONTH $\omega_{x} = \omega_{x}$, $\omega_{y} = \omega_{z} = 0$; $H = I_{x} \omega_{x}^{i} - I_{x} \omega_{y}^{i} - I_{zz}^{i}$, $\omega_{x}^{k} = m\epsilon^{2}\omega\left(\frac{1}{4}\frac{i}{k} + \frac{2}{3\pi}\frac{i}{k} + \frac{2}{3\pi}\frac{k}{k}\right)$ (1)

 $EQ.(18,22): \stackrel{\dot{H}}{\underline{H}} = (\stackrel{\dot{H}}{\underline{H}})_{Cxy'z} + \stackrel{\dot{D}}{\underline{D}} \times \stackrel{\dot{H}}{\underline{H}}$ $\stackrel{\dot{H}}{\underline{H}} = m\epsilon^* \alpha \left(\frac{1}{4} \stackrel{\dot{i}}{\underline{i}} + \frac{2}{3\pi} \stackrel{\dot{i}}{\underline{j}} + \frac{2}{3\pi} \stackrel{\dot{k}}{\underline{k}} \right) + \omega \stackrel{\dot{i}}{\underline{i}} \times m\epsilon \omega \left(\frac{1}{4} \stackrel{\dot{i}}{\underline{i}} + \frac{2}{3\pi} \stackrel{\dot{k}}{\underline{k}} \right)$

 $= m\epsilon'\alpha(\frac{1}{4}i + \frac{2}{3\pi}j + \frac{2}{3\pi}k) + \frac{2}{3\pi}m\epsilon'\omega'(k-j)$ $H = \frac{1}{4}m\epsilon'\alpha i + \frac{2}{3\pi}m\epsilon'(\alpha-\omega')j + \frac{2}{3\pi}m\epsilon'(\alpha+\omega)k$ (2)

 $\frac{2.54}{31} = \frac{2}{311} m^{\frac{1}{2}} (\alpha - \omega)^{\frac{1}{2}}$ $\frac{2}{311} m^{\frac{1}{2}} (\alpha + \omega)^{\frac{1}{2}}$

 $\Sigma M_{A} = \Sigma (M_{A})_{eff} : 52 i \times (B_{y} + B_{z} k) + M_{0} i = H_{0}$ $52 B_{y} k - 52 B_{z} + M_{0} i = \frac{1}{4} m 2^{2} \alpha i + \frac{2}{37} m 2^{2} (\alpha - \alpha) j + \frac{2}{37} m 2^{2} (\alpha + \alpha) k$ (3)

EQUATE COEFF, OF i: $M_0 = \frac{1}{4} \text{MIZ}^2 \alpha$ SINCE ASSEMBLY ROTATES THROUGH $\theta = 2\pi \text{ to A}$ IN 26! $\theta = \frac{1}{4} \alpha t^2$, $\alpha = 2\theta/t^2 = 4\pi/4 = \pi \text{ rad/s}^2$. THUS! $M_0 = \frac{1}{4} (0.37267 \text{ lb} \cdot \text{s}/\text{s}/\text{t}) (\frac{2}{3} \text{s}/\text{t})^4 (\pi \text{ rad/s}/\text{s}) = 0$; 1301 lb·ft $M_0 = (0.1301 \text{ lb·ft})^2$

(b) DYNAMIC PERCTIONS AT t=0

EQUATING THE COEFF. OF J AND & IN (=) AT 1 TIME
W=0 AND \(\alpha = \text{Trad/s}; \)

 $B - 5 \times B_z = \frac{2}{5\pi} m t' (\pi rad/s), B_z = -\frac{2}{15} (0.372.67)(\frac{2}{3}) = -0.0331 | b$ $B - 5 \times B_z = \frac{2}{3\pi} m t' (\pi rad/s), B_z = +0.0331 | b$ $B = (0.0331 | b) \frac{1}{2} - (0.0331 | b) \frac{1}{2} = 0.0331 | b$

IF=IF+1: A+=0, A=-(0.033/16)2+(0.033/16)&

18.78 GIVEN: ASSEMBLY OF PROB. 18.77

FIND: DYNAMIC REACTIONS AT A AND B AT t = 25. SEE SOLUTION OF FROB. 18,77 FOR DERIVATION OF EN (3) 5284K-5282 + Moi = 4m2 ai + 2m2 (a-w) + 2m (a+o) k WHERE m = 0,37267 16.53/ft AND . E== ft SINCE ASSENBLY ROTHIES THROUGH &= 2 ft rad in 2 3; 0 = 1 xt, x = 20/t = 41/4 = 11 rad/s AT t = 25: W = at = (Arrad/s)(25) = 27 rad/s EQUATING THE COEFF. OF & AND K IN EQ (3) AND SUBSTITUTING THE ABOVE VALUES! (1) -52 Bz = 2 m E2 (17-4174) B = - 2 (0, 37267)(3)(1-477) = +0.383 b By = 产加点 (1+471) B 5 & B, = 2 m 2 (17+471) By = 2 (0.37267)(2)(1+47) = 0,449 B B=(0.449/b)j+(0.383/b)k ΣF= Σ(F)++: A+B=0 A=-(0,44916) - (0,38316) K

18.79 GIVEN:

PLYWHEEL RIGIDLY ATTICHED TO CRANKSHIPFT OF AUTOMOBILE ENGINE IS EQUIVALENT TO 400-MIN-DIAM, 15-MIN-THICK STEEL FILL (DENSITY = 7860 Kg/m³). AUTUMOBILE TRAVELS ON UNBANKED CHEVE OF 200-M RADIUS AT GOKM/h WITH PLYWHEEL ROTATING AT 2700 pm. FIND:

MAGNITUDE OF COUPLE EXERTED BY FLYWHEEL ON CHANKS HAFT, ASSUMING AUTOMOBILE TO HAVE

- (a) REAR-WHEEL DRIVE WITH ENGINE MOUNTED LUKINSTONELY,

 $\frac{1}{2} = \frac{1}{2} (\pi 2^{l} t) = \frac{1}{2} (\pi 2^{l} t) 2^{l}$ $\frac{1}{2} = \frac{1}{2} (\pi 2^{l} t) = \frac{1}{2} (\pi 2^{l} t) (\pi 2^{l}$

ANGULAR HOMENTUM ABOUT G: H = I, W, L + I, W, i

Ea. (18.22): $\dot{H}_{G} = (\dot{H}_{G})_{GXYZ} + \dot{\Omega} \times \dot{H}_{G} = 0 + \omega_{g} \dot{t} \times (I\omega_{g} \dot{t} + I\omega_{g})$ $\dot{H}_{G} = -\dot{I}_{X} \omega_{X} \dot{\omega}_{Y} \dot{K} = -(0.29632 \, kg.m)(282.14 \, rad/s)(0.125 \, rad/s) \dot{K}$ $\dot{H}_{G} = -(10.47 \, N \cdot m) \dot{K}$

THE COUPLE EXERTED ON THE PLYWHEEL, THEREFORE, MUST BE M = HG = - (10, 47 N·m) & AND THE COUPLE EXERTED BY THE FLYWHEEL IS - M = (10,47 N·m) &

ANSWER: 10,47 N.M.

(CONTINUED)

18.79 continued

(b) PRONT-WHEEL DRIVE (TRANSVERSE MOINTING)

18 WE ASSUME THE GAME DIRECTION

OF MOTION OF THE CASE & IN PROTE

 WE ASSUME THE SAME DIRECTION OF MOTION OF THE CAR AS IN PARTA REFERRING TO THE NUMERICAL VALUES FOUND IN PAINT a: $\omega_2 = 282.74 \text{ rad/s} \\
\omega_3 = 0.125 \text{ rad/s} \\
\bar{L} = 0.29632 \text{ Kg·m}^2$

ANGULAR MOMENTUM ABOUT 6:

 $\begin{array}{l} H_{G} = \bar{I}_{g}\omega_{g}\dot{j} + \bar{I}_{g}\omega_{g}\dot{k} \\ EO. (18.22)! \quad \dot{H}_{G} = (\dot{H}_{G})_{GN/2} + 2\times H = 0 + \omega_{g}\dot{j}\times(\bar{J}_{g}\omega_{g}\dot{j} + \bar{J}_{g}\omega_{g}\dot{k}) \\ \dot{H}_{G} = \bar{I}_{g}\omega_{g}\omega_{g}\dot{i} = (0.29632\,kg.m)(0.025rcd/s)(282.74\,rad/s)\dot{i} \\ \dot{H} = (10.47\,Nm)\dot{i} \end{array}$

THE COUPLE EXEUTED ON THE FLYWHEEL, THEREFORE,

MUST BE M = Hg = (10.47 N·m)1, AND THE COUPLE

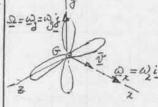
EXERTED BY THE FLYWHEEL IS M = -(10.47 N·m)i

ANSWER! 10,47 N·m

18.80 | GIVEN:

FOUR-BLADED AIRPLANE PROPELLER
WITH MI = 160 kg AND E = 600 mm ROTATES AT 1600 pm.
AIRPLANE IS TRAVELING IN CIRCULAR PATH WITH
C = 600 m AT V = 540 km/h.
FIND:

MAGNITUDE OF COUPLE EXERTED BY PROPELLER ON



WE ASSUME SENSE SHANN
FOR ω_2 , ω_3 , AND $\overline{\omega}$ $\overline{v} = 540 \text{ km/h} = 150 \text{ m/s}$ $\omega_2 = 1600 \text{ rpm} \left(\frac{277 \text{ rad}}{60 \text{ s}}\right)$ $\omega_3 = \frac{7}{6} = \frac{150 \text{ m/s}}{600 \text{ s}} = 0.25 \text{ rad/s}$

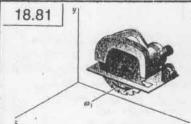
J=mk2=(160kg)(0.8m)2=102,4 kg.m2
ANGULAR MOMENTUM ABOUT G:

 $H_{G} = \bar{J}_{1} \omega_{2} \hat{i} + \bar{J}_{2} \omega_{3} \hat{a}$ $EQ (18.22): \dot{H}_{G} = (\dot{H}_{0})_{G} \omega_{1} + \Omega \times H_{G} = 0 + \omega_{3} \hat{j} \times (\bar{J}_{1} \omega_{2} \hat{i} + \bar{J}_{2} \omega_{3} \hat{a})$ $\dot{H}_{G} = -\bar{J}_{2} \omega_{2} \omega_{3} \hat{k} = -(102.4 \text{ kg·m}^{2})(167.55 \text{ rad/s})(0.25 \text{ rad/s}) \hat{k}$

 $\frac{H}{G} = -(4.289 \text{ N·m}) K = -(4.29 \text{ KN·m}) K$

THE COUPLE EXERTED ON THE PROPELLER, THEREFIXE, MUST BE M = H = - (4,24 KNim) K, AND THE COUPLE EXERTED BY THE PROPELLER ON ITS SHAFT IS - M = (4.29 KNim) K.

ANSWER: 4.29 KN.M

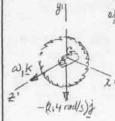


FOR BLADE AND ROTOR OF MOTOR OF PORTABLE SAW!

W=2,51b, K=1.5 in. BLADE ROTATESA'S SHOWN AT RATE W, = 1500 rpm

FIND: COUPLE M THAT WOKKER MUST EXERT ON HANDLE TO ROTATE SAW WITH CONSTANT Q = - (2,4 rad/s) }.

USING AXES CENTERED AT MASS CENTER G OF BLADE AND ROTOR AND ROTATING WITH CASING!



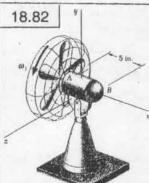
W= W1 = 1500 rpm (217 rad/s = 507 rad/s Q = Wy = W2 = -2.4 rods \overline{I}_{2} , $=\frac{W}{7}\overline{k}^{2} = \frac{2.5 \text{ /b}}{32.2 \text{ f//s}} \left(\frac{1.5}{12} \text{ ft}\right)^{2}$ = 1,2/3/X/03/bift.5

ANGULAR MOMENTUM A EUUT GI

 $H_C = I_y, \omega_y j + I_z, \omega_z k$

EQ. (18,22): = I, Q, W, i = (1,2131×10 16.fts)(-2,400d/s)(507 rad/s) i

THE COUPLE THAT THE WORKER MUST APPLY IS M = HG M =-(0.457 16.ft) i



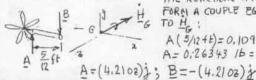
GIVEN:

FOR BLADE AND RUTOR OF MOTOR OF OSCILLATING FAN: W= 800, k = 3 in. BEARING SUPPORTS AT A AND B AKE 5 III. APART. BLADE RUTHTES AT RATE W = 1800 rpm.

FIND !

DYNAMIC REACTIONS AT A AND B WHEN MUTOR CASING HAS ANG. VEL. W= (0,6 rod/s) j.

ANGULAR MOMENTUM A FOUT MASS CENTER; Ho= Izwi+Izwj = Izwi+Izwj EQ. (18,22): $H_G = (H_G)_{GXYZ} + \Omega \times H_G = O + \omega_Z j \times (\bar{j} \omega_i i + \bar{j} \omega_j j)$ $H_G = -\frac{7}{2}\omega_1\omega_2 = -\frac{(8/16)^4b}{322795}(\frac{3}{12}5t)^2(1800 rpm)\frac{27100}{605}(0.6 rad/s) \pm$ = - (0, 10976 16.ft) k



THE REACTIONS AT A AND B FORM A COUPLE EQUIVALENT

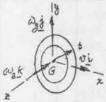
A (5/12+E) = 0. 10976 16.ft A= 0,26343 1b = 4.21 07

GIVEN: 18.83

AUTO MOBILE TRAVELS AROUND UNBANKED CURVE WITH Q= 150 M AT SPEED V = 95 km/h.

PUR EACH WHEEL: M = 27 kg, DIAM, = 575 mm R= 225mm. TRANSVERSE DISTANCE BETWEEN WHEELS = 1.5 m.

FIND! ADDITIONIL REACTION AR EXERTED BY FROUND ON EACH OUTSIDE WHEEL DUE TO MOTION OF CAR.



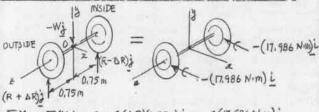
FOR EACH WHEEL! V = 95 Km/h = 26.389 m/s N = 26.369 01/3 =0,17593 rade 2 = - 26.389 m/5 = - 91.787 rad/s I=mk=(22kg)(0,225m)=1.1138 kg.ni

ANGULAR MOMENTUM OF EACH WHEELS H = 1, w, j + 1, w, k

EQ.(18.22): H=(H) +QXH=0+ajx(Iaj+Ia,k)

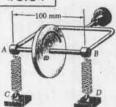
H= I w w = 1: (1.1138 kg. m²)(0,17593 rad/s)(-91.787 rad/s) =-(17.986 N·m) i

EQUATIONS OF MUTION FOR TWO WHEELS ON SAME AXLES



ZM = Σ(Mo)ey: -2(ΔR)(0.75m) = -2(17.986 N·m)] AR = 23,98 N DR = 24.0 NT



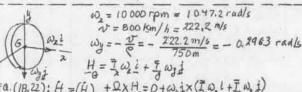


GIVEN:

TYPE OF AIRCRAFT TURN INDICATOR. UNIFORM DISK: M = 2008, 2= 40 mm SPINS AT RATE OF 10000 FPM. EACH SPRING HAS 500-N/M CONSTANT. SPRINGS EXERT EQUAL FORCES ON YOKE AB IN STRAIGHT FLIGHT PATH. FIND:

ANGLE OF ROTATION OF YOKE IN HORIZ-

ONTAL TURN OF 750-M RADIUS TO THE RIGHT WITH U= BOOKM/h. DOES A MOVE UP OR DOWN?



Ea.(18.22): $H_{G} = (H_{G})_{G+g_2} + \Omega \times H_{G} = 0 + \omega_{g_2} \times (\overline{I}_{G})_{g_2} + \overline{I}_{g_3} \omega_{g_3}$

H=-Iwwyk=-=(0.2 kg)(0.04m)(1047,2 rod/s)(-0.2963 od/s)k

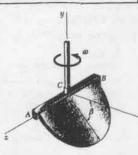


WEHAVE (0.1m) F = H = 0.049645 N·m F = 0.49645 N

F=kz: z= 0.49645N = 0.9929 mm $\theta = \frac{Z}{6A} = \frac{0.9929 \, \text{mm}}{50 \, \text{mm}} = 0.01986 \, \text{rad} = 1.14$

SINCE SPRING AT A PULLS DOWN, A 15 MOVING U.P.

18.85 and 18.86



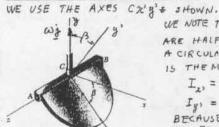
GIVEN:

SEMICIRCULAR PLATE WITH &= 120 nim 15 HINGED TO CLEVIS WHICH ROTATES WITH CONSTANT W.

PROBLEM 18.85

FIND: (a) B WHEN W= 15 rad/s. (b) LARGEST W FOR WHICH PLATE REMAINS VERTICAL (\$=90) PROBLEM 18.86: FIND W FOR WHICH B = 50°.

MOMENTS AND PRODUCTS OF INTERTIA



WE NOTE THAT I, AND I, ARE HALF THUSE FOR A CIRCULAR PLATE, AND 50 15 THE MASS M. THUS I, = 1 mt Iy = 1 m2 BECAUSE OF SYMMETRY ALL PRODUCE OF INSPI ARE EQUAL TO ZERO!

I,y, = I, = I = 0

ANGULAR HOMENTUM ABOUT C

 $H_c = I_1, \omega_2, i' + I_1, \omega_1, \delta$ = 1 m2 (-w sin p) i' + 1 m2 (w cos p) i = 4m2w (-sin Bi+2cospj')

SINCE C IS A FIXED POINT, WE CHIVUSE EQ. (18.28): EM = (Hc)cx'y'2+ 1x Hc = 0 + wix Hc

OR, SINCE &= -L' SINB + j'cos B:

ZMc = ω(-i'sinp+j'cosβ)×+mzω(-sinβi'+2cosβj') $= \frac{1}{4} m \epsilon^2 \omega^2 \left(-2 \sin \beta \cos \beta k + \cos \beta \sin \beta k \right)$ ZM = - 4 me w sin posp K

BUT \(\Sigma \in - mg \vec{x} \) cosp \(\text{t} \)
= - mg \(\frac{4z}{377} \) cosp \(\text{t} \) (2) EQUATING (1) AND(2):

1 m 22 w sin 3 cos \$ = 4 mg2 cos \$

 $\omega f sin\beta = \frac{16}{3\pi} \frac{g}{2} = \frac{16}{3\pi} \frac{9.81 \text{ m/s}^2}{0.12 \text{ m}}$ asin B = 138,78 5 (3)

PROBLEM 18.85

(a) LET W=15rad/s IN(3): sin = = 138,78 = 0,61681

(b) LET \$=90° IN(3): W=138.7852 W=11.78 rad/s

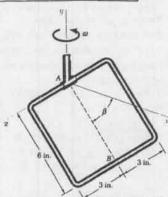
PROBLEM 18.86

LET B = 50° IN EQ. (3):

 $\omega^2 = \frac{138.78 \, \text{s}^{-1}}{5 \, \text{in } 50^\circ} = 181,17 \, \, \text{s}^2$

W = 13,46 rad/s

18.87 and 18.88



GIVEN:

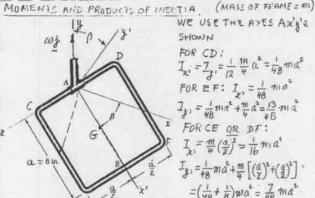
ROD BENT TO FORM 6-in. SQUARE FRAME WHICH IS ATTACHED BY COLLAR AT A TO SHAFT ROTATIVE WITH CONSTANT W.

PROBLEM 18,87:

(a) B WHEN W= 9, Bradis (b) LARGEST W FOR WHICH B = 90°.

PROBLEM 18.88:

FIND W FOR WHICH B= 48°,



WE USE THE AYES AX'4'2 SHOWN FOR CD: $I_{2} = \frac{7}{3} = \frac{1}{12} \frac{m}{4} \alpha^{2} = \frac{1}{48} m \alpha^{2}$

FOR EF: 1, = 1 ma I_3 , = $\frac{1}{48}ma^2 + \frac{m}{4}a^2 = \frac{13}{48}ma^2$ FORCE OR DF:

 $I_{2} = \frac{m}{4} \left(\frac{\sigma_{2}}{2} \right)^{2} = \frac{1}{16} m_{1} a^{4}$ $I_{g} = \frac{1}{48} m a^{2} + \frac{m}{4} \left[\left(\frac{a}{2} \right)^{2} + \left(\frac{a}{2} \right)^{2} \right]^{4}$

=(48+8)ma = 7 ma

FOR ENTIRE FRAME: $I_{23} = \left[\frac{1}{4B} + \frac{1}{4B} + 2\left(\frac{1}{16}\right)\right] ma^{2} = \frac{1}{6} ma^{2}; I_{g} = \left[\frac{1}{4B} + \frac{13}{8B} + 2\left(\frac{7}{4B}\right)\right] ma^{2} = \frac{7}{12} ma^{2}$ BECAUSE OF SYMMETRY: Ixig = Ix = Ix = 0

ANGULAR MOMENTUM ABOUT A

H= Ix, w2, i'+ Iy, w, i'= 1 ma'(-wsin B)i'+ 7 ma'(wwsB)j SINCE A IS PIXED, WE USE ED. (18.28);

ZMa=(HA)AX'V'Z+ IX XHA= O + WJ X HA OR, SINCE & = - i'sing + i'cosp :

EMA = ω (-i's in β + j'cosp.) x /2 maω (-2 sinβ i' +7 ωςβ j') = 12 mater (-75in/scosp k +2 cosps sin/s k)

IMA = - 5 maw sin Boosp K (1)

BUT ZMA = - mg (cosp k EQUATING (1) AND (2): 5 1110 Wsin B cos 3 = - 1 mgacos 3

Wisin B = 6 4 = 6 32.2 ft/s (6/12)ft wsin B=77.28 52 (3)

PROBLEM 18.87 (a) LET $\omega = 9.8 \text{ rad/s}$ IN (3): $\sin \beta = \frac{77.18}{(9.8)^2} = 0.80466$ B= 53.6"

(b) LET B=90 IN (3): W= 77.28 52

W = 8,79 rad/s

PROBLEM 18,88

LET B= 48° IN EQ. (3): W= - 77.285 = 103,995

W= 10,20 rad/5 €

18.89 and 18.90 = 400 mm

GIVEN:

950-4 GEAR A CONSTRAINED TO ROLL ON FIXED GEAR B. BUT FREE TO ROTATE ABOUT AD. AXLE AD CONNECTED BY CLEVIS SHAFT DE WHICH ROTATES WITH CONSTANT WI. (GEAR A CAN BE CONSIDERED AS THIN DISK.)

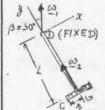
PROBLEM 18,89:

FIND LARGEST ALLOWABLE W. IF GENK A IS NOT TO LOEF COLLIACT WITH GEAR B.

PROBLEM 18.90:

FIND FORCE F EXERTED BY GEAR B ON GEAR A WHEN W, = 4 rad/s. (F 15 1 CD.)

ANCULAR VELOCITY OF GEAR A



WE USE THE AXES GZ & SHOWN AND EXPRESS THAT Y = 0:

U = WXDC = 0

w= (w, cos p+ wz) j+w, sing i DC = - (Lj +EL

THUS:

 $N = -[(\omega, \cos\beta + \omega_z)j + \omega, \sin\beta i] \times (Lj + \varepsilon i) = 0$

 $(\omega_1 \cos \beta + \omega_2) \approx k - (\omega_1 \sin \beta) L k = 0$

THUS: $\omega_1 \cos \beta + \omega_2 = (\omega_1 \sin \beta)(L/2)$ SUBSTITUTE INTO (1): (2) ω = ω, sinβ(++ + 1)

ANGULAR MOMENTUM ABOUT D

TD= Ix wzi+Iy wy = m(k+ &) w, sinp i+ m & w, & sin Bi Hp=mw, sinp[(1+2) + +2 2 L)

SINCE D IS A FIXED POINT, WE USE EQ. (18.28):

ZM = (H) + 1 × HD = 0 + (W, sin Bi + W, ws Bi) × HD = mw2 sinB[= ELsinp - (L+ + +) cos [3] k (4)

PROB. 18.89: WHEN FORCE EXERTED BY GEAR B ON GEAR A DECOMES ZERO!



(5) ZMD = - mg Lsin & K EQUATING (4) AND (5)1 m Wisin B [+ 2 Lsinp - (= + +) cosp] = - my LsinB Wi [(Li+ 12) cos B - 1 2 Lsin B] = gl WITH L= 0,4 m, t= 0.08 m, p=30, 1= 9.8/n/2 O. 13:95 N, = 3.924 w,= 5,45 rad/5 €

PROB. 18,90:

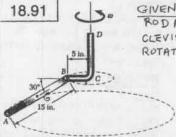
MOMENT OF FIRE TD = + FVL + +22 K THUS, ED. (5) A BONE MUST BE REPLACED BY ZMp = (FVL2+22-mgLsinp)k EQUATING (4) AND (6):

mw, sing [theing- (L+th) cosp)

= FVL+2 - mg Lsinp WITH L=0.4m, 2=0.0 am, 13=30, g= 9.81ms, m=0.95kg, co= 4 rad/s: 0.95 (4) sin 30° (-0.13195) = F VO. 1664-(0.95)(9.81)(0.4) sin 30°

0,40792 F = 1,863 9- 1,0028 = 0,8611-

tan r=字=0,2、 x=11,31°, B-8=18,7° F=2,11N& 18,7°



GIVEN:

ROD AB IS ATTACHED BY A CLEVIS TO ARM BCD WHICH ROTATES WITH CONSTANT W.

FIND:

MAGNITUDE OF 00

LET L = 15 in. = 1,25 ft THEN: BC= 5 In. = L/3

ANGULAS HOHENTUM ABOUT G

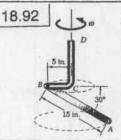
co = wsin 30 i + w cos30 }

H= Iwi+Iwj= 0+ 1m Lw 65303 = 0+ Wx H = (Wsin30 i+ wws30 j) x 1 m L wcos30 j 12 mL w sin 30° cos 30 k

EWVATIONS OF MOTION

ma=mEQ2

+) ZMB= Z(MB)++: mg (\$ 10530") = HG + (ma) (\$ 51930"). 1 mgL cos30 = 1 ml wsin30 cos30 + m (2 cos30 + 1) w (2 sin30) 1 \$ cos 30 = (3 sm 30 cos 30" + 6 sin 30") W 32.2 ft/s cos30 = 0.2276702, W= 48,994 w=7.00 rad/s



GIVEN;

ROD AB IS ATTACHED BY A CLEVIS TO ARM BCD WHICH ROTATES WITH CONSTANT W.

FIND: MAGNITUDE OF @ LET L= 15 in. = 1,25 fb THEN: BC =5 in. = L/3

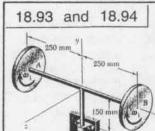
ANGULAR MOMENTUM ABOUTG:

w=-wsin30°i+w cos30° } H=Iwi+I,wj=0+12m12wcos.30j (-wsin30 i+ wws30 j) x 1 mLw cos30 j H = - 12 91 (W sin 30 cos 30 K

ERUATIONS OF MOTION ma-mrw T= = cos 30 - = 1 cos30 mg

+) EMB= E(MB)++ : mig ({ cos30)= HG + (mā)({ sin30) 1 mg L cos 30 = 17 mL w's in30 as 30+ m(L cos 30- 1/3) w (1/2 sin 30) 12 = cos 30° = (3 sin 30° cos 30 - 2 sin 30°) w

1 32,2 H/3 cos30° = 0.061004 02, ω= 182.85 w= 13.52 rod/s



FOR EACH DISK: M = 5 kg, & = 100 mm \(\omega_1 = 1500 \text{ rpm} \) PROB 18.93:

FOR W2 = 45 rpm FIND
DYNAMIC REHCTIONS
HT C AND IF
(a) BOTH DISKS ROTHTE
AS SHOWN
(b) DIRECTION OF SPIN
OF B IS REVERSED

PROB. 18,941

FIND MAX. ALLOWABLE WZ

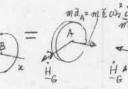
IF DYNAMIC REACTIONS

AT C AND D ARE NOT TO

EXCEED 250 N EACH.

ANGULAR MOMENTUM OF EACH LISK ABOUT ITS HASS CENTER

EQUATIONS OF MUTION



SINCE MÀ AND MÀ CHICEL OUT, EFFECTIVE FILCES REDUCE TO COUPLE 2H = m & W, W, K

ma==- m210=1

IT FOLLOWS THAT THE REACTIONS FORM AN EQUIVALENT COUPLE WITH $-C = D = (m \epsilon \omega)_1 \omega_2 / 0.3 \text{ m}) i$ (3)

PROBLEM 18,93

(0) WITH m = 5 kg, &= 0.1 m, W = 1500 rpn = 50 il rad/s, AND

W = 45 rpm = 1.5 Tl rad/s, Eo. (3) YIELDS

C = D = (5 kg)(0.1 m) (50 Tl rad/s)(1.5 Tl rad/s) V 0.3 m) = 123,37 N

C=-(123,4N) L; D=(123,4N) L

(b) WITH DIRECTION OF SPIN OF B REVERSED, ITS
ANGULAR MOMENTUM WILL ALSO BE REVERSED AND
THE EFFECTIVE FORCES (AND, THUS, THE APPLIED
FORCES) REDUCE TO ZERO!

C = D = 0

PROBLEM 18,94

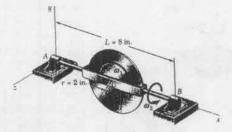
THE NOTE C = D = 250 N IN EQ. (3) YIELDS $m \varepsilon^2 \omega_1 \omega_2 = 250 N$

WITH M= 5 kg, &= 0.1 m, W, = 1500 rpm = 50 TT rad/s

 $\omega_2 = \frac{(250 \text{ N}(0.3 \text{ m}))}{(5 \text{ kg})(0.1 \text{ m})^2 (50 \text{ Trad/g})} = 9.5493 \text{ rad/s}.$

 $\omega_z = 91.2 \text{ rpm}$

18.95 and 18.96



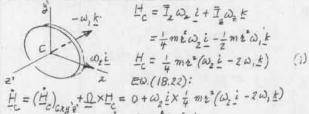
GIVEN: 10-02 DISK SPINS AT RATE W. = 750 pm

FOR W2 = 6 ron/s FIND THE DYNAMIC REACTIONS AT A AND B.

PROBLEM 18,96:

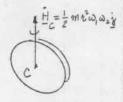
FIND MAX. ALLOWABLE W. IF DYNAMIC REACTIONS AT A AND B ARE NOT TO EXCEED 0,25 16 EACH.

ANGULAR MUMENTUM PEOUT C



He = 7 me a, we have

A A B B



(2)

 $\sum M_{A} = \sum (M_{A})_{cij} : BL = \frac{1}{L} m t^{2} \omega_{i} \omega_{L} \qquad A = B = \frac{m t^{2} \omega_{i} \omega_{z}}{2L}$ (3)

PROBLEM 18.95 LETTING $m = \frac{W}{3} = \frac{(10/16)16}{32.2 \, \text{M}_{5}} = 0.01941 \, \text{16.5}/\text{ft}, \ \xi = \frac{1}{6} \, \text{ft}, \ L = \frac{2}{3} \, \text{ft}$ $\omega_1 = 750 \, \text{rpm} = 25 \, \text{ft} \, \text{rad/s}, \ \omega_2 = 6 \, \text{rad/s} \, \text{(N Eq. (3))};$

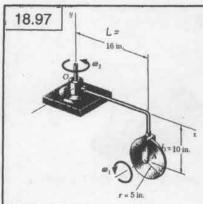
A = B = (0,01941 16.51/ft)/1/1/1/257 rod/s (6 md/s) = 0,1906 16

A = (0,1906/6)k; B = -(0.1906/6)k

PROBLEM 18.96

LETTING A = B = 0,25/b, m = 0.01941 | $b \le 1/4t$, $b = \frac{1}{6} 5t$, $L = \frac{2}{3} 5t$ AND $\omega_1 = 750$ FPM = 25π ray/s IN EQ. (3) AND SOWING
POR ω_2 :

 $\Omega_2 = \frac{2(\frac{2}{3}ft)(0.2516)}{(0.019411659ft)(\frac{1}{6}ft)^2(2517 \text{ mid/s})} = 7.872 \text{ rad/s}$ $\omega_z = 7.87 \text{ rad/s}$

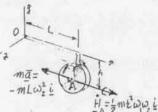


DISK OF WEIGHT W=BIL ROTHTES OT CONSTIANT W,= 12 mad/s. ARM DA RUTHTES HT CONSTANT W2 = 4 mod/s.

FIRCE - COUP. & SYSTEM REPRESENTING DYNAMIC REACTION ATSUPPORTO.

ANGULAR HOMENTUN ABOUT A

EQUATIONS OF MOTIC



IF= E(F) + : R = -mLwzi

ZM = Z (Mo)eff: MR = HA + (Li-hj) × (-mLWz'i) = 1 m & wiw i - m hLw k. (4)

WITH GIVEN DATA:

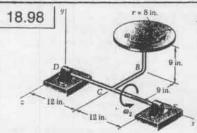
 $m = \frac{W}{g} = \frac{816}{32.2464} = 0.24845 \text{ 16.5} \text{ 15.5} \text{ 15.5} \text{ 16.5} \text{ 16.$ W = 12 rad/s, Wz = 4 rad/s, 2 = 12 ft.

EQ.(3): R = - (0.24845 16.5/ft) (4 rad/s)" = - (5,300 16) L

EQ.(4): MR = 1 (0.24845 16.5/4+)(5/12 ft) (12 rad/s)(4 rad/s) 1 -- (0,74845/bis3/ft)(\$ft)(\$ft)(4rad/s)2k

= (1.0352 16.5t) i - (4.417 16.ft) k

FORCE-COUPLE AT O: R=-(5,30 lb)i 3 MR=(1.035 lb,5t)i-(4,42 lb.ft)k

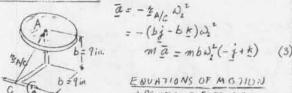


GIVEN DISK OF WEIGHT W=616 ROTATES AT CONSTAILT WI = 16 rolls ARM AGC IS WELDED TO SHAFT DUE WHICH ROTATES AT CONSTANT W. = 8111/5.

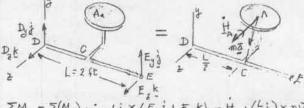
FIND: DYNAMIC REACTIONS AT DANDE.

ANGULAR MONFHTUM AROUS A HA=I, WEL+ In, W. = 上加とるシュナーかかいう H = + m/2 (Wz 1+2W1) HA = (HA) AXY + D × H = 0 + WZ L XHA HA = Wzix + m& (Wi+2W,j) Ha = 1 mix w, w, k (2)

EFFECTIVE FORCE 111 a



APPLIED PORCES ARE
HUNGENT TO EFFECTIVE FORCES



ZMD = Z(MD)esi: Li x (Exi+ = k) = Ha + (Zi) x niz REULLING EUS. (2) AND (3): Lix(E, i+ E, k) = 1 m2 w, W= K + 1 i x mbw2 (-i+k)

LE, K-LE, j = 1 m2 w, w, k-1 mbl w, k-1 mbl w. }

FOUNTING COEFF, OF UNIT VECTORS! $E_y = \frac{1}{Z} m \left[\left(\frac{\epsilon^2}{L} \right) \omega_1 \omega_2 - b \omega_2^2 \right]$ E = 1 m b 10 2 WITH GIVE' DATA: m = 616. = 0.18634 165/5t, 2=3ft,

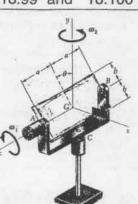
L=2ft, b=0.75ft, W, = 16 rad/s, Wz= 8 rad/s 1 Ey = -1.022 1b, Ez = 4.472 16

E=-(1,82216)j+(4,4716)k < ZF=ZFett: D+E=ma

RECALLING (3) AND - GIVEN DATA! D= ma - E= mbw; (-j+k)-E-= (0,18634)(0.75)(8) (-j+k) + (1.822 b) j-(4.472 lb) k = (1.821-8,944)) + (8.944-4,472) k

D=-(7.1216)j+(4.4716)k

18.99 and 18.100

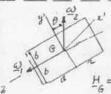


GIVEN:

ADVERTISING PANEL m = 48 ks, 2a = 2.4m, 2b=1.6m. MOTOR AT A KEERS PANEL ROTATING ABOUT AB AT CONSTANT RATE W .. MOTOR AT C KEEPS FRAME RUTATING AT CONSTANT W2 . PANEL COMPLETES FULL REVOLUTION IN 65. PRAME COMPLETES FULL REVOLUTION IN 125, PROBLEM 18,99: EXPRESS DYNAMIC REACTION AT D AS FUNCTION OF A.

PROBLEM 18, 100: SHOW THAT (a) DYNAMIC REACTION AT D IS INDEPENDENT OF LENGTH 2a.

(b) AT ANY INSTANT MI/M = WE/ZWI, WHERE MI AND M2 PARE THE MACHITUDES OF THE COUPLES EXERTED BY THE MUTANS AT A AND C, KE' PECTIVELY.



USING AXES GXY'& WITH Z' PERPENDICULAR TO PANEL ! Wz = Wz sin 0, Wz = Wz cos 0, Wz = W, $H_G = I_x, \omega_x, \omega + I_x, \omega_y, \delta + I_z \omega_z k$

H = 1m (a+b) w, sind i'+ + 1 m a w cos + j + 5 m b w, k

TO THE ORIGINAL FRAME GXYZ, WE NOTE 1'= costi + sint) i' = -sinti + cosoj

SUBSTITUTE IN (1):

Ha = = m (a+b) W, sind (cost i + sind)+ + = matuz coso (-sint) + coso) + + mbwik

H = 1 m (6 w, sin + cost i + (a + b sin + b) w j + b w, k)

THE FIRST TERM IS OBTAINED BY DIFFERENTIATING &. WITH RESIECT T , ASSUMING FRAME GRYZ TO BE FIXED!

-6 GRYZ = 5 m [6 Wz (coso-sin20) oi + 26 wz sino wsf of] DBSTEVING THAT B = W, AND SUBSTITUTION HATO (2):

H= 1mbw, wz (cost -sint) i+ & sint cost j]+ Wejx = mb (w, sin & cobi + w, k)

H = = 1 m b w w 1 (cos 0 - sin't) i + 2 sin 8 cos 0 } +

I milb (Wiwzi - Wisin troosb K).

(4) $H_{c} = \frac{1}{3} mb \left(2\omega_{1} \omega_{2} \cos \theta i + \omega_{1} \omega_{2} \sin 2\theta j - \frac{1}{2} \omega_{2}^{2} \sin 2\theta k \right)$ LEATTER AT I FINST BE EQUAL TO COUPLE HA PROBLEM 18.99, WITH GIVEN DATA:

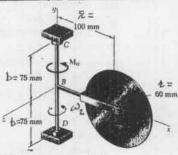
1 = H = 3 (48 kg) (0.8 m)2 [2 (2)/27 kcs Di + (21)/27) sinzej - 2 (27) sinzej +

M = (11,23 Nim) cos di + (5,61 Nim) sin 20 j - (1,404 Nim) sin 20 k PROBLEM 18,100

(a) Eq. (4) DOES NOT CONTAIN Q

(b) FROM(4): M= 1 mb w2 sin 20, M2= 3 mb w2 sin 20 MI/MZ = WZ/2WI

18.101 and 18.102



PROBLEM 18.1012

BIVEN: 3- Ky DISK SPINS AT CONSTANT W, = 60 rad/siq ARM AB AND SHAFT ARE AT REST WHEN M=(0.40 N.m) } 15 APPLIED FOR 25.

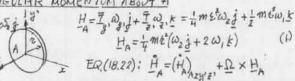
DYNAMIC REACTIONS AT C AND D AFTER MO IS REMOVED.

(2)

PROBLEM 18, 102

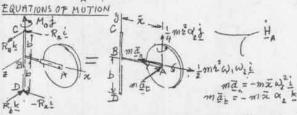
GIVEN: 3-Kg DISK SPINS AT CONSTANT W, = 60 rad/s. ARM AB AND SHAFT ARE AT REST WHEN M IS APPLITE FOR 3 5. WITH ANG. VELOCITY OF SHAFT REACHING 18 rad/s. FIND! (a) Mo, (b) DYNAMIC REACTIONS, AT C AND D AFTER M IS REMOVED.

ANGULAR MOMENTUM ABOUT A



SINCE DISK HAS AN ANG. ACCEL. OF = 021, WE HAVE ALSO, Q = W28

mt x, j + w, 2 x + m 2 (W, j + 2W, E)



FROM SYMMETRY AND INSPECTION OF EFFECTIVE PORTES, WE FIND THAT THE COMPONENTS OF THE REACTIONS AT CAND DARE EQUAL IN MAGNITUDE AND DIRECTED AS SHOWN. $ZM_y = \sum (M_y)_{eff}: M_0 = \frac{1}{4}mt^2\alpha_1 + \sum (m\bar{\lambda}\alpha_2) = m(\frac{1}{4}z^2\pi^2)\alpha_2$ $M_0 = (3kg)[\frac{1}{4}(0.06m)^2 + (0.1m)^2]\alpha_2$ $M_0 = 0.0327\alpha_2$ (5) $\sum F_{x,i} = \sum (F_x)_{eff}$: $2R = m \times \omega^2 = (3kg)(0.1m) \omega_2^2$, $R_i = 0.15 \omega_2^2$ ZM= Z(M), 1: 26R, = 1 m & W, W2

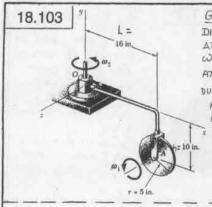
 $R_z = (mz/46)\omega_1 \omega_2 = \frac{(3 \text{ kg}/0.06 \text{ m})(\text{kp rad/s})\omega_2}{4(0.075 \text{ m})}$, $R_z = 2.16 \omega_2$ (5)

PROBLEM 18,101 $\frac{1}{LZT} \frac{N}{N}_0 = 0.40 \, \text{N·m} \, \text{IN (3)} : \alpha_2 = \frac{0.40}{4.0327} = 12.232 \, \text{rad/s}^6$ FOR t=25: 02= 02t=(12.232 rad/51)(25) = 24.464 rad/s EGS. (4) AND(5): Rx = 0,15 W= B9,8 N; Rx = 2,16 W, = 52,8 N C=-(89,8 N) i+(52,8 N)k; D=-(89,8 N)i-(52,8 N)k

PROBLEM 18.102

W2 = 0, t: 18 rad/s = 0, (35), 0 = 6 rad/5 (a) E0.(3): M = 0.0327(6) = 0.1962 N·m M=(0.1962 N·m) & (b) EQ. (4): R = 0.15 (18 mod/s) = 48.6 N

ER(5)1R= 2,16(18 rad/s) = 38,88 N C=-(48,6N):+(38,9 H)k; D=-(48,6N):-(3B,9 N)k



DISK OF WEIGHT W= 8 lb.

AT INSTANT SHOWN ω_1 = 12 rad/s AND DEREASES

AT RATE OF 4 rad/s*

DUE TO BEARING FRICTION,

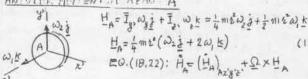
ARM OA RUTHTES AT

CONSTIANT ω_2 = 4 rad/s.

E FIND:

FIND: FORCE-COURESYSTEM REPRESENTING DYNAMIC REACTION ATSUPPORT 0.

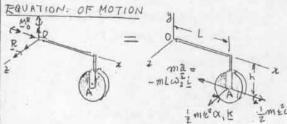
ANSVIEW HELLENT OF AROUT A



WHERE THE FIRST TTOP IS OBTAINED BY DIFFECENTIATING HA ASSUMING THE FRAME AXY'Z'TIZE FIXED:

(HA)AX'Y'Z = 1 m2 w/k = 1 m2 x/k with x = 4 rads

Thus: $\underline{H}_{A} = \frac{1}{2} m x^{2} x_{1} \underline{k} + \omega_{2} \underline{1} \times \frac{1}{4} m x^{2} (\omega_{2} \underline{1} + 2\omega_{1} \underline{k})$ $\underline{H}_{A} = \frac{1}{2} m x^{2} (\omega_{1} \omega_{2} \underline{i} + \alpha_{1} \underline{k})$ (2)



 $\Sigma \vec{F} = \Sigma(\vec{F})_{eff}; \quad \vec{R} = -m L \omega_z^L \dot{L}$ $\Sigma \vec{M} = \Sigma(\vec{M}_0)_{eff}; \quad \vec{M}_0^R = \dot{H}_A + (L \dot{L} - h \dot{J}) \times (-m L \omega_z^L \dot{L})$ $\underline{M}_0^R = \frac{1}{2} m z \dot{\omega}_1 \omega_z \dot{L} + \frac{1}{2} m z^2 \sigma_1 \dot{K} - \epsilon l h L \omega_z^L \dot{K}$ $M_0^R = \frac{1}{2} m z^2 \omega_1 \omega_z \dot{L} + m (\frac{1}{2} z^2 \sigma_1 - h L \omega_z^L) \dot{K}$ (4)

WITH GIVEN DATA: $m = \frac{W}{3} = \frac{B/b}{32.24t/s} = 0.24845 |b.s^2/st, L = \frac{4}{3}ft, h = \frac{5}{6}ft, t = \frac{5}{12}ft$

W, = 12 rod/s, X, = - 4 rad/s2. W_ = 4 rad/s

EQ.(3): $R = -(0.24845 /b.5^2/ft)(\frac{4}{3}ft)(4 rad/s)^2 = -(5.300 /b) i$

 $EQ (4): M_{0}^{R} = \frac{1}{L} (0.24845/b.5/ft) (\frac{5}{12}ft)^{4} (12 \text{ rad/s}) (4 \text{ rad/s}) + \\
+ (0.24845/b.5^{3}ft) [\frac{1}{L} (\frac{5}{12}ft)^{2} (-4 \text{ rad/s}^{2}) - (\frac{5}{6}ft) (\frac{4}{3}ft) (4 \text{ rad/s})^{2}] \frac{1}{L} \\
= (1.0352/b.ft) \frac{1}{L} - 0.24845 (0.34722 + 17.778) \frac{1}{L} \\
M_{0}^{R} = (1.0352/b.ft) \frac{1}{L} - (4.503/b.ft) \frac{1}{L}$

FORCE-COUPLE AT 0: R=-(5,3016)i; MR = (1.03516.ft)i-(4.5016.ff)k

18.104

r = 8 in.

n = 9 in.

12 in.

n = 9 in.

GIVEN:

DISK OF WEIGHT W = 6 1b

ROTATES WITH CONSTANT $\omega_{i} = (16 \, \text{rad/s}) \frac{1}{6}$.

AT INSTANTS HOWN, SHAFT

DCE HAS $\omega_{i} = (8 \, \text{rad/s}) \frac{1}{6}$.

AND $\omega_{i} = (6 \, \text{rad/s}^{2}) \frac{1}{6}$.

FIND:

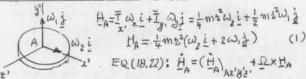
(a) COUPLE APPLIED

TO SHAFT

(b) DYNAMIC REACTIONS

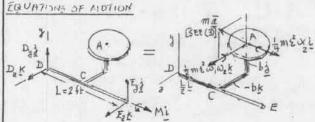
AT D AND E.

ANGULAR MOMENTUM ABOUT A



WHEN - THE FIRST TERM IS OFTHING BY DIFFERENTIATING

 $(\exists_{A})_{Ax'y'z'} = \frac{1}{4} mz^{2} \omega_{z} \dot{L} = \frac{1}{4} mz^{2} \alpha_{z} \dot{L}$ $THUS: \dot{H}_{A} = \frac{1}{4} mz^{2} \alpha_{z} \dot{L} + \omega_{z} \dot{L} \times \frac{1}{4} mz^{2} (\omega_{z} \dot{L} + 2\omega_{z} \dot{L})$ $\dot{H}_{z} = \frac{1}{4} mz^{2} \alpha_{z} \dot{L} + \frac{1}{2} mz^{2} \omega_{z} \dot{L} \qquad (2)$



ΣMD= Σ(MDk++: Lix(E3+E2K)+Mi=(½i+bj-bK)xmb[(α,-ω²)j+(α,+ω²)K+HA LE3K-LE32+Mi=½mbL[(α,-ω²)K-(α,+ω²)j+2mb°α,i+ - me²α,i+½m ε²α,i+½m ε²ω,ωε Κ

EQUATING COUPF. OF UNIT VECTORS:

(4)

M=m(2b+4z) a,

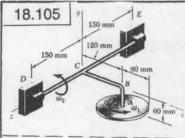
 $D - LE_2 = -\frac{1}{2} mbL(\alpha_2 + \omega_2^2), \quad E_2 = \frac{1}{2} mb(\alpha_2 + \omega_2^2)$ (5)

 $E_{y} = \frac{1}{2} m \left[b L \left(\alpha_{z} - \omega_{z}^{2} \right) + \varepsilon^{2} \alpha_{1} \omega_{z} \right]$ $E_{y} = \frac{1}{2} m \left[b \left(\alpha_{z} - \omega_{z}^{2} \right) + \frac{\delta^{2}}{L} \omega_{1} \omega_{z} \right]$ (6)

 $\sum F_{y} = \sum (F_{y})_{t+1} : D_{y} + E_{y} = m a_{y} \qquad D_{y} = m + (\alpha_{2} - \omega_{2}^{*}) - E_{y}$ $D_{y} = \frac{1}{2} m \left[b(\alpha_{2} - \omega_{2}^{*}) - \frac{t^{2}}{L} \omega_{1} \omega_{2} \right] \qquad (7)$

 $\sum F_z = \sum (F_z)_{eH}: D_z + E_z = md_z \qquad D_z = mb(\alpha_z + \omega_z^2) - E_z$ $D_z = \frac{1}{2}mb(\alpha_z + \omega_z^2)$

WITH GIVEN DATA: $m = 6/82, z = 0.18634, L = 2ft, b = \frac{3}{4}ft, z = \frac{2}{3}ft, \omega_1 = 16, \omega_2 = 8, \alpha_2 = 6$ (a) M = 1.382 16.5t M = (1.382 16.4t) M = (1.483 16.4t) M = (1.483



GIVEN: 2.5-kg DISK ROTATES WITH $\omega_1 = \omega_1 j$, $\alpha_2 = -(15 \text{ mod/s}^2) j$. SHAFT DCE ROTATES WITH SCANSTANT $\omega_2 = (12 \text{ mod/s}) k$, FIND:

DYNAMIC REACTIONS AT D
AND E WHEN W, HAS
DECREASED TO 50 rad/s.

ANGULAR MOMENTUM ABOUT A



HA= Ig. Wg. j+ Iw K = I mr W, j+ im work

EU. (18.22): H= (H)

THE FIRST TERM IS THE RATE OF CHANGE OF HA WITH RESPECT TO THE

ROTATING FRAME AXYZ:

$$\begin{split} (\stackrel{\leftarrow}{H}_{A})_{A \times Y^{12}}, &= \frac{1}{2} m z^{2} \tilde{\omega}_{1} \stackrel{\rightarrow}{j} = \frac{1}{2} m z^{2} \omega_{1} \stackrel{\rightarrow}{j} ; \quad \text{ALSO} : \Omega = \omega_{2} \stackrel{\leftarrow}{k} \\ \text{T HUS} : \stackrel{\leftarrow}{H}_{A} &= \frac{1}{2} m z^{2} \omega_{1} \stackrel{\rightarrow}{j} + \omega_{2} \stackrel{\leftarrow}{k} \times (\frac{1}{2} m z^{2} \omega_{1} \stackrel{\rightarrow}{j} + \frac{1}{2} m z^{2} \omega_{2} \stackrel{\leftarrow}{k}) \\ \stackrel{\leftarrow}{H}_{A} &= \frac{1}{2} m z^{2} \omega_{1} \stackrel{\rightarrow}{j} - \frac{1}{2} m z^{2} \omega_{1} \omega_{2} \stackrel{\leftarrow}{i} = \frac{1}{2} m z^{2} (-\omega_{1} \omega_{2} \stackrel{\rightarrow}{i} + \omega_{1} \stackrel{\rightarrow}{k}) \\ &= \frac{1}{2} (2.5 kg) (0.08 m)^{2} \left[- (50 radk) (12 rad/s) \stackrel{\rightarrow}{i} + (-15 rad/s) \stackrel{\rightarrow}{j} \right] \\ \stackrel{\leftarrow}{H}_{A} &= -(4.8 N \cdot m) \stackrel{\rightarrow}{i} - (0.120 N \cdot m) \stackrel{\rightarrow}{j} \end{split}$$

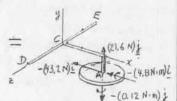
ACCELERATION OF MASS CENTER USING C AS THE FIXED ORIGIN, AND SINCE $\alpha_z = 0$: $\bar{\alpha} = -\frac{v_{a/c}}{v_{a/c}} \omega_z^2 = -\frac{(0.12 \text{ m})^2 - (0.06 \text{ m})^2_d}{(12 \text{ rad/s})^4}$ $\bar{\alpha} = -\frac{(17.28 \text{ m/s}^2)^2}{(12 \text{ rad/s})^4}$

THUS: mā = (2,5 kg)ā mā = - (43,2 N)i + (21.6N)j (2)

Dyj alsm E ELL

Dyj alsm E ELL

Dyj alsm E ELL



 $\Sigma \stackrel{M}{=} \Sigma \stackrel{M}{=} \Sigma \stackrel{M}{=} 0_{eff}$: - $(0.3 \text{ m}) \stackrel{k}{=} \times (E_x \stackrel{i}{=} + E_y \stackrel{i}{=}) = -(4.8 \text{ N·n}) \stackrel{i}{=} -(0.12 \text{ N·m}) \frac{1}{2} + \frac{1}{2} A/D \times m \stackrel{a}{=}$

-0.3 Ex = +0.3 Ey = -4.8 i - 0.12j + +(-0.15k+0.12i-0.06j)x(-43.2i+21.6j)

-0.3Fz = +0.3 Eg = -4.8i -0.12f+6.48j +3.24i +2,542K-2,542K

ERVATING THE COEFF, OF THE UNIT VECTORS:

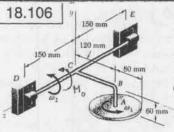
-0.3 = -0.36 $E_2 = -21.2 \text{ N}$

0.3 Ey = - 1.56 Ey = - 5.70 N

 $\sum F_{x} = \sum (F_{x})_{eff}$: $D_{x} = -21.2N = -45.2N$ $D_{x} = -22.0N$ $\sum F_{y} = \sum (F_{y})_{eff}$: $D_{y} = 5.20N = 21.6N$ $D_{y} = 26.8N$

ANSWER D=-(21.0N) i+(26.8 N); E=-(21.2 N) i-(5.20 N) j

(ANSWER GIVEN WITH RESPECT TO ROTATING CXYE AXES)



GIVEN:

2.5-kg DISK ROTATES WITH

CONSTANT \(\omega = (50 md/s) \).

AT INSTANT SHOWN, SHAFT

DCE ROTATES WITH

\(\omega = (12 rad/s) \).

\(\omega = (8 md/s) \).

\(\omega \).

\(\

(1)

(b) DYNAMIC REACTIONS AT D AND E

ANGULAR MUMENTUM TENUT A



$$\begin{split} & \underline{H}_{A} = \underline{I}_{3}, \omega_{3}, \underline{i}_{3} + \underline{I}_{2}, \omega_{2}, \underline{k} = \underline{i}_{3} m \ell^{2} \omega_{1} \underline{j}_{3} + \underline{i}_{3} m \ell \omega_{2} \underline{k} \\ & E O.(19.22); \quad \underline{H}_{A} = (\underline{H}_{A})_{AX'\underline{3}'\underline{2}'} + \underline{D} \times \underline{H}_{A} \\ & \text{THE EIRST TERM IS THE 11 } = 3 = 6 \text{ CHOUSE} \end{split}$$

THE FIRST TERM IS THE 1 . TO SECHETICE OF MA WITH RESPECT TO THE FRANCE AZY'E' WHICH ROTATES AT $\Omega = \omega_z k$.

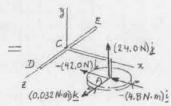
$$\begin{split} &(\stackrel{.}{H}_{A})_{A\times'y'z'} = \frac{1}{4}m\kappa^{2}\mathring{\omega}_{z} \stackrel{k}{\underline{k}} = \frac{1}{4}m\epsilon^{2}\alpha_{z} \stackrel{k}{\underline{k}} \\ &\text{THUS: } \stackrel{.}{\underline{H}}_{A} = \frac{1}{4}m\epsilon^{2}\alpha_{z} \stackrel{k}{\underline{k}} + \omega_{z} \stackrel{k}{\underline{k}} \times (\frac{1}{2}m\epsilon^{2}\omega_{1} \stackrel{i}{\underline{j}} + \frac{1}{4}m\epsilon^{2}\omega_{z} \stackrel{k}{\underline{k}}) \\ &\stackrel{.}{\underline{H}}_{A} = \frac{1}{4}m\epsilon^{2}\alpha_{z} \stackrel{k}{\underline{k}} - \frac{1}{2}m\epsilon^{2}\omega_{1}\omega_{z} \stackrel{i}{\underline{i}} = \frac{1}{4}m\epsilon^{2}(-2\omega_{1}\omega_{z} \stackrel{i}{\underline{i}} + \alpha_{z} \stackrel{k}{\underline{k}}) \\ &= \frac{1}{4}(2.5kg)(0.08m)^{2}[-2(50 \text{ rod/s})(12 \text{ rad/s}) \stackrel{i}{\underline{i}} + (8 \text{ rad/s}) \stackrel{k}{\underline{k}}] \end{split}$$

HA = -(4.8 N·m) + (0.032 N·m) K ACCELERATION OF MHSS CENTER USING C AS THE FIXED ORIGIN:

 $\bar{a} = \frac{\alpha_{X}}{2} \times \frac{\alpha_{A/C}}{4} - \frac{\alpha_{A/C}}{4} \omega_{X}^{2} = \frac{(8 \operatorname{rad}/s)^{4}}{4} \times \left[(0.12 \, \text{m})^{\frac{1}{2}} - (0.06 \, \text{m})^{\frac{1}{2}} \right] - \left[(0.12 \, \text{m})^{\frac{1}{2}} - (0.06 \, \text{m})^{\frac{1}{2}} \right] \left(12 \, \operatorname{rad}/s \right)^{4}$ $= (0.96 \, \text{m/s}^{2})^{\frac{1}{2}} + (0.98 \, \text{m/s}^{2})^{\frac{1}{2}} + (1.98 \, \text{m/s}^{2})^{\frac$

 $= (0.96 \text{ m/s}^*)_{\dot{a}}^{\dot{a}} + (0.48 \text{ m/s}^*)_{\dot{b}}^{\dot{a}} - (17.28 \text{ m/s}^*)_{\dot{b}}^{\dot{a}} + (8.64 \text{ m/s}^*)_{\dot{a}}^{\dot{a}}$ $\bar{a} = -(16.8 \text{ m/s}^*)_{\dot{b}}^{\dot{a}} + (9.6 \text{ m/s}^*)_{\dot{a}}^{\dot{a}}$

THUS; ma=(2.5 kg) & ma=-(42,0N)i+(24.0N)j (2)



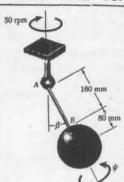
$$\begin{split} & \sum \underline{M}_{D} = \sum (\underline{M}_{D})_{eff}: \\ & - (0.3 \text{ m}) \underbrace{k \times (E_{\lambda} \underbrace{i} + E_{\lambda} \underbrace{i}) + M_{0} \underbrace{k}}_{c} = - (4.8 \text{ N·m}) \underbrace{i} + (0.032 \text{ N·m}) \underbrace{k} + \underbrace{2}_{AD} \times m\underline{a} \\ & - 0.3 \underbrace{E_{\lambda} \underbrace{i}}_{c} + 0.9 \underbrace{E_{\lambda} \underbrace{i}}_{c} + M_{0} \underbrace{k}_{c} - 4.8 \underbrace{i}_{c} + 0.032 \underbrace{k}_{c} + \\ & + (-0.15 \underbrace{k}_{c} + 0.12 \underbrace{i}_{c} - 0.06 \underbrace{j}_{c}) \times (-42 \underbrace{i}_{c} + 2.4 \underbrace{j}_{c}) \end{split}$$

(a) (b) Mo = 0.032+2.88-2.52 = 0.392 N·m

Mo=(0,392 N.m)k

(b) $Q - 0.3 E_{\chi} = 6.30$ $E_{\chi} = -21.0 N$ Q) $0.3 E_{y} = -4.8 + 3.6 = -1.20$ $E_{y} = -4.00 N$ $\sum_{\lambda} = \sum_{\lambda} (F_{\lambda})_{eff}$: $D_{\chi} - 21.0 N = -42.0 N$ $D_{\lambda} = -21.0 N$ $\sum_{\lambda} F_{y} = Z(F_{y})_{eff}$: $D_{y} - 4.00 N = 24.0 N$ $D_{\chi} = 25.0 N$

D= -(21.0 N) i + (28.0 N) j; E= -(21.0 N) i - (4.00 N) j (ANSWER GIVEN WITH RESPECT TO ROTATING CZYZ AYES) 18.107 and 18.108 GIVEN:



SOLID SPHERE WELDED TO END OF ROD AB OF NEGLIGIBLE MASS SUPPORTED BY BALL AND SUCKET AT A. SPHERE PRECESSES AT CONSTANT RATE OF 50 FPM AS SHOWN.

PROBLEM 18.107: PIND RATE OF SPINY, KNOWING THAT B = 25°. PROBLEM 18, 108:

FIND B, KNOWING THAT RATE OF SPIN 15 4 = 800 PM.

18.109 and 18.110

CONE SUPPORTED BY BALL AND SOCKET AT A.



PROBLEM 18.109;

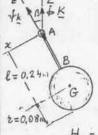
GIVEN!

PRECESSES AS SHOWN AT CONSTANT RATE OF HOPPM WITH B= 40° FIND!

RATE OF SPINY PROBLEM 18,110

FIVEN: = 3000 rpm, /5 = 60

TWO POSSIBLE VALUES OF 4



ANGULAR VELOCITIES: SPHERE: W= &K+Vk ω= - \$ sinβ i + (V+\$ wsB) k FRAME AXY 2: IL = &K IL = - PSINB i + + WSB k ANGULAR MOMENTUM ABOUT A

H = I W 1+ I W, K H = - m (32+ 1") +sinsi+ = m2 (*14005) k

SINCE A 15 FIXED, WE USE = Q. (18.28): ZMA=(HA)Ary + DXHA = 0 + (- + sin Bi + + ws B) k X HA = (- +sinpi++ uspk) xm[-(25+6)+sinpi+226(4++cospk) = m + sinp = (+++cosp)-(= ++++)+cosp]

BUT IMA = - lf x (-mgK) = - mglsin bj

EQUATING (1) HND (2): n b sin p [2 2 (+ + + cos p) - (= 2 + e) + cos p = - mg (sin p = 2 (++ + cos)= (= 2 + 2) + cos > - 36

GIVEN DATA: (NOTE THAT & IS MEGATIVE) 2=0.08 m, C=0.24 m, g= 9.81 m/s, +=-50 rpm = -5.236 rad/s 256 ×10 (4- 5,236 cos B) = 60.16 ×10 (-5,236 cos B) + 449.7×10

(4)

4=- 117,01 cosp + 175.65

PROBLEM 18,117 WIN PERT, EN (4) TIELDS Y=- 117,81 cus 25"+ 175.65 = + 68,875 rad/s = . 657.7 rpm V= 658 rpm

PROBLEM 16.108

WITH 4- 800 cpm = 83,776 cm/s, EQ. (4) READS 83,776 = - 117.81 cosp + 175,65 cosp = 0,77985 B = 38.753"

B= 38.8°

ANGULAR VELOCITIES CONE: W= OK + FK w=- +sing + (V+ +cosp)k FRAME AZYZ: Q =- 4K D= -+ singi+ + wsph ANGULAR MOMENTUM ABOUT A H = I W i + I W K

H = - 3 m (4+h) + sin pi+ 3 m2 (++ + cos s) k SINCE A IS FIVED, LE USE ER. (18.28): ZM = (H) + 1 × H = 0 + (- + sin Bi + + cos pk) x H = (-+ sin pi++cospk) x -= m(=+h)+sinpi+=m=(+++osp)k) = 3 m + sin s[1 t (V++ cos/s)-(+++)+cos/s]i BUT EM = - fik x (-mg K) = - 7 mg h sin pj (2)

EQUATING (1) AND (2): 3 m + sin | [+ + + (v+ + cosp) - (+ + h) + cos | = - + mghsin | 12 (++ + cosp) - (+++) +cosp = - 5 gh 12 4- (h- 2) + cos p = - 5 8h

WITH 4= 4 St, h= 3 ft, g = 32,2 ft/s, AND MULTIPLYING BY 32, 4-17.5 PCOSB = - 966/2 (3)

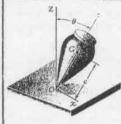
PROBLEM 18,109 LETTING \$= - 40 rpm = -4,1888 rad/s, \$= 40° IN (3), V-17.5(-4.1888)cos 40°=- 966/(-4.1888) V=-56.154+230.616=174.46 rad/s = 1666,0 rpm V=1666 rpm

PRUBLEM 18,110

LETTING 1 = 3000 rpm = 314.16 rad/s, B = 60 in (3), 314,16-17,5 + cas 60° = - 966/+ 8.75 - 314.16 - 966 = 0 \$ - 35,904 p - 110,4 = 0 \$= \frac{1}{35,904 \pm \langle (35,904)^2 + 4(110.4)} = \frac{1}{35,904 \pm 41,602} \$ = +38.753 rad/s = + 370 rpm = (370 rpm) K (SENSE OPPOSITE TO SENSE SHOWN)

OR += -2.849 rad/s = -27.2 rpm φ = -(27.2 rpm) K (SAME SENSE AS SHOWN) 18.111 and 18.112

TOP SUPPORTED AT FIXED POINT O.



PROBLEM 18.111:

GIVEN: m = 853, K = 21mm, K = 45mm C = 37,5 mm, 8 = 30°, RATE OF SPIN ABOUT & AXIS = = 1800 rpm.

TWO POSSIBLE RATES & OF STEADY PRECESSION.

PROBLEM 18, 112 GIVEN: I = I , I = I , W = RECTANGULAR COMPONENT CFW ALONG & AXIS

(a) SHOW THAT (Iω, - I' + cos θ) + = WC (b) SHOW THAT I + + WC IF + >> +

(c) FIND PERCENT EFFOR WHEN EXPRESSION UNDER b IS USED TO APPROXIMATE THE SLOWER & OF PROB. IB. III.

WE RECALL FROM PAGE 1150 THE FOLLOWING EUS. W = - \$ 5111 + i + W. K Ha = - I'sindi + Iwak (19.41) == + sinbi + + cos ok (18.42)

SINCE O IS A FIXED POINT, WE USE EW (18.28): $\Sigma \overrightarrow{H}_0 = (\overrightarrow{H}_0)_{0 \times 17} + \overrightarrow{U} \times \overrightarrow{H}_0 = 0 + \overrightarrow{U} \times \overrightarrow{H}_0$

= (- \$sindi + \$cost k) x (- I \$sindi + IW, E) = (IW psin 8 - I + costsill) j

= (Iw - 1 + 1010) + sint ; WHERE & 15 I PLANE OZE AND POINTS AWAY

(2) BUT IM = ckx(-WK) = We sindj EQUATING CUS. (1) AND (2):

(3) (IW, - I + COSA) + = WG

PROELEH 18,111

SINCE I=mk2, I'=mk2, W=mg, EO (3) YIELDS

(k2 W2 - k2 + cos a) + = go WHERE W = Y + \$ cos 0

WITH GIVEN DATH AND V = 1800 pm = 60 Trads: [(0.021)*(607+ + 0030)-(0.045)*+0030) += 9.81(0.0375)

 $[(0.045)^{4}, (0.021)^{8}]\cos 30^{8} + (0.021)^{8}(0.074) + 9.81(0.0375) = 0$ \$ - 60.597 \$ + 268.17 = 0

SOLVING: \$ = 30.249 ± 25.492

\$= 55,79 radis AND \$= 4,807 radis

ANSWEE: 533 Fpm AND 45.9 Fpm

PROBLEM 18,112

(a) SEE DERIVATION OF EN. (3) ABOVE

(b) FON \$>> \$, W2 \$ \$, AND EO. (3) REDUCES TO (I + - I' + ax +) + = WC

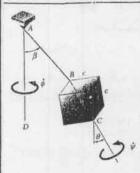
AND, WITH Y >> 4, TO (Q.E.D.) TY+ = WC

(c) WITH DATA OF PROB. 18.11, ABOVE EQUATION VITLED SOLVING (5) FOR C \$\div \frac{1}{2}\$, \frac{1}{2}\$ \$\frac{1}{2}\$ \$\frac{1}{2 $\phi = \frac{Wc}{IV} = \frac{Mgc}{Mk_L^2 V} = \frac{9.81(0.0375)}{(0.021)^2 (6077 \text{ rad/s})} = 4.4555 \text{ rad/s}$ = 42.26 rpm

% ERRUK = 100 42.26-45.90 = -7.9 %

18.113 and 18.114

SOLIL C' RE ATTACHED TO CORD AB



PROBLEM 18.113:

c = Bomm, B = 30° $\psi = 40 \text{ rad/s}, \ \phi = 5 \text{ rad/s}.$ FIND: 8

PROBLEM 18.1141

c = 120 mm, AB = 240 mm 8 = 25°, B = 40° FIND:

(a) v, (b) \$

WE RECALL FROM SEC. 9,17 THAT, SINCE THE 3 PRINCIPL MOMENTS OF INFITTA OF A CUES ARE EQUAL, IT'S MOMENT OF MEETIN ABOUT ANT LINE THEODIGH G 15 ALSO I - - DIC USIN'S GROZ AXES WIT- Z ALONG CB, X IN ABD PLANE AND & L ABD AND FOINTING ANDAY, WE HAVE

CUBE: W = + sindi+ (+++ ws 8) } H = 1 mc (\$ sin 0 i + (V+ \$ cos 0)k) FRAME GZYZ:

1 = \$sinbi+ + cost K

EQ.(18.22):

He = (He)exAs + Tx x Ae H = O+(&sinti+&costk) X Emc [dsinbi+(+++ costs)]

H = +mc + sino [-(+++ coso) +++ coso] =- Imc & Tsint j

EQUATIONS OF MOTION 13 csing mg

EF = E(F) + 1 HORIZ. COMP. : Tsing = ma VERTICAL COMP. Tcos & -mg = 0 TCOS B = mg (3) (4)

a=qtanp DIVIDE (2) BY (3): tan B = a +9 ZMB= [(MB)ess: -- mg 13c sin 0 = 6 mc + Vsin 0 - (mg tan 1) 12 c cos 0 DNIDE BY mg 13 c cos 0 AND SOLVE FOR tan 0:

tan 0 = tox B 1 + (c \$ 1/3 1/3 g)

PROBLEM 18,113 LETTIN & B=30°, C= 0.08m, += 5rad/s, 4=40 rad/s, 9=9.81m/s IN (5): tan 8 = 0,43942 $\theta = 23.7^{\circ}$

PROBLEM 18, 114 $\overline{a} = \overline{z} + \overline{z}$ RECALLING (4): $\overline{4} = \overline{a} = \frac{g \tan \beta}{\overline{z}}$ LETTING 13=40", 0=25", AB=0.24 m, C=0.12 m, g=9.81 m/32;

P = 6.4447 rad/s (b) \$ = 6.44 rad/s

 $\psi = \frac{40.752}{(0.12)(6.4447)} = 52.694 \text{ rad/s}$ (a) V= 52,7 rad/s €

4

SOLID SPHERE ATTACHED 18.115 and 18.116 TO CORD AB. PROBLEM 18, 115; GIVEN: c = 3 In., B=40, \$= 6 rad/s ANGLE B, KNOWING THAT (a) \(=0, (b) \(=50 \) rad/s, (c) \(=-50 \) rod/s. PROBLEM 18.116: GIVEN: C=3 in., AB = 15 in., 0=20, B= 35° (a) V, (b) \$ USING GXYZ AXES WITH & ALONG CB, X IN ABD PLANE AND & LABD AND POINTING AWAY: SPHERE: W= + sin Di+(+++ coso) k H===mc [+sin+(+++cos+)6] FRAME Gzyz: Q= &sinti + &costk EQ.(18,22); HG = (HG)GZYZ + QXHG (AB) sin B C sin O = O+ (&sinti++cost) xH H= (+sindi++cost k) x = mc2[+sindi+(+++cost)k] H===mc+oin+[-(+++ws+)++cosoj] Ha=-= mc+ Vsino } (1) EQUATIONS OF MOTION 2mc中Ysinの ΣF = Z(F)ett: HORIZ, COMP.: TsinB=ma (2) VERTICAL COMP Tcos & - W = 0 C shirt -T cos & = mg (3) DIVIDE (2) BY (3): tan B = a a=qtans (4) +9 EMB = ZIMB)eH: -mgcsin+= = mc+ Vsino - (mgtan) ccost DIVIDE BY MIG COST AND SOLVE FOR Tand: $\tan \theta = \frac{1}{1 + (2c \phi \dot{v}/5g)}$ (5) PROBLEM 18,115 LETTING (3=40°, c= 4ft, \$= 6 rad/s, \$= 32.2 ft/s2 IN (3)) tan 0 = tan 40 / (1 + 0.018 634 V) (a) POR = 0; 0 = 40,0° tan 0 = tan 40° (b) FOR \(\frac{1}{2} = 50 \text{ ral/s}: \text{ tan 0} = 0.43438 \\
(c) \(\text{FOR} \quad \text{V} = -50 \text{ ral/s}: \text{ tan 0} = 12.285 \\ 0 = 23.5 B = 85.3° PROBLEM 18, 116 $\bar{a} = \bar{z} \dot{\phi}^2$ RECALLING(4): $\dot{\phi} = \frac{\bar{a}}{\bar{z}} = \frac{g \tan \beta}{(AB) \sin \beta}$



(PRECESSI .: OF THE EQUINDXES) GIVEN:

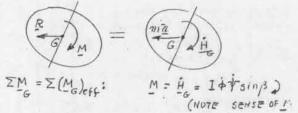
RATE OF PRESENCE OF ENETH ABOUT GA = I rev IN 25 800 yr FOR EARTH! Pave = 5.51 2 ave = 6370 km, I = = m tave

FIND: AVERAGE VALUE OF COURTE M DUE TO GRAVITH TIONAL ATTRACTION OF SUN, MOON, AND PLANETS.

WE USE GIYZ AXES (WITH A POINTING AWAY) TUTAL ANG. VEL. OF EARTH w = - 4 singi + (+++ cosp) 5 FRAME GXY 4: 12 = - + sings + + cosp & Ho=- I psinpi+ I (++ cxp) b

EQ.(18,25): $\frac{\dot{H}}{G} = (\frac{\dot{H}}{G})_{GAYI} + \frac{\Omega}{G} \times \frac{\dot{H}}{G} = 0 + (-\dot{\phi}\sin\beta i + \dot{\phi}\alpha\alpha\beta k) \times \dot{H}$ H= (-4 sin Bi+ + cosp &) x [- I + sin Bi+ I (4+ + cos A) = I + sin B (++ + + x 05 B - + x 05 B) j H = I & Ysin Bi

EQUATIONS OF MOTION



WITH GIVEN DATA! m= 4720 = 471 (6.37 × 10 m) (5.51 × 10 kg/m3) = 5,9657×1024 kg = (5.9657×102+ kg)(6.37×10 in)= = 96.827 × 1036 kg.mi (25 BUOYT) (365,24 day/yr) (24 h/day) (3600 s/h) V = (23,93 h)(3600 s/h) = 72,935 × 10 rad/s, /3 = 23.45°

M= Hs= I + TsinB = (96.827×1036kg.m)(7.717×101251)(72.935×1051) sin 23,45 = 21.69 x 1021 N.m

M=21.7 x 10 N.m

(AB) sin B + c sin B

(b) \$= 5,30 rad/s

(a) \v = 56,1 rad/s

WITH B = 35", 0 = 20", AB=1.25ft, c = 0.25ft, g = 32.2 ft/s":

C + V = 2.5 g (tan B -1) = 2.5 (32.2) (tan 35° -1) = 74.366

\$ = 5,3006 rad/s

SOLVING (5) FOR CAT,

V= 74.366 = 56,12 and/s

18.118



GIVEN:

PROJECTILE WITH M = 20kg R = 50 mm, R = 200 mm \$ = 600 m/s (HORIZONTAL) DRAG = D = 120 N (HORIZONTA) 13 = 3°, c = 150 mm V=6000 rpm

(a) APPROXIMATE "ALUE OF KATE OF PRECESSION , (b) EXAL I VALUES OF TWO FOR BUE KILLS OF FRECESCION

SINCE THE DRAG D IS A PORCE CONSTANT IN MAGNITUDE AND DIRECTION (LIKE THE WEIGHT OF A TOIT), IT WILL PRECESS, LIKE A TOP, SHOT AN AXIS GZ PARPELET TO THAT FOR LE.



USING THE AXES -XH3, WITH A PRINTERS ANDRY

w= +sinpi+(4++cosp)k He= I & sin pi+I (+++cosp)k

FRANF GXX 2: 12 = \$ sin \$ i + \$ wsp &

EQ (18.22): HG = (HG) 3442 + 12 XHG = 0 + 12 XHG H = (+ sinpi + + cosps) x [= + sinpi + = (V+ + cosp) k] = \$\dots inp [- \bar{I}_{\pi} (\bar{Y} + \dots \beta) + \bar{I}_{\phi} \dots \beta) \bar{j}

THUS! ZM = H = +sinp (I2 - I2) + cosp - I2 + J & ON THE OTHER HAND.

ZM=ckx(-DK) = - cDsings

 $ZM_G = Z(M_G)_{eff}: -cD = \phi [(\bar{I} - \bar{I}) + \cos \beta - \bar{I} - \bar{V}]$ (3)

(a) APPROXIMATE VALUE OF &

CINICE V>> b, WE MAY NEGLECT THE FIRST TERM IN THE EXPONET IN (3). WE DRITAIN

I. + V = c D

WITH GIVEN DATA: T = m K = (20kg)(0.05m)=0.05kgm C=0.15m, D=120 N, Y= 6000 rpm = 200 TT rad/s:

0.05 \$ (200 Ti) = (0.15)(120) \$ = 0.5730 rad/s \$= 5.47 rpm

(b) EXACT VALVES OF &

USING EG. (3) WITH THE ABOVE DATA AND WITH 13 = 3° AND I = m k = (20 K) (0,2 m) = 0.8 kg·m*; -(0.15m)(120N) = \$ [(0.8-0.05) \$ cos 3-0.05(200 R)

 $0.74897 \dot{\phi}^2 - 31.416 \dot{\phi} + 18 = 0$ \$ - 41.945 \$ + 24.033 = 0 \$ = \frac{1}{2} (41.945 \tau \((41.945)^2 - 4(24.033)) = 1 (41.945 ± 40.783) rad/s

φ = 41.364 rad/s 1. φ = 0.58101 rad/s \$= 395 rpm AND \$ = 5,55 rpm

GIVEN: 18.119

AXISYMMETRICAL BODY UNDER NO FORCE I - MOMENT OF INERTIA ABOUT AXIS OF SYMMETRY - TRANSVERSE AXIS THRU G. HE = ANG. MOM. ABOUT G.

SHOW THAT:

AND W = He cos & (I'-I)

FROM EQ. (18,40), PAGE 1146:

(1) ω, = - + sin θ

FROM THE PIRST OF ERS. (18.47), PAGE 1147!

Wz = - He sint (2)

EQUATING THE R.H. MEMBERS OF (1) AND (2):

 $-\phi \sin \theta = -\frac{H_6 \sin \theta}{T^2} \qquad \dot{\phi} = \frac{H_6}{T^2}$ (Q.E.D.)(3)

FROM FIG. 18.21: V=W, - \$ 205A (4)

FROM ERS. (18.48): No = - He cost

FROM ER. (3) AMOVE! \$cos 8 = He COS H (6)

SI BSTITUTE FROM (5) AND (6) INTO (4):

1 = Hg cas 0 (- - 1) ψ = H6 COSO (1'-1) (Q.E.D)

18.120

AXISTMMETRICAL BUDY UNDER NO FORCE I = MOMENT OF INCRTIA ABOUT AXIS DESYMMETRY I'= - --TRANSVERSE AXIS THRU G B = ANGIF BETWIEN AXES OF PRECESSION & SPIN W, = COMPONENT OF W ALONG AXIS OF SYMMETRY

SHOW THAT:

 $\dot{\phi} = \frac{I \, \omega_e}{T^* \cos \theta}$ (a)

(b) EQ. (18,44) IS SATISFIED

(a) SEE SOLUTION OF PRUB. 18,119 FOR DEPINITION OF EW. (3): (3)

FROM POS. (18.48): HG = I We

SUBSTITUTE FOR H_G IN (3): $\dot{\phi} = \frac{I \, \Omega_s}{I' \cos \theta}$

(Q. E.D.)

(b) FROM RELATION JUST DETAINED, NE HAVE $I\omega_2 - I' \phi \cos \theta = 0$

WHICH SHOWS THAT, FOR AN AXISYMMETRICAL BODY UNDER NO FORCE, THE R. H. MEMBER

ZM = (IW2 - I' + cost) + sint (18,44) IS EQUAL TO ZERO. BUT, SINCE THERE

IS NO FORCE, WE ALSO HAVE ZM = 0 AND EQ. (18.44) IS SATISFIED. (Q.E.D.)

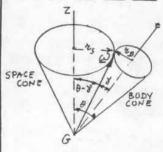
18.121

GIVEN:

AXISYMMETRICAL BODY UNDER NO FORCE I = MUMENT OF INERTIA ABOUT AXIS OF SYMMETRY. - TRANSVERSE AXIS THRU G W, = COMPONENT OF W ALONG AXIS OF SYMMETRY. SHOW THAT :

ANGULAR VELOCITY W IS OBSERVED FROM THE BODY TO ROTATE ABOUT THE AXIS OF STRIPLETRY AT THE RATE

$$m = \frac{I' - I}{I'} \omega_z$$



ASSUMITIS DIRECT PRECESSION (I'>I) WE CONSIDER THE SPACE AND BODY CONES, THE PLANE ZG = ROTHTES ABOUT THE ZAXE AT THE FATE &; SO WILL THE VECTOR W CONTAINED IN THAT FLANE THUS, THE TIP OF W

WILL DESCRIBE AN ARC OF CIRCLE OF LENGTH & & AL IN THE TIME At. BUT, ACCORDING TO THE DEFINITION OF N, THE VECTOR W IS OBSERVED TO ROTATE AT THE RATE IN WITH RESPECT TO THE BODY. THUS THE TIP OF O WILL DESCRIBE AN ARE OF CIRCLE OF LENGTH & n At IN THE TIME At. SINCE THE BODY CONE ROLLS ON THE SPACE CONE, WE HAVE $r_s \neq \Delta t = t_B n \Delta t$

BUT, FROM THE SKETCH ABOVE,

2= W Sin (8-8) AND 2= W Sin 8

SUBSTITUTING INTO (1):

$$\frac{1}{4} \sin(\theta - \delta) = n \sin \delta
n = \frac{1}{4} \frac{\sin(\theta - \delta)}{\sin \delta}$$
(2)

WE RECALL THE RELATION DERIVED IN PAGE 18.12D: WE WRITE THE TRIGONOMETRIC IDENTITY

$$\dot{\phi} = \frac{\Gamma \omega_a}{\Gamma' \cos \theta}$$

$$n = \frac{I \omega_2}{I' \cos \theta} \frac{\sin \theta \cos x - \sin x \cos \theta}{\sin x}$$
$$= \frac{I \omega_1}{I'} \left(\frac{\tan \theta}{\tan x} - I \right)$$

RECALLING FROM EQ. (18,49) THAT $\frac{\tan \theta}{\tan \theta} = \frac{I'}{I}$, WE HAVE

$$n = \frac{I}{I'} \left(\frac{I'}{I} - 1 \right) \omega_2$$

$$n = \frac{I' - I}{I'} \omega_{\epsilon} \left(Q, E, D_1 \right)$$

NOTE. FOR I > I' (RETROGRADE PRECESSION), WE WOULD FIND $m = \frac{I - I'}{T'} \omega_z$

18.122

GIVEN:

AXISYMMETRICAL BODY UNDER NO FORCE AND IN RETROGRADE PRECESSION (I>I'), SHOW THAT :

(a) RATE OF RETROGRADE PRECESSION CANNOT BE LESS THAN, TWICE THE RATE OF SPIN: 14 > 2 T), (b) THE AXIS OF SYMMETRY IN FIG. 18,24 CHN NEVER LIE WITHIN THE SPACE CONE .

(a) WE RECALL THE RELATION DERIVED IN PROB. 18,120:
$$\hat{\phi} = \frac{I N_2}{7^2 \cos \theta}$$
 (1)

I' p coso = I We

SUBSTITUTING W=+++ COSO, WE HAVE 1' \$ cos0 = I (+ \$ cos0)

SOLVING POR +, $\oint = -\frac{I}{I - I}, \frac{\psi}{\cos \theta}$ OR = - 300 0 (2)

FOR RETRUGENDE PRETESSIONS I'/I < 1 ON THE OTHER HAND, THE SMALLEST POSSIBLE VALUE OF I/I IS 1/2 (WHICH CORRESPONDS TO THE CASE OF A FLAT DISK OR ANNULUS).

$$\frac{1}{2} \le \frac{\underline{\underline{I}}}{\underline{\underline{I}}} < 1 \quad oR \quad \frac{1}{2} \ge 1 - \frac{\underline{\underline{I}}}{\underline{\underline{I}}} > 0$$

$$oR \quad \frac{1}{1 - (\underline{\underline{I}}/\underline{\underline{I}}^2)} \ge 2$$

RECALLING THAT SEC & 21, WE MUST HAVE FROM (2) 10122141 (Q.E.D.)

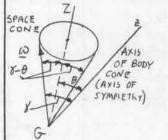
(b) WE RECALL EQ. (10.49):

$$\tan \delta = \frac{I}{I}, \tan \theta$$
.
SINCE $\frac{I}{I} \ge \frac{1}{2}$ AS SHOWN ABOVE, $\frac{I}{I}, \le 2$ AND
 $\tan \delta \le 2 \tan \theta$ (3)

 $\tan (8-\theta) = \frac{\tan 8 - \tan \theta}{1 + \tan 8 \tan \theta}$

SUBSTITUTING INTO (2) AND EXPANDING Sin (8-8): SINCE & < T AND B < T, WE HAVE I+ tand tant >1

THUS
$$tan(8-\theta) \leq \frac{tan \theta}{1}$$



tan (8-+) = tan ti 8-8 5 A

THE Z AXIS CANNOT LIE WITHIN THE

S PACE CONE (SEE SKETCH) (RE.D.)

(FREE PRECESSION OF THE EARTH) 18.123 GIVEN:

I = MOM. OF INTERTIA OF EARTH ABOUT AXIS OF SYMMETRY. - - - - TRANSVERSE 1XIS

I'-0,9967I

RELATION DE ZIVED IN PROB. 18.121:

(FOR I > I')

WHERE WZ = COMPONENT OF WO OF EARTH ALONG AXIS DE SYMMETRY , AND M = RATE AT WHICH WIS OBSERVED FROM THE EARTH TO LOTHTE ABOUT ITS AXIS OF SYMMETRY.

FIND:

PERIOD OF PRECESSION OF NORTH POLE.

$$\begin{array}{l} PERIOD OF PRECESSION \\ = \frac{2\pi}{n} = \frac{1'}{I-I'}, \frac{2\pi}{\omega_2} = \frac{I'}{I-I'}, (1 \, day) \end{array}$$

 $\frac{I'}{I-I'} = \frac{0.9967 \ I}{0.0033 \ E} = 302$

THUS: PERIOD OF PRECESSION = 302 days





GNEN:

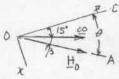
FOOTBALL KICKED WITH HORIZONTAL ANG. VEL. 10 OF HAGNITUDE 200 rpm. RATIO OF AXIAL AND

TRANSVERSE MOMENTS OF INERTIA 15 I/I'=1/3.

(A) ANGLE B BETWEEN WAND PRECESSION AXIS OA.

(B) RATES OF PRECESSION AND SPIN.

(a) USING REFERENCE FRAME OXYZ WITH I FOUNTING AWAY



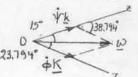
Wx = W sin 150 = W ros 15" H,= I'w, = I'w sin 15' Hy = I'Wy = 0 H2 = I W2 = I W COS 15

Hz = I'w sin15" = 1 tan15= 3 tan15

0 = 38,794° tant = 0,80385

B= 0-15° = 38.794°-15°=23.794° B=23.8"

(b) USING THE OBLIQUE COMPONENTS OF W ALONG OA AND OC:



LAW OF SINES: w Sin 15° Sin 23,794"

SETTING W = 200 rpm. WE FIND

RATE OF PRECESSION = += 02.6 rpm RATE OF SPIN = 7 = 128,8 cpm 18.125

GIVEN:

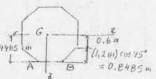
2500 - kg E+ Tall ITE , 2,4-m HIGH WHA PERSONAL BASE k = K = 0,00 m, ka = 0,98m. SHITELLINE FFINITION ENTE OF BEITER HE GY WHEN 20-14 THE METER. ATA MILL ENGERSON SE FO. 25, EXPELLING FIFE. IN PRITIVE I TO THE MAL

FIND: (a) PRECESSION AXIS (b) \$, (c) \$.

INITIAL ANG, VELOCITY! $\underline{\omega}_0 = \left(36 \frac{\text{rev}}{h}\right) \left(\frac{8\pi \text{ md}}{1 \text{ rev}}\right) \left(\frac{1 \text{ h}}{3600 \text{ s}}\right) \underline{j} = \left(0.062832 \text{ mH/s}\right) \underline{j}$ INITIAL ANG. MINERTON

(H) = mky w = (2500 kg)(0,98m)(0,062832 rad/s) }

= (150.86 kg·n1/5) } ANG. IMPUL 5: Madt = " (1.4485m) KX X2(-20N)1(25) M Dt = (115,88 kg.m/s) i



PRINCIPLE OF IMPULSE AND MOMENTUIT

FINAL MOMENTUM: H = (Hc) + M at = (150.86 kg - m/s) & + (115.88 kg - m/) i

H = (115,88 kg·ni/s) i+(150,86 kg ni/s) j (1)

WE RECALL THAT

H=I wi + I wij + I wi K

H = (2500 kg)(0,90m) w i + (2500 kg)(0,98m) wy + IW k

EQUATING THE WEFF OF 1 j t IN (1) HND (2);

2025 W, = 115.88 Wa = 57.225 × 10 rad/s 2401 Wy = 150.86 · Wy = 62,832 × 15 2 rad/s (3) I, W2 = 0

SPINI F PRECESSION AXIS

AXIS HED (a) FROM Eq. (1): tan 0 = Hx = 115.88 0=37,529 THUS: 0= 52,5", 0= 37,5", 0= 90°

FRUA E41(3): tay 8 = 11/2 = 57.225 8=42326 62.832

W = 1/W2 + W1 = 84.986 × 15 TAN/S PRECESSION LAW OF SINES! 5in(5-0)

84.986×10 sin37,529 Sin 42,396° Sin 4,797

SOLVING FOR & AND Y;

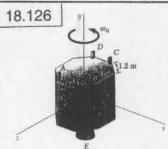
(b) \$= 93.941 × 10 3 rad/s (c) V= 11.667 × 10 rad/s

\$= 53,8 rev/h V= 6.68 rcv/h

WE CHECK FROM DIAGRAM THAT PRECESSION IS RETROGRADE. (IT HAD TO BE, SINCE by > k AND, THUS, I > I')

SPIN AXIS

8-0



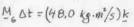
GIVEN: 2500-19 511TELLITE, 2,4-m HIGH WITH OCTUGONAL BASE. K = K = 0. Tom, Ku = 0.98m. SATELLITE SPINNING AT RATE OF 36 rev/h ABOUT GY WHEN ZU-N THRUSTERS AT A AND D ARE ACTIVITED FOR 2 5, EXPELLING FUEL IN POSITIVE & DIRECTION

FIND: (a) PRECESSION AXIS, (b) \$ (c) \$.

INITIAL ANG. VELUCITY: 00 = (36 FeV) (2 Frad) (3600 s) j = (0.062032 rad/s) j INITIAL ANG. HUMENTUM:

(HG) = m ky w = (2500 kg)(0.98m) (0.062832 rank) } = (150.86 kg·n1/5)j

ANG. IMPULSE: Ma St=- (0.6m) i x2 (-20N) j (25)





PRINCIPLE OF IMPULSE AND MOMENTUM

FINAL MOMERTUM:

HG=(HG)+MG Dt = (150.86 kg·m²/s) j+(48.0 kg·n1/5)k(1) WE RECALL THAT

Ha= I20, 1+ I, w, 1+ I, w, 1

H = I, W, 1 + (2500 kg)(0.98m) W, j+(2500 kg)(0.90m) W, E

EQUATING THE COEFF. OF 1, 1, 1 IN(1) AND (2):

I, W, = 0

Wy = 62,832 x 10 rad/c W2 = 23.704 X10 radk

2401 Wy = 150.86 20250, = 48.0



(a) PROM EN. (1): tan 0 = Hy $\frac{H_{e}}{H_{y}} = \frac{48.0}{150.86}$ B=17.650 THUS: \$ = 90", 0, = 17.65", \$ = 72.35"

FROM EQS. (3): tan 8 = 8 = 20.669° 62,832 = 67.155 × 10 rad/s

PREC. . SPIN 14XIS

LAW OF SINES!

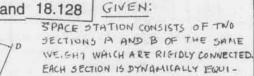
67.155×10 5in 20.669° sin 3.019° SCLVING FIR & AND V:

\$ = 78,177 × 10 rad/s ¥ = 11.665 × 10 rad/s

\$ = 44.8 rev/h V= 6.68 rev/h

WE CHECK FROM DINGRAM THAT PRECESSION IS RETROFRADE (IT HAD TO BE, SINCE Ky>k, AND, THUS, I>I',)

18.127 and 18.128



VALENT TO A HUMO GENEOUS CYLINDER STATION IS PRECESSING ABOUT &D AT THE CONSTANT RATE OF 2 rev/h. PROBLEM 18. 127:

FIND THE RATE OF SPIN OF THE STATION ABOUT CC'.

PROBLEM 18, 128; IF CONNECTION IS SEVERED BETWEEN A AND B, FIND FOR SECTION A:

(a) THE ANGLE BETWEEN CC' AND THE PRECESSION AXIS, (b) \$, (c) Y.

FOR ENTIRE STATION: I'= 1/2 11 (3a2+ L1) I = f ma EQ. (18,49): $\tan \delta = \frac{I}{I}, \tan \theta = \frac{6(9)^4}{3(9)^4}$ 3(9) + (90) tan 40" = 58.252 × 10 stan 40°, 8=2.7984°

PROBLEM 18.127

LAW OF SINES: Jin B 5in8 Sin(0-8) WITH \$ = 2 rev/h:

SOLVING FOR W AND VI w = 26,332 rev/h V= 24.8 rev/h

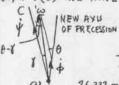
PROBLEM 18,128 FOR SECTION A:

(a) ANGLE BETWEEN SPIN AXIS AND W IS 57 ILL 8 = 2.7984" NOW: $\frac{I}{I} = \frac{6\alpha^4}{3a^2 + L^2} = \frac{6(9)^4}{3(9)^2 + L^2}$ $\overline{3(9)^2+(45)^2}=0.21429$

EQ.(18.49); $tan Y = I tan \theta$ tan8 = 0.21429 tan &

tant = tan 8 = tan 2.7984° = 0.22811 D= 12.850 A=12.85°

(b) AND(c) WE HAVE W= 26.332 rev/h, 8= 2,7984. AND 8 = 12,850°



LAW OF SINES: Sin 8 Sin (0-8)

26.332 rev/h_ Sin 12,850 - sin 2,7984° 5in 10.052°

SOLVING FOR + AND Y:

(b) \$ = 5,781 rev/h

\$ = 5.78 rev/h

V= 20.665 RV/h

1 = 20.7 rev/h

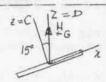
18.129

GIVEN:

COIN SPINS AT THE RATE OF GOO PPM ABOUT AND GC PERPENDICULAR TO COIN AND PRECESSES ABOUT VERTICAL DIRECTION GD.

FIND:

(a) ANGLE BETWEEN W AND GD, (b) RATE OF PRECESSION ABOUT GD.



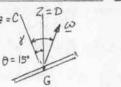
!T FULLOWS PROM THE ABOVE STATEMENT THAT HE IS DIRECTED AS SHOWN AND THAT THE ANGLE BETWEEN THE AXES OF SPIN AND PRECESSION IS 0 = 15"

$$I = I_{\tilde{e}} = \frac{1}{2} m \tilde{z}^{\epsilon}$$

EQ. (18.49):

$$tan8 = \frac{I}{F} tan \theta = 2 tan 15°$$

(a) ANGLE BETWEEN () AND GD



THE ANGLE & WE HAVE FOUND IS THE ANGLE BETWEEN W AND GC. THE ANGLE BETWEEN W AND GD 15

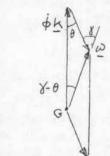
$$\delta - \theta = 2.8, 187^{\circ} - 15^{\circ}$$

= 13. 187°
 $\delta - \theta = 13.19^{\circ}$

(b) RATE OF PRECESSION

THE RATE OF SPIN 15 V = 600 rpm RESOLVING THE AN EULAR VELOCITY W INTO

ITS SPIN COMPONENT Y & AND ITS PRECESSION COMPONENT &K, WE DRAW THE FOLLOWING DIAGRAM!



LAW OF SINES: sin(Y-7) 5in(1-h) =(600rpm) sin 28.187 Sin 13, 1876 ф= 1242 грт ◀

WE NOTE FROM DIAGRAM THAT THE PRECESSION IS RETROGRADE

THIS COULD HAVE BEEN ANTICI PATED, SINCE I/I' = 2 > 1.

18.130 SOLVE SAMPLE PROB. 18.6, ASSUMING THAT THE METEURITE STRIKES THE SATELLITE AT C WITH Y = (2000 m/s) L.

(a) ANGULAR VELOCITY AFTER IMPACT

FROM SAMPLE PROB. 18.6:

I = I, = = = ma

$$I' = I_{2} = I_{3} = \frac{5}{4} ma^{4}$$

ANG, MOMENTUM AFTER IMPACT

$$H_G = \frac{1}{2} \times M_0 \cdot V_0 + I \cdot W_0 \cdot K$$

BUT
$$H_6 = I'\omega_2 i + I'\omega_8 j + I\omega_2 k$$
 (2)

EQUATING THE CUEFF, OF THE UNITVECTORS IN (1) HAD (2) $\omega_z = 0 \qquad \omega_y = -\frac{m_0 v_0 a}{\tau^2} = -\frac{H}{5} \frac{m_0 v_0}{m a}$

$$\omega_{g} = -\frac{m_{0}v_{0}u}{I^{2}} = -\frac{1}{5}\frac{m_{0}u}{m_{0}}$$

$$\omega_{g} = \omega_{0} + \frac{m_{0}v_{0}u}{I} = \omega_{0} + 2\frac{m_{0}v_{0}}{m_{0}}$$

GIVEN DATA: W = 60 rpm = 6.283 rad/s

$$\frac{m_b}{m_1} = 0.001$$
 $a = 0.8 m$ $v_o = 2000 m/s$

WE FIND

$$\omega_{\lambda} = 0$$
 $\omega_{\dot{\beta}} = -2 \text{ rad/s}$

$$\omega = \sqrt{\omega_{g}^{2} + \omega_{z}^{2}} = 11.459 \text{ rad/s}$$
 $\omega = 109.4 \text{ rpm}$

(1)

$$\cos \delta_{\lambda} = 0$$
 $\cos \delta_{\beta} = \frac{i \omega_{\beta}}{i \omega} = -0.17453$ $\cos \delta_{\gamma} = \frac{i \omega_{\beta}}{i \omega} = 0.48464$

$$\chi_2 = 90^\circ$$
, $\chi_y = 100.05^\circ$, $\chi_z = 10.05^\circ$

(6) PRECESSION AXIS

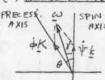
SINCE IT IS DIRECTED ALONG HE, WE USE DW (1) AND WRITE

$$\begin{aligned} H_{\chi} &= 0 \; , \; H_{\psi} &= -m_0 \, V_0 \, a = -\frac{m}{1000} \; (2000)(0.8) = -(1.6) \, m \\ H_{\chi} &= I \, \omega_0 + m_0 \, V_0 \, a = \frac{1}{2} \, m \, \alpha^2 \, \omega_0 + m_0 \, V_0 \, a \\ &= \frac{1}{2} \, m \, (0.8)^4 (6.283) + (1.6) \, m = (3.6106) \, m \end{aligned}$$

$$H_{z} = \sqrt{H_{y}^{1} + H_{z}^{2}} = (3.7492) \text{ as}$$

 $\cos \theta_{y} = 0$, $\cos \theta_{y} = \frac{H_{y}}{H_{x}} = -0.40515$, $\cos \theta_{z} = \frac{H_{z}}{H_{x}} = 0.91425$ DIRECTION OF PRECESSION AYIS IS

(C) RATES OF PRECESSION AND SPIN



0 = 02 = 23,9° 8=82 = 10,05° B-8= 13.85°

sin (0-8) Sin 23, 9° Sin 10,05° 5113850 SOLVING FOR & AND IF

RATE OF PRECESSION = \$ = 47.1 rpm

RATE OF SPIN = Y = 64,6 rpm

18.131 and 18.132

GIVEN:



DISK OFMASS M IS FREE TO ROTATE ABOUT A B.

FORK-ENDED SHAFT OF NEGLIGIBLE MASS IS FREE TO ROTATE
IN BEARING C.

PROBLEM 18.131:

INITIALLY, \$\theta_0 = 90^\theta_0^\theta_0 = 0,
\$\phi_0 = 8 \text{ rad/s}.

IF DISK SLIGHTLY DISTURBED FIND IN ENSUING MOTION

(a) MINIMUM VALUE OF \$\phi_0\$.

PROBLEM 18.132:

INITIALLY 0=30°, 0=0, = Bradle.

FIND IN ENSUING MOTION:

(a) RANGE OF VALUES OF 0, (b) MINIMUM &, (c) MAXIMUM &.



USING THE AXES GXYZ: W = 0 1 + + sin 0 1 + + cos 0 K CONSERVATION OF ANGULAR MOMENTUM:

SINCE DISK IS FREE TO KOTATE
ABOUT THE Z AXIS, WE HAVE $H_Z = Constant \qquad (1)$ BUT $H_Z = H_Z \sin \theta + H_Z \cos \theta$

 $H_Z = I_g \omega_g \sin\theta + I_z \omega_z \cos\theta = \frac{1}{4} m d + \sin^2\theta + \frac{1}{2} m d + \cos^2\theta$ $= \frac{1}{4} m d + (\sin^2\theta + 2\cos^2\theta) = \frac{1}{4} m d + (1+\cos^2\theta)$ USING THE INITIAL CONDITIONS, EU (1) YIELDS $+ (1+\cos^2\theta) = + (1+\cos^2\theta_0) \qquad (2)$

CONSERVATION OF ENERGY

SINCE NO WORK IS DONE, WE HAVE T = constant (3)

WHERE

 $T = \frac{1}{2} \left(I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} + I_{\lambda} \omega_{\lambda}^{2} \right)$ $T = \frac{1}{2} \left(\frac{1}{4} m \alpha^{2} \dot{\theta}^{2} + \frac{1}{4} m \alpha^{2} \dot{\phi}^{2} \sin^{2} \theta + \frac{1}{2} m \alpha^{2} \dot{\phi}^{2} \cos^{2} \theta \right)$ $= \frac{1}{8} m \alpha^{2} \left[\dot{\theta}^{2} + \dot{\phi}^{2} \left(\sin^{2} \theta + 2 \cos^{2} \theta \right) - \frac{1}{8} m \alpha^{2} \left[\dot{\theta}^{2} + \dot{\phi}^{2} \left(1 + \cos^{2} \theta \right) \right] \right]$

USING THE INITIAL CONDITIONS, INCLUDING $\theta_0 = 0$, EU. (3) YIELDS $\theta^2 + \frac{1}{4}^2(1+\cos^2\theta) = \frac{1}{4}^2(1+\cos^2\theta_0)$ (4)

PROBLEM 18, 131

(a) WITH $\theta_0 = 90^\circ$ AND $\phi_0 = 8 \text{ rad/s}$, EQ.(2) YIELDS $\dot{\phi} = \frac{8}{1 + \cos^* \theta}$ $\dot{\phi}$ IS MINIMUM FOR $\theta = 0$: $\dot{\phi}_{min} = 4.00 \text{ rad/s}$

(b) Eq.(4) YIELDS $\dot{\theta}^2 = 64 - \dot{\phi}^2 (1 + \cos^2 \theta) = 64 (1 - \frac{1}{1 + \cos^2 \theta})$

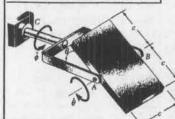
 $\dot{\theta}_{mn}^{2} = 6.4(1-\frac{1}{2}) = 32$ $\dot{\theta}_{mnx} = 5.66 \, rad/5$

PROBLEM 18.132

(a) WITH $\theta_0 = 30^\circ$, $\phi_0 = 8 \text{ rad/s}$ (N (2) $\phi(1+\cos^2\theta) = 14$ $\phi = 14/(1+\cos^2\theta)$ (5) SUBSTITUTE IN (4) AND SOLVE FOR θ^2 : $\theta^2 = 112 - \frac{196}{1+\cos^2\theta}$ (6) SINCE $\theta^2 \ge 0$, WF MUST HAVE $1+\cos^2\theta \ge \frac{196}{112} = 30^\circ \theta \le 30^\circ$ (b) FROM (5), ϕ 15 MINIMUM FOR $\theta = 0$: $\phi = 7.00 \text{ rad/s}$

(C) FROM (6), & is MAXIMUM FOR 0=0: \$\display = 7.00 rads

18.133 and 18.134



GIVEN:

PLATE OF HASS M 10 FREE TOROTH TE ABOUT AB.

FURK-ENDED SHAFT OF NEGLIGIBLE MASS IS FREE TO ROTATE IN BEARING C.

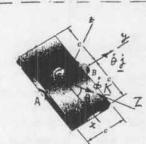
PROBLEM 18.133:

INITIALLY 8, = 30,

\$\text{\text{\$\texi{\$\text{\$\text{\$\text{\$\text{\$\}}\exititt{\$\t

FIND IN ENSUING MOTION (a) RANGE OF VALUES OF & (b) MINIMUM VALUE OF &, (c) MAXIMUM VALUE OF &.
PROBLEM 18, 134 .

INITIALLY \$ = 0, \$ = 0, \$ = 6 rad/s. IF PLATE
IS SLIGHTLY DISTURBED, FIND IN ENSUING MOTION
(a) MINIMUM VALUE OF \$, (b) MAXIMUM VALUE OF \$.



USING THE AXES GZYZ

\(\O = \frac{1}{2} \odots \od

MOMENTUM

SINCE PLATE IS FREE TO

ROTATE ABOUT Z AXIS.

Hz = constant (1)

BUT Hz = Hz cos 0 + Hz sin 0

 $\begin{aligned} H_Z &= I_z \omega_z \cos \theta + I_z \omega_z \sin \theta = \frac{1}{12} m c^2 + \cos^2 \theta + \frac{5}{12} m c^2 + \sin^2 \theta \\ &= \frac{1}{12} m c^2 + \left(\cos^2 \theta + 5 \sin^2 \theta\right) = \frac{1}{12} m c^2 + \left(1 + 4 \sin^2 \theta\right) \\ VSING THE INITIAL CONDITIONS, EQ (1) YIELDS \\ &+ \left(1 + 4 \sin^2 \theta\right) = \frac{1}{12} (1 + 4 \sin^2 \theta_0) \end{aligned} \tag{2}$

CONSERVATION OF ENERGY

SINCE NO WORK ISDONE, WE HAVE T = constant (3)

WHERE $T = \frac{1}{2} (I_2 \omega_2^2 + I_4 \omega_2^2 + I_4 \omega_2^2)$

 $T = \frac{1}{L} \left(\frac{1}{12} m c^{2} \dot{\phi}^{2} \cos^{3}\theta + \frac{1}{3} m c^{2} \dot{\theta}^{2} + \frac{5}{18} m c^{2} \dot{\phi}^{2} \sin^{3}\theta \right)$ $= \frac{1}{L_{H}} m c^{2} \left[4 \dot{\theta}^{2} + \dot{\phi}^{2} (\cos^{2}\theta + 5 \sin^{2}\theta) \right] = \frac{1}{L_{H}} m c^{2} \left[4 \dot{\theta}^{2} + \dot{\phi}^{2} (1 + 4 \sin^{2}\theta) \right]$

USING THE INITIAL CONDITIONS, INCLUDING $\theta = 0$, EQ. (3) YIELDS $4\theta^2 + \frac{1}{2}(1+4\sin^2\theta) = \frac{1}{2}(1+4\sin^2\theta)$ (4)

PROBLEM 18.133

(a) WITH \$6 = 30 AND \$6 = 6 rads IN (2) AND (4):

 $\phi(1+4\sin^2\theta)=12$ $4\dot{\theta}^2+\dot{\phi}^2(1+4\sin^2\theta)=72$ (2,4) ELIMINATING $\dot{\phi}^2$ AND SOLVING FOR $\dot{\theta}^2$: $\dot{\theta}^2=18-\frac{36}{1+45}$ (5)

FOR $\theta \ge 0$: $1+4\sin^2\theta \ge 2$, $\sin\theta \ge \frac{1}{4}$, $30^\circ \le \theta \le 150^\circ$. (b) FROM (2'), ϕ ISMIN. FOR $\theta = 90^\circ$: $\phi_{\min} = 2.40$ rad/s

(c) FROM (5), \$ IS MAX. FOR \$=90: \$\frac{1}{2}\$ = 3,29 red/s

PROBLEM 18.134

(a) WITH θ=0, φ=6 rad/s, EQ (2) TIELDS \$= 6 / 1+45in²θ

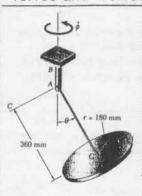
\$\frac{1}{4}\$ IS MINIMUM FOR θ=90°: \$\frac{1}{4}\$ min = 1,200 rad/s

(b) EQ.(4) YIELDS: 40 = 36 - 4(1+45in b) = 36(1 - 1+45in b)

 $4\theta_{may}^2 = 36(1-\frac{1}{5})$ $\theta_{may}^2 = 7.20$ $\theta_{max} = 2.68 \text{ rad/s}$

18.135 and 18.136

GIVEN:



DISK WELDED TO RUDA -OF NEGLIGIBLE MASS COMPECTED BY CLEVIS TO SHAFT AB. ROD AND DISK FREE TO KOTATE ABOUT AC; SHAFT FREE TO ROTATE ABOUT VERTICAL AXIS. INITIALLY, 0 = 40°, 0 = 0. PROBLEM 18, 135: KNOWING THAT \$ = 2 \$0 FIND: (a) +min, (b) + PROBLEM 18.136:

USING AXES AZYZ, WITH Y POINTING INTO PAPER. w= +sin Bi- Of + +cost k 12 = 1y = 17 ma2, 12 = 2 ma H= IW= 17 ma + 5100 H2 = I, W2 = 1 ma + cost

KNOWING THAT Danin = 30°,

FIND: (a) \$, (b) \$ max

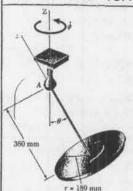
CONSERVATION OF ANG, HOM, ABOUT Z SINCE ONLY FORCES ARE REACTIONAT A AND W=-mg K, WE HAVE ZM = O AND H = constant. THUS. $H_7 = H_2 \sin \theta + H_2 \cos \theta = \frac{1}{4} ma^2 \phi (17 \sin^2 \theta + 2 \cos^2 \theta)$ Hz= 1 ma + (2 + 15 sin 0) = constant USING THE INITIAL CONDITIONS, EQ. (1) YIELDS + (2+15 sin A) = 17 4

CONSERVATION OF ENERGY T= (I, W2+ I, W2+ I, W2) = = 1 ma (174 sin 0+170+ 2 + cos 0) T= = ma2 [(2+15 sin' 0) +2+17 62] V=-2mgacost USING THE INITIAL CONDITIONS, WE WRITE T+V= const. (R+15 sin2 0) + 17 82-16 \$ cost = 17 82

PROBLEM 18, 135 (a) LET = + max=2+ IN(2): 2+ (2+15 sin 0)=17+ 2+15 sin 0 = 8.5, sin 0 = VO. 4333. 0 = 41,169° Omin = 41,2° (b) LET #=41,169°, 6=0, +=2 + IN (3): (2+15 sin 41.1690)(40) - 16 \$ cos 41.169 = 17% $17 \phi_0^2 = 12.044 \left(\frac{9.81}{6.18}\right) \quad \phi_0^2 = 38.63$ φ = 6.21 rad/s

PROBLEM 18,136 (a) LET B=30" IN (2): \$\phi(2+3,75)=17\$\phi_0, \$\phi=\frac{17}{5.75}\$\phi_0\$ LET 0=30°, 0 = 0 IN (3); $(2+3.75) \dot{\phi}^2 - 16 \left(\frac{9.81}{0.18}\right) \cos 30^\circ = 17 \, \Phi_0$ 5.75 (17) \$ -872 cos 30° = 174 17(17 -1) \$ = 872 cos30° 40 = 4,74 49 mays 4,76 rad/s (b) FROM (4): \$\dagger_{\text{along}} = \frac{17}{5175}(4.7649)

GIVEN: *18.137 and *18.138



DISK WELDED TO ROD AS OF NEGLIGIBLE MASS SUPPORTED BY BALL AND SOCKET AT A. INITIALLY, 0=90°, += 6=0. PROBLEM 18, 137: KNOWING THAT TO = 50 rads, FIND: (a) Dmin , (b) \$ AND IF FOR B = 0 min PROBLEM 18, 138: KNOWING THAT Omin = 30, FIND: (a) y, (b) & AND Y FOR 0 = Onin

ARE THE REACTION AT A AND THE WEIGHT W=-MgK AT 6,

USING AXES AZYZ, WITH & AXIS POINTING INTO PAPER ω=+sin01-0j+(+++cos0)k $I_{1} = I_{2} = \frac{17}{4} ma^{2}, \quad I_{2} = \frac{1}{2} ma^{2}$ H = ma [17 \$ sin 8 i - 17 8 j +2 (V+ \$ cos 0) k] CONSERVATION OF ANG. MOMENTUM SINCE THE ONLY EXTERNAL FORCES

WE HAVE $ZM_{7}=0$, $ZM_{8}=0$. SINCE Z IS PART OF A NEWTONIAN FRAME, IT FOLLOWS THAT HZ= CONST.; BECAUSE OF THE AXISMAMETRY OF THE DISK, IT ALSO FOLLOWS THAT H_= CONST. (SEE PROB. 18.139). VSING THE INITIAL CONDITIONS, WE WRITE #+ + cas B = To $H_2 = const.$ NOTING THAT HZ = HAK = MA [17 + 5 in + 2 (+ + + cos +) cos +] AND SUBSTITUTING FROM(1) FOR THE INSIDE PARENTHESIS 17 \$ sin2 0 + 2 \$ cos 0 = 0 Hy=wnst. CONSERVATION OF ENERGY $T = \frac{1}{2} \left(J_z \omega_A^2 + J_z \omega_y^2 + J_z \omega_z^2 \right) = \frac{1}{2} \frac{\sin \alpha}{4} \left[17 \dot{\phi}^2 \sin^2 \theta + 17 \dot{\theta} + 2 \left(\dot{\psi} + \dot{\phi} \cos \theta \right) \right]$ T= 1 ma (17 4 sin' 0 + 17 0 + 2 40) 1=-2 mg a cos o T+Y=const : 17+ sin'8 +170 + 270 - 16 = cool = 170 \$ sin20 + \$2 = 16 9 cost

PROBLEM 18.137 (a) PROM (2): $\phi = -\frac{2}{17} \psi_0 \frac{asb}{sin^2 \theta} = -\frac{2}{17} (50 \text{ rad/s}) \frac{asb}{sin^2 \theta}$ CARRY INTO (3) AND LET \$ = 0 FOR &= Dain : $\frac{10 \times 10^{3}}{16 \times 17} \frac{0.18}{7.81} \cos \theta = 1 - \cos^{2} \theta$ COS' + 0.67458 COS 8 - 1 = 0

COSO= (-0.67458 ± 2,1107) = 0,71806 0=44,105° OR - 1, 3926 (IMPOSSIBLE) 0 = 44.1" (6) SUBSTITUTING 76 = 50 rad/s AND 0 = 44. 105" IN (2) AND (1) EQ.(2): \$=-\frac{2}{17}(50)\frac{\cos 44,105°}{} \$=-8.72 rad/s 5in 44.105° EQ.(1): V= 50- (-8.72) COS H 4.105° V= 56.3 rad/s PROBLEM 18,138 LET 0=30°, 0=0 IN (3): \$\display=\frac{16}{17} \frac{4,81 m/s}{0.18m} \frac{ccs30}{5in^230}, \display=\frac{+}{2} \frac{13,330}{5in^230} FROM (2), WE NOTE THAT \$ <0 FOR \$ >0. THUS: \$=-13.33 rad/s Eq.(2): $\psi_0 = -\frac{17}{2}(-13.33) \frac{\sin^2 30}{\cos 30} = 32.708$, $\psi_0 = 32.7 \text{ ma/s}$ EU.(1): 1 = 32.700 - (-13,33) cos30°

₩ = 44,3 rad/s ~

\$ = 14.09 rad/s

*18.139



GIVEN:

TOP WITH FIXED POINT D P. D. T = EULERIAN HNGLES I = MDM, OF INERTIA ABOUT & AXIS - TRANSVERSE AXE THROUGH D.

SHOW THAT!

 $I'\dot{\phi}\sin^2\theta + I(\dot{\psi} + \dot{\phi}\cos\theta)\cos\theta = \alpha \quad (1)$ $I(\dot{\psi} + \dot{\phi}\cos\theta) = \beta$

(b) W= const. AND \$= FUNCTION OF 8



WEUSE FRAME DRUZ WITH Y AXIS POINTING INTO PAPER

ANG VELOCITY OF TOP ω=- φsinθ i+θj+(Y+4cosθ) k(A) ANG. VELOCITY OF PRAME! A =- + sinti+tj++cost (B) ANG, MOMENTUM ABOUT OF

Ho= Iwai + Iywyj+ Izwzk H =- I' + sin 0 i + I' + I (+ + coso) K

(a) WE RECALL IM = Ha (18.27)SINCE THE ONLY EXTERNAL FORCES ARE THE REACTION AT O AND THE WEIGHT W= - mg K AT G, WE HAVE ZM = 0 AND FRUM (18,27) H = D. SINCE THE Z AXIS IS PART OF A NEWTONIAN FRAME OF REFERENCE. IT FOLLOWS THAT Hy = constant. BUT Hy = H - K. SUBSTITUTING FOR HO FROM (C) AND NOTING THAT i.K = - sin b, j. K = O, k.K = cost, WE HAVE H= H.K =- 1'+ sint (-sint)+0+I(++465t) cost THUS: I'+sin'0 + I(+++cos +)cos += α (1) € WHERE ON IS A CONSTANT.

WE OBSERVE THAT WE ALSO HAVE \(\Sigma_4 = 0 \), BUT WE CANNOT CONCLUDE THAT H, = const. , & INCE THE & AXIS IS NOT PART OF A NEWTONIAN FRAME OF REFERENCE USING ER. (1828), WE WRITE

ENO = (HO) OxYZ + TXHO (18,28)

SUBSTITUTING FROM (B) AND (C) INTO (18,28), IM = - I'd (+ sino) + I') + Id (++ + wso) + + CONSIDERING ONLY THE COEFFICIENTS OF K, WE OBTAIN EM= Ift (++cost)-I'+ + sint + I++ sint =0 BUT THE SECOND AND THIRD TERMS CANCEL OUT, DUE TO THE AXISYMMETRY OF THE TOP.

IM= IA(+++coso) = 0 I (++ cos 0) = B

WHERE A IS A CONSTANT (b) FROM EQ. (A) WE HAVE W = Y++ COSO AND, IN VIEW OF (2): W= = B/I = constant

SUBSTITUTING FOR I (++ + cos +) FROM (2) INTO (1): I'+ sin2 0 + B cos 0 = 0 \$ = 8-BC05 B (FUNCTION OF B) (5) I' Sint A

* 18.140

GIVEN:

TOP OF PROB. 18, 139

SHOW THAT:

(a) A THIRD EQUATION OF MOTION CAN BE OBTHINED FROM THE PRINCIPLE OF CONSERVATION OF ENERGY.

(b) BY ELIMINATING & AND Y FROM THAT EQUATION AND EQS(1) AND(2) OF PROB. 18,140 AN EQUATION \$= f(8) CAN BE OBTAINED, WHERE

$$f(\theta) = \frac{1}{l'} \left(2E - \frac{\beta^2}{l} - 2mgc \cos \theta \right) - \left(\frac{\alpha - \beta \cos \theta}{l' \sin \theta} \right)^2$$
 (1)

(c) BY INTRODUCING THE VARIABLE Z = cost, THE MAX. AND MIN. VALUES OF & CAN BE OBTAINED BY SOLVING THE CUBIC EQUATION

$$\left(2E - \frac{\beta^2}{I} - 2mgcx\right)(1 - x^2) - \frac{1}{I'}(\alpha - \beta x)^2 = 0$$
 (2)

(2) CONSERVATION OF ENERGY

 $T = \frac{1}{2} \left(I' \omega_{\chi}^2 + I' \omega_{\chi}^2 + I \omega_{\chi}^2 \right)$ REFERRING TO EQ. (A) OF PROB. 18. 139:

T=+[I'+"sin'8 + I'+ I (+++cos 8)"],

T+V = E: +[I'+ sin'+ I'+ I(+++cos 6)]+mgccos 0=E(

(b) SUBSTITUTING IN (6) FOR & FROM EQ. (5) OF PROB. 18.39 AND FOR (4+4 cost) FROM RW. (2) OF PROB. 18. 139, AND MULTIPLYING BY Z:

 $I'\left(\frac{\partial -\beta \cos \theta}{1'\sin^2 \theta}\right)^2 \sin^2 \theta + J'\dot{\theta}^2 + I\left(\frac{\beta}{I}\right)^2 + 2 \operatorname{mgc} \cos \theta = 2E$

$$\frac{(\alpha - \beta \cos \theta)^{2}}{I' \sin^{2} \theta} + I' \dot{\theta}^{2} + \frac{\beta^{2}}{I} + 2 mg \cos \theta = 2E$$

SOLVING FOR O, WE OBTAIN

WHERE

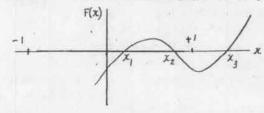
$$f(\theta) = \frac{1}{I'} \left(2E - \frac{\beta^2}{I} - 2mgc\cos\theta \right) - \left(\frac{\alpha - \beta\cos\theta}{I'\sin\theta} \right)^2$$

(C) SETTING COSO = 2 $J(\theta) = \frac{1}{I}, \left(2E - \frac{\beta^4}{I} - \lambda \operatorname{ang} c\lambda\right) - \frac{(\alpha - \beta\lambda)^2}{I^{12}(1 - \lambda^2)}$

LETTING f(B)= O AND MULTIPLYING BY I' (1-2), WE OBTAIN THE CUBIC EQUATION F(2)=0:

$$\left(2E - \frac{\beta^2}{I} - 2mgcx\right)(1 - x^2) - \frac{1}{I'}(\alpha - \beta x)^2 = 0$$

SOLVING THIS EQUATION WILL YIELD THREE VALUES OF IL . THE TWO VALUES COMPRISED BETWEEN - I AND +1 CORRESPOND TO THE MAX. AND MIN. VALUES OF B.



* 18.141 and * 18.142

1.142 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6 | 1.6

GIVEN: SOLID CONE.
INITIALLY, \$\theta_{=} 30^{\chi}, \theta_{=} 0
\$\forall_{=} 300 rad/s. USING EQ.(2)

OF PROB. 18. 140 AND

PROBLEM 18. 141:

KNOWING THAT \$\phi_{=} 20 rad/s.

FIND: (a) \$\theta_{max}\$,

(b) CORRESPONDING \$\psi_{AND}\$ \$\phi_{AND}\$

FIND: (a) \$\theta_{max}\$,

(b) CORRESPONDING \$\psi_{AND}\$ \$\phi_{AND}\$ \$\phi_{AND}\$

FIND: (a) \$\theta_{max}\$,

(b) CORRESPONDING \$\psi_{AND}\$ \$\phi_{AND}\$ \$\phi_{AND}\$

(c) VALUE OF \$\theta_{FR}\$ \$\phi_{FR}\$ \$\

WE FIRST DETERMINE THE FOLLOWING CONSTANTS: $I = \frac{3}{10} \text{ m } t^2 = \frac{3}{10} \text{ m } (0.25 \text{ ft})^4 = (18.75 \times 10^3 \text{ ft}^2) \text{ m}$ $I' = \frac{3}{5} \text{ m } (\frac{1}{4} x^2 + h^2) = \frac{3}{5} \left[\frac{1}{4} (0.25 \text{ ft})^2 + (0.75 \text{ ft})^2 \right] \text{ m}$ $= (346.875 \times 10^3 \text{ ft}^2) \text{ m}$ $c = AG = \frac{2}{4} h = \frac{3}{4} (0.75 \text{ ft}) = 562.5 \times 10^3 \text{ ft}$

FROM EGS. (2) AND (4) OF THE SOLUTION OF PROB. 18. 139
AND FROM EQ. (6) OF THE SOLUTION OF PROB. 18. 140,
USING THE APPROPRIATE INITIAL CONDITIONS.

 $\frac{PR0BLEM 18.141}{\beta = I(\dot{\psi}_{0} + \dot{\phi}_{0}\cos\theta_{0}) = (18.75\times10^{3}) m(300+20\cos30^{\circ})} = (5.94976)m$ $\propto = I'\dot{\phi}_{0}\sin^{2}\theta_{0} + \beta\cos\theta_{0} = (346.875\times10^{3}) m(20)\sin^{2}30^{\circ} + (5.94976)m\cos530^{\circ} = (6.88702) m$ $E = \frac{1}{2}[I'\dot{\phi}_{0}^{2}\sin^{2}\theta_{0} + I'\dot{\theta}_{0}^{2} + I(\dot{\psi}_{0} + \dot{\phi}_{0}\cos\theta_{0})^{2}] + mg\cos\theta_{0}$

 $E = \frac{1}{2} \left[1 + \frac{1}{6} \sin^2 \theta_0 + 1 + \frac{1}{16} + 1 + \frac{1}{16} \cos \theta_0 \right] + mq = 2000$ $= \frac{1}{2} \left[(3+6, 875 \times 10^{-3}) m (20)^3 \sin^3 30^2 + 0 + \frac{(5, 94976 \text{ in})^3}{18.75 \times 10^{-3}} \right] + m (32,2)(562,5 \times 10^{-3}) \cos 30^\circ = (977,020) m$

SUBSTITUTE IN EQ. (2) OF PROB. 18,140: $(2E - \frac{B^2}{2} - 2mgcx)(1-x^2) - \frac{1}{2}, (\alpha - \beta z)^2 = 0$ $(66.0593 - 36.225x)(1-x^2) - 2.88288(6.88702 - 5.94976z) = 0$ (a) SOLVING: $\chi = 0.743151$ (b) PR. (5) OF PROB. 18 134.

(b) PR.(5) of PROB. 18.139: $\phi = \frac{\alpha - \beta \cos \theta}{L^2 \sin^2 \theta} = \frac{6.88762 - 5.94976 \cos 42.0}{(346.875 \times 10^{-3}) \sin^2 42.0} = 15.8748 \text{ rad/s}$

FROM EQ.(2): $\psi = \frac{1}{I} - \frac{1}{4}\cos\theta = \frac{5.44476}{18.75 \times 10^{-3}} - (15.8748)\cos 42.0^{\circ}$ $-\psi = 306 \text{ rad/s}$; $\psi = 15.87 \text{ rad/s}$

PROBLEM 18, 142

 $E = \frac{1}{2} \left[1^3 \dot{\phi}_0^2 \sin^2 \theta_0 + 1^3 \dot{\theta}_0^2 + 1 \left(\dot{\phi}_0 + \dot{\phi}_0 \cos \theta_0 \right)^2 \right] + \text{Mig } \cos \theta_0$ $= \frac{1}{2} \left[(346.875 \times 10^{-3}) m \left(-4 \right)^2 \sin^2 30^2 + 0 + \frac{(5.56005 m)^2}{18.75 \times 10^{-3} m} \right] + m (32.2) (562.5 \times 10^{-3}) \cos 30^2 = 840.76 m$

SUBSTITUTE IN EQ.(2) OF PROB. (B. 140: $(2E - \frac{R^2}{2} - 2mgcx)(1-x^2) - \frac{1}{17}(\alpha - \beta x)^2 = 0$ $(32,765 - 36,225 \times)(1-x^2) - 2,88208(4,46827 - 5,56005x)^2 = 0$

(a) SOLVING: x = 0.37166, $\theta_{max} = 68.18$, $\theta_{max} = 68.2$ °
(b) F0.(5): $\dot{\phi} = \frac{\alpha - \beta \cos \theta}{1' \sin^2 \theta} = \frac{k.46827 - 5.56005 \cos 68.18}{(346.875 \times 10^{-3}) \sin^2 68.18} = 8.0335 \text{ rank}$ F0.(2): $\dot{\psi} = (\beta/1) - \dot{\phi} \cos \theta = 296.536 - 8.0335 \cos 68.18 = 293.55 \text{ rank}$ $\dot{\psi} = 294 \text{ rank} + 3.03 \text{ rank} = 3.03 \text{ rank}$

(c) ϕ REVERSES FOR $\alpha - \beta \cos \theta = 0$, $\cos \theta = \frac{4.46827}{5.56005}$, $\theta = 36.5°$

18.143



GIVEN:

RIGID BODY OF ARBITRARY SHAPE SUPPORTED AT ITS MASS CENTER O AND SUBJECTED TO NO FURCE (EXCEPT AT SUPPURT O).

SHOW THAT;

T= constant
PROJ. OF \(\Omega\) ALONG \(H_0 = constant \)

(b) TIP OF \(\Omega\) DESCRIBES CURITE ON
FIXED PLANE (THE INVARIABLE FLAME)
PERP. TO \(H_0 \) AND AT DISTANCE 2T/H0
FROM O.

(c) WITH RESPECT TO PRINCIPAL AXES

OXYZ ATTACHED TO FODYS \(\Omega\)

(a) Ho = constant (IN MAGNITIWE & UIR!)

AFTERNIS TO DESCRIBE A CURVE ON ELLIPSOID OF EQUATION

[4] 4] 4] + 1 4 = 2T (POINSOT ELLIPSOID)

(a) From Eq. (18. 27): $\Sigma M_0 = H_0$ SINCE $\Sigma M_0 = 0$: $H_0 = constant$ (1) T+V=const.; SINCE V=const, T=constant (2)

WE RECALL FROM PROB. 18.37 THAT HOW = 2T

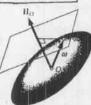
BUT $H_0 \omega = H_0 \omega \cos \beta$ THUS $H_0 \omega \cos \beta = 27$ PROJ. OF ω ON $H_0 = \omega \cos \beta = \frac{27}{H_0} = \text{const}$ (3)

(b) IT FOLLOWS FROM (3) THAT THE TIP OF W MUST REMAIN IN A PLANE L HO ATIA DISTHINCE 2T/HO PROM O.

(C) FROM EW. (18.20): $T = \frac{1}{2} \left(I_{\lambda} \omega_{\lambda}^{2} + I_{y} \omega_{y}^{2} + I_{z} \omega_{z}^{2} \right)$ FROM (2) IT FOLLOWS THAT $I_{\lambda} \omega_{\lambda}^{2} + I_{z} \omega_{y}^{2} + I_{z} \omega_{z}^{2} = 2T = const.$

EO.(4) IS THE EQUATION OF AN ELLIPSOID ON WHICH THE TIP OF ω MUST LIE. THIS IS POINSOT ELLIPSOID. COMPARING EQ.(4) WITH EO.(9.44) OF SEC. 9.17, WE NOTE THAT POINSOT ELLIPSOID HAS THE SAME SHAPE AT THE ELLIPSOID OF INERTIA OF THE BODY, BUT A DIFFERENT SIZE.

18.144



GIVEN:

POINSOT ELLIPSOID AND INVAKIABLE PLANE DEFINED IN PROB. 18.143. SHOW THAT:

(a) THE ELLIPSOID IS TANGENT TO THE PLANE,

(b) AS THE BODY MOVES THE POWSOT ELLIPSOID ROLLS ON THE INVAKIABLE PLAYE.

(a) AT THE TIP OF ω THE DIRECTION OF THE NORMAL TO THE ELLIPSOID IS THAT OF <u>Brad</u> $F(\omega_{\lambda},\omega_{y},\omega_{z})$, where F Denotes the LEFT-HAND MEMBER OF EQ. (4) OF PROB. 18.143. FROM SEC. 13.7: g rad $F = \frac{2F}{D\omega_{z}} + \frac{2F}{2D\omega_{z}} \pm \frac{2F}{2D\omega_{z}} \pm \frac{2F}{2D\omega_{z}}$

= 2 [] w, 1 + 2 [w, j + 2 [w, k = 2 (] w, 1 + I] w, j + I w, k) = 2 H

THUS, THE NOKITAL TO POINSOT ELLIPSOID IS PARALLEL TO HO, IT FOLLOWS THAT
POINSOT ELLIPSOID IS TANGENT TO THE INVARIABLE PLANE (CONTINUED)

* 18.144 continued

(b) THE POINSOT ELLIPSOID IS PART OF THE BODY WHOSE MOTION IS BEING ANALYZED, AND ITS POINT OF CONTACT WITH THE INVPRINGLE PLANE IS THE TIP OF THE VECTOR W. SINCE W DEFINES THE INSTANTANEOUS AXIS OF ROTATION, THE POINT OF CONTACT HAS ZERO VELOCITY, THUS, THE POINSOT ELLIPSOID ROLLS ON THE INVARIABLE PLANE (WITH IT'S CENTER O REMAINING FIXED).

* 18.145 GIVEN:

AXISYMMETRICAL RIGID BODY SUPPORTED AT ITS MASS CENTER O AND SUBJECTED TO NO FORCE (EXCEPT AT SUPPORT 0).

USING THE RESULTS OBTAINED IN PROBS. 18,143-144, SHOWTHAT THE POINSOT ELLIPSUID IS AN ELLIPSUID OF REVOLUTION AND THE SPACE AND BODY CONES ARE BOTH CIRCULAR AND TANGENT TO EACH OTHER. FURTHER SHOW THAT

(a) THE TWO CONES ARE TANGENT EXTERNALLY AND THE PRECESSION IS DIRECT WHEN I < I' WHERE I = MOM OF INERTIA ABOUT AXIS OF SYMMETRY

I'= - - -- TRANSVERSE AXIS (b) THE SPACE CONE IS INSIDE THE BUDY CONE AID THE PRECESSION IS RETROGRADE WHEN I>I'.

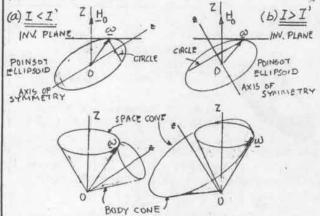
CHOOSING & ALONG THE AXIS OF SYMMETRY, WE HAVE I = I = I' AND I = I . SUBSTITUTE INTO (4) OF PIB. 143: $I'(\omega_x^2 + \omega_y^2) + I\omega_1' = const.$

WHICH IS THE FOUNTION OF AN ELLIPSCID OF REVOLUTION. IT FOLLOWS THAT THE TIP OF W DESCRIBES CIRCLES ON BOTH THE POINSOT ELLIPSOID AND THE INVARIABLE ELLIPSOIDS IN THE STANDARD FORM PLANE, AND THAT THE VECTOR W ITSELF DESCRIBES CIRCULAR BODY AM SPACE CONES

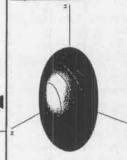
THE POINSOT ELLIPSOID, THE INVARIABLE PLANE AND THE BODY AND SPACE CONES ARE SHOWN BELOW

FUR CASES a AND b:

DIRECT PRECESSION



* 18.146



GIVEN:

RIGID BODY OF ARBITRARY SHAPE PHD ITS POINSOT ELLIPSOID (CF. PROBS. 18,143 AND 18,144, SHOW THAT:

(a) CURVE DESCRIBED BY TIP OF W ON POINSOT ELLIPSOID 15 DEFINED BY

 $I_x \omega_x^2 + I_y \omega_y^2 + I_z \omega_z^2 = 2T = \text{constant}$ (1) $I_{\tau}^{2}\omega_{x}^{2} + I_{y}^{2}\omega_{y}^{2} + I_{x}^{2}\omega_{x}^{2} = H_{O}^{2} = \text{constant}$

AND CAN THUS BE OBTAINED BY INTERSECTING THE POINSOT ELLIPSOID WITH THE ELLIP SOID DEFINED BY (2)

(b) ASSUMING Ix> Iy > Iz, THE CURVES (CALLED POLHODES) OBTAINED FOR VARIOUS VALUES OF HO HAVE THE SHAPES INDICATED IN FIGURE

(C) THE BODY CAN ROTATE ABOUT A FIXED AXIS ONLY IF THAT AXIS COINCIDES WITH ONE OF THE PRINCIPAL AYES, THIS MOTION BEING STABLE IF THE AXIS IS THE MAJOR OR MINOR AXIS OF THE POINS OF ELLIPSOID (& OR & AXIS) AND UNSTABLE IF IT IS THE INTERMEDIATE AXIS (& AXIS).

(a) EQ.(1) IN STATEMENT EXPRESSES CONSERVATION OF ENERGY; THIS IS EQ. (4) OF PROB. 18, 143. WE NOW EXPRESS THAT THE MAGNITUDE OF H IS CONSTANT:

 $H_0^2 = H_x^2 + H_y^2 + H_z^2 = I_x^2 \omega_x^2 + I_y^2 \omega_y^2 + I_z^2 \omega_x^2 = const.$ WHICH IS EW (2) IN STATEMENT, SINCE THE COORDINATES WINY, WI OF THE TIP OF W MUST SATISFY BOTH FOS. (1) AND (2), THE CURVE DESCRIBED BY THE TIP OF W IS THE INTERSECTION OF THE TWO ELLI PSOIDS.

(6) WE NOW WRITE THE EQUATIONS OF THE TWO

$$\frac{z^2}{a^2} + \frac{\dot{y}^2}{b^2} + \frac{z^3}{c^2} = 1$$

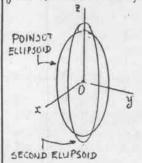
WHERE A, b, C ARE THE SEMI-AXES OF THE ELLIPSOID.

WE HAVE FOR POINSOT ELLIPSOID: (3)

 $\frac{\omega_{x}^{2}}{H_{0}^{2}/I_{x}^{2}} + \frac{\omega_{y}^{2}}{H_{0}^{2}/I_{y}^{2}} +$ FOR SECUND ELLIPSOID:

SINCE WE ASSUMED THAT I, > I, WE HAVE 27/1 < 27/1 < 27/1 AND HO/1 < HO/1 < HO/1

THUS, FOR BOTH ELLIPSOIDS, THE MINDR AXIS TO DIRECTED ALONG THE X AXIS, THE INTERMEDIATE AXIS ALONG THE A AXIS, AND THE MAJOR, AXIS ALONG THE & AXIS.



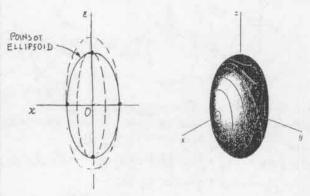
HOWEVER, BECAUSE THE RATIO OF THE MAJOR TO THE MINOR SEMIAXIS IS VI/I, POR THE POINSOT ELLIPSOID AND 1/12 FOR THE SECOND ELLIPSOID, THE SHAPE OF THE LATTER WILL BE MORE " PROHOUNCED".

(CONTINUED)

RETROGRADE PRECESSION

* 18.146 continued

THE LARGEST ELLIPSOID OF THE SECOND TYPE TO BE IN CONTACT WITH THE POINSOT ELLIPSOID WILL BE OUTSIDE THAT ELLIPSOID AND TOUCH IT AT ITS POINTS OF INTERSECTION WITH THE x AXIS, AND THE SMALLEST WILL BE INSIDE THE POINSOT ELLIPSOID AND TOUCH IT AT ITS POINTS OF INTERSECTION WITH THE z AXIS (SEE LEFTHAND SKETCH). ALL ELLIPSOIDS OF THE SECOND TYPE COMPRISED BETWEEN THESE TWO WILL INTERSECT THE POINSOT ELLIPSOID ALONG THE POLHODES AS SHOWN IN THE RIGHT-HAND FIGURE.

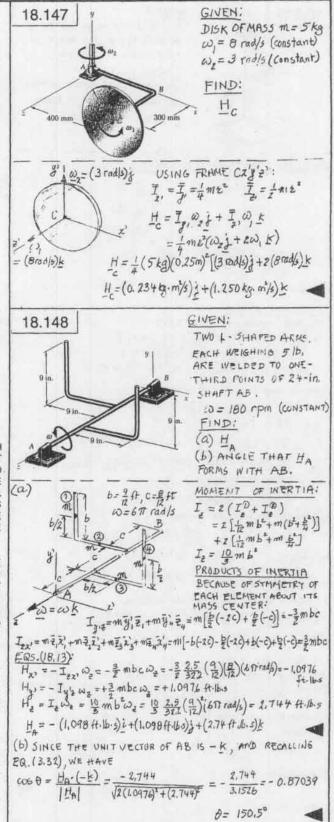


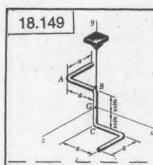
NOTE THAT THE ELLIPSOID OF THE SECOND TYPE WHICH HAS THE SAME INTERMEDIATE AXIS AS THE POINSOT ELLIPSOID INTERSECTS THAT ELLIPSOID ALONG TWO ELLIPSES WHOSE PLANES CONTAIN THE y AXIS. THESE CURVES ARE NOT POLHODES, SINCE THE TIP OF WILL NOT DESCRIBE THEM, BUT THEY SEPARATE THE POLHODES INTO FOUR GROUPS: TWO GROUPS LOOP AROUND THE MINOR AXIS (x AXIS) AND THE OTHER TWO AROUND THE MAJOR AXIS (z AXIS).

(c) IF THE BODY IS SET TO SPIN ABOUT ONE OF THE PRINCIPAL AXES, THE POINSOT ELLIPSOID WILL REMAIN IN CONTACT WITH THE INVARIABLE PLANE AT THE SAME POINT (ON THE x, y, OR z AXIS); THE ROTATION IS STEADY. IN ANY OTHER CASE, THE POINT OF CONTACT WILL BE LOCATED ON ONE OF THE POLHODES AND THE TIP OF $\underline{\omega}$ WILL START DESCRIBING THAT POLHODE, WHILE THE POINSOT ELLIPSOID ROLLS ON THE INVARIABLE PLANE.

A ROTATION ABOUT THE <u>MINOR</u> OR THE <u>MAJOR</u> AXIS (x OR z AXIS) IS <u>STABLE</u>: IF THAT MOTION IS DISTURBED, THE TIP OF $\underline{\omega}$ WILL MOVE TO A VERY SMALL POLHODE SURROUNDING THAT AXIS AND STAY CLOSE TO ITS ORIGINAL POSITION.

ON THE OTHER HAND, A ROTATION ABOUT THE INTERMEDIATE AXIS (z AXIS) IS <u>UNSTABLE</u>: IF THAT MOTION IS DISTURBED, THE TIP OF W WILL MOVE TO ONE OF THE POLHODES LOCATED NEAR THAT AXIS AND START DESCRIBING IT, DEPARTING COMPLETELY FROM ITS ORIGINAL POSITION.AND CAUSING THE BODY TO TUMBLE.





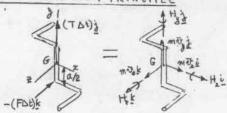
ROD OF MASS IN 15 HIT AT C IN NEGATIVE & DIRECTION. IMPULSE = - (F Dt) & .

FIND

IMMEDIATELY AFTER IMPACT
(a) ANG, YELOCITY OF ROD,

(b) VELUCITY OF G.

IMPULSE-MOMENTUM PRINCIPLE



(WEIGHT IS OMITTED, SINCE MONIMPULSIVE)

(a) ANGULAR VELOCITY

EQUATE MOMENTS ABOUT G:

MOHENTS AND PRODUCTS OF INERTIA:

$$\bar{I}_{2} = \frac{1}{12} \frac{m}{5} \alpha^{2} + 2 \frac{m}{5} \left(\frac{1}{12} \alpha^{2} + 2 \frac{\alpha^{2}}{4}\right) + 2 \frac{m}{5} \left(\alpha^{2} + \frac{\alpha^{3}}{4}\right) = 0.75 \text{ ma}$$

$$\bar{1}_{xy} = \frac{m}{5} \left(\frac{a}{2} \right) \left(\frac{a}{2} \right) + \frac{m}{5} \left(-a \right) \left(\frac{a}{2} \right) + \frac{m}{5} \left(\frac{a}{2} \right) \left(-\frac{a}{2} \right) + \frac{m}{5} (a) \left(-\frac{a}{2} \right) = -0.5 \, \text{ma}$$

$$\begin{split} \mathbf{I}_{d^2} &= \frac{m}{5} \left(\frac{a}{2} \right) - \frac{a}{1} \right) + \frac{m}{5} \left(-\frac{a}{2} \right) \frac{d}{1} = -0.1 \, \text{ma}^2 \\ \mathbf{I}_{e^2} &= \frac{m}{5} \left(-\frac{a}{2} \right) - a \right) + \frac{m}{5} \left(\frac{a}{2} \right) \frac{d}{1} = 0.2 \, \text{ma}^2 \end{split}$$

EQS. (18.17) AND DIVIDING BY ma":

$$H_{\lambda} = \overline{I}_{\lambda} \omega_{\lambda} - \overline{I}_{\lambda \gamma} \omega_{\gamma} - \overline{I}_{\lambda \gamma} \omega_{\lambda} : \frac{F \Delta t}{2 \pi i \alpha} = 0.35 \omega_{\lambda} + 0.3 \omega_{\beta} - 0.2 \omega_{\lambda}$$
 (1)

$$H_y = -I_{2y} \omega_2 + I_{y} \omega_y - I_{y} \omega_z$$
: $O = 0.3 \omega_z + \frac{2}{5} \omega_y + 0.1 \omega_z$ (2)

$$H_1 = -\tilde{\mathbf{I}}_2 \omega_2 - \tilde{\mathbf{I}}_{y2} \omega_y + \tilde{\mathbf{I}}_z \omega_z$$
: $0 = -\partial z \omega_z + \partial z \omega_y + \partial z \omega_z$

SOLVING FOS. (1), (2), (3) SIMULTHNEOUSLY:

$$\omega_2 = \frac{30}{6} \frac{F\Delta t}{ma}$$
 $\omega_y = -\frac{15}{6} \frac{F\Delta t}{ma}$ $\omega_z = \frac{10}{6} \frac{F\Delta t}{ma}$

74105: W = FAT (30 1-15 + 10 k)

(b) YELOCITY OF G

WE PIRST NOTE THAT THE GIVEN CONSTRAINTS REQUIRE THAT \$ -0. EQUATING THE COMPONENTS OF IMPULSE

AND MOMENTUM!

$$\sqrt{5} = 0$$
 $T\Delta t = 0$

$$\bar{v_2} = -\frac{F\Delta t}{m}$$

THEREFORE!

$$\bar{v} = -\frac{F\Delta t}{m} \kappa$$



GIVEN:

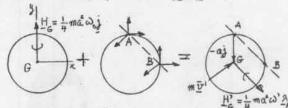
DISK OF MASS m SUPPORTED BY BALL AND SOCKET AT A ROTATES WITH CONSTANT $\omega = \omega_0 j$ WHEN OBSTRUCTION IS INTRODUCED AT B. IMPACT PERFECTLY PLASTIC (e=0).

FIND:

IMMEDIATELY AFTER IMPACT (a) ANGULAR VELOCITY OF DISK, (b) VELOCITY OF G.

IMPULSE - MOMENTUM PRINCIPLE

WE NOTE THAT IDIAM = 4 mat AND AB = 12 (1-1)



WE NOTE THAT U'= W'X AG = W'AAg X (-ag) = - 1 Wat

(a) EDUATE MOMENTS ABOUT AB OF ALL VECTORS AND COUPLES!

2AB. Hg + 0 = 2AB. (-aj x m v') + 2AB. Hg

L(i-i) - 1 mov i = 1(l-i). [-aix(-1-mwak)] + 2. H'

$$\frac{1}{\sqrt{2}}(\underline{i}-\underline{i}) \cdot \frac{1}{4} ma^{2} \omega_{0} \underline{i} = \frac{1}{\sqrt{2}}(\underline{i}-\underline{i}) \cdot [-a\underline{j} \times (-\frac{1}{\sqrt{2}} m\omega^{i} a\underline{k})] + \frac{1}{2} \underline{A}\underline{B} \underline{H}_{6}^{i}$$

$$-\frac{1}{4\sqrt{2}} ma^{2} \omega_{0} = \frac{1}{2} ma^{2} \omega^{i} + \frac{1}{4} ma^{2} \omega^{i}$$

$$\omega^{i} = -\frac{1}{2\sqrt{2}} \omega_{0}$$

$$\underline{\omega}' = \omega' \ \underline{2}_{AB} = -\frac{1}{3\sqrt{2}} \omega_0 \frac{1}{\sqrt{2}} (\underline{i} - \underline{j}), \quad \underline{\omega}' = \frac{1}{6} \omega_0 (-\underline{i} + \underline{j})$$

(b) RECALLING THAT T'= W'X AG,

$$\underline{\vec{v}}' = \frac{1}{6}\omega_o(-i+j)\dot{x}(-aj)$$
 $\underline{\vec{v}}' = \frac{1}{6}\omega_o ak$

18.151 GIVEN:

DISK OF PROB. 18.150

FIND:

KINETIC ENERGY LOST WHEN DISK HITS OBSTRUCTION.

BEFORE IMPACT :

$$T_0 = \frac{1}{2} I_{2|AM} \omega_0^2 = \frac{1}{2} I_{2|AM} \omega_0^2 = \frac{1}{8} ma^2 \omega_0^2$$

AFTER IMPACT:

BUT, FROM ANSWERS TO PROB. 18.150:

$$v'^{2} = (\frac{1}{6}\omega_{0}a)^{2} = \frac{1}{36}\omega_{0}^{2}a^{2}$$

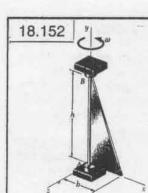
$$\omega'^{2} = \omega_{2}^{12} + \omega_{3}^{12} = \frac{\omega_{0}^{2}}{36}(1+1) = \frac{1}{18}\omega_{0}^{2}$$

THEREFORE:

$$T' = \frac{1}{2}m(\frac{1}{36}\omega_0^2a^2) + \frac{1}{2}\frac{md}{4}(\frac{1}{18}\omega_0^4) = \frac{1}{48}md\omega_0^4$$

KINETIC ENERGY LOST

= 5 maw

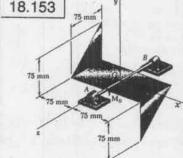


FROM BACK COVER:

GIVEN:

TRIANGULAR PLATE OF MASS M WELDED TO SHAFT SUPPORTED BY BEARINGS AT A AND B. PLATE ROTATES AT CONSTANT

DYNAMIC REACTIONS AT A AND B.



GIVEN:

SHEET-METAL COMPONENT OF MASS m = 600g. LENGTH AB = 150 mm . COMPONENT AT REST WHEN M = (49,5 mN·m) k 16 APPLIED.

IND: DYNAMIC REACTIONS AT A AND B (DJUST AFTER COUPLE IS APPLIED (b) 0.65 LATER

COMPUTATION OF MONTH AND PRODUCT OF INTETIA (Ig) AREA = 12 boh, A=26h, (I) AREA TAREA $(I_g)_{MASS} = \frac{1}{12} b^3 h \left(\frac{m}{\frac{1}{2}bh}\right) = \frac{1}{6} m b^3$

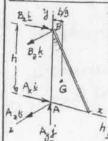
FROM SHMPLE PROB. 9.6 (PAGE 485 OF STATICS) ! (Izy) AREA = 1/4 6 h, (Izy) MASS = (Ixy) AREA 1 = 1/4 b h (1/4 h) = 1/2 mbh

WE ALSO NOTE THAT INE = 0

ANGULAR MOMENTUM HA

SINCE W = 0, W = W, W = 0, EW. (18.13) YIELD H=-Izywg=-1211 bhw, H= Izwy=6mbw, H=0 H=-12mbhwi++mbwj

EQUATIONS OF HUTION



(WEIGHT CHITTED FOR DYNAMIC REACTIONS) FRAME OF REPERENCE AZY + ROTATES WITH Q=W=Wj.

EQ.(18.28): EM = (HA)Azy+ DX HA

= 0 + WZ X HA

RECALLING (1) AND COMPUTING EMA: hjx (B, i+B, k) = wjx(-mbh wi+mb wj) -hBk+hBi = 1 mbhw*k

EQUATING THE COLFT. OF THE UNIT VECTORS:

$$B_{\lambda} = -\frac{1}{12} 7.16 \omega^{2} \qquad B_{\mu} = 0$$

$$B = -\frac{1}{12} 4116 \omega^{2} \dot{L}$$

ZQ. (18.1):

WHERE a = - Zwi = - 1 bwi ∑F=ma

THUS:

$$\underline{A} + \underline{B} = -\frac{1}{3}b\omega^2 i$$

$$\underline{A} = -\frac{1}{3}b\omega^2 i - (-\frac{1}{12}mb\omega^2 i)$$

$$\Delta = -\frac{1}{4} mb\omega^2 i$$

b=0.075m

MOMENT AND PRODUCTS OF INERTIA RECTANGLE 2: MASS = 3 m Iz = 1/2 (2m)(2b) = = = mb

I12 = I32 = 0

TRIANGLE 1: MASS = 7 m

FROM BACK COVER : $(\bar{I}_z)_{ARCH} = \frac{1}{36}b^4, A = \frac{1}{2}b^3, (\bar{I}_z)_{AASS} = (\bar{I}_z)_{ARCH} + (\bar{I}_z)_{ARCH} +$ $(\bar{I}_2)_{MASS} = \frac{1}{36}b''(\frac{\pm m}{\pm b^2}) = \frac{1}{108}mb^2$

FROM SAMPLE PROB. 9.6 (PAGE 485 8FSTATIS) (] 1/2) AREA = - 1/2 b", (] 1/2) MASS = - 1/2 b" (1/2) = - 1/2 m b"

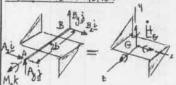
THEREFORE: $I_{z} = \frac{1}{2} + \frac{m}{6} d^{z} = \frac{1}{100} mb^{z} + \frac{m}{6} \left[b^{z} + \left(\frac{b}{3} \right)^{z} \right] = \left(\frac{1}{100} + \frac{10}{54} \right) mb^{z} = \frac{7}{36} mb^{z}$ $I_{32} = \bar{1}_{3'2}, + \frac{m}{6} \bar{y} \bar{z} = -\frac{1}{216} m b^3 + \frac{m}{6} (\frac{b}{3})(-\frac{b}{6}) = -\frac{1}{72} m b^3$ Izz= + + + = = = = 0+ + (-b)(-b) = 1 mb

TRIANGLE 3: BY SYMMETRY, SAME AS TRIANGLE 1.

FOR ENTIRE COMPONENT: I = = = mb + z (36 mb =) = 11 mb I = 2(- 1/2 mb') = - 1/6 mb' I = 2 (36 mb) = 1 mb

ANGULAR MUMENTUM Ho EUS. (18.7) WITH Wz = Wy = 0, Wz = W : H=-12 w2=-18mbw, H=-72 == 36mbw, H=70=118mbw HG = 1/36 mb w (-21+1+22k) (1)

EQUATIONS OF MUTION



ED. (18.22) AND USING (1): H = (H) + P ×H = 1 mb a(-2i+j+22t)+ 0 k x 1 m + w (-2i+j+22k)

(2)

EQUATING MUMENTS ABOUT B .:

26 x (A i + Ay i) + M & = = = = mb [(-2i+j+2k) x - (2j+i) w] 26 A j - 26 A i + Mok = 1 mb2 [- (2x + w) + (x - 20) + 22 x k]

EQUATING THE COEFT OF THE UNIT VECTORS:

 $A_i = \frac{1}{72} mb \left(2\alpha + \omega^2\right)$ 1 - 26Ay = - 1 mb (20+02)

1 26 Az = 36 mb (x-200) Az= 12 =116 (01-202) (3) (4)

(1) 10 = 11 mb x a = 18Mo/11mb (CONTINUED)

18.153 continued

WE RECALL THE RESULTS OBTAINED: $A_{3} = \frac{1}{72} mb (2\alpha + \omega^{2})$ $A_{2} = \frac{1}{72} mb (\alpha - 2\omega^{2})$ $\alpha = 18 M_{0} / 11 mb^{2}$ (2)
(3)
(4)

WITH GIVEN DATA: M = 0.0495 N·m, m = 0.6 kg. b = 0.075m:

 $EQ.(4): \ N = \frac{1}{72} (0.6)(0.675)(10.6)(0.675)^2 = 24 \ rad/5^2$ $EQ.(3): \ A_{\lambda} = \frac{1}{72} (0.6)(0.675)(24 - 2i\delta^2) = (15 - 1.25\omega^2)10^3 \text{N} \qquad (3')$ $EQ.(2): \ A_{\lambda} = \frac{1}{72} (0.6)(0.675)(2 \times 24 + \omega^4) = (30 + 0.625 \omega^3)10^3 \text{N} \qquad (2')$

(a) JUST AFTER COUPLE IS A PPLIED: LETTING (D=0 IN (3") HID (2"): Az = 15x10" N, Ay = 30 x10 K THUS: A = (15,00 m) + (30.0 m) 2

ZF=ma: A+B=0, B=-(15,00mN)i-(30.0mN)j

(b) AFTER 0.65: LETTING 10 = xt = (2+rad/s*)(0.6=)= 14.40 rad/s IN (3') AND (2'): A = (15-1.25(14.40)*) 10 N = -244,2 mN, Ay=(30+0.625(14.40)*) 10 N = 159.6 mN

THUS: $A = -(244 \text{ m N})\dot{c} + (159.6 \text{ m N})\dot{d}$ $\Sigma T = m \bar{a} : A + B = 0, B = (244 \text{ m N})\dot{c} - (159.6 \text{ m N})\dot{d}$

18.154

GIVEN!

RING ATTACHES BY COLLAR AT A TO VERTICAL SHAFT ROTATING AT CONS-TANT RATE W.

FIND:

(a) CONSTANT AND TO B THAT PLANE OF RING FORMS WITH VERTICAL WHEN Q = 12 rod/s, (b) MAX, VALUE OF Q) FOX WHICH (3 = 0.



ANGULAR MOMENTUM HE
USING THE PRINCIPAL AXES GZYZ WITH

X PERPENSIONLAR TO PLANT OF RING:

H = I Wzi + I D j + I W; E

= mEW sinß i + I mE'W cosß i

H = mEW (sinß i + 1 cosß i)

(1)

A A H

EQ. (18.22) AND USING (1):

H

H

G = 0 + ω × H

= ω(sins i + cosps) × miω (sins i + i cosps)

= miω (isins i + cosps) × miω (sins i + i cosps)

= - i miω (isins ω p - sins μω p) k

= - i miω sin β cosps k

Z=29in/3 Z=4cosp EQUATING MOMENTS ABOUT A: *) WZ=(mZw) Z+H.

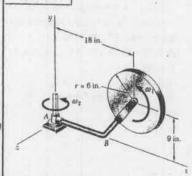
 $mg z \sin \beta = m(z \sin \beta)\omega^{2}(z \cos \beta) + \frac{1}{2}mz^{2}\omega^{3} \sin \beta \cos \beta$ $g = \omega^{2}z \cos \beta + \frac{1}{2}\omega^{3}z \cos \beta$ $\cos \beta = \frac{1}{3}\frac{1}{6}\omega^{3}$ (2)

(a) LETTING g = 32.2 ft/s, &= 0.25 ft, w = 12 rad/s:

 $\cos \beta = \frac{3}{3} \frac{32.2}{(0.25)(12)^2} = 0.59630$ $\beta = 53.4$

(b) SOLVING EQ.(2) FOR W AND LETTING $g = 32.24t/s^2$, e = 0.25 + t, p = 0.1 $\omega^2 = \frac{2g}{32} = \frac{2(32.2)}{3(0.25)} = 85.87$ $\omega = 9.27 \text{ ma/s}$

18.155



GIVEN:

10-16 DISK ROTATES
AT CONSTANT RATE $\omega_1 = 15 \text{ rad/s}.$ ARM ABC ROTATES
AT CONSTANT RATE $\omega_2 = 5 \text{ rad/s}.$ FIND:

FORCE-COUPLESYSTEM
REPRESENTING THE
DYNAMIC REACTION
AT SUPPORT A.

ANGULFR MUNICITUM OF DISK ABOUT C.



USING THE PRINCIPAL CENTROIDAL AXES $Cx'y'e': \qquad \omega = \omega_2 \underline{i} + \omega_1 \underline{k}$ $ANG.VELOCITY OF FRAME Cx'y'z': \qquad \Omega = \omega_2 \underline{i}$ $H_C = \overline{I}_{\omega_2} \underline{i} + \overline{I}_{y,\omega_3} \underline{i} + \overline{I}_{z,\omega_2,\underline{k}}$ $= 0 + \frac{1}{4} m t^z \omega_2 \underline{i} + \frac{1}{2} m t^z \omega_1 \underline{k}$ $H_C = \frac{1}{4} m t^z \left(\omega_2 \underline{i} + 2\omega_1 \underline{k} \right) \qquad (1)$

RATE OF CHANGE OF HC RQ (18.22) AND USING (1):

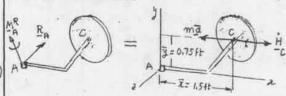
 $\frac{\dot{H}_{c}}{\dot{H}_{c}} = \left(\frac{\dot{H}_{c}}{\dot{L}_{c}}\right)_{cxyz} + \frac{\Omega}{\Omega} \times \frac{\dot{H}_{c}}{\dot{L}_{c}} = 0 + \omega_{2}\dot{g} \times \frac{1}{4}mz^{2}(\omega_{2}\dot{g} + 2\omega_{1}z)$ $\frac{\dot{H}_{c}}{\dot{L}_{c}} = \frac{1}{4}mz^{2}\omega_{1}\omega_{2}\dot{L}$

WITH GIVEN DATA:

 $\frac{H}{c} = \frac{1}{2} \frac{1016}{32.211/s^2} (\frac{1}{2}ft)^{6} (15 rad/s) (5 rad/s) = (2,9115 16.ft) =$

 $\frac{COMPUTATION OF m2}{\bar{a} = -\bar{x} \omega_z^2 i = -(1.5ft)(5 md/s)^2 i = -(37.5 ft/s^2) i$ $m \bar{a} = \frac{10 lb}{32.2 ft/s} (-37.5 ft/s^2) i = -(11.646 lb) i$

ERVATIONS OF MUTION

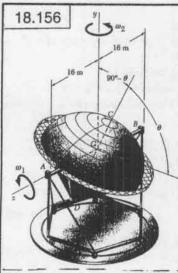


ΣF= ΣFeff! R= ma = -(11.646 /b) i

$$\begin{split} & \sum_{A} = \overline{X} (\underline{M}_{A})_{eff}; \\ & \underline{M}^{R} = \underline{H}_{c} + (\overline{z}_{L}^{\perp} + \overline{y}_{J}^{\perp}) \times m\underline{a} \\ & = (2.911516.ft) \underline{i} + [(1.5ft) \underline{i} + (0.75ft) \underline{j}] \times (-11.64616) \underline{i} \\ & = (2.911516.ft) \underline{i} + (8.73416.ft) \underline{k} \end{split}$$

FORCE-COUPLE SYSTEM AT A:

RA = - (11.65 16) ; MR = (2,91 16.ft) 1+ (8.73 16.ft) k



SOLAR-ENERGY CONCENTRATOR; $m=30\,\mathrm{Mg}$ RADII OF GYRATION ABOUT CD: $\bar{k}=12\,\mathrm{m}$ ABOUT AB: $\bar{k}'=10\,\mathrm{m}$ $\omega_1=0.20\,\mathrm{rad/s}$ (constant) $\omega_2=0.25\,\mathrm{rad/s}$ (constant)

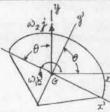
FIND FOR B=60°:

(a) FORCES EXERTED ON CONCENTRATOR AT A AND B,

(b) COUPLE M. L.

APPLIED TO CONCENTRATOR AT THAT

ANGULAR MOMENTUM ABOUT G



USING THE PRINCIPAL AXES Gz'y'z. $\omega_{z} = \omega_{z}\cos\theta$, $\omega_{z} = \omega_{z}\sin\theta$, $\omega_{z} = \omega_{z}$ $H_{G} = I_{z}\omega_{z}z' + I_{z}\omega_{z}z' + I_{z}\omega_{z}t$ $H_{G} = -I'\omega_{z}\cos\theta z' + I\omega_{z}\sin\theta z' + I'\omega_{z}t$ where $I = mk^{2}$ AND $I' = mk^{2}$

WE NOW RETURN TO THE REFERENCE FRAME Gay? ATTACHED TO THE STEEL FRAMEWORK ($\Omega = \omega_2$).

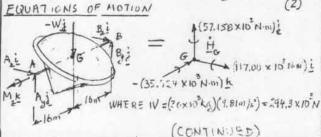
 $H_{6} = -I'\omega_{1}\cos\theta(\underline{i}\sin\theta - \underline{j}\cos\theta) + I\omega_{2}\sin\theta(\cos\theta + \underline{j}\sin\theta) + I'\omega_{1}K$ $H_{6} = (I-I')\omega_{1}\sin\theta\cos\theta\underline{i} + (I'\cos^{2}\theta + I\sin^{2}\theta)\omega_{2}\underline{j} + I'\omega_{1}K$ (1)

RATE OF CHANGE OF H_{6}

WE NOTE THAT WI AND WI ARE CONSTANT, BUT THAT & IS A FUNCTION OF & WITH DERIVATIVE &= WI

$$\begin{split} &E \otimes \{(B,22):\\ &\stackrel{\cdot}{H}_{G} = (\stackrel{\cdot}{H}_{G})_{G2M_{2}^{2}} + \stackrel{\cdot}{\Omega} \times \stackrel{\cdot}{H}_{G} = (I-I')\omega_{2} (\cos^{2}\theta-\sin^{2}\theta) \stackrel{\cdot}{\theta} \stackrel{\cdot}{L} +\\ &+2 (I-I')\omega_{2} \sin\theta\cos\theta \stackrel{\cdot}{\theta} \stackrel{\cdot}{\delta} + \omega_{2} \stackrel{\cdot}{\chi} \times [(I-I')\omega_{2} \sin\theta\cos\theta \stackrel{\cdot}{L} +\\ &+ (I\cos^{2}\theta+I\sin^{2}\theta)\omega_{2} \stackrel{\cdot}{\delta} + I'\omega_{1} \stackrel{\cdot}{K}]\\ &= (I-I')\omega_{1}\omega_{2} (\cos2\theta \stackrel{\cdot}{L} + \sin2\theta \stackrel{\cdot}{\theta}) - \frac{1}{2} (I-I')\omega_{2}^{2} \sin2\theta \stackrel{\cdot}{K} + I^{2}\omega_{1}\omega_{2} \stackrel{\cdot}{L}\\ \stackrel{\cdot}{H}_{G} = [I'+(I-I')\cos2\theta]\omega_{1}\omega_{2} \stackrel{\cdot}{L} + (I-I')\omega_{1}^{2}\omega_{1} \sin2\theta \stackrel{\cdot}{K} + I^{2}\omega_{1}\omega_{2} \stackrel{\cdot}{L}\\ \stackrel{\cdot}{W}_{C} + (I-I')\cos2\theta]\omega_{1}\omega_{2} \stackrel{\cdot}{L} + (I-I')\omega_{1}^{2}\omega_{1} \sin2\theta \stackrel{\cdot}{K} + I^{2}\omega_{1}^{2}\omega_{1} \stackrel{\cdot}{L}\\ \stackrel{\cdot}{W}_{C} + (I-I')\cos2\theta]\omega_{1}\omega_{2} \stackrel{\cdot}{L} + (I-I')\omega_{1}^{2}\omega_{1} \sin2\theta \stackrel{\cdot}{K} - \frac{1}{2} (I-I')\omega_{2}^{2}\sin2\theta \stackrel{\cdot}{K}\\ \stackrel{\cdot}{W}_{C} + (I-I')\omega_{2}^{2}\omega_{1} \stackrel{\cdot}{L}\\ \stackrel{\cdot}{W}_{C} + (I-I')\omega_{1}^{2}\omega_{2} \stackrel{\cdot}{L}\\ \stackrel{\cdot}{W}_{C} + (I-I')\omega_{2}^{2}\omega_{1} \stackrel{\cdot}{W}_{C} \stackrel{\cdot}{W}$$

 $\dot{H}_{G} = (117,00 \times 10^{3} \text{ N·m}) i + (57.158 \times 10^{3} \text{ N·m}) j - (35.724 \times 10^{3} \text{ N·m}) k$



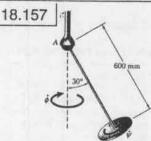
18.156 continued

ΣM_B = Σ(M_B)ess: (32 m) K × (A₁ + A₂;) + (16 m) K × (-29 4.3 × 10 k) s + M₂ k = H_G (32 m) A₂; - (32 m) A₃; + (16 m) (29 4.3 × 10 k); + M₂ k = = (117 × 10 N·m) i + (57. 158 × 10 N·m); - (5.7 24 × 10 N·m) K EQUATING THE COEFF. OF THE UNIT VECTORS:

(1) $-(32 \text{ m})A_3 + (16 \text{ m})(294.3 \times 10^3 \text{ N}) = .117 \times 10^3 \text{ N·m}$ $A_4 = .143.49 \times 10^3 \text{ N}$ (32 m) $A_2 = .57.158 \times 10^3 \text{ N·m}$ $A_3 = .1.7862 \times 10^3 \text{ N}$

K M2=-35,724×103 N·m

(b) COUPLE M, K M, K=-(35.724 x103 N·m) K M K=-(35.7 kil·m) K



GIVEN:

2-kg DISK OF 150-MM

DIAMETER ATTACHED TO ROD

SUPPORTED BY BALL AND

SOCKET AT A.

\$\displaintering 36 rpm AS SHOWN

FIND:

RATE OF SPIN \$\displaintering\$

1=0.6 m 2:0.075m

USING THE FRAME AZYZ (WITH THE

g AXIS POINTING TOWARD US), PAND

NOTING THAT THE PRECESSION K STEADY

EQ. (18 43) YIELDS

\$\(\text{SM} = \Omega \times \text{H}_0
\)

(1)

 $\begin{array}{l} \underline{\Sigma}\underline{M}_{A} = \underline{\Omega} \times \underline{H}_{A} & (1) \\ \text{WHERE} & \underline{H}_{A} = \underline{I}_{A} & \underline{L} + \underline{I}_{A} & \underline{L} \\ \underline{H}_{A} = \underline{I}_{A} & \underline{L} + \underline{I}_{A} & \underline{L} \\ \underline{H}_{A} = \underline{I}_{A} & \underline{L} + \underline{I}_{A} & \underline{L} \\ \end{array}$

THUS: $\sum_{A} = (-4\sin\beta) + 4\cos\beta k \times [1(+4\cos\beta) + 1(+4\cos\beta)k]$ $\sum_{A} = [1(+4\cos\beta) - 1(+\cos\beta) + \sin\beta j$ (2)

BUT ZM = ABX-mgK = -lkx-mg(-sinfi + wspk) =-mg(sinf); EQUATING THE R.H. MEMBERS OF (2) AND (2):

$$\begin{split} \left[I\left(\Psi + \phi\cos\beta\right) - I'\phi\cos\beta\right] & \phi\sin\beta = -mg\ell\sin\beta \\ \left[I\Psi + (I-I')\phi\cos\beta\right] & \phi = -mg\ell \\ & \Psi = I'-I\phi\cos\beta - \frac{mg\ell}{T^2} \end{split} \tag{4}$$

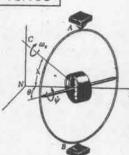
FROM GIVEN DATA: $I = \frac{1}{2}me^{\frac{1}{2}} = \frac{1}{2}(2kg)(0.075m)^{\frac{1}{2}} = 5.625 \times 10^{\frac{1}{2}}$ $I' = m(\ell' + \frac{6}{4}) = 2[(0.6)^{\frac{1}{2}} + \frac{1}{4}(0.075)^{\frac{1}{2}}] = 0.72281$, I' = 127.5 $\dot{\phi} = -36 \text{ rpm} = -1.277 \text{ rad/s}$, $\beta = 30^{\frac{1}{2}}$

EQ.(4): $\sqrt{-(127.5)(-1.27)}\cos 30^{\circ} - \frac{2(9.81)(0.6)}{(5.625 \times 10^{-3})(-1.27)}$ = -416.27 + 555.13 = 138.86 rad/5

¥=1326 rpm

491

18.158



GIVEN:
GYROOM PASS CONSISTING OF ROTOR
SPINNING AT RATEY ABOUT AXIS
MOUNTED IN GIMBAL RUTATING
FREELY ABOUT VERTICAL AB.
B=ANGLE FORMED BY AXIS OF
ROTOR AND MERIDIAN NS.

ROTOR AND MERIDIAN NS.

A = LATITUDE = ANGLE FORMED

BY NS ANDUNE OC PARALLEL TO

GARTH AXIS

ME = ANG. VELOCAT OF EARTH ABOUT ITS AXIS.

SHOW THAT

(a) THE EQUATIONS OF MOTION OF THE GYROCOM PASS ARE

 $I'\ddot{\theta} + I\omega_{\kappa}\omega_{\epsilon}\cos\lambda\sin\theta - I'\omega_{\epsilon}^{2}\cos^{2}\lambda\sin\theta\cos\theta = 0$ $I\omega_{\kappa} = 0$

WHERE WY = RECTANGULAR COMPONENT OF FOTAL ANG VELOCITY

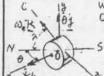
A) ALONG AXIS OF ROTOR

(b) NEGLECTING TERMS IN WE AND FOR SMALL VALUES OF B,

$$\ddot{\theta} + \frac{1\omega_s\omega_e\cos\lambda}{I'}\theta = 0$$

AND THAT AXIS OF RUTOR OSCILLATES ABOUT THE LINE NS.

(A) ANGULAR MOMENTUM ABOUT O.



WE SELECT A FRAME OF REFERENCE DZY & ATTACHED TO THE GIMBAL.

THE ANG. VELOCITY OF OZYZ WITH RESPECT TO A NEWTONIAN FRAME IS $\Omega = \omega_e K + \theta \hat{g}$

WHERE

R = - cost sint i + sin 2 j + cost costk

THUS: $Q = -iQ\cos\lambda\sin\theta i + (\theta + iQ\sin\lambda)j + iQ\cos\lambda\cos\theta k$ (1)

THE ANE. VELOCITY WO OF THE ROTOR IS OBTAINED BY ADIING ITS SPIN I'K TO Q. SETTING I + We COSAWSO = WE HAVE

W=-Wecosasinti+(0+Wesina)j+Wek (2)

THE ANE MOMENTUM HO OF THE ROTOR IS

H = I, w, i + I, w, j + I, w, k

WHERE I, = I, AND I, = I, RECALLING (2) WE WRITE

 $H_0 = -1'\omega_c \cos \lambda \sin \theta \, i + I' (\dot{\theta} + \omega_c \sin \lambda) \, j + I \omega_c \, k \quad (3)$

 $ZM = -I'\omega_{e}\cos \alpha \cos \theta \dot{\theta} \dot{\theta} + I'\dot{\theta}\dot{\theta} + I\dot{\omega}_{e}\dot{\kappa} +$

+ $-\omega_e \cos \lambda \sin \theta$ $\theta + \omega_e^2 \sin \lambda$ $\omega_e \cos \lambda \cos \theta$ (4) -1'\omega \cos \gamma \sin 0 I'(\delta + \omega \sin \lambda) I \omega_2

WE LESERVE THAT THE ROTOR IS FREE TO SPIN ABOUT THE # AXIS AND FREE TO ROTATE ABOUT THEY AXIS. THEREFORE THEY AND # COMPUNENTS OF ZMO MUST BE ZERO. IT FOLLOWS THAT THE COEFFICIENTS OF AND K IN THE R.H. MEMBER OF ER.(4) MUST ALSO BE ZERO.

(CONTINUED)

18.158 continued

SETTING THE COEFF. OF & IN THE R.H. MEMBER OF EQ. (4) EQUAL TO ZERO!

I' # + (-I'We cos A sind) (We cos A cos 8) -

- (- We cospsint) Iw= 0

 $I'\theta' + I\omega_{e} \Omega_{e} \cos \beta \sin \theta - I'\omega_{e}^{e} \cos \beta \sin \theta \cos \theta = 0$ (5)

SETTING THE COFFE OF & EQUAL T ZERO!

 $I\dot{\omega}_2 + (-\omega_e \cos A \sin \theta) I^2(\theta + \omega_e \sin A) - (-I\dot{\omega}_e \cos A \sin \theta) \dot{\theta} + \omega \sin A) = 0$

OBSERVING THAT THE LAST TWO TERMS CANCEL OUT, WE HAVE

 $I\tilde{\omega}_{\lambda} = 0$ (Q.E.D.) (6)

(b) IT POLLOWS FROM EQ. (6) THAT $\omega_{,} = constant \tag{7}$

REWRITE EQ. (5) AS FULLOWS :

 $I'\theta' + (I\omega_x - I'\omega_x \cos \lambda \cos \theta)\omega_x \cos \lambda \sin \theta = 0$

IT IS EVIDENT THAT $\omega_2 >>> \omega_c$. WE CAN THEREFORE NEGLECT THE SECOND TERM IN THE PARTNINESIS AND WRITE I' B' + I $\omega_a \omega_c \cos \lambda \sin \theta = 0$

or $\theta + \frac{I \omega_2 \omega_0 \cos \lambda}{2} \sin \theta = 0$ (B)

WHERE THE COEFFICIENT OF SINB IS A CONSTANT.
THE ROTOR, THEREFORE, OSCILLATES ABOUT THE
LINE NS AS A SIMPLE PENDULUM.

FOR SHALL OSCILLATIONS, SIN 8 \$ 8, AND EQ. (8)

YIELDS $\ddot{\theta} + \frac{I \omega_z \omega_c \cos \theta}{\theta} \theta = 0$ (0.E.D.)(9)

EQ. (9) IS THE EQUATION OF SIMPLE HARMONIC MOTION WITH PERIOD

$$\mathcal{C} = 2\pi \sqrt{\frac{I'}{I\omega_2\omega_c\cos\lambda}} \tag{10}$$

SINCE ITS ROTOR OSCILLATES ABOUT THE LINE NS,
THE BYRD COMPASS CAN BE USED TO DETERMINE
THE DIRECTION OF THAT LINE. WE SHOULD NOTE,
HOWEVER THAT FOR VALUES OF A CLOSE TO 90°0R-90,
THE PERIOD OF OSCILLATION BECOMES VERY LARGE
AND THE LINE ABOUT WHICH THE ROTOR OSCILLATES
CANNOT BE DETECHINED. THE GYRD COMPASS,
THEREFORE, CANNOT BE USED IN THE POLAR
REGIONS.

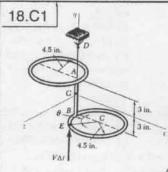


FIGURE SHOWN MADE OF WIRE WEIGHING \$ 02/10 CORD AB. IMPULSE FAt = (0.5 16.5) } 15 APPLIED AT E FIND MIMEDIATELY AFTER IMPACT, FOR VALUES OF & FROM O TO ING IN 10° INCKEMENTS (a) VELOCITY OF G. (b) ANGULAR VELOCITY

ANALYSIS

LET m' = MASS PER UNIT LENGTH 2a = LENGTH OF ROD AB t = RADIVS OF EACH RING

COMPUTATION OF MASSES!

AB:
$$m_{AB} = 2\alpha m'$$
 (1)
EACH RING: $m_{R} = 2\pi \epsilon m'$ (2)

EACH RING:
$$m_R = 2\pi \epsilon m^2$$
 (2)
ENTIRE FIGURE: $m = m_{AB} + 2m_R$ (3)

MOMENTS OF INERTIAL

AB:
$$(I_{\chi})_{AB} = (I_{2})_{AB} = \frac{1}{3} \pi_{AB} a^{2}, (I_{g})_{AB} = 0$$
 (4)

EACH RING:
$$(I_z)_R = \frac{1}{2} m_R z^2 + m_R a^2 = m_R (\frac{1}{2}z^2 + a^2)$$
 (5)
 $(I_y)_R = m_R z^2 + m_R z^2 = 2m_R z^2$ (6)

$$(\vec{1}_{z})_{R} = \frac{1}{2}m_{R}z^{2} + m_{R}(z^{2} + a^{2}) = m_{R}(\frac{3}{2}z^{2} + a^{2})$$
 (7)

ENTIRE FIGURE :

$$I_{2} = (I_{2})_{AB} + 2(I_{2})_{R}, \quad I_{3} = 2(I_{3})_{R}, \quad I_{2} = (I_{2})_{AB} + 2(I_{3})_{R} \quad (B)$$

PRODUCTS OF INERTIAL

THE DALY NON-ZERO PRODUCTS OF INFERTIA ARE (IZY) P

$$I_{xy} = 2(I_{xy})_{R} = -z n_{R} ra$$
 (9)

IMPULCE- MOMENTUM PRINCIPLE:

EQUATING IMPULSE AND MOMENTUM AFTER IMPACT

$$F\Delta t = m\bar{v}$$
: $(F\Delta\theta j = m\bar{v}$
 $\bar{v} = F\Delta t j (FOR ALL VALUES OF \theta) (10)$

EQUATING MORIENT OF IMPULSE ABOUT G AND ANGULAR HOHENTUM HA APTER IMPACT (NOTE THAT THERE IS MU IMPULSIVE PILLE EXCEPT F)

$$H_6 = \left[2(1-\cos\theta) \right] - a \right] + 2\sin\theta \left[\frac{1}{2} \right] \times F\Delta t \right]$$

$$= -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \sin\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t (1-\cos\theta) \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[\frac{1}{2} + 2 F\Delta t \cos\theta \right] \times F\Delta t = -2 F\Delta t \cos\theta \left[$$

THUS: H,=- + FAt sin 0, Hy=0, Hz= + PAt(1-6050) (11) USING EQS. (18.7) AND RECALLING THAT INT THE POS

$$I_{\lambda} \omega_{\lambda} - I_{\lambda \gamma} \omega_{\gamma} = H_{\lambda} \tag{12}$$

$$-I_{yy}\omega_{x}+I_{y}\omega_{y}=0 \qquad \qquad (13)$$

$$I_{\frac{1}{2}}\omega_{\frac{1}{2}}=H_{\frac{1}{2}} \tag{14}$$

(CONTINUED)

18.C1 continued

SOLVING ERS. (12) AND (13) SIMULTANEOUSLY FOR WA AND Wy , AND FO. (14) FOR WZ , WE OBTAIN

$$\omega_{2} = \frac{T_{1} H_{2}}{I_{2} I_{3} - I_{2}^{2}}, \quad \omega_{3} = \frac{I_{23} H_{2}}{I_{2} I_{3} - I_{24}}, \quad \omega_{5} = \frac{H_{6}}{I_{6}}$$
 (15)

OUTLINE OF PROGRAM

ENTER $m' = \frac{[(5/8)/16]1b}{32,2 \text{ ft/s}^4}$, $a = \frac{3}{12} \text{ ft, } t = \frac{45}{12} \text{ ft, } \text{ Fat= 0.5 lb/s}$ COMPUTE MAB, MR, AND M FROM EQS. (1), (2), AND (3)

COMPUTE (Ix)AB AND(IA)AB PROM ENS. (4) COMPUTE $(I_2)_R$, $(I_3)_R$, AND $(I_2)_R$ FROM EQS. (5), (6), AND (7)

COMPUTE I, I, AND I PROM ERS. (8) AND IXY PROMER(9) COFIPUTE TO = FOL/M AND PRINT

FOR \$= 0 TO \$ = 180° AND USING 10° INCREMENTS: CALCULATE HE AND HE FROM POS. (11)

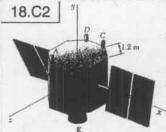
CALCULATE WZ, WY, AND WZ FROM E QS. (15) AND TABULATE

PROGRAM OUTPUT

(b)

(a) Velocity of mass center vbar = 79.07 ft/s (directed upward)

Angular velocity (Omega)x (Omega)y (Omega degrees rad/s rad/s rad/s 0.00 0.00 0.00 0.00 10.00 -54.88 18.29 1.8 20.00 -108.10 36.03 7.1 30.00 -158.03 52.68 15.9 40.00 -203.16 67.72 27.8 50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 76.2 80.00 -311.26 103.75 98.3	8
degrees rad/s rad/s rad/s 0.00 0.00 0.00 0.00 10.00 -54.88 18.29 1.8 20.00 -108.10 36.03 7.1 30.00 -158.03 52.68 15.5 40.00 -203.16 67.72 27.8 50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	8
10.00 -54.88 18.29 1.8 20.00 -108.10 36.03 7.1 30.00 -158.03 52.68 15.5 40.00 -203.16 67.72 27.8 50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	1
10.00 -54.88 18.29 1.8 20.00 -108.10 36.03 7.1 30.00 -158.03 52.68 15.5 40.00 -203.16 67.72 27.8 50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	1
20.00 -108.10 36.03 7.1 30.00 -158.03 52.68 15.9 40.00 -203.16 67.72 27.8 50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	
30.00 -158.03 52.68 15.5 40.00 -203.16 67.72 27.6 50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	
40.00 -203.16 67.72 27.6 50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 76.2 80.00 -311.26 103.75 98.3	4
50.00 -242.12 80.71 42.5 60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	
60.00 -273.72 91.24 59.4 70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	
70.00 -297.00 99.00 78.2 80.00 -311.26 103.75 98.3	
80.00 -311.26 103.75 98.3	T/2
90.00 -316.06 105.35 118.9	750
100.00 -311.26 103.75 139.6	
110.00 -297.00 99.00 159.6	8
120.00 -273.72 91.24 178.4	
130.00 -242.12 80.71 195.4	
140.00 -203.16 67.72 210.1	
150.00 -158.03 52.68 222.0	3
160.00 -108.10 36.03 230.8	
170.00 -54.88 18.29 236,1	
180.00 0.00 -0.00 237.9	



PROBE WITH m = 2500 kg. k= 0.98 m, kg=1.06 m, k=1.02 m. 500-N MAIN THRUSTER E; 20-N THRUSTERS A, B, C, D CAN EXPEL FUEL IN & DIRECTION. PROBE HAS AND VELOCITY W=Wz L+Wz K

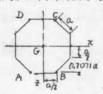
FIND WHICH TWO OF THE 20-N THRUSTERS SHOULD BELSED TO REDUCE ANG, VELOCITY TO ZERO AND FOR HOW LUNG EACH OF THEM SHOULD BE ACTIVATED, ASSUMING (a) \(\omega = (0.040 \text{ rad/s}) i + (0.060 \text{ rad/s}) k, As IN PROB. 18.33, (b) w = (0.060 rad/s)i - (0.040 rad/s)k, AS IN PROB. 18.34, (c) w = (0,060 rad/s) i + (0.020 rad/s) &,

(d) w = - (0.060 rad/s) i - (0.020 rad/s) k.

ANALY515

INITIAL ANG, MOMENTUM:

HG=Iwi+0+Iwk=mkwi+mkwk (1) THUS H = m K = W Hy = 0 Hz=mkzwz ANGULAR IMPULSE OF TWO 20-N THRUSTERS:



LET US ASSUME THAT A AND B ARE ACTIVATED.

ANE, IMPULSE ABOUT G = 2A × (-FDtA) 3 + 28 × (-FDtB) 2 = (-0,5ai+1,2071ak)x(-Fot)j+ . + (0.5a1+1.2011ak) x (-FOta) à

ANE. IMP. = 1.2071 a F (DtA+ DtB) i + 0.52 F (DtA- DtB) K IMPULSE-MOMENTUM PRINCIPLE

WE MUST HAVE HG + ANE. IMP. = 0

OR, USING COMPONENTS: H, +1.2071aF(Ata+Ata)=0 H2 + 05 a F (1 tA - D tB) = 0

1,2071aF Ata-Ata = - D. Saf

SOLVING THESE EQUATIONS SIMULTANGOUSLY.

DtA = - He+0.41421 Hx, OtB = H2-0.41421 Hz

IF DtA > 0, ASSUMPTION IS CORPECT, A SHOULD BE USED; IF At <0, ASSIMITION IS WRONE; C SHOULD SE USEL AND ACTIVATED FIR Dtc = | Ltal.

SIMILARLY, IF At > 0, B SHOULD BE USED, AND IF ATBCO D SHOULD BE USED WITH AT = | Atal.

OUTLINE OF PROGRAM

ENTER PART: a, b, c, or d

ENTER m = 2500 kg, k = 0.98 m, kz = 1,02 m

ENTER a = 1,2 m, F = 20 N

ENTER VALUES OF WX AND W2

COMPUTE HX AND H, FIOM EOS, (1)

COMPUTE At AND At FROM EQS. (3)

IF DtA > 0, PRINT DtA; IF NOT, PRINT Dt= | DtA |

IF Dt , PRINT Dt, IF NOT, PRINT Dt = | Dt |

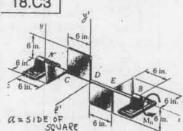
PROGRAM OUTPUT

(a) CANDB; At = 8,1605; At = 4,8455

(6) A AND D; DtA=1.8495; DtD=6.8215 (C) C AND D; At = 4.6543; At = 0.31883

(d) A AND B; Dta = 4.6545; Dta = 0.31885





GNEN:

A COUPLE M = (0.03 lb+ff)i 15 APPLIED ATT = OTO 2.7 - 16 ASSEMBLY OF SHEET ALUMINUM OF UNIFORM THICKNESS

FIND:

(a) COMPONENTS ALONG THE ROTATING & ANDE AXES OF THE DYNAMIC REAC-

TIONS ATA AND & FROM t=0 TO t= 2s AT O.13 INTERVALS, (b) THE TIME (WITH 3 SIGNIFICANT FIGURES) AT WHICH THE Z COMPONENTS OF THESE REACTIONS ARE EDUAL TO ZERO.

ANALYS 15

WE COMPUTE THE MOMENT AND PRODUCTS OF INERTIA OF THE ASSEMBLY WITH RESPECT TO THE CENTROIDAL AXES DZX'Z'. WE FIRST COMPUTE THE MUMENT AND PRODUCTS OF AREAS FOR FACH SOUARE : (Ix) AREA = 1 a, (Iz), AFFA + 4", (Iz) AREA = 0

FOR EACH TRIANGLE: (I) ATEA = 12 04, [IN) AREA = 0

 $(I_{\chi_{\overline{\lambda}}})_{\mu\rho\pi\lambda} = \frac{1}{2}a^2\overline{\lambda}'\overline{\xi}' + \overline{I}_{\chi''\mu\nu} = -\frac{1}{2}a^4(\frac{1}{3}a)(\frac{1}{3}a) + \frac{1}{72}a^4 = -\frac{15}{72}a^4$ [CF. 5P9.6] FOR ENTIRE ASSEMBLY

(1) AREA = 2(1/3 a")+2(1/2 a") = 5 a" $(I_{xy})_{AREA} = 2(-\frac{1}{4}a^4) = -\frac{1}{2}a^4$ $(I_{xy})_{AREA} = 2(-\frac{15}{72}a^4) = -\frac{15}{76}a^4$

THE MASS MORIENT AND PRODUCTS OF INTERTIA ARE OBTAINED BY MULTIPLYING THESE EXPRESSIONS BY THE HASS M OF THE ASSEMBLY AND DIVIDING BY ITS AREA, WHICH IS EQUAL TO 32:

In = - 1 mat, In = - 5 mat (1) I = 5 ma,

WE DETERMINE HO AND ITS DERIVATIVE HO SETTING W_= W, Wy = W2 = O IN EDS. (18,7), JE HAVE

H2=1,00, Hx=-Ixy,00, Hz=-Izz,0

H=(Ii-In, 1-In, k)w EB (18.22): H = (HD) DZY'E+ + AX HD

H = (I, i - I, y, j - I, p, k) is + wix (I, i - I, y d - I, r, k) w =(I, i - I, i - I, x) x - I, w + I, w +

H = I, a i + (I, a, a) - I, a, a) j - (I, w+ I, a) k EQUATIONS OF MOTION



IMB = Z[Mpless: Moi - 4ai x (A, i+A, t) = Ho Moi-4anyk+4an=i= Ix i+(I, 0-1,x)i-(Ixw+Ix)E EQUATING THE COSFF. OF THE ONIT VECTORS!

Q = Mo/I, 1 M = I, a (2) FROM WHICH WE OBTAIN w=at (3)

(4)

(k) Ay = (Izy, w+ I a)/4a (5) Σ F = Σ(F) = 0: A+B=0

(6) THUS: By = - Ay By = - Az

(CONTINUED)

18.C3 continued

OUTLINE OF PROGRAM

(a) ENTER Mo= 0.03 lb.ft, W= 2,7 lb, a= 0.5 ft COMPUTE m = W/32.2

COMPUTE Ix, Ixy, , Ixe, FROM EQS. (1)

COMPUTE OF FROM EQ. (2)

FOR t = 0 TO t = 2s AT O.1-s INTERVALS:

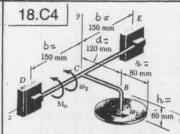
COMPUTE W FROM EQ. (3)

COMPUTE Ay, As, By, B, FROM EQS. (4), (5), AND (6)
AND THEOLATE VS &

(b) DETERMINE BY INSPECTION THE TIME INTERVAL IN WHICH AZ AND BZ CHAPISE SIGN AND RUN THE PROGRAM OVERTHAT INTERVAL, USING 0.01-5 INCREMENTS. REPROT THIS PROCEDURE, USING 0.001-5 INCREMENTS. THE DESIRED VALUE OF \$\frac{t}{2}\$ THAT FOR WHICH |Az| AND |Bz| ARE SMALLEST.

PROGRAM OUTPUT

(a)	t	Ay 1b	A2 1b	By 1b	Bz 1b
	0.00000	-0.00750	0.00900	0.00750	-0.00900
	0.10000	-0.00796	0.00861	0.00796	-0.00861
	0.20000	-0.00935	0.00745	0.00935	-0.00745
	0.30000	-0.01167	0.00552	0.01167	-0.00552
	0.40000	-0.01492	0.00282	0.01492	-0.00282
-	0.50000	-0.01909	-0.00066	0.01909	0.00066
	0.60000	-0.02419	-0.00491	0.02419	0.00491
	0.80000	-0.03022	-0.00993	0.03022	0.00993
	0.90000	-0.04506	-0.01573	0.03718	0.01573
	1.00000	-0.05387	-0.02964	0.05387	0.02230
	1.10000	-0.06361	-0.03775	0.06361	0.03775
	1,20000	-0.07427	-0.04664	0.07427	0.04664
	1.30000	-0.08586	-0.05630	0.08586	0.05630
	1.40000	-0.09838	-0.06673	0.09838	0.06673
Wille-	1.50000	-0.11183	-0.07794	0.11183	0.07794
Gran.	1.60000	-0.12620	-0.08992	0.12620	0.08992
	1.70000	-0.14150	-0.10267	0.14150	0.10267
	1.80000	-0.15773	-0.11619	0.15773	0.11619
	1.90000	-0,17489	-0.13049	0.17489	0.13049
	2.00000	-0.19297	-0.14556	0.19297	0.14556
(b)	t	Ay	Az	Ву	Bz
(2)	8	1b	1b	1b	1b
	0.40000	-0.01492	0.00282	0.01492	-0.00282
	0.41000	-0.01529	0.00250	0.01529	-0.00250
	0.42000	-0.01568	0.00218	0.01568	-0.00218
	0.43000	-0.01607	0.00186	0.01607	-0.00186
	0.44000	-0.01648	0.00152	0.01648	-0.00152
	0.45000	-0.01689	0.00118	0.01689	-0.00118
	0.46000	-0.01731	0.00082	0.01731	-0.00082
100	0.47000	-0.01774	0.00046	0.01774	-0.00046
	0.48000	-0.01818	0.00010	0.01818	-0.00010
	0.49000	-0.01863	-0.00028	0.01863	0.00028
	0.50000	-0.01909	-0.00066	0.01909	0.00066
	t	Ay 1b	Az	Ву	Bz
	.00	1ь	1b	16	1b
	0.48000	-0.01818	0.00010	0.01818	-0.00010
	0.48100	-0.01823	0.00006	0.01823	-0.00006
F	0.48200	-0.01827	0.00002	0.01827	-0.00002
MAKE	0.48300	-0.01832	-0.00001	0.01832	0.00001
	0.48400	-0.01836	-0.00005	0.01836	0.00005
	0.48500	-0.01841	-0.00009	0.01841	0.00009
	0.48600	-0.01845	-0.00013	0.01845	0.00013
	0.48700	-0.01850	-0.00016	0.01850	0.00016
	0.48900	-0.01854	-0.00020	0.01854	0.00020
	0.49000	-0.01859	-0.00024	0.01859	0.00024
	0.45000	0.01003	-0.00028	0.01863	0.00028



GIVEN: DISK: m=2.5 kg, k=80 mm $\omega_1=60$ rad/s at t=0 and DECREASES AT RATE OF 15 rad/s. At t=0, $\omega_2=0$ AND COUPLE $M=(0.5N \cdot m)k$ 15 APPLIED TO SHAFT DCE. FIND: (a) COMPONENTS ALONG-

THE ROTATING 2 AND 3 AXES OF THE DYNAMIC REACTIONS AT D AND E FROM t=0 TO t=4s AT 0.2-S IN TERVALS, (b) THE TIMES t_1 AND t_2 (WITH 3 SIGNIFICANT FIGURES) AT WHICH E_2 AND E_4 ARE RESPECTIVELY ENUAL TO ZERO.

$$\begin{split} &\frac{A NALYSI5}{H_A = \overline{I}_y \omega_j + \overline{I}_z \omega_z k} = \frac{1}{2} m_z^2 \omega_l + \frac{1}{4} m_z^2 \omega_z k \\ &EQ. (18,22); \quad \dot{H}_A = (\dot{H})_{2yz} + \underline{\Omega} \times \dot{H}_A \\ &\dot{H}_A = \frac{1}{2} m_z^2 \omega_l + \frac{1}{4} m_z^2 \omega_z k + \omega_z k \times (\frac{1}{2} : n_z^2 \omega_l + \frac{1}{4} m_z^2 \omega_z k) \\ &= \frac{1}{2} m_z^2 \alpha_l + \frac{1}{4} m_z^2 \alpha_z k - \frac{1}{2} m_z^2 \omega_l \omega_z i \\ &\dot{H}_A = \frac{1}{2} m_z^2 \alpha_l + \frac{1}{4} m_z^2 \alpha_z k - \frac{1}{2} m_z^2 \omega_l \omega_z i \\ &\dot{H}_A = \frac{1}{2} m_z^2 (-\omega_l \omega_z L + \alpha_l + \alpha_l + 0.5 \alpha_z k) \end{split}$$
(1)

 $m\bar{a} = m(\alpha_2 \times \alpha_{Nc} - \omega_1^2 \alpha_{Nc}) = m\alpha_1 k \times (d_1 - h_2^2) - m\omega_1^2(d_1 - h_2^2)$ $= \#(d\alpha_1 j + h\alpha_2 l - d\omega_2^2 l + h\omega_2^2 j)$ $m\bar{a} = m(h\alpha_1 - d\omega_2^2) l + m(d\alpha_1 + h\omega_2^2) j \qquad (2)$

FOURTION: OF MOTION $D_{1} = \frac{1}{2b} = \frac{1}{2mz'\alpha_{1}} = \frac{1}{2$

$$\begin{split} & \sum \underline{M}_{D} = \underline{Z}[\underline{M}_{D}]_{eff}; \\ & - 2b \, k \, \times \left(\underline{E}_{\underline{i}} + \underline{E}_{\underline{j}} \underline{i}\right) + \underline{M}_{0} \, \underline{k} = -\frac{1}{2} m \epsilon^{2} \omega_{i} \, \omega_{j} \, \underline{i} + \frac{1}{2} m \epsilon^{2} \alpha_{i} \underline{j} + \frac{1}{4} m \epsilon^{2} \alpha_{i} \underline{k} + \frac{1}{4}$$

 $-2bE_{2}j+2bE_{3}i+M_{b}k=-\frac{1}{2}m\xi^{2}\omega_{1}\omega_{2}i+\frac{1}{2}m\xi^{2}\omega_{3}j+\frac{1}{4}m\xi^{2}\alpha_{2}k=\\ -mb(h\alpha_{2}-d\omega_{2}^{2})j+mb(d\alpha_{2}+h\omega_{2}^{2})i+md(d\alpha_{2}+h\omega_{2}^{2})k+mh(h\alpha_{2}-d\omega_{2}^{2})k$

EQUATE THE COEFF. OF THE UNIT VECTORS: $M_0 = m(\frac{1}{4}t^2 + d^2 + h^2)\alpha, \qquad \alpha_2 = \frac{M_0}{m(\frac{1}{4}t^2 + d^2 + h^2)}$ (3)

 $\tilde{L} = \frac{m}{2b} \left(-\frac{1}{2} t^2 \omega_1 \omega_2 + b d \omega_2 + b h \omega_2^2 \right)$ (5)

 $\Sigma F = Z(F)_{eff}: \quad \mathcal{D} + E = m\bar{a}$ $\mathcal{D}_{i} + \mathcal{D}_{j} + E_{i} + E_{j} = m(h\alpha_{i} - d\omega_{i}^{*})_{i} + m(d\alpha_{j} + h\omega_{i}^{*})_{j}$ $\Xi VICTORS$

WE RECALL FROM THE GIVEN DATA THAT

m = 2.5 kg, t = 0.08 m, b = 0.15 m, d = 0.12 m, h = 0.06 m

M₀ = 0.5 N·m ω₀ = 60 rad/s α₁ = -15 rad/s*

(8)

NOD NOTE THAT AT TIME t $\omega_2 = \alpha_e t$ (9)

(CONTINUED)

18.C4 continued

OUTLINE OF PROGRAM

- (a) ENTER DATA SHOWN IN (8) ON PREVIOUS PAGE COMPUTE 02 FROM EW. (3) FOR t = 0 TO t = 4 s AT 0.Z-s INTERVALS COMPUTE W, AND W, FROM EQS. (9) COMPUTE Ex AND Eg FROM EQS. (4) AND (5) COMPUTE DE AND DE FROM EQS. (6) AND 67) AND TABULATE VS t.
- (b) TO FIND THE TIME t, AT WHICH Ex = 0, DETERMINE BY INSPECTION THE TIME INTERVAL IN WHICH E, CHANGES SION AND RUN THE PROGRAM OVER THAT INTERVAL, USIN & 0,01-5 INCREMENTS, REPEAT THIS PROCEDURE USING SELECT FOR t, THE TIME 0.001-5 INCREMENTS AT WHICH Ex 15 SMALLEST. A SIMILAR PROCEDURE IS USED TO DETERMINE

THE TIME to AT WHICH EX = 0.

PROGRAM OUTPUT

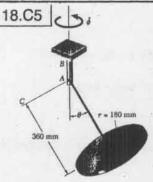
(a)	(a)	Dx (N)	Dy (N)	Ex (N)	Ey (N)
	0.0000	0.3653	1.5306	1.1653	1.5306
	0.2000	-0.2594	4.9450	0.5406	-1.2591
	0,4000	-2.1337	8.6576	-1.3337	-3.0975
	0.6000	-5.2574	12.6685	-4.4574	-3.9846
	0.8000	-9.6305	16.9775	-8.8305	-3.9204
	1.0000	-15.2532	21.5848	-14.4532	-2.9050
	1.2000	-22.1253	26.4902	-21,3253	-0.9384
	1.4000	-30.2469	31.6939	-29.4469	1.9796
	1.6000	-39.6181	37.1958	-38.8180	5.8488
	1.8000	-50,2386	42.9958	-49.4386	10.6693
	2,0000	-62.1087	49.0941	-61.3087	16.4411
	2.2000	-75.2282	55.4906	-74.4282	23.1641
	2.4000	-89.5972	62,1854	-88.7973	30.8384
	2.6000	-105.2158	69.1783	-104.4158	39.4640
	2.8000	-122.0838	76.4694	-121.2837	49.0409
	3.0000	-140.2012	84.0588	-139,4012	59.5690
	3.2000	-159.5681	91.9463	-158.7681	71.0483
	3.4000	-180.1846	100.1321	-179.3845	83.4790
	3.6000	-202.0505	108.6160	-201.2504	96.8609
	3.8000	-225.1659	117.3982	-224.3658	111.1942
	4.0000	-249.5307	126.4786	-248.7306	126.4786

(b) LAST STEP IN DETERMINATION OF t,

(s)	Dx (N)	Dy (N)	Ex (N)	Ey
0.2700	-0.7733	6.2105	0.0267	(N) -2.0107
0.2710	-0.7817 -0.7902	6.2289	0.0183	-2.0206
0.2730	-0.7987	6.2472	0.0098 0.0013	-2.0305 -2.0403
0.2750	-0.8073 -0.8158	6.2839	-0.0073 -0.0158	-2.0501 -2.0599
0.2760	-0.8244 -0.8331	6.3207	-0.0244	-2.0697 -2.0795
0.2780	-0.8418 -0.8505	6.3575	-0.0418 -0.0505	-2.0892 -2.0989
0.2800	-0.8592	6.3943	-0.0592	-2.1086

LAST STEP IN DETERMINATION OF EZ

	t	Dx	Dy	Ex	Ey
	(s)	(N)	(N)	(N)	(N)
	1.2700	-24.8258	28.2776	-24.0258	-0.0253
-	1.2710	-24.8654	28.3034	-24.0655	-0.0114
	1.2720	-24.9052	28.3292	-24.1052	0.0025
	1.2730	-24.9449	28.3550	-24.1449	0.0165
	1.2740	-24.9847	28.3808	-24.1847	0.0304
	1.2750	-25.0245	28.4066	-24.2245	0.0444
	1.2760	-25.0644	28.4325	-24.2644	0.0584
	1.2770	-25.1042	28.4583	-24.3042	0.0724
	1.2780	-25.1442	28,4842	-24.3441	0.0865
	1.2790	-25.1841	28,5100	-24.3841	0.1006

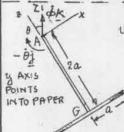


GIVEN:

DISK WELDED TO ROD A 6 OF NEGLIGIBLE MASS CONVECTED BY CLEVIS TO SHAFT AB. ROD-AND DISK PREE TORUTATE ABOUT AC; SHAPT AB PREE TO ROTATE ABOUT VERTICAL AXIS. MITIALLY, $\theta = \theta_0$, $\dot{\theta} = 0$, $\phi = \dot{\phi}$ EIND:

(a) MINIMUM VALUE On OF B DURING ENSUING MOTION AND TIME REQUIRED FOR 8 TO RETURN TO DO (PERIOD) -

(b) ANG VEL & FOR VALUES OF & FROM & TOOM USING I INCREMENTS CONSIDER SUCCESSIVELY THE INITIAL CONDITIONS (i) 0=90, 4=5rad/s, (ii) 0=90, 4=10 rad/s, (iii) 0=60,4=5 rad/s.



ANA LYSIS USING THE ROTATING PRAMEAXY ES W= +sin Di-Dj++cosk $H = I \omega_2 = I' \phi \sin \theta$ H = I, D, = I & cos o

(1) WHERE I = + ma (2) I'= 1 ma+m(2a)= 17 ma

CONSERVATION OF ANG. MOM. ABOUT Z SINCE THE PORCES CONSIST OF REACTION AT A AMD WEIGHT W = - mg K AT G, WE HAVE ZM = O AND H = CONSTANT SINCE Hy= H_sin0+H_cost = I'+ sin'0+I+cost, WE HAVE (I'sin'0+Ias'0) + = (I'sin'0,+Ias'0,) +

Q = I'sin't + I cos't (34 SETTING (4) AND Qo = I'Sin'to + I cos'to (5)

AND SOLVING FOR +: +=(Q,/Q)+

CONSERVATION OF ENERGY

T+V= E=CONSTANT: - (1,0)+ I, 0)+ Ie 0,+ W(-1a cost)=E \$ (I' + sin'0 + I' + I + Cas'0) - 2 mg a cos0 = E $(I'sin'\theta + Icos'\theta) + I'\theta' - 4mgacos\theta = 2E$

RECALLING (3) AND SUBSTITUTING FOR & FROM (5): (Q + 10/Q) + 1' + - 4mg a cost = 2E (6)

SOLVING FOR A: 62 = 1 (2 E + 4 mg a coso - Q0 4 WHICH IS OF THE FORM $\hat{\theta}^2 = f(\theta)$ (7) (8) WHERE $f(\theta) = \frac{1}{T}(2E + 4mg a \cos\theta - \frac{90}{10})$

AND Q IS THE FUNCTION OF B DEFINED IN (3). THE CONSTANT 2E IS OBTAINED AT MAKING B = 0, 0 = 0, MD Q=Q, IN EO. (6): E= 2 & 8 -2mg a cos 0, PROM (7) WE WRITE

(10) $d\theta = \dot{\theta} = \sqrt{3(\theta)}$

(a) THE TIME & NE DED POR & TO DECREASE TO DO LS OBTAINED THROUGH NUMERICAL INTEGRATION, On BEING DEFINED BY THE PACT THAT $f(\theta_n)=0$ (f CHANGES SIGN) (b) FOR EACH DESIRED VALUE OF B, COMPUTE Q PORMER(3) AND & FROM Ea. (5). (CONTINUED)

18.C5 continued

OUTLINE OF PROGRAM

ENTER 2:0.18 m, g = 9.81 m/s*. ASSUME m = 1.

ENTER INITIAL CONDITIONS: 80 AND PO
ENTER DECREMENT AB YOU WISH TO USE

COMPUTE I AND I' PROM (1) AND (2)

COMPUTE Q FROM (4) AND E PROM (9)

FOR 8=8, TO 8=8m (WHEN \$(8) CHANGES SIGN), AND

USING DECREMENTS AB:

COMPUTE Q FROM (3)

COMPUTE \$(0) FROM (8)

CARRY OUT NUMERICALLY THE INTEGEN HOW

INDICATED IN (10)

AT 2°INTERVALS, COMPUTE \$PROM (5) AND PRINT

THE VALUES OF B AND \$

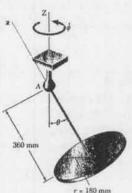
THE PERIOD OF THE OSCILLATION IN 8 15 OFTAINED BY DOUBLING THE VALUE OF & WHEN & REACHES ITS MINIMUM VALUE On.

PROGRAM SUTPUT

111

(1)		(ii)			
TH0= 90 PHID0= 5 DTH= .1		THO= 90 PHIDO= 10 DTH= .1			
Theta	Precession	Theta	Precession		
(degrees)	(rad/s)	(degrees)	(rad/s)		
90.000 88.000 86.000 84.000 82.000 80.000 76.000 74.000 72.000 66.000 64.000 62.000 56.000 56.000 56.000 56.000 56.000 56.000 58.000 56.000	5.000 5.005 5.022 5.087 5.137 5.137 5.138 5.272 5.359 5.460 5.575 5.707 5.855 6.021 6.207 6.415 6.647 6.905 7.193 7.513 7.705 13.683 13.704 20.04	(degrees) 90.000 88.000 88.000 84.000 82.000 80.000 78.000 76.000 76.000 66.000 64.000 62.052 Theta min = Period = 0.5 THO= 60 PHID= 5 DTH= .1 Theta (degrees) 60.000 58.000 54.000 52.000 58.000 56.000 54.000 52.000 50.000 54.000 52.000 50.000 54.000 52.000 50.000 54.000 52.000 50.000 54.000 52.000 50.000 54.000 52.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000 50.000	10.000 10.011 10.043 10.097 10.174 10.273 10.397 10.545 10.719 10.920 11.151 11.413 11.709 12.042 12.404 62.1 degrees		
		40.000 38.000 36.894	8.082		
			8.945 6.9 degrees		

18.C6



GIVEN:

NEGLISTIES MASS SUPPORTED BY

BALL AND SOCKET AT A.

INITIALLY, $\theta = \theta_0$, $\dot{\theta} = 0$, $\dot{\psi} = \dot{\psi}_0$,

AND $\dot{\psi} = \dot{\phi}_0$.

FIND:

(a) MITHIT IN VILLE θ_m OF θ IN

ENSUING MOTION AND PERIOD (TIME

REQUIRED FOR θ TO RETURN TO θ_0).

(b) $\dot{\psi}$ AND $\dot{\psi}$ FOR VALUES OF θ FROM θ_0 TO θ_m USING 2^n INCREMENTS.

CONSIDER SUCCESSIVELY THE INTIAL

(i) 00= 90°, V= 50 rad/s, += 0

DICK WELDED TO ROD HE OF

(iii) $\theta_0 = 90^\circ$, $\psi_0 = 50 \text{ rad/s}$, $\psi_0 = 50^\circ$, $\psi_0 = 5 \text{ rad/s}$ (iv) $\theta_0 = 90^\circ$, $\psi_0 = 80 \text{ rad/s}$, $\psi_0 = 5 \text{ rad/s}$ (v) $\theta_0 = 90^\circ$, $\psi_0 = 80 \text{ rad/s}$, $\psi_0 = 5 \text{ rad/s}$

CONDITIONS

(iv) $\theta_0 = 90^\circ$, $\psi_0 = 10 \text{ rad/s}$, $\phi_0 = 5 \text{ rad/s}$ (v) $\theta_0 = 60^\circ$, $\psi_0 = 0$, $\phi_0 = 5 \text{ rad/s}$ (vi) $\theta_0 = 60^\circ$, $\psi_0 = 50 \text{ rad/s}$, $\phi_0 = 5 \text{ rad/s}$

ANALYSIS

F ZI OK

- OJ CA

2a

VSING THE ROTATING PRAISE

Azyz WITH Y AXIS RSINTING

INTO THE PAPER: $\omega = \dot{\phi} \sin \theta \, \dot{\nu} - \dot{\theta} \dot{j} + (\dot{Y} + \dot{\phi} \cos \theta) \dot{k}$ $H_a = I_z \omega_z \dot{i} + I_y \omega_y \, \dot{q} + I_z \omega_z \, \dot{k}$ $= I^{\dot{\phi}} \sin \theta \, \dot{i} - I^{\dot{\phi}} \dot{\theta} \dot{j} + I^{\dot{\psi}} \dot{\psi} \dot{\phi} \cos \theta) \dot{k}$ WHERE $I = \frac{1}{2} m a^{\dot{\phi}}$ (1) $I^{\dot{\phi}} = \frac{17}{4} m a^{\dot{\phi}} + m(2a)^{\dot{\phi}} = \frac{17}{4} m a^{\dot{\phi}}$ (2)

CONSERVATION OF ANBULAR MOMENTUM
SINCE THE ONLY EXTERNING FORCES ARE THE REACTION

HT A AND THE WEIGHT W = -TUPK ATG, WE HAVE $ZM_Z = 0$ AND $ZM_z = 0$. SINCE Z IS PART OF A NEWTONIAN FRAME OF REFERENCE, IT FOLLOWS THAT $H_Z = CONSTAINT$. RECAUSE OF THE AXISYMMETRY OF THE DISK, IT ALSO FOLLOWS THAT $H_Z = CONSTAINT$ (SEE PROB. 18, 139). WE WRITE

 $H_z = const$: $I(\mathring{V} + \mathring{\varphi} cos\theta) = \beta$ (3) WHERE FROM INIT. COND.: $\beta = I(\mathring{V}_0 + \mathring{\varphi} cos\theta_0)$ (4) $H_z = const$: $H_z sin\theta + H_z cos\theta = \alpha$

 $I'\dot{\phi}\sin^2\theta + I(\dot{\Psi} + \phi \omega \cdot \theta) cD \cdot \theta = \omega$ $RECALLING (3) WE HAVE I'\dot{\phi}\sin^3\theta + \beta \cos\theta = \omega \qquad (5)$ $PROH INITIAL CONDITIONS: <math>\alpha = I'\dot{\phi}\sin^3\theta + \beta \cos\theta \qquad (6)$ $SOLVING (5) FOR \dot{\phi}: \dot{\phi} = \frac{\alpha - \beta \cos\theta}{I'\sin^3\theta} \qquad (7)$ $CONSERVATION OF ENERGY I's in^2\theta$

 $T = \frac{1}{2} \left(I_x \omega_x^2 + I_y \omega_y + \frac{1}{2} \omega_y^2 \right)$ $= \frac{1}{2} \left[I^3 \dot{\Phi}^2 \sin^2 \theta + I^3 \dot{\theta}^2 + I \left(\dot{\Psi} + \dot{\Phi} \cos \theta \right)^2 \right]$ SUBSTITUTE FOR () FROM (3): $T = \frac{1}{2} \left(I^3 \dot{\Phi}^2 \sin^3 \theta + I^3 \dot{\theta}^3 + \frac{A^2}{I} \right) \qquad V = -mg(2a) \cos \theta - E$ $T + V = E: \frac{1}{2} \left(I^3 \dot{\Phi}^2 \sin^3 \theta + I^3 \dot{\theta}^3 + \frac{A^3}{I} \right) - 2mg \cos \theta = E$ $FROM INIT (MID): E = \frac{1}{2} \left(I^3 \dot{\Phi}_0 \sin^3 \theta + \frac{A^3}{I} \right) - 2mg a \cos \theta_0 \qquad (9)$ $SOLVING (B) FOR \dot{\theta}^2: \dot{\theta}^2 = f(0)$

 $f(\theta) = \frac{1}{I} (2E - \frac{A^2}{I} + 4mg a \cos \theta) - \phi \sin^2 \theta$ (CONTINUED)

(11)

18.C6 continued

SUBSTITUTING FOR & FROM (7) INTO (11) WE HAVE $f(\theta) = \frac{1}{I}, (2E - \frac{\beta^2}{I} + 4 \operatorname{mg} \alpha \cos \theta) - \left(\frac{\alpha - \beta \cos \theta}{I' \sin \theta}\right)^2$ (12)

FROM 20. (10) WE WRITE

$$\frac{d\theta}{dt} = \dot{\theta} = \sqrt{5(6)} \qquad \dot{t} = \int_{\theta_0}^{\theta} \frac{d\theta}{\sqrt{5(\theta)}} \tag{13}$$

(A) THE TIME 1 & NEEDED POR & TO DECREASE TO 8m IS OBTHINED THROUGH NUMERICAL INTEGRATION, BON BEING DEPINED BY THE PART THAT \$ (Bin) = 0, THAT IS, THAT f(0) CHATTEES SIGN FOR B = 0m (b) FOR EACH DESIDED VALUE OF B, COMPUTE \$ PROM EU. (7).

OUTLINE OF PROGRAM

ENTER a = 0.18m, g = 9.81 m/s. ASSUME m = 1. ENTER INITIAL CONDITIONS: 80, VO, AND \$0. COMPUTE I AND I' FROM (1) AND (2) COMPUTE (3 FROM (4), O FROM (6), AND E FROM (4) FOR H = HO TO 0 = Om (WHEN 5(0) CHANGES SIGN). AND USING DECREMENTS do :

COMPUTE & FROM (7) COMPUTE & (0) FROM (11)

CARRY OUT NUMERICALLY THE INTEGRATION DEFINED

AT 2º INTERVALS, PRINT THE VALVES OF B, +. AND, FROM (3), OF \$ = \$ - \$ cos +

THE PERIOD OF THE OSCILLATION IN A 15 DETAINED BY DOUBLING THE WHENE OF & CORRESPONDING TO D = 0m .

PROGRAM OUTPUT

(i)	(ii)

,			,		
TH0=90 PS DTH=0.10	ID0=50	PHIDO= 0	TH0=90 PS DTH=0.10	IDO= 0	PHIDO= 5
	Spin	Precess.	Theta	Colo	Precess.
degrees			degrees		
	20012	2011/2012	acartica	rau, o	LOW/ D
90.00	0.00		90.00	5.00	0.00
88.00	-0.21	50.01	88.00		
86.00	-0.41	50.03			-0.35
84.00	-0.62	50.06			-0.53
82.00	-0.83	50.12			-0.71
80.00	-1.05	50.18	80.00		-0.90
78.00	-1.28	50.27			-1.09
76.00		50.37	76 00	5 21	-1 20
		50.48	74.00 72.00 70.00	5.41	-1.49
72.00		50.62	72.00	5.53	-1.71
70.00	-2.28	50.78	70.00	5.66	-1.94
68.00	-2.56	50.96	68.00	5.82	-2.18
66.00	-2.87	51.17	66.00	5.99	-2.44
64.00	-3.19	51.40	64.00		
62.00	-3.54	51.66 51.96	62.00	6.41	
60.00	-3.92	51.96	60.00	6.67	-3.33
		52.30	58.00	6.95	-3.68
56.00		52.68	56.00	7.27	-4.07
		53.11	54.00	7.64	
		53.59	52.00	8.05	
		54.14	50.00	8.52	-5.48
48.00	-7.13	54.77	48.00	9.05	-6.06
46.00		55.49	46.00	9.66	-6.71
44.11		56.26	44.00	10.36	-7.45
Theta min			42.00	11.17	-8.30
Period =	0.668 \$	3	40.00	12.10	
			38.23	13.06	-10.26
			Theta min		
			Period = .	0,687 8	

-8					
- 1	U	L	L	L)

TH0=90 PSID0=50 PHID0= 5 TH0=90 PSID0=10 PHID0= 5 DTH=0.10 DTH=0.10 Theta Spin Precess. Spin rad/s degrees rad/s 5.00 50.00 90.00 5.00 88.00 88.00 4.96 86.00 4.61 49.68 84.00 49.54 84.00 4.93 82.00 4.93 80.00 4.10 49.29 4.94 78.00 3.95 49.18 78.00 4.97 76.00 5.01 74.00 3.66 48.99 74.00 5.06 72.00 3.52 48.91 72.00 48.84 70.00 5.21 68.00 3.25 48.78 5.30 66.00 3.12 48.73 48.69 64.00 5.55 62.00 2.87 5.71 60.00 2.75 48.63 2.62 48.61 58.00 6.09 2.49 56.00 48.61 6.32 54.00 48.61 54.00 6.89 52.00 50.00 2.08 48.66 50.00 48.00 1.93 48.71 46.00 8.08 44.00 1.59 48.85 44.00 8.61 42.00 1.40 48.96 42.00 40.00 40.00 9.92 38.00 0.96 49.24 36.00 49.44 36.00 0.39 34.00 12.87 32.00 0.04 49.97 32.00 14.25 -0.38 30.00 50.33 30.00 15.93 28.00 -0.88 50.78 28.23 17.71 Theta min = 28. 51.34 Theta min = 28.2 Period = 0.655 B 24.00 -2.26 52.06 22.00 -3.24 53.00 20.00 18.00 -6.23 55.92 16.00 -8.62 58.28 14.00 -12.09 12.00 -17.44 67.06

75.90

92.19

127 40 -89.03 138.60 = 5.62 degrees

(V)

-42.61 -77.82

10.00

8.00

6.00

5.62

Theta min Period = 0.542 8

(vi)

(tu)

rad/s

10.00

9.83

9.66

9.48

9.31

9.14

8.97

8.79 8.61

8.42

8.22

8.01

7.57

7.32

6.78

5.76

5.35

4.38

3.81

2.40

0.52

-0.67

-2.09

-3.79

-5.60

				100	
TH0=60 PS	ID0= 0	PHIDO= 5	TH0=60 PS DTH=0.10	ID0=50	PHIDO= 5
Theta	Spin	Precess.	Theta	Spin	Precess.
degrees	rad/s	rad/s	degrees	rad/s	rad/s
60.00	5.00	0.00	60.00	5.00	50.00
58.00	5.20	-0.26	58.00	4.96	49.87
56.00	5.43		56.00	4.92	49.75
54.00	5.69	-0.84	54.00	4.90	49.62
52.00	5.98	-1.18	52.00	4.89	49.49
50.00	6.32	-1.56	50.00	4.89	49.36
48.00	6.70	-1.98	48.00	4.90	49.22
46.00	7.14	-2.46	46.00	4.92	49.08
44.00	7.64	-2.99	44.00	4.96	48.93
42.00	8.22	-3.61	42.00	5.02	48.77
40.33	- 8.77	-4.18	40.00	5.10	48.59
heta min	= 40.3	degrees	38.00	5.20	48.40
eriod =	0.661 B		36.00	5.33	48.19
			34.00	5.49	47.95
			32.00	5.70	47.67
			30.00	5.96	47.34
			28.00	6,28	46.95
			26.00	6.70	46.48
			24.00	7.23	45.90
	101		22.00	7.92	45.16
			20.00	8.84	44.19
			18.00	10.10	42.90
			16.00	11.86	41.10
			14.00	14.44	38.49
			12.00	18.43	34.47
			10.00	25.06	27.82
			8.00	37.27	15.59
			6.01	63.40	
			Theta min		
			Period =	0.520 €	

19.1 19.4 GIVEN: GIVEN: BLOCK W=30 lb PARTICLE IN SIMPLE HARMONIC MOTION SPRING &= 20 1b/in. AMPLITUDE = 40 IN. PERIOD = 1.4 S. INITIAL DEFLECTION = Z.I In . \$ 20 lb/in EIND: RELEASED FROM REST HAXIMUM VELOCITY, Um EIND: MAIMUM ACCELERATION, QM (a) PERIOD TW, AND FREQUENCY, FA SIMPLE HAR MONIC MOTION (b) HAXIMUM VELOCITY, UM $x = x_{m} \sin(\omega_{n} t + \phi)$ AND ACCELERATION, am Wn= 211/En = 211/(145) = 4.480 rAD/S (a) Ym = AMPLITUDE = 40 in. = 3,333 ft X= Xm SIN (Wnt+0) X=(3.333) SIN (4.480++b) Wn= 1 k/m le=2016/in=240/6 X= Xm Wn cos Wn t+ b) 2m=15= 2mWn Wn=V(240 16/ft)/(30 16)/(32.2/6) \$ 20 lb/in. Um= (3.333 ft) (4.480 h) Wn= 16.050 rAD/S Tn= Wn Nm= 14.96 ft/5" $\ddot{x} = -\chi_m \omega_n^2 \sin(\omega_n t + \phi) \quad \ddot{\chi}_m = \alpha_m = -\chi_m \omega_n^2$ Tu= 211/16.050=0,39155 Qm= (3,333 ft) (4.480 r6 an= 67.1 ft/52 fn= 1/cn= 1/0.391 = 2.55 HZ 19.2 (b) 2m=2.1 in = 0.175 ft GIVEN: PARTICLE IN SIMPLE HARMONIC MOTION MAXIMUM ACCELERATION 72 m/s2 X= 0.175 SIN (16 USOT+0) FREQUENCY f = 8 HZ. HAXIMUM VELOCITY FIND: Um= xm Wn= (0.175 ft) (16 050 rAN/S) AMPLITUDE, XM Um= 2.81 ft/s MAXIMUM VELOCITY: N'm am = 1/2 m w = (0.175 ft) (16.050 +AD15)? SIMPLE HARMONIC MOTION am=45.1ft/s2.4 X= Xm sin(wat+0) 19.5 GIVEN: Wn=2TTfn=(2TT)(8HZ,)=16 TrAD/S BLOCK M= 32 kg x= xmwn cos(wn++) Nm=xmwn SPRING IR= 12 len/m $\chi = -\chi_m(\omega_n)^2 \sin(\omega_n t + \phi)$ $\alpha_m = \chi_m \omega_n^2$ $\alpha_m = 7.2 \text{ m/s}^2 = \chi_m (16 \text{ Trap/s})^2$ TINITIAL VELOCITY Un= 250 mm/s $k = 12 \, \text{kN/m}$ Km=(7.2 m/s2)/(1617 rAD/s)2 INITIAL DISPLACEMENT= 0 Km=2.849x10 m Xm= 2.85mm Um= Xm wn= (2.849 mm) (16TT rAD/S) (a) PERIOD TO AND FREQ. FO Um= 143,2 mm/s (b) ANDLITUDE KM 19.3 MAXIMUM ACCELERATION, UM GIVEN: (a) X= Xm SIN (Witt 0) U=O PARTICLE IN SIMPLE HARMONIC MOTION WN= VEIM = VIZXIO3 N/M AMPLITUDE = 300 MM MAXIMUM ACCELERATION = 5 m/s2 Wn= 19.365 FADIS FIND: Tn= 211/wn Tn= 211/19:365 1 No=250 mm/s MAXIMUM VELOCITY, Um FREQUENCY F Tn= 0.3245 SIMPLE HARHONIC MOTION fn=1/tn=1/0.324=3.08Hz K=Km SIN (Wnt+0) K= 0.300 M (b) @t=0 x=0 &=v=zsomm/s X= (0.300) SIN (Wht + d) (M) THUS x=(0.3)(wn) cos(wnt+0) (m/s) Yo= 0 = Km SIN (W, (0)+6) X=-(0.3) (Wni2sIN (Wnt+0) (M/s) AND DEO 1am1=(0.3 m/s)(Wn)2 am=5m/s2 20=0= xm wn cos (wn (0)+0)= xm wn Wn= |am1/(0.3m)=(5m/s2)/(0.3m)=16.667m0/s2 Un= 0.250 m/s= 12m (19.365 rap/s) WN= 4.082 rAD/S FN=Wn/ZTT 2m=(0,250 m/s)/(19.365 rAD/s) fn=(4.082 rAD/5)/(211 rAD/CYCLE)=0.649.7 HZ Xm=12.91:X10-3 m Km= 12.91 mm fu=0.650 HZ am= 4mwn=(12.91x103m)(19,365+AD/5)2 Um= KmWn=(0.3m)(4.0BZrAD/5) Nm= 1.225 M/S am= 4.84 m/s2

Tn=0.3915

19.6 GIVEN: PENDULUM IN SIMPLE HARHONIC HOTION PERIOD THE 1.35 HAXIMUM VELOCITY, Um= 15 In./S (a) AMPLITUDE OF THE HOTION, QM IN DEGREES (b) THE MAXIMUH TANGENTIAL ACCELERATION (04) M (0) SIMPLE HARHONIC HOTION e= em sin(watto) WN= SHICN=(SH)/(1.35) Wn= 4.833 rAD/S 8= 0m wh cos (whitto) Om= OmWn Um= Lôm= Lom Wn Om= 1Jm/lwn (1) FOR A SIMPLE PENDULUM wn=1 912 1= 3/w= 32.2 ft/s2 (4.833 HAD/S)2 l= 1,378 ft FROM (1) Om= Nm (EWn=(15/12 Pt/s)/(1.378 Pt) 4.833 H @w= 0.18769 FAB = 10.750 (b) Qt=10 HAX TAI GENTIAL ACCELERATION OCCURS WHEN O IS HAXIMUM, 0=-Omwisin (witto) BHAX OMWIZ, (at) HAX= LOMWIZ (at) HAX = (1.378 ft) (0.1879 HAD) (4.833 MD) (a+) m = 6.04 ft/52 19.7 GIVEN: SIMPLE PENDULUM L= 800 mm, ⊖HAX= 6° FIND: (a) FREQUENCY OF OSCILLATION. (b) HAXIMUH VELOCITY NIM OF THE BOB Wn= V 9/R = V(9.81 m/52) (0.8 m) (a) Wn= 3,502 rAD/s fn= Wn/2TT=(3.502 FAD/S)/2TT fu= 0.557 Hz (b) == = sin(wnt+ p) = Omwn cus(wnt+0) DW= OWWN N'm= (8m=10mWn=(0.8m)(6)(11 rad) (3.502 (ad)) (b) squaeine (1) and (2) AND ADDING, Um= 293.4 XID m/S

19.8

GIVEN

PACKAGE A IN SIMPLE HARHONIC HOTION AT A FREQUENCY WHICH IS THE SAKE AS THE HOTOR WHICH DRIVES IT. PEAK ACCEPATION = 150 ft/sz AMPLITUDE=2.310.

FIND:

REQUICED SPEED OF THE MOTOR IN PPM HAXIMUM VELOCITY OF THE TABLE (PACKAGE) IN SIMPLE HARMONIC HOTION

CHAX = XHAX WIZ 150 ft/s=(2.3 ft) Wn2

WW= (782.6 rAD/5) Wn = 27.98 rAD/s $f_{N} = \frac{\omega_{N}}{2\pi} = \frac{27.98}{2\pi} = 4.452 \text{ Hz} (CYCLE/S)$

1 RPM = 1 CYCLE/(IMIN.) (605/MIN.) = 1/60 (HZ.)

(fHz)/(160Hz)= 4.452 = 267 RPM =

HAXIHUM VELOCITY WHAX=XMAXWN=(2.3 ft)(27.98 HAD/S)

UHAY= 5.36 ft/s

19.9 GIVEN:

PARTICLE MOTION K= SSINZt+4coszt (M,S)

FIND:

(a) PERIOD, To

(b) AMPLITUDE, YM

(C) PHASE ANGLE, O

FOR SIMPLE HARMONIC HOTION X= Xm SIN (Wnt+0)

DOUBLE ANGLE FORMULA (TRIGONOMETRY) SIN (A+B)=(SINA) (COEB)+(SINB)(COSA)

LET A= wnt , B= &

THEN X= XM SIN (WILLO) X= Xm(>INWnt)(cos \$)+Xm(SIN\$)(cos wit)

X= (Xmcosd)(SINW, t)+(XmSIND)(COSWnt)

GIVEN X= 55INZt + 4coszt

COMPARING Wn=2 Xmcosp = 5 KmSIND = 4 (2)

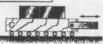
(a) $T_N = \frac{2\Pi}{\omega_N} = \frac{2\Pi}{(2 \text{ rAO/s})} = \Pi s$ T=3.14 5

Xm (605) + Xm 51N3 + = A2+5 > Xm= 6.40 m

(C) DIVIDE (2) BY (1) TAN 0=4 0= 38.7°

Um = 293 mm/s





GIYEN:

TABLE C HOVES IN SIMPLE TARMONIC MOTION WITH AMPUTUDE 31 45= 0.65 BETWEEN BLOCK BAND C

FIND:

LARGEST FREQUENCY ALLOWED FOR NO SUDING NEWTON'S LAW

N=W

Fe = ma BLOCK B MOVES IN SIMPLE HARMONIC MOTION WITH THE SAHE FREQUENCY AS C WHEN THERE IS NO SLIDING

X= Xm SIN (Whttp) R=-Xmw2 SIN(wnt+0) HAXIMUM ACCUERATION a= xmw2

Ff= M Kmwn FOR NO SLIDING

Ff>45W

OR 45W > W XmWn

Wu > 459/2m Wn2> (0.65) (32.2 ft/s2) /(3/12 ft) = 83.72 (rAD/S)2 Wu> 9.150 fu= Wn/2TT = (9.150)/2TT = 1.456Hz

19.11



k= 12 GCN/W

GIVEII:

INITIAL DISPLACEMENT OF THE BLOCK = 300 MM DOWNWARD

FIND:

1.5 5 AFTER THE BLOCK IS RELEASED, (a) TOTAL DISTANCE TRAVELED BY THE BLOCK (b) ACCELERATION OF

(a) x= xm sin (watto) (EQUIL) 32 leg V(0)=0.3 M 32 kg (x(0)=0

k = 12 kN/m

Wn= /k/m = /(12x103N/m) Wn=19.365 FAD/S Tn=211/Wn=(211)/19.365)

THE BLOCK

Tn= 0.3245 5 INITIAL CONDITIONS k(0)=0.3m, k(0)=0

0.3= KmsIN(0+0) X10; = 0 = Km w, cos(0+0) D=11/2

K(t)=(0.3)51N(19.365++ 17/2) Tn= 0.3245 5 7K(1.55)=(0.3) SIN[(19.365)(1.5)+T1/2]=-0.2147 M 8-(1.55)=(0.3)(19.365)COS[(9.365))1.57+1] = 4.057 m/s

IN ONE CYCLE, BLOCK TRAVELS -0.2147m (4)(0.3m)=1.2 m 0.6984 M _ - O (EQUIL) TO TRAVEL 4 PYCLES IT TAKES (4 c4c)(0.3245 5/24c) = 1.2980 S

THUS , TOTAL DISTANCE TRAVELED 15 4(1.2)+0.6+(0.3-0.2147)=5.49M

(b) X(1.5)=-(0.3)(19.365)² SIN(19.365)(1.5)4=]= 80.5 m/s}

19.12



GIVEN:

WA= 316, k= 216/11 INITIAL VELOCITY OF A= 90 In./s 1

FIND:

(a) TIME REQUIRED FOR THE BLOCK TO HOVE 3 IN UPWARD (b) CORRESPONDING VELOCITY AND ACCELERATION

unu K= Km sin (wnt+d) Wn= 1/1 R=216/10=2416 Wn=V=16/66 (316/32296/52) 1=31n. Wn=16.05 TAD/S (EQUILIBRIUM)

\$(0) = 90 in./s 4(0)= 0 74(0)=0= Km Sin(0+0) Φ=0

X(0)= Km Wh cos (0+0) 1(0)=90=7.5ft/s

7.5 = 4m (16.05) Km=0.4673 ft $X = (0.4673) \sin(16.05 t) (ft, s)$ (1)

(a) AT X= 3/12=0.25ft 0.25 = 0.4673 SIN(16.05t) t= 510 (0.25)/(0.4673) = 0.03525

(b) K= Kmwn cos(wnt) X=-Kmwn2sinwnt t=0.0352, R=(0.4673)(16.05)(05)(16.05)(.0352)] X= 6.34 ft/s 1

x=-(.4673)(16.05) SIN(16.05)(.0352)]=-64A FE/S 19.13

REFER TO FIGURE IN PROBLEM 19.12 ABOVE GIVEN:

WA=316, k=216/in, U0=90 in. 15 P (SAME AS 19.12) FIND:

AFTER 0.90 5, POSITION, VELOCITY AND ACCELERATION OF THE BLOCK

X= Xm SIN (Watto) $\hat{x} = \chi_{m} \omega_{n} \cos(\omega_{n} t + \phi)$ R=-Kmw= Ren (wnt+0)

SINCE THE GIVEN DATA IS THE SAME AS IN PROBLEM 19.12 ABOYE, THE EQUATION OF MOTION IS THE SAME AS EQUATION (1) IN 19.12 \$=0, X m = 0.4673 ft, Wn=16.05 RADIS AND X, X, X ARE + 1

X=(0.4673) SIN (16.05 t) (ft,5)

AT 0.905 K= (0.4673) SIN(14.05)(0.90) = 0.445 ft 1 ◀

 $\chi = (0.4673)(16.05)\cos[(6.05)(0.90)] = -2.27 \text{ ft/s}$

 $\chi^{\infty} = -(0.4673)(16.05)^{2} \sin[(16.05)(090)] = 114.7 \text{ ft/s}^{1}$



L= 800 mm AT t= 0, 0=+5°, 0=0 ASSUME SIMPLE HARMONIC MOTION

FIND:

(a) o (a) N AND a of the BOR

 $\Theta = \Theta_m \sin(\omega_n t + \phi)$ $\omega_n = \sqrt{\frac{9}{2}} = \sqrt{\frac{9.81 \text{ m/s}}{0.8 \text{ m}}}$ ENITIAL CONDITIONS

θω)=5°=(5) (π)/180 LAD Å(0)=0 Wn=3.502 FAD/S

0(0) = 511 = 0m SIN(0+4)

 $\hat{\Theta}(0) = O = \Theta_{\text{M}} \omega_{\text{N}} \cos(0+\hat{\phi})$ $\hat{\phi} = \pi/2$ $\Theta_{\text{M}} = \sup_{n=0}^{\infty} \epsilon_{\text{AD}}$

田= TRO SIN(3.502+ 年)

AT t= 1.65 \(\text{\tin}\etat{\text{\ti}\text{\text{\text{\text{\text{\text{\text{\text{\text{\text{\tin}\text{\text{\text{\text{\text{\text{\text{\text{\text{\text{\tin}\eta}\text{\text{\text{\text{\text{\text{\text{\text{\text{\tinit}\text{\text{\text{\text{\text{\text{\text{\text{\tinit}\text{\text{\text{\text{\text{\text{\text{\text{\text{\text{\text{\tinit}\text{\text{\text{\tinit\text{\text{\text{\text{\text{\texi}\text{\text{\text{\text{\text{\text{\text{\text{\tinit}\text{\texi}\tin}\text{\text{\text{\text{\text{\texi}\tint{\text{\tiin}\tintet{\text{\texi}\tint{\text{\tin\tintet{\text{\text{\tin\tiin}\tintet

(b) 40, +8, +8 9(1.6)=0.06786 RAD 2 0=28 04=28

 $\hat{\Theta} = \Theta_{\text{m}} \omega_{\text{n}} \cos(\omega_{\text{n}} t + \phi) = \left| \frac{\epsilon_{\text{II}}}{180} \right| (3.502) \cos[3.502](1.6) + \frac{\pi}{2}$

U=L= (0.800 m)(0.1923 0.1923 280)/s

 $\widetilde{\Theta} = -\Theta_{\mathsf{m}} \omega_{\mathsf{n}}^{2} \sin(\omega_{\mathsf{n}} t + \phi) = -\left(\frac{511}{180}\right)(3.502)^{2} \sin\left[(3.502)(1.6) + \frac{11}{2}\right]$

0= -0.8319 BAD/52

a= V(a+)2+(an)2

 $Q_t = 10^{0} = (0.8 \text{ m})(-0.8319 \text{ RAD/5}^2) = 0.6655 \frac{\text{M}}{\text{SZ}}$ $Q_N = 10^{2} = (0.8 \text{ m})(0.19223 \text{ RAD/5}^2 = 0.02956 \frac{\text{M}}{\text{SZ}})$

 $Q = \sqrt{(0.6655)^2 + (0.02956)^2} = 0.6662 \text{ m/s}^2$

a=0.666 m/52

19.15

GIVEN:

M= 5kg, UNATTACHED TO THE SPRING WHEN COLLAR IS PUSHED DOWN IBOMM OR HORE AND RELEASED IT LOSES CONTACT WITH THE SPRING

EIND:

(a) THE SPRING CONSTANT &

(b) POSITION, VELOCITY AND ACCELERATION O. 165 AFTER IT IS PUSHED DOWN 180 mm AND RELEASED.

0=0 0=+g1 - a=+g1 - x=0 + x=0 - x=0,180 m

 $\chi = \chi_m \sin(\omega_n t + \phi)$ $\chi_0 = \chi_m \sin(0 + \phi) = 0.180 \text{ m}$ $\chi_0 = 0 = \chi_m \cos(0 + \phi)$

Φ= Π/Z Km= 0.180 m γ= 0.180 sin(ω, t+<u>π</u>)

WHEN THE COLLAR TUST LEAVES THE SPRING, ITS ACCELERATION IS Q V AND U=0

&=(0.180) wn cos (wnt+II)

V0=0

U=0 0=(0.180) Wn cos (Wnt+17/2)

(Wnt+11/2) = T1/2

a=-g=(0.180)(wn)2SIN(wn t+ 11/2)

 $-g = (-0.180)(\omega_n^2)$ $\omega_n = \sqrt{\frac{9.81 \text{ m/s}^2}{0.180 \text{ m}}}$ $\omega_n = 7.382 \text{ PAD/s}$

wn=Vk/m

k=mw= (5kq)(7.382:40/5)=272.5N/M

k=273 N/m

(b) Wn= 7.382 BAD/S X= 0.180 SIN[(7.382)++ T/2]

At t= 0.16 s POSITION

X=0.18051N[(7.392)(0.16)+ T/2]=0.06838M

X= 68.4 MM

berom edrinbeinh sællion

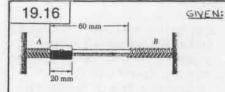
VECOCITY: \$= (0.180) (7.382) COS[(7.382)XO.16)+ T/2]=-1.729W

2= 1.229 m/s 1

ACCELERATION

X= - (0.180)(7.382) SIN[(7.382)(0.16)+1]]=-3.726 M

x= 3.73 m/52 1



Mc=8kg

R=600 N/M

FOR EACH SPRING

INITIAL DEFLECTION

OF SPRING A

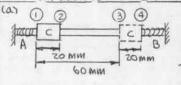
= 20 mm.

NO FRICTION

FIND;

(a) PERIOD

(b) POSITION OF C AFTER 1.5 5



FOR EITHER SPRING $T_{N} = \frac{2\pi}{\sqrt{k/m_{c}}}$ $T_{n} = \frac{2\pi}{\sqrt{600 \, N/m}/8 \, kg}$ $T_{N} = 0.72555$

COMPLETE CYCLE 15 1234,4321

TIME FROM 1 TO 2 IS TWA WHICH IS THE SAME AS TIME FROM 3 TO 9, 4 TO 3 AND 2 TO 1
THUS THE TIME DURING WHICH THE SPRINGS ARE COMPRESSED IS 4 (Told) = TN = 0.72555

YELOCITY AT 2 08 2).

U=0 T=0 V=1 & x2=1 (600 N/m) (0.020 m)²

$$T_2 = \frac{1}{2} M U_2^2 = \frac{1}{2} (8 \text{kg}) (U_2)^2$$
 $T_2 = 4 U_2^2$

 $V_2 = 0$ $\overline{V_1} + V_1 = \overline{V_1} + V_2$ 0 + 0.1732 m/s 0.1732 m/s

TIME FROM 2 TO 3 15 t = (0.020m) =0.115455

THUS

TOTAL TIME FOR A COMPLETE CYCLE IS $T_c = T_{n+2} t_{2-3} = 0.7255 + 2(0.11545) = 0.9564$

Te= 0.9565

(b) FROM (a), IN 0.9564 THE SPRING A IS AGAIN FULLY (CHAPPESSED. SPRING B IS COMPRESSED)
THE SECOND THE IN IS CYCLES OR (I.S) (0.9564) =
1.4346 S. AT IS 5 THE COLLAR IS STILL IN
CONTACT WITH SPRING B HOVING TO THE LEFT
AND IS AT A DISTANCE AX FROM THE
MAXIMUM DEFLECTION OF B EQUAL TO $\Delta X = 20 - 20 \cos \left[\frac{211}{10.7255} (1.5-1.4346)\right]$

AX= 20 - 16.877 = 3.123 mm

THUS COLLAR C 15 60-3,123= 56,877 MM FROM ITS INITIAL POSITION

FROM INITIAL POSITION

19.17 16 kN/m 8 kN/m

GIVEN:

HASS AND SPEINGS AS SHOWN
AFTER THE HASS IS POLLED

DOWN AND RELEASED FROM
REST THE AMPUTUDE OF THE
RESULTING MOTION IS 45 mm

(a) THE PERIOD AND FREQUENCY OF THE MUTION

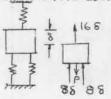
(b) THE PAXIFUM YELDCITY

AND ACCELERATION OF THE

BLOCK

(a) DETERMINE THE CONSTANT & OF A SINGLE SPRING EQUIVALENT TO THE THREE SPRINGS

FIND:



k = 32 kN/m $\omega_{n} = \sqrt{\frac{1}{2} \times 10^{3} \text{ N/m}}$ $\omega_{n} = \sqrt{\frac{1}{2} \times 10^{3} \text{ N/m}}$

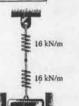
 $(1N=1 \text{ leg} \cdot 1 \text{ m/s}^2)$ $\omega_N = 30.237 \text{ rAD/s}$ $C_N = 2 \text{ T/} \omega_N = 2 \text{ T/} 30.23 = 0.2085$ $f_N = 1/C_N = 4.81 \text{ Hz}$

P= & 8 = 16 8+88+88

(b) K= Km51N(Wnt+\$) K0=0.045 M= Km Wn= 30.24 RAD/s

7=0.04531N (30.24+4) \$=(0.045)(30.24)(205(20.24+4)) U=1.361 M/s \$=-(0.045)(30.237)371V(30.24+4) QMAx=41.1 M/s

19.18



MASS AND SPEINGS AS SHOWN AMPLITUDE OF MOTION IS 45 MM AFTER HASS IS TULLED DOWN AND RELEASED FROM REST

FIND:

GIVEN:

(a) PERIOD AND CREQUENCY OF MOTION

(b) HAXIHUM VELOCITY AND ACCHERATION

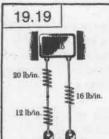
(a) DETERMINE THE CONSTANT & OF A SINGLE SPING EQUIVALENT TO THE TWO SPEINGS SHOWN $S = 5_1 + 8_2 = P + P =$

fu= tn= 0.416= 2.41HZ

(b) Wn=2\pi fn=2\pi (z:41)=15.1z RAD/S X= 0.045 SIN (15.1z++\$) X= (0.045) (15.1z > cos (15.1z+\$)

x'=-(0.045)(15.13)2SIN(15.13+40)

QHAX = 10.29 m/s2



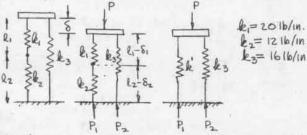
30 16 BLOCK AT t=0, %=1.75 in. DOWN WARD, U=0

FIND:

(a) PERIOD AND FREQUENCY OF KOTION

(b) MAXIMUM DELOCITY AND **LCCELERATION**

DETERMINE THE CONSTANT & OF A SINGLE SPRING EQUIPMENT TO THE THREE SPRINGS SHOWIN



SPEINGS I AND Z (FORCE IN EACH SPEING IS THE SAME) $P_1 = P_1 + P_1$ 8= 81+82

& IS THE SPRING CONSTANT OF A SINGLE SPEING EQUIVALENT TO SPEINGS I AND Z SPRINGS & AND 3 (DEFLECTION IN GACH SPRING IS THE SAKE) P=RS P=RS P= P1 + P2 P= 62 S

> les= le'8+le, 2 k=k+l2= k1l2+k3

R=(20)(12) + 16 = 23.5 lb/In = 282 lb/ft (20+12)

(a) $C_n = \frac{zT}{\sqrt{le/m}} =$ = 0.3615(2821/ft) (3016/32.2 ft/s2

fn= 1/cn= 1/0.361= 2.77 Hz

(b) X=Xm SIN (Wnt+)

Km=1,75 in = 0.1458 ft

X=(0.1458)(SIN 17.41++6)

X=(0.1458)(17.40) SIN(17.41+6)

 $\sum_{k=-(0.1458)(17.40)^2}^{\infty} (0.458)(17.41) = 2.54 \text{ ft/s}$

QHAX = 44.14 17/5

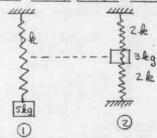
amax 441 Pt/52

19.20 GIVEN:

> J-AG BLOCK ATTACHED TO A SPRING FIXED AT THE OTHER END VIBRATES WITH A PERIOD THE GBS SPEING CONSTANT & IS INVERSELY PROPORTIONAL TO THE SPRINGS LENGTH.

FIND:

THE PERLOD FOR A 3 kg BLOCK ATTACHED TO THE CENTER OF THE SAME SPRING FIXED AT BOTH ENDS



EQUIVALENT SPRING CONSTANT Q= 2 R+2 R= 4 R (DEFLECTION OF EACH SPRING 15 THE SAHE)

(TN=6.8=2TT/k/5kg) (Cn)=2TT/4k(5kg)

k=(217) (5kg) (6.85)2 k= 4.269 N/m

(Tn)=2TT/V44.269N/m/(3A) (Cn)= 2.635

19.21 GIVEN:



SYSTEM AS SHOWN IS HOVED O. SIN. DOWNWARD AND RELEASED FROM

FERIOD FOR RESULTING MOTION 15 Cn= 1.55

FIND:

(a) CONSTANT &

(b) MAXIMUM VELOCITY AND ACCELERATION OF THE BLOCK

SINCE THE FORCE IN EACH SPRING IS THE SAME, THE CONSTANT & OF A SINGLE EQUIVALENT SPRING IS le'= le/2.5 (SEE PROB 19.19) 本=立庭+生生

Wn=2Tfn=(2T)(2.77)=7.40 100 (a) Tn=1.5 s.=2TT//k/m $k = \frac{(3\pi)^2}{(1.55)^2} \left(\frac{3016}{37.2 \text{ Pt/s}^2} \right) (7.5) = 40.868 \text{ lb/ft}$ R= 40,916/ft

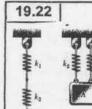
(b) X= Kmsin(wnt+\$) X= xmwneus(wnt+0) UHAX= Kmwn Wn = 21 = 211 = 4.189 RADIS

YM= 0.811.= 0.06667ft WHAX = (,06667 FE) (4189 RAD/S) WHAX=0.7279 PHS

X=-xmwn2cos(wnt++)

(4.189)2 = (0.0667ft)(4.189)2

| anax = 1.170 ft/5



PERIOD FOR SPRINGS IN SERIES, T5= 55 PERIOD FOR SPRINGS IN PARALLEL, Tp=ZS

RATIO OF SPRING CONSTANTS RILEZ

19.23 CONTINUED

(0.6049)(k,+1.2)= &,

le,=1,838 leN/m

(b)
$$T = \frac{2\pi}{\sqrt{k_1/m}} (0.4.5)^2 = \frac{(2\pi)^2 m}{(1.838x)^2}$$

 $M = (0.95)^2 (1.838 \times 10^3 \text{ N/m})$

 $M = 37.7 \, kg$

EQUIVALENT SPRINGS

PAVILLEL &p= &,+lez

$$\tau_{5} = \frac{2\pi}{\omega_{5}} = \frac{2\pi}{\sqrt{k_{5}/m}} \qquad \tau_{p} = \frac{2\pi}{\omega_{p}} = \frac{2\pi}{\sqrt{k_{p}/m}}$$

$$T_p = \frac{2\pi}{\omega_p} = \frac{2\pi}{\sqrt{k_p/m}}$$

$$\left(\frac{T_s}{T_p}\right)^2 = \left(\frac{5}{2}\right)^2 = \frac{k_p}{k_s} = \frac{(k_1 + k_2)}{(k_1 k_2)/(k_1 + k_2)} = \frac{(k_1 + k_2)}{k_1 k_2}$$

(6.25)(k, k2) = k2 + 2/2, k2+ k2

$$k_1 = (4.25)k_2 \mp \sqrt{(4.25)^2k_2^2 - 4k_2^2}$$

$$\frac{k_1}{k_2} = 2.125 + \sqrt{3.516}$$

le. 1 kg = 4

19.23

BIVEN:

PERIOD = 0.75, = T AFTER &2 IS REHOVED PERIOD = 0.95. = T'

FIND:

(a) k,

(b) MASS OF A

$k_0 = 1.2 \, \text{kN/m}$

EDUIVALENT SPRINGS

FI =
$$k_1$$
 Fz = k_2 S

A Fi F1+F2 = F = k_2 S

 k_2 k_3 k_4 k_5 k_6 k_6 k_6 k_6 k_6

(a) BOTH SPRINGS
$$T = \frac{2\pi}{Vkelm} = 0.7 \text{ s}$$
 k_1 ALONE, $T = \frac{2\pi}{Vk_1/m} = 0.9 \text{ s}$

$$\frac{T}{T} = \frac{0.7}{0.9} = \sqrt{\frac{k_1}{ke}} = \sqrt{\frac{k_1}{k_1+k_2}}$$

$$\left(\frac{17}{9}\right)^{\frac{2}{3}} = 0.6049 = \frac{k_1}{k_1 + 1.2}$$

19.24

GIVEN: PERIOD FOR SYSTEM SHOWN 15 T= 1.65

PERIOD AFTER A 7-kg COLLAR IS PLACE DON A, IS T'= 2.15

FIND:

(a) MASS OF A

(b) k

T= 211 _ 163 INITIALLY V6e/ma

AFTER The MASS IS ADDED TO A,

$$\tau' = \frac{2T}{\int k/(m_h + \tau)} = 2.1 \text{ s}$$

(a)

 $\left(\frac{2.1}{1.6}\right)^2 = M_1 + \frac{7}{1.6}$

(1.7227) (MA) = MA+ 7

MA= 9.69 lea

(6)

k= (2112 (ma)/(T)2 b= (211)2 (9.69 kg)/(1.65)2

le = 149.4 kg/52

k= 149.4 N/m



FOR SYSTEM SHOWN
PERIOD C= 0.25
AFTER Rez IS REMOVED
AND BLOCK A IS CONNECTED
TO RI, PERIOD C'= 0.125



FIND:

(a) lei (b) WEIGHT OF BLOCK A.

EQUIVALENT SPRING CONSTANT FOR

$$ke = \frac{k \cdot k_2}{(k_1 + k_2)}$$

$$T = \frac{2\pi}{V \cdot k_2 / M_A} = \frac{2\pi}{V(k_1 \cdot k_2) / (M_A)(k_1 + k_2)}$$

FOR & ALONE

(a)
$$\frac{\tau}{\tau} = \sqrt{\frac{(k_1 + k_2)(k_1)}{(k_1 \cdot k_2)}} = \sqrt{\frac{k_1 + k_2}{k_2}}$$

$$k_2 \left(\frac{\tau}{\tau}\right)^2 = k_1 + k_2$$

T/C' = 0.2/0.12 = 1.6667, $k_2 = 2016/m$ (20 16/in.) $(1.66675)^2 = k_1 + 2016/m$

le = 35.6 16/m.

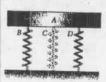
(b)
$$\tau' = \frac{2\pi}{\sqrt{k_1/m_A}} \qquad m_A = W_A/g$$

$$W_{A} = \frac{(32.2 \text{ ft/s}^2)(0.12 \text{ s})^2(426.7 \text{ lb/ft})}{(2\pi)^2}$$

WA = 5.01 lb

19.26

GIVEN:



WA = 100 lb

kB = kB = k = 120 lb/ft

FREQUENCY REHAINS.

THE SAHE WHEN AN

BOID BLOCK IS ADDED

TO A AND A SPRING OF

CONSTAIT & IS ADDED

TO THE SYSTEM

FIND: kc

FREQUENCY OF THE ORIGINAL SYSTEM

SPRINGS BANDD ARE IN PARALLEL

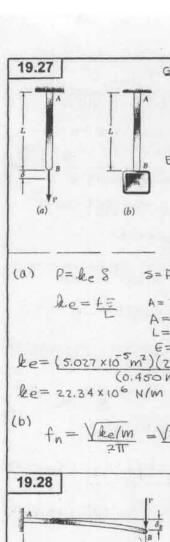
$$W_n^2 = \frac{ke}{MA} = \frac{24016/At}{(10016/32.2 ft/s^2)}$$

Wn= 77.78(rAD/s)2

FREQUENCY OF NEW SYSTEM

SPRINGS A, BANDC ARE IN PARALLEL

&c= 192.0 16/ft



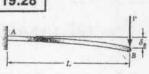
GIVEN: 8 = PL/AE L= 450 mm E= 200 GPa ROD DIAMETER = 8 mm. m= 8 & q

EIND: (a) EQUIVALENT SPEING CONSTANT OF THE ROD, (Re) (b) FREQUENCY OF VEETICAL VIREATIONS OF THE 8-20 MASS

(a)
$$P = ke S$$
 $S = PL/AF, P = (AE) S$
 $ke = \frac{1}{L}$ $A = \pi c^2/4 = \pi (8 \times 10^3 \text{m})^2/4$
 $A = 5.027 \times 10^{-5} \text{m}^2$
 $L = 0.450 \text{ m}$
 $E = 200 \times 10^9 \text{ N/m}^2$
 (0.450 m)

$$f_{N} = \frac{\sqrt{ke/m}}{2\pi} = \frac{\sqrt{22.3 \times 10^{6}/8}}{2\pi} = 265.96 \text{ Hz}$$

$$f_{N} = 266 \text{ Hz}$$



GIVEN: SB=PL3/SEI L= 10 ft E= 29×106/6/11/2 I= 12,4 In4

FIND: (a) EQUIVALENT SPRING CONSTANT

(b) FREQUENCY OF A 570-16 BLOCK AT B

(a)
$$P = k_e \delta_B$$
 $S_B = PL^3/3 \in I$, $P = (3 \in I) \delta_B$
 $k_e = \frac{3 \in I}{L^3} = (3)(\frac{29 \times 10^6 \text{ lb/m}^2}{(10 \times 12 \text{ in})^3})(12.4 \text{ in}^4)$

ke= 624.3 lb/m

Rp=624.3 lb/1n.

(b)
$$f_n = \sqrt{\frac{ke/m}{z\pi}}$$
 $k_e = 624.3 \, lb/in$ $= 7.49 \, 2 \times 10^3 \, lb/ft$ $f_n = \sqrt{\frac{7.492 \times 10^3 \, lb/ft}{2\pi}}$ (520 lb/(32.2 ft/s²))

fn=3.428 H3

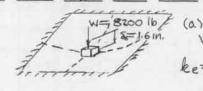
fu= 2.43 4z

19.29

GIVEN:

STATIC DEFLECTION OF THE FLOOR OF A BUILDING UNDER AN 8200-16 PIECE OF HACHINERY EQUALS &= 1.6 IN.

FIND: (a) EQUIVALENT SPEING CONSTANT & (b) THE SPEED IN RPM OF THE HACHINGRY THAT SHOULD BE AVOIDED SO AS NOT TO COINCIDE WITH THE NATURAL FREQUENCY OF THE SYSTEM.



W=ke &s Re= W- 8200 16

Rp= 5130 16/11. 1/(5130 X12 16/ft) (8200 lb/32.Zft/s2) fn= 2.473 Hz

I HZ= I CYCLE/5= 60 RPM

SPED=(2.473 HZ) (60 RPM) = 148.4 RPM

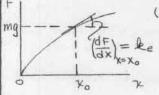
19.30

FORCE-DEFLECTION EQUATION FOR A HON-LINEAR SPRING, F=5x1/2 (N,m)

FIND:

(a) STATIC DEFLECTION X. UNDER A 120-9 BLOCK

(b) FREQUENCY OF VIBRATION OF THE BLOCK FOR SHALL OSCILLATIONS



(a) $mg = (0.170 kg)(9.81 m/s^2)$ mg = 1.177 N $F = mq = 5 \times 0^{1/2}$ Vo= (1.177)=0.0554 m

Vn= 55.4 mm

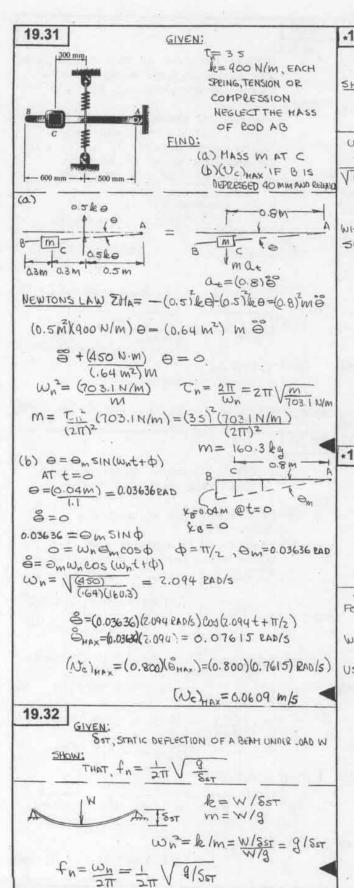
(dF) = 10:618 N/M

ke=10.618 N/M

$$f_{N} = \frac{\sqrt{ke/m}}{2\pi} = \sqrt{\frac{(0.618 \text{ N/m})(.120 \text{kg})}{2\pi}}$$

fn= 1.4971 Hz

fu= 1.497 Hz.



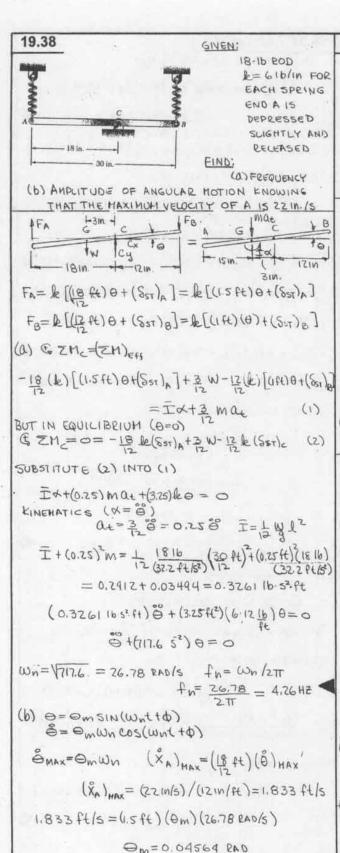
GIVEN: 4 V & J T/2 d 4 SHOW BY EXPANDING THE INTEGRAND OF THE ABOVE EQUATION, TN = ZTT (1+ 1 5102 PM) USING THE BINOHIAL THEOREM, WE WRITE -= 1-214 (DM) 214 0] 11-21N3 (BM) 21N3 4 = 1+ = sin20m sin20+ WHERE WE NEGLECT TERMS OF ORDER HIGHER THAN 2 SETTING SINZO = 1 (1-coszó), WE HAVE TN = 4 / 1 / [1+ 1/2 SIN2 0/ [1/(1-cos24)]] do = 4 \ [1+ 1 512 0m - 1 512 0m coszojah [++4(SIN2 @m)+- & SIN2 @m SIN24] 1=+7(211, 0M) =+0] CN= SIL/ 3 (1+721/00) +19.34 TN=211 (1+ 1 SIN 20m) (PROB. 19.33) FIND: AMPLITUDE BY OF A PENDULUM FOR WHICH THE PERIOD OF A SIMPLE DENDUCUM IS & PERCENT LONGER THAN THE PERIOD OF THE SAME PENDULUM FOR SMALL OSCILLATIONS FOR SHALL OSCILLATIONS (Th) = 2TT & WE WANT Th= 1.005(Th) = 1.005 271 \ USING THE FORMULA OF PEOB 19.33, WE WRITE Tn=(Cn) (1+ 1 SIN2 DM) = 1.005(Cn) SIN2 0m = 4[1.005-1] = 0.02

21NOW = 1.05

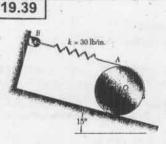
Om = 8.130°

Om= 16,30

19.35 GIVEN 19.37 CONTINUED IN MOTION & ZMA = (ZMA)eff DATA OF TABLE 19.1 PENDULUM LENGTH, 2= 750 mm (0.7) [kg[0.70+(ssr)d-mg]+1.4[kc(1.40+(ssr)c]= FIND: (A) PERIOD FOR SHALL OSCILLATIONS (b) PERIOD FOR AMPLITUDE O == 60° -IX-(0.7)(max) BUT IN EQUILIBRIUM (8=0) (C) PERIOD FOR AMPLITUDE OM= 900 9) ZHA=0=0.7[kg(SsT)B-mg]+1.4 kc(SsT)c (2) (EQ. 19.18 FOR SHALL OSCILLATIONS) (a) T= 211/2 $T_n = 2\pi \sqrt{\frac{0.750 \text{ m}}{4.81 \text{ m/s}^2}} = 1.737 \text{ s}$ SUBSTITUTING (2) INTO (1) Th=1.7375 IX+0.7 mat + (0.7) Les 0+(1.4) le 0=0 (b) FOR LARGE OSCILLATIONS (Eq. 19.20) EINEHATICS (4=8) てn=(2片)(2TTV용)=2片(1,7375) FOR 0 = 60° 1 = 1.686 (TABLE 19.1) [I+ m(0.7)2] & +(0.7)2ka+(14)2kc] 0=0 Tu(60°) = 2 (1.686) (1.7375) = 1.8645 I= 12 ml= 12 (5kg)(1.4m)=0,8167kg-m2 Tn(60)=1.8645 (c) FOR OM=90°, K= 1.854 (.7)2m=(0.49 m2)(5 kg)=2.45 kg-m2 Th= 2 (1.854)(1.7375) = 2.055 (0.7)2 kg + (1.4)2 kc = (.49 m2)(500 N/m)+(146 m2)(204) . 19.36 GIVEN: DATA OF TABLE 19.1 PERIOD = 25, AMPLITUDE = 900 = 245+1215.2=1460.2 N.M FINO: [0.8167+2.45] 0+1460.20 = 0 LENGTH LOF A SIMPLE PENDULUM (In.) FOR LARGE OSCILLATIONS (EQ 19.20) \$ + (460.7 N·m) = 0 FOR Om=900 でい= (学)(SUL人号 K=1.854 (TABLE 19.1) 0+4470=0 (N/kg/m=5-2 l=(25)(322 ft/s2) = 2,342 ft, (= 28.1 IN. (4)(1.854) W= \44752 = 21.14 PAD/5 19.37 $f_n = \frac{\omega_n}{2\pi} = \frac{21.14}{2\pi} = 3.36 Hz$ GIVEN: 5-kg ROD AC SPRING B, R= STOONIN (b) == = m SIN(Wort+o) SPRINGC, &= 620N/W (TENSION OR COMP.) 0=(0m)(wn) cos(wnt+0) WHEN END CIS HAXIMUM ANGULAR VELOCITY, Om= Om WIN DEPOESSED SLIGHTLY (a) FREQUENCY OF HAXIMUM VELOCITY AT C MOITAGBIV (RJ= 1.4 Sm=(1.4 m)(0m)(wn) (b) AMPLITUDE OF POINT C KNOWING $\theta_{\rm m} = \frac{(0.9 \text{ m/s})}{(1.4 \text{ m})(21.14 \text{ RAD/s})} = 0.03041 \text{ RAD} \quad \omega_{\rm m} = 21.14 \text{ RAD/s}$ THAT ITS HAXINUM VELOCITY IS 0,9 m/s HAXIMUM AMPLITUDE AT C $(X_c)_m = (1.4 \text{ m})(\Theta_m) = (1.4 \text{ m})(0.03041)$ FB= &B (YB+(SET)B) = &B (0.70+(SET)B) (Xc)m=0.0426 m Fc=lec(xc+ (Sst)c)=lec (1.40+(Sst)c)



Om= 2.610



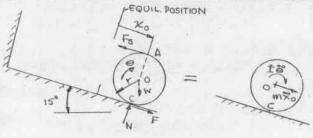
GIVEN:
30-16 CYCINDER
ROLLS WITHOUT
SCIDING.
DATA AS SHOWN.
INITIAL DISDUCCHED
= 210. DOWN

(a) FIND:

(a) PERIOD

(b) MAXIMUM

ACCELERATION OF C



SPRING DEFLECTION, $x_A = x_0 + x_{A/0}$ $x_{A/0} = r\theta$ $\theta = x_0/r$ $x_A = 2x_0$ $x_A = 2x_0$ $x_A = 2x_0$ $x_A = 2x_0$

(FY ZMc=(ZM)eff

BUT IN EQUILIBRIUM, X0=0 ZNE=0=-2 PA Sott WSINIS (2)

SUBSTITUTE (2) OTAL (1) OTAL (5) STUTITEBUE = x0/r, 8 = x0/r

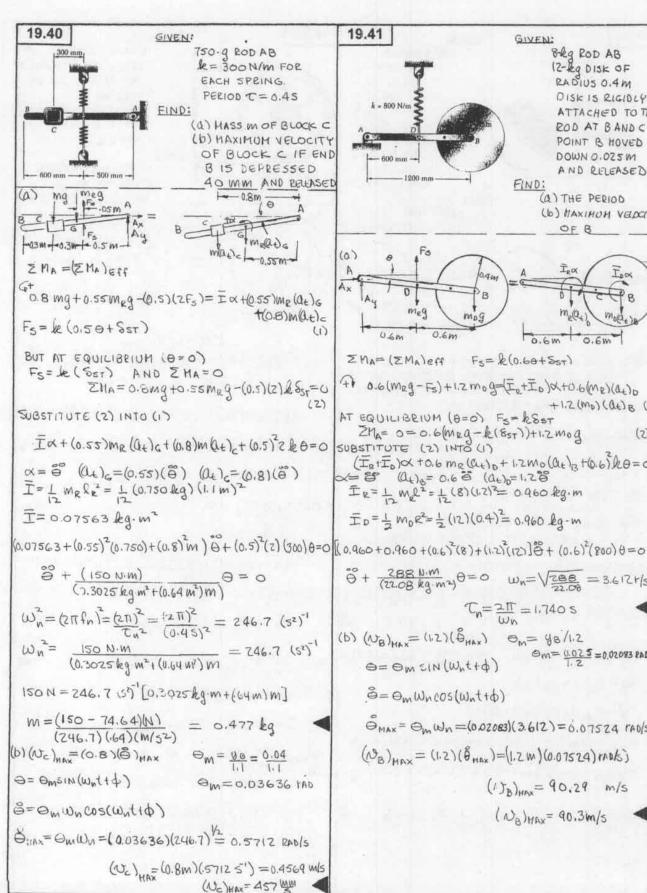
rmx° + I x° + 4rkx° = 0 I = 1 mr2
3 mrx° + 4rkx° = 0

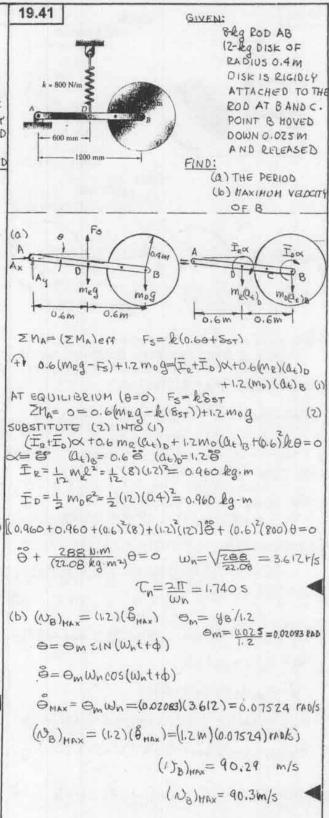
(b) $x_0 = (x_0)_m \sin(\omega_n t + \phi)$ Qt=0 $x_0 = \frac{2}{5}$ ft $x_0 = 0$

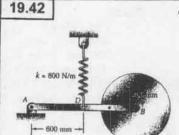
 $\tilde{x}_{0} = (x_{0})_{m} \omega_{n} \cos(\omega_{n} t + \phi)$, t = 0 , $0 = (x_{0})_{m} \omega_{n} \cos \phi$ t = 0 $x_{0}(0) = 1$ $f \in (x_{0})_{m} \sin \phi = (x_{0})_{m}(1)$

 $\mathcal{L}^{0} = -(k^{0})^{m} \mathcal{M}^{n}_{s} \geq \ln (m^{\nu} + \varphi)$ $(k^{0})^{m} = f \cdot f + \varphi$

(Qo)MAX=(Xo)MAX=-(Xo)MW =-(ft)(32.15-1)=171.7 ft/s2

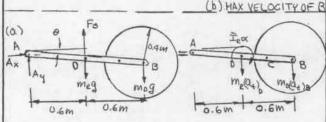






8kg ROD AB 12 kg DISK OF PADIUS 9.4 M PIN C REMOVED AND DISK CAN ROTATE FREELY ABOUT PIN B. POINT B HOVED DOWN D.UZSIM AND RECEASED

FIND: (a) PER100



NOTE: THIS PROBLEM IS THE SAME AS PROB 19.41 EXCEPT THAT THE DISK DOES NOT ROTATE. SO THAT THE EFFECTIVE HOHENT IN = 0. ZMA=(ZMWeff Fs=k(0,60+5=T)

A (0.6) (Meg-Fs)+1.2Mpg= I = x + (0.6) (me) (0.10) +1.3 (MD) (Ot) B (1) AT EQUILIBRIUM (0=0) FS=168ST

2 ZMA=0=0.6 (Meg-SST) + 1.2 Mbg SUBSTITUTE (2) INTO (1) (2)

 $I_{e} \times +0.6 \text{ mg}(a_{e})_{0}+1.2 \text{mp}(a_{t})_{0}+(0.6)^{2} \text{ LO} = 0$ $\times = 6 \text{ (a_{t})_{0}} = 0.66 \text{ (a_{t})_{0}} = 1.26 \text{ }$ $I_{e} = 1 \text{ mg} \text{ L}^{2} = 1 \text{ (e)}(1.2)^{2} = 0.960 \text{ leg m}$

[0.960+(0.6)2(8)+(1.2)212)] + (0.6)2(800) ==0

 $\Theta + \frac{(288 \text{ N·m})}{21.12 \text{ kg-m}^2} \Theta = 0 \quad W_M = \sqrt{\frac{288}{21.12}} = 3.693 \text{ kg}$

 $T_{N} = \frac{211}{W_{N}} = \frac{211}{3.693} = 1.7015$

(b) (UB) MAX (d) (d) $\Theta_{M} = \frac{48}{1.2} = \frac{.025}{1.20} = 0.02063$ and

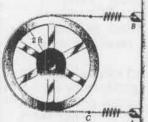
== Om SIN (Witto)

== 0mwn cos(wnt+ 4)

SMAX= OMWn= (0.02083)(3.(93)=0.07694 YADIS (b) MAX = (1.2)(B) = 0.2 X07694) = 0.9233 m

(NB) = 92.3 mm/s

19.43

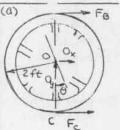


GIVEN:

600-16 FLYWHEEL OF PADIUS OF GYPATION=204 &= 75 lb/in, FOR EACH POINT C IS PULLED TO THE RIGHT I IM. AND RELEASED

FIND:

(a) PERIOD OF VIBRATION (b) HAXIMUM ANGULAE VELOCITY OF THE FLYWHOLD



ZMo=(ZNo)eff Fc=k((8st)-20) FB=k(20+(8st)B)

AV 2Fc-2FB=IB

2k[(8x)=20]-2k[(8x)a+20]=I0°

AT EQUILIBRIUM (0=0) Fo= & (Bst) , Fc= & (Bst)e

SUBSTITUTE (2) INTO (1)

T# + 8k0=0

 $\bar{T} = m \, \bar{k}^2 = \frac{(600 \, lb) (20/12 \, ft)^2}{(32.2 \, ft/s^2)} = 51.76 \, lb \, ft \cdot s^2$

&=(8+2)(75×12 16/ft)= 7200 16.ft

 $C_{\rm N} = \frac{2\pi}{\omega_{\rm N}} = \frac{2\pi}{\sqrt{139.1}} = 0.533 \text{ s}$

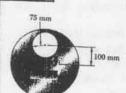
(b) 0=0msin(wnt+d)

= = = www cos(watto)

OMAX = Om Wn Om = Xc/r = 1/12 Wn= V139.1 = 11.79 EADIS ON= 0.04167 RAD

Smax = (0.04167)(11.79)= 0.491 rab/s

W = 0.491 rab/s

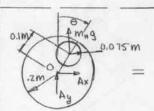


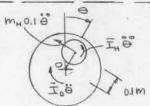
GIVEN:

DISK ATTACHED TO A
FRICTIONLESS PIN AT ITS
GEOMETRIC CENTER
AS SHOWN

FIND:

(a) PERIOD OF SHALL
OSCILLATIONS
(b) LENGTH OF A SIMPLE
PENDULUM OF THE
SAME PERIOD





ZMo=(ZMo)eff

 $(f - m_H g (0.1) SINO = I_0 O - I_H O - (0.1)^2 m_H O$ $m_b = S + \pi R^2 = (S + \pi)(.2)^2 + (0.04) \pi S + M_H = S + \pi R^2 = (S + \pi)(0.075)^2 + (0.005625) \pi S + M_H = S + \pi R^2 = (S + \pi)(0.075)^2 + (0.005625) \pi S + M_H = \frac{1}{2} m_D R^2 = \frac{1}{2} (0.005625) \pi S + M_H = \frac{1}{2} m_H R^2 = \frac{1}{2} m_H R$

(0.005625) WEE (9.81)(1) 0=0

727.9 ×10 0+ 5.518×10 0=0

 W_N^2 : $\frac{5.518 \times 10^{-3}}{727.9 \times 10^{-6}} = 7.581$

Wn= 2.753 RAD/S

 $T_N = \frac{2\pi}{\omega_N}$ $T_N = \frac{2\pi}{2.753} = 2.285$

(b) PERIOD OF A SIMPLE PENDULUM

 $T_{n} = 2\pi \sqrt{l/g}$ $l = (T_{n}/2\pi)^{2}g$ $l = [(2.753)/2\pi)^{\frac{3}{2}}(9.81 \text{ m/s}^{2})$ l = 1.294 m

19.45



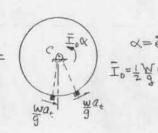
GIVEN:

WEIGHTS WAT A AND B, AND DISK W FOR B = 0, PERIOD = To

EIND:

ANGLE & FOR A PERIOD OF 200

Ay Con Ay B B



ZMc=(\SMc)eff

If $w + \sin(\beta - \theta) - \sin(\beta + \theta) = -2w + \sin \theta \cos \beta$

SINOZO at= rô

 $\omega_r = \sqrt{\frac{2\omega_0 \cos \beta}{2\omega + w/2}} = \sqrt{\frac{49\cos \beta}{4 + w/\omega}r}$ $\beta = 0 \quad C_0 = \frac{2\pi}{\omega_0} = 2\pi / \sqrt{\frac{49}{4 + w/\omega}r}$

 $C_{h} = 2\pi / \sqrt{\frac{\cos \beta}{(4+\frac{1}{16})r}} = 2C_{0} = 4\pi / \sqrt{\frac{49}{4+\frac{1}{16}(\omega)}r}$ $\cos \beta = \left(\frac{1}{2}\right)^{2} = \frac{1}{4}$

B=75.50

19.46 REFER TO FIGURE IN PROB 19.45

GIVEN:

w=0.116, W=316, r=4in., β=60°

FIND:

FREQUENCY OF SHALL OSCILLATIONS

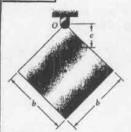
FROM DERIVATION IN DEOB 19.45 (EQ. 1)

 $\omega_{N} = \sqrt{\frac{249\cos\beta}{(24 + W/\omega)r}}$

 $W_{N} = \sqrt{\frac{(4)(37.2 \text{ ft/s}^2)\cos 60^{\circ}}{(4+3/0.1)(4/12)}} = 2.384 \text{ f/s}$

fn= Wn/2 T= 2.384/21T

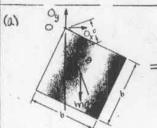
1'n= 0.379 Hz.

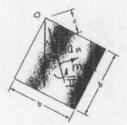


SQUARE PLATE, b=0.3M

FIND: (a) PERIOD OF SMALL

OSCILLATIONS ABOUT O (t) DISTANCE C FROMO TO A POINT A FROM 'N-ICH THE PLATE CHOULD BE SUSPENDED TO MINIMIZE THE PERIOD





$$\Sigma M_0 = (\Sigma M_0)_{eff}$$
 $\alpha = \emptyset$ $\alpha_L = (0G)(\alpha)$

$$\overline{L} = \frac{1}{6} m b^2$$

$$\alpha = \emptyset$$

$$\alpha_L = (0G)(\alpha)$$

$$\alpha_L = (0G)(\alpha)$$

$$\alpha_L = (0G)(\alpha)$$

4) (OSKSINO) (mg) = -(OS) mat - IX SINORO brz/zm (brz/z) + + m b + (brz/z)mg 0=0 (b)(=+t)m0+ 12/2 mg0=0 0+ (12/2) 9 0=0 b=0.3m

$$(T_{N})_{0} = \frac{2\pi}{(\omega_{N})_{0}} = 2\pi \sqrt{\frac{(2/3)b}{(\sqrt{2}b)g}} = 2\pi \sqrt{\frac{4(3m)}{3\sqrt{2}(98)g}}$$

(Cn) = 1.0675

(b) SUSPENDED ABOUT A

ZMA=(EMA)ess at=(06-c)x

A (06-c)(SINB) (Mg)=-(06-C) max-IX

((b 15/2-c)2++ b2) m8+(c12-c)mg 0=0

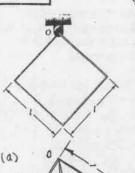
 $(C_n)_A^2 = \frac{(2\pi)^2}{\omega_n^2} = \frac{4\pi^2 \left[(b R_1/2 - c)^2 + b^2/6 \right]}{(b R_2 - c)}$

FOR HINIHOM DERIOD dC2/a = 0 0= 2(p12/2-c)(-1)(p12-c)-(-1)[(p12/2-c)+16/6]

(br/2-c)2+ 62/6=0 b=0.3m

briz-c= b = 0.08966m C= 89.7 MM

19.48

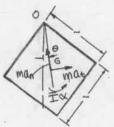


GIVEN:

THIN WIRE, L=1.2 ft FIND:

(a) PERIOD ABOUT O (6) PERIOD ABOUT A POINT AT THE HIDPOINT OF ONE OF THE

(a)



M=MASS OF THE FRAME

ZNo=(ZMo)eff X=0 Q=(OG)(X)

2 6' + m/4

06= 2 12/2. Q=(1 12/2) 0

FOR ONE LEG (IG)= IG+ (MA) (R/Z)2 (工品)=脚門卡子(馬)=脚見?(男)

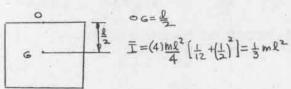
FOR COMPLETE WIRE FRAME

I = 4 (I6),=(4) M 22(3) = \frac{1}{3} m 22

G-Mglasino=IX+(MGt)/2 SINOXO (3+1)ml20+ mgl [2/20=0

 $T_{N} = \frac{2\Pi}{\omega_{N}} = \frac{2\Pi}{\sqrt{\frac{\sqrt{2/2} \cdot 3}{(5/6) \cdot 2}}} = 2\Pi \sqrt{\frac{5(1.2 \text{ ft})}{(3\sqrt{2})(32.2 \text{ ft}(5^{2})}}$ TA= 1,3175

(b) FOR FRAME SUSPENDED FROM MIDPOINT



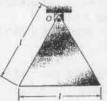
- mg & sino = IB+ (m&B) & = (3+4) ml20 I m 1200 + mg & 0 = 0

 $C_N = \frac{2\pi}{\omega_N} = \frac{2\pi}{\sqrt{\frac{6}{6}}} = 2\pi \sqrt{\frac{7}{6}} \left(\frac{12 \text{ ft}}{322 \text{ ft/s}^2} \right) = 1.310 \text{ s}$

Tn= 1.310 S



UNIFORM EQUILATERAL
TRIANGLE OF SIDE L=0.3 M



FIND:

(a) PERIOD IF PLATE IS

SUSPENDED FROM ONE

OF ITS VERTICES

OF ITS VERTICES
(b) PERIOD IF PLATE IS
SUSPENDED FROM THE
MIDPOINT OF ONE OF ITS
SIDES

19.50

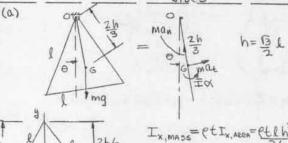
B 0 r - 250 mm

GIYEN:

ROD AB OF NEGLIGIBLE
MASS ATTACHED TO A DISK
OF MASS M. AB=1=0.650M
V= 0.250 M

FIND:

THE PERIOD OF SMALL
OSCILLATIONS IF
(a) THE DISK IS FREE TO
ROTATE IN A BEARING AT A
(b) THE DISK IS RIVETED AT A



h e 2h/3

 $I_{x,MASS} = (tI_{x,AREA} = (tI)^3$ M = StA = St Ib

 $T_{x, mass} = mh^{2}$

Iy, mads = 8tIy, area Iy, $nrea = \frac{h\ell^3}{48}$

 $T_{y, MASS} = \frac{m\ell^2}{24}$

 $I = I = I_x + I_y = \frac{mh^2 + ml^2}{18}$ $h = 0.5/2 \qquad I = mn^2 [3/4 + 1] = mn^2 [3/4 + 1]$

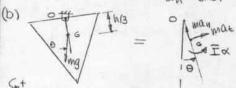
 $h = \ell \sqrt{3}/2 \qquad \overline{L} = M\ell^2 \left[\frac{3/4}{18} + \frac{L}{24} \right] = M\ell^2/12$ $\alpha = \Theta \qquad \alpha_{\pm} = \frac{13}{3}\ell\Theta \qquad \qquad \text{SINO} \approx \Theta$

Ct ZMo= (ZMo)eff -mg 13 1 SINO= IX+13 2 mat

(12 + 13) ml20 + mg 13/3 l 0 = 0

 $\omega_{n}^{2} = \frac{\sqrt{3}/3}{5/12} \frac{9}{9} = (\sqrt{3})(4) \frac{(9.81 \text{ m/s}^{2})}{5} = 45.31 \text{ s}^{2}$

 $W_{n} = 6.731 \text{ r/s}$ $T_{n} = \frac{2\pi}{\omega_{N}} = \frac{2\pi}{6.731} = 0.933 \text{ s}$



七百年夏里=0

 $\omega_{N}^{2} = \sqrt{3} \frac{9}{2} = \sqrt{3} \frac{9100 \text{ M/s}^{2}}{0.300} = 56.635^{-1} \quad \omega_{N} = 7.5258 \text{ Ms}$

 $T_{N} = \frac{2\pi}{\omega_{N}} = \frac{2\pi}{2756} = 0.8355$



l=0,650m el man r=0.250m

 $I = \frac{1}{2}mr^2 = \frac{1}{2}(0.250)^2 m = \frac{m}{32}$ $A = 1 = 1 = 0.650 \times x = \frac{1}{2}$

(A) THE DISK IS FREE TO ROTATE AND IS IN CURVILINEAR TRANSLATION
THUS IX=0

EMB=(ZHB)eff

(t - mglsine = lmat sine ≈ 0) $ml^2 ≈ + mgle = 0$

 $W_{N}^{2} = \frac{9}{L} = \frac{9.81 \text{ m/s}^{2}}{0.650 \text{ m}} = 15.092$ $W_{N} = 3.885$ $C_{N} = \frac{2\pi}{W_{N}} = \frac{2\pi}{3.885} = 1.6175$

(b) WHEN THE DISK IS RIVETED AT A, IT POTATES AT AN ANGULAR ACCELERATION &

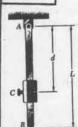
ZMB=Z(MB)eff

(\$ -mglsin0 = Ix+ lmat I = 1 mr2 (1 mr2+ ml2) = + mgl0 = 0

 $\omega_{N}^{2} = \frac{9 l}{(F^{2}/2 + l)^{2})} = \frac{(9.81 \text{ m/s}^{2})(0.650 \text{ m})}{[(0.250^{2})/z + (0.650)^{2}]} = [4.053 \, 5^{2}]$

Wn= 3.749 Hs

 $T_{N} = \frac{2\pi}{\omega_{N}} = \frac{2\pi}{3.749} = 1.676 S$

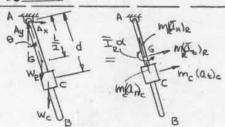


GIVEN:

COLLAR C WEIGHT, WE= 216 ROD AB WEIGHT, WE= 616, L= 3ft

FIND:

PERIOD OF SHALL OSCILLATIONS
WHEN,
(a) d= 3 ft
(b) d= 2 ft



EMA = (EMA)eff

$$\frac{8 + (\frac{1}{2} + \frac{1}{100}) g}{(\frac{12}{3} + \frac{1}{100} \frac{1}{6})} \theta = 0$$

$$\Theta + \frac{(3+\frac{1}{3}d^2)}{(3+\frac{1}{3}d^2)} \Theta = 0$$

$$T_{\rm N} = 2\pi / \omega_{\rm N} = 2\pi \sqrt{\frac{(3+d^2/3)}{\frac{3}{2}+d/3}(q)}$$

$$T_n = 2\pi \sqrt{\frac{(3+3)}{(\frac{3}{2}+1)(32.2)}} = 1.715 \text{ s.}$$

$$T_N = 2\pi \sqrt{\frac{(3+4/3)}{(3/2+2/3)(37.2)}} = 1.566 \text{ S}$$

19.52



GIVEN:

COMPOUND PENDULUM WHICH OSCILLATES ABOUT O RECENTRODAL PADIUS OF GYRATION GA = \mathbb{R}^2/\mathbb{F}

SHOW THAT:

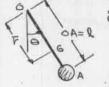
OF LENGTH OA.





t) $\geq N_0 = Z$ (N_0) eff: -WFSIND = IK + MOLF $-MgFSIND = ML^2 + MT^2 + MT$

FOR A SIMPLE PENDULUM OF LENGTH OA= &



 $\ddot{\Theta} + \dot{Q} \Theta = 0 \qquad (2)$

COMPARING EQUATIONS (1) AND (2) $= \overline{F^2 + \overline{A}^2}$

GA= 1-F= 12/F (QED)

19.53 | GIVEN;

COMPOUND PENDULUM AS IN PROB. 19.52 SHOWN ABOVE

SHOW THAT:

SHALLEST PERIOD OF COCILLATION OCCURS

SEE SOLUTION TO PROBIG. 52 FOR DERIVATION OF 6+ 9F FRATE

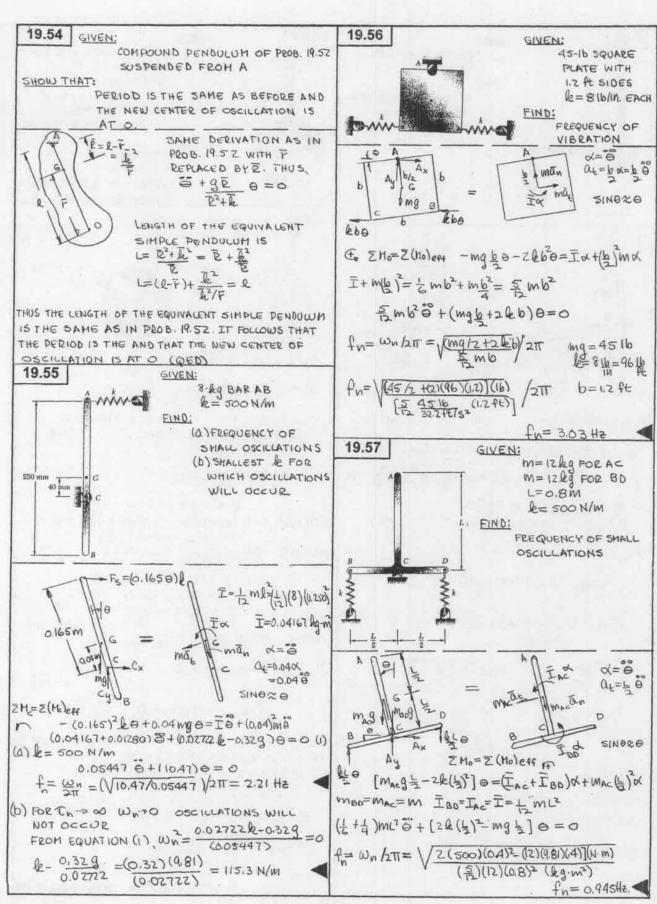
FOR SHALL OSCILLATIONS SINDRO AND

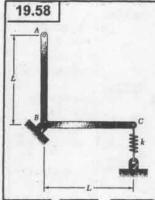
The BILL STILL STIL

FOR SHALLEST TO WE HUST HAVE TIL

$$\frac{d\left(\overline{r}+\frac{\overline{k}^{2}}{\overline{r}}\right)}{d\overline{r}} = \left|-\frac{\overline{k}^{2}}{\overline{r}^{2}}\right| = 0$$

F= R GED)





ROD ABC OF TOTAL MASS M

FREQUENCY OF SHALL OSCILLATIONS IN TERMS OF MIL AND



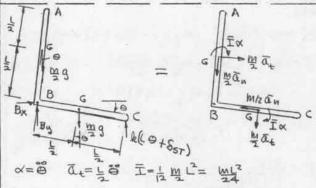
19.59

GIVEN:

ROB AB LENGTH Q= 0.650 M HASS OF AB IS NEGLIGIBLE AB IS DISPLACED 20 FROM THE POSITION SHOWN AND RELEASED

FIND:

MAXIMUM VELOCITY OF A IF THE DISK IS. (A) FREE TO ROTATE ABOUT A (b) PIVETED TO AB AT A



A ZMB=Z(HB)eff

SINOZO COSOZI

mg = SINO+mg = (050- L (L+5+SsT)(050= 2 10+2ma, L MgL 0+mgL - kl20-kl28st=ml28+ml28 (1)

BUT FOR EQUILIBRIUM (0=0) ZMB=0=mgh-kl2SsT

(2)

SUBSTITUTE (2) INTO (1)

00 + (lel2-mg/4)0=0 m173

W= 3k - 39 Wn= 13 / 1 - 9

fu= Wulatt

fn= 13 / 1 - 9



1=0.650IN 0 =0.250 M

 $\bar{\mathbf{I}} = \bar{\mathbf{I}}^2 \mathbf{W} \mathbf{L}_S = \bar{\mathbf{I}}^2 (0.520)_S \mathbf{W} = \frac{35}{M}$ Q= 2x=0.650 0

(a) THE DISK IS FREE TO ROTATE AND IS IN CURVILINEAR TRANSLATION. THUS IX=0 ZMB=Z(HB)eff

F - mglsino = lmax

ml25+mgl0=0 w= 9 FROM 19.17 THE SOLUTION TO THIS EQUATION IS e= em sin (watto)

At t=0, 0= 2. II = II PAD, 8=0

8 = Omwncos(wnt+0) 0= 0 www cos \$ \$ = I II = OM SIN(0+II)

OM = II PAD

THUS O= I SINWALL]

(UA) HAX = l & HAX = LOMWN L= 0.650 M

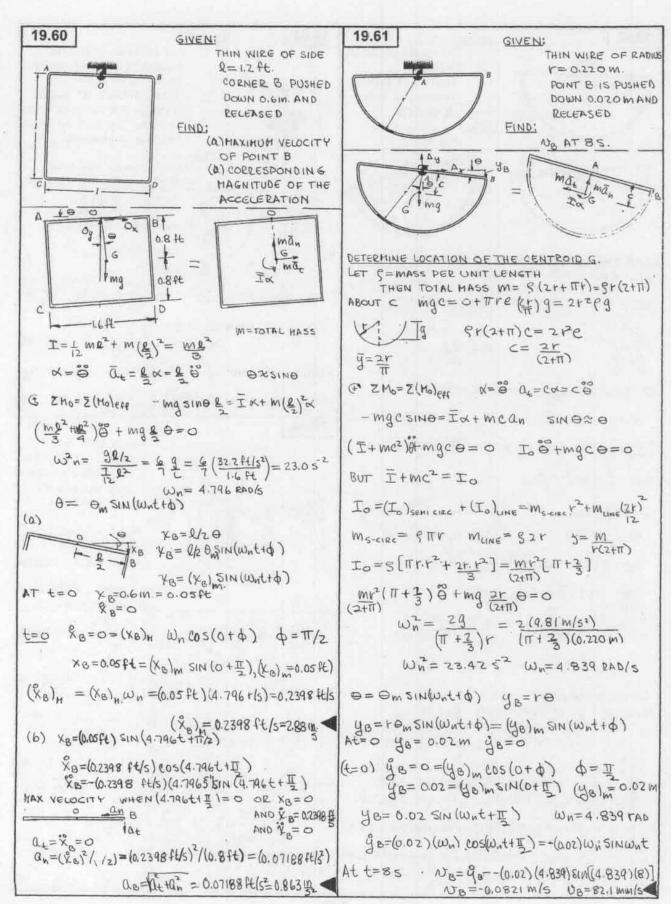
(NA) HAX = (0.650 M) (1) (19.81M/52)

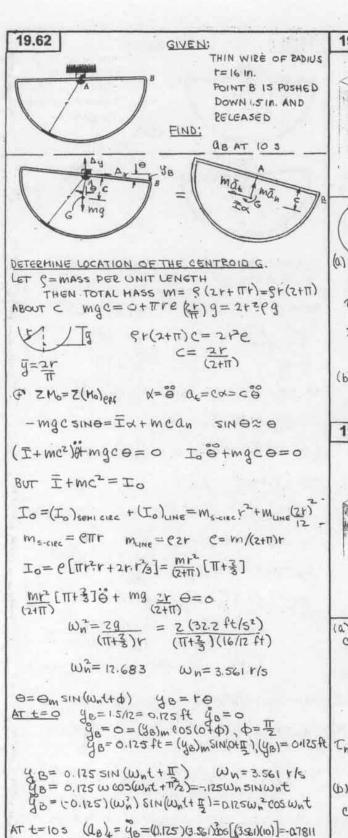
(NA) HAX = 0.08815 M/S MM 1.88 = XAH (AU) b) FOR DISK RIVETED AT A (IX INCLUDED) (7 m rs+m rs) Q+md r0 = 0 ZHO=Z(Ho)epf G-mglsino=IX+ CMQ+

= I SIN(wnt+ I) (SEE (a))

(NA) MAX = I BMAX = LOMWN

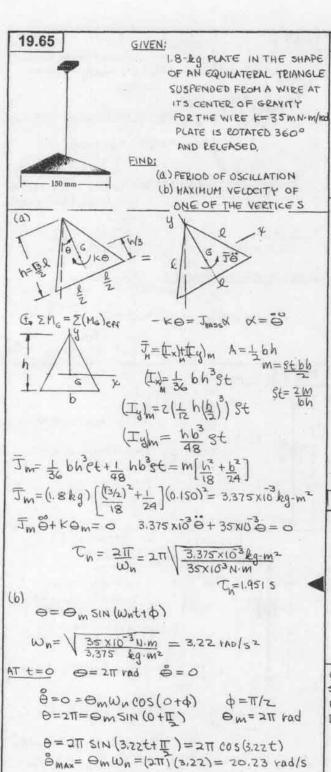
(UA) MAX= (0.650 M) (9.81 M/S) (0.650 M) = 0.0851 M/S (0,200/2 + 6.650 2) M2 (UA)MAX = 85.1 MM





 $0.8 = \left[(0.0)_{6}^{2} + \frac{0.0}{2} \right]^{\frac{1}{2}} = \left[(0.7811)_{7}^{2} + \left[\frac{(0.3874)_{7}^{2}}{(16/12)} \right] = 0.789 \text{ ft/s}^{2}.$

19.63 GIVEN: DISK OF RADIUS F= 120 MM IS WELDED TO POD AB WHICH IS FIXED AT A AND B. DISK POTATES BO WHEN A 500-MN·M IS APPLIED PERIOD THE 1.35 WHEN THE COUPLE IS BEHOVED (A) THE MASS OF THE DISK (b) PERIOD IF ONE ROD IS REMOVED K= I = 0.5 NM (8)(T/180) K=3.581 N.W/120 ZMo=Z(Mo)eff 0+ K0=0 KO=JÖ (211)2 (1.35)2(35810 m/r) C"= 部"= JU/K J= 0.1533 N.m.s2 = 1mr=1m (0.120m)2 m=(0.1533 N·m·52)(2) = 21.3 kg (0,120 m) (b) WITH ONE BOD REMOVED K'= K/2=3.581 =1.791 U.M C= 211 / 1/K1 = 211 / (0.1293 N.m.2) = 1.8382 19.64 GIVEN: 10-16 ROD CD OF LENGTH 1=2,2ft WELDED ROD FIXED AT. A AND B WITH K= 18 lb.ft/tad FIND: PERIOD OF SHALL OSCILLATIONS IF THE EQUILIBRIUM POSITION (a) VERTICAL AS SHOWN LATHOSIAOH (9) A ZM = E(Mc)eff - KO - Md ZINO=24+MTg x= & at= & x= & 6 (J+m/2) 18 HK+mge/2/8=0 J+W(8/2=Jc=1 ml2 (1016) (2.241) = 0,501 lbfts (32.2 ft/52) Th= 211 / JE (K+11942 TN= 211 VIBHOULD 16452 Tu=0.8265 man 1JX ×=0 E ZMc=Z(Hc)eff - K(0+0st)+mylk = Jo+m(l/2) = Joo (UB) = 4B=(0.125)(3561) 51N(3 X61)(10) =0.3874 ft/ BUT IN EQUILIBRIUM (0=0) ZMC=0=-KOST+Mg42 THUS JOB+KO=0 TN=211 = 211 \ 1/k = 211 \ 0.501 | b.f652 = 1.048 \$

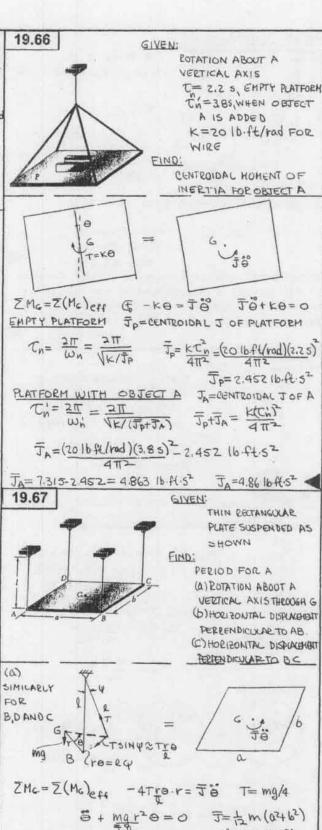


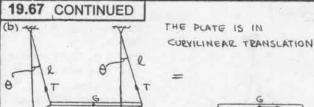
 $t = \frac{h13}{51N300} = 2h/3 = f3/3Q$

(UA) = FOHAX

(Ua) = (13/3)(0.150m)(20.23 rod/s)

(NA) HAX = 1.752 M/S





B,A mg SINDAD COSDAI

LÖ COSDA & LÖ = TL

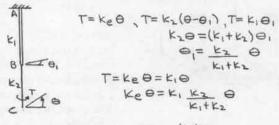
+ 12E=0=4 (TCOSO)-mg=0 T= mg/4

+ → Z = Z(Fn)eff -4T SINO = M ā 28+90=0 Tn=2TV\$

(C) SINCE THE OSCILLATION ABOUT AXES PARALLEL TO AB (AND CD) IS INDEPENDENT OF THE LENGTH OF THE SIDES OF THE DUATE, THE PERIOD OF YIBRATION ABOUT AXES PARALLEL TO BC (AND AD) IS THE SAME TO BE TO BE THE SAME

19.68 GIVEN: 2.2-kg CIRCULAR DISK Y=0.8m WIRE AB, K_1=10N·m/rad WIRE BC, K_2=5N·m/rod FIND: PERIOD OF OSCILLATION ABOUT AXIS AC

EQUIVALENT TORSIONAL SPRING CONSTANT



Ke = K1 K2

NEWTONS LAW $k_{e\theta} = 10$ $z_{H_c} = z_{H_c} = z_{H_c$

 $C_{N} = \frac{m_{N}}{2\pi} = \sqrt{\frac{m_{L_{3}}}{2 \text{ Fe}}} = 511 \sqrt{\frac{5[(10)(2)/(10+2)]N \cdot M}{(2.5 \text{ Fd})(0.8 \text{ m})^{2}}}$ $C_{N} = \frac{m_{N}}{2\pi} = \sqrt{\frac{m_{L_{3}}}{2 \text{ Fe}}} = 211 \sqrt{\frac{5[(10)(2)/(10+2)]N \cdot M}{(2.5 \text{ Fd})(0.8 \text{ m})^{2}}}$

19.69

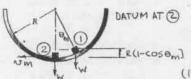


GIVEN:

PARTICLE WHICH HOYES WITHOUT FRICTION INSIDE A CURYED SURFACE

FIND:

PERIOD OF SHALL OSCILLATIONS



POSITION()

Ti=0

Vi=WR(1-cos@m)

SMALL OSCILLATIONS

(1-cos@m)=251129_=21/2

VI=WEOZ

 $\frac{Posifion @}{V_m = R@m} T_2 = \frac{1}{2} m V_m^2 =$

 $_{M}=R\theta_{M}$ $T_{2}=\frac{1}{2}MN_{M}^{2}=\frac{1}{2}MR^{2}\theta_{M}^{2}$ $V_{2}=0$

CONSERVATION OF ENERGY TITY = T2+V2

O+ WREIM = IME BM + O BM=WADM

Mg R = IM R WADM

W= MG

W= VG/R

Wn= V9/R Tn= 211 = 211 VP Wn

19.70 GIVEN:

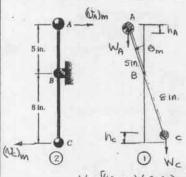


1402. SPHERE A 1002. SPHERE C ROD AC OF NEGUGIBLE WEIGHT

FIND:

PERIOD OF SMALL OSCILLATIONS

OF THE ROD



DATUM AT (1)

POSITION(1)

TI = 0

VI = Who-WAHA

HC = BC (I- COSOM)

HA = BA (I- COSOM)

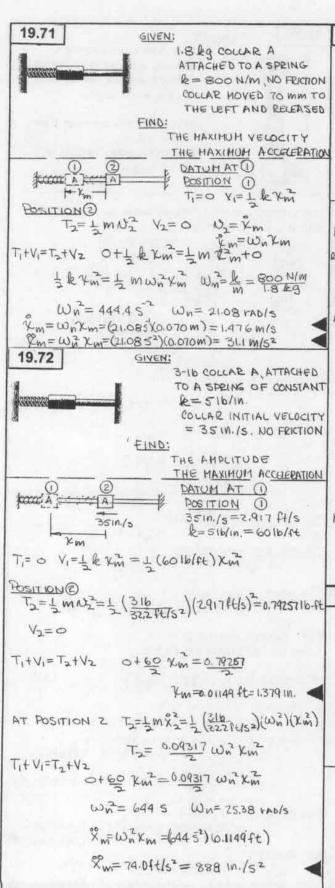
SHALL ANGLES

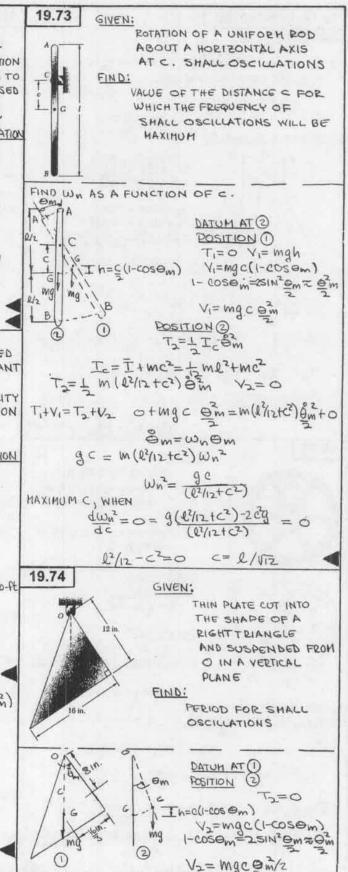
I-COSOM & OM/2

VI-[WC)(BC)-(WARA))

ハー(101P)(をけ)-(119p)(をけ)) るが

 $V_{1} = (0.4167 - 0.3646) \underbrace{0}_{m}^{m} = 0.05208 \underbrace{0}_{m}^{m}$ $V_{2} = 0 \quad T_{2} = \underbrace{1}_{m} \underbrace{m_{0}} \underbrace{(0)_{m}^{m}} + \underbrace{1}_{m} \underbrace{m_{0}} \underbrace{(v_{0})_{m}^{m}}, \underbrace{(v_{0})_{m}^{m}} = \underbrace{2}_{m}^{m} \underbrace{0}_{m}^{m}$ $T_{2} = \underbrace{1}_{2} \underbrace{m_{0}} \underbrace{(0)_{1}^{m}} \underbrace{(0)_{1}^{m}} + \underbrace{1}_{2} \underbrace{m_{0}} \underbrace{(0)_{1}^{m}} \underbrace{(0)_{1}^{m}} \underbrace{0}_{m}^{m} \underbrace{0$

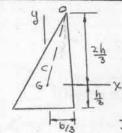




19.74 CONTINUED

POSITION (1) $T_{1} = \frac{1}{2} J_{0} \stackrel{\circ}{\Theta}_{m} \quad V_{1} = 0$ CONSERVATION OF ENERGY $T_{1} + V_{1} = T_{2} + V_{2}$ $1 J_{0} \stackrel{\circ}{\Theta}^{2} + 0 = 0 + mg c \stackrel{\circ}{\Theta}_{m}^{2}$ $1 J_{0} \stackrel{\circ}{\Theta}^{2} = mg c \stackrel{\circ}{\Theta}_{m}^{2}$

DETERMINE CAND TO



$$c = \left[\frac{(2b)^{2}}{3} + \frac{(b)^{2}}{3} \right]^{\frac{1}{2}}$$

$$c = \frac{1}{3} \left[4h^{2} + b^{2} \right]^{\frac{1}{2}}$$

$$h = 12 \text{ in} = 1 \text{ ft}$$

$$b = 16 \text{ in} = 4/3 \text{ ft}$$

$$C = \left[4(1)^{2} + \frac{(4/3)^{2}}{3} \right]^{\frac{1}{2}} = \frac{\sqrt{52}}{9}$$

$$J_{0} = J + \text{in} C^{2}$$

 $\overline{J} = St[\overline{I}_x + \overline{I}_y] = St[\frac{1}{36}bh^3 + \frac{1}{36}hb^3]$ $m = St + \frac{1}{2}bh$

 $2f = \frac{\rho N}{3m}$ $\underline{J} = \frac{18}{M} [N_3 + \rho_5] = \frac{18}{M} [1 + (4/3)] = \frac{165}{522} M$

 $T_0 = \frac{25}{162}M + \frac{52}{81}M = \frac{129}{162}M$ 16.ft. S^2

 $\omega_{N}^{2} = \frac{mgc}{J_{0}} = \frac{m(32.2)(\frac{152}{9})}{m(129/162)} = 32.4 \, \text{s}^{2} \quad \omega_{N} = 5.592 \, \text{g}$

 $T_N = \frac{2\Pi}{\omega_N} = \frac{2\Pi}{5.592} = 1.1045$

19.75



GIVEN:

85-16 FLYWHEEL PERIOD = 1.265 FOR SMALL OSCILLATIONS

FIND:

CENTROIDAL MOMENT OF



POSITION O T=1J. & Y=0

Position (2) $T_2 = 0 \quad V_2 = mgh$ $h = r(1 - cose_m) = r 2 sin^2 e_m/2$ $\approx r e_m^2 / 2$ $V_2 = mg r e_m^2 / 2$

CONSERVATION OF ENERGY

TITYI=TZTYZ 1 0 0 m+ 0= 0+mg rom/z

êm=wnom Townon=mgron

 $W_{N}^{2} = \frac{Mgr}{J_{0}}$ $C_{N}^{2} = \frac{4\pi^{2}}{\omega_{N}^{2}} = \frac{(4\pi^{2})J_{0}}{Mgr}$

 $J_0 = \overline{J} + Mr^2 = \overline{J} + M$

 $\overline{J} = \frac{(C_{1}^{2})(Mgr) - Mr^{2}}{4\pi^{2}} = \frac{(1.265)^{2}(8516)(724)}{(8516)(724)} \frac{(8516)(724)}{(8224)5} \frac{(6)}{(8)} = 0$

J=1.994-0.8983=1.09616.4652

19.76



GIVEN:

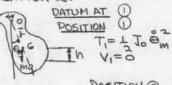
FOR SHALL OSCILLATIONS PERIOD ABOUT A = T_A =0.8955 PERIOD ABOUT B= T_B =0.8055 T_A + T_B =0.270 M

FIND:

(a) LOCATION OF THE MASS CENTER G (b) CENTROIDAL RADIUS OF GYRATION E.

CONSIDER GENERAL PENDULUM OF CENTROIDAL PADIUS OF GYRATION &.





 $\begin{array}{cccc}
& & & & & & & & \\
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 & & & & & & \\
\hline
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& & & & & \\
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 & & & & \\
V_2 = mg & & & & \\
\hline
 & & & & \\
V_2 = mg & & & & \\
\hline
 & & & &$

 $T_1+V_1=T_2+V_2$ $\frac{1}{2}J_0\tilde{\Theta}_M+O=O+\frac{1}{2}Mg\bar{F}\tilde{\Theta}_M^2$ $\tilde{\Theta}_M=W_0\Theta_M$

Jo= J+mF2= mk2+mF2

FOR THE ROD SUSPENDED AT A $C_n = 0.89$ $S = 2\pi \sqrt{\frac{R^2 + F^2}{9F}}$ $F = r_0$ (1)

FOR THE ROD SUSPENDED AT B $T_N = 0.805 S = \sqrt{\frac{12+12}{415}} F = V_0 \qquad (2)$

But $r_0 + r_0 = 0.270 \text{ m}$ (3) FROM (1) $r_0^2 + r_0^2 = g r_0 (0.895)^2$ (1')

FROM (2) R+10= 9 10 (0.805)2 (2')

SUBTRACING (2') FROM (1')

12-152= (9/4π2) (0.8017a-0.648 Tb) (4) DIVIDING (4) BY (3) MEMBER BY HEMBER

ra-rb= 10.270 (9/4172) (0.80170-0.64816)

 $r_a - r_b = \frac{9.81/4\pi^2}{0.270} (0.801 r_b - 0.648 r_b) = 0.7372 r_b - 0.5963 r_b = 0.6510 r_a$ (5)

5085TITUTE FOR 16 FROM (5) INTO (3)

10-10.6510 10-0,270 10-0.1635 M

16) FROM (1')

162 = (9.81)(0.1635)(0.895/2π)² - (0.1635)²

162 = 0.03254 - 0.02673 = 0.05812 m²

163 = 0.03254 - 0.02673 = 0.05812 m²

164 = 0.03254 - 0.02673 = 0.05812 m²

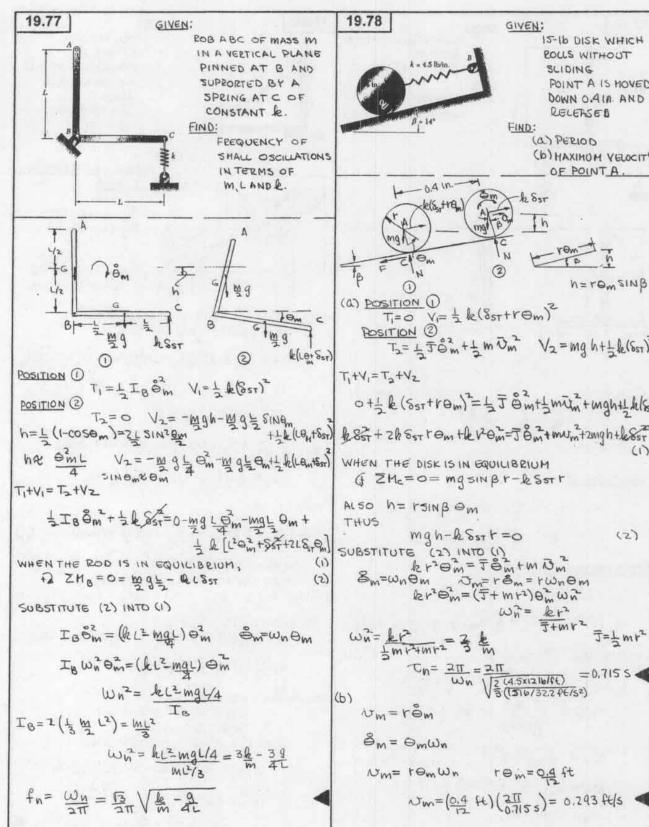
165 = 0.03254 - 0.02673 = 0.05812 m²

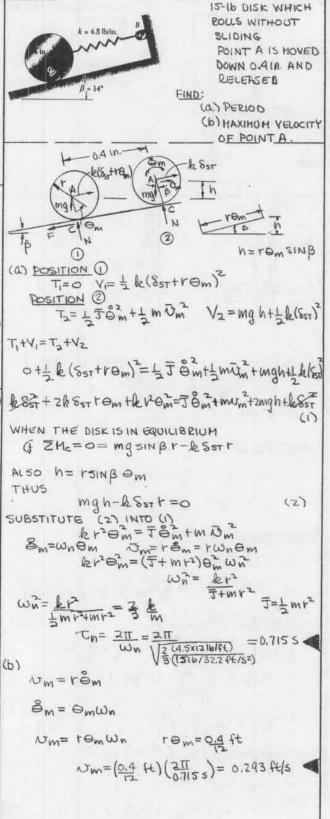
165 = 0.03254 - 0.02673 = 0.05812 m²

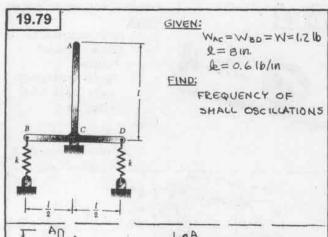
167 = 0.03254 - 0.02673 = 0.05812 m²

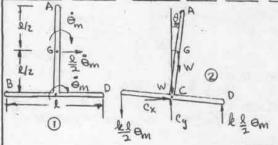
168 = 0.03254 - 0.02673 = 0.05812 m²

169 = 0.03254 - 0.02673 = 0.05812 m²









Position () $1-\cos\theta = 2\sin\theta = \frac{2}{2}m$ $T_1 = 2(\frac{1}{2} \int_{-\infty}^{\infty}) + \frac{1}{2} m(\frac{1}{2} \frac{2}{6})^2$ $V_1 = 0$

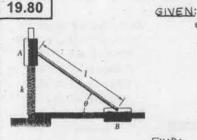
POSITION @

 $T_2 = 0$ $V_2 = -W_{\frac{1}{2}} (1 - \cos \frac{1}{m}) + \frac{1}{2} k (\frac{1}{2} \theta_m)^2$ $V_2 = -\frac{W_{\frac{1}{2}}}{2} \frac{\theta_2^2 m}{4} + \frac{1}{4} k (\frac{1}{2} \theta_m)^2$

CONSERVATION OF ENERGY

 $T_{1}+V_{1}=T_{2}+Vz$ $\frac{1}{2}(z\overline{J})\mathring{\theta}_{m}^{2}+\frac{1}{2}M\frac{d^{2}\theta_{m}^{2}}{4}\theta_{m}+0=0-\underline{\omega}L\frac{\theta_{m}^{2}}{2}+\underline{U}^{2}\theta_{m}^{2}$ $\mathring{\theta}_{m}=\omega_{n}\theta_{m}\quad \overline{J}=\frac{1}{12}\underline{\omega}g$ $(\underline{\omega}g+\underline{\omega}g)Q^{2}\omega_{n}^{2}\theta_{m}^{2}=(-\underline{\omega}L+\underline{U}^{2})\theta_{m}^{2}$ $\underline{\omega}_{n}^{2}=-\underline{\underline{\omega}}+\underline{U}^{2}\underline{Q}=\underline{G}(-\underline{q}+\underline{U}^{2})\theta_{m}^{2}$ $\underline{\omega}_{n}^{2}=-\underline{\underline{\omega}}+\underline{U}^{2}\underline{Q}=\underline{G}(-\underline{q}+\underline{U}^{2})\theta_{m}^{2}$ $\underline{\omega}_{n}^{2}=-\underline{\underline{\omega}}(-\underline{3}z,z,\underline{f}Us^{2})+(0.6\times12.16/f_{E})$ $\underline{\omega}_{n}^{2}=-\underline{\underline{G}}(-\underline{3}z,z,\underline{f}Us^{2})+(0.6\times12.16/f_{E})$ $\underline{\omega}_{n}^{2}=-\underline{\underline{G}}(-\underline{4}8,3+193.2)=173.9$ $\underline{\omega}_{n}=13.19 \text{ rad/5}$

fn= Wn = 13.19 = 2.10 Hz.



BLE ROD AB
L= 0.6 M
COLLARS A AND B
OF NEGLIGIBLE
HASS
L= 1.7 LN/M
0= 40° AT
EQUILIBRIUM

Symbol of VIBRATION

VERTICAL ROD

YELSING

Sy = $1 \cos \theta \cos \theta$ Sig =

POSITION (MAXIMUM VELOCITY SEM)

TI = 1 TISEM + 1 W(SE)2+(SE) T

TI = 1 (12 ml2)(SE) + 1 m[(2 SINE)2+(2006)](SE) T

TI = 1 ml2[12+4](SEM)2=1 ml2 (SEM)2

VI = 1 & (85T)2+ mg T

POSITION (ZERO YELOCITY, HAXIMUM SOM)

T3=0

V2= 1/2 (Syt8st)2+mg(y-Sq)

TitVi=Tz+Vz

1/2 ml+1/2 [60] +1/2 f Ssr + mgy = 0+1/2 f Sy + 6st) + mg(y-Sq)

ml2/3 (80) +1/2 f Ssr + mgy = k(Sy+2 Sy Ssr + Ssr) + mg(y-Sy)

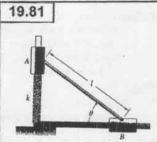
BUT WHEN THE ROD IS IN EQUILIBRIUM,

GZM8= mg x - kSsr x = 0 mg=2 f Ssr (2)

SUBSTITUTE (2) INTO (1)

MEZ J(SOM) = RSYM SYM=1 COSO SOM

MZZ (SOM) LLCOSO (SOM) Z

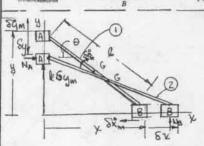


L= 0.6 m ma=mo=m= 8 kg k= 1.7 kn/m 0= 40° 200 AB OF

ROD AB OF NEGLIGIBLE HASS

EIND:

PERIOD OF VIBRATION



HORIZONTAL POD Y = 1 SIN 0 8y = 1 COS 0 80 80 = 1 COS 0 X = 1 COS 0 8X = -1 SIN 0 80 8X - 1 SIN 0 80

POSITION® (MAXIMUM VELOCITY, S&m) $T_1 = \frac{1}{2} m (S_{m}^2)^2 + \frac{1}{2} m (S_{m}^2)^2$ $T_1 = \frac{1}{2} m [(1050)^2 + [15100)^2] (S_{m}^2)^2$ $T_1 = \frac{1}{2} m 1^2 (S_{m}^2)^2$ $V_1 = 0$

(OS MUXIXAM, YTIDORY 0835) (NOITIZOR

V2 = 2 h Sym

 $T_1 + V_1 = T_2 + V_2$

= ml2 (Som) = 0= = 2 h Sym

Sym= 1 cose Som

ml2(80m)= kl2cos20(80m)2

SIMPLE HAPHONIC HOTION

SO = SOM SIN (WINTHO)

Sốm=SOMWn

ml2(80m) Wn2 = Ll2cos20 (50m)2

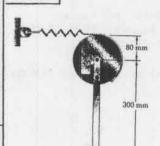
Wn= le coszo

Wn2 = 1200 N/M cos240 = 88.02 5-2

 $C_{N} = \frac{2\pi}{\omega_{N}} = \frac{2\pi}{\sqrt{8802}} = 0.66915$

Cn= 0.670 s.

19.82

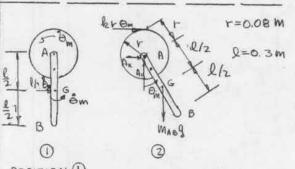


GIVEN:

MAB = 3 kg MDISK = 5 kg SPEING IS UNSTRETCHED IN THE POSITION SHOWN &= 280 M

FIND:

PERIOD OF SHALL



 $\frac{\text{Position} \ \widehat{\mathbb{I}}}{T_{i} = \frac{1}{2}} \overline{\mathbb{I}}_{\text{Disk}} \overset{\circ}{\Theta}_{\text{M}}^{\text{N}} + \frac{1}{2} (T_{\text{N}})_{\text{ROD}} \overset{\circ}{\Theta}_{\text{M}}^{\text{N}}$

V1=0

JOISK = 1 MOF2 (TAROO = 1 MAG)

Position @ Tz=0

V2= 1 & (rem)2+mag & (1-cos om)

1-cose m=2 SIN2 OM & OM

V2= = 1 le +0 = + mag(2)0 m

T1+V1=T2+V2

= (= mor2+3mapl2) = m+0 = 0+1 & r262m+1 maggen

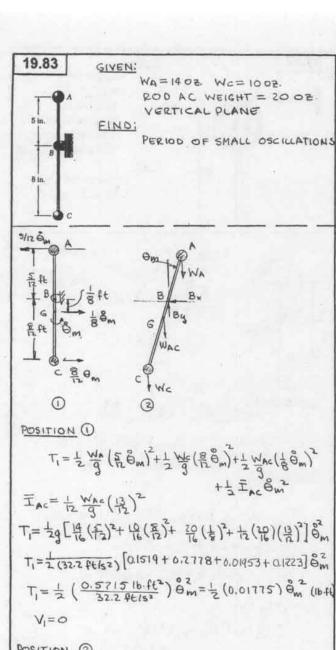
& m= wnom

(=mo+2+3max (2) W" 02 m= (k+2+ Mag 2) 03 m

Wn=ler2+MABgl Imov2+13 MABl2

 $W_{\rm N}^2 = \frac{6.207}{0.106} = 58.55$

 $T_{N} = \frac{2\pi}{\omega_{N}} = \frac{2\pi}{\sqrt{58.55}} = 0.821 S$



POSITION (2)

T2 = 0

V2=-WA = (1-COSOM) + WE = (1-COSOM) + WACE (1-cosom)

1-cosem= 521N, 0m & 0m

V=[傷(長)(長)+(鬼)(長)+(鬼)(台)] @m (Bft) Vz=[-0.3646+0.4167+0.1563] 0mm

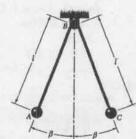
V2= 0.2084 8m/2

T, + Y1 = T2+ Y2 1 (0.01775) 0m+0=0+0.2084 0m

& m= Wn & m WW= 017084 = 11.738

 $T_N = \frac{217}{VW_N} = \frac{217}{V11.738} = 1.8345$

19.84



GIVEN:

SPHERES AND ROD AR ALL OF HASS M B= 400 l= 0.5 m

FIND:

FREQUENCY OF SHALL OSCILLATIONS

DATUM pm

POSITION (1)

T==== m(Vx)m+== m(Vc)m+==(2Ia)(Bm)+=(2MEBm) Ic= 12 12 (Va) m= (Vc) m= 18 m TI= M/28 + (M/2+ M/2) Om = 7 m/2 Om VI=-2mglcosp-mglcosp=-=mglcosp

POSITION @ T2=0

> V2= - mg l cos (β-om) - mg l cos (β-om) - mg l cos (β+om) - mg l z cos (β+om) V2=- 5 mgl [P.OSB COSOM + SINBSINOM +copcosom-singsin om) V2=- 2 mgl cosp cosom

COSOM & 1-02m/2 (SMALL ANGLES)

V2====mgl cosp [1-02m/2]

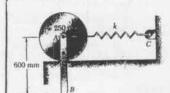
T1+ V1= T2+ Y2

76 ml20m- 5mglcoso = 0-5mgleosp(1-gm)

Om=Wn &m

I & WZ Om = \$ - 9 COSB Om W= 15 9 cosp Wn2 = 15 (9.81 m/s2) cos 400 = 16.10 52

fn= \frac{\omega_N}{2\pi} = \frac{\sqrt{16.10}}{2\pi} = 0.639 Hz

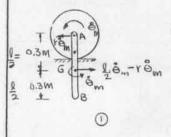


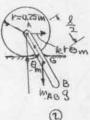
GIVEN:

0.8 leg ROD BOLTED TON
1.2 leg DISK
L= 12 N/M
DISK ROLLS WITHOUT
SLIDING.

FIND:

PERIOD OF SMALL OSCILLATIONS





POSITION ()

 $T_{i} = \frac{1}{2} (\overline{T}_{G}) \frac{\partial}{\partial m} + \frac{1}{2} m_{i} (\frac{1}{2} - r)^{2} \frac{\partial}{\partial m} + \frac{1}{2} (\overline{T}_{G}) \frac{\partial^{2}}{\partial l s \kappa} m + \frac{1}{2} m_{i} r^{2} \frac{\partial}{\partial m}$ $(\overline{T}_{G})_{AB} = \frac{1}{2} m \ell^{2} = \frac{1}{12} (0.8) (0.6)^{2} = 0.024 \text{ kg} \cdot m^{2}$ $m_{AB} (\frac{1}{2} - r)^{2} = (0.8) (6.3 - 0.25)^{2} = 0.002 \text{ kg} - m^{2}$ $(\overline{T}_{G})_{DISK} = \frac{1}{2} m_{DISK} r^{2} = \frac{1}{2} (1.2) (0.25)^{2} = 0.0375 \text{ kg} m^{2}$ $mr^{2} = 1.2 (0.25)^{2} = 0.0750 \text{ kg} - m^{2}$ $T_{i} = \frac{1}{2} [0.024 + 0.002 + 0.0375 + 0.0750] \frac{\partial^{2}}{\partial m}$ $T_{i} = \frac{1}{2} [0.1385] \frac{\partial^{2}}{\partial m}$ $V_{i} = 0$

POSITION 2

T2= 0

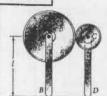
 $V_{2} = \frac{1}{2} k (r \Theta_{m})^{2} + M_{RB} g \frac{1}{2} (1 - \cos \Theta_{m})$ $1 - \cos \Theta_{m} = 2 \sin^{2} \Theta_{m} \approx \frac{6}{2} m^{2} (shall angles)$ $V_{2} = \frac{1}{2} (12 \text{ N/m}) (0.25 \text{ m})^{2} \Theta_{m}^{2} + (8 \text{ kg}) (9.8 \text{ kg}) (0.6 \text{ m}) \frac{6}{2} m^{2}$ $V_{2} = \frac{1}{2} [0.750 + 2.354] \Theta_{m}^{2} = \frac{1}{2} (3.104) \theta_{m}^{2} \text{ D.m}$

 $T_1 + U_1 = T_2 + V_2$ $\Theta_m^2 = \omega_n^2 \Theta_m^2$ $\frac{1}{2} (0.1385) \Theta_m^2 \omega_n^2 + 0 = 0 + \frac{1}{2} (3.104) \Theta_m^2$

WN= (3.104 N.M) = 22.415-2

$$T_{N} = \frac{2\pi}{W_{N}} = \frac{2\pi}{\sqrt{22.41}} = 1.3275$$

19.86

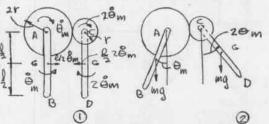


GIVEN:

RODS AB AND CD EACH OF MASS M AND LENGTH & ATTACHED TO GEARS A ANDOMASS OF GEAR A = 4 M MASS OF GEAR C= M

FIND:

PERIOD OF SHALL OSCILLATIONS



LET BA=OM 20m=Gc)m
20m=Gc)m

POSITION ()

TI = 1 IA 8 + 1 Ic (28 m) + 1 I IAB 8 + 1 I CO (28 m)

+ 1 MAB (18 m) + 1 MAB (18 m) + 1 MAB (18 m)

$$\begin{split} & \vec{I}_{A} = \frac{1}{3} (4m) (2r)^{2} = 8mr^{2} \\ & \vec{I}_{c} = \frac{1}{3} (m) (r)^{2} = \frac{1}{3} mr^{2} \\ & \vec{I}_{AB} = \frac{1}{3} m \ell^{2} \quad \vec{I}_{CB} = \frac{1}{2} m \ell^{2} \\ & \vec{I}_{1} = \frac{1}{3} m \left[8r^{2} + (r^{2}/2)4 + \ell^{2}/(2 + \ell^{2}/3 + \ell^{2}/4 + \ell^{2}) \right] \\ & \vec{I}_{1} = \frac{1}{3} m \left[10r^{2} + \frac{5}{3}\ell^{2} \right] \vec{\Theta}_{m}^{2} \\ & \vec{V}_{1} = 0 \end{split}$$

POSITION @

T,=0

V1= mgg (1-cosom)+mge (1-co40m)

SHALL ANGLES 1-COSOM= ZSIN' OM = OM

V=1mgl(gm+20m)=1mgl 50m = 20m

TI + VI=T2+12 62m=W202m

= m [10+2+ \ 22] Wn2 0 2 + 0 = 0+ 5 mg 1 50 2m

 $W_{N}^{2} = \frac{2gl}{10r^{2}+5gl^{2}} = \frac{3gl}{12r^{2}+2l^{2}}$

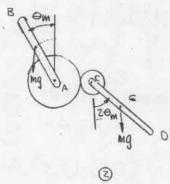
 $\mathcal{L}^{N} = \frac{\sqrt{m^N}}{3 \, \mu} = 3 \, \mu \sqrt{\frac{3 \, \delta \sigma}{15 \, k_2 + 5 \, \sigma_2}}$

GIVEN:

RODS AB AND BC GACH OF MASS M GEAR A OF MASS 4M GEAR C OF MASS W

FIND: PERIOD OF SHALL OSCILLATIONS

(mes) 1



KINEHATICS 204=0c 2 roa= roc 78,= 8c LET OA=Om 20m=(0c)m 28 m= (0.)m

POSITION ()

T= 1 In 8m+1 I (10) + 1 Ing 8m + 1 Ico (28m)2 + 5 mas (20m) + 2 mo (120m)

IA= 12 (4m)(2+)= 8m+2 Ic= 7 (m)(hs) = 7 m Ls

IAB= 12 Ml2 Ico= tzml2

T== 1 m[8+2+(+2/2)4+22/12+22/3+22/4+22] = m

T_= 1 m[10+2+ 522] 0m V=0

POSITION 3 T2=0 12= -mg & (1-cosem) +mg & (1-cos20m)

SMALL ANGLES 1-COS OM= ZSINZOM & OM

1-cos 20m=251020m220m220m N= - mgg =2m+mgg 20m=1 mg/3 0m

TitVI=Tz+Vz Om=WNOM

1 m[10r2+ 322] 0 mwn+0=0+1 mg 130m

Wn= 381 = 998 107+522 6072+1012

Th 211 271 160 12+10 12

19.88

GIVEN:

10-16 ROD CD DISKS A AND B EACH WEIGH ZO 16 AC OF NEGLIGIBLE WEIGHT NO SLIDING

FIND: PERIOD OF SMALL OSCILLATIONS

WA=WB=WDISE

T= = = 2(I)0m+= (2wbise)(+0m)+= Io0m (In) = = = Moise 1= = = (20)(1)= 10 Ico = 12 WGO (2 = 1 (10) (3) = 15 T== == [20+40+15+5] 0m $T = \frac{1}{2}g(70)\theta_{M}^{2}$

V1=0 POSITION 3

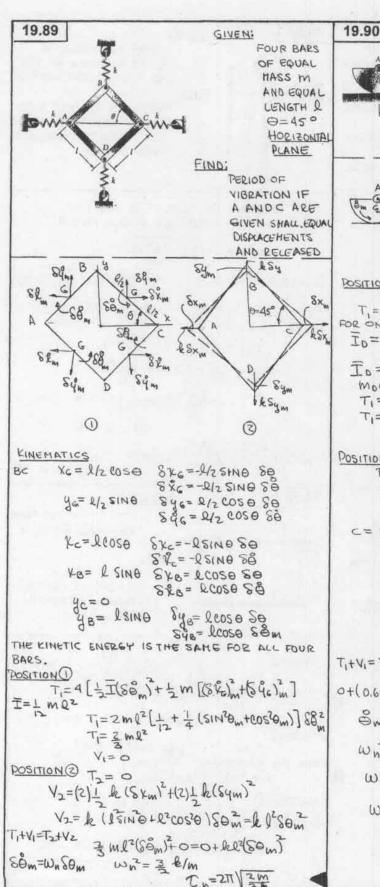
Vz= Wco & (1-cosom)

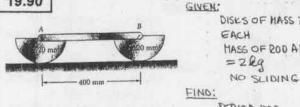
SHALL ANGLES 1-COSOM=25IN29 2 0m 12=7 Meol & m = 7 (10) (1.2) & m = 7 128 m

Ti+V=Tz+Vz Om=WnOm

79401m202+0=0+7120m W7= 159

 $T_N = \frac{2\pi}{\sqrt{(\mu_N)}} = 2\pi \sqrt{\frac{70}{(15)(32.2)}} = 2.395.$





DISKS OF HASS 3 49 EACH MASS OF 200 AB = 2 Rg

SHALL OSCULATIONS

PERIOD FOR

POSITION (1)

T,=2(1) IO 0 m+26/mo(r-c)20m+1 mr +20m ID = (To) - Moc2 = 1 Mor- motor) = Mo[12 1612]

To= 0.3199 Mor2 Mo(r-c)= mo +2(1-4)=0.3313 Mo +2 T1=[(0.3199+0.3313) Mo+2+0.5m++2]6m Ti= (0.6512 Mo+0.5 Mr] 12

POSITION 3 T2=0

Vz= 2 mog c (1-cosom)

C= 4r 1-cosom= 251N'Qm = Qm (SHALL

V= 2 mog 4 = 0m

 $V_2 = M_0 r \frac{q(4)}{3\pi} = M_0 r (9.81)(4) = 4.164 m_0 r$

T,+V= T2+VZ

0+(0.6512motosmr)+202 = 0+4.164 mor

Om=Wnom

 $\omega_{n}^{2} = \frac{4.164 \, \text{Mp+}}{0.6512 \, \text{Mp+}.5 \, \text{Mr}) t^{7}} \frac{(4.164)(3)}{[0.6512)(3) + 0.5(2)]6.120}$ $W_{N}^{2} = \frac{12.490}{0.3544}$ = 35.24

Wn = 5.936

 $T_N = \frac{2\pi}{\omega_N} = \frac{2\pi}{5.936} = 1.0585$

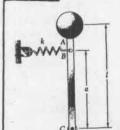
GIVEN:

SPHEEE OF WEIGHT W BAR ABC OF NEGLIGIBLE WEIGHT

EIND:

(4) FREQUENCY OF SHALL OSCILLATIONS

(b) SHALLEST VALUE OF a FOR WHICH OSCILLATIONS WILL OCCUP. 19.92



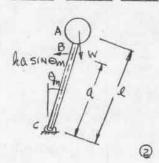
GIVEN:

t=0.8HE WHEN W=2lb f=1.5HE WHEN W=2lb

FIND:

FOR GIVEN &, a AND LITHE LARGEST VALUE OF W FOR WHICH OSCILLATIONS WILL OCCUP.

lêm A



POSITION 1

(a)

 $T_1 = \frac{1}{2} m (l \theta_m)^2 = \frac{1}{2} m l^2 \theta_m^2$ $V_1 = 0$

(1)

DOSITION @

T=0 V2=1 le (asina) - w (1-cosom)

SMALL ANGLES SINOM & OM

V2= 1 & a2 02 - Wl 02 = 1 [ka2-Wl] 02m

Ti+V1=T2+V2

 $\frac{1}{2} m \ell^{2} \hat{\Theta}_{m}^{2} + 0 = 0 + \frac{1}{2} [\ell a^{2} - W \ell] \hat{\Theta}_{m}^{2}$ $\hat{\Theta}_{m} = \omega_{n} \hat{\Theta}_{m} \quad m = W / g$ $\frac{W}{g} \ell^{2} \omega_{n}^{2} \hat{\Theta}_{m}^{2} = \ell a^{2} - W \ell$ $W_{n}^{2} = \frac{\ell a^{2} - W \ell}{(W \ell)^{2}} = \frac{9 \ell \ell [\ell a^{2} - 1]}{(W \ell)^{2}}$

 $\int_{M} = \frac{3\pi}{1} \Omega^{N} = \frac{3\pi}{1} \sqrt{\frac{3}{1} \left(\frac{MS}{VO_{3}} - 1\right)}$

(p) tn=0

ka2-1 > 0 WL a > VWL/k SEE SOLUTION TO DROB 19.91 FOR THE FRQUENCY IN TERMS OF W, L, Q AND &

fn=1.5 Hz W=216 1.5= = 1 \frac{1}{211} \frac{9/2 \left(\frac{\ha^2}{22} - 1 \)

fn=0.8Hz W=41b 0.8= 1/2π √9/4 (ka²-1) (2)

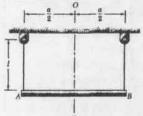
 $\frac{\left(\frac{1.5}{0.8}\right)^{2} = \left(\frac{ka^{2}-1}{2a}\right) / \left(\frac{ka^{2}-1}{4a}\right)^{2} }{3.516 ka^{2} - 3.516 = ka^{2} - 3$

3.516 Ka - 5.516 = ha - 1 2e - 1

 $\frac{ka^{2}\left(\frac{3.516}{4} - \frac{1}{2}\right) = 2.516}{ka^{2} = 6640}$

 $f_{N} = \frac{1}{2\pi} \sqrt{9/(\frac{6.640}{W} - 1)}, f_{N} = 0, \frac{6.640}{W} - 1 = 0$

19.93

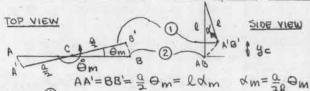


GIVEN:

PIPE SUSPENDED FROM TWO CABLES AT A AND B

FIND:

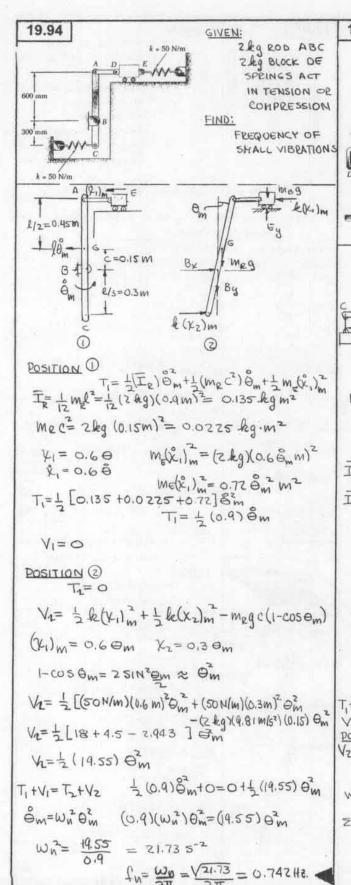
PREQUENCY
VIBRATION FOR A
SMALL ROTATION
ABOUT OO'

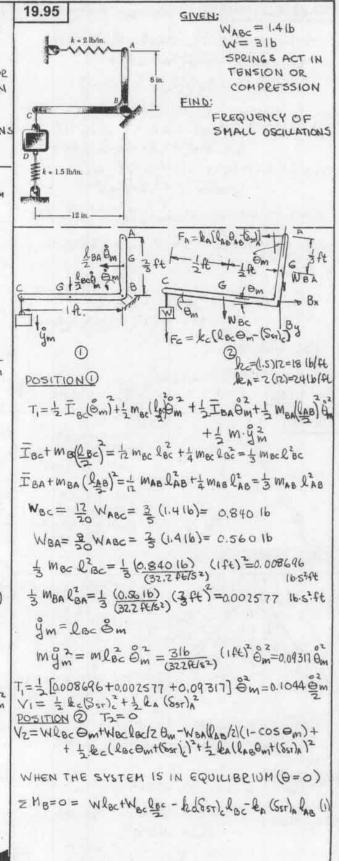


POSITION (T) $T_i = 0$ $V_i = mg y_c = mg l (1-cos x)$ SHALL ANGLES 1-cos $d_i = 2 \sin \frac{1}{2} \cos \frac{1}{2} \cos \frac{1}{2} \cos \frac{1}{2}$ $V_i = mg l (\frac{0^2}{8l^2}) \frac{1}{60^2} m^2$

 $\frac{\text{POSITION } @}{\text{en}} \tau_2 = \frac{1}{2} \vec{\Xi} \hat{\Theta}_{\text{m}}^2 = \frac{1}{2} \left(\frac{1}{12} \text{ m } \alpha^2 \right) \hat{\Theta}_{\text{m}}^2 \quad \forall_z = 0$

 $mg l(a^{2}/8e^{2}) + 0 + \frac{1}{24}ma^{2}w_{n}^{2}\Theta_{m}^{2}$ $w_{n}^{2} = 3g/l \qquad f_{n} = \frac{1}{2\pi}\sqrt{3g/l}$





19.95 CONTINUED I-COS OM = 2 SINZOM = OL

V2=[Wlac+ Wactac/2)]0m -[Wan(las/2)(0m/2)]+ + = Relacom - Relathon + 1 Rec (Sor) + + 1 ka (200m - & ASSA 1/20 M+ 1 ka (SST) 2

TAKING EQUATION (1) INTO ACCOUNT 1/2=-[WBA(RAB/2) 02m/2+1 kr 18c 0m + 1 kc(Ssr)2+1 ka 02 20m+1 ka (Ssr)2

 $V_2 = \frac{1}{2} \left[-0.560 \left(\frac{1}{3} \right) + 18 \left(1 \right)^2 + 24 \left(\frac{2}{3} \right)^2 \right] \oplus_{M}^{2}$ + 1 kg(Ssr) 2 +1 ka(Ssr) 2

V2= 1 [0.1867+18+10.67] 0 + 1 hc (65) 2+1 ka 651)2

V== = [28.48] 0 + = le(Ssr) + = la (Ssr)2

T, + V, = T3+V2

1 (0.1044) 0 m + 1 lectosi 2 + 1 latosi) = 1 (28.48) 0 m + 1 lectosi) 2 + 1 latosi) 2

Om=Wn Om

0.1044 Wn 0m = 24,92 0m W= 28.48 = 272.8 52 fn= 12728 = 2.63 Hz.

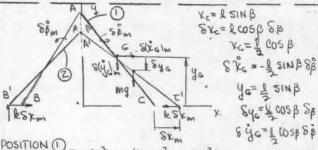
+19.96



RODS AB AND AC EACH OF MASS W & HTDNEJ ONA

FIND:

PERIOD WHEN A IS GIVEN A SHALL DOWN DEFLECTION AND RELEASED

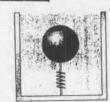


T = 2 [I I (SE) + 1 M((SX)) = + (SE) =] = (12 ml2+m22(51N2B+cos2B)) Epim = 3 ml2 SBm V1=0

POSITION @ TI=0 = le (28xm)== = (4 1cosp)= 212cos3B 8 Bm=Wn 8 Bm

I me wi Ski+0=0+2kleoste Spi T, + V, = T21 V2 112= 6Recos2A

Ch Valenicos B = 2TT M



GIVEN:

SUBHERGED SPHERE V= YOLUNG OF THE STYCKE KINETIC ENERGY = 7 PVU WHERE P= MASS DENSITY AND N = WELDCITY OF THE SPHERE SPHERE HASS = 500-3 R= 500 N/M HOLLOW OF SPHERE RADIUS = 80mm

FIND:

(A) DETLIOD WHEN DISPLACED YERTICALLY AND RELEASED

(b) PEZIOD WHEN THE TANK IS ACCELERATED UPWARD AT 8m/52

THIS IS NOT A DAMPED YIBRATION. HOWEVER THE KINETIC ENERGY OF THE FLUID MUST BE INCLUDED



POSITION (2) T2=0 V2= L lexin

POSITION () TI = TSPORT + TELVID = 1 M& UM+ 4 CV UM 5 W. 5 Um + 1 8 VUm+0=0+ 1 12 7m Ti+ Vi= T2+ V2 Um= 2m= xmwn 12 (ms+ = 5V) xm w= = = 1 12 xm, wn= ms+ = PV WA = 500 N/M \$8V = \frac{1000 kg}{3} (\frac{4}{3}\tau (0.08mi) (0.5kg+(= SV) = gv=1.0723kg Wn = 318 5-2 Th= 2it $=\frac{2\pi}{\sqrt{318}}=0.3525$

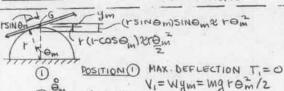
b) Acceleration does not change has, Th=0.352 s 19.98



PLATE ON A SEMI-CIRCULAR CYLINDER AS SHOWN

FIND:

PERIOD FOR SMALL OSCILLATIONS

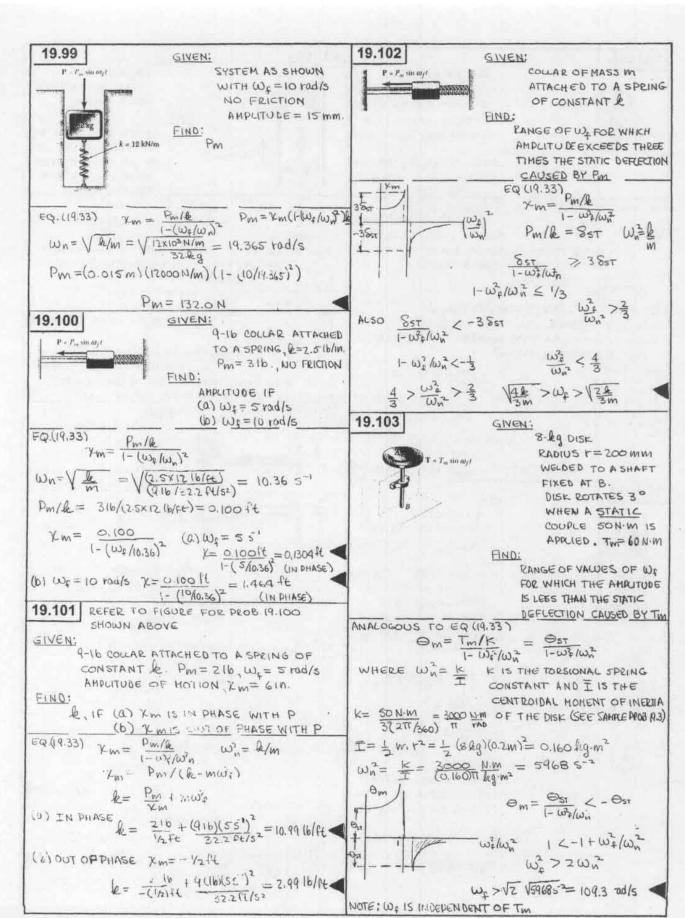


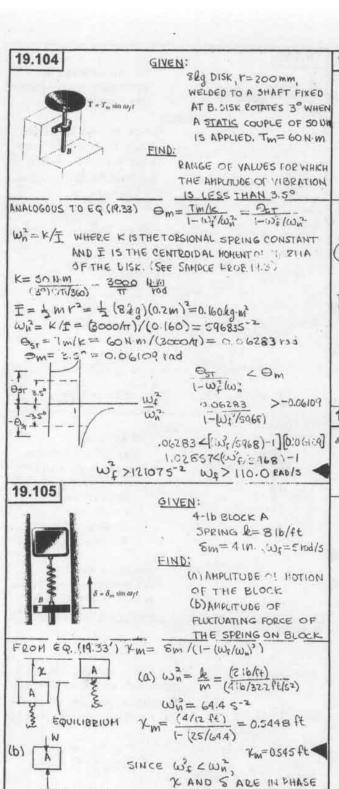
MOTADA POSITION® (0=0) T2=1=0m=1(12)ml20m ôm=Wn ⊕m T2 = 1 (/2) Ml2 Wir Om

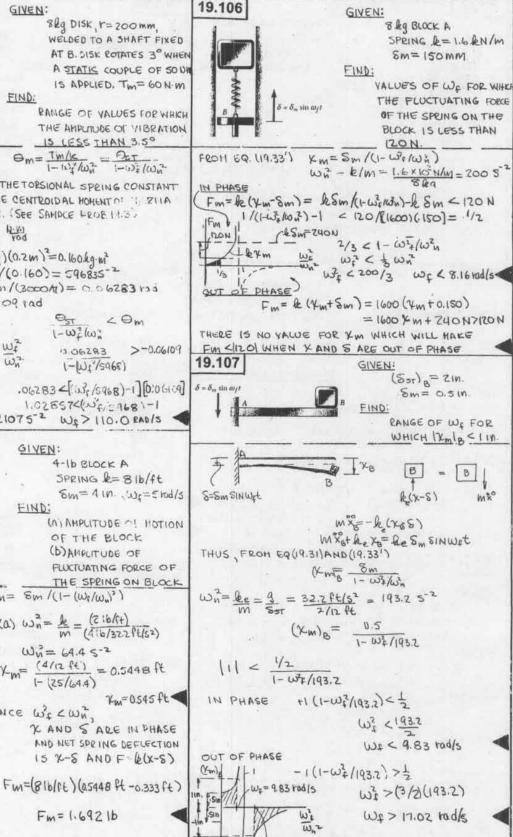
T,+V,=T2+V2 0+1mgrom = = = (12)ml2w20m

In= 211 / 12/12gr

Th= TTR

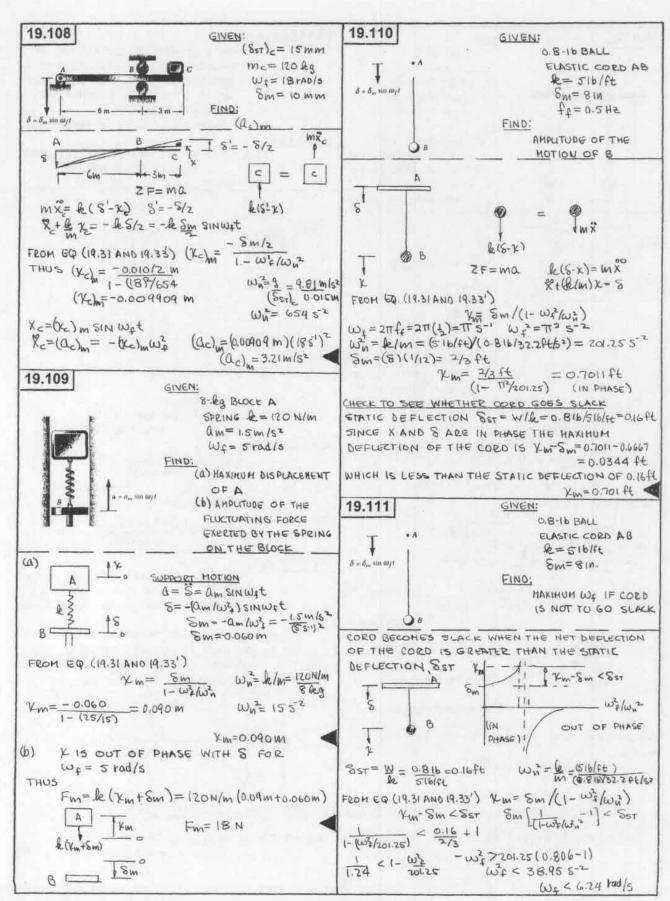


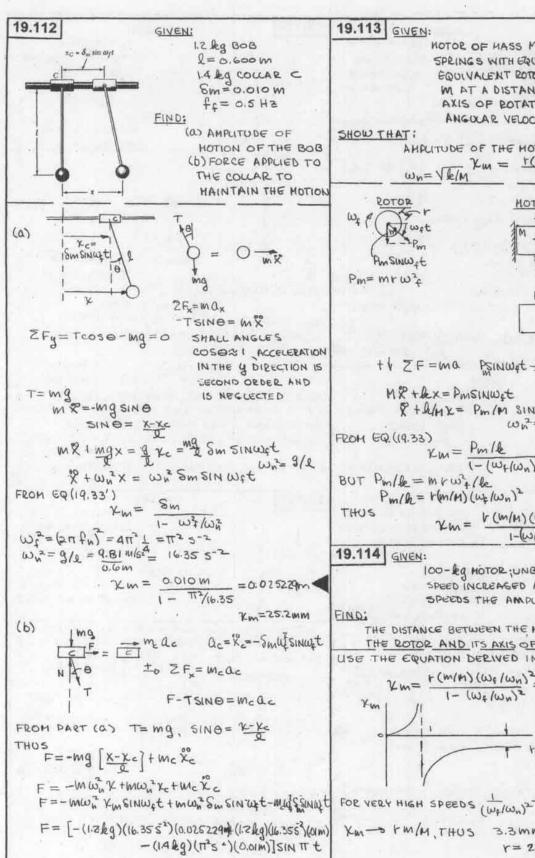




we=17.02 rad/s

Fm= 1.692 1b





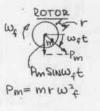
F= -0.437 SINTT (N)

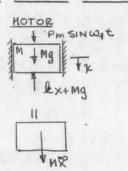
19.113 GIVEN:

HOTOR OF HASS M SUPPORTED BY SPRINGS WITH EQUIVALENT CONSTANT & EQUIVALENT ROTOR HASS UN BALANCE M AT A DISTANCE I FROM THE AXIS OF POTATION. ANGOLAR VELOCITY OF MOTOR, W.

SHOW THAT :

AMPLITUDE OF THE MOTION OF THE MOTOR Xm = r(m/m)(we/wis 1-(wa/wn)





+ V ZF = ma PSINWAT-lex = nxº

M&+lex= PMSINWET & + l/mx = Pm/m sin wet w== le/n

FROM EQ.(19.33)
$$K_{M} = \frac{P_{m}/k}{(-(\omega_{f}/\omega_{n})^{2})}$$
RUT D. (19.33)

BUT Pm/le=mrw3/le Pm/h= +(m/m) (wx/wn)2

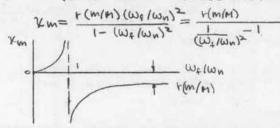
XM = 1-(m/N) (m+/m") = DED

19.114 GIVEN:

100-leg HOTOR; UNBALANCED 15-leg ROTOR. SPEED INCREASED AND AT VERY HIGH SPEEDS THE AMPLITUDE NEARS 3.3 MM

FIND:

THE DISTANCE BETWEEN THE HAS S CENTER OF THE ROTOR AND ITS AXIS OF ROTATION USE THE EQUATION DERIVED IN PROB 19.113 (ABOVE)



Km-> +M/M, THUS 3.3 mm= + (15/100) r= 22 mm

19.115 GIVEN:

SPRING SUPPORTED HOTOR WHOSE SPEED IS INCREASED FROM 200 TO 300 RAM AMPLITUDE DUE TO UNBALANCE INCREASES CONTINUOUSLY FROM 2.5 TO 8 MM

FIND:

SPEED AT RESONANCE

FROM PROB. 19.113 2m= +(m/M)(w=/w_)2 2.5 = r(m/m)(200/wn)2 1-(200/Wn)2 8 = K(m/M) (300/m) 2 1- (300/00)2 $\frac{2.5}{8.0} = \frac{1 - (300/\omega_n)^2}{1 - (200/\omega_n)^2} \left(\frac{200}{300}\right)^2$

0.703-0.703(200/wn)=1-(300/wn)2 1 [90x103 28.125 X103] = 0.2969

Wn= 708,4 Wn=457 rpm

RESONANCE WHEN WE = WN

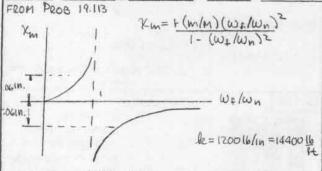
Wc=457tpm ◀

19.116 GIVEN:

400-16 HOTOR SUPPORTED BY SPRINGS WITH TOTAL &= 12001blin. ROTOR UNBALANCE IS 102, 8 IN FROM THE AXIS OF POTATION

FIND:

RANGE OF ALLOWABLE VALUES OF HOTOR SPEED IF THE AMPUTUDE OF VIBRATION IS NOT TO EXCEED 0.06 IN



wn= k/m=(14400 16/ft) = 1159.25-2 (40016/32.2ft/s2)

+ m/M = (8/12 ft) (1/16 16) = 104.17 × 10-6 ft

(0.06/12ft) < 104.17 x10 ft (wolu) X=0.06 in 1-(W=/W=)2 47.998-47.998 (W+/Wn)2 (W+/Wn)

(Wf/W) = 47.998

W € 0.9897 (1159.2) 12 War < 0,9897

wf € 33.69 rad/s Wf = (33.69 rad /60 5) (1 2TT RAD/REV) = 322 RAM

xm=-0.06in (-0.06/12ft)> 104.17x106tt(m3/m2) 1-m2/w2= (WF/Wn) = 47.998=

Wf 7(1.0106)(1159.2)1/2 Wf 734.40 rod/s=329 rpm

19.117



GIVEN:

220-16 HOTOR UNBALANCE OF THE ROTOR= 202,411 FROM THE AXIS OF ROTATION RESONANCE AT 400 PM

FIND:

AMPUTUDE AT (a) 800 rpm (b) 200 rpm (c) 425 rpm FROM PROB 19.113

Km= + (m/M) (W=/wn)2

RESONANCE AT 400 PM HEANS THAT WH = 400 MM v (m/n) = (4 in.)(2/16)/(220) = 2.2727×103 in. (a) $(\omega_f/\omega_n)^2 = (800/400)^2 = 4$

Km=2.2.7.7X1030.(4) =0.003031M

(b) (w, 16, 1)= (200/400)= 1/4

7-m= 2.272(10 (4)=0.000758111.

(c)(wf/wn)=(925/400)=1.1289

Km=2272x103(1.1289) =-0.0199011. 1-1.1289

19.118



GIVEN:

180-leg MOTOR UNBALANCE OF THE ROTOR = 289 ISO MM FROM AXIS OF POTATION STATIC DEPERTION SST=12MM

FIND:

HASS OF A PLATE ADDED TO THE BASE OF THE HOTOR SO THAT AMPLITUDE OF VIBRATION IS LESS THAN GOXIOGM FOR HOTOR SPEEDS ABOVE 300 rpm.

FROM PROB 19.113

Km= K(m/M) (wa/wn)2 1- W7/W2

SINCE HWIZER Xm= (mr/b) W} / (1-W}/W)

BEFORE THE PLATE IS ADDED, $\omega_n^2 = \frac{9}{8} = \frac{9.81 \, \text{m/s}^2}{0.012 \, \text{m}}$

le= HWm=(180 leg)(817.55) Wu= 817.5 5-2 k= 147.15 ×103 N/m

mr/R=(28x103kg)(0.150m)/(47.15x103N/m) = 28.542×10-9 m.52

AFTER THE PLATE IS ADDED THE NATURAL FREDUENCY OF THE SYSTEM CHANGES SINCE THE HASS CHANGES W"= Je/M'

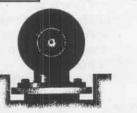
SINCE THE VIBRATION IS TO BE LESS THAN GOXIO'M FOR HOTOR SPEEDS ABOVE 300 FPM, WE HAVE Xm= -60x106 m= (28.542x159 m.52)(300-2115)

1- (300,217/60)

- 2.1299+21299 (98696)=1

 $\omega_n^2 = \frac{2.1299 (986.96)}{2.1299} = 671.65^2 =$ M'=(147.15×103N/m) /(671.65-2)= 219.1 &g AM=M'-H=219-180= 39,1 Ra

GIVEN:

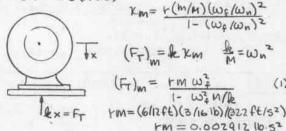


400-16 HOTOR UNBALANCE OF 3 02. 6 IN FROM AXIS OF ROTATION FORCE TRANSHITTED TO FOUNDATION LIMITED TO 0.2 10 WHEN MOTOR IS BUN AT 100 FPM AND ABOVE

FIND:

(a) HAXIMUH ALLOWABLE SPRING CONSTANT & OF A PAD PLACED BETWEEN THE HOTOR AND THE FOUNDATION (b) CORRESPONDING AMPLITUDE OF THE FLUCTUATING FORCE WHEN THE HOTOR IS RUN AT 200 rpm

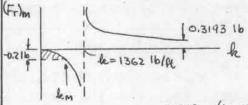
(a) FROM PROB. (19.113)



AT We = 100 rpm = 100 (21/60) = 10.472 rad/s

$$(F_{T})_{M} = \frac{(0.002412 | bb^{2})(10.472 s^{-1})^{2}}{1 - (10.472 s^{-1})^{2}(4001b/32.2 ft/s^{2})}$$

(FT) m= 0.31928 1- 1362/k



-. Z = 0.319 /(1-1362/fe) -0.2+0.2 (1362)/A=0.31928

$$le_{m} = \frac{(0.2)(1362)}{0.51928} = 525 \text{ lb/ft}$$

(b) AT 200 rpm, W= (200)(211)/60=20.94 rad/s FROM (1), AND USING & FOUND IN PART (a)

$$\begin{array}{c} (F_{T})_{M} = \underbrace{(0.002912 \text{ lb} \cdot \text{S}^{2}) \left(20.94 \text{ S}^{-1}\right)^{2}}_{1 - (20.94 \text{ S}^{-1})^{2} \underbrace{(40016/32.2ft/6^{2})}_{(52516/4t)} \\ (F_{T})_{M} \\ \downarrow f_{T} = 100 \text{ rpm} \\ (F_{T})_{M} = -0.1361 \text{ lb} \\ f_{T} = 200 \text{ rpm} \\ f_{T}/f_{H} \\ -0.1361 \text{ ft} \end{array}$$

19.120 GIVEN:

180-REG MOTOR, SUPPORTED BY SPRINGS OF TOTAL CONSTANT &= 150 &N/M UNBALANCE OF THE ROTOR IS 28-9 AT 150 MM

FIND:

RANGE OF SPEEDS FOR WHICH THE FLUCTUATING FORCE (FT) IS LESS THAN ZON

FROM PROB (19.113) 1- (wf/wn)2 Km= +(m/m) (w))

(Fr) m= lexm

FT= +m W} /(1-(4/Wn)2) rm=(0.150 m)(0.028 eg) = 0.0042 m. eg Wn= k/n=(150x103 N/m)/(180 kg)=833.35-2 (Ft= (0.0042) (W3)]/(1- W2/833.3)



 $(F_{\tau})(1-\omega_{\xi}^{2}/833.3) = 0.0042\omega_{\xi}^{2}$

 $W_t^2 = (F_t)_m [(F_t)_m | (853.3) + 0.0042]$ FOR (FT) = 20 N

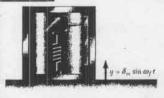
Wf < 0.024+0.0042 = 709.2 52 WE 5 26.63 rod/s

W< 26.63 (60) = 254 rpm

FOR (Fr) = -20 N $w_t^2 > \frac{-20}{-0.024 + .0042} = 10105^{-1}$ We 731.78 tad/s

W>31.78 (60) = 303 rpm (211)

19.121



GIVEN: fu= 120 Hz ZM = AMPLITUDE RELATIVE TO THE BOX IS USED AS A MEASURE OF SM

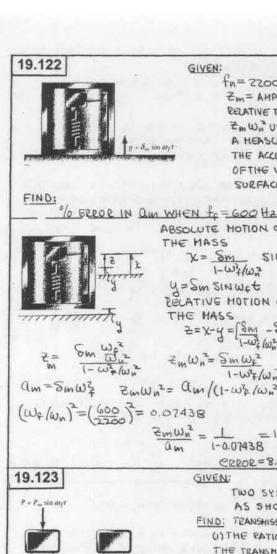
(a) % ERROR FOR fr= 600H (b) for ZERO GREEN

x = (Sm 2) SIN WET

y= Sm SINWAT Z = RELATIVE HOTION Z= 2-4= [5m - 8m] sinut Zm= Sm[1-willow] = Sm willing

 $\frac{\omega_1^*/\omega_n^{2^*}}{(-\omega_1^*/\omega_n^{2^*})} = \frac{(600/120)^2}{1-(600/120)^2} = \frac{25}{24} = 1.0417$ 1-(600/120)2 ERROR=4.17%

(b) == 1 = \frac{1-w_{\frac{1}{2}}/w_{\frac{1}{2}}}{w_{\frac{1}{2}}/w_{\frac{1}{2}}} tt= = tn= 12 (150)= 84.9HF 1=2 Wifluz



fu= 2200 HZ Zm= AMPUTUBE RELATIVE TO THE BOX Zm Wi USED AS A HEASURE OF THE ACCELERATION OFTHE VIBRATING SURFACE, QUIT & W

ABSOLUTE MOTION OF ! THE MASS X = Sm SINWET

4= Sm SINWet RELATIVE MOTION OF THE MASS 2=x-y=[8m -Sm]sinwit

1-W= (w= Zmwn= Smwz

1-W4/Wm Zmwn2 = am/(1-w2/wn2)

> ZmWn = = 1.0804 1-0.07438

> > CREOR = 8.04 %

TWO SYSTEMS

GIVEN:

AS SHOWN FIND: TRANSHISSIBILY LIE TO OITAS SHT(U THE TRANSHITTED FORCE TO THE IM PRESSED FORCE (2) RATIO OF THE TRANSMITTED DISPLACEMENT TO THE IMPRESSED DISPLACEMENT

SHOW THAT:

TO REDUCE TRANSMISSIBILITY, WE W. > VZ

(1) FROM EQ. (19.33) X m= Pm/R 1-(mx/mn)2 FORCE TRANSHITTED, Prim= lexm= le TRANSHISSIBILITY = (Pr)m = 1-(wp/wn)2

(2) FROM EQ. (19.33') DISPLACEMENT TRANSMITTED 7-M = 1-(We/Wh)2 TRANSMISSIBILITY = XM = 1-WEIWN

FOR (A)m OR KM TO BE LESS THAN I

11-(w, 10,)2/ < 1 1<11-(44/4)31 5 F (nw12W)

19.124

GIVEN:

60-16 DISK

e=0.006 in. R = 40,000 16/FE FOR THE SHAFT WHICH ROTATES AT A CONSTANT ANGULAR VEWCITY WE ABOUT AB

FIND: (a) WE FOR RESONANCE (b) DEFLECTION + WHEN may 0021 = 10

WE

G DESCRIBES A CIRCLE ABOUT THE AXIS AB OF RADIUS THUS an= (HE) WE

DEFLECTION OF THE SHAFT IS. THUS F=ler

F=man ler=m(He)WE w= le m= le/wn2 Kr= K2 (rte) Wi

1- w=2/w=

(a) RESONANCE OCCURS WHEN WE = WIN, i.e. +>0 Wn= / le/m = /(0,000 16/ft) = 146.5 md/s (6016/32.2ft/s2

w = w= (146.5)(60) = 1399.1 (2TI) = (0.006 IN) (1200) = 0.01669IM. 1- w26/00,2 1- (1200/13991)2

GIVEN:

19.125

FIND:

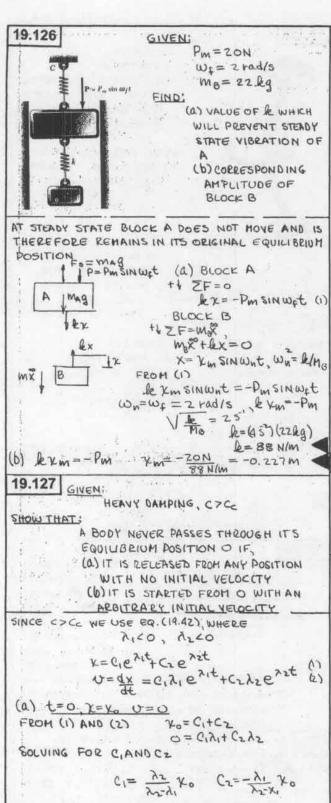


(a) SPEED U AT WHICH

MASS OF TRAILER AND LOAD = 250 kg TRAILER SUPPORTED BY TWO SPRINGS EACH OF 10 = 10 PM ROAD SURFACE IS A SINE CURVE WITH AHRITUDE OF 40 MM AND WAVELENGTH 5 M

LESONANCE WILL OCCUR (b) AMPLITUDE OF THE TRAILER VIBRATION AT U= 50 km/h

TOTAL SPRING CONSTANT le = 2 le = 20 len/m wi= le/m = (20,000 N/m)/(250 leg) = 805 y=Sm Sin } S_= 0.040 M x=nt y= 8m sin will t= 2/0 WE= STI/C THUS X=5My = 0.04 SIN WET m 12 FRONI EQ (19.331) xm= Sm/(+w2/w2) ey=Smsinwet PESONANCE W== 2TTV=Wn=1805



SUBSTITUTING FOR CIAND CZ IN (1)

x= x2 [22e lt xie xxt]

19.127 CONTINUED FOR X=0 WHEN t=0, WE HUST HAVE

\[\lambda_1 \in \frac{1}{2} \in \lambda_2 \in \frac{1}{2} \in 12 = e (12-11)+ (3) BECALL THAT ALCO, AZCO. CHOOSING ALAND AZ SO THAT A, CA, CO, WE HAVE 0 < 12- 11 ONA 12-2170 THUS A POSITIVE SOLUTION FOR too FOR EQ. (3) CANNOT EXIST SINCE IT WOULD REQUIRE THAT & RAISED TO A POSITIVE POWER BE LESS THAN 1. WHICH IS IMPOSSIBLE. THUS X IS NEVER O. THE X-t CURVE CURVE X FOR THIS CASE IS SHOWN (b) t=0, X=0, U=Up EQ. (1) AND (2), YIELD + O=CI+CZ No=CIAI+CZAZ SOLVING FOR CIANDCZ, CI= No CZ= UZ SUBSTITUTING (NTO (1) Us | (e xzt exit) FOR 1=0, tax p hat e hit FOR C>Cc, A, \$ Az THUS NO SOLUTION CAN EXIST FOR & AND X IS NEVER O THE X-t CURVE FOR THIS HOTION IS AS SHOWN 19.128 GIVEN: HEAVY DAMPING C >CC SHOW THAT: A BODY REJEASED FROM AN ARBITRARY POSITION WITH AN ARBITRARY VELOCITY CANNOT PASS THROUGH ITS EQUILIBRIUM POSITION HORE THAN ONCE. SUBSTITUTE THE INITIAL CONDITIONS , t=0, X= %, V= Vo IN EQS (1) AND(2) OF PROB. 19.127 Ko=Ci+Cz No=CiritCzzz SOLVING FOR CLANDCZ, CI= (10-12x0) (2=(10-12x0) AND SUBSTITUTING IN (1) x= 1/2-1/2 [(0,-1,1/2) ext (0,-12/2) ext] FOR K=0, ++00 (Uo- xixo)ent = (Uo-raxo)ent e(22-21) = (20 - 22 /20) THIS DEFINES ONE VALUE OF t ONLY FOR K=0,

WHICH WILL EXIST IF THE NATURAL LOG IS POSITIVE L.E IF No-1240 >1 . ASSUMING 2/4240

THIS OCCURS IF No Chiko

SLOPE= 1, Xo.

19.129 GIVEN

LIGHT DAMPING, C<CE

SHOW THAT :

THE RATIO OF ANY TWO SUCCESSIVE HAXIHUM DISPLACEMENTS Ky AND Ky IN FIG. 19.11 IS A CONSTANT AND THAT THE NATURAL LOGARITHM OF THIS PATIO CALLED THE LOGARITHHIC DECREMENT IS,

 $\ln \frac{\chi_n}{\chi_{n+1}} = \frac{2\pi (c/c_c)}{\sqrt{1-(c/c_c)^2}}$

FOR LIGHT DAMPING, CCCE
EQ (19.46) X=X0 E FAM) SIN (Wot+ 0) AT GIVEN HAX. DISPLACEMENT, t=tn, x=Kn SIN (Wet+++)=1, Xn=X0 expenses AT NEXT MAX. DISPLACEMENT, t=tn+1, X=Xn+1 SIN (Wotnerto)=1 Xn+= x0e-cczmstn+1 BUT Wotner-Wotn=2TT

PATIO OF SUCCESSIVE DISPLACEMENTS: Kn = Ko e Smtn = e Sm(tu-tu+i) + Sm 200 YNTI YO Q-SIM THI

THUS In Kn = CIT mwo

FROM EQS. (1945) WD= WN VI- (8)2 AND (1941) WD = CE VI-(E)2

tn+-tn=211/WD

ln ko = CTT 2M (VI-(E)2 ln Ku = 217 (c/c)

19.130 GIVEN:

LIGHT DAMPING C/C. < 1

SHOW THAT;

SHOW THAT THE LOGARITHMIC DECREMENT CAN BE EXPRESSED AS 1/R On (Yn/Yutk), WHERE & IS THE NUMBER OF CYCLES

AS IN PROB. 19.129 FOR HAXIMUM DISPLACEMENTS Xn AND Xute AT to AND totle, SIN(wototo)=1 AND SIN (Wothtet 1) = 1. Kuta= Ko e- (zm)(tuta) Xn= Xo e- Kramten

PATIO OF MAXIMUM DISPLACEMENTS

Yn/Yn+&= Yo & C/2m)th = e = e = m)(thtmb)

BUT Wotn+&-Wotn=& 12TT) th-th+&=& 2TT

BUT Wotn+&-Wotn=& 12TT)

THUS $\frac{V_n}{V_n} = \frac{1}{2}m(\frac{2k\pi}{\omega_0})$; $\ln k_0 = k \frac{C\pi}{m\omega_0}$

BUT FROM PROB. 19. 129 EQ.(1) LOG DECREMENT = ln Kn = CTT KNOT = WWO

COMPARING WITH EQ (2) LOG DEZREMENT = I In Kn Q.E.D. 19.131 GIVEN:

LIGHT DAMPING, CCCE To= 211/WD

SHOW THAT

- (1) TIME BETWEEN A MAXIMUM POSITIVE DISPLACEMENT AND THE FOLLOWING MAX NEGATIVE DISPLACEMENT IS TOLZ
- (b) TIME BETWEEN TWO SUCCESSIVE ZERO DISPLACEMENTS IS TO/2
- (C) TIME BETWEEN A HAXIMUM POSITIVE DISPLACEMENT AND THE POLLOWING ZERO DISPLACEMENT IS GREATER THAN TO/4

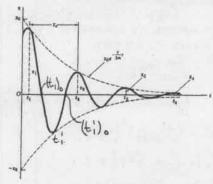


FIG. 19,11

x= to e (c/zm)t EQ. (19.46) SIN(Wot+4)

(a) MAXIMA (POSITIVE OF NEGATIVE) WHEN L=0

= 40 (-c/2m) = 12m)t (wot +0) + To wo e (c/zmt cos watto) THUS ZEED VELOCITIES OCCUR ATTINGS WHEN 1=0, OR TANLWotto)= 2mwo/c

THE TIME TO THE FIRST ZERO VELOCITY, t, 15

t = (TAN (2mwold) - b) /wb (1) THE TIME TO THE NEXT ZERO VELOCITY WHERE THE

DISPLACEMENT IS NEGATIVE, IS t,'=[TAN"(2mwolc)-++1](W.13)

SUBTRACTING (2) FROM (3) t,'- t,= T/WO = TT. CO = Co/2 Q.E.D BETWEEN READINGS OF THE MAXIMUM DISPLACEMENT (b) ZERO DISPLACEMENTS OCCUP WHEN

SIN (Wetto) = 0 OR AT INTERVALS OF $W_0t+\phi=\Pi$, 2Π NTT THUS, (+110=(1-4)/WO AND +1)0=(217-4)/WD THE BLTWEEN O'S = (+1)0-(+1)0=211-11 = 11 to = TO O.E.O

TAN (wpt+4) PLOT OF EQ.(1)

(c) THE FIRST HAXIMA OCCUPS AT 1 (Wotito) THE FIRST ZERO OCCURS AT (WOLLI) + 4)=TT FROM THE ABOVE PLOT (Walt, +4) - (Wat, +4) > I OR (t,)0-t, > 17/2W0 (t,) -t, > To/4 Q.E.D SIMILAR PROOFS CAN BE HADE FOR SUBSEQUENT HAX AND MIN



GIVEN:

BLOCK IN EQUILIBRIUM AS SHOWN IS DEPRESSED 1.2 IN. AND RECEASED AFTER 10 CYCLES THE HAXINUM DISPLACEMENT OF THE BLOCK IS 0.5 IN.

FIND:

(a) THE DAMPING FACTOR C/Cc
(b) THE YAWE OF THE
CONFFICIENT OF VISCOUS DAMPING

FROM PROB 19.130 AND 19.129

(/k)/ ν , (ν n/ ν n+ ν) = 2 TC/ ν C

WHERE k = NUMBER OF CYCLES=10 $\sqrt{1-(c/c_c)^2}$ (a) FIRST MAXIMA 15, ν = 1.2 in.

THUS, ν = 1 ν = 1.2 = 2.4 ν + 1.10 = 0.5 = 2.4

$$\frac{1}{10} \ln 2.4 = 0.08755 = \frac{2\pi c/c_c}{\sqrt{1 - (c/c_c)^2}}$$

$$1 - (c/c_c)^2 = \left(\frac{2\pi}{0.08755}\right)^2 (c/c_c)^2$$

$$\left(\frac{c}{c_c}\right)^2 \left[\frac{2\pi}{0.08755}\right]^2 + 1 = 1$$

 $\left(\frac{C}{C_c}\right)^2 = 1/(5150+1) = 0.0001941$ $C/C_c = 0.01393$

OR Cc = 2 V&M Cc = 2 V/8-16/4) (916/322ft/52) Cc = 2,991 16.5/ft

FROM (a) &= 0.01393 C= (0.01393)(2.991)
C=0.0417 lbs/ft

19.133 GIVEN:

SUCCESSIVE MAXIMUM DISPLACEMENTS OF A SPRING-HASS-DAGHPOT SYSTEM ARE 25, 15, AND 9MM M= 18 kg, l= 2100 N/M

EINO:

(a) THE DAMPING FACTOR C/Ce

(6) THE COFFEICIENT OF YICOUS DAMPING C.

(a) From Prob 19.29 In 2n = 211 (c/cc)

FOR Ku=25 mm AND Fu+= 15mm

$$\lim_{t \to 0} \frac{25}{15} = 0.5108 = \frac{2\pi (c/cc)}{\sqrt{1 - (c/cc)^2}}$$

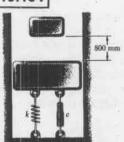
 $\left(\frac{c}{c}\right)_{3}\left[\left(\frac{(0.2108)}{5\pi}\right)_{3}+1\right]=1$

 $\left(\frac{C}{Cc}\right) = \frac{1}{(151.3+1)} = 0.006566, \frac{C}{Cc} = 0.0810$

(b) Cc = 21 (EQ. 19.41)

Cc= 2 Vkm = 2 V2100 N(m)(18 kg) = 0.3888 N.S

FROM (a) $\frac{c}{c_c} = 0.810$ c = (0.810)(.3888) = 31.5 N·S/m 19.134



SIVEN:

4-leg BLOCK A
9-leg BLOCK B
le= 1500 N/M
C= 230 N·S/M
BLOCK A IS DROPPED
FROM AN 800 MM
HEIGHT ONTO B WHICH
IS AT REST
NO REBOUND

EIND:

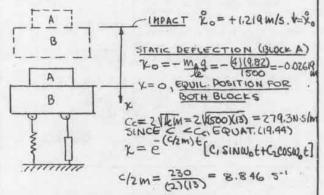
NAXIMUM DISTANCE BLOCKS MOVE AFTER IMPACT

VELOCITY OF BLOCK A JUST BEFORE IMPACT $N_A = \sqrt{29 h} = \sqrt{2(9.61)(0.8)} = 3.962 \text{ m/s}$ VELOCITY OF BLOCK S A AND B IMMEDIATELY

AFTER IMPACT

CONSERVATION OF HOMENTUM

(4)(3,962)+0=(4+9)0' (4)(3,962)+0=(4+9)0'



FROM TOP OF PAGE 1221 (5) = 1 - (5)

$$\omega_0 = \sqrt{\frac{1500}{13} - \left(\frac{230}{2203}\right)^2} = 6.094 \text{ rad/s}$$

 $\chi = e^{-8.846t}$ (c, SIN 6.094 t+Cz cos 6.094t) INITIAL CONDITIONS $\chi_0 = -0.02619$ m (t=0) $\chi_0 = +1.219$ m/s $\chi_0 = -0.02619 = e^{\circ}[c_1(0)+c_2(1)]$

 $\hat{x}(0) = -8.8468 \left[(0,0) + (-.02619)(1) \right] + e^{-6.8460} \left[(0,0) + (-.02619)(1) \right] = 1.219$ 1.219 = (-8.846)(-0.02619) + 6.09461 -8.844 $C_1 = 0.16202$

7= = 0.16202 SIN6.094t - 0.026190056.094t)
HAXIMUM DEFLECTION OCCURS WHEN \$=0

2=0=-8.846@ (0.1620251Nb.094t-0.02619C056.094t) + e88466(6.094)[0.1620 COS6.094t+0.02619CNL09

0=[(-8.846)(.16202)+(6.094)(0.02619)] SIN 6.094tm +(-8.846)(-0.02619)+(6.094)(0.1620)(0056.099+m

19.134 CONTINUED

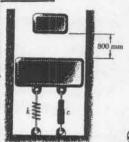
0=-1,274 SIN 6,094+ 1,219 COS 6,094+ TAN 6.094t = 1.219 = 0.957

TIME AT HAX DEFLECTION= t = TAN 0.957=0.1253 S

-(8.846)(0.1523)[0.165051N(6.094)(.1553) -0.02619 005(6.0947(1253)]

1 m=(0.3301) (0.1120-0.0189)=0.307m BLOCKS MOVE, STATIC DEFLECTION + KM TOTAL DISTANCE = 0.02619+0.307 =0.0569 M = 56.9 mm

19.135 GIVEN:



4-RA BLOCK A and BLOCK B R = 1500 N/M C= 300 N·s/m BLOCK A IS DROPPED FROM AN 800 MM HEIGHT XM = (-0.1219)e ONTO B WHICH IS AT REST NO REBOUND

FIND:

HAXI HUM DISTANCE BLOCKS MOVE AFTER IMPACT

VELOCITY OF BLOCK A JUST BEFORE IMPACT NA= VZ9h = VZ(9.81)(0.8) = 3.962 m/s VELOCITY OF BLOCK S A AND B IMMEDIATELY

AFTER IMPACT CONSERVATION OF HOHENTUM

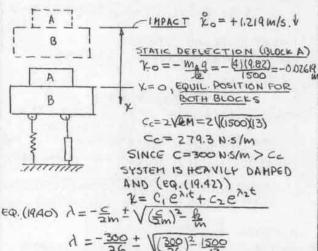
> MAUA+MBUB=WA+MB)U' (4)(3A62)+0=(4+9)~1

> > λ=41.538 ± 4.213

x= c. = 15.751t+ c. = 7.325t

λ=-15.751 λ2=-7.325

21=1.219 m/s = 20



19.135 CONTINUED

ENITIAL CONDITIONS 12= -0.02619m /2=1.219 m/s X(0)= X0=-0.02619= C1e"+ C2e" 12(0) = 20= 1.219= (-15.751)C,+(-7.325)Cz SOLKING SIMULTANEOUS LY FOR CIAND CZ C1= -0.1219 C2= 0.09571

x(t)=-0.1219e +0 +0.095710 HAXIMUM DEFLECTION WHEN X=0 2=0=(-1219)(-15.75)e-15.75+#(0.09571)(-7.325)e -7.325 tm 0=1,920e -0,701e

1.920 = (-7.325+15.75) Em 2.739=e 8.425 tm 0,701 en 2,739 = tm 8,425

tm=0.11965 - (15.75)(1196) - (7.325)(.1196) +6.09571)e

Km= -0.01851+ 0.03986 = 0.02136 M

TOTAL DEFLECTION = STATIC DEFLECTION + KM TOTAL DEFLECTION =0.02619+0.02136 = 0.0475 m= 47.5 mm

19.136 GIVEN:

GUN BARREL WEIGHT = 1500 16 RECUPERATOR CONSTANT C = 1100 lbs Ift

FIND:

(a) CONSTANT & FOR RECUPERATOR TO RETURN THE BARREL TO ITS FIRING POSITION IN THE SHORTEST TIME WITHOUT OSCILLATION (b) THE TIME NEEDED FOR THE BARREL TO HOVE TWO THIRDS OF THE WAY FROM ITS

MAXIMUM-RECOIL POSITION TO ITS FIRING POSITION (a) A CRITICALLY DAMPED SYSTEM REGAINS ITS EQUILIBRIUM POSITION IN THE SHORTEST TIME THUS C=Cc=1100=2m/== ZVEW Eq (19.41)

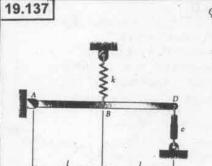
 $b = \frac{(c_{c/2})^2}{(1500 16 132.2 ft/s^2)} = \frac{(1500 16 132.2 ft/s^2)}{(1500 16 132.2 ft/s^2)}$

FOR A CRITICACLY DAMPEN

m from WE TAKE t= O AT HAXIMUM DEFLECTION. YO THUS & (0) = 0 , K(0) = K0

I NITIAL CONDITIONS x(0)= X0= (C+0) & C = K0 X= (Yotcztlewnt &=-wn (xo+czt) =wnt cze-wnt x(0)=0=-ωηχο+Cz Cz=ωηχο χ= γο(1+ωητ)εωητ

W'N= 11.806-1 ±= 2.289/11.806=0.19395

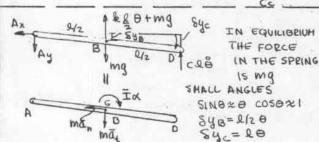


GIVEN:

200 OF MASS W PINNED AT A

FIND: INTERMS OF m, le, AND C

(a) DIFFERENMAL EQUATION OF MOTION (b) CRITICAL DAMPING COEFFICIENT



ZMA=(ZMA)ess (a) NEWTONS LAW

+ mg 2/2 - (lel/2 0+mg) 2/2-cl 8 1 = Id+ma 4/2 KINEHATICS a= 2/2 x = 2/2 0

 $[\bar{x} + m(2/2)^{3}] + C(2)^{3} + k(2/2)^{3} = 0$

I+m(e/2)2= = = ml2

6 + (3 c/m) 0 + (3 le/4 m) 0 = 0 (b) SUBSTITUTING 0= ext INTO THE DIFFERENTIAL EQUATION OBTAINED IN (a), WE OBTAIN THE CHARACTERISTIC EQUATION. 2 +(3c/m)2+32/4m=0

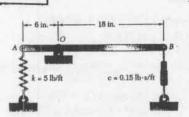
AND OBTAIN THE ROOTS

$$\lambda = -3c/m + \sqrt{(3c/m)^2 - (3k/m)}$$

THE CRITICAL DAMPING COEFFICIENT Co, IS THE VALUE OF C IN THE RADICAL TO ZERO. THUS

Cc = Vem/3

19.138



GIVEN:

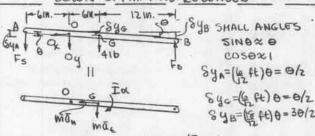
4-16 ROD AB PINNED AT O AND SUPPORTED BY A SPRING AT A. DIHENSIONS AND OTHER CONSTANTS AS SHOWN

FIND

FOR SHALL OSCILLATIONS

(a) THE DIFFERENTIAL EQUATION OF MOTION (b) THE FORMED BY THE ROD WITH THE HOZIZONTAL 5'S AFTER ENDB IS DUSHED

DOWN O. 9 IN. AND RELEASED



(a) NEWTONS LAW ZM=(ZMo)ess

1 - (6 ft) Fs +(6 ft) (4)- (8 ft) Fo = IX+(8 ft) mae

$$F_S = k(Sy_A + (S_{ST})_A) = k(\frac{9}{2} + (S_{ST})_A)$$

$$F_D = CS\dot{y}_B = C 3/2 \delta$$

KINEHATICS X= & Q= (12 ft) X= = THUS FROM (1)

[m+m] + + (3/2)200 + (k/2)(g+6st)-z=0

BUT IN EQUILIBRION ZNO=0 A R(Sor) (4)-4)(6)=0, &(Sor) = 2

EQ (2) BECONES (7/12) m & + (9/4) C + (2/4) 6 = 0 12 m=(12)(4/322)=0.07246, 9/4 C=(9/4)(15)=0.3375 12/4= 5/4=1.25

0.072460 +0.33750+1.250=0 (b) SUBSTITUTING EXTINTO THE ABOVE DIFFERENTIAL EQUATION 0.07246 x2+0.3375 x+1.25=0

A=(-0,3375 7 V(,3375)3-4(,07246)(1,25))/2(0704 2= (-0.3375 7 V-0.2484)/(2)(0.07246)

λ = -2.329 ± 3.439 L

SINCE THE KOOTS ARE COMPLEX AND CONTUGATE (LIGHT DAMPING), THE SOLUTION TO THE DIFFERENTIAL EQUATION IS (50.19.46),

$$\Theta = \Theta_0 e^{\frac{2.329 t}{5 \text{IN}(3.439 t + \phi)}}$$
 (3)

LEONTINDED)

19.138 CONTINUED

INITIAL CONDITIONS (Syp(0)) = 0,911. Q(0) = (540)/181n = 0.9 @(0) = 0.05 rad A(0) = 0 FROM (3)

0(0)=0.05 = 00 51N b

6(0) = 0 = -2.329 0, SIND +3.439 0, COS \$ TAN 0 = 3,439/2.329

> 0= 0.9755 rad SIN (.9755) = 0.06039 rad

SUBSTITUTING INTO (3)

0= 0.06039 e 2.329+ 5IN (3,439 t+0,9752)

AT t= 5 \$ (-2.329N5) @(5)=0.06039@ 51N [3.439)(5)+0.9752]

0(5)=-0.333 ×10 4 rad ⊕(5)=(0.019 09X10) ABOUE HOEIZONTAL

19.139 GIVEN:

1100-16 HACHINE SUPPORTED BY TWO SPEINGS EACH WITH &= 3000 16 PERIODIC FORCE APPLIED OF 30-16 AT 2.8 HZ. C= 110 16/ft

FIND:

AMPUTUDE OF STEADY STATE VIBRATION

Eq. (19.52) Ym= Pm \(\int(1e-m\omega^2)^2+(C\omega^2)^2 TOTAL SPEING CONSTANT &= (2)(3000 16/FE) = 6000 lb/ft

 $W_f = 2\pi f_c = 2\pi (2.8) = 5.6\pi \text{ rad/s}$

m = w/q = 1100 lb/(32.7 ft/s2) = 34.161 lb/s2/ft

7 m = 3016 V((6000-(34.161)(5.611)2)2+(11045.611)2)(6)2

 $\chi = \frac{30}{\sqrt{20.914\times10^6 + 3.745\times10^6}}$

X= 0.00604 ft

Zm= 0.0725 in.

19.140 GIVEN:

1100-16 HACHINE SUPPORTED BY TWO SPEINGS PERIODIC FORCE OF 30 16 APPLIED AT 2.8 HZ. C= 110 16/FL AMPCITUDE OF VIBRATION, Xm=0.0511.

FIND:

SPEING CONSTANT OF EACH SPRING

EQ. (19.52) 2m= Pm V(k-mut)2+(cwf)2 [(k-mw=)2+(cw=)2)x==Pm2

k= V(Pm/xm)2-(cwf)2 + mwf2

 $M^{t=3}\pi t=3\pi (s.8)=2.611$ $M=\frac{3}{M}=\frac{35.5 \text{ Hz}}{100 \text{ P}}=34.191 \frac{45.5}{5}$ $k = \sqrt{\frac{3016}{.05(12)}} + (34.161)(5.6\pi)^2$

R= V 51.84×106-3.745×106 +

R= 6935+10573=1750816/FE

&12= 8750 1b/ft

19.141 GIVEN:

FORCED VIBRATING SYSTEM

FIND:

VALUES OF C/CC FOR WHICH THE MAGNIFICATION FACTOR WILL DECREASE AS WE/WIN INCREASES

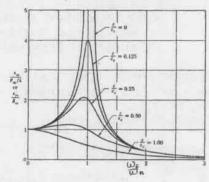


FIG. 19.12

EQ. (19.53)

MAG. FACTOR KM Pm/ke V(1-(W=16m)2)2+[2(c/c)(W=16m)2 FIND VALUE OF SICE FOR WHICH THERE IS

NO HAXIMUM FOR YM AS WITH INCREASES

 $\frac{d\left(\frac{V_{m}}{P_{m}/k_{c}}\right)^{2}}{d\left(\frac{W_{c}}{W_{m}}\right)^{2}} = -\frac{\left[2\left(1-\left(W_{c}/W_{m}\right)^{2}\right)(-1)+4c^{2}/c_{c}^{2}\right]}{\left[\left[1-\left(W_{c}/W_{m}\right)^{2}\right]^{2}+\left[2\left(C/c_{c}\right)\left(W_{c}/W_{m}\right)^{2}\right]\right]^{2}}$

 $-2 + 2(\omega_{5}/\omega_{n})^{2} + 4c^{2}/c_{c}^{2} = 0$

(wf/wn)= 1-2 c2/c22 c2/c2 ≥ 1 THERE IS NO MAXIMUM FOR AND THE HAGNIFICATION

FACTOR WILL DECREASE AS AS WE WIN INCREASES

C/CC7/1/VZ

C/C=7,0.707

19.142 GIVEN:

FORCED YIBRATING SYSTEM SHALL C/CC

SHOW THAT

HAXINUM AMPLITUDE OCCURS WHEN WE 2WM AND THAT THE CORRESPONDING VALUE OF THE HAGNIFICATION FACTOR IS & C/Cc.

EQ. (19.53')

HAG. FACTOR = $\frac{R_{m}}{P_{m}/R} = \frac{\sqrt{(1-(\omega_{f}/\omega_{h})^{2})^{2}+(2\cdot(c/c_{h})(\omega_{f}/\omega_{h})^{2})^{2}}}$

FIND YALUE OF WEIWN FOR WHICH KIM 15 A HAXIMUM

 $0 = \frac{d \left(\frac{k_{m}}{R_{m}/k_{0}} \right)^{2}}{d \left(\omega_{f} / \omega_{h} \right)^{2}} = - \frac{\left[z \left(1 - \left(\omega_{f} / \omega_{h} \right)^{2} \right) \left(-1 \right) + 4 \left(c^{2} / c_{c^{2}} \right) \right]}{\left\{ \left[1 - \left(\omega_{f} / \omega_{h} \right)^{2} \right]^{2} \left\{ z \left(c / c_{c} \right) \left(\omega_{f} / \omega_{h} \right) \right]^{2} \right\}^{2}}$

-2+2(ws/wn)2+4(c/ce)2=0

FOR SMALL C/Ce W/Www Wiscon

FOR $\omega_f/\omega_n = 1$, $\frac{\gamma_m}{P_m/k} = \frac{1}{\sqrt{[1-1]^2 + [2(c/c)1]^2}}$

(Pm/4) = 1 Cc

19.143 GIVEN:

15-leg MOTOR SUPPORTED BY FOUR SPEINGS EACH OF CONSTANT R= 45 RN/m

MOTOR UNBALANCE IS EQUIVALENT TO HASS OF ZOQ AT 125 MM FROM AXIS OF BOTATION

FIND;

AMPLITUDE OF STEADY STATE VIBRATION AT A SPEED OF 1500 PPM ASSUMING,

(a) NO DAMPING

(b) DAMPING FACTOR C/c= 1.3

EQ. (19.52) 1-m=V(k-mw})2+ (cw)2 W¢ 125mm=r W12=[1500)(217)/60]=2467452 k=(4)(4500)=1800001 Pm sinust Pm= m'rw= (0.02 kg) (0.125m) (246743) Pm=61.685 N 209=m' (a) C=0

61.685 N Zm= -[180000-15(24674)](N/M) Ym=-0.324×103m=-0.324mm

(b) FOR c/c= 1.3 Cc= 3286 NS/m C=(13)(3286)=4272 NS

61.685 N V[180000-15(24674)]2+(4272)2(24674)

Km=0.0884 XIU m=0.0884 mm

19.144

GIVEN:

18-kg MOTOR BOLTED TO A BEAM HAS A STATIC DEFLECTION ST- 1.5mm UNBALANCE IS EQUIVALENT TO A HASS OF ZOG LOCATED 125 MM FROM AXIS OF ROTATION

FIND:

AMPUTUDE AT A HOTOR SPEED OF 900 FPM (a) FOR NO DAMPING

(b) FOR C/Cc = 0.055 EQ. (19.52)

 $\kappa_{\text{m}} = \sqrt{(k - m\omega_f^2)^2 + (c\omega_f)^2}$

(m= 20g ROTOR rwet r= 125 mm Pm SIN Wet

Wif = [(900) (211)/60]=8882.6 52 FIND SPRING CONSTANT & FOR THE BEAM R= M9 =(18kg)(9.81 /2)

SST (1.5X10-3 M) le= 117720 N/m

Pm=m'rw=6.020kg)(0.125 m)(8882.652) Pm= 22,20 N

(a) C=0 22.20N 2m = ((117720-(18)(8882.6 N/m) 12m=-0.527 x103 m=-0.527 mm

(b) FOR c/ce = 0.055 EQ.(19.41) Cc=2/4 =2/km =2/117720)(18) Cc= 2911 N.5/m C= 0.055 C= (0.055)(2911) = 160.12 N/M

 $\gamma_{\text{LM}} = \frac{2.7.121}{V[117720-(18)(8883)]^2+(160.12)(8883)]}$

 $= \frac{22.21}{\sqrt{(1.779 \times 10^{9}) + (0.2278 \times 10^{9})}} = 0.000496W$ 2-w=0.496mm€

19.145

GIVEN:

100-16 MOTOR BOLTED
TO BEAM WHICH HAS
A STATIC DEFLECTION
SST= 0.25 IN.
UNBALANCE IS 4 02.
AT 3 IN.
AMPLITUDE XM=
0.010 IN AT 300 PPM

FIND:

(b) COEFFICIENT OF DAHDING

(b) COEFFICIENT OF DAHPING C

 $V_{m} = \frac{1}{\sqrt{(1-(\omega_{f}/\omega_{h})^{2})^{2}+(2(c/c_{c})(\omega_{f}/\omega_{h})^{2})^{2}}}$ $\omega_{f} = 31n. \qquad \omega_{n}^{2} = \frac{9}{8\pi} = \frac{32.2 \text{ ft/s}^{2}}{(0.25/62\text{ ft})}$ $\omega_{f} t \qquad \omega_{n}^{2} = \frac{9}{8\pi} = \frac{32.2 \text{ ft/s}^{2}}{(0.25/62\text{ ft})}$ $\omega_{f} t \qquad \omega_{n}^{2} = \frac{987.2 \text{ ft/s}^{2}}{(0.25/62\text{ ft})}$ $\omega_{f}^{2} = \frac{987.2}{15806} = 0.63.87 \text{ s}^{-2}$

 $P_{m} = \frac{m'r \omega_{f}^{2}}{4} = \frac{4(46)}{(32.2 \text{ ft/s}^{2})} (\frac{3}{12} \text{ ft}) (987.2 5^{2})$ $P_{m} = 1.916 \text{ lb}$

 $k = \omega_n^2 m = (1546)(100/32.2) = 4801 \text{ lb/ft}$ $P_m/k = 1.916/4801 = 0.0003991 \text{ ft}$

 $\frac{0.01}{12} = \frac{0.0003991}{\sqrt{(1-.6387)^2 + (4)(0.6387)(c/c_c)^2}}$ $0.2293 = 0.1305 + 2.555 (c/c_c)^2$ $(c/c_c)^2 = 0.0988 = 0.03867$

 $(c/c_c)^2 = \frac{0.0988}{2.555} = 0.03867$

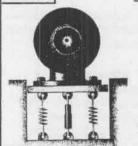
(b) EQ. (19.41) Cc= 2 MWn

Cc = 2 (10016 32.2 ft/s2) (1546) /2

Cc= 244.21b.5/ft

 $\frac{C}{C_c} = 0.1957$ $C = (244.2)(.1967) = 48.0 \frac{165}{ft}$

19.146



GIVEN:

BY FOUR SPRINGS EACH OF CONSTANT &= 90 km/m
DASHPOT C= 6500 N·s/m
AMPLITUDE Y_m=2.1 mm
AT A SPEED OF 1200 FPM
MASS OF THE ROTOR
M'=15 kg

FIND:

DISTANCE BETWEEN THE HASS CENTER OF THE ROTOR AND THE AXIS OF SHAFT

P_M

 $R_{m} = \frac{V(k-m\omega_{+}^{2})^{2}+(c\omega_{+})^{2}}{V(k-m\omega_{+}^{2})^{2}+(c\omega_{+})^{2}}$

 $m_3^{\xi} = 12.1612_{-5}$ $m_3^{\xi} = [1500)(54)/(60)_{\xi}$

Pm=m'ewf

R = 4 (90,000 N/m)=360,000 N/m

Pm=(15kg)(e)(1579152)

Pm= 236870.e

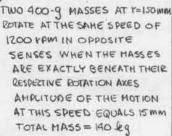
 $0.0021 = \frac{2368702}{\sqrt{[360,000-(100)(15791)]^{2}(6500^{2}(15791)]}}$

(1.4674 × 106)(0,0021)=(236870)e

e=0.1301 m e=(3.01 mm

19.147





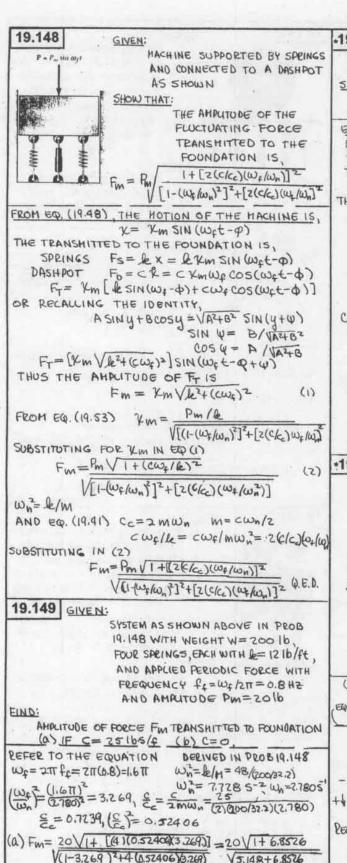
FIND:

(a) THE COMBINED SPRING CONSTANT & (b) THE DAMPING FACTOR C/C2

(a) $\phi = \pi/2$, AT 1200 FPM $\phi \in \varphi$, (19.54) TAN $\phi = \frac{2(C/C_2)(W_2/W_1)}{1 - (W_2/W_2)^2}$

SINCE $\phi = \pi/2$ TAN $\phi = \infty$ THUS $1 - (\omega_e/\omega_w)^2 = 0$

 $\omega_{n}^{2} = \frac{1}{M} \qquad \omega_{n} = \omega_{1} = (1200 \times 217) (60 = 40 \text{ IT 5}^{-1})$ $\omega_{n}^{2} = \frac{1}{M} \qquad \omega_{n} = (140 \text{ kg}) (40 \text{ IT 5}^{-1})^{2} = 2210 \text{ kg}$ $\omega_{n}^{2} = \frac{1}{M} \qquad \omega_{n}^{2} = (140 \text{ kg}) (40 \text{ IT 5}^{-1})^{2} = 2210 \text{ kg}$ $\omega_{n}^{2} = \frac{1}{M} \qquad \omega_{n}^{2} = \frac{1}{$



19.150 GIVEN: STEADY STATE VIBRATION UNDER A HARHONIC FORCE SHOW THAT: MECHANICAL ENERGY DISSIPATED PER CYCLE 15 E=TICKING ENERGY IS DISSIPATED BY THE DASHPOT FROM EQ (19.48) THE DEFLECTION OF THE SYSTEM IS X= XmSIN(Wet-0) THE FORCE ON THE DASHPOT, FO = CX Fo= < xmwf cos(wct-op) THE WORK DONE IN A COMPLETE CYCLE WITH T = ZIT/W E= (27/W) Fo dx (L.E FORCEX DISTANCE) dx=xmwccos(wet-q)dt E= (cxmwfcoswft-p)dt cos2ωot-φ)=(1-2 cos(ωtt-φ))/2 E= CX m W 1 [1-2005(w+t-0) d+ $E = \frac{C \chi_{m}^{2} \omega_{f}^{2} \left[t - 2 \frac{\sin(\omega_{f} t - \phi)}{\omega_{f}} \right]^{2\pi/\omega_{f}}}{2\pi/\omega_{f}} \left[\frac{2\pi}{\omega_{f}} - \frac{2}{\omega_{f}} \left(\frac{\sin(2\pi - \phi) - \sin(\phi)}{2\pi/\omega_{f}} \right) \right]$ E= ITC XIM WE Q. E.D. +19.151 GIVEN: SPRING - DASHPOT SYSTEM AS SHOWN WITH HASS IN HOUING AT N. OVER A ROAD WITH A SINUSOIDAL CROSS SECTION 'OF AMOUTUDE Sm AND WAVELENGTH L. LAYDIFFERENTIAL EQUATION OF VERTICAL DISPLACEMENT OF MASS IM (b) EXPRESSION FOR THE AMPLITUDE OF M (a) +1 ZF=ma: W-le(Ss++x-8)-c(dx-ds)=m dx RECALLING THAT W= & SST, WE WRITE m dix + cdx + lex= les+cds (1)

(CONTINUED)

Fm= 16.18 lb

(b) C=0, Fm=(20)/15,148 = 8.81 1b

• 19.151 CONTINUED

MOTION OF WHEEL IS A SINE CURVE, $S=S_M SIN \omega_t t$ THE INTERVAL OF TIME NEEDED TO TRAVEL A DISTANCE L AT A SPEED U, IS t=UU, which is the PERIOD OF THE ROAD SURFACE.

Thus $\omega_f = 2\pi/c_f = \frac{1}{2\pi} = 2\pi u/L$

AND 8= 8m sinwft

ds=Sm211 cos wet

THUS EQ. (1) IS.

mdik+cdx+lex=(lesinux+cu)cosux+)Sm

(b) FROM THE IDENTITY

 $\cos \phi = A \sqrt{A^2 + B^2}$

WE CAN WRITE THE DIFFERENTIAL EQUATION

M d2x + C dx + lex= Sm \ \k^2 + (cux)^2 SIN(\warphi t+\warphi)

W = TAN' Cwx

THE SOLUTION TO THIS EQUATION & IS (ANALOGOUS TO EQ'S 19.47 AND 19.48, WITH PM = Sm V&+(CW+)2)

X= Xm SIN (Wft-Q+ 4) JUHERE ANALOGOUS

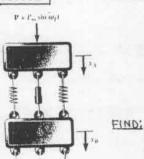
TO EQ'S (19.52))

x= 8/1/5+(cm2)2

TAN Q= cws

TAN W= cws/B

19.152 GIVEN:



BLOCKS A AND B HAVE
HASS IM
THREE SPRINGS, EACH HAVE
CONSTANT &
THREE DASHPOTS, EACH
HAVE CONSTANT C.
BLOCK A ACTED UPON
BY A FORCE PRISINGLE

DIFFERENTIAL EQUATIONS
DEFINING THE DISPLACEMENTS
FA AND FOR OF THE BLOCKS
FROM THEIR EQUIL BRIDM

POSITION. $|P_{M}SINW_{p}t|$ $|A| = |A| |m \tilde{x}_{A}|$ $|A| = |B| |m \tilde{x}_{B}|$ $|A| = |B| |m \tilde{x}_{B}|$ $|A| = |B| |m \tilde{x}_{B}|$ $|A| = |B| |m \tilde{x}_{B}|$

* 19.152 CONTINUED

SINCE THE ORIGINS OF COORDINATE ARE CHOSEN FROM THE EQUILIBRIUM POSITION, WE HAY OMIT THE INITIAL SPRING COMPRESSIONS AND THE EFFECT OF GRAVITY

FOR LOAD A

+ 1 ZF=man; Pinsinwet+zle(xg-ka)+c(xg-ka)=m xa

FOR LOAD B

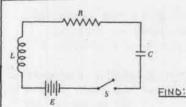
+ TF=MaB; -2 & (xB-XA)-c(XB-XA)-hxB-ZCXB= mxBB
REARRANGING EQS (1) AND (2), WE FIND:

GIVEN:

mx + c(x - x)+ 2 & (x - x 8) = Pm sinust

mxg+3cxg-cxa+3kxg-2kxa=0

19.153



R, L, C CIRCUIT

AS SHOWN WITH

SUDDENLY APPLIED

VOLTAGE E WHEN

THE SWITCH IS

CLOSED

VALUES OF R FOR
WHICH OSCILLATIONS
WILL TAKE PLACE
WHEN THE SWITCH S
IS CLOSED

FOR A HECHANICAL SYSTEM OSCILLATIONS TAKE PLACE IF C<CC. (LIGHTLY DAMPED)

BUT FROM EQ. (19.41).

cc=zmVk/m=zVkm

THEREFORE

FROM TABLE 19.2:

SUBSTITUTING IN (1) THE ANALOGOUS ELECTRICAL VALUES IN (2), WE FIND THAT OSCILLATIONS WILL TAKE PLACE IF,

P < 2 \((1/c)(L)

R < 2 VL/C

19.154 GIVEN:

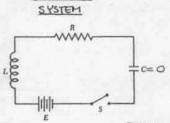
R,L,C CIRCUIT OF FIG. PROB 19.153 WITH CAPACITOR C REMOVED

FIND:

IF SWITCH S IS CLOSED AT t=0

- (a) THE FINAL VALUE OF THE CURRENT IN THE CIRCUIT
- (b) THE TIME t AT WHICH THE CURRENT WILL HAVE REACHED (1- YE) TIMES IS FINAL YALUE.

 (I.E THE TIME CONSTANT)



ELECTRICAL

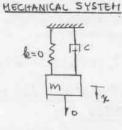


TABLE 19.2 FOR ANALOGUE CLOSING SWITCH & IS EQUIVALENT TO SUDDENLY APPLYING A CONSTANT FORCE OF HAGNITUDE P TO THE HASS.

(a) FINAL VALUE OF THE CURRENT CORRESPONDS
TO THE FINAL VELOCITY OF THE HASS, SINCE THE
CAPACITANCE IS REPORTE SPRING CONSTANT
IS ALSO REPORTED.

$$P = \frac{dx}{dt} + \sqrt{z} = m \frac{d^2x}{dt^2} \qquad (1)$$

FINAL VEWCITY OCCURS WHEN dix =0

FROM TABLE 19.2: U-OL, P-OE, C-OR
THUS

LFINAL E/R

(b) REARRANGING EQ. (1), WE HAVE

M dix + cdx = P

die + cdx = P

substitute dx = Ae-x+ 2; dx = Aze-x+

 $m[-A\lambda e^{-\lambda t}] + c[Ae^{-\lambda t}tP] = P$ $-m\lambda + c = 0$ $\lambda = c/m$

THUS dx = A = K/m/t + P

ATT t=0 dx =0 0 = A+P/c A=-P/c

N= dx = [1-e(c/m)t]

FROM TABLE 19.2: NOU, POE, COR, MOL

FOR (=(E/R)(1-1/e),(R/L)t=1 t= 1

19.155

GIVEN:

MECHANICAL SYSTEM SHOWN

DRAW:

THE ELECTRICAL ANALOGUE



kol/a kol/a kol/a kol/a kol/a kol/a kol/a kol/a WE NOTE THAT BOTH THE

SPRING AND THE DASHPOT

EFFECT THE HOTION OF

POINT A. THUS ONE LOOP

IN THE ELECTRICAL CIRCUIT

SHOULD CONSIST OF A

CAPACITOR (N=1/C) AND A

RESISTANCE (C=R)

THE OTHER LOOP CONSISTS

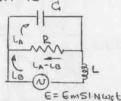
OF (PMSINWFT = EMSINWFT), AN

INDUCTOR (M=> L) AND THE

RESISTOR (C->R)

SINCE THE RESISTOR IS COMMON TO BOTH WOPS, THE CIPCUIT IS

CIRCUIT



19.156 GIVEN:

HECHANICAL SYSTEM SHOWN FIND:
THE ELECTRICAL ANALOGUE

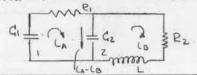


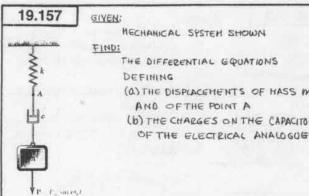


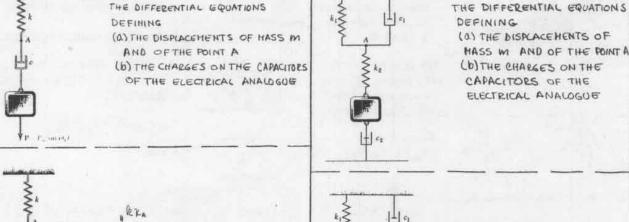
LOOP : (ROINT A) & - & & & C - R.

LOOP Z (HASS W) & > & M - L, C2 - 82

WITH (&2 - 1/42) COMMON TO BOTH LOOPS,





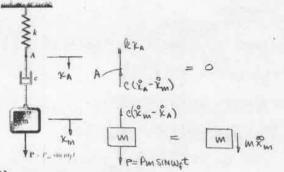


19.158

GIVEN:

FIND:

MECHANICAL SYSTEM SHOWN



(a) HECHANICAL SYSTEM

POINTA
+1 ZF=0
$$kx_A + c\frac{d}{d\epsilon}(x_A x_m) = 0$$

(b) ELECTRICAL ANALOGUE

FROM TABLE 19.2

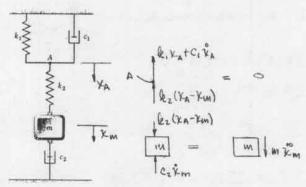
SUBSTITUTING INTO THE RESULTS FROM PART (A) THE ANALOGOUS ELECTRICAL CHARECTEISTICS.

$$(1/c)g_n + R \frac{d}{dt}(g_n - g_m) = 0$$

$$L \frac{d^2g_m}{dt^2} + R \frac{d}{dt}(g_m - g_n) = E_m \sin \omega_n t$$

NOTE:

THESE EQUATIONS CAN ALSO BE OBTAINED BY SUMMING THE VOLTAGE DROPS AROUND THE LOOPS IN THE CIRCUIT OF PROB 19.155



(a) HECHANICAL SYSTE!

POINT A
$$k_1 x_A + C_1 \frac{dx_A}{dt} + k_2 (x_A - x_m) = 0$$

$$c_1 \frac{dx_A}{dt} + (k_1 + k_2) x_A - k_2 x_m = 0$$

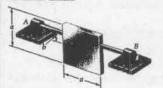
MASS M $\Sigma F = MQ$ $e_2(\chi_A - \chi_m) - C_2 \frac{d\chi_m}{dt} = M \frac{d^2\chi_m}{dt^2}$ md2xm + Czdxm + kz(xm-za)=0

(b) ELECTRICAL ANALOGUE

SUBSTITUTING INTO THE RESULTS FROM PART (a) USING THE ANALOGOUS ELECTRICAL CHARACTERISTICS FROM TABLE 19.2 (SEE LEFT).

$$L \frac{d^{2} g_{m}}{d t^{2}} + R_{z} \frac{d g_{m}}{d t} + \frac{1}{C_{z}} (g_{m} - g_{A}) = 0$$



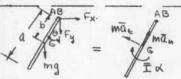


GIYEN:

THIN SQUARE PLATE
OF SIDE
OSCILLATIONS ABOUT
AB AT A DISTANCE
D FROM G

EIND:

(a) Recion if b= a/z
(b) a second value of b which gives the sahe period as in (a)



MEWIDNS LAW

ZMB= (ZIIAB)eff

$$(\overline{1}+mb^{2}) \stackrel{\circ}{\Theta} + mgb\Theta = 0$$

$$\overline{1}_{x} = \frac{1}{12} ma^{2} a \stackrel{G}{\longrightarrow} x$$

$$\stackrel{\circ}{\Theta} + \frac{gb}{\frac{1}{12}a^{2}+b^{2}} \Theta = 0$$

$$T_{N} = \frac{2\pi}{\omega_{N}} = 2\pi \sqrt{\frac{\alpha^{2} + 126^{2}}{129b}}$$
 (1)

(a) WHEN b= a/2.

$$C_{N}=2\pi\sqrt{\frac{\alpha^{2}+12(\alpha^{3}/4)}{12g(\alpha/2)}}=2\pi\sqrt{\frac{2\alpha}{3g}}$$

(b) EQUATING THE RESULT FROM PART (a) TO EQ. (1) AND SQUARING BOTH SIDES,

$$\frac{a^2 + 12b^2}{12gb} = \frac{2a}{3g}$$

36b2-(24a)(b)+3a2=0

$$b^2 - (\frac{2}{3}a)b + \frac{a^2}{12} = 0$$

$$b = +\frac{2}{3}a \pm \sqrt{\frac{4}{9}a^2 - \frac{2}{3}} = \frac{a}{2}, \frac{a}{6}$$

b= a/6

19.160



GIVEN:

HALF SECTION OF A SOCIO CYLINDER IS ROTATED THROUGH A SHALL ANGLE AND RELEASED

FIND:

PERIOD OF OSCILLATION (NUSUIONE)

POSITION (NUSUIONE)

POSITION (NUSUIONE)

V=0 T_=0

V_= mg c (1-cose) ~ mg c 0

V_= 0

V_= 0

T_=1 m 0^2 + 1 T 0

T_=1 m 0^2 + 1 T 0



iv=(r-c)6

NSTANTANEOUS CENTER

$$\begin{split} I_0 &= \vec{I} + mc^2 \quad \vec{I} = I_0 - mc^2 = \frac{1}{2} m r^2 - mc^2 = m \left[\frac{r^2 - c^2}{2} \right] \\ T_2 &= \frac{1}{2} m \left[(r - c)^2 + (\frac{r^2}{2} - c^2) \right] \hat{\theta}_M^2 \\ T_2 &= \frac{1}{2} m \left[r^2 - 2cr + c^2 + \frac{r^2}{2} - c^2 \right] \hat{\theta}_M^2 \\ c &= \frac{4r}{3\pi} \quad T_2 &= \frac{1}{2} m \left[\frac{3}{2} r^2 - 2 \left(\frac{4r}{5\pi} \right) r \right] \hat{\theta}_M^2 \\ T_3 &= \frac{1}{2} m r^2 \left[\frac{3}{2} - \frac{8}{3\pi} \right] \hat{\theta}_{=0.3256}^2 m r^2 \hat{\theta}_M^2 \end{aligned}$$

 $T_1+U_1=T_2+U_2$

$$0 + mg(\frac{4r}{3\pi})\frac{\theta_{m}^{2}}{2} = 0.3256 m r^{2} \theta_{m}^{2}$$

FOR SHALL OSCILLATIONS

 $g + 0.2122 \Theta_{m}^{2} = 0.3256 \text{m} t^{2} \Theta_{m}^{2} \omega_{n}^{2}$ $\omega_{n}^{2} = \frac{0.2122}{0.3256} \frac{q}{r} = 0.6518 \frac{q}{r} 5^{2}$ $\omega_{n} = 0.8073 \sqrt{\frac{q}{r}}$ $U_{n} = \frac{2\pi}{\omega_{n}} = \frac{2\pi}{0.8073} \sqrt{\frac{r}{q}}$

$$T_{ij} = 7.78 \sqrt{\frac{r}{g}}$$
 s

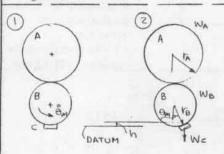
19.161

GIVEN:

WA=30 lb , WB=12 lb WC= 5 lb , ATTACHED TO B NO SCIPPING



PERIOD OF SHALL OSCILLATIONS



SMALL OSCILLATIONS h= rg(1-cosom) & rg 00/2

$$T_{1} = \frac{1}{2} m_{c} (r_{8} \delta_{m})^{2} + \frac{1}{2} \overline{I}_{8} \delta_{m}^{2} + \frac{1}{2} \overline{I}_{A} (\frac{r_{6}}{r_{A}} \delta_{m})^{2}$$

$$\overline{I}_{8} = \frac{m_{8} r_{6}^{2}}{2} \quad \overline{I}_{A} = \frac{m_{A} r_{A}^{2}}{2}$$

$$T_{1} = \frac{1}{2} \left[m_{c} r_{6}^{2} + m_{8} r_{6}^{2} / 2 + (m_{A} r_{A}^{2} / 2) (r_{6} / r_{A}^{2}) \right] \delta_{m}^{2}$$

$$T_{1} = \frac{1}{2} \left[(m_{c} + m_{8} / 2 + m_{A} / 2) r_{6}^{2} \delta_{m}^{2} \right]$$

$$V_{1} = 0$$

POSITION @ TL= 0

V2= mag h= mag 0 2/2

 $T_1+V_1=T_2+V_2$

= [(mc+ mb/z+ ma/z] 12 wn om + 0 =

0+ mgt = 0 2 /2

Wn2 = Me g Mc+ (Me+ma)/2 Te

 $w_{\rm N}^2 = \frac{5}{5 + (12 + 30)/2} \frac{(32.2 \text{ ft/s}^2)}{(6/12) \text{ ft}}$

Wn= 12.39 5-2

$$T_n = \frac{2\pi}{\omega_n} = \frac{2\pi}{\sqrt{0.39}} = 1.785 \text{ s.}$$

19.162

GIVEN:



4.02 GYROSCOPE ROTOR; Th= 6.00 S WHEN ROTOR IS SUSPENDED FROM A WILE AS SHOWN WHEN 1.25 IN. DIAMETER

WHEN IZS IN DIAMETER SPHERE IS SUSPENDED IN THE SAME FASHION THE PERIOD (CN) = 3.805.

FIND:

RADIUS OF GYRATION & OF THE POTOR

K= SPEING CONSTANT OF THE WIRE

FOR SPHERE OR POTOR



. O.

ZMo=(ZHo)eff

$$\omega_{N}^{2} = \frac{1}{2} \qquad \Gamma_{N} = \frac{2\pi}{\omega_{N}} = 2\pi \sqrt{\pm/k} \quad (1)$$

E = m = (4 (16 16) = 7.764 x 103 = 2

FROM (1) $6=2\pi\sqrt{2}$

SPHERE = 3 m +2 SP N+ = 490 16/ft3

 $M = \frac{4}{3} \pi \frac{[(1.25/2)/(12)ft]^3}{[32.2 ft/52]} [490 16/ft^3]$

M= 9.006 x 10-3 16.52/ft

 $\bar{I}_{s}^{2} = \frac{2}{5} (9.006 \times 10^{3} \text{ lb·s}^{2}/\text{ft}) [(1.25/2)/(12)\text{ft}]^{2}$ $\bar{I}_{s}^{2} = 9.772 \times 10^{6} \text{ lb·s}^{2}.\text{ft}$

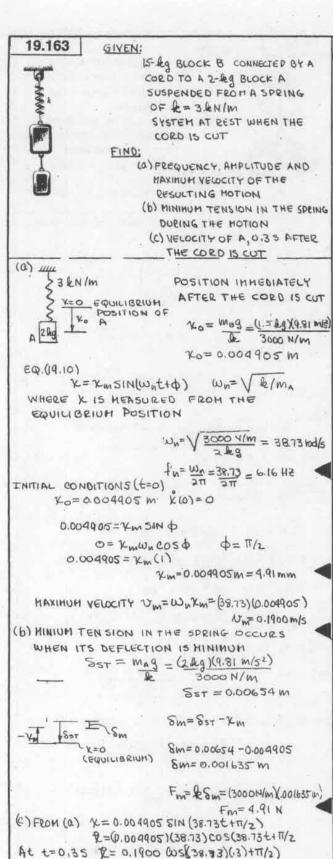
DIVIDE EQ. (2) BY EQ. (3) AND SQUARING,

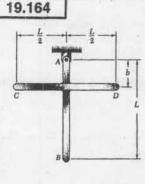
$$\left(\frac{6}{3.80}\right)^2 = \frac{(7.764 \times 10^3 \text{ lb s}^2/\text{fe}) \frac{1}{4}}{9.772 \times 10^6 \text{ lb s}^2 \text{ft}}$$

$$\overline{le}^{2} = \frac{(9.717 \times 10^{6})}{(7.764 \times 10^{3})} \left(\frac{6}{3.80}\right)^{2} = 3.138 \text{ ft}^{2}$$

$$\overline{le}^{2} = 0.0560 \text{ ft}$$

Te=0.672 in.





GIVEN:
TWO RODS EACH OF
MASS IM AND LENGTH
L, WELDED TOGETHER
TO FORM THE ASSEMBLY

SHOWN

FIND:

(A) THE DISTANCE D

FOR WHICH THE

FREQUENCY OF SMALL

OSCILLATIONS IS

MAXIMUM

(b) THE CORRESPONDING

HAXIMUM FREDUENCY

C T G VAB M91 M9 UZ

LIZ OM POSITION ()

 $T_{i} = \frac{1}{2} \left[m \vec{\upsilon}_{co}^{2} + m \vec{\upsilon}_{AB}^{2} + \vec{\Gamma}_{co} \vec{\vartheta}_{m}^{2} + \vec{\Gamma}_{AB} \vec{\vartheta}_{m}^{2} \right]$ $\vec{\upsilon}_{co} = \vec{\upsilon} \vec{\vartheta}_{m} \quad \vec{\upsilon}_{AB} = (L/z) \vec{\vartheta}_{m}$ $\vec{\Gamma}_{co} = \vec{\Gamma}_{AB} = \frac{1}{12} m L^{2}$

T1= 1 m [b2+(L/2)2+ 12 L2++2 L2] \$ 2 = m [b2+52/12] 6 4

POSITION (2)

V= mg b (1-cosom) + mg y/z(1-cosom)

SHALL ANGLES 1-cosom= 2 sin²(0m/z)= 0m/z

Vz= mg = (b+L/2)

Ti+Vi=Ti+V2 = m[62+512] & m+0=0+ mg[6+42] & m

 $\Theta_{m} = \omega_{n} \Theta_{m}$ $\omega_{n}^{2} = \frac{g(b+1/2)}{(b^{2}+5/(2L^{2})}$ (1)

MAXWIZ WHEN dwildb = 0

 $\frac{db}{db} = \frac{(b+s/(2)^2)q-q(b+l/2)(2b)}{(b^2+s/(2)^2)^2} = 0$ $b = -L + \sqrt{(2+(2)/(2))^2} = 0$ $b = -L + \sqrt{(2+(2)/(2))^2} = 0$

b= 0.316L (b) FROM EQ (i) AND THE ANSWER TO (a)

 $f_{N} = \frac{9[0.316+0.5]}{[0.316)^{2}+5/12]L} = 1.5808/L$ $f_{N} = \frac{\sqrt{\omega_{N}}}{2\pi} = \frac{\sqrt{1.580}}{2\pi} \sqrt{9}(L = 0.200\sqrt{9}/L + \frac{1}{2})$

2 (0.3) = 0.1542 W/S V

19.165 GIVEN:

SPRING SUPPORTED HOTOR SPEED INCREASED FROM 700 FPM TO 500 FPM.

AMPLITUDE OF VIBRATION DECREASE
CONTINUOUSLY FROM 8 MM TO 2.5 MM

EIND:

(a) RESONANT SPEED

(b) AMPLITUDE OF STEADY STATE VIBRATION AT 100 MA

(A) FOR A HOTOR WITH A ROTOR UNBALANCE THE AMPLITUDE OF VIBRATION IS GIVEN BY (SEE SAMPLE PROB 14.5)

AT ZOO FPM

$$-8 = \frac{(1 - (200)^2/k}{(1 - (200)^2/k})$$
 (1)

AT 500 rpm $-2.5 = \frac{Mr(500)^2/k}{(1-(500/f_w)^2)}$ (2)

DIVIDING EQ. (1) BY EQ. (2) TERM BY TERM,

$$\frac{3.5}{8} = \frac{(-(200/t^{N})_{5})(200)_{5}}{(-(200/t^{N})_{5})(200)_{5}}$$

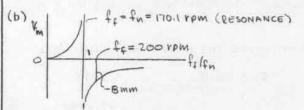
$$(3.2)(1-(2001fn)^2) = 0.160(1-(5001fn)^2)$$

fn= 170.14 rpm

fu= 170.1 rpm

RESONANCE WHEN ff=fn

ff = 170.1 rpm



AT 200 rpm $\omega_f = 2\pi(200)$ $\omega_f = 20\pi \text{ rab}$

$$-8 = \frac{\text{Mr}}{1 - (200/170.14)^2} (EQ.1)$$

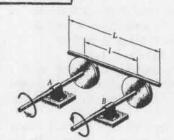
 $\frac{mr}{le} = \frac{(-8)(-0.3818)}{(20\pi/3)^2} = 0.006963$

AT 100 PPM

$$\chi_{\rm m} = \frac{(0.006963)(\frac{10}{3})^2}{1 - (\frac{100}{170.14})^2} = 1.1666 \, \rm mm$$

2 m= 1.167 mm

19.166



GIVEN:

ROD OF HASS IN AND
LENGTH L RESTS
ON TWO PULLEYS
WHICH POTATE
IN OPPOSITE
DIRECTIONS AS
SHOWN
ME COEFFICIENT
OF KINETIC FRKING
BETWEEN THE 200
AND THE PULLEYS

FIND:

FREQUENCY OF VIBRATION IF THE ROD IS GIVEN A SHALL DISPLACEMENT TO THE RIGHT

+
$$12F_y = \overline{\Sigma}(F_y)_{eff}$$
:
 $N_A = (\frac{1}{2} + \frac{\overline{N}}{2}) mg - mg = 0$
 $N_A = (\frac{1}{2} - \frac{\overline{N}}{2}) mg$

FA=Mf NA=1(1- x)mg

$$\mp$$
 ZF = Z(F_x\eff

$$f = \frac{\omega_n}{2\pi} = \frac{2\pi}{2\pi} \sqrt{\frac{2469}{2}}$$

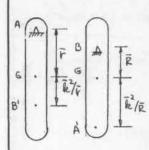
19.167

GIVEN:



COMPOUND PENDULUM WITH ENIFE EDGES AT A AND B A DISTANCE L APART COUNTERWEIGHT D IS ADJUSTED SO THAT THE PERIOD IS THE SAME WHEN EITHER KNIFE EDGE IS USED

SHOW THAT!



FROM PROB 19.52 THE LENGTH OF AN EQUIVALENT SIMPLE PENDULUM IS:

$$L_{A} = \overline{F} + \overline{k}^{2}/F$$
AND
$$L_{B} = \overline{K} + \overline{k}^{2}/\overline{K}$$
BUT $T_{A} = T_{B}$

$$\frac{2\pi}{8}\sqrt{\pi s} = 2\pi\sqrt{\frac{26}{9}}$$
THUS

FOR LA=LB

$$\vec{F} + \frac{\vec{D}_{L}^{2}}{\vec{F}} = \vec{P} + \frac{\vec{D}_{L}^{2}}{\vec{P}}$$

$$\vec{F}^{2} P + \vec{D}^{2} \vec{P} = \vec{F} \vec{D}^{2} + \vec{D}^{2} \vec{F}$$

$$Fe(\bar{r}-\bar{e})=\bar{k}^2(\bar{r}-\bar{e})$$

THUS FR = Te

THUS AG = GA' AND BG = GB'

THAT IS , A = A' AND B=B'

NOTING THAT LA= LB= L

$$\tau = 2\pi \sqrt{\frac{4}{9}}$$

19.168

GIVEN:

400-leg motor supported by Four springs, each spring has a constant of 150 len/m unbalance is 23 g at 100 mm from the axis of rotation

FIND:

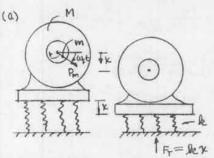
FOR A SPEED OF 800 FPM

(A) THE AMPUTUDE OF THE FLUCTUATING

FORCE TRANSMITTED TO THE FOUNDATION

(b) THE AMPUTUDE OF THE VERTICAL

MOTION OF THE MOTOR



FLOH EQ (19.33)

$$Y_{m} = \frac{1 - (\omega_{\phi}/\omega_{m})^{2}}{1 - (\omega_{\phi}/\omega_{m})^{2}} \tag{1}$$

THUS
$$F_T = \int_{\mathbb{R}} \mathcal{X}_M = \frac{P_m}{1 - \omega_t^2/\omega_n^2}$$
 (2)

&= (4) (150,000 N/m) = 600,000 N/m

$$m_s^t = (sut^t)_s^t [(su)(800/60)]_s^t = 10.18.3_s^t$$

Pm=mrw==(0.023 kg)(0.100 m)(7018 52) Pm=16.14 M

SUBSTITUTING THE ABOVE VALUES INTO GQ. 2

$$F_T = \frac{16.14}{1 - (7018/1500)} = -4.388 \text{ N}$$

 $F_T = 4.39 \text{ N}$

(b)
$$\chi_{m} = Fr/k = \frac{(4.388 \text{ N})}{(600,000 \text{ N/m})}$$

 $\chi_{m} = 0.00731 \times 10^{3} \text{ m}$

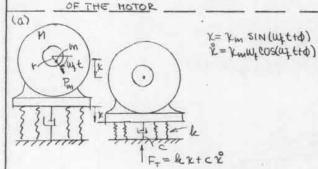
12m=0.00731 mm

19.169 GIVEN:

400- kg HOTOR SUPPORTED BY FOUR SPRINGS EACH WITH &= 150 &N/M. AND A DASHPOT WITH C= 6500 N.S/M UNBALANCE IS 23 9 AT 100 MM FROM THE AXIS OF ROTATION

FIND:

FOR A SPEED OF 800 rpm (A) AMPLITUDE OF THE FLUCTUATING FORCE TRANSMITTED TO THE FOUNDATION (b) AMPLITUDE OF THE YERTICAL HOTION



FT=lex+cx=lxmsin(w++++)+cxmwecoscopt

k= 4 (150,000 N/m)= 600,000 N/m Wn = le/M = 600,000/400 = 1500 5-2 mit = (54th)=(54 (800)/60]= 2018 2-5 Pm=m+w== (0.023kg)(0.100m)(7018)=16.14 N

FROM (2)
$$\gamma_{cm} = \frac{16.14}{\sqrt{(600,000-400)(7018)^2+(6500)^2(7018)}}$$

7-10 x 10-6 m (3)

FROM (1) (FT) = 7.10×10 6 m V(600,000)2+(6500)2(7018) (FT) = 5.75 N

(b) FROM (3)

Xm= 0.00710 mm

NOTE: COTIDARING RESULTS WITH PROB. 19.168 IN WHICH THERE IS NO DASHPOT, THE AMPLITUDE OF THE FORCE HAS INCREASED WHILE THE AMOUTUDE OF VERTICAL MOTION DECREASES.

19.170



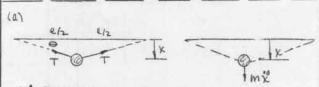
GIVEN!

SHALL HASS IN ATTACHED TO AN ELASTIC CORD OF LENGTH & IN A HORIZONTAL PLANE

TENSION IN THE CORD REMAINS CONSTANT AS THE BALL IS GIVEN A SHALL DISPLACE MENT PERPENDICULAR TO THE CORD AND RELEASED

FIND:

- (a) DIFFERENTIAL EQUATION OF MOTION OF THE BALL
- (b) THE PERIOD OF VIBRATION



FOR SHALL X SIND & TANO = X /(2/2)

$$M_{X}^{*} + \frac{1}{2T} x = 0$$

$$C_{n} = \frac{2\pi}{\omega_{n}} = \frac{2\pi}{\sqrt{4T/m}} = \pi \sqrt{\frac{ml}{T}}$$

19.C1 GIVEN:

PERIOD OF A SIMPLE PENDULUM OF LENGTH & IS, $C_{N} = 2\pi \sqrt{\frac{g}{3}} \left[1 + \left(\frac{1}{2} \right)^{2} C^{2} + \left(\frac{|X3}{2X4} \right)^{2} C^{4} + \left(\frac{|X3X5}{2X4X6} \right)^{2} C^{4} + \cdots \right]$

WHERE C = SIN 10 AND OM IS THE AMPLITUDE

FIND:

THE SUM OF THE SERIES IN BRACKETS USING SUCCESSIVELY 1, 2, 4,8 AND 16 TERMS FOR VALUES OF & M FROM 30°TO 120° USING 30° INCREMENTS. EXPRESS RESULTS WITH FIVE SIGNIFICANT FIGURES

REWRITE GIVEN SERIES INTERHS OF M= 1,2,3

$$AMPLITUDE = \Theta_{m} \qquad C = \frac{1}{2} \sin \Theta_{m}$$

$$LET \quad \gamma = 2\pi \sqrt{\frac{2}{3}} \left[B \right] \qquad \omega_{MBEE} \quad B = \left[1 + \left(\frac{1}{2} \right)^{2} e^{2} + \left(\frac{1 \times 3}{2 \times 7} \right)^{2} e^{4} + \left(\frac{1 \times 3 \times 5}{2 \times 7 \times 6} \right)^{2} e^{6} + \cdots \right]$$

WE MAY COMPUTE & AS FOLLOWS:

$$n = 1: \qquad \mathcal{B} = \left[1 + \left(\frac{2n-1}{2n} \cdot \mathbf{c}\right)^{2}\right]$$

$$n = 2: \qquad \mathcal{B} = \left[1 + \left(\frac{2n-1}{2n} \cdot \mathbf{c}\right)^{2}\right]$$

$$n = 3: \qquad \mathcal{B} = \left[1 + \left(\frac{2n-1}{2n} \cdot \mathbf{c}\right)^{2}\right]$$

AT EACH STEP THE QUANTY ABOVE THE ___ IS THE CHANGE IN B AND IS DENOTED BY "DELTA

AND THE QUARTY
$$\left(\frac{2n}{2n-1}c\right)$$
 is Denoted by FRIOR = $\frac{2n-1}{2n}c$

OUTLINE OF PROGRAM

CALCULATE C= \frac{1}{2} SIN \text{\Theta}_M, FOR \text{\Theta}_M = 30°

CALCULATE B, USING THE ALGORITHM ABOVE

FOR N = 1,2,4,8,16

PRINT B FOR \text{\Theta}_M AND N

REPEAT FOR \text{\Theta}_M = 60°,90° AND 120°

PROGRAM OUTDUT

		CARL CONTRACTOR CONTRACTOR	
Amplitude =	30 degrees	Amplitude =	98 degrees
N	Bracket	N	Bracket
-1	1.01675	1	1.12500
3.	1.01738	2	1.16616
. 4	1.01741	4	1.17704
8	1.01741	8	1.18822
16	1.01741	16	1.18834
Amplitude =	60 degrees	Amplitude =	120 degrees
N	Bracket	N	Bracket
1	1.06250	1	1.19758
1 2	1.87129	P	1.26660
4	1.07311	Ā	1.33146
8	1.07318	8	1.36468
16	1.07318	16	1.37248
***********			1.07548

19.C2

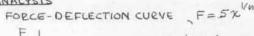
GIVEN:

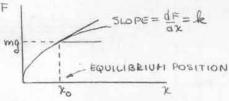
FORCE DEFLECTION EQUATION FOR A CLASS OF SPRING IS F= 5 x1/n WHERE F IS IN NEWTONS AND & IS THE DEFLECTION IN HETERS

FIND:

FOR A BLOCK OF MASS IN SUSPENDED FROM THE SPRING AND IS GIVEN A SHALL DOWNWARD DISPLACEMENT FROM ITS EQUIL BRICH POSITION, THE FREQUENCY OF VIBRATION OF THE BLOCK FOR M = 0,2,0.6 AND 1.0 kg AND FOR VALUES OF IN FROM ITOZ USING 0.7 INCREMENTS

ANALYSIS





$$k = \frac{dF}{dx} = \frac{5}{N} x^{N-1} = \frac{5}{N} x_0^{\frac{1-N}{N}}$$

$$\omega_N = \sqrt{\frac{k}{N}} = \sqrt{\frac{5}{N}} x_0^{\frac{1-N}{2N}}$$

$$f_N = \frac{\omega_N}{2\pi} = \frac{1}{2\pi} \sqrt{\frac{5}{N}} x_0^{\frac{1-N}{2N}} \qquad (1)$$

FOR ANY MY, THE EQUILIBRIUM POINT IS F= mg= 5 x01/4

$$K^{0} = \left(\frac{2}{md}\right)_{N} \tag{5}$$

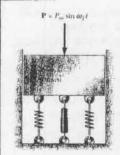
OUTLINE OF PROGRAM

- 1. CALCULATE TO FROM EQ. (2) FOR M=0.2 kg AND N=Z
- 2. SUBSTITUTE YO FROM (2) INTO (1),
- 3. CALCULATE IN AND PRINT IN . M AND IN
- 4. REPEAT STEPS 1-3 FOR N=1.8,1.6,1.4, 1.2 AND 1.0
- 5. PEPEAT STEPS 1-4 FOR M= 0.6 AND 1.0 kg

PROGRAM OUT PUT

n	m (kg)	f (Hz)
2.0	0.20	0.898
1.8	0.20	0.862
1.6	0.20	0.833
1.4	0.20	0.811
1.2	0.20	0.798
1.0		0.796
2.0		
1.8	0.60	0.321
1.6	0.60	0.346
1.4	0.60	0.376
1.2	0.60	0.413
1.0	0.60	0.459
2.0	1.00	0.180
1.8	1.00	0.203
1.6	1.00	0.230
1.4	1.00	0.263
1.2	1.00	0.304
1.0	1.00	0.356
	- Call 5.30.	

19.C3



GIVEN:

MACHINE ELEHENT SUPPORTED BY SPRINGS AND CONNECTED TO A DASHPOT IS SUBJECTED TO A PERIODIC FORCE AS SHOW N

EIND:

FOR FREQUENCY RATIOS WE/WA EQUAL TO 0.8,1.4 AND 2.0 AND FOR DAMPING FACTORS C/CC EQUAL TO O, I, AND Z. THE TRANSMISSIBILITY TIM= FIM / PIM WHERE FIM IS THE MAXIMUM FORCE TRANSMITTED TO THE SHT OF MOITAGMOOR MAXIMUM VALUE Pm.

ANALYSIS

FROM PEOB. 19.148,

$$T_{m} = \frac{P_{m}}{F_{m}} = \sqrt{\frac{1 + [2(c/c_{c})(\omega_{c}/\omega_{n})^{2}]}{[1 - (\omega_{c}/\omega_{n})^{2}]^{2} + [2(c/c_{c})(\omega_{c}/\omega_{n})]^{2}}}$$

OUTLINE OF PROGRAM (USING THE ABOUF PROGRAM)

- I INPUT C/CC = 0
- 2. INDUT UZ/WW= 0.8
- 3. CALCULATE TIM AND PRINT FOR CICE AND WI/WA THE VALUE OF TW
- 4. REPEAT STEPS 2 AND 3 FOR W. IW. = 1.4 AND THEN FOR WILW = Z.O
- 5. REPEAT STEPS ITHROUGH 4 FOR C/C= 1.0 AND THEN FOR C/c= 2.0

PEOGRAM OUTPUT

wflwn	clce	Tm
PREQ. RATIO	DAMPING FACTOR	TRAN. RATIO
0.80	0.0	2.778
1.40	0.0	1.042
2.00	0.0	0.333
0.80	1.0	1.150
1.40	1.0	1.004
2.00	1.0	0.825
0.80	2.0	1.041
1.40	2.0	1.001
2.00	2.0	0.944

19.C4 GIVEN:

15-Leg HOTOR SUPPORTED BY FOUR SPRINGS EACH OF CONSTANT 60 RN/m. UNBALANCE EQUALS 20 g AT 125 mm FROM AKIS OF ROTATION.

FIND:

AMPLITUDE AND ACCREPATION FOR MOTOR SPEEDS OF 1000 TO 2500 FPM USING 100 FPM INCREMENTS

ANALYSIS

FROM EQ.(19.33)
$$\gamma_{m} = \frac{p_{m}/\ell_{c}}{1 - (\omega_{f}/\omega_{n})^{c}}$$
 (1)

WHERE DM = M & WE (SAMPLE PROB. 19.5)

$$k = 4 \times 60,000 \text{ N/m} = 240,000 \text{ N/m}$$

 $k = \frac{1}{240,000} = 240,000 \text{ N/m} = 16000 \text{ S}^2$
 $k = \frac{1}{240,000} = 240,000 \text{ N/m} = 16000 \text{ S}^2$

SUBSTITUTE THE ABOVE VALUES INTO (1)

$$7 = \frac{(2500 \times 10^{-6} M_{\odot}^{2})/(240,000)}{1 - M^{2}/16000} M (2)$$

$$Q_{m} = W_{1}^{2} K_{m} M/S^{2}$$
 (3)

$$W_f = (eph)(2\pi)/60$$
 (4)

OUTLINE OF PROGRAM

- 1. USING EQ. (2) AND NOTING EQ. (4) INPUT AN INITIAL VALUE OF HOTOR SPEED OF 1000 PPM.
- 2. CALCULATE XM
- 3. CALCULATE FROM EQ. (3), am
- a. PEINT rom, Km AND am
- 5. REPENT STEPS I THROUGH 4 FOR HOTO R SPEEDS OF 1100 TO 2500 FPM IN STEPS OF LOO FPM

PROGRAH OUT PUT

TO OBTAIN THE UNITS CORRESPONDING TO THE ANSWERS, BELOW, MULTIPLY EQ. (2) BY LOOD, AND IF THE RESULT (INMM) IS USED IN EQ. (3), DIVIDE IT BY 1000.

SPEED (RPM)	AMP. (mm)	ACCEL. (m/s**2)
1000	0.363	3.98
1100	0.810	10.75
1200	12.615	199.21
1300	-1.219	22.60
1400	-0.652	14.02
1500	-0.474	11.70
1600	-0.388	10.88
1700	-0.337	10.67
1800	-0.303	10.77
1900	-0.280	11.07
2000	-0.262	11.51
2100 -	-0.249	12.05
2200	-0.239	12.66
2300	-0.230	13.35
2400	-0.223	14.10
2500	-0.217	14.90

19.C5 GIVEN:

DASHPOT HAVING A COEFFICIENT OF DAHPING C = 2.5 & N/S IS CONNECTED TO THE MOTOR BASE AND THE GROOND

FIND:

AMPLITUDE AND ACCELERATION FOR HUTOL SPEEDS OF 1000 TO 2500 FPM USING 100 FPM INCREMENTS

ANALYSIS

FROM EQ. (9.52)
$$\gamma_{m} = \frac{Pm}{\sqrt{(l_{e}-M\omega_{\phi}^{2})^{2}+(C\omega_{\phi})^{2}}}$$
 (1)

k= 4×60,000 N/m = 240,000 H/m

SUBSTITUTE INTO (1)

$$\lambda^{-1} = \frac{\sqrt{(540'000 - 12(m^{\frac{1}{2}}))_{2} + (s200)_{2}m^{\frac{1}{2}}}}{5200 \times 10_{2}m^{\frac{1}{2}}}$$
 w

$$\omega_t = (\text{RPM})(z\pi)/60 \tag{4}$$

(3)

OUTLINE OF PROGRAM

- 1. USING & Q. (2) AND NOTING EQ.(4), INPUT AN INITIAL VALUE OF HOTOR SPEED OF 1000 FPM
- 2. CALCULATE Xm (IN METERS)
- 3. CALCULATE FROM 50. (3) THE ACCELERATION
- 4. PRINT FPM, Xm, am
- 5. EFFERT STEPS I THROUGH 4 FOR HOTOR SPEEDS OF 1100 TO 2500 FPM IN INCREMENTS OF 100 FPM

PROGRAM OUT PUT

SEE NOTE AT LEFT

SPEED (RPM)	AMP. (mm)	ACCEL. (m/s**2)
1000	0.1006	1.103
1100	0.1140	1,513
1200	0.1257	1,984
1300	0.1353	2.507
1400	0.1430	3.074
1500	0.1491	3.679
1600	0.1538	4.318
1700	0.1574	4.987
1800	0.1601	5,688
1900	0.1621	6.419
2000	0.1637	7.180
2100	0.1649	7.972
2200	0.1657	8.796
2300	0.1664	9.653
2400	0.1669	10.542
2500	0.1673	11.464

19.C6

5 m

GIVEN:

TRAILER AND LOAD MASS

= 250 kg.
SUPPORTED BY TWO
SPRINGS EACH WITH

R= 10 kN/m

ROAD IS A SINE
CURVE WITH AN
AMPLITUDE OF 40 MM
AND WAVE WRIGTH OF
S M.

FIND:

- (6) AMPLITUDE OF VIBRATION AND MAXIMUM VERTICAL ACCELERATION OF THE TRAILER FOR SPEEDS OF 10 TO 80 LM/N USING S LM/N INCREMENTS
- (b) USING APPROPRIATE SHALLER INCREMENTS

 DETERHINE THE RANGE OF VALUES OF THE SPEED

 FOR WHICH THE TRAILER WILL LEAVE THE GROUND.

(a) In			
- m		FROM E	0 (19.331)
Q 19	= Smsmwgt	1/m=	1-(w+/wn)2
9	2 k=2.10,0 = 20,0	100 N/M	
0	Ts =20000		
+ ;	Sm=0.040V		X
- 2	= 5m	F 5	c 2
		y= Sm	21117

THUS
$$x_{\rm in} = \frac{40 \text{ m/m}}{1 - \left(\frac{2\pi \pi \sigma (1000)}{3600}\right)^2/80} \text{ m/m}$$
 (1)

(b) WHEN Y AND I ARE IN PHASE THEY HAVE THE SAME SIGN (LE Km+)

SST = 0.1226 m = 122.6 mm
THUS WHEN $\chi_{m} > 122.6 + 40 = 162.6$ mm THE
TRAILER WILL LEAVE THE GROUND
WHEN $\gamma_{m} < 122.6 + 40 = 122.6$ when
TRAILER WILL LEAVE THE GROUND WHEN $\gamma_{m} < 122.6 + 40 = -82.6$ mm

19.C6 CONTINUED

SP

OUTLINE OF PROGRAM

(a) INPUT TO EQ. I VALUES OF VELOCITY FROM
10 TO 80 km/h IN 5 km/h INTERVALS
AND PRINT THE RESULTS
PROGRAM OUTPUT

SPEED	(km/h)	AMPLITUDE	(mm)
1.0	0.0	47	,19
1	5.0	60	.85
	0.0	102	.36
	5.0	832	1.11
	0.0	-107	7.88
	5.0	-46	5.20
	0.0	-27	7.84
	5.0	-19	1.19
	0.0	-14	1.25
	5.0	-11	1.09
-	0.0	-1	3.92
	5.0	-	7.36
	0.0		5.19
	5.0		5.29
	0.0	-10	4.57

(b) FROM PART (b) OF THE ANALYSIS WE NOTE THAT IF Xm7167.6 mm or Xm <-826.mm
THE TRAILER WILL LEAVE THE GROWND. FROM THE RESULTS OF PART (a) WE NOTE THAT THIS OCCURS BETWEEN THE VELOCITIES OF 20 km/h and 35 mi/h
RERUN EQ. (1) FOR VELOCITIES OF 20 km/h
TO 35 km/h at intervals of 0.1 km/h and

	ERESULTS		
DEED (RH/N)	AHPUTUDE (MM)	SPEED (Rum/h)	ANDUTUDE (M)
22.2	160.41	26.8	-425.78
22,3	164.89.	26.9	-391.60
22.4	169.65	27.0	-362.54
22.5 LOSE	174.72	27.1	-337.35
22.6	ONTACT180.13	27.2	-315.35
22.7	185.90	27.3	-295,98
22.8	192.09	27.4	-278,79
22.9	198.73	27.5	-263.44
23.0	205.88	27.6	-249,64
23.1	213.60	27.7	-237.17
23.2	221.96	27.8	-225.85
23.3	231.04	27.9	-215.53
23.4	240.94	28.0	-206,08
23.5	251.77	28.1	-197.39
23.6	263.68	28.2	-189.37
23.7	276.82	28.3	-181.96
23.8	291.41	28.4	-175.08
23.9	307.70	28.5	-168.68
24.0	326.00	28.6	-162.72
24.1	346.70	28.7	-157.14
24.2	370.31	28.8	-151.91
24.3	397.49	28.9	-147.00
24.4	429.12	29.0	-142.39
24.5	466.39	29.1	-138.04
24.6	510.94	29.2	-133.94
24.7	565.14	29.3	-130.06
24.8		29.4	-126.38
24.9	632.52 718.53	29.5	-122.90
25.0		29.6	-119.59
25.1	832.14	29.7	-116.45
25.2	989.16	29.8	-113.45
	1220.36	29.9	-110.60
25.3	1594.54		-107.88
25.4	2303.69	30.0	-105.28
25.5	4161.94	30.1	-102.80
25.6	******	30.2	-100.42
25.7	******	30.3	-98.14
25.8	******	30.4	
25.9	******	30.5	-95.96
26.0	******	30.6	-93.86
26.1	******	30.7	-91.85
26.2	-878.93	30.8	-89.91
26.3	-747.58	30.9 LOSES	TACT -88.05
26.4	-650.06		
26.5	-574.79	31.1	-84.54
26.6	-514.95	31.2	-82.88
26.7		31.3	-81.27
20.1	-466.22		